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3D-BASIS-ME:Computer Program for Nonlinear Dynamic Analysis of Seismically Isolated Single and Multiple Structures and Liquid Storage Tanks

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P.C. Tsopelas, M.C. Constantinou and A.M. Reinhorn

Technical Report NCEER-94-0010 April 12, 1994

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P.C. Tsopelas¹, M.C. Constantinou² and A.M. Reinhorn³

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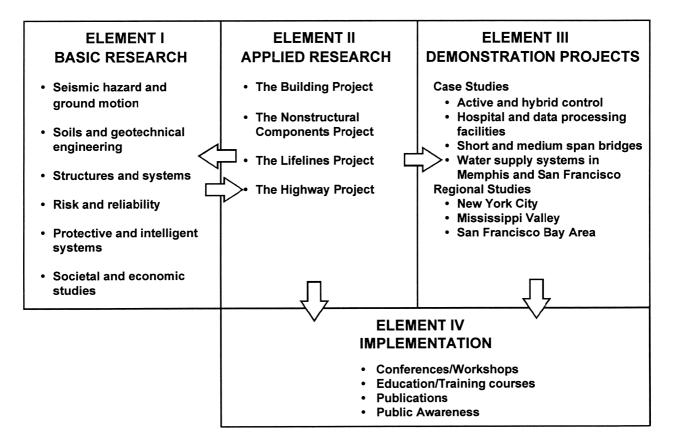
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PREFACE

The National Center for Earthquake Engineering Research (NCEER) was established to expand and disseminate knowledge about earthquakes, improve earthquake-resistant design, and implement seismic hazard mitigation procedures to minimize loss of lives and property. The emphasis is on structures in the eastern and central United States and lifelines throughout the country that are found in zones of low, moderate, and high seismicity.

NCEER's research and implementation plan in years six through ten (1991-1996) comprises four interlocked elements, as shown in the figure below. Element I, Basic Research, is carried out to support projects in the Applied Research area. Element II, Applied Research, is the major focus of work for years six through ten. Element III, Demonstration Projects, have been planned to support Applied Research projects, and will be either case studies or regional studies. Element IV, Implementation, will result from activity in the four Applied Research projects, and from Demonstration Projects.



Research in the **Building Project** focuses on the evaluation and retrofit of buildings in regions of moderate seismicity. Emphasis is on lightly reinforced concrete buildings, steel semi-rigid frames, and masonry walls or infills. The research involves small- and medium-scale shake table tests and full-scale component tests at several institutions. In a parallel effort, analytical models and computer programs are being developed to aid in the prediction of the response of these buildings to various types of ground motion.

Two of the short-term products of the **Building Project** will be a monograph on the evaluation of lightly reinforced concrete buildings and a state-of-the-art report on unreinforced masonry.

The **protective and intelligent systems program** constitutes one of the important areas of research in the **Building Project**. Current tasks include the following:

- 1. Evaluate the performance of full-scale active bracing and active mass dampers already in place in terms of performance, power requirements, maintenance, reliability and cost.
- 2. Compare passive and active control strategies in terms of structural type, degree of effectiveness, cost and long-term reliability.
- 3. Perform fundamental studies of hybrid control.
- 4. Develop and test hybrid control systems.

This is the latest in a series of NCEER technical reports documenting the development of the 3D-BASIS computer program, which is designed for nonlinear dynamic analysis of seismically isolated structures. In this report, the program is extended to include the simulation of the hysteretic behavior of friction pendulum bearings and linear and nonlinear viscous fluid dampers. The effects of overturning moment and vertical ground acceleration on the behavior of sliding bearings are also included.

ABSTRACT

3D-BASIS-ME is a special purpose program for the nonlinear dynamic analysis of seismically isolated multiple buildings and liquid storage tanks. New features of this program, which do not exist in the currently available class of 3D-BASIS programs, are new elements for modeling hysteretic stiffening behavior, for modeling the behavior of spherical sliding isolation systems and for modeling linear and nonlinear viscous fluid dampers. Furthermore, the effects of vertical ground motion and overturning moment on the behavior of sliding bearings have been included.

ACKNOWLEDGEMENTS

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SECTION 1

INTRODUCTION

3D-BASIS-ME represents an enhanced version of program 3D-BASIS-M (Tsopelas et al. 1991), which is an extension of program 3D-BASIS (Nagarajaiah et al. 1989, 1991b and 1993).

The 3DBASIS class of computer programs were developed for the nonlinear dynamic analyses of seismically isolated structures. Program 3D-BASIS was designed to analyze a single superstructure isolated building. Program 3D-BASIS-M was designed to analyze single as well as multiple superstructures with a single isolation basemat. It is suitable for the dynamic analysis of isolated structures which consist of several parts separated by thermal expansion joints. The program may also be used in the dynamic analysis of isolated liquid storage tanks in which the liquid-tank system is modeled by two multi-degree-of freedom systems, representing respectively the impulsive and convective effects.

Program 3D-BASIS-ME maintains the features of programs 3D-BASIS-M with the following enhancements:

- 1. The effects of overturning moment and vertical ground acceleration on the behavior of sliding bearings have been included.
- 2. A new stiffening hysteretic element with bidirectional interaction has been included. This element may be used in modeling the behavior of high damping rubber bearings at large strains.
- 3. A new element capable of modeling the behavior of spherical sliding isolation systems(such as the Friction Pendulum or FPS bearings) has been included.
- 4. A new viscous element has been included that produces output force which is proportional to a power of the velocity of motion of one end of the element with respect to the other end.

This report describes the enhanced program 3D-BASIS-ME and demonstrates its capabilities through a series of example analyses of an isolated liquid storage tank.

SECTION 2

OVERVIEW OF PROGRAM 3D-BASIS

Program 3D-BASIS (Nagarajaiah et al. 1989, Nagarajaiah et al. 1991b) was developed as a public domain special purpose program for the dynamic analysis of base isolated building structures. The basic features of program 3D-BASIS are:

- 1. Elastic superstructure,
- 2. Detailed modeling of the isolation system with spatial distribution of isolation elements,
- 3. Library of isolation elements which include elastomeric and sliding bearing elements with bidirectional interaction effects and rate loading effects,
- 4. Time domain solution algorithm for very stiff differential equations, and
- 5. Bidirectional excitation.

These features are maintained in the extended 3D-BASIS-M program.

2.1 Superstructure Modeling

The superstructure is assumed to remain elastic at all times. Coupled lateral-torsional response is accounted for by maintaining three degrees of freedom per floor, that is two translational and one rotational degrees of freedom. Two options exists in modeling the superstructure :

- a. Shear type representation in which the stiffness matrix of the superstructure is internally constructed by the program. It is assumed that the centers of mass of all floors lie on a common vertical axis, floors are rigid and walls and columns are inextensible.
- b. Full three dimensional representation in which the dynamic characteristics of the superstructure are determined by other computer programs (e.g. ETABS, Wilson et al. 1975) and imported to program 3D-BASIS. In this way, the extensibility of the vertical elements, arbitrary location of centers of mass and floor flexibility may be implicitly accounted for. Still, however, the model for dynamic analysis maintains three degrees of freedom per floor.

In both options, the data needed for dynamic analysis are the mass and the moment of inertia of each floor, frequencies, mode shapes and associated damping ratios for a number of modes. A minimum of three modes of vibration of the superstructure need to be considered.

A recently developed version of 3D-BASIS, called 3D-BASIS-TABS (Nagarajaiah et al. 1993), incorporates the modeling approach of ETABS (Wilson et al. 1975) into 3D-BASIS and allows for the calculation of time histories of superstructure member forces and joint displacements.

2.2 Isolation System Modeling

The isolation system is modeled with spatial distribution and explicit nonlinear force-displacement characteristics of individual isolation devices. The isolation devices are considered rigid in the vertical direction and individual devices are assumed to have negligible resistance to torsion.

Program 3D-BASIS has the following elements for modeling the behavior of an isolation system:

- 1. Linear Elastic element.
- 2. Linear viscous element.
- 3. Hysteretic element for elastomeric bearings and steel dampers.
- 4. Hysteretic element for sliding bearings.

2.2.1 Linear Elastic Element

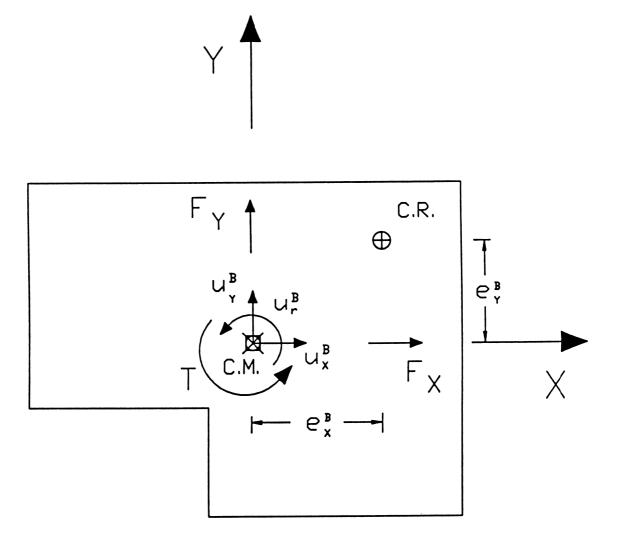
All linear elastic devices of the isolation system are combined in a single element having the combined properties of the devices. These are the translational stiffnesses, K_x and K_y and the rotational stiffness, K_r , with respect to the center of mass of the base. Furthermore, eccentricities e_x^B and e_y^B of the center of resistance of the isolation system to the center of mass of the base need to be specified.

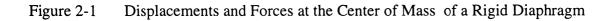
The forces exerted at the center of mass of the base by the linear elastic element are given by the following equations (with reference to Figure 2-1)

$$F_{x} = K_{x}(u_{x}^{B} - e_{y}^{B} u_{r}^{B})$$
(2.1)

$$F_{y} = K_{y}(u_{y}^{B} + e_{x}^{B} u_{r}^{B})$$
(2.2)

$$T = K_r u_r^B + K_y e_x^B u_y^B - K_x e_y^B u_x^B$$
(2.3)





2.2.2 Linear Viscous Element

The linear viscous element is used to simulate the combined viscous properties of the isolation devices. All linear viscous devices are combined in a single viscous element having translational damping coefficients C_x and C_y and rotational damping coefficient C_r . Furthermore, eccentricities $e_x^{\ C}$ and $e_y^{\ C}$ are defined in a manner similar to those of the linear elastic element. The forces exerted by the linear viscous element at the center of mass of the base are given by :

$$F_{x} = C_{x} (\dot{u}_{x}^{B} - e_{y}^{C} \dot{u}_{r}^{B})$$
(2.4)

$$F_{y} = C_{y}(\dot{u}_{y}^{B} + e_{x}^{C} \dot{u}_{r}^{B})$$
(2.5)

$$T = C_r \dot{u}_r^B + C_y e_x^B \dot{u}_y^B - C_x e_y^B \dot{u}_x^B$$
(2.6)

2.2.3 Biaxial Hysteretic Element for Elastomeric Bearings and Steel Dampers

The forces along the orthogonal directions which are mobilized during motion of elastomeric bearings or steel dampers are described by :

$$F_{x} = \alpha \frac{F^{y}}{Y} U_{x} + (1 - \alpha) F^{y} Z_{x}, \qquad F_{y} = \alpha \frac{F^{y}}{Y} U_{y} + (1 - \alpha) F^{y} Z_{y}$$
(2.7)

in which, α is the post-yielding to pre-yielding stiffness ratio, F^{y} is the yield force and Y is the yield displacement, as illustrated in Figure 2-2. Z_x and Z_y are dimensionless variables governed by the following system of differential equations which was proposed by Park et al. 1986 :

$$\begin{cases} \dot{Z}_{x} & Y \\ \dot{Z}_{y} & Y \end{cases} = \begin{cases} A & \dot{U}_{x} \\ A & \dot{U}_{y} \end{cases} - \begin{pmatrix} Z_{x}^{2}(\gamma Sgn(\dot{U}_{x}Z_{x}) + \beta) & Z_{x}Z_{y}(\gamma Sgn(\dot{U}_{y}Z_{y}) + \beta) \\ Z_{x}Z_{y}(\gamma Sgn(\dot{U}_{x}Z_{x}) + \beta) & Z_{y}^{2}(\gamma Sgn(\dot{U}_{y}Z_{y}) + \beta) \end{pmatrix} \begin{cases} \dot{U}_{x} \\ \dot{U}_{y} \end{cases}$$
(2.8)

in which A, γ and β are dimensionless quantities that control the shape of the hysteresis loop. Furthermore, U_x , U_y and \dot{U}_x , \dot{U}_y represent the displacements and velocities that occur at the isolation element.

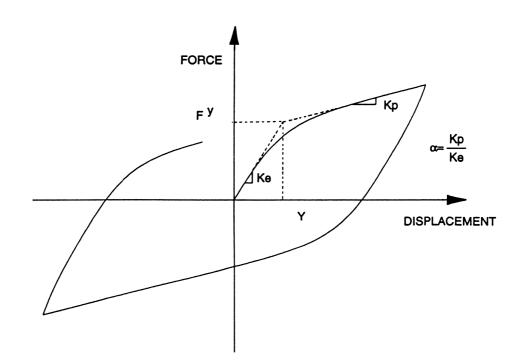


Figure 2-2 Hysteretic Element for Elastomeric Bearings and Steel Dampers. For Elastoplastic Behavior $\alpha = 0$.

Constantinou et al. 1990 have shown that when motion commences and displacements exceed the yield displacement, Equation 2.8 has the following solution provided that $A/(\beta + \gamma) = 1$:

$$Z_{\rm r} = \cos\theta, \qquad Z_{\rm v} = \sin\theta \tag{2.9}$$

where θ is the angle specifying the instantaneous direction of motion

$$\theta = \tan^{-1}(\dot{U}_{y}/\dot{U}_{x}) \tag{2.10}$$

Equations 2.7 and 2.9 indicate that the interaction curve of the element is circular. To demonstrate this, consider motion along an angle θ with respect to the X-axis so that $U_x = U \cos \theta$ and $U_y = U \sin \theta$. By substituting Equations 2.9 into Equations 2.7, it is easily shown that the resultant of mobilized forces is independent of θ and given by

$$F = (F_{x}^{2} + F_{y}^{2})^{1/2} = \left\{ (1 - \alpha)^{2} F^{y^{2}} + \alpha^{2} \frac{F^{y^{2}}}{Y^{2}} U^{2} + 2\alpha (1 - \alpha) \frac{F^{y^{2}} U}{Y} \right\}^{1/2}$$
(2.11)

Equation 2.11 clearly describes a circle. At the lower limit of inelastic behavior, i.e. U = Y, Equation 2.11 reduces to $F = F^y$ which demonstrates that the yield force of the element is equal to F^y in all directions. This desirable property is possible only when $A/(\beta + \gamma) = 1$ (Constantinou et al. 1990). In particular, A = 1 and $\beta = 0.1$ and $\gamma = 0.9$ are suggested.

This element may be used in modeling the behavior of low damping rubber bearings, high damping rubber bearings in the range of strain prior to stiffening and lead-rubber bearings.

2.2.4 Biaxial Element for Sliding Bearings

For flat sliding bearings, the mobilized forces are described by the equations (Constantinou et al. 1990, Mokha et al. 1993)

$$F_x = \mu_s N Z_x, \quad F_y = \mu_s N Z_y \tag{2.12}$$

in which N is the vertical load carried by the bearing and μ_s is the coefficient of sliding friction which depends on the bearing pressure, direction of motion as specified by angle θ (Equation 2.10) and the instantaneous velocity of sliding \dot{U}

$$\dot{U} = (\dot{U}_x^2 + \dot{U}_y^2)^{1/2}$$
(2.13)

The conditions of separation and reattachment and biaxial interaction are accounted for by variables Z_x and Z_y in Equation 2.8.

The coefficient of sliding friction is modeled by the following Equation suggested by Constantinou et al. 1990 :

$$\mu_{s} = f_{\max} - (f_{\max} - f_{\min}) \exp(-a | U |)$$
(2.14)

in which, f_{max} is the maximum value of the coefficient of friction and f_{min} is the minimum (at $\dot{U} = 0$) value of the coefficient of friction as shown in Figure 2-3. Furthermore, a is a parameter which controls the variation of the coefficient of friction with velocity. Values of parameters f_{max} , f_{min} and a for interfaces used in sliding bearings have been reported in Constantinou et al. 1990 and Mokha et al. 1991. In general, parameters f_{max} , f_{min} and a are functions of bearing pressure and angle θ , though the dependency on θ is usually not important.

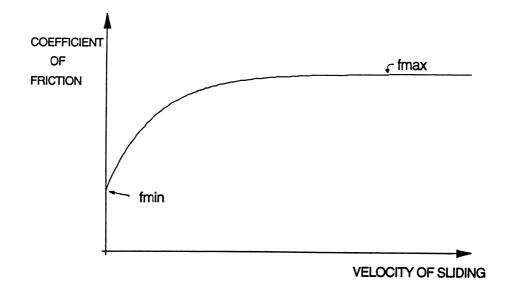


Figure 2-3 Model of Coefficient of friction in Program 3D-BASIS. The Model Collapses to the Coulomb Model when $f_{max} = f_{min}$.

2.2.5 Uniaxial Model for Elastomeric Bearings, Steel Dampers and Sliding Bearings

The biaxial interaction achieved in the models of Equations 2.7 to 2.10 and 2.12 to 2.14 may be neglected by replacing the off-diagonal elements in Equation 2.8 by zeroes. This results in two uniaxial independent elements having either sliding or smooth hysteretic behavior in the two orthogonal directions.

SECTION 3

PROGRAM 3D-BASIS-M

Program 3D-BASIS-M (Tsopelas et al. 1991) is an extension of program 3D-BASIS for the dynamic analysis of base isolated structures with multiple building superstructures on a common isolation system. This section concentrates on the development of the equations of motion of the multiple superstructure isolated system and the method of solution.

3.1 Superstructure and Isolation System Configuration

The model used in the analysis of the system (superstructure and isolation system) has been discussed in Section 2. The same options available in 3D-BASIS were adopted in program 3D-BASIS-M. The basic assumptions considered in modeling the system are :

- 1. Each floor has three degrees of freedom. These are the X and Y translations and rotation about the center of mass of each floor. These degrees of freedom are attached to the center of mass of each floor.
- 2. There exists a rigid slab at the level that connects all the isolation elements. The three degrees of freedom at the base are attached to the center of mass of the base.
- Since three degrees of freedom per floor are required in the three-dimensional representation of the superstructure, the number of modes required for modal reduction is always a multiple of three. The minimum number of modes required is three.

The degrees of freedom of the floors and base and the configuration of a multiple building isolated structure are illustrated in Figures 3-1 and 3-2. A global reference axis is attached to the center of mass of the base (Figure 3-1). The coordinates of the center of mass of each floor of each superstructure are measured with respect to the reference axis. The center of resistance of each floor is located at distances e_{xj} and e_{yj} (eccentricities) with respect to the center of mass of the floor (Figure 3-2). All degrees of freedom (two translations and one rotation at each floor and base) are attached to the centers of mass as shown in Figures 3-1 and 3-2. Displacements and rotations of each floor are measured with respect to the base, whereas those of the base are measured with respect to the ground as shown in Figure 3-3.

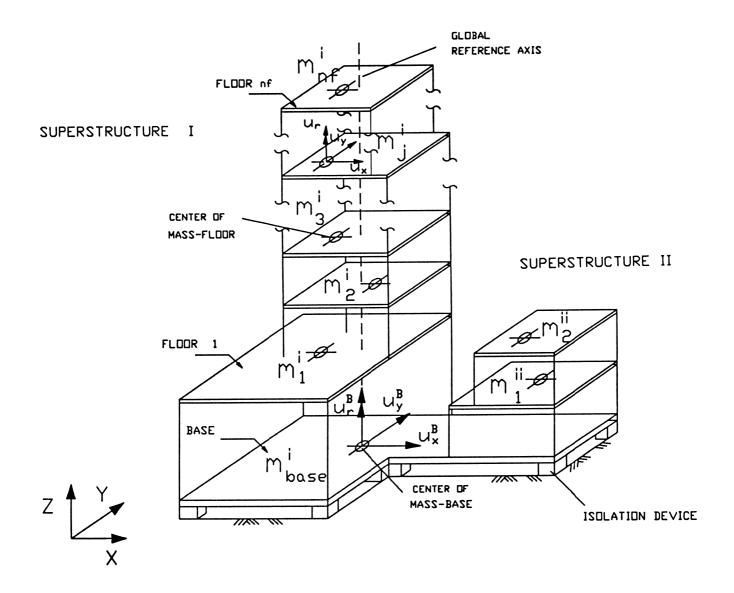


Figure 3-1 Multiple Building Isolated Structure.

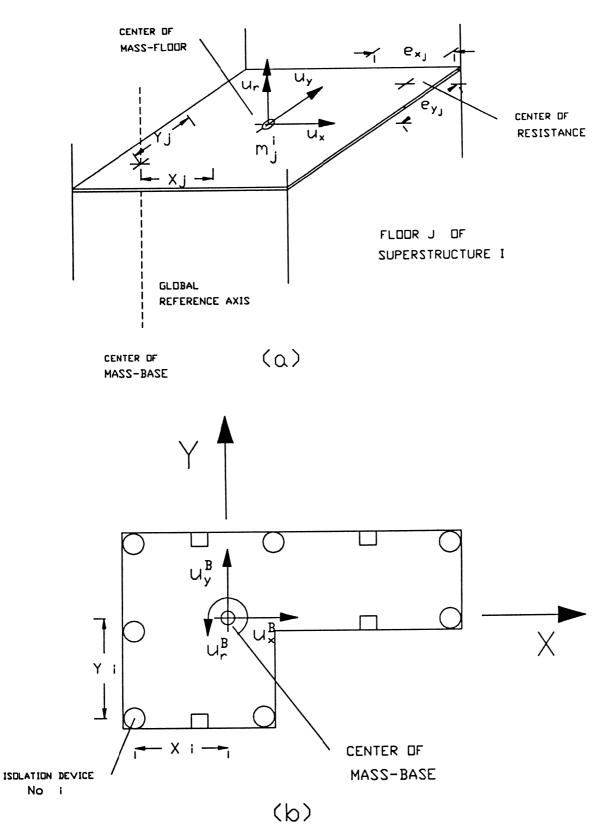


Figure 3-2 Degrees of Freedom and Details of a Typical Floor and Base : (a) Isometric View of Floor j of Superstructure i; (b) Plan of Base.

As in program 3D-BASIS, the extended 3D-BASIS-M program has two options for the representation of the superstructure. In the first option, each superstructure is represented by a shear building representation. In this representation, the stiffness characteristics of each story of each superstructure are represented by the story translational stiffnesses, rotational stiffness and eccentricities of the story center of resistance with respect to the center of mass of the floor (see Figure 3-2). Furthermore, and only for the shear type representation, it is assumed that the centers of mass of all floors of each superstructure lie on a common vertical axis. This common vertical axis is located at distances X_j and Y_j with respect to the global reference axis which is located at the center of mass of the base (see Figures 3-1 and 3-2). Of course, the shear representation implies that the floors and the base are rigid and all vertical elements are inextensible.

In the second option, all restrictions of the shear type representation other than that of rigid floor and base are relaxed. A complete three dimensional model of each superstructure is developed externally to program 3D-BASIS-M using appropriate computer programs (e.g. ETABS, Wilson et al. 1975). The dynamic characteristics of each superstructure in terms of frequencies and mode shapes are extracted and imported to program 3D-BASIS-M.

Modeling of the isolation system in program 3D-BASIS-M is identical to that in program 3D-BASIS. Spatial distribution and biaxial interaction effects are included.

3.2 Analytical Model and Equations of Motion

A multiple building base isolated structure and the coordinates (displacements) used in the basic formulation is shown in Figure 3-3. \mathbf{u}_{j}^{i} is the relative displacement vector of the center of mass of floor (j) of superstructure (i) with respect to the base, \mathbf{u}_{b} is the relative displacement vector of the center of the center of the center of the base with respect to the ground and \mathbf{u}_{g} is the ground displacement vector. Each one of the these vectors has translational X, Y components and rotation about the vertical axis.

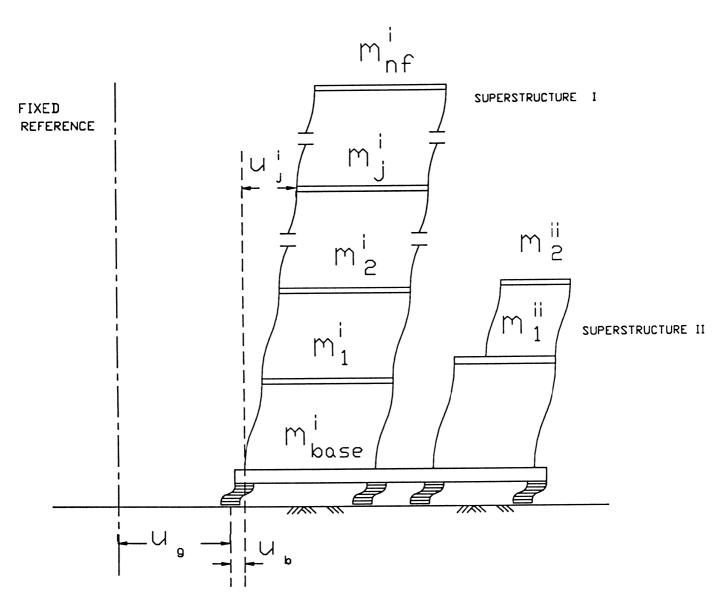


Figure 3-3 Displacement Coordinates of Isolated Structure.

The equations of motion of the part of the structure above the base (superstructures) are :

$$\mathbf{M}_{N_b \times N_b} \ddot{\mathbf{u}}_{N_b \times 1} + \mathbf{C}_{N_b \times N_b} \dot{\mathbf{u}}_{N_b \times 1} + \mathbf{K}_{N_b \times N_b} \mathbf{u}_{N_b \times 1} = -\mathbf{M}_{N_b \times N_b} \mathbf{R}_{N_b \times 3} \{ \ddot{\mathbf{u}}_b + \ddot{\mathbf{u}}_g \}_{3 \times 1}$$
(3.1)

In the above equations **M**, **C** and **K** are the combined mass, damping and stiffness matrices of the superstructure buildings, **u** is the combined displacement vector relative to the base and **R** is a transformation matrix which transfers the base $(\ddot{\mathbf{u}}_b)$ and ground $(\ddot{\mathbf{u}}_g)$ acceleration vectors from the center of mass of the base to the center of mass of each floor of each superstructure building. The

subscripts in Equation 3.1 denote the dimension of the matrices. N_b is the number of degrees of freedom in the part above the base. It is equal to the total number of degrees of freedom minus the three degrees of freedom of the base. In extended form, Equations 3.1 are expressed as

$$\begin{pmatrix} \mathbf{m}^{1} & 0 & 0 & 0 & 0 \\ 0 & \dots & 0 & 0 & 0 \\ 0 & 0 & \mathbf{m}^{i} & 0 & 0 \\ 0 & 0 & 0 & \dots & 0 \\ 0 & 0 & 0 & 0 & \mathbf{m}^{ns} \end{pmatrix} \begin{bmatrix} \mathbf{\ddot{u}}^{1} \\ \dots \\ \mathbf{\ddot{u}}^{i} \\ \dots \\ \mathbf{\ddot{u}}^{ns} \end{bmatrix} + \begin{pmatrix} \mathbf{c}^{1} & 0 & 0 & 0 & 0 \\ 0 & \dots & 0 & 0 & 0 \\ 0 & 0 & \mathbf{c}^{i} & 0 & 0 \\ 0 & 0 & 0 & \mathbf{c}^{ns} \end{pmatrix} \begin{bmatrix} \mathbf{\dot{u}}^{1} \\ \dots \\ \mathbf{\ddot{u}}^{ns} \end{bmatrix}$$

$$+ \begin{pmatrix} \mathbf{k}^{1} & 0 & 0 & 0 & 0 \\ 0 & \dots & 0 & 0 & \mathbf{c}^{ns} \\ 0 & 0 & \mathbf{c}^{ns} \end{pmatrix} \begin{bmatrix} \mathbf{\dot{u}}^{1} \\ \dots \\ \mathbf{\ddot{u}}^{ns} \end{bmatrix}$$

$$+ \begin{pmatrix} \mathbf{k}^{1} & 0 & 0 & 0 & 0 \\ 0 & \dots & 0 & 0 & \mathbf{c}^{ns} \end{pmatrix} \begin{bmatrix} \mathbf{\dot{u}}^{1} \\ \dots \\ \mathbf{\dot{u}}^{ns} \end{bmatrix}$$

$$+ \begin{pmatrix} \mathbf{m}^{1} & 0 & 0 & 0 & 0 \\ 0 & \dots & 0 & 0 & 0 \\ 0 & 0 & \mathbf{m}^{i} & 0 & 0 \\ 0 & 0 & 0 & \dots & 0 \\ 0 & 0 & 0 & 0 & \mathbf{m}^{ns} \end{pmatrix} \begin{bmatrix} \mathbf{\dot{u}}^{1} \\ \dots \\ \mathbf{\dot{r}}^{i} \\ \dots \\ \mathbf{\dot{r}}^{ns} \end{bmatrix} \begin{bmatrix} \mathbf{\ddot{u}}_{b} & + & \mathbf{\ddot{u}}_{g} \end{bmatrix}$$

In Equations 3.2, \mathbf{m}^{i} , \mathbf{c}^{i} , and \mathbf{k}^{i} and the mass, damping and stiffness matrices of superstructure (i). These matrices are of dimensions $3nf^{i}$ where nf^{i} is the number of floors in superstructure (i). It should be noted that matrices \mathbf{m}^{i} are diagonal and contain the mass and mass moment of inertia of each floor. The range of index (i) varies between one and ns, the number of superstructures. \mathbf{u}^{i} is the displacement vector of superstructure (i) relative to the base. Further, \mathbf{r}^{i} is the transformation matrix which transfers the base and ground acceleration vectors from the center of mass of the base to the center of mass of each floor of superstructure (i) :

(3.2)

(3.3)

$$\mathbf{r}^{i} = \begin{pmatrix} \mathbf{R}_{nf^{i}} \\ \vdots \\ \mathbf{R}_{j^{i}} \\ \vdots \\ \mathbf{R}_{1} \end{pmatrix}$$

where

$$\mathbf{R}_{j} = \begin{pmatrix} 1 & 0 & -\mathbf{Y}_{j} \\ 0 & 1 & \mathbf{X}_{j} \\ 0 & 0 & 1 \end{pmatrix}$$
(3.4)

in which \mathbf{X}_{j} , \mathbf{Y}_{j} are the distances to the center of mass of floor (j) of superstructure (i) from the center of mass of the base (see Figure 3-2).

The equilibrium equation of dynamic equilibrium of the base is:

$$\mathbf{R}_{3\times N_{b}}^{\prime}\mathbf{M}_{N_{b}\times N_{b}}\{\ddot{\mathbf{u}}_{N_{b}\times 1}+\mathbf{R}_{N_{b}\times 3}\{\ddot{\mathbf{u}}_{b}+\ddot{\mathbf{u}}_{g}\}_{3\times 1}\}+\mathbf{M}_{b_{3\times 3}}\{\ddot{\mathbf{u}}_{b}+\ddot{\mathbf{u}}_{g}\}_{3\times 1}+\mathbf{C}_{b_{3\times 3}}\{\dot{\mathbf{u}}_{b}\}_{3\times 1}+\mathbf{K}_{b_{3\times 3}}\{\mathbf{u}_{b}\}_{3\times 1}+\{\mathbf{f}_{N}\}_{3\times 1}=0$$
(3.5)

in which \mathbf{M}_{b} is the mass matrix of the base, \mathbf{C}_{b} is the resultant damping matrix of viscous elements of the isolation system, \mathbf{K}_{b} is the resultant stiffness matrix of elastic elements of the isolation system at the center of mass of the base and \mathbf{f}_{N} is a vector containing the forces mobilized in the nonlinear elements of the isolation system.

Employing modal reduction :

$$\mathbf{u}_{3nf^{i}}^{i} = \Phi_{3nf^{i} \times ne^{i}}^{i} \mathbf{Y}_{ne^{i} \times 1}^{i}$$
(3.6)

where Φ^i is the orthonormal modal matrix relative to the mass matrix of superstructure (i), \mathbf{Y}^i is the modal displacement vector of superstructure (i) relative to the base and neⁱ is the number of eigenvectors of superstructure (i) retained in the analysis.

Combining Equations 3.2 to 3.6, the following equation is derived

$$\begin{pmatrix} \mathbf{I} & \Phi^{T}\mathbf{M}\mathbf{R} \\ \mathbf{R}^{T}\mathbf{M}\Phi & \mathbf{R}^{T}\mathbf{M}\mathbf{R} + \mathbf{M}_{b} \end{pmatrix}_{(M_{b}+3)\times(M_{b}+3)} \begin{cases} \ddot{\mathbf{Y}} \\ \ddot{\mathbf{u}}_{b} \end{cases}_{(M_{b}+3)\times1} + \begin{pmatrix} 2\xi\omega & \mathbf{0} \\ \mathbf{0} & \mathbf{C}_{b} \end{pmatrix}_{(M_{b}+3)\times(M_{b}+3)} \begin{cases} \dot{\mathbf{Y}} \\ \dot{\mathbf{u}}_{b} \end{cases}_{(M_{b}+3)\times1} + \begin{pmatrix} \mathbf{0} \\ \mathbf{f}_{N} \end{pmatrix}_{(M_{b}+3)\times1} = - \begin{pmatrix} \Phi^{T} & \mathbf{M} & \mathbf{R} \\ \mathbf{R}^{T} & \mathbf{M} & \mathbf{R} \end{pmatrix}_{(M_{b}+3)\times3} \{ \ddot{\mathbf{u}}_{g} \}_{3\times1}$$

$$(3.7)$$

in which M_b is the total number of eigenvectors for all superstructures retained in the analysis, and ξ and ω are the matrices of modal damping and eigenvalues for all eigenvectors of all superstructures, respectively. Furthermore, I denotes an identity matrix and **0** denotes a null matrix.

Equation 3.7 may be written as :

$$\tilde{M}\tilde{\tilde{y}}_t + \tilde{C}\tilde{\tilde{y}}_t + \tilde{K}\tilde{y}_t + f_t = \tilde{P}_t$$
(3.8)

in which subscript t denotes that the equation is valid at time t. Extending Equation 3.8 to time $t+\Delta t$, where Δt is the time step, we have

$$\tilde{M}\ddot{\tilde{y}}_{t+\Delta t} + \tilde{C}\tilde{\tilde{y}}_{t+\Delta t} + \tilde{K}\tilde{y}_{t+\Delta t} + f_{t+\Delta t} = \tilde{P}_{t+\Delta t}$$
(3.9)

Taking the difference between Equations 3.8 and 3.9 gives the incremental equation of equilibrium $\tilde{M}\Delta\ddot{\tilde{y}}_{t+\Delta t} + \tilde{C}\Delta\dot{\tilde{y}}_{t+\Delta t} + \tilde{K}\Delta\tilde{y}_{t+\Delta t} + \Delta f_{t+\Delta t} = \tilde{P}_{t+\Delta t} - \tilde{M}\ddot{\tilde{y}}_{t} - \tilde{C}\dot{\tilde{y}}_{t} - \tilde{K}\tilde{Y}_{t} - f_{t}$ (3.10)

Accordingly, the response of the multiple building superstructure and base is represented by the modal coordinate vectors \ddot{y}_t , \dot{y}_t and \tilde{y}_t .

3.3 Method of Solution

The modified Newton-Raphson solution procedure with tangent stiffness representation is widely used in nonlinear dynamic analysis programs and rapidly converges to the correct solution when the nonlinearities of the system are mild. However the method fails to converge when the non-linearities are severe (Stricklin et al. 1971, Stricklin et al. 1977). Additional studies by Nagarajaiah et al. 1989 reported the failure of this method to converge when nonlinearities stemmed from sliding isolation devices.

The pseudo-force method is used in the present study as originally adopted in the program 3D-BASIS by Nagarajaiah et al. 1989. This method has been used for nonlinear dynamic analysis of shells by Stricklin et al. 1971 and by Darbre and Wolf 1988 for soil structure interaction problems. More details and the advantages of this method in the analysis of base isolated structures have been presented by Nagarajaiah et al. 1989, 1990, 1991a and 1991b. In the pseudo-force method, the incremental nonlinear force vector $\Delta f_{t+\Delta t}$ in Equation 3.10 is unknown. It is, thus brought on the right hand side of Equation 3.10 and treated as pseudo-force vector.

3.4 Solution Algorithm

The differential equations of motion are integrated in the incremental form of Equations 3.10. The solution involves two stages :

- Solution of the equations of motion using the unconditionally stable (for both positive and negative tangent stiffness - Cheng 1988) Newmark's constant-average-acceleration method (Newmark 1959).
- (ii) Solution of the differential equations governing the nonlinear behavior of the isolation elements using an unconditionally stable semi-implicit Runge-Kutta method suitable for stiff differential equations (Rosenbrock 1964). The solution algorithm of the pseudo force method with iteration is presented in Table 3-I.

3.4.2 Varying Time Step for Accuracy

The solution algorithm has the option of using a constant time step or variable time step. The time step is reduced from Δt_{slip} (time step at high velocity) to a fraction of its value at low velocities to maintain accuracy in sliding isolated structures. The time step is reduced based on the magnitude of the resultant velocity at the center of mass of the base :

$$\Delta t_{stick} = \Delta t_{slip} \left[1 - \exp\left(-\frac{\dot{u}^2}{B}\right) \right]$$
(3.11)

in which, \dot{u} is the resultant velocity at the center of mass of the base, Δt_{stick} is the reduced time step when the base velocity is low ($\Delta t_{slip} > \Delta t_{slick} > \Delta t_{slip}/nl$, nl is an integer to introduce the desired reduction) and B is a constant to define the range of velocity over which the reduction takes place. It is important to note that the reduction in the time step is not continuous as indicated by Equation 3.11 but rather at discrete intervals of velocity. This procedure is adopted for computational efficiency.

TABLE 3-I SOLUTION ALGORITHM

A.Initial Conditions:

1. Form stiffness matrix $~\tilde{K},$ mass matrix $~\tilde{M}$, and damping matrix $\tilde{C}.$ Initialize $~\tilde{u}_{_0},~~\tilde{u}_{_0}$ and $~\tilde{u}_{_0}.$

2. Select time step Δt , set parameters $\delta = 0.25$ and $\theta = 0.5$, and calculate the integration constants:

$$a_1 = \frac{1}{\delta(\Delta t)^2}; \quad a_2 = \frac{1}{\delta\Delta t}; \quad a_3 = \frac{1}{2\delta}; \quad a_4 = \frac{\theta}{\delta\Delta t}; \quad a_5 = \frac{\theta}{\delta}; \quad a_6 = \Delta t(\frac{\theta}{2\delta} - 1)$$

3. Form the effective stiffness matrix $\mathbf{K}^* = a_1 \mathbf{\tilde{M}} + a_4 \mathbf{\tilde{C}} + \mathbf{\tilde{K}}$ 4. Triangularize \mathbf{K}^* using Gaussian elimination (only if the time step is different from the previous step).

B.Iteration at each time step:

1. Assume the pseudo-force $\Delta f_{t+\Delta t}^i = 0$ in iteration i = 1.

2. Calculate the effective load vector at time $t + \Delta t$:

$$\mathbf{P}_{t+\Delta t} = \Delta \mathbf{P}_{t+\Delta t} - \Delta f_{t+\Delta t}^{t} + \mathbf{M}(a_{2}\tilde{\mathbf{u}}_{t} + a_{3}\tilde{\mathbf{u}}_{t}) + \tilde{\mathbf{c}}(a_{5}\tilde{\mathbf{u}}_{t} + a_{6}\tilde{\mathbf{u}}_{t})$$
$$\Delta \tilde{\mathbf{P}}_{t+\Delta t} = \tilde{\mathbf{P}}_{t+\Delta t} - (\tilde{\mathbf{M}}\tilde{\mathbf{u}}_{t} + \tilde{\mathbf{C}}\tilde{\mathbf{u}}_{t} + \tilde{\mathbf{K}}\tilde{\mathbf{u}}_{t} + \mathbf{f}_{t})$$

3. Solve for displacements at time $t + \Delta t$: $\mathbf{K}^* \Delta \mathbf{u}_{t+\Delta t}^i = \mathbf{P}_{t+\Delta t}^*$

4. Update the state of motion at time $t + \Delta t$:

$$\ddot{\mathbf{u}}_{t+\Delta t} = \ddot{\mathbf{u}}_{t} + a_1 \Delta \tilde{\mathbf{u}}_{t+\Delta t}^i - a_2 \dot{\tilde{\mathbf{u}}}_{t} - a_3 \ddot{\tilde{\mathbf{u}}}_{t}; \quad \dot{\tilde{\mathbf{u}}}_{t+\Delta t} = \dot{\tilde{\mathbf{u}}}_{t} + a_4 \Delta \tilde{\mathbf{u}}_{t+\Delta t}^i - a_5 \dot{\tilde{\mathbf{u}}}_{t} - a_6 \ddot{\tilde{\mathbf{u}}}_{t}; \quad \tilde{\mathbf{u}}_{t+\Delta t} = \tilde{\mathbf{u}}_{t} + \Delta \tilde{\mathbf{u}}_{t+\Delta t}^i$$

5. Compute the state of motion at each bearing and solve for the nonlinear force at each bearing using semi-implicit Runge-Kutta method.

6. Compute the resultant nonlinear force vector at the center of mass of the base $\Delta f_{t+\Delta t}^{i+1}$.

7. Compute

$$Error = \frac{\|\Delta f_{t+\Delta t}^{i+1} - \Delta f_{t+\Delta t}^{i}\|}{Ref. \operatorname{Max} .Moment}$$

Where $\|.\|$ is the euclidean norm 8. If Error \geq tolerance, further iteration is needed, iterate starting form step B-1 and use $\Delta f_{t+\Delta t}^{i+1}$ as the pseudo-force and the state of motion at time t, $\tilde{\mathbf{u}}_{t}$, $\dot{\tilde{\mathbf{u}}}_{t}$ and $\ddot{\tilde{\mathbf{u}}}_{t}$.

9. If Error \leq tolerance, no further iteration is needed, update the nonlinear force vector:

$$\mathbf{f}_{t+\Delta t} = \mathbf{f}_t + \Delta \mathbf{f}_{t+\Delta t}^{t+1}$$

reset time step if necessary, go to step B-1 if the time step is not reset or go to A-2 if the time step is reset.

SECTION 4

ENHANCEMENTS IN PROGRAM 3D-BASIS-ME

4.1 Stiffening Biaxial Hysteretic Element

The element is appropriate for modeling the behavior of high damping rubber bearings. Typically, these bearings exhibit higher stiffness at large strains. The element is formed by combining the elastoplastic version ($\alpha = 0$) of the biaxial hysteretic element of Section 2.2.3 and a stiffening bilinear spring.

The resultant force, F, in the stiffening bilinear spring is described by

$$F = \begin{cases} K_{1}U & , & U \leq D_{1} \\ \frac{(K_{2} - K_{1})}{(D_{2} - D_{1})^{2}} gn(U) + K_{1}U & , & D_{1} < U \leq D_{2} \\ \frac{(K_{1} - K_{2})(D_{1} + D_{2})}{2} gn(U) + K_{2}U & , & U > D_{2} \end{cases}$$
(4.1)

where K_1 is the tangent stiffness which is mobilized for displacements less than the limit D_1 and K_2 is the higher tangent stiffness which is mobilized for displacements larger than the limit D_2 , as illustrated in Figure 4-1. Furthermore, U is the resultant displacement

$$U = (U_x^2 + U_y^2)^{1/2}$$
(4.2)

The components of the force F in the two orthogonal directions are

$$F_{xs} = F \cos \theta, \qquad F_{ys} = F \sin \theta$$
(4.3)

where

$$\theta = \theta^*$$
 when U_x , $U_y > 0$ (4.4*a*)

$$\theta = \theta^* + \pi/2 \qquad \text{when} \qquad U_x < 0, \quad U_y > 0 \tag{4.4b}$$

 $\theta = \theta^* + \pi$ when U_x , $U_y < 0$ (4.4c)

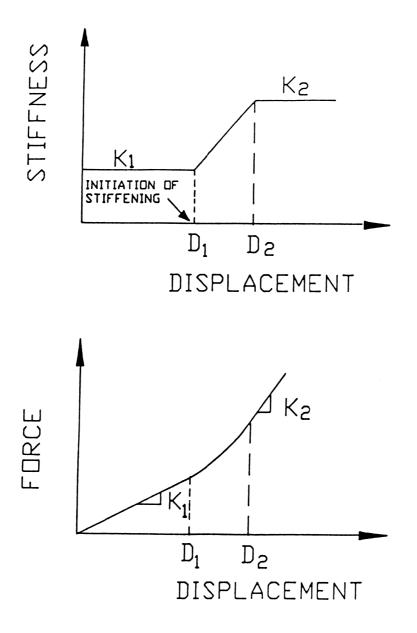


Figure 4-1 Model of Stiffening Bilinear Spring.

$\theta = -\theta^*$	when	$U_x > 0,$	$U_y < 0$	(4.4 <i>d</i>)
$\theta = \pi/2$	and $U = U_y$	when	$U_x = 0$	(4.4 <i>e</i>)
$\theta = 0$	and $U = U_x$	when	$U_y = 0$	(4.4 <i>f</i>)
and				

$$\theta^* = \tan^{-1} \left(\frac{|U_y|}{|U_x|} \right)$$
(4.5)

The complete model consits of the combination of components given by Equations 2.7, with $\alpha = 0$ and $F^y=Q$, and 4.3 :

$$F_x = Q Z_x + F_{xs}, \qquad F_y = Q Z_y + F_{ys}$$
 (4.6)

These relations are depicted graphically in Figure 4-2. The uniaxial version of the model is recovered by replacing the off-diagonal elements in Equation 2.8 by zeroes and by enforcing Equations 4.4e or 4.4f.

To illustrate the capabilities of this model, we consider the modeling of the behavior of a high damping rubber bearing based on data from testing of scaled specimens. The test data on the scaled specimens, obtained at pressure of 10 MPa and frequency of 0.5 Hz, are: tangent shear modulus for shear strain $\gamma = 0.5$ to 1.0 G=0.8 MPa, equivalent damping ratio (per 1991 UBC) $\beta = 0.10$ at shear strain $\gamma = 1.0$, displacement limits $D_1=1.2T$ and $D_2=1.3T$, yield displacement Y=0.07T, where T=total rubber thickness. Furthermore, the tangent stiffness beyond the displacement limit D_2 is $K_2=2K_1$.

The bearing to be modeled is made of the same material and has the same shape factor as the tested scaled specimens. The bearing has bonded rubber diameter D=500 mm and total rubber thickness T=150 mm.

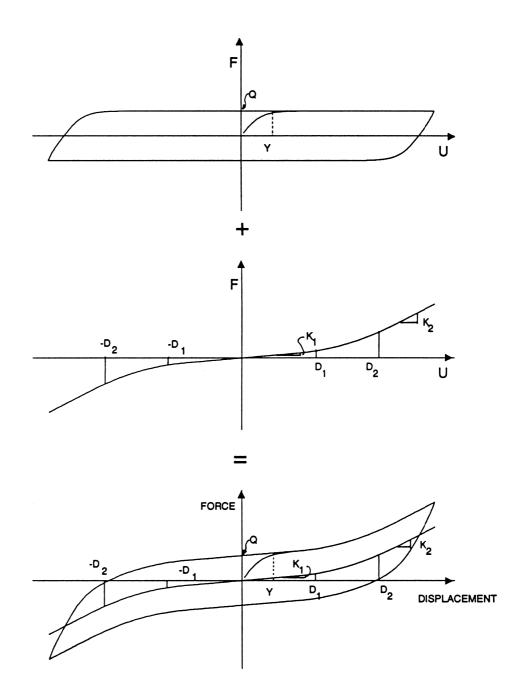


Figure 4-2 Stiffening Hysteretic Model in Program 3D-BASIS-ME.

The tangent stiffness K_1 (see Figure 4-2) is determined from

$$K_1 = \frac{GA}{T} \tag{4.7}$$

where $A = \pi D^2/4$ is the bonded rubber area. The force Q (see Figure 4-2) is determined from

$$Q = \frac{\pi \beta K_1 U}{2 - \pi \beta} \tag{4.8}$$

where $\beta = 0.1$ (equivalent viscous damping ratio) at U=T (shear strain $\gamma = 1.0$). It follows that $K_1=1.05$ kN/mm, $K_2=2.10$ kN/mm, Q=29.35 kN, $D_1=180$ mm, $D_2=195$ mm, Y=10.5 mm. The mathematical model of Equations 4.1 to 4.6 is constructed from these data and analytically determined loops of force vs displacement are shown in Figures 4-3 to 4-5.

In Figure 4-3 the imposed displacement is harmonic with amplitudes of 240 mm and 176.8 mm along the X axis. The computed loop in the X direction shows the anticipated stiffening behavior in the motion with U_0 =240 mm, whereas in the loop for displacement amplitude less than D_1 , it does not. In Figure 4-4 the imposed motion is also harmonic with amplitudes of 240 mm and 176.8 mm along a 45° axis. The loops in that direction are identical to that of Figure 4-3. The loops at the largest displacement amplitude in the X and Y directions show stiffening behavior despite that the displacement amplitude is 169.7 mm, thus less than the limit D_1 =180 mm. This of course, was expected since the amplitude of the resultant displacement is 240 mm, thus more than the limit D_1 .

Figure 4-5 shows loops of force vs displacement in bi-directional motion of elliptical shape (X and Y displacements of motion out of-phase). The peculiar shape of the loop in the Y direction bears a similarity to loops recorded in tests with bidirectional motion of other isolation devices (Mokha 1993).

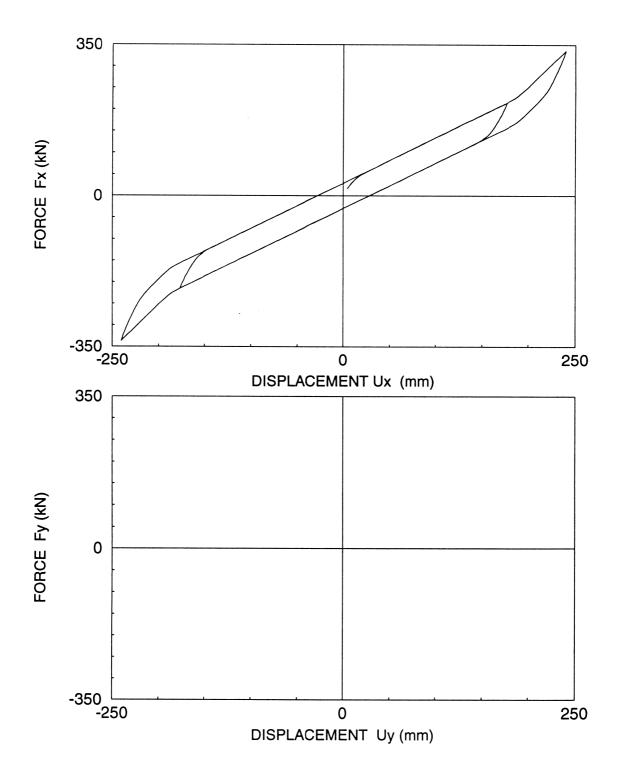


Figure 4-3 Force-Displacement Loops of High Damping Rubber Bearing in X and Y Directions for Motion $U_x = U_0 \sin(2\pi ft)$, $U_y = 0$, f = 0.5Hz, $U_0 = 240mm$ and 176.8 mm.

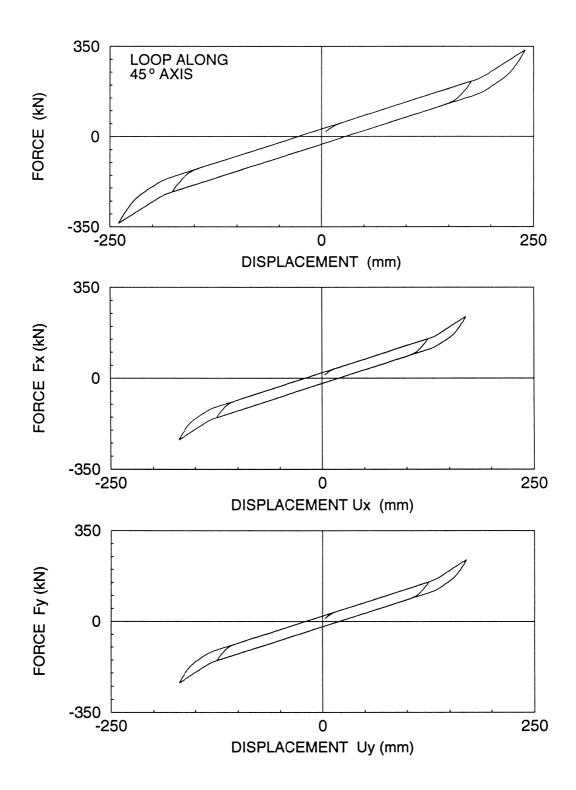


Figure 4-4 Force-Displacement Loops of High Damping Rubber Bearing Along 45° Angle and in X and Y Directions for Motion $U_x = U_0 \sin(2\pi ft), U_y = U_0 \sin(2\pi ft), f = 0.5Hz,$ $U_0 = 125mm$ and 169.7 mm.

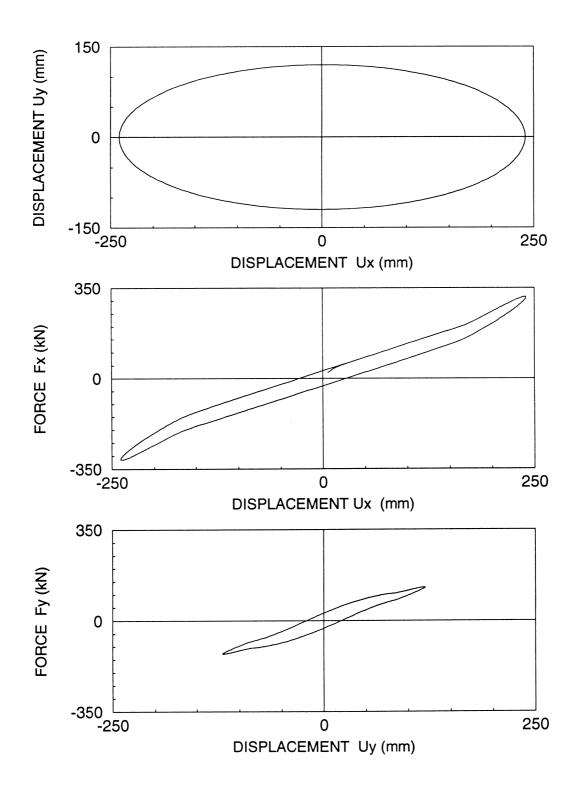


Figure 4-5 Force-Displacement Loops of High Damping Rubber Bearing in X and Y Directions for Motion $U_x = 2U_0 \sin(2\pi ft), U_y = U_0 \cos(2\pi ft), f = 0.5Hz, U_0 = 120mm$.

4.2 Element for Friction Pendulum (FPS) Bearing

The principles of operation of the FPS bearing have been established by Zayas et al. 1987, Mokha et al. 1990 and Constantinou et al. 1993. These principles are, of course, valid for all types of spherical sliding bearings. A cross section view of an FPS bearing is shown in Figure 4-6. The bearing consists of a spherical sliding surface and an articulated slider which is faced with a high pressure capacity bearing material. The bearing may be installed as shown in Figure 4-6 or upside-down with the spherical surface facing down rather than up. In both installation methods the behavior is identical.

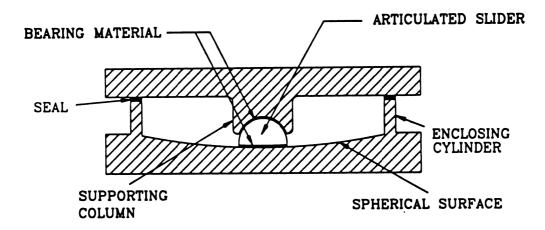


Figure 4-6 FPS Bearing Section.

The force-displacement relation of an FPS bearing in any direction is given by

$$F = \frac{N}{R}U + \mu_s N sgn(\dot{U})$$
(4.9)

in which R is the radius of curvature of the spherical sliding surface, N is the normal load and μ_s is the coefficient of the sliding friction. In cases in which the normal load may be assumed to

be constant and equal to the carried weight W_i , modeling of an FPS bearings may be accomplished by combining the linear elastic element of Section 2.2.1, using stiffness $K_{x_i} = K_{y_i} = W_i/R$, and the biaxial element for sliding bearing of Section 2.2.4, using N=W_i. To reduce computational effort, all the linear elastic elements may be combined in a global element described by translational stiffnesses $K_x = K_y = \Sigma W_i/R$ (ΣW_i =total weight) and corresponding rotational stiffness K_r and associated eccentricities e_x^B and e_y^B (see Section 2.2.1 and Nagarajaiah et al. 1989 and 1991). This has been the approach followed in programs 3D-BASIS and 3D-BASIS-M.

In general, the vertical load on an isolation bearing does not remain constant but rather varies as a result of the vertical ground motion and the effect of overturning moment. For vertically rigid structures, the normal load on an FPS bearing is

$$N = W_i \left(1 + \frac{\ddot{U}_v}{g} + \frac{N_{OM}}{W_i} \right)$$
(4.10)

where W_i is the weight, \ddot{U}_{ν} is the vertical ground acceleration (positive when the direction is upwards) and N_{OM} is the additional axial force due to the overturning moment effects (N_{OM} is positive when compressive).

The direct effects of variations in the normal load on the behavior of the FPS bearing are to instantaneously change the stiffness and friction force. Another indirect effect is to change the coefficient of friction which is pressure dependent. Modeling of the behavior of FPS bearings to this detail is important in the accurate estimation of the forces in individual bearings. However, use of $N=W_i$ rather than Equation 4.10 results in nearly the same global isolation system response and superstructure response. This has been demonstrated by comparison of analytical results to shake table results of a seven-story model in which the axial forces on individual bearings varied from 0 to $2W_i$, W_i being the gravity load (Al-Hussaini et al. 1994).

The forces in the FPS element of program 3D-BASIS-ME are described by

$$F_x = \frac{N}{R}U_x + \mu_s NZ_x, \qquad F_y = \frac{N}{R}U_y + \mu_s NZ_y$$
(4.11)

which Z_x and Z_y are described by Equation 2.8 and N is described by Equation 4.10. Program 3D-BASIS-ME requires user-supplied routines to

- a) Calculate the additional axial force on individual bearings from overturning moments about the two horizontal orthogonal axes, and
- b) Describe the variation of coefficient f_{max} in Equation 2.14 with bearing pressure.

Details of these routines are given in Section 4.6.

4.3 New Biaxial Element for Sliding Bearings

The new biaxial element for flat sliding bearings in program 3D-BASIS-ME is again described by Equations 2.12 to 2.14 and 2.8 with the exeption that N is not constant but rather described by Equation 4.10. The element requires the user-supplied routines described in Sections 4.2 and 4.6. It should be noted that when \ddot{U}_{ν} is not given and when the user-supplied routine returns zero for the additional axial load N_{OM} (eq. 4.10), the model collapses to the original constant normal load (N=W_i) model of programs 3D-BASIS and 3D-BASIS-M.

4.4 Linear Elastic Element

This element can be used to model the behavior of helical steel springs, rubber springs or other devices that exhibit linear elastic behavior.

The model of linear elastic element in program 3D-BASIS-ME is identical with the one available in programs 3D-BASIS and 3D-BASIS-M. In those programs, the properties of the linear elastic elements were combined automatically by the program in one global element, whereas in 3D-BASIS-ME the program is dealing with each element independently. The forces generated in each element are

$$F_x = K_x U_x, \qquad F_y = K_y U_y \tag{4.12}$$

where K_x , K_y and U_x , U_y are the stiffnesses and displacements of the element in X and Y directions, respectively.

It should be noted that the option of using one global linear elastic element that combines the properties of a linear elastic isolation system is also available in the program 3D-BASIS-ME (see Section 2.2.1 and Appendix A, Section C2).

4.5 Viscous Element

This element is suitable for modeling the behavior of Fluid Viscous Dampers or other devices displaying viscous behavior. Specifically, fluid dampers which operate on the principle of fluid orificing produce an output force which is proportional to the power of the velocity. That power can take values in the range of 0.5 to 2.0 (Constantinou et al. 1992).

The mobilized forces on a viscous element are described by

$$F_x = C_x \mid \dot{U}_x \mid^{\alpha} sgn(\dot{U}_x) \tag{4.13}$$

$$F_{y} = C_{y} \left| \dot{U}_{y} \right|^{\alpha} sgn(\dot{U}_{y})$$

$$(4.14)$$

where C_x , C_y and \dot{U}_x , \dot{U}_y are damping coefficients and velocities experienced by viscous elements placed along the X or Y directions respectively, and α is a coefficient taking real positive values. For $\alpha = 1$, the linear viscous element is recovered. It should be noted that program 3D-BASIS-ME allows only the placement of dampers along the principal directions.

In the case of linear viscous devices, an alternative approach is possible. The properties can be combined in one global linear viscous element located at the center of mass of the base (see Section 2.2.2 and Appendix A, Section C3).

4.6 User-Supplied Routines in Program 3D-BASIS-ME

4.6.1 Routine for Additional Axial Load Due to Overturning Moment Effects

The routine (a function) has the form

FOVM(OVMX,OVMY,XP,YP,I)

in which I is the bearing number, XP and YP are arrays containing the bearing coordinated (XP(I)=X coordinate of bearing I etc.), and OVMX and OVMY are the overturning moments about the X and Y axes, as illustrated in Figure 4-7. Function FOVM is called by the main program at all time steps. The function returns to the main program the additional axial load FOVM on bearing I. FOVM is positive when compressive.

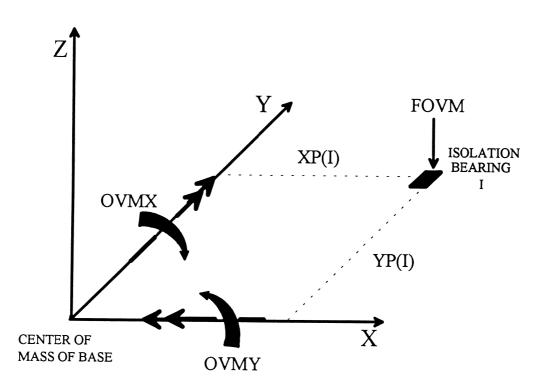


Figure 4-7 Definition of Overturning Moments OVMX and OVMY, and Additional Force FOVM.

It should be noted that we have assumed a unique relation between overturning moments and additional axial load on bearings. The user is cautioned that this is a simplification of a complex phenomenon. However, it is a commonly used engineering approximation. The report of Al Hussaini et al. 1994 provides valuable insight into the behavior of slender isolated structures with FPS bearings which are subjected to strong overturning moments.

To exclude the effect of the overturning moment on the additional axial force, function FOVM should be as follows:

FUNCTION FOVM(OVMX,OVMY,XP,YP,I) IMPLICIT REAL *8 COMMON/MAIN1 / NB,NP,MNF,MNE,NFE,MXF DIMENSION XP(NP),YP(I) FOVM=0.D0 RETURN END

This is the default version of function FOVM in 3DBASIS-ME.

4.6.2 Routine for Describing the Dependency of Parameter f_{max} on Bearing Pressure

Constantinou et al. 1990 and 1993 described the dependency on bearing pressure of the parameters in the model of friction in Equation 2.14. Specifically, the coefficient of sliding friction is given by

$$\mu_{s} = f_{\max} - (f_{\max} - f_{\min}) \exp(-a \mid \dot{U} \mid)$$
(4.15)

where *a* is nearly independent of pressure, whereas f_{\min} is dependent on pressure for unfilled and glass-filled PTFE but nearly independent of pressure for the PTFE-composites used in the FPS bearings. Parameter f_{\max} is generally dependent on bearing pressure. Since parameter f_{\max} describes the maximum friction force that is transmitted through the bearing, its dependency on pressure is explicitly modeled in program 3D-BASIS-ME. However, the much less significant dependency on pressure of parameters *a* and f_{\min} is neglected.

The user-supplied routine (function) has the form

FFMAX(FRMAX,FRMIN,FNOR,I)

in which I is the bearing number, FNOR is the normal load on bearing I, which includes the gravity, vertical ground motion and overturning moment effects, normalized by the weight W_i on the bearing. Furthermore, FRMAX and FRMIN are, respectively, the supplied, through the INPUT, parameters f_{max0} and f_{min0} under almost zero static pressure of bearing I. Function FFMAX returns the value of f_{max} at the bearing pressure resulting from the instantaneous normal load. Note that parameter f_{min} is assumed independent of pressure, that is $f_{min0} = f_{min}$.

For example, consider the case in which the dependency on pressure of parameter f_{max} is neglected.

Function FFMAX should be

FUNCTION FFMAX(FRMAX,FRMIN,FNOR,I) IMPLICIT REAL *8 COMMON/MAIN1 / NB,NP,MNF,MNE,NFE,MXF FFMAX=FRMAX RETURN END

This is the default version of function FFMAX in 3DBASIS-ME.

Consider now the case of pressure dependent parameter f_{max} . Figure 4-8 shows the assumed dependency on pressure of parameter f_{max} . It is typical of the behavior of sliding bearings (Soong and Constantinou 1994). An accurate representation of the variation of parameter f_{max} with pressure can be accounted for by using the following expression

$$f_{\max} = f_{\max 0} - (f_{\max 0} - f_{\max p}) \tanh(\varepsilon p)$$
(4.16)

where p is the pressure, f_{maxp} is the maximum coefficient of friction at very high pressures, f_{max0} is the vakue of the coefficient at zero pressure, ε is a constant that controls the transition of f_{max} between very low and very high pressures.

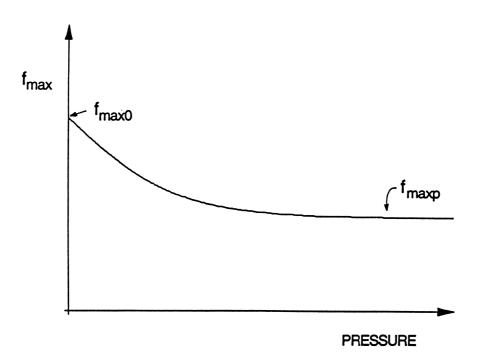


Figure 4-8 Variation of Friction Parameter f_{max} with Pressure.

As an example, Constantinou et al. 1993 gave the following values for the parameters of a bearing at pressure of 17.2 MPa : $f_{max0}=0.12$, $f_{maxp}=0.05$, $\epsilon=0.012$ (p is in units of MPa). For this case function FFMAX should be of the form :

FUNCTION FFMAX(FRMAX,FRMIN,FNOR,I) IMPLICIT REAL *8 COMMON/MAIN1 / NB,NP,MNF,MNE,NFE,MXF DIMENSION P(500) DATA/P(J)=17.2,J= 1,../ etc. etc.

PRES=FNOR*P(I) FFMAX=FRMAX-0.07*DTANH(0.012*PRES) RETURN END

Note that P(J) contains the bearing pressure under static conditions of bearing J. Quantity PRES is the instantaneous bearing pressure in units of MN/m^2 or MPa.

4.7 Validation of Model of FPS Bearing

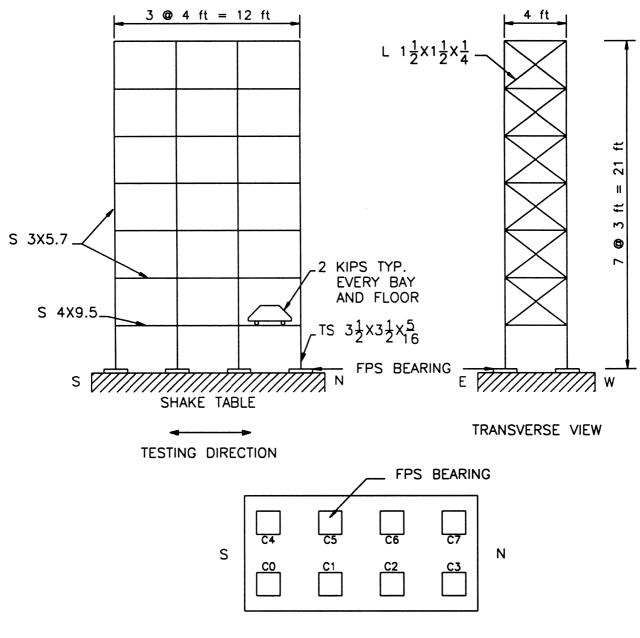
The validity of the model of FPS bearings in program 3D-BASIS-ME is investigated by comparison of the predictions of the model to experimental results. The experimental results were obtained in shaking table testing of an isolated structure, in which the FPS bearings were subjected to lateral motion under normal load of varying magnitude.

Al-Hussaini (1994) reported test results of a 7-story model structure supported by eight FPS bearings and tested in a variety of structural system configurations. One of these configurations is shown in Figure 4-9. The 7-story structure is a moment resisting frame with the isolators placed directly below the columns without connecting them to form an isolation basemat. The structure had a total weight of 212 kN (47.5 kips). The bearings had radius of curvature R=248 mm (9.75 in) and were loaded to an averege bearing pressure of about 110 MPa (16 ksi), for which the friction coefficient f_{max} was measured to be 0.06. Length scale factor in the experiments was 4.

The columns of the model above the isolation bearings were instrumented with strain gages so that measurements of axial and shear force could be made. In one test the table was excited with the 1971 San Fernando earthquake, record at Pacoima Dam, component S74W. While the command signal consisted of only horizontal motion, the shake table responded with additional vertical, roll and pitch motions, as a result of the large model weight and demand for high table velocity. The recorded horizontal and vertical acceleration histories of the table are shown in Figure 4-10. The recorded loops of bearing shear force versus bearing displacement of two FPS bearings (one interior and one exterior) are shown in Figure 4-11.

The loops have been also analytically constructed from the recorded histories of bearing displacement and axial bearing force by using Equations (4.11), (2.8), (2-14) and (4.15). That is U_x = recorded bearing displacement, $U_y = 0$, N = recorded axial force, and \dot{U}_x and \dot{U}_y were determined by numerical differentiation of the displacement histories. The parameters used were : $f_{max0} = 0.12$, $f_{maxp} = 0.05$, $f_{min} = 0.04$, $\varepsilon = 0.012$ (*MPa*)⁻¹ and a = 0.0429 s/mm. The bearing pressure under

4-17



PLAN VIEW OF ISOLATION LEVEL

Figure 4-9 Model in Shake Table Testing of Al-Hussaini (1994).

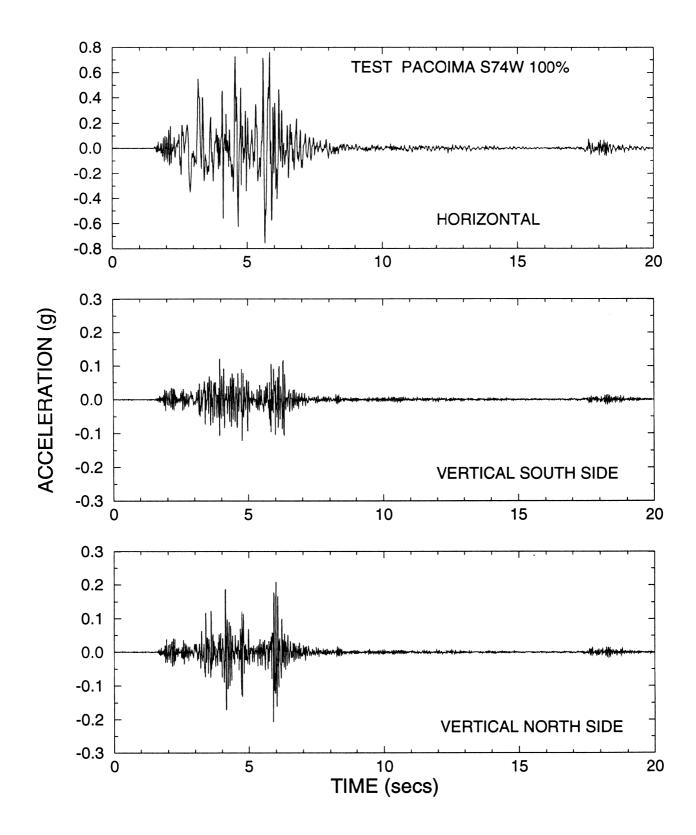


Figure 4-10 Recorded Horizontal and Vertical Accelerations of Shake Table in Test of Model Structure with Pacoima Dam S74W Input.

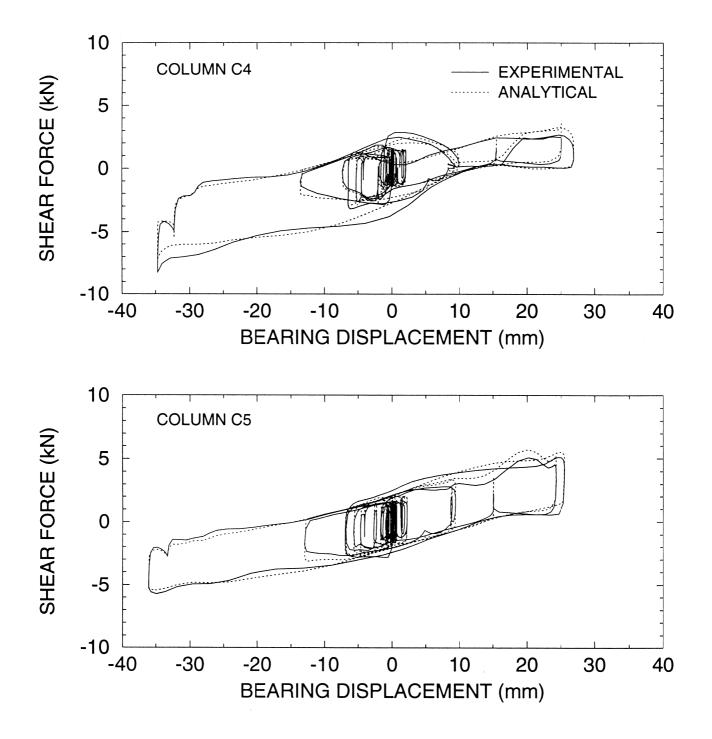


Figure 4-11 Recorded and Analytically Predicted Force-Displacements Loops of Exterior (C4) and Interior (C5) Bearings in Pacoima Dam S74W Test.

static conditions was set at 120 MPa for the interior bearing and at 220 MPa at the exterior bearing. The analytical results are compared to the experimental results in Figure 4-11. The agreement is very good.

In another test the structure was excited with the 1940 El Centro earthquake, components S00E and vertical. Figure 4-12 shows the recorded table accelerations. The shake table response was unstable with extremely high frequency vertical motion, which reached a peak acceleration 0.6 g (it should have been only 0.21 g). The recorded loops of the bearing shear force versus bearing displacement of one exterior and one interior bearings are shown in Figure 4-13. The analytically determined loops, obtained by the same model and using the same parameters, are compared to the experimental ones in Figure 4-13. Again the agreement is very good.

Of interest is to note in Figure 4-13 the significant variations in shear force of the interior (C5) bearing. These variations could not be caused by variations in the friction force alone. Rather, they are caused by variations in both the restoring force (that is, force NU/R in Equation 4.9) and friction force.

Finally, the dynamic response of the tested model in the 1940 El Centro S00E plus vertical input (Figure 4-12) was computed with program 3D-BASIS-ME. The analytical model was based on the experimentally determined modal properties of the structure (Al-Hussaini 1994). The overturning moment effects on the axial bearing load was accounted for by assuming a linear distribution of axial load. Time histories of isolation system displacement and base shear-displacement loops are compared in Figure 4-14. They compare well.

We may conclude that satisfactory experimental evidence has been provided for the validity of the FPS bearing (and other spherical sliding bearings) model in 3D-BASIS-ME.

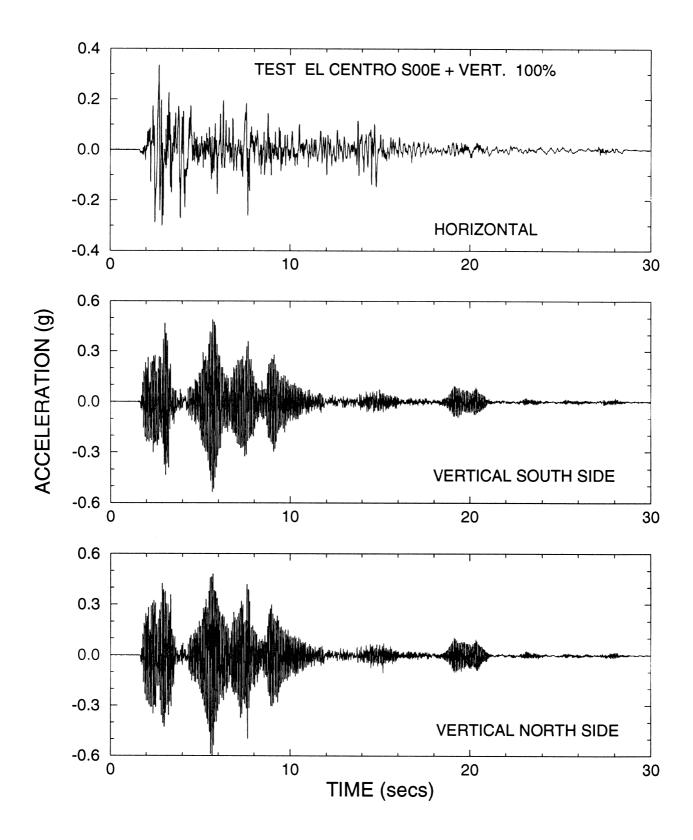


Figure 4-12 Recorded Horizontal and Vertical Accelerations of Shake Table in Test of Model Structure with El Centro S00E plus Vertical Input.

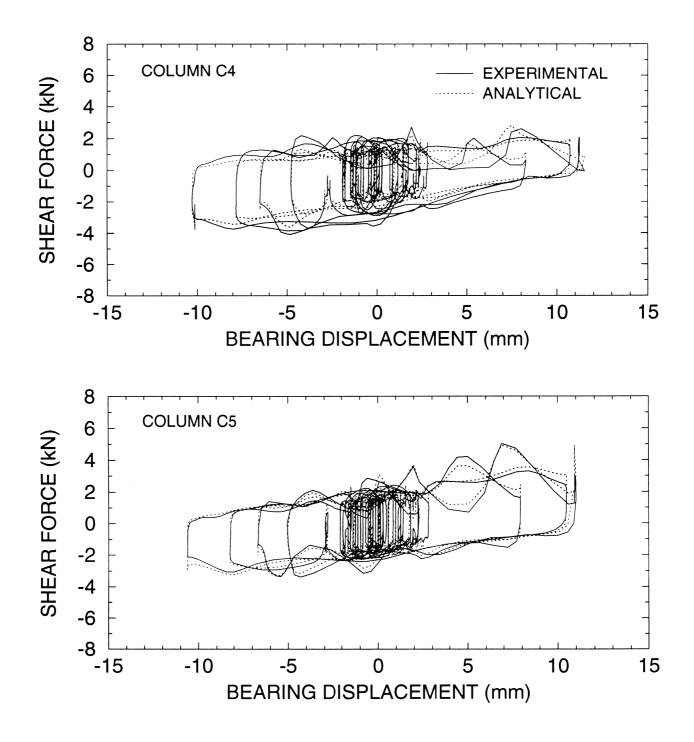


Figure 4-13 Recorded and Analytically Predicted Force-Displacements Loops of Exterior (C4) and Interior (C5) Bearings in El Centro S00E plus Vertical Test.

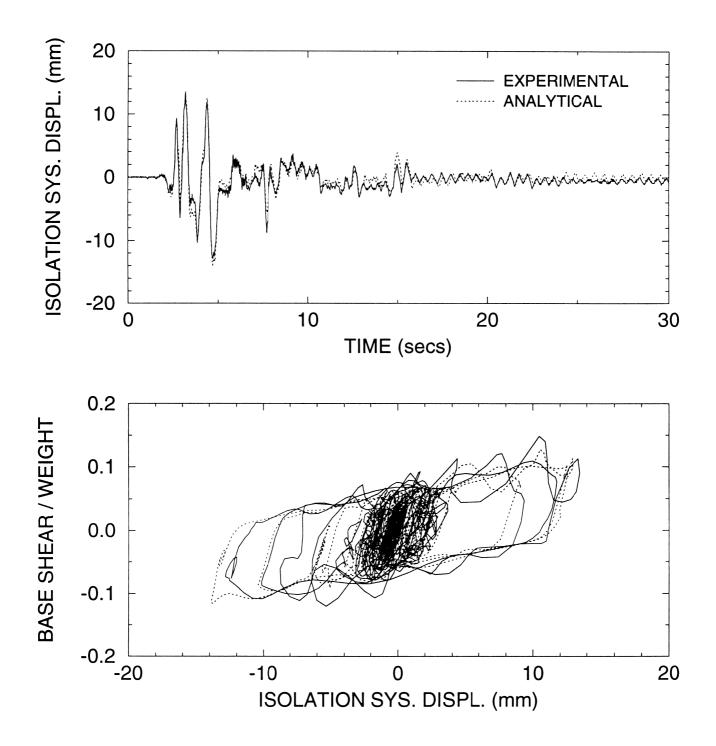


Figure 4-14 Comparison of Experimental and Analytical Response of Isolation System of Tested Model in El Centro S00E plus Vertical Input.

SECTION 5

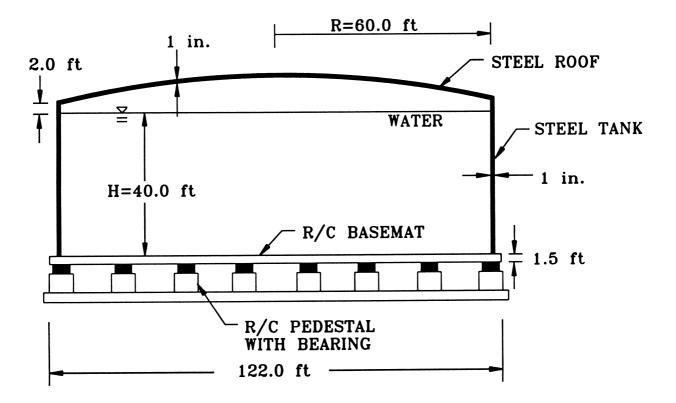
EXAMPLES

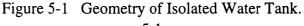
5.1 Introduction

Examples are presented which demonstrate the modeling details of isolation devices and their implementation in program 3D-BASIS-ME. Detailed input and output from the program are presented for each case.

5.2 Isolated Structure

The isolated structure is a water tank as illustrated in Figure 5-1. Unit weights are : for water 62.75 lb/ft³, for steel 490 lb/ft³ and for concrete 150 lb/ft³. The weights are : water (for full tank) 28387.4 kips, steel tank 646.5 kips, steel roof 477.3 kips and concrete basemat 2629.8 kips with a total isolated weight of 32141 kips.





Four different isolated systems are considered. They are :

- (a) High damping rubber bearings,
- (b) Low damping rubber bearings with linear viscous fluid dampers,
- (c) Low damping rubber bearings with non-linear viscous fluid dampers and
- (d) FPS bearings.

Each isolation system consists of 52 bearings or 52 bearings plus 24 fluid dampers in the configurations shown in Figures 5-2 and 5-3.

5.3 Mathematical Model of Tank

The mathematical model of the tank is based on the mechanical analog of Haroun and Housner, 1981 which takes into account the deformability of the tank wall and sloshing of the fluid. In the mathematical model used in the present examples, only the fundamental sloshing and fundamental tank-fluid modes of vibration are considered. Based on the theory of Haroun and Housner, 1981, the following were determined:

Sloshing Mode : Sloshing weight 16317 kips, sloshing period 6.89 secs, damping ratio (assumed) 0.005.

Fluid-tank Mode: Weight 12000 kips, period 0.162 secs, damping ratio (assumed) 0.02. The model of the tank is illustrated in Figure 5-4. It should be noted that the convective fluid is rigidly attached to the concrete basemat, raising its weight to 3824 kips.

5.4 Design of Isolation Systems

The design of the isolation systems does not follow a common design basis and their safety is not assessed. Rather, the design demonstrates the capabilities of the computer program rather than the capabilities of the isolation systems.

The properties of the isolation systems are determined in the stage of least stiffness and characteristic strength. That is, for rubber the properties under scragged and fresh conditions are used. Furthermore, all quantities, such as friction coefficient and shear modulus of rubber are obtained from

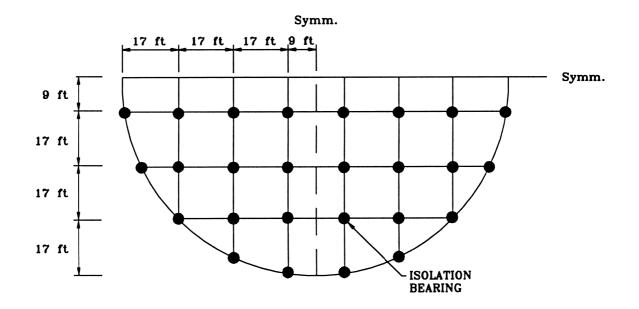


Figure 5-2 Configuration of Isolation System in High Damping Rubber Bearing and FPS System.

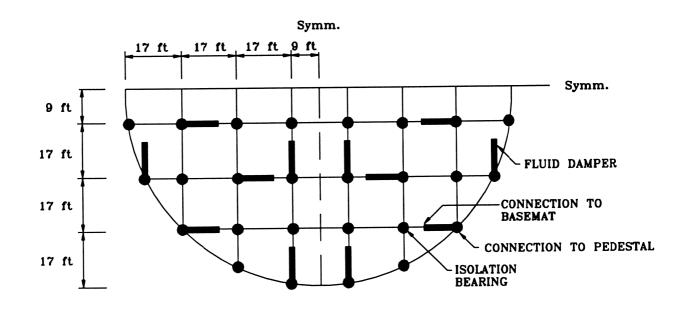


Figure 5-3 Configuration of Low Damping Rubber Bearing - Fluid Damper Isolation System.

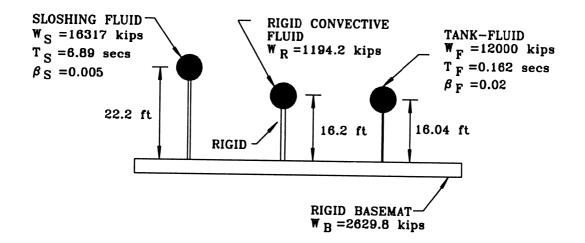


Figure 5-4 Mathematical Model of Fluid Tank.

representative mean values at normal temperatures and fresh conditions, with a further reduction for variability of properties. Thus, the analyzed stage is the one which results in the maximum response of the isolation system (i.e. bearing displacement). A complete analysis would require further analysis for a second stage of properties with the highest stiffness and characteristic strength. For this second stage it is necessary to consider the unscragged properties of rubber, aging and low temperature effects, and furthermore increase these properties for variability. This second stage results in the maximum response in the isolated superstructure.

5.4.1 High Damping Rubber Bearing System

The system consists of 52 bearings in the configuration of Figure 5-2. The bearing construction is shown in Figure 5-5. The bearings have stiffening hysteretic behavior as shown in Figure 4-2 with $K_2=2K_1$, $K_1=AG/T$ (Equation 4.7), $D_1=1.2T$, $D_2=1.25T$ and Y=0.06T, where A=bonded rubber area, T=total rubber thickness and G=115 psi. Furthermore, the characteristic strength Q is determined from Equation 4.8 and an assumed damping ratio $\beta=0.10$ at shear rubber strain of 1.0.

The properties for each bearing are

 $K_1 = 7.45$ kip/in $K_2 = 14.90$ kip/in Q = 13.19 kips Y = 0.57 in $D_1 = 11.40$ in $D_2 = 11.88$ in

5.4.2 Low Damping Rubber Bearing and Linear Viscous Fluid Damper System

The system consists of 52 bearings and 24 linear viscous fluid dampers in the configuration of Figure 5-3. The rubber bearing construction is shown in Figure 5-5. The behavior of the bearing is linear elastic and viscous with stiffness K=AG/T where G=96 psi. Thus, K=7.63 kip/in. The viscous behavior is accounted for by assuming a damping ratio in the isolation system equal to 0.03.

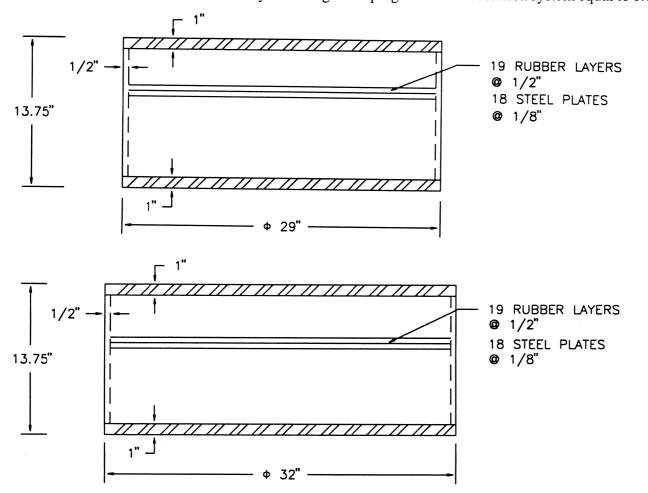


Figure 5-5 Construction of (a) High Damping Rubber Bearing, (b) Low Damping Rubber Bearing.

Each fluid damper has linear viscous behavior with force F_i proportional to velocity, \dot{U}_i (i= X or Y, dampers are placed along principal directions):

$$F_i = C_L \dot{U}_i \tag{5.1}$$

where C_L = 3.61 kip-s/in. Approximate dimensions of a fluid damper with constant C_L = 3.61 kip-s/in, stroke of ±15 in., rated load of 200 kips and ultimate load of 500 kips are shown in Figure 5-6. It should be noted that for twelve dampers C_{TOTAL} =43.32 kip-s/in. Thus, for a SDOF system with K_{TOTAL} =52X7.63=396.76 kip/in and weight of 15824 kips (excluding the weight of the very flexible sloshing mode), the damping ratio is 0.17. This, together with 0.03 damping inherent in the rubber bearing, gives a total viscous damping of 0.20 of critical.

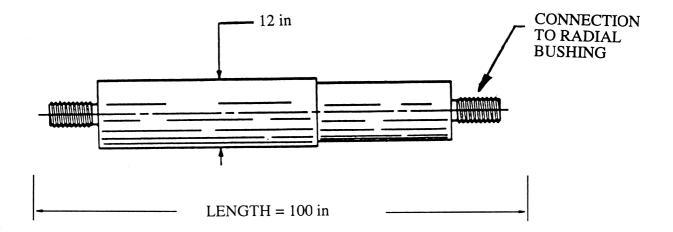


Figure 5-6 Approximate Dimensions of Fluid Damper with Stroke of ±15 in and Ultimate Load of 500 kips.

5.4.3 Low Damping Rubber Bearing and Nonlinear Viscous Fluid Damper System

The system consists of 52 bearings and 24 nonlinear viscous fluid dampers in the configuration of Figure 5-3. The bearing construction is that of Figure 5-5 with K=7.63 kip/in and viscous damping ratio of 0.03. Each fluid damper has force-velocity relation

$$F_i = C_N |U_i|^{\alpha} sgn(\dot{U}_i)$$
(5.2)

where $\alpha=0.5$ and $C_N=26.67 \text{ kip}(s/in)^{1/2}$. A damper with this damping constant, stroke of ±15 in. and ultimate load of 500 kips has approximately the same dimensions as the damper of Figure 5-6.

The difference between the nonlinear and linear viscous fluid dampers is illustrated in Figure 5-7. For this motion with peak velocity approximately equal to the one calculated in the analyses, the two dampers reach nearly the same peak force. However, the nonlinear damper dissipates more energy per cycle. This often desirable feature of nonlinear dampers has long being exploited in the shock isolation of military hardware. Furthermore, nonlinear dampers with α equal to approximately 0.5 are used together with rubber bearings in the seismic isolation system of the San Bernandino County Medical Center Replacement Project, of which construction is scheduled to start in late 1994.

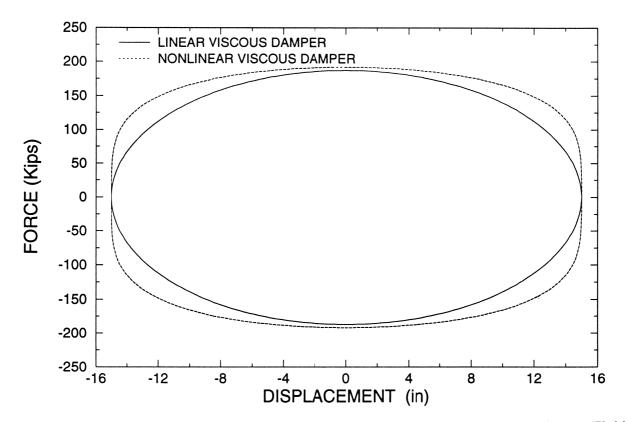


Figure 5-7 Comparison of Force-Displacement Loops of Linear and Nonlinear Viscous Fluid Dampers for Harmonic Motion of Frequency of 0.55 Hz and Amplitude of 15 in.

5.4.4 Friction Pendulum (or FPS) System

The system consists of 52 bearings in the configuration of Figure 5-2. The bearing construction is shown in Figure 5-8. The radius of curvature of the concave sliding surface is R=82.4 in. Average bearing pressure (for full tank) is 15 ksi. The coefficient of friction follows Equation 2.14 with f_{min} =0.03, a=0.8 sec/in and f_{max} =0.045 at pressure of 15 ksi. Parameter f_{max} is pressure dependent. Figure 5-9 depicts the variation of parameter f_{max} with bearing pressure.

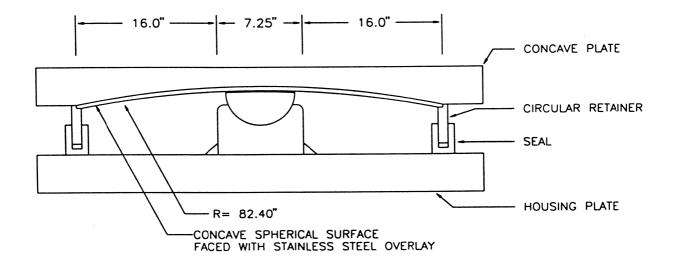


Figure 5-8 Construction of FPS Bearing.

5.5 Model of Isolated Tank in 3D-BASIS-ME

To reduce computational effort, the 52 isolation bearings are grouped into one cluster of 26 bearings at the center of the base and four clusters of 6.5 bearings each at a distance of 68.38 feet (820.56 in) from the center. In this way, the rotational stiffness of the five clusters of bearings is equal to that of the 52 bearings in the configuration of Figure 5-2. Furthermore, an eccentricity of 0.01 times the tank's plan dimension or 14.4 in. is induced in the X direction as illustrated in Figure 5-10.

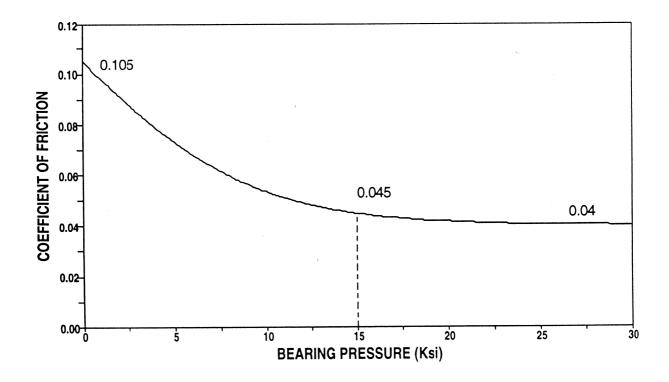


Figure 5-9 Dependency of Parameter f_{max} (coefficient of friction at high velocity of sliding) on Bearing Pressure of FPS Bearings.

The fluid dampers are also grouped into four clusters as shown in Figure 5-11. Each cluster consists of three fluid dampers placed in the X-direction and another three fluid dampers placed in the Y-direction. That is, the damping constants used for each cluster of dampers is $C_x=C_y=10.83$ kip-s/in for the linear dampers and $C_x=C_y=80.0$ kip(s/in)^{1/2} for the nonlinear dampers.

In the analysis of the isolation systems with low damping rubber bearings, additional viscous damping of 0.03 of critical is used to account for the energy dissipation capability of the bearings. This is included in the analysis as a global linear viscous element (see Section 2.2.2) with $C_x=C_y=7.65$ kip-s/in and $C_T=2374932$ kip-s-in.

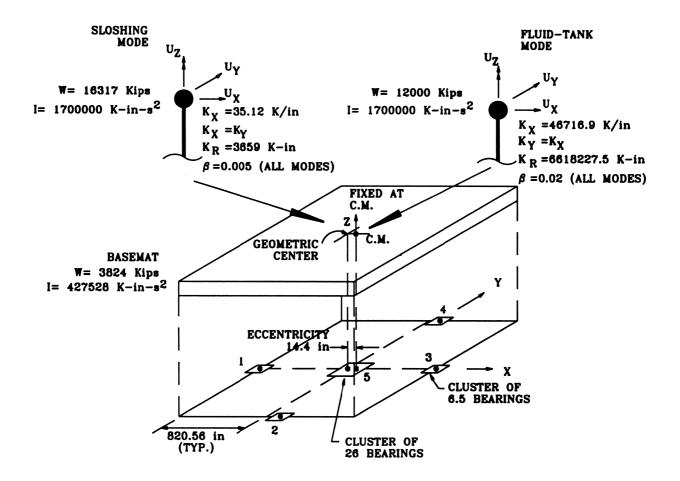


Figure 5-10 Model in 3D-BASIS-ME. Clusters of Bearings are used for Reducing the Computational Effort.

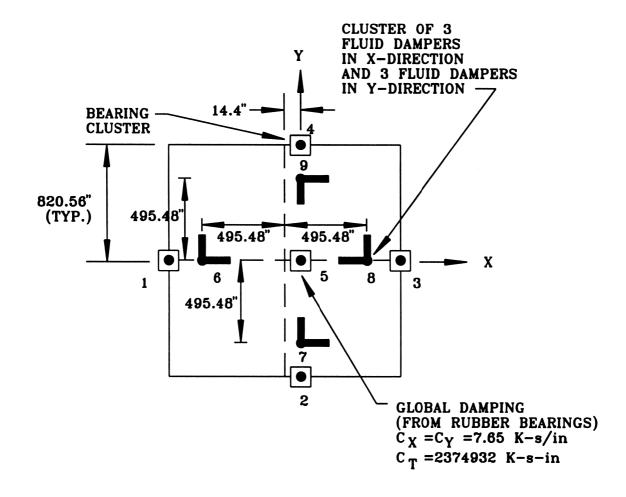


Figure 5-11 Clusters of Fluid Dampers in 3D-BASIS-ME. Model of Low Damping Rubber Bearing - Fluid Damper System. Rubber Bearings are Modeled as Linear Elements with Global Viscous Damping.

5.6 Seismic Excitation

Analyses are performed using the Pacoima Dam record from the 1971 San Fernando earthquake as input. Component S16E (PGA=1.17 g, PGV=44.58 in/s, PGD=14.83 in) is applied in the X direction and component S74W (PGA=1.08 g, PGV=22.73 in/s, PGD=4.26 in) is applied in the Y direction. The vertical component (PGA=0.71 g, PGV=22.95 in/s, PGD=7.60 in) is used only in the analysis of the FPS system. The very strong vertical component of this earthquake is known to influence the response of isolated structures with the FPS system. Specifically, Zayas et al. 1987 studied experimentally the response of three different isolated model structures with and without the vertical motion effects. Only the Pacoima Dam motion had some influence on the peak superstructure shear force, which amounted to an increase of about 20% over the case without vertical motion.

The vertical ground motion is included in the analysis of the FPS system because the effects of varying normal force on the FPS bearings is a well understood phenomenon (Al-Hussaini 1994, Constantinou 1993). Varying normal forces affect other types of isolation bearings. However, these effects are not well understood and have not been incorporated in computer program 3D-BASIS-ME. For such cases, a designer should bound the response by performing analyses which account for plausible variations in the characteristics of the isolation devices.

This seismic excitation is severe, even for an isolated structure. The use of this excitation illustrates the capabilities of program 3D-BASIS-ME in capturing the effects of strong vertical ground motion on the response of sliding systems and in capturing the effects of stiffening behavior at large strains of high damping rubber bearings. Furthermore, the nature of the motion, being a near-fault high velocity motion, demonstrates the usefulness of nonlinear viscous dampers.

5.7 Results of Dynamic Analysis

Dynamic analyses are performed for the five isolation systems under the following conditions:

(a) High damping rubber bearing system without the effect of vertical ground motion and overturning moments,

- (b) Low damping rubber bearing system with linear fluid dampers without the effect of vertical ground motion and overturning moments,
- (c) As (b) above but with the properties of bearings and fluid dampers represented by one global stiffness and one global damping element,
- (d) Low damping rubber bearing system with nonlinear fluid dampers without the effect of vertical ground motion and overturning moments,
- (e) FPS system without the effect of vertical ground motion and overturning moments,
- (f) FPS system with the effect of vertical ground motion and overturning moments.

Detailed input and output of program 3D-BASIS-ME for each case is presented in Appendix B. A summary of the results is presented in Table 5-I. Figures 5-12 to 5-14 present representative force-displacement loops of the four systems. It should be observed that analysis of the low damping rubber bearing-linear viscous fluid damper system by explicit representation of the isolation devices or by global representation gives nearly identical results.

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Table 5-I Summary of Dynamic Analysis Results

AK PER CITY CE ¹ Kips)		58 20)8 ² 40 ³	57 90		
PEAK DAMPER VELOCITY AND FORCE ¹ (in/s, kips)		51.58 186.20	51.08 ² 184.40 ³	47.57 183.90		
SLOSHING DISPL. IN X / Y DIRECTIONS (in)	40.41 12.09	37.76 7.89	37.79 7.87	39.03 5.93	40.53 9.82	40.72 9.97
RESULTANT FLUID-TANK SHEAR FORCE (kips)	8879.0	5052.0	5045.0	5245.0	6287.0	7228.0
RESULTANT SLOSHING FLUID SHEAR FORCE (kips)	1423.0	1337.0	1338.0	1382.0	1433.0	1439.0
RESULTANT ISOLATION SYSTEM SHEAR FORCE (kips)	11090.0	5821.0	5792.0	6029.0	7352.0	0.0067
RESULTANT CORNER BEARING DISPL. (in)	19.49	12.73	12.78 2	11.53	15.79	15.89
RESULTANT CENTER BEARING DISPL. (in)	19.32	12.60	12.58 2	11.40	15.51	15.64
TYPE OF ANALY- SIS	Without Vertical Motion and Overturn- ing Moment Effects	Without Vertical Motion and Overturn- ing Moment Effects	Without Vertical Motion and Overturn- ing Moment Effects, Global Representation of Isolation System	Without Vertical Motion and Overturn- ing Moment Effects	Without Vertical Motion and Overturn- ing Moment Effects	With Vertical Motion and Overturning Moment Effects
ISOLATION SYS- TEM	High Damping Rubber Bearing	Low Damping Rubber-Linear Fluid Damper	Low Damping Rubber-Linear Fluid Damper	Low Damping Rubber-Nonlinear Fluid Damper	FPS	SdH

¹ Force of one damper, ² At artificial (without stiffness) bearings, ³ Calculated from velocity

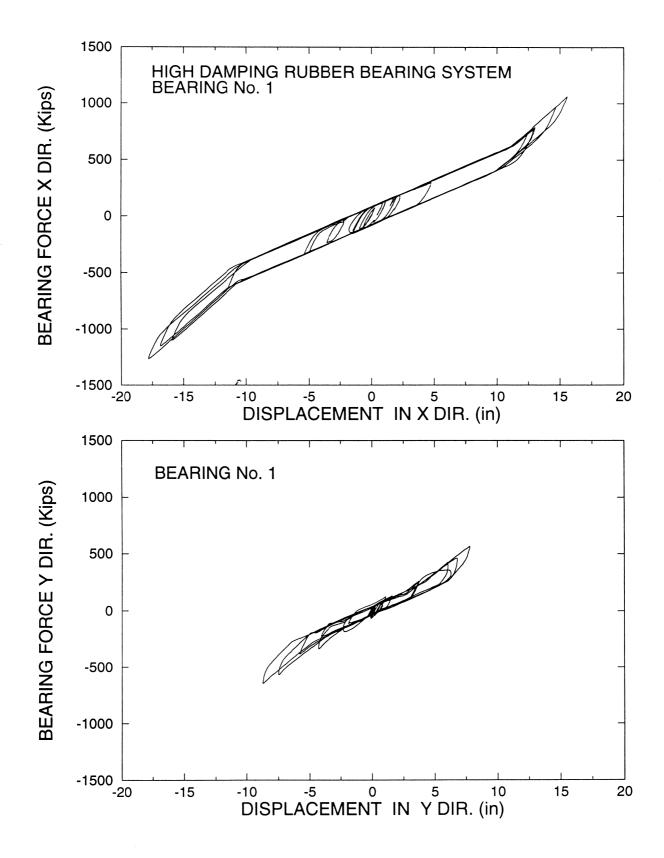


Figure 5-12 Force-Displacement Loops of High Damping Rubber Bearing System.

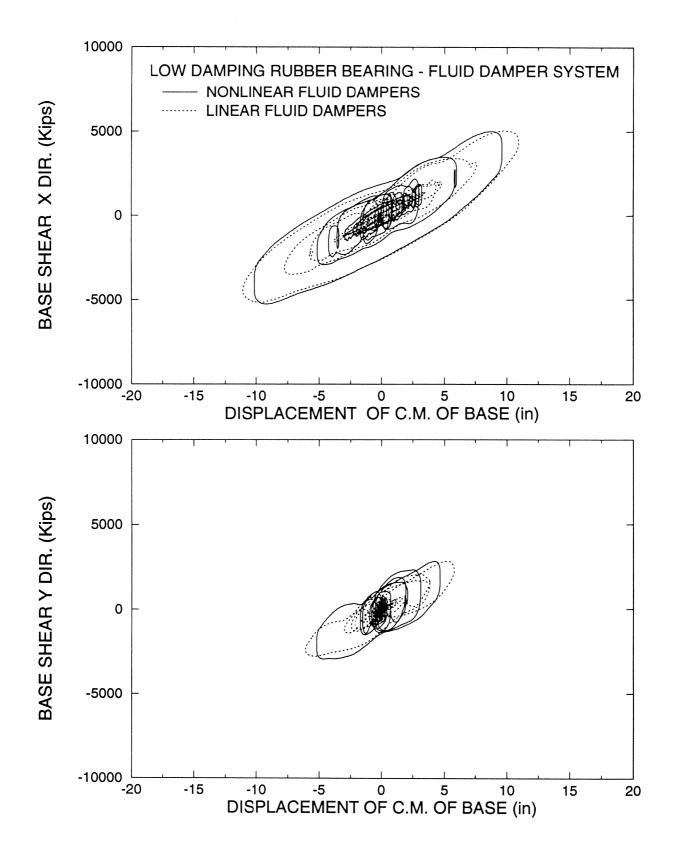


Figure 5-13 Base Shear Force-Isolation System Displacement Loops of Low Damping Rubber Bearing System with Fluid Dampers.

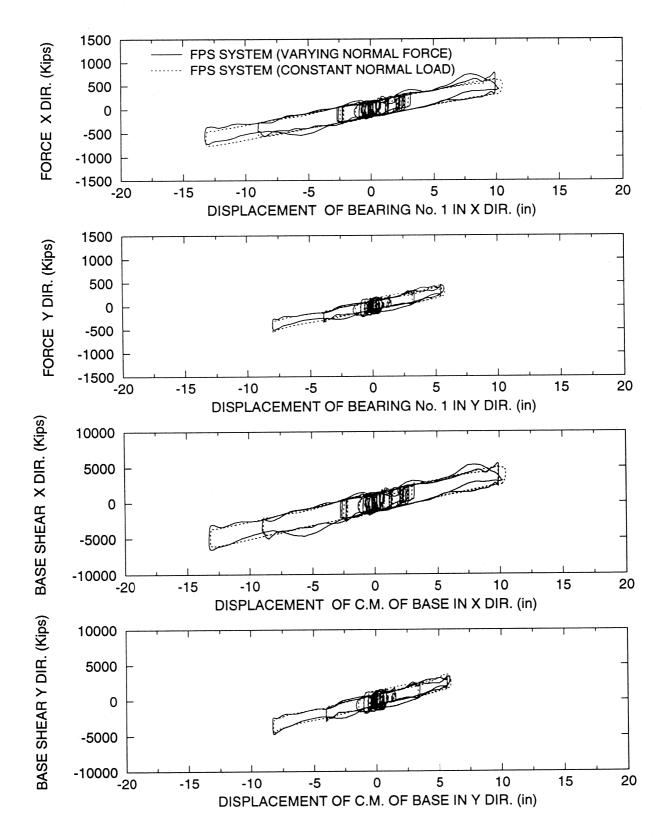


Figure 5-14 Force-Displacement Loops of FPS System With and Without the Effects of Vertical Ground Motion and Overturning Moments.

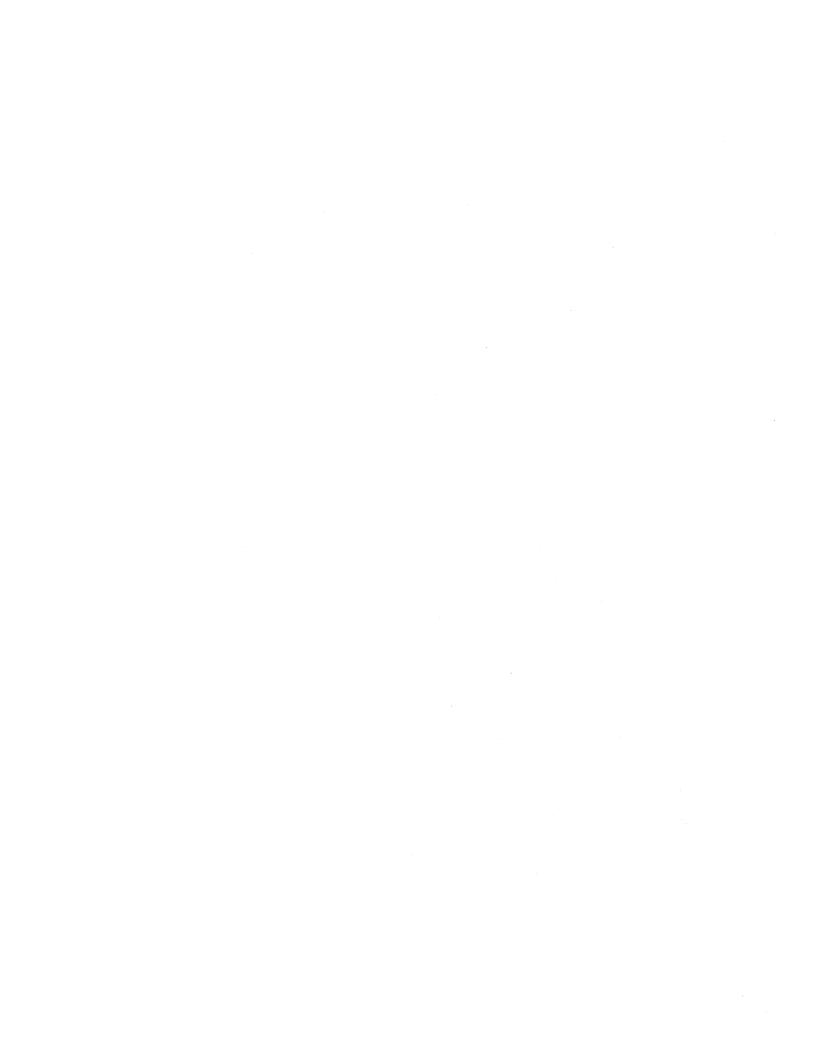
SECTION 6

SUMMARY

Program 3D-BASIS-ME is capable of analyzing single isolated structures, multiple building isolated structures on a common isolation basemat and isolated liquid storage tanks. New elements for modeling Friction Pendulum (FPS) bearings, high damping rubber bearings with stiffening behavior and nonlinear viscous dampers have been included in the program. Furthermore, program 3D-BASIS-ME accepts vertical ground motion and user supplied routines for describing the overturning moment effects on the axial bearing forces and for describing the dependency of the coefficient of friction on bearing pressure. This information is utilized by the program in modeling the behavior of sliding bearings.

The validity of the FPS bearing model in 3D-BASIS-ME has been established by comparisons of its predictions to experimental results under combined lateral displacement and varying normal load.

The capabilities of the program have been demonstrated through the analysis of an isolated liquid storage tank. The mathematical model included the effects of convective and impulsive modes of vibration.



SECTION 7

REFERENCES

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APPENDIX A 3D-BASIS-ME PROGRAM USER'S GUIDE

A.1 INPUT FORMAT FOR 3D-BASIS-ME

Input file name is 3DBME.DAT and the output file is 3DBME.OUT. Free format is used to read all input data. Earthquake records are to be given in files WAVEX.DAT and/or WAVEY.DAT and/or WAVEZ.DAT. Dynamic arrays are used. Double precision is used in the program for accuracy. Common block size has been set to 100,000 and should be changed if the need arises. All values are to be input unless mentioned otherwise. No blank cards are to be input.

A.2 PROBLEM TITLE

One card	
TITLE	TITLE up to 80 characters

<u>A.3 UNITS</u>

One card LENGTH,MASS,RTIME

- LENGTH = Basic unit of length up to 20 characters
- MASS = Basic unit of mass up to 20 characters
- RTIME = Basic unit of time up to 20 characters

A.4 CONTROL PARAMETERS

A.4.1 Control Parameters - Entire structure

One card ISEV,NB,NP,INP,G

- ISEV = 1 for option 1 Data for Stiffness of the superstructures to be input.
 ISEV = 2 for option 2 Eigenvalues and eigenvectors of the superstructures (for fixed base condition) to be input.
 NB = Number of superstructures on the common base.
 NP = Number of bearings.
 INP = Number of bearings at which output is desired.
 G = Gravitational acceleration.
 Notes: 1. For explanation of the option 1 and the option 2 refer to section 3.1.
 - 2. Number of bearings refers to the total number of bearings which could be a combination of linear elastic, viscous, smooth bilinear, sliding bearings etc. .

A.4.2 Control Parameters - Superstructures

NB cards NF(I),NE(I),I=1,NB

- NF(I)= Number of floors of superstructure I excluding base. (If NF<1 then NF set = 1)
- NE(I)= Number of eigenvalues of superstructure I to be retained in the analysis.(If NE<3 then NE set = 3)
- Notes: 1. Number of eigenvectors to be retained in the analysis should be in groups of three the minimum being one set of three modes.

A.4.3 Control Parameters - Integration

one card TSI,TOL,FMNORM,MAXMI,KVSTEP

- TSI = Time step of integration. Default = TSR (refer to A.4.5)
- TOL = Tolerance for the nonlinear force vector computation. Recommended value =0.001.
- FMNORM= Reference moment for convergence.
- MAXMI = Maximum number of iterations within a time step.
- KVSTEP=Index for time step variation.KVSTEP = 1 for constant time step.KVSTEP = 2 for variable time step.
- Note: 1. The time step of integration cannot exceed the time step of earthquake record.
 - 2. If MAXMI is exceeded the program is terminated with an error message.
 - 3. Compute an estimate of FMNORM by multiplying the expected base shear by one half the maximum base dimension.

A.4.4 Control Parameters - Newmark's Method

One card GAM,BET

- GAM = Parameter which produces numerical damping within a time step. (Recommended value = 0.5)
- BET = Parameter which controls the variation of acceleration within a time step. (Recommended value = 0.25)

A.4.5 Control Parameters - Earthquake Input

One card INDGACC,TSR,LOR,XTH,ULF

- INDGACC = Index for earthquake time history record. INDGACC = 1 for a single earthquake record at an angle of incidence XTH. INDGACC = 2 for two independent earthquake records along the X and Y axes. INDGACC = 3 for two independent earthquake records along the X and Z (vertical) axes. (X axes excitation at angle of incidence XTH. INDGACC = 4 for three independent earthquake records along X, Y and Z (vertical) axes.
- TSR = Time step of earthquake record(s).
- LOR = Length of earthquake record(s) (Number of data in earthquake record)
- XTH = Angle of incidence of the earthquake with respect to the X axis in anticlockwise direction (for INDGACC=1).
- ULF = Load factor.
- Notes: 1. Four options are available for the earthquake record input:

a. INDGACC = 1 refers to a single earthquake record input at any angle of incidence XTH. Input only one earthquake record (read through a single file WAVEX.DAT). Refer to D.2 for wave input information.

b. INDGACC = 2 refers to two independent earthquake records input in the X and Y directions, e.g. El Centro N-S along the X direction and El Centro E-W along the Y direction. Input two independent earthquake records in the X and Y directions (read through two files WAVEX.DAT and WAVEY.DAT). Refer to D.2 and D.3 for wave input information.

c. INDGACC = 3 refers to two independent earthquake records input in the X and Z directions, e.g. El Centro N-S along the X direction and El Centro Vertical along the Z direction. Input two independent earthquake records in the X and Z directions (read through two files WAVEX.DAT and WAVEZ.DAT). Refer to D.2 and D.4 for wave input information.

d. INDGACC = 4 refers to three independent earthquake records input in the X, Y and Z directions, e.g. El Centro N-S along the X direction and El Centro E-W along the Y directionand El Centro Vertical along the Z direction. Input three independent earthquake records in the X, Y and Z directions (read through three files WAVEX.DAT, WAVEY.DAT and WAVEZ.DAT). Refer to D.2 to D.4 for wave input information.

2. The time step of earthquake record and the length of earthquake record has to be the same in X, Y and Z directions for INDGACC = 2 or 3 or 4.

3. Load factor is applied to the earthquake records in the X, Y and Z directions.

B.1 SUPERSTRUCTURE DATA

Go to B.2 for option 1 - three dimensional shear building representation of superstructure.

Go to B.3 for option 2 - full three dimensional representation of the superstructure. Eigenvalue analysis has to be done prior to the 3D-BASIS-ME analysis using computer program ETABS.

Note: 1. The same type of group, B2 or B3, must be given for all superstructures (the same option, either 1 or 2, must be used for all superstructures).
2. The data must be supplied in the following sequence:
B2 or B3, B4, B5, B6 and B7 for superstructure No. 1, then repeat for superstructure No. 2, etc. for a total of NB superstructures.

B.2 Shear Stiffness Data for Three Dimensional Shear Building (ISEV = 1)

B.2.1 Shear Stiffness - X Direction (Input only if ISEV = 1)

NF cards SX(I),I=1,NF

SX(I) = Shear stiffness of story I in the X direction.

Note: 1. Shear stiffness of each story in the X direction starting from the top story to the first story. One card is used for each story.

B.2.2 Shear stiffness in the Y Direction (Input only if ISEV = 1)

NF cards SY(I),I=1,NF

SY(I) = Shear stiffness of story I in the Y direction.

Note: 1. Shear stiffness of each story in the Y direction starting from the top story to the first story.

B.2.3 Torsional stiffness in the $\boldsymbol{\theta}$ Direction

(Input only if ISEV = 1)

NF cards ST(I),I=1,NF

- ST(I) = Torsional stiffness of story I in the θ direction about the center of mass of the floor.
- Note: 1. Torsional stiffness of each story in the θ direction starting from the top story to the first story.

B.2.4 Eccentricity Data - X Direction (Input only if ISEV = 1)

NF cards EX(I),I=1,NF

EX(I) = Eccentricity of center of resistance from the center of mass of the floor I. Default = 0.0001.

B.2.5 Eccentricity Data - Y direction (Input only if ISEV = 1)

NF cards EY(I),I=1,NF

- EY(I) = Eccentricity of center of resistance from the center of mass of the floor I. Default = 0.0001.
- Note: 1. The case of zero eccentricity in both the X and Y directions cannot be solved correctly by the eigensolver in the program, hence if both the eccentricities are zero, a default value of 0.0001 is used.

B.3 Eigenvalues and Eigenvectors for Fully Three Dimensional Building (ISEV = 2)

B.3.1 Eigenvalues (Input only if ISEV = 2)

NE cards W(I),I=1,NE

W(I) = Eigenvalue of Ith mode.

Note: 1. Input from the first mode to the NE mode. 2. Eigenvalues are frequencies squared (ω^2 in rad²/s²)

B.3.2 Eigenvectors (Input only if ISEV =2)

NE cards (E(K,J),K=1,3*NF),J=1,NE

E(K,J)= Value corresponding to Kth floor of eigenvector of Jth mode.

Note: 1. Input from the first mode to the NE mode. 2. Eigenvectors must be normalized with respect to the mass matrix of superstructure $(\Phi^T M \Phi = \{1\})$.

B.4 Superstructure Mass Data

B.4.1 Translational Mass

NF Cards CMX(I),I=1,NF

CMX(I)= Translational mass at floor I.

Note: 1. Input from the top floor to the first floor.

B.4.2 Rotational Mass (Mass Moment of Inertia)

NF Cards CMT(I),I=1,NF

CMT(I)= Mass moment of inertia of floor I about the center of mass of the floor.

Note: 1. Input from the top floor to the first floor.

B.5 Superstructure Damping Data

NE Cards DR(I),I=1,NE

DR(I)= Damping ratio corresponding to mode I.

Note: 1. Input from the first mode to the NE mode.

B.6 Distance to the Center of Mass of the Floor

NF cards XN(I),YN(I),I=1,NF

- XN(I) = Distance of the center of mass of the floor I from the center of mass of the base in the X direction.
- YN(I) = Distance of the center of mass of the floor I from the center of mass of the base in the Y direction. (If ISEV = 1 then XN(I) and YN(I) set 0)
- Note: 1. Input from the top floor to the first floor.

B.7 Height of the Base and Different Floors

NF+1 cards H(I),I=1,NF+1

- H(I) = Height from the ground to the floor I.
- Note: 1. Input from the top floor to the base.

C.1 ISOLATION SYSTEM DATA

C.2 Stiffness Data for Linear Elastic Isolation System

	One card SXE,SYE,STE,EXE,EYE
SXE =	Resultant stiffness of linear elastic isolation system in the X direction.
SYE =	Resultant stiffness of linear elastic isolation system in the Y direction.
STE =	Resultant tortional stiffness of linear elastic isolation system in the θ direction about the center of mass of the base.
EXE =	Eccentricity of the center of resistance of the linear elastic isolation system in the X direction from the center of mass of the base.
EYE =	Eccentricity of the center of resistance of the linear elastic isolation system in the Y direction from the center of mass of the base.
Note:	 Data for linear elastic elements can also be input individually (refer to C.5.1). See reports by Nagarajaiah et al. 1989 and 1991 for definitions.

C.3 Mass Data of the Base

One Card CMXB,CMTB

CMXB =	Mass of the base	in the translational	direction.
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CMTB = Mass moment of inertia of the base about the center of mass of the base.

C.4 Global Damping Data

One card	
CBX,CBY,CBT,ECX,ECY	

- CBX = Resultant global damping coefficient in the X direction.
- CBY = Resultant global damping coefficient in the Y direction.
- CBT = Resultant global damping coefficient in the θ direction about the center of mass of the base.
- ECX = Eccentricity of the center of global damping of the isolation system in the X direction from the center of mass of the base.
- ECY = Eccentricity of the center of global damping of the isolation system in the Y direction from the center of mass of the base.

Note: 1. Data for viscous elements can also be input individually (refer to C.5.2).

2. See reports by Nagarajaiah et al. 1989 and 1991 for definitions.

C.5 Coordinates of Bearings

NP Cards XP(NP),YP(NP),I=1,NP

- XP(I) = X Coordinate of isolation element I from the center of mass of the base.
- YP(I) = Y Coordinate of isolation element I from the center of mass of the base.

Note: 1. If NP equals zero then skip Section C.5.

C.6 Isolation Element Data

The isolation element data are input in the following sequence:

1. Coordinates of isolation elements with respect to the center of mass of the base. One card containing the X and Y coordinates of each isolation element is used. The first card in the sequence corresponds to element No. 1, the second to element No. 2, etc. up to element No. NP.

2. The second set of data for the isolation elements consists of two cards for isolation element. The first card identifies the type of element and the second specifies its mechanical properties. Two cards are used for isolation element No. 1, then another two for element No. 2, etc. up to No. NP. The first of the two cards for each element always contains two integer numbers. These numbers are stored in array INELEM(NP,2) which has NP rows and two columns. The card containing these two numbers will be identified in the sequel as INELEM(K,1),INELEM(K,2)

where K refers to the isolation element number (1 to NP), INELEM(K,1) denotes whether the element is uniaxial (unidirectional) or biaxial (bidirectional). INELEM(K,2) denotes the type of element :

- INELEM(K,1)=1 for uniaxial element in the X direction
- INELEM(K,1)=2 for uniaxial element in the Y direction
- INELEM(K,1)=3 for biaxial element
- INELEM(K,2)=1 for linear elastic element
- INELEM(K,2)=2 for viscous element
- INELEM(K,2)=3 for hysteretic element for elastomeric bearings/steel dampers
- INELEM(K,2)=4 for hysteretic element for flat sliding bearings (friction force and f_{max} independent of instant changes in normal force)
- INELEM(K,2)=5 for hysteretic element for flat sliding bearings (friction force and f_{max} depend on instant changes in normal force)
- INELEM(K,2)=6 for FPS bearing element
- INELEM(K,2)=7 for stiffening hysteretic element
- Note: 1. Uniaxial element refers to the element in which biaxial interaction between the forces in the X and Y directions is neglected rendering the interaction surface to be square, instead of the circular interaction surface for the biaxial case.
 2. If NP equals zero then skip Section C.6.

C.6.1 Linear Elastic Element

One card INELEM(K,1),INELEM(K,2)

INELEM(K,1) = 1 or 2 or 3

INELEM(K,2) = 1 (Refer to C.6 for further details).

One card PS(K,1),PS(K,2)

- PS(K,1)= Shear stiffness in the X direction for biaxial element or uniaxial element in the X direction (leave blank if the uniaxial element is in the Y direction only).
- PS(K,2)= Shear stiffness in the Y direction for biaxial element or uniaxial element in the Y direction (leave blank if the uniaxial element is in the X direction only).
- Note: 1. Biaxial element means elastic stiffness in both X and Y directions (no interaction between forces in X and Y direction).

C.6.2 Viscous Element

One card INELEM(K,1),INELEM(K,2)

INELEM(K,1) = 1 or 2 or 3

INELEM(K,2) = 2 (Refer to C.6 for further details).

One card PC(K,1),PC(K,2),PC(K,3)

- PC(K,1)= Damping coefficient in the X direction for biaxial element or uniaxial element in the X direction (leave blank if the uniaxial element is in the Y direction only).
- PC(K,2)= Damping coefficient in the Y direction for biaxial element or uniaxial element in the Y direction (leave blank if the uniaxial element is in the X direction only).
- PC(K,3)= Power that velocity is raised (α in the Equations 4.13 and 4.14). Usual values in the range of 0.5 to 1.2. If given value is 1.0 then the linear viscous element is recovered.
- Note: 1. Biaxial element means elastic stiffness in both X and Y directions (no interaction between forces in X and Y direction).

C.6.3 Hysteretic Element for Elastomeric Bearings/Steel Dampers

One card INELEM(K,1),INELEM(K,2)

INELEM(K,1) = 1 or 2 or 3

INELEM(K,2) = 3 (Refer to C.6 for further details).

One card (ALP(K,I),I=1,2),(YF(K,I),I=1,2),(YD(K,I),I=1,2)

- ALP(K,1)= Post-to-preyielding stiffness ratio (leave blank if the uniaxial element is in the Y direction only);
- YF(K,1) = Yield force (leave blank if the uniaxial element is in the Y direction only);
- YD(K,1) = Yield displacement; in the X direction for biaxial element or uniaxial element in the X direction (leave blank if the uniaxial element is in the Y direction only);
- ALP(K,2)= Post-to-preyielding stiffness ratio (leave blank if the uniaxial element is in the X direction only);
- YF(K,2) = Yield force (leave blank if the uniaxial element is in the X direction only);
- YD(K,2) = Yield displacement; in the Y direction for biaxial element or uniaxial element in the Y direction (leave blank if the uniaxial element is in the X direction only).

C.6.4 Biaxial Hysteretic Element for Sliding Bearings (Friction Independent of Instant Change of Normal Load)

One card INELEM(K,1),INELEM(K,2)

INELEM(K,1) = 1 or 2 or 3

INELEM(K,2) = 4 (Refer to C.6 for further details).

One card (FMAX(K,I),I=1,2),(FMIN(K,I),I=1,2),(PA(K,I),I=1,2),(YD(K,I),I=1,2),FN(K)

- FMAX(K,1)= Maximum coefficient of sliding friction (leave blank if the uniaxial element is in the Y direction only);
- FMAX(K,2)= Maximum coefficient of sliding friction (leave blank if the uniaxial element is in the X direction only);
- FMIN(K,1)= Minimum coefficient of sliding friction (leave blank if the uniaxial element is in the Y direction only);
- FMIN(K,2)= Minimum coefficient of sliding friction (leave blank if the uniaxial element is in the X direction only);
- PA(K,1) = Constant which controls the transition of coefficient of sliding friction from maximum to minimum value (leave blank if the uniaxial element is in the Y direction only);
- PA(K,2) = Constant which controls the transition of coefficient of sliding friction from maximum to minimum value (leave blank if the uniaxial element is in the X direction only);
- YD(K,1) = Yield displacement; in the X direction for biaxial element or uniaxial element in the X direction (leave blank if the uniaxial element is in the Y direction only);
- YD(K,2) = Yield displacement; in the Y direction for biaxial element or uniaxial element in the Y direction (leave blank if the uniaxial element is in the X direction only).
- FN(K) = Initial normal force at the sliding interface.

C.6.5 New Biaxial Hysteretic Element for Sliding Bearings (Friction Depends on Instant Change of Normal Load)

One card INELEM(K,1),INELEM(K,2)

INELEM(K,1) = 1 or 2 or 3

INELEM(K,2) = 5 (Refer to C.6 for further details).

One card (FMAX(K,I),I=1,2),(FMIN(K,I),I=1,2),(PA(K,I),I=1,2),(YD(K,I),I=1,2),FN(K)

- FMAX(K,1)= Maximum coefficient of sliding friction at almost zero pressure (f_{max0} in Equation 4.16) (leave blank if the uniaxial element is in the Y direction only);
- FMAX(K,2)= Maximum coefficient of sliding friction at almost zero pressure (f_{max0} in Equation 4.16) (leave blank if the uniaxial element is in the X direction only);
- FMIN(K,1)= Minimum coefficient of sliding friction (independent of pressure) (leave blank if the uniaxial element is in the Y direction only);
- FMIN(K,2)= Minimum coefficient of sliding friction (independent of pressure) (leave blank if the uniaxial element is in the X direction only);
- PA(K,1) = Constant which controls the transition of coefficient of sliding friction from maximum (f_{max}) to minimum (f_{min}) value (leave blank if the uniaxial element is in the Y direction only);
- PA(K,2) = Constant which controls the transition of coefficient of sliding friction from maximum (f_{max}) to minimum (f_{min}) value (leave blank if the uniaxial element is in the X direction only);
- YD(K,1) = Yield displacement; in the X direction for biaxial element or uniaxial element in the X direction (leave blank if the uniaxial element is in the Y direction only).
- YD(K,2) = Yield displacement; in the Y direction for biaxial element or uniaxial element in the Y direction (leave blank if the uniaxial element is in the X direction only).
- FN(K) = Initial normal force at the sliding interface (static condition).

C.6.6 Element for Friction Pendulum Bearing (FPS)

One card INELEM(K,1),INELEM(K,2)

INELEM(K,1) = 1 or 2 or 3

INELEM(K,2) = 6 (Refer to C.6 for further details).

One card ALP(K,3),(FMAX(K,I),I=1,2),(FMIN(K,I),I=1,2),(PA(K,I),I=1,2),(YD(K,I),I=1, 2),FN(K)

- ALP(K,3) = Radious of curvature of the concave surface of the bearing;
- FMAX(K,1)= Maximum coefficient of sliding friction at almost zero pressure (f_{max0} in Equation 4.16) (leave blank if the uniaxial element is in the Y direction only);
- FMAX(K,2)= Maximum coefficient of sliding friction at almost zero pressure (f_{max0} in Equation 4.16)(leave blank if the uniaxial element is in the X direction only).;
- FMIN(K,1)= Minimum coefficient of sliding friction (independent of pressure)(leave blank if the uniaxial element is in the Y direction only);
- FMIN(K,2)= Minimum coefficient of sliding friction (independent of pressure)(leave blank if the uniaxial element is in the X direction only).;
- PA(K,1) = Constant which controls the transition of coefficient of sliding friction from maximum (f_{max}) to minimum (f_{min}) value (leave blank if the uniaxial element is in the Ydirection only);
- PA(K,2) = Constant which controls the transition of coefficient of sliding friction from maximum (f_{max}) to minimum (f_{min}) value (leave blank if the uniaxial element is in the X direction only).;
- YD(K,1) = Yield displacement; in the X direction for biaxial element or uniaxial element in the X direction (leave blank if the uniaxial element is in the Y direction only);
- YD(K,2) = Yield displacement; in the Y direction for biaxial element or uniaxial element in the Y direction (leave blank if the uniaxial element is in the X direction only).
- FN(K) = Initial normal force at the sliding interface (static condition).

C.6.7 Stiffening Biaxial Hysteretic Element

One card INELEM(K,1),INELEM(K,2)

INELEM(K,1) = 1 or 2 or 3

INELEM(K,2) = 7 (Refer to C.6 for further details).

One card ALP(K,3),ALP(K,4),ALP(K,5),ALP(K,6),ALP(K,7),YD(K,1)

- ALP(K,3) = Characteristic strength (Q of Equation 4.6);
- ALP(K,4) = Tangent stiffness K₁ (see Equation 4.1);
- ALP(K,5) = Tangent stiffness K₂ (see Equation 4.1);
- $ALP(K,6) = Displacement limit D_1 (see Equation 4.1);$
- $ALP(K,7) = Displacement limit D_2 (see Equation 4.1);$
- YD(K,1) = Yield displacement;

D.1 EARTHQUAKE DATA

D.2 Unidirectional Earthquake Record

File:WAVEX.DAT

LOR cards X(I),I=1,LOR

X(I) = Unidirectional acceleration component.

Note: 1. If INDGACC as specified in A.4.4 is 1 or 3, then the input will be assumed at an angle XTH specified in A.4.4. If INDGACC as specified in A.4.4 is 2 or 4, then X(LOR) is considered to be the X component of the bidirectional earthquake.

D.3 Earthquake Record in the Y Direction for the Bidirectional Earthquake

File:WAVEY.DAT (Input only if INDGACC = 2 or 4)

LOR cards Y(I,1),I=1,LOR

Y(I,1) = Acceleration component in the Y direction.

D.4 Earthquake Record in the Z (Vertical) Direction

File:WAVEZ.DAT (Input only if INDGACC = 3 or 4)

LOR cards Y(I,2),I=1,LOR

Y(I,2) = Acceleration component in the Z direction.

E.1 OUTPUT DATA

E.2 Output Parameters

One card LTMH,KPD,IPROF

- LTMH = 1 for both the time history and peak response output.
- LTMH = 0 for only peak response output.
- KPD = No. of time steps before the next response quantity is output.
- IPROF= 1 for accelerations-displacements profiles output.
- IPROF= 0 for no accelerations-displacements profiles output.

E.3 Isolator output

INP cards IP(I),I=1,INP

- IP(I) = Bearing number of bearings I at which the force and displacement response is desired.
- Note: 1. If INP equals zero then skip Section E.3.

E.4 Interstory drift output

The following set of cards must be imported as many times as the number of superstructures NB.

One card ICOR(I),I=1,NB

ICOR(I)= Number of column lines of superstructure I at which the interstory drift is desired. ICOR(I) cards CORDX(K),CORDY(K),K=1,ICOR(I)

CORDX(K)= X coordinate of the column line at which the interstory drift is desired.

CORDY(K)= Y coordinate of the column line at which the interstory drift is desired.

Note: 1. Maximum number of columns at which drift output may be requested is limited to six for each superstructure (maximum value for ICOR(I) is six)
2. The coordinates of the column lines are with respect to the reference axis at the center of mass of the base.

APPENDIX B

INPUT-OUTPUT OF 3D-BASIS-ME

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HIGH DAMPING RUBBER BEARING SYSTEM

WITHOUT VERTICAL GROUND MOTION AND OVERTURNING MOMENT EFFECTS

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INPUT
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secs

52 IMPROVED DAMPING RUBBER BEARINGS in Kips/in*sect2 1 2 5 5 386.22 1 3 1 3 0.005 10 1 500 1 0.5 0.25 2 0.02 2000 0 1 35.12 35.12 3659.0059 0.0 0.0 42.24794 1700000.00000 0.005 0.005 0.005 0.0 0.0 284.40, 18.00 46716.90 46716.90 6618227.50 0.0 0.0 31.07037 1700000.00000 0.02 0.02 0.02 0.0 0.0 210.48, 18.00 00000 9.90109 427528.00000 00000 -834.96 0.00 0.00 -820.56 0.00 806.16 0.00 820.56 -14.40 0.00 3 7 85.735 48.425 96.85 11.40 11.88 0.57 37 85.735 48.425 96.85 11.40 11.88 0.57 3 7 48.425 96.85 11.40 11.88 0.57 85.735 37 85.735 48.425 96.85 11.40 11.88 0.57 37 342.94 193.7 387.4 11.40 11.88 0.57 021 1 2 3 4 5 1 -834.96 0.00 1 -14.40 0.00

WAVEX.DAT

-0.30719E+01	-0.10240E+02	-0.48048E+01	0.10200E+02	-0.28356E+01
-0.81918E+01	0.12051E+02	-0.10161E+02	-0.13430E+02	0.14611E+02
-0.11618E+02	-0.13430E+02	0.14257E+02	-0.98065E+01	-0.72072E+01
0.14532E+02	0.12248E+02	-0.56712E+01	-0.27962E+01	0.57500E+01
-0.14178E+01	0.15753E+01	0.20755E+02	0.11027E+01	-0.28159E+02
0.72072E+01	0.40526E+02	0.28317E+02	0.46866E+01	-0.10003E+02
-0.23394E+02	-0.17053E+02	0.63014E+00	0.17014E+02	0.12997E+02
0.90582E+00	0.94520E+00	-0.58681E+01	-0.23630E+01	0.88613E+01
0.16108E+02	-0.15753E+01	-0.32058E+02	-0.88219E+01	0.91764E+01
0.40959E+01	0.36666E+02	0.31389E+02	-0.16738E+02	-0.37139E+02
-0.10318E+02	0.47654E+01	0.83493E+01	0.22764E+02	0.98459E+01

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WAVEY.DAT

0.12878E+02	0.22055E+01	-0.22842E+01	-0.16935E+01	0.88219E+01
0.31507E+01	-0.20361E+02	-0.16502E+02	0.46079E+01	0.78767E+01
0.36233E+01	-0.14572E+02	-0.10122E+02	0.16187E+02	0.19652E+02
-0.66952E+00	-0.94914E+01	-0.26781E+01	-0.24418E+01	-0.13587E+02
-0.14926E+02	0.12721E+02	0.29538E+02	0.83493E+01	-0.24851E+02
-0.17526E+02	0.90582E+01	0.17368E+02	-0.60257E+01	-0.21779E+02
-0.24378E+02	-0.10437E+02	0.26190E+02	0.50686E+02	0.10082E+02
-0.44582E+02	-0.47772E+02	-0.19731E+02	0.87825E+01	0.41234E+02
0.38753E+02	-0.40565E+01	-0.36745E+02	-0.37375E+02	-0.10673E+02
0.21464E+02	0.27017E+02	0.39384E+01	-0.22527E+02	-0.35209E+02
-0.29301E+02	0.67346E+01	0.46236E+02	0.47890E+02	0.44897E+01

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PROGRAM 3D-BASIS-M.....A GENERAL PROGRAM FOR THE NONLINEAR DYNAMIC ANALYSIS OF THREE DIMENSIONAL BASE ISOLATED MULTIPLE BUILDING STRUCTURES

DEVELOPED BY...P. C. TSOPELAS, S. NAGARAJAIAH , M. C. CONSTANTINOU AND A. M. REINHORN DEPARTMENT OF CIVIL ENGINEERING STATE UNIV. OF NEW YORK AT BUFFALO

NATIONAL CENTER FOR EARTHQUAKE ENGINEERING RESEARCH STATE UNIVERSITY OF NEW YORK, BUFFALO VERSION 3D-BASIS-ME, JANUARY 1993

DEVELOPED BY...P. C. TSOPELAS, M. C. CONSTANTINOU AND A. M. REINHORN DEPARTMENT OF CIVIL ENGINEERING STATE UNIV. OF NEW YORK AT BUFFALO

52 IMPROVED DAMPING RUBBER BEARINGS

UNITS LENGTH : in MASS : Kips/in*sec12 TIME : secs ~ U 19

വ	0.00500		0.50000 0.25000 10.00000 1.00000 500	5		0.02000 2000 1.00000 0.00000
INDÉX = 1 FOR 3D SHEAR BUILDING REPRES. INDEX = 2 FOR FULL 3D REPRESENTATION NUMBER OF ISOLATORS, OUTPUT IS DESIRED=	TIME STEP OF INTEGRATION (NEWMARK)= INDEX FOR TYPE OF TIME STEP=	INDEX = 1 FOR CONSTANT TIME STEP INDEX = 2 FOR VARIABLE TIME STEP	GAMA FOR NEWMARKS METHOD BETA FOR NEWMARKS METHOD TOLERANCE FOR FORCE COMPUTATION REFERENCE MOMENT OF CONVERGENCE MAX NUMBER OF ITERATIONS WITHIN T.S	INDEX FOR GROUND MOTION INPUT=	INDEX = 1 FOR X DIR. INPUT INDEX = 2 FOR X & Y DIR. INPUT INDEX = 3 FOR X & Z DIR INPUT INDEX = 4 FOR X , Y & Z DIR. INPUT	TIME STEP OF RECORD

SUPERSTRUCTURE : 1

.....STIFFNESS DATA.....

0.00001 ECCENT Y ECCENT X 3659.00590 ECCENT X STIFF R STIFFNESS (THREE DIMENSIONAL SHEAR BUILDING) LEVEL STIFF X STIFF Y ROTATIONAL MASS 35.12000 SUPERSTRUCTURE MASS...... 35.12000 -

0.00000

0.00000

0.00000

1700000.00000

42.24794

-

ECCENT Y

SUPERSTRUCTURE DAMPING.....

1 0.00500 2 0.00500

0.00500

ო

HEIGHT.....HEIGHT

284.400 18.000 -0

EIGENVALUES AND EIGENVECTORS (3D SHEAR BUILDING REPRESENTATION)....

PERIOD	0.135432E+03 0.689137E+01 0.689137E+01
EIGENVALUE	0.215236E-02 0.831283E+00 0.831283E+00
MODE NUMBER	- 0 Q

e			
N	0.0000000	0.1538499	0.000000
	0.0000000 0.1538499 0.0000000	0.0000000 0.0000000 0.1538499	0.0007670 0.0000000 0.0000000
- -	0.000000	0.000000	0.0007670
HAPES	×	≻	۲
MODE SHAPES LEVEL	-	-	-

2 SUPERSTRUCTURE :

STIFFNES	STIFFNESS DATA				
SIJFFNESS (1 LEVEL	HKEE UIMENSIUNAL STIFF X	SILFFNESS (THREE DIMENSIONAL SHEAR BUILDING)	STIFF R	ECCENT X	ECCENT Y
÷	46716.90000	46716.90000	6618227.50000	0.00001	0.00000
SUPERSTRUCTUR LEVEL	SUPERSTRUCTURE MASS	ROTATIONAL MASS	ECCENT X	ECCENT Y	
÷	31.07037	1700000.00000	0.00000	0.00000	
SUPERSTRUCTUR MODE SHAPE	SUPERSTRUCTURE DAMPING MODE SHAPE DAMPING RATIO				
- 0 0	0.02000 0.02000 0.02000				

210.480 18.000

- 0

HEIGHT.....HEIGHT

STIFFNESS DATA FOR LINEAR-ELASTIC ISOLATION SYSTEM.....

0.00000	0.00000	0.00000	0.00000	0.00000
STIFFNESS OF LINEAR-ELASTIC SYS. IN X DIR. =	STIFFNESS OF LINEAR ELASTIC SYS. IN Y DIR. =	STIFFNESS OF LINEAR ELASTIC SYS. IN R DIR. =	ECCENT. IN X DIR. FROM CEN. OF MASS=	ECCENT. IN Y DIR. FROM CEN. OF MASS=

MASS AT THE CENTER OF MASS OF THE BASE TRANSL. MASS ROTATIONAL MASS

MASS 9.90109 427528.00000

ЕС≺ ECX GLOBAL ISOLATION DAMPING AT THE CENTER OF MASS OF THE BASE..... X F

0.00000
0.00000
0.00000
0.00000
0.00000
DAMPING

ISOLATORS LOCATION INFORMATION.....

0.0000	-820.5600	0.0000	820.5600	0.0000
-834.9600	0.0000	806.1600	0.0000	- 14 . 4000
-	7	ო	4	പ

.....ELEMENT TYPE =

VIELD DISPL.	0.57000 0.57000 0.57000 0.57000
D2	11.88000 11.88000 11.88000 11.88000 11.88000
10	11.40000 11.40000 11.40000 11.40000 11.40000
IRCE-DISPLACEMENT LOOP PARAMETERS	96.85000 96.85000 96.85000 96.85000 96.85000 387.40000
PLACEMENT LOOP	48.42500 48.42500 48.42500 48.42500 48.42500 193.70000
VONLINEAR ELEMENT FORCE-DISF ISOLATOR CHR. STRENGTH	85.73500 85.73500 85.73500 85.73500 85.73500 342.94000
NONLINEAR ISOLATOR	- 0 0 7 D

TIME HISTORY OPTION=

-

INDEX = O FOR NO TIME HISTORY OUTPUT INDEX = 1 FOR TIME HISTORY OUTPUT

																	MAX RES. DISPL. SQRT(DX12+DY12) TIME DISPLACEMENT 5.570 0.1925E+02 5.575 0.1949E+02 5.575 0.1939E+02 5.570 0.1916E+02
:TORY •••••••= 2 • OPTION= 1) = 1 2 3 4 5	BUILDING REPRESENTATION)	10D	000	ņ	00	18	00	OF MASS OF LEVELS 4 RESPECT TO THE BASE)	Y TIME ROTATION	-02 3.925 0.2539E-03	Y TIME ROTATION	-01 4.7103417E-03	: BASE Y TIME ROTATION -01 3.9252539E-03		MAX DISPL Y TIME X DIRECT Y DIRECT 3.915 - 1564E+02 - 8738E+01 3.915 - 1585E+02 - 8949E+01 3.915 - 1564E+02 - 9153E+01 3.915 - 1564E+02 - 8949E+01 3.915 - 1544E+02 - 8949E+01
TIME HIS	FOR NO PROFILES OUTPUT FOR PROFILES OUTPUT	FORCE-DISPLACEMENT TIME HISTORY DESIRED AT ISOLATORS NUMBERED	AND EIGENVECTORS (3D SHEAR	EIGENVALUE	0.383307E+01 0.318444E+01 0.150358E+04 0.162038E+00 0.150358E+04 0.162038E+00	5	0.0000000 0.1794018 0.0000000	0.0000000 0.0000000 0.1794018	0.0007670 0.0000000 0.0000000	RELATIVE DISPLACEMENTS AT CENTER C (WITH	RE: 1 DISPL X TIME DISPL	0.4041E+02 4.7351209E+02	RE: 2 DISPL X TIME DISPL	1743E+00 3.9009193E	DISPLACEMENTS AT CENTER DF MASS DF B . TIME DISPL X TIME DISPL Y : 5.5651780E+02 3.9158949E+01	ING DISPLACEMENTS	MAX DISPL X TIME X DIRECT Y DIRECT 5.5651780E+027321E+01 5.5651798E+027502E+01 5.5651780E+027502E+01 5.5651762E+027502E+01
NO OF TIME OUTPUT IS DE ACCELERATION	INDEX = 0 F INDEX = 1 F	FORCE-DISPLA AT ISOLATORS	EIGENVALUES	MODE NUMBER	- 0 G	MODE SHAPES Level	• •	1 × 0	1 R O	MAX. RELATIV	SUPERSTRUCTURE LEVEL TIME	1 9.315	SUPERSTRUCTURE LEVEL TIME	1 5.550	MAX. DISPLAC LEVEL TIME BASE 5.565	MAXIMUM BEARING	ISOLATOR TIME 1 5.561 2 5.561 3 5.561 4 5.562

5.575 0.1932E+02

3.915 -.1564E+02 -.8946E+01

5.565 -.1780E+02 -.7499E+01

ີ. ເ PROFILES OF TOTAL ACCELERATION AND DISPLACEMENT AT TIME OF MAX BASE DISPLACEMENTS

MAXIMUM BASE DISPLACEMENT IN X DIRECTION TIME : 5.565

SUPERSTRUCTURE : 1 X

≻

ACCEL -4.4877 113.4474	ACCEL 116.5551 113.4474	N ACCEL -6.2798 130.3809 ACCEL 136.4874 130.3809
DISP 5.3307 -7.5024	۲ DISP -0.0774 -7.5024	Y DIRECTION DISP Y 7.6564 -8.9492 -0.0909 -8.9492 -8.9492
ACCEL -10.2606 252.2362 2	ACCEL 258.9142 252.2362	DISPLACEMENT IN E : 1 312 -23.2372 448 210.4397 E : 2 X ACCEL 513 226.5920 448 210.4397
DISP 12.6955 -17.7974 RUCTURE :	X DISP -0.1724 -17.7974	BASE DISPL 3.915 RUCTURE : DISP 28.1312 -15.6448 RUCTURE : DISP -0.1513 -15.6448
LEVEL DISP 1 12.69 BASE -17.79 SUPERSTRUCTURE	LEVEL 1 BASE	MAXIMUM BASE D TIME : 3.915 SUPERSTRUCTURE LEVEL DISP 1 28.13 BASE -15.64 SUPERSTRUCTURE LEVEL DISP 1 -0.15 BASE -15.64

PROFILES OF TOTAL ACCELERATION AND DISPLACEMENT AT TIME OF MAX ACCELERATION IN EACH BUILDING

SUPERSTRUCTURE : 1

MAX ACCELERATION IN X DIRECTION TIME : 9.300
 LEVEL
 DISP
 ACCEL
 DISP
 ACCEL

 1
 40.3988
 -33.5979
 2.8356
 -2.2247

 BASE
 -7.7057
 102.8309
 -4.2112
 43.0927

MAX ACCELERATION IN Y DIRECTION TIME : 4.725
 LEVEL
 DISP
 ACCEL
 DISP
 ACCEL

 1
 -4.4987
 3.7221
 -12.0864
 10.0587

 BASE
 15.5430
 -214.3719
 7.9607
 -123.5597

SUPERSTRUCTURE : 2

MAX ACCELERATION IN X DIRECTION

TIME : 5.545

ACCEL	112.6542	108.3166	
DISP	-0.0747	-7.3411	
ACCEL	262.1055	244.8050	
DISP	-0.1743	-17.7337	
LEVEL	-	BASE	

MAX ACCELERATION IN Y DIRECTION TIME : 3.900

ACCEL	138.2470	126.1989
DISP	-0.0919	-8.9285
ACCEL	236.9759	211.0198
DISP	-0.1579	- 15.8107
LEVEL	-	BASE

.MAXIMUM STRUCTURAL SHEARS.....

.MAXIMUM STORY SHEARS.....

Z MOMENT TIME RES. SHEAR SQRT(FX12+FY12) 9.295 0.1423E+04 3.675 -.1685E+01 TIME 4.725 0.4250E+03 FORCE Y TIME 9.300 -.1419E+04 SUPERSTRUCTURE : 1 LEVEL TIME FORCE X -

Z MOMENT TIME RES. SHEAR SQRT(FX12+FY12) TIME FORCE Y TIME SUPERSTRUCTURE : 2 LEVEL TIME FORCE X

5.550 0.8879E+04 3.900 0.4295E+04 4.690 0.2291E+04 5.545 0.8144E+04 -

PROFILES OF TOTAL ACCELERATION AND DISPLACEMENT AT TIME OF MAX STRUCT SHEAR IN EACH BUILDING

SUPERSTRUCTURE : 1

MAX STRUC SHEAR IN X DIRECTION TIME : 9.300

ACCEL	-2.2247	43.0927
DISP	2.8356	-4.2112
ACCEL	-33.5979	102.8309
DISP	40.3988	-7.7057
LEVEL	-	BASE

MAX STRUC SHEAR IN Y DIRECTION TIME : 4.725

ACCEL	10.0587	-123.5597
DISP	-12.0864	7.9607
ACCEL	3.7221	-214.3719
DISP	-4.4987	15.5430
LEVEL	-	BASE

SUPERSTRUCTURE : 2

MAX STRUC SHEAR IN X DIRECTION TIME : 5.545

ACCEL	112.6542	108.3166
DISP	-0.0747	-7.3411
ACCEL	262.1055	244.8050
DISP	-0.1743	-17.7337
LEVEL	-	BASE

MAX STRUC SHEAR IN Y DIRECTION TIME : 3.900

ACCEL	138.2470	126.1989
DISP	-0.0919	-8.9285
ACCEL	236.9759	211.0198
DISP	-0.1579	-15.8107
LEVEL	-	BASE

PROFILES OF TOTAL ACCELERATION AND DISPLACEMENT AT TIME OF MAX BASE SHEARS

MAXIMUM BASE SHEAR IN X DIRECTION TIME : 5.555

SUPERSTRUCTURE : 1

							MAX STRUCTURAL SHEAR X DIRECTION	TIME MAX STUCTURAL SHEAR 9.300 -1419.4434	INERTIA FORCES -1419.4434 1018.1382 FORCE AT C.M. OF ENTIRE BASE	TIME MAX STUCTURAL SHEAR 5.545 8143.7152	INERTIA FORCES 8143.7152 2423.8368 FORCE AT C.M. OF ENTIRE BASE	MAX STRUCTURAL SHEAR Y DIRECTION	TIME MAX STUCTURAL SHEAR 4.725 424.9596	INERTIA FORCES 424.9596 -1223.3757 FORCE AT C.M. OF ENTIRE BASE
X DISP ACCEL DISP ACCEL -0.1740 261.5005 -0.0763 114.9470 -17.7815 247.1465 -7.4272 110.0184	BASE SHEAR IN Y DIRECTION 3.900	- ., ?	X A DISP ACCEL DISP ACCEL 28.3480 -23.4494 7.7809 -6.4016 -15.8107 211.0198 -8.9285 126.1989	: 2	X DISP ACCEL DISP ACCEL -0.1579 236.9759 -0.0919 138.2470 -15.8107 211.0198 -8.9285 126.1989	**************************************	MAX OVERTURNING MOMENT X DIRECTION	E DVERTURNING MOMENT 00 -403689.7058	INERTIA FORCES -1419.4434 1018.1382	E DVERTURNING MOMENT 45 1714089.1752	INERTIA FORCES 8143.7152 2423.8368	MAX OVERTURNING MOMENT Y DIRECTION	E DVERTURNING MOMENT 25 120858.5028	INERTIA FORCES 424.9596 -1223.3757
LEVEL D 1 - BASE -1	MAXIMUM BA TIME : 3.	SUPERSTRUCTURE	LEVEL D 1 2 BASE -1	SUPERSTRUCTURE	LEVEL D 1	В	-16	SUPR/STURE TIME 1 9.300	FLOOR 1 BASE	SUPR/STURE TIME 2 5.545	FLOOR 1 BASE	MAX D'	SUPR/STURE TIME 1 4.725	FLOOR 1 BASE

γ DISP 5.2631 -7.4272 DISP LEVEL DISP ACCEL 1 13.0038 -10.5406 BASE -17.7815 247.1465 ACCFI SUPERSTRUCTURE : 2 X LEVEL DISP

7

ACCEL -4.4421 110.0184

×

B-16

FORCE AT C.M. OF ENTIRE BASE	TIME Y DIR Y DIR	
STUCTURAL SHEAR 4295.3853 4295.3853 4295.3853 1249.5070	ME X DIR	
3.900 MAX STUC 3.900 INERTIA	Y DIR TIME Y DIR TIME	
SUPR/STURE TIME OVERTURING MOMENT T. 2 3.900 904092.6905 3 1 1.000 904092.6905 3 1 4295.3853 3 3 1 4295.3853 1249.5070 3	AXIMUM INTERSTORY DRIFT RATIOS' FOR EACH SUPERSTRUCTURE MAXIMUM INTERSTORY DRIFT RATIOS' FOR EACH SUPERSTRUCTURE SUPERSTRUCTURE : 1 CORDINATES OF COLUMN LINES WITH RESPECT TO MASS CENTER OF BASE C/L : 1 X COOR : -834.960 C/L : 1 X COOR : -344.960 C/L : 1 X COOR : -144.00 COLUMN LINES MITH RESPECT TO MASS CENTER OF BASE C/L : 1 X COOR : -144.00 COLUMN LINES COLUMN LINES TIME X DIR TIME X DIR TIME X DIR TIME COLUMN LINES COLUMN LINES COLUMN LINES TEVEL TIME X TIME X DIR TIME X DIR TIME TO COLUMN LINES COLUMN LINES COLUMN LINES TO COLUMN LINES TO COLU	

LOW DAMPING RUBBER BEARING - LINEAR VISCOUS FLUID DAMPER SYSTEM

WITHOUT VERTICAL GROUND MOTION AND OVERTURNING MOMENT EFFECTS

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INPUT
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52 LOW DAMPING RUBBER BEAR. + 24 LINEAR FLUID DAMPERS
                                               secs
                    Kips/in*sect2
in
1 2 9 9 386.22
1 3
1 3
0.005 50 1 500 1
0.5 0.25
2 0.02 2000 0 1
        35.12
        35.12
     3659.0059
0.0
0.0
      42.24794
 1700000.00000
 0.005 0.005 0.005
0.0
0.0
284.40, 18.00
46716.90
     46716.90
    6618227.50
0.0
0.0
        31.07037
   1700000.00000
 0.02 0.02 0.02
0.0
0.0
210.48, 18.00
00000
     9.90109
 427528.00000
7.65, 7.65, 2374932, 0 0
-834.96 0.00
   0.00 -820.56
 806.16 0.00
   0.00 820.56
         0.00
  -14.40
 -509.88
   0.00 -495.48
 481.08 0.00
  0.00 495.48
  3 1
49.595 49.595
  3 1
49.595 49.595
  3 1
49.595 49.595
  3 1
49.595 49.595
  3 1
 198.38 198.38
  32
 10.83 10.83 1.0
  3 2
 10.83 10.83 1.0
  3 2
 10.83 10.83 1.0
  32
 10.83 10.83 1.0
021
 123456789
 1
           0.00
 -834.96
 1
 -14.40
           0.00
```

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WAVEX.DAT

-0.30719E+01	-0.10240E+02	-0.48048E+01	0.10200E+02	-0.28356E+01
-0.81918E+01	0.12051E+02	-0.10161E+02	-0.13430E+02	0.14611E+02
-0.11618E+02	-0.13430E+02	0.14257E+02	-0.98065E+01	-0.72072E+01
0.14532E+02	0.12248E+02	-0.56712E+01	-0.27962E+01	0.57500E+01
-0.14178E+01	0.15753E+01	0.20755E+02	0.11027E+01	-0.28159E+02
0.72072E+01	0.40526E+02	0.28317E+02	0.46866E+01	-0.10003E+02
-0.23394E+02	-0.17053E+02	0.63014E+00	0.17014E+02	0.12997E+02
0.90582E+00	0.94520E+00	-0.58681E+01	-0.23630E+01	0.88613E+01
0.16108E+02	-0.15753E+01	-0.32058E+02	-0.88219E+01	0.91764E+01
0.40959E+01	0.36666E+02	0.31389E+02	-0.16738E+02	-0.37139E+02
-0.10318E+02	0.47654E+01	0.83493E+01	0.22764E+02	0.98459E+01
• • • • • • • • • • • •				

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WAVEY.DAT

0.12878E+02	0.22055E+01	-0.22842E+01	-0.16935E+01	0.88219E+01
0.31507E+01	-0.20361E+02	-0.16502E+02	0.46079E+01	0.78767E+01
0.36233E+01	-0.14572E+02	-0.10122E+02	0.16187E+02	0.19652E+02
-0.66952E+00	-0.94914E+01	-0.26781E+01	-0.24418E+01	-0.13587E+02
-0.14926E+02	0.12721E+02	0.29538E+02	0.83493E+01	-0.24851E+02
-0.17526E+02	0.90582E+01	0.17368E+02	-0.60257E+01	-0.21779E+02
-0.24378E+02	-0.10437E+02	0.26190E+02	0.50686E+02	0.10082E+02
-0.44582E+02	-0.47772E+02	-0.19731E+02	0.87825E+01	0.41234E+02
0.38753E+02	-0.40565E+01	-0.36745E+02	-0.37375E+02	-0.10673E+02
0.21464E+02	0.27017E+02	0.39384E+01	-0.22527E+02	-0.35209E+02
-0.29301E+02	0.67346E+01	0.46236E+02	0.47890E+02	0.44897E+01

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BY...P. C. TSOPELAS, M. C. CONSTANTINOU AND A. M. REINHORN DEPARTMENT OF CIVIL ENGINEERING STATE UNIV. OF NEW YORK AT BUFFALO 3D-BASIS-M.....A GENERAL PROGRAM FOR THE NONLINEAR DYNAMIC ANALYSIS OF THREE DIMENSIONAL BASE ISOLATED MULTIPLE BUILDING STRUCTURES 52 LOW DAMPING RUBBER BEAR. + 24 LINEAR FLUID DAMPERS DEVELOPED BY...P. C. TSOPELAS, S. NAGARAJAIAH , M. C. CONSTANTINOU AND A. M. REINHORN DEPARTMENT OF CIVIL ENGINEERING STATE UNIV. OF NEW YORK AT BUFFALO NATIONAL CENTER FOR EARTHQUAKE ENGINEERING RESEARCH State University of New York, Buffalo 3D-BASIS-ME, JANUARY 1993 ************** DATA************* Kips/in*sect2 secs VERSION ₽ UNITS LENGTH : DEVELOPED PROGRAM MASS TIME

N 0 -

OUTPUT

თ	0.00500		0.50000 0.25000 50.00000 1.00000	7		0.02000 2000 1.00000 0.00000
INDEX = 1 FOR 3D SHEAR BUILDING REPRES. INDEX = 2 FOR FULL 3D REPRESENTATION NUMBER OF ISOLATORS, OUTPUT IS DESIRED. =	TIME STEP OF INTEGRATION (NEWMARK)= INDEX FOR TYPE OF TIME STEP=	INDEX = 1 FOR CONSTANT TIME STEP INDEX = 2 FOR VARIABLE TIME STEP	GAMA FOR NEWMARKS METHOD	INDEX FOR GROUND MOTION INPUT=	INDEX = 1 FOR XDIR. INPUTINDEX = 2 FOR X & YDIR. INPUTINDEX = 3 FOR X & ZDIR INPUTINDEX = 4 FOR X , Y & ZDIR. INPUT	TIME STEP OF RECORD

SUPERSTRUCTURE : 1

.....STIFFNESS DATA.....

F X ECCENT Y	0.00001 0.00000	ίT Υ	0.00000	
ECCENT X	0.0	ECCENT	0.0	
STIFF R	3659 . 00590	ECCENT X	0 . 00000	
STIFFNESS (THREE DIMENSIONAL SHEAR BUILDING) EVEL STIFF X STIFF Y	35.12000	ROTATIONAL MASS	1700000.00000	
HREE DIMENSIONAL STIFF X	35.12000	SUPERSTRUCTURE MASS	42.24794	SUPERSTRUCTURE DAMPING MODE SHAPE DAMPING RATIO
STIFFNESS (T LEVEL	-	SUPERSTRUCTUR LEVEL	-	SUPERSTRUCTURE MODE SHAPE

0.00500 0.00500 - N

0.00500

ო

HEIGHT.....HEIGHT

284.400 18.000 - 0

EIGENVALUES AND EIGENVECTORS (3D SHEAR BUILDING REPRESENTATION)....

PERIOD	0.135432E+03 0.689137E+01 0.689137E+01
EIGENVALUE	0.215236E-02 0.831283E+00 0.831283E+00
MODE NUMBER	- 0 0

	0 0.689137E	
O.831283E+O	0.831283E+00	
2	e	

7	0.1538499 0.0000000
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0.0000000 0.0000000 0.1538499

0.0007670 0.0000000 0.0000000 α -

2 .. SUPERSTRUCTURE

STIFFNI	STIFFNESS DATA				
STIFFNESS (LEVEL	(THREE DIMENSIONAL STIFF X	STIFFNESS (THREE DIMENSIONAL SHEAR BUILDING) EVEL STIFF X STIFF Y	STIFF R	ECCENT X	ECCENT Y
-	46716.90000	46716.90000	6618227.50000	0.00001	0.00000
SUPERSTRUCTI LEVEL	SUPERSTRUCTURE MASS	ROTATIONAL MASS	ECCENT X	ECCENT Y	
-	31.07037	1700000.00000	0.00000	0.00000	
SUPERSTRUCT MODE SHAPE	SUPERSTRUCTURE DAMPING MODE SHAPE DAMPING RATIO				
- 0 0	0.02000 0.02000 0.02000				
HE IGHT LEVEL	· · · · НЕ І GHT				

210.480 18.000

- 0

STIFFNESS DATA FOR LINEAR-ELASTIC ISOLATION SYSTEM.....

0.00000	0.00000	0.00000	0.00000	0.00000
IN X DIR. =	IN Y DIR. =	IN R DIR. =	MASS=	MASS=
STIFFNESS OF LINEAR-ELASTIC SYS. IN X DIR.	STIFFNESS OF LINEAR ELASTIC SYS. IN Y DIR.	STIFFNESS OF LINEAR ELASTIC SYS. IN R DIR.	ECCENT. IN X DIR. FROM CEN. OF MASS=	ECCENT. IN Y DIR. FROM CEN. OF

MASS AT THE CENTER OF MASS OF THE BASE TRANSL. MASS ROTATIONAL MASS

MASS 9.90109 427528.00000

AL ISOLA	TION DAMPING AT X	THE CENTER OI	GLOBAL ISOLATION DAMPING AT THE CENTER OF MASS OF THE BASE E χ		ЕСҮ
DAMPING	7.65000	7.65000	7.65000 2374932.00000	0.00000	0.00000

ISOLATORS LOCATION INFORMATION...... ISOLATOR X Y

B-26

0.0000	-820.5600 0.0000	820.5600	0.0000	0.0000	-495.4800	0.0000	495.4800
-834.9600	0.0000 806.1600	0.0000	- 14 . 4000	-509.8800	0.0000	481.0800	0.0000
- 0	ה ה	4	വ	9	7	80	თ

....ELEMENT TYPE =

• • • • • • • • • • • • • • • • • • • •	STIFFNESS Y	49.59500	49.59500	49.59500	49.59500	198.38000
ELEMENT PARAMETERS	STIFFNESS X	49.59500	49.59500	49.59500	49.59500	198.38000
LINEAR ELASTIC	ISOLATOR	-	2	e	4	ß

....ELEMENT TYPE =

VISCOUS ELEMENT PARAMETERS..... ISOLATOR DAMP-COEF X DAMP-COEF Y POWER OF VELOCITY

υ Γ α σ	0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0	10.83000 10.83000 10.83000 10.83000	0000	10.83000 10.83000 10.83000 10.83000		1.00000 1.00000 1.00000
******	ουτρυτ	S	*****	* *		
TIME HISTORY	ORY OPTION			".	-	
INDEX = INDEX =	O FOR NO TIME HIST 1 FOR TIME HISTORY	oRY OUT	OUTPUT PUT			
ND. OF TI OUTPUT IS ACCELERAT	ND. OF TIME STEPS AT WHICH OUTPUT IS DESIRED ACCELERATION-DISPLACEMENTS	CH TIME HIS	HISTORY	н н • • •	0 -	
INDEX = INDEX =	O FOR NO PROFILES 1 FOR PROFILES OU	ES OUTPUT OUTPUT	Т			
FORCE-DISPLA(AT ISOLATORS	CEMENT TIME NUMBERED	HISTORY	DESIRED	4 6 4	ით 4 თ	വ
EIGENVALUES	JES AND EIGENVECTORS	TORS (3D	SHEAR	BUILDING RI	REPRESENTATION)	ON)
MODE NUMBER	SER EIGENVALUE	ЭE	PERIOD			
- N Q	0.389307E+01 0.150358E+04 0.150358E+04		0.318444E+01 0.162038E+00 0.162038E+00			
MODE SHAPES LEVEL	res 1	0	e			
۲ ۲	0.0000000 0.1	0.1794018 (0.000000			
1 Y	0.0000000 0.0	0.0000000 0	0.1794018			
−	0.0007670 0.0	0.0000000	0.000000			
MAX. RELA	RELATIVE DISPLACEMENTS	AT	CENTER OF MASS (WITH RESPECT	0F TO	LEVELS THE BASE)	
SUPERSTRUCTURE LEVEL TIME	RUCTURE : 1 TIME DISPL X	TIME	DISPL Y	TIME	ROTATION	
۰ ٥	.890 0.3776E+02	4.780 -	7885E+01	3.925 (0.1870E-03	
SUPERSTRUCTURE LEVEL TIME	ICTURE: 2 Me displ x	TIME	DISPL Y	TIME	ROTATION	
- 3.	3.7409760E-01	3.785 -	4893E-01	3.920 (0.2575E-03	

MAX. DISPLACEMENTS AT CENTER OF MASS OF BASE LEVEL TIME DISPL X TIME DISPL Y TIME ROTATION BASE 3.850 -.1111E+02 3.905 -.6094E+01 3.925 -.1869E-03

۲ ACCEL ACCEL Y TIME SUPERSTRUCTURE : 1 LEVEL TIME ACCEL X TIME

9.880 -.3139E+02 4.765 0.6557E+01 3.170 0.5592E-06 -

ACCEL TIME TIME ACCEL Y SUPERSTRUCTURE : 2 LEVEL TIME ACCEL X

α

3.740 0.1468E+03 3.785 0.7357E+02 3.895 -.1006E-02 -

MAX. ACCELERATIONS AT CENTER OF MASS OF BASE LEVEL TIME ACCEL X TIME ACCEL Y TIME ACCEL R BASE 3.720 0.1435E+03 3.825 0.7270E+02 8.525 -.1009E-01

PROFILES OF TOTAL ACCELERATION AND DISPLACEMENT AT TIME OF MAX BASE DISPLACEMENTS

MAXIMUM BASE DISPLACEMENT IN X DIRECTION TIME : 3.850

structure : 1 X Y	DISP ACCEL DISP ACCEL 23.1487 -19.1878 5.1369 -4.2166 -11.1062 126.9532 -5.9316 68.4944	structure : 2 X Y	DISP ACCEL DISP ACCEL -0.0851 107 6845 -0.0463 69 5495	126.9532 -5.9316	MAXIMUM BASE DISPLACEMENT IN Y DIRECTION TIME : 3.905	structure : 1 x Y
SUPERSTRUCTURE	LEVEL D1 1 23 BASE -11	SUPERSTRUCTURE	LEVEL DI	BASE -11	MAXIMUM BAS TIME : 3.9	SUPERSTRUCTURE

	ACCEL	-3.8247	64.0398			ACCEL	63.0781	64.0398
7	DISP	4.7075	-6.0943		>	DISP	-0.0421	-6.0943
	ACCEL	- 18 . 6650	110.4702	2		ACCEL	110.8643	110.4702
×	DISP	22.6028	-10.7910	SUPERSTRUCTURE :	×	DISP	-0.0740	-10.7910
	LEVEL	-	BASE	SUPERST		LEVEL	-	BASE

PROFILES OF TOTAL ACCELERATION AND DISPLACEMENT AT TIME OF MAX ACCELERATION IN EACH BUILDING

SUPERSTRUCTURE : 1

MAX ACCELERATION IN X DIRECTION TIME : 9.880 LEVEL DISP ACCEL DISP ACCEL 1 37.7621 -31.3932 -4.7726 4.0137 BASE 2.8223 -8.1532 0.1258 -7.7618

MAX ACCELERATION IN Y DIRECTION TIME : 4.765
 LEVEL
 DISP
 ACCEL
 DISP
 ACCEL

 1
 3.5186
 -2.9080
 -7.8803
 6.5571

 BASE
 7.6046
 -62.1106
 3.8444
 -48.8624

SUPERSTRUCTURE : 2

MAX ACCELERATION IN X DIRECTION TIME : 3.740 LEVEL DISP ACCEL DISP ACCEL 1 -0.0976 146.7578 -0.0463 69.6603 BASE -9.7787 141.5995 -4.6387 71.8017

MAX ACCELERATION IN Y DIRECTION TIME : 3.785 LEVEL DISP ACCEL DISP ACCEL 1 -0.0945 142.0019 -0.0489 73.5698 BASE -10.6610 140.6156 -5.3194 69.2991

.MAXIMUM STRUCTURAL SHEARS.....

 SUPERST. No
 TIME
 FORCE
 X
 TIME
 FORCE
 Y
 TIME
 Z
 MOMENT

 1
 9.880
 -.1326E+04
 4.765
 0.2770E+03
 3.170
 0.9507E+00

 2
 3.740
 0.4560E+04
 3.7785
 0.2286E+04
 3.895
 -.1710E+04

.MAXIMUM STORY SHEARS......

Z MOMENT TIME RES. SHEAR SQRT(FX+2+FY+2) 9.900 0.1337E+04 3.170 0.9507E+00 TIME 4.765 0.2770E+03 > FORCE TIME 9.880 -.1326E+04 SUPERSTRUCTURE : 1 LEVEL TIME FORCE X -

TIME RES. SHEAR SQRT(FX12+FY12) 3.750 0.5052E+04 3.895 -.1710E+04 Z MOMENT TIME 3.785 0.2286E+04 7 FORCE TIME 3.740 0.4560E+04 SUPERSTRUCTURE : 2 LEVEL TIME FORCE X -

PROFILES OF TOTAL ACCELERATION AND DISPLACEMENT AT TIME OF MAX STRUCT SHEAR IN EACH BUILDING

SUPERSTRUCTURE : 1

MAX STRUC SHEAR IN X DIRECTION TIME : 9.880 LEVEL DISP ACCEL DISP ACCEL 1 37.7621 -31.3932 -4.7726 4.0137 BASE 2.8223 -8.1532 0.1258 -7.7618

MAX STRUC SHEAR IN Y DIRECTION TIME : 4.765 LEVEL DISP ACCEL DISP ACCEL 1 3.5186 -2.9080 -7.8803 6.5571 BASE 7.6046 -62.1106 3.8444 -48.8624

SUPERSTRUCTURE : 2

MAX STRUC SHEAR IN X DIRECTION TIME : 3.740

ACCEL 69.6603 71.8017 DISP -0.0463 -4.6387 MAX STRUC SHEAR IN Y DIRECTION TIME : 3.785 ACCEL 146.7578 141.5995 -0.0976 -9.7787 DISP BASE LEVEL 1

DISP -0.0489 -5.3194 ACCEL 142.0019 140.6156 DISP -0.0945 -10.6610 LEVEL BASE -

ACCEL 73.5698 69.2991

PROFILES OF TOTAL ACCELERATION AND DISPLACEMENT AT TIME OF MAX BASE SHEARS

MAXIMUM BASE SHEAR IN X DIRECTION TIME : 3.740

	ACCEL -4.4510 71.8017		ACCEL 69.6603	71.8017			ACCEL 1.5636	-67.5392		ACCEL -71.4801 -67.5392
2	Y DISP 5.3280 -4.6387	2	DISP -0.0463	-4.6387	DIRECTION	>	DI-SP - 1.9595	5.2116	>	DISP 0.0475 5.2116
-	ACCEL -19.1239 141.5995	2	ACCEL 146.7578	141.5995	≻ NI	.	ACCEL 0.0192	-103.9023	7	ACCEL - 104 . 5 14 1 - 103 . 9023
SUPERSTRUCTURE :	x DISP 22.8825 -9.7787	SUPERSTRUCTURE :	DISP -0.0976	-9.7787	I BASE SHEAR 3.160	SUPERSTRUCTURE :	DISP -0.4681	10.9384	SUPERSTRUCTURE :	DISP 0.0697 10.9384
SUPERST	LEVEL 1 BASE	SUPERST	LEVEL 1	BASE	MAXIMUM TIME :	SUPERST	LEVEL 1	BASE	SUPERST	LEVEL 1 BASE

MAX OVERTURNING MOMENT X DIRECTION

	FORCE AT C.M. OF ENTIRE BASE		FORCE AT C.M. OF ENTIRE BASE	DIRECTION		FORCE AT C.M. OF ENTIRE BASE		FORCE AT C.M. OF ENTIRE BASE				TIME Y DIR	
TIME MAX STUCTURAL SHEAR 9.880 - 1326.2994	INERTIA FORCES -1326.2994 -80.7251	TIME MAX STUCTURAL SHEAR 3.740 4559.8195	INERTIA FORCES 4559.8195 1401.9893	MAX STRUCTURAL SHEAR Y I	TIME MAX STUCTURAL SHEAR 4.765 277.0254	INERTIA FORCES 277.0254 -483.7911	TIME MAX STUCTURAL SHEAR 3.785 2285.8421	INERTIA FORCES 2285.8421 686.1370				Y DIR TIME X DIR	
				DIRECTION					FOR EACH SUPERSTRUCTURE		WITH RESPECT TO MASS CENTER OF BASE	Y DIR TIME X DIR TIME 2924E-01	
OVERTURNING MOMENT -377199.5405	INERTIA FORCES -1326.2994 130.7251	OVERTURNING MOMENT 959750.8119	INERTIA FORCES 4559.8195 1401.9893	MAX DVERTURNING MOMENT Y DI	DVERTURNING MOMENT 78786.0332	INERTIA FORCES 277.0254 -483.7911	DVERTURNING MOMENT 481124.0512	INERTIA FORCES 2285.8421 686.1370	.MAXIMUM INTERSTORY DRIFT RATIOS' FOR EACH	. 	COORDINATES DF COLUMN LINES WITH F C/L : 1 X COOR : -834.960 Y CODR : 0.000	+ + ·	
SUPR/STURE TIME 1 9.880	FLOOR 1 BASE	SUPR/STURE TIME 2 3.740	FLOOR 1 BASE	MAX DVERT	SUPR/STURE TIME 1 4.765	FLOOR 1 BASE	SUPR/STURE TIME E 2 3.785 C	FLOOR A 1 BASE	. MAXIMUM INTER	SUPERSTRUCTURE	COORDINATES DF C/L : 1 X COC Y COC	COLUMN LINES LEVEL TIME 1 9.890 C	SUPERSTRUCTURE

COORDINATES OF COLUMN LINES WITH RESPECT TO MASS CENTER OF BASE

C/L : 1 X COOR : -14.400 Y COOR : 0.000

COLUMN LINES

Y DIR TIME X DIR TIME Y DIR TIME X DIR TIME LEVEL TIME X DIR TIME Y DIR 1 3.740 0.5071E-03 3.790 0.2706E-03 -

LOW DAMPING RUBBER BEARING - LINEAR VISCOUS FLUID DAMPER SYSTEM

GLOBAL REPRESENTATION OF ISOLATION SYSTEM

WITHOUT VERTICAL GROUND MOTION AND OVERTURNING MOMENT EFFECTS

·

E MAX STUCTURAL SHEAR 47 4127.0021	INERTIA FORCES 4127.0021 882.9128 FORCÉ AT C.M. OF ENTIRE BASE	Y DIR TIME X DIR TIME Y DIR	
SUPR/STURE TIME DVERTURNING MOMENT 2 3.847 868651.3953 3.847 3.847	FLOOR INERTIA FORCES 1 4127.0021 BASE 882.9128	MAXIMUM INTERSTORY DRIFT RATIOS' FOR EACH SUPERSTRUCTURE .MAXIMUM INTERSTORY DRIFT RATIOS' FOR EACH SUPERSTRUCTURE SUFERSTRUCTURE : 1 SUPERSTRUCTURE : 1 COLOMN LINES C/L : 1 × COOR : -834.960 C/L : 1 × COOR : -834.960 COLUMN LINES LEVEL TIME * DIR TIME * DIR TIME * DIR TIME SUPERSTRUCTURE : 2 COLUMN LINES WITH RESPECT TO MASS CENTER OF BASE C/L : 1 × COOR : -14.400 C/L : 1 × COOR : -14.400	

INPUT

```
52 LOW DAMPING RUBBER BEAR. + 24 LINEAR FLUID DAMPERS - GLOBAL REPR.
                      Kips/in*sect2
 in
                                          secs
1 2 0 0 386.22
1 3
1 3
0.005 50 1 500 1
0.5 0.25
2 0.02 2000 0 1
         35.12
         35.12
     3659.0059
0.0
0.0
      42.24794
1700000.00000
0.005 0.005 0.005
0.0
0.0
284.40, 18.00
      46716.90
      46716.90
    6618227.50
0.0
0.0
        31.07037
   1700000.00000
0.02 0.02 0.02
0.0
0.0
210.48, 18.00
396.76 396.76 123270280.0
                            14.4 0
     9.90109
427528.00000
50.97 50.97 13010010.6 14.4 0.0
0 2 1
1
-834.96
           0.00
1
-14.40
           0.00
```

WAVEX.DAT

-0.30719E+01	-0.10240E+02	-0.48048E+01	0.10200E+02	-0.28356E+01
-0.81918E+01	0.12051E+02	-0.10161E+02	-0.13430E+02	0.14611E+02
-0.11618E+02	-0.13430E+02	0.14257E+02	-0.98065E+01	-0.72072E+01
0.14532E+02	0.12248E+02	-0.56712E+01	-0.27962E+01	0.57500E+01
-0.14178E+01	0.15753E+01	0.20755E+02	0.11027E+01	-0.28159E+02
0.72072E+01	0.40526E+02	0.28317E+02	0.46866E+01	-0.10003E+02
-0.23394E+02	-0.17053E+02	0.63014E+00	0.17014E+02	0.12997E+02
0.90582E+00	0.94520E+00	-0.58681E+01	-0.23630E+01	0.88613E+01
0.16108E+02	-0.15753E+01	-0.32058E+02	-0.88219E+01	0.91764E+01
0.40959E+01	0.36666E+02	0.31389E+02	-0.16738E+02	-0.37139E+02
-0.10318E+02	0.47654E+01	0.83493E+01	0.22764E+02	0.98459E+01

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WAVEY.DAT

0.12878E+02	0.22055E+01	-0.22842E+01	-0.16935E+01	0.88219E+01
0.31507E+01	-0.20361E+02	-0.16502E+02	0.46079E+01	0.78767E+01
0.36233E+01	-0.14572E+02	-0.10122E+02	0.16187E+02	0.19652E+02
-0.66952E+00	-0.94914E+01	-0.26781E+01	-0.24418E+01	-0.13587E+02
-0.14926E+02	0.12721E+02	0.29538E+02	0.83493E+01	-0.24851E+02
-0.17526E+02	0.90582E+01	0.17368E+02	-0.60257E+01	-0.21779E+02
-0.24378E+02	-0.10437E+02	0.26190E+02	0.50686E+02	0.10082E+02
-0.44582E+02	-0.47772E+02	-0.19731E+02	0.87825E+01	0.41234E+02
0.38753E+02	-0.40565E+01	-0.36745E+02	-0.37375E+02	-0.10673E+02
0.21464E+02	0.27017E+02	0.39384E+01	-0.22527E+02	-0.35209E+02
-0.29301E+02	0.67346E+01	0.46236E+02	0.47890E+02	0.44897E+01

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52 LOW DAMPING RUBBER BEAR. + 24 LINEAR FLUID DAMPERS - GLOBAL REPR. DEVELOPED BY...P. C. TSOPELAS, M. C. CONSTANTINOU AND A. M. REINHORN DEPARTMENT OF CIVIL ENGINEERING STATE UNIV. OF NEW YORK AT BUFFALO 3D-BASIS-M.....A GENERAL PROGRAM FOR THE NONLINEAR DYNAMIC ANALYSIS OF THREE DIMENSIONAL BASE ISOLATED MULTIPLE BUILDING STRUCTURES DEVELOPED BY...P. C. TSOPELAS, S. NAGARAJAIAH , M. C. CONSTANTINOU AND A. M. REINHORN DEPARTMENT OF CIVIL ENGINEERING STATE UNIV. OF NEW YORK AT BUFFALO NATIONAL CENTER FOR EARTHQUAKE ENGINEERING RESEARCH STATE UNIVERSITY OF NEW YORK, BUFFALO JANUARY 1993 3D-BASIS-ME, Kips/in*sect2 secs VERSION <u>_</u> UNITS LENGTH MASS TIME PROGRAM

"		DATA=
GS	ND. OF ISOLATORS	SUPERSTRUCTURE STIFFNESS DATA=
• • • • • • • • •	•	STRUCTURE
NO. OF BUILDINGS	SOLATORS	
NO. OF E	NO. OF 1	INDEX FOR

N 0 -

0	0.00500		0.50000 0.25000 50.00000 1.00000	7		0.02000 2000 1.00000
INDEX = 1 FOR 3D SHEAR BUILDING REPRES. INDEX = 2 FOR FULL 3D REPRESENTATION NUMBER OF ISOLATORS, OUTPUT IS DESIRED=	TIME STEP OF INTEGRATION (NEWMARK)= INDEX FOR TYPE OF TIME STEP=	INDEX = 1 FOR CONSTANT TIME STEP INDEX = 2 FOR VARIABLE TIME STEP	GAMA FOR NEWMARKS METHOD	INDEX FOR GROUND MOTION INPUT=	INDEX = 1 FOR X DIR. INPUT INDEX = 2 FOR X & Y DIR. INPUT INDEX = 3 FOR X & Z DIR INPUT INDEX = 4 FOR X , Y & Z DIR. INPUT	TIME STEP OF RECORD

SUPERSTRUCTURE : 1

....STIFFNESS DATA.....

STIFFNESS (TH LEVEL	REE DIMENSIONAL S STIFF X	STIFFNESS (THREE DIMENSIONAL SHEAR BUILDING) .EVEL STIFF X STIFF Y	STIFF R	ECCENT X	ECCENT Y
.	35.12000	35.12000	3659.00590	0.00001	0.00000
SUPERSTRUCTURE LEVEL	SUPERSTRUCTURE MASS LEVEL TRANSL. MASS	ROTATIONAL MASS	ECCENT X	ECCENT Y	
1 42.24	42.24794	1700000.00000	0.00000	0.00000	

SUPERSTRUCTURE DAMPING.....

1 0.00500 2 0.00500

ო

284.400 18.000

0

-

EIGENVALUES AND EIGENVECTORS (3D SHEAR BUILDING REPRESENTATION)....

PERIOD	0.135432E+03 0.689137E+01 0.689137E+01
EIGENVALUE	0.215236E-02 0.831283E+00 0.831283E+00
MODE NUMBER	- 0 G

MODE SHAPES Level

e		
8	1538499 0.0000000	0.1538499
		0000000 0.0000000
-	0.0000000 0.	0.000000
	×	≻
LEVEL	-	-

1 R 0.0007670 0.0000000 0.0000000

SUPERSTRUCTURE : 2

HEIGHT.....HEIGHT LEVEL HEIGHT

1 210.480 0 18.000

STIFFNESS DATA FOR LINEAR-ELASTIC ISOLATION SYSTEM....

396.76000	396.76000	123270280.00000	14.40000	0.00000
11	H	11	".	".
DIR.	DIR.	DIR.		
×	- > 7	2 2	ASS.	ASS.
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šΥS.	šΥS.	šΥS.	P	P
01 CI	с С	с С	CEN.	CEN.
LAST	LAST	LAST	MOS	SOM
Ē	ω	Ξ	Ē	Ē
STIFFNESS OF LINEAR-ELASTIC SYS. IN X DIR.	STIFFNESS OF LINEAR ELASTIC SYS. IN Y DIR.	STIFFNESS OF LINEAR ELASTIC SYS. IN R DIR.	ECCENT. IN X DIR. FROM CEN. OF MASS	ECCENT. IN Y DIR. FROM CEN. OF MASS=
 11	 11	 	×	≻
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VES!	VES!	VES		
IFFI	IFF	IFF	CEN	CEN
ST	ST	ST	О Ш	О Ш

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MASS AT THE CENTER OF MASS OF THE BASE TRANSL. MASS ROTATIONAL MASS

427528.00000 9.90109 MASS

ECY	0.00000
ECX	14.40000
GLOBAL ISOLATION DAMPING AT THE CENTER OF MASS OF THE BASE	50.97000 13010010.60000
HE CENTER OF	50.97000
TION DAMPING AT T X	50.97000
GLOBAL ISOLA	DAMP I NG

Ŧ		0 -			REPRESENTATIO
TIME HISTORY OPTION=	INDEX = 0 FOR NO TIME HISTORY OUTPUT INDEX = 1 FOR TIME HISTORY OUTPUT	NO. OF TIME STEPS AT WHICH TIME HISTORY OUTPUT IS DESIRED= ACCELERATION-DISPLACEMENTS PROFILES OPTION=	INDEX = 0 FOR NO PROFILES OUTPUT INDEX = 1 FOR PROFILES OUTPUT	FORCE-DISPLACEMENT TIME HISTORY DESIRED AT ISOLATORS NUMBERED	EIGENVALUES AND EIGENVECTORS (3D SHEAR BUILDING REPRESENTATION

.... (NC

ŭ		
EIGENVALUES AND EIGENVECTORS (3D SHEAR BUILDING	PERIOD	0.318444E+01 0.162038E+00 0.162038E+00
AND EIGENVECTORS	EIGENVALUE	0.389307E+01 0.150358E+04 0.150358E+04
EIGENVALUES	MODE NUMBER	- N Ø

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2

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MODE SHAPES LEVEL

. 0000000
0
1794018
•
0
0.0000000
×

- 1 Y 0.000000 0.000000 0.1794018
- 1 R 0.0007670 0.0000000 0.0000000

MAX. RELATIVE DISPLACEMENTS AT CENTER OF MASS OF LEVELS
(WITH RESPECT TO THE BASE)

SUPERSTRUCTURE : 1 LEVEL TIME DISPL X TIME DISPL Y TIME ROTATION 1 9.890 0.3779E+02 4.780 -.7865E+01 3.885 -.2904E-03

SUPERSTRUCTURE : 2 LEVEL TIME DISPL X TIME DISPL Y TIME ROTATION 1 3.735 -.9751E-01 3.780 -.4862E-01 3.875 -.3981E-03

MAX. DISPLACEMENTS AT CENTER DF MASS DF BASE LEVEL TIME DISPL X TIME DISPL Y TIME ROTATION BASE 3.850 -.1108E+02 3.905 -.6082E+01 3.885 0.2902E-03

MAXIMUM BEARING DISPLACEMENTS

0.000 0.0000E+00 TIME MAX DISPL Y TIME X DIRECT Y DIRECT 0.000 0.0000E+00 0.0000E+00 MAX DISPL X TIME X DIRECT Y DIRECT 0.000 0.0000E+00 0.0000E+00 ISOLATOR TIME

MAX RES. DISPL. SQRT(DX↑2+DY↑2) DISPLACEMENT

MAXIMUM BEARING VELOCITIES

MAX VELOCITY X ISOLATOR TIME X DIRECT Y DIRECT TIME X DIRECT Y DIRECT 1 0.000 0.0000E+00 0.0000E+0000E+0000E+0000E+0000E+0000E+0000E+0000E+0000E+0000E+00000E+00000E+0000E+0000E

MAX RES. VELDCITY SQRT(VX12+VY12)

TIME VELDCITY 0.000 0.0000E+00

MAXIMUM BEARING FORCES

0.000 0.0000E+00 TIME TIME X DIRECT Y DIRECT 0.000 0.0000E+00 0.0000E+00 ≻ MAX FORCE MAX FORCE X ISOLATOR TIME X DIRECT Y DIRECT 1 0.000 0.0000E+00 0.0000E+00

MAX RES. FORCE SQRT(FX12+FY12) FORCE

MAX. TOTAL ACCELERATIONS AT CENTER OF MASS OF LEVELS

SUPERSTRUCTURE : 1 LEVEL TIME ACCEL X TIME ACCEL Y TIME ACCEL R 1 9.880 -.3142E+02 4.770 0.6541E+01 3.660 0.9082E-06

SUPERSTRUCTURE : 2 LEVEL TIME ACCEL X TIME ACCEL Y TIME ACCEL R

1 3.735 0.1466E+03 3.780 0.7311E+02 3.855 0.1555E-02

MAX. ACCELERATIONS AT CENTER OF MASS OF BASE LEVEL TIME ACCEL X TIME ACCEL Y TIME ACCEL R BASE 3.770 0.1442E+03 3.815 0.7241E+02 8.520 0.1833E-01 PROFILES OF TOTAL ACCELERATION AND DISPLACEMENT AT TIME OF MAX BASE DISPLACEMENTS

		ACCEL -4.2100 67.4256		ACCEL 69.3254 67.4256			ACCEL -3.8195 63.5302		ACCEL 62.7202 63.5302
X DIRECTION	:	y DISP 5.1287 -5.9176	:	v DISP -0.0461 -5.9176	Y DIRECTION	>	DISP 4.7008 -6.0816	2	DISP -0.0418 -6.0816
DISPLACEMENT IN	-	ACCEL - 19.1693 128.0581	2	ACCEL 126.4401 128.0581	DISPLACEMENT IN	Ŧ	ACCEL - 18.6530 108.6323	2	ACCEL 110.8292 108.6323
BASE 3.850	SUPERSTRUCTURE :	x DISP 23.1247 -11.0773	SUPERSTRUCTURE	x DISP -0.0843 -11.0773	BASE 3.905	SUPERSTRUCTURE :	DISP 22.5865 -10.7694	SUPERSTRUCTURE :	DISP -0.0740 -10.7694
MAXIMUM TIME :	SUPERST	LEVEL 1 BASE	SUPERST	LEVEL 1 BASE	MAXIMUM TIME :	SUPERST	LEVEL 1 BASE	SUPERST	LEVEL 1 BASE

PROFILES OF TOTAL ACCELERATION AND DISPLACEMENT AT TIME OF MAX ACCELERATION IN EACH BUILDING

SUPERSTRUCTURE : 1

MAX ACCELERATION IN X DIRECTION TIME : 9.880 LEVEL DISP ACCEL DISP ACCEL 1 37.7935 -31.4193 -4.7617 4.0047 BASE 2.7896 -3.7790 0.1157 -7.2207

-

MAX ACCELERATION IN Y DIRECTION TIME : 4.770 LEVEL DISP ACCEL DISP ACCEL 1 3.5225 -2.9124 -7.8634 6.5410 BASE 7.5914 -65.4995 3.8376 -48.1184

SUPERSTRUCTURE : 2

MAX ACCELERATION IN X DIRECTION TIME : 3.735

ACCEL	69.5194	70.9963	
DISP	-0.0462	-4.5405	
ACCEL	146.6272	139.0588	
DISP	-0.0975	-9.6144	
LEVEL	-	BASE	

MAX ACCELERATION IN Y DIRECTION TIME : 3.780

ACCEL 73.1071	68.9791
DISP -0.0486	-5.2394
ACCEL 140.9062	143.2994
DISP -0.0938	- 10.5507
LEVEL 1	BASE

.MAXIMUM STRUCTURAL SHEARS.....

SUPERST. No TIME FORCE X TIME FORCE Y TIME Z MOMENT 1 9.880 - 1327E+04 4.770 0.2763E+03 3.660 0.1544E+01 2 3.735 0.4556E+04 3.780 0.2271E+04 3.855 0.2644E+04

.MAXIMUM BASE SHEARS.....

TIME FORCE X TIME FORCE Y TIME Z MOMENT TIME RES. SHEAR SQRT(FX12+FY12) 3.735 -.5128E+04 3.155 0.2803E+04 8.525 -.7353E+04 3.745 0.5792E+04

MAXIMUM STORY SHEARS......

TIME RES. SHEAR SQRT(FX12+FY12) 9.905 0.1338E+04 MOMENT 3.660 0.1544E+01 N TIME ≻ FORCE TIME SUPERSTRUCTURE : 1 LEVEL TIME FORCE X

4.770 0.2763E+03 9.880 -.1327E+04 -

Z MOMENT TIME RES. SHEAR SQRT(FX12+FY12) TIME FORCE Y TIME SUPERSTRUCTURE : 2 LEVEL TIME FORCE X

3.740 0.5045E+04 3.855 0.2644E+04 3.780 0.2271E+04 3.735 0.4556E+04 -

PROFILES OF TOTAL ACCELERATION AND DISPLACEMENT AT TIME OF MAX STRUCT SHEAR IN EACH BUILDING

SUPERSTRUCTURE : 1

MAX STRUC SHEAR IN X DIRECTION TIME : 9.880
 LEVEL
 DISP
 ACCEL
 DISP
 ACCEL
 ISP
 ACCEL
 ISP
 ACCEL
 ISP
 ACCEL
 ACCEL</

SUPERSTRUCTURE : 2

MAX STRUC SHEAR IN X DIRECTION TIME : 3.735 LEVEL DISP ACCEL DISP

ACCEL

69.5194 70.9963 -0.0462 -4.5405 146.6272 139.0588 -0.0975 -9.6144 BASE -

MAX STRUC SHEAR IN Y DIRECTION TIME : 3.780

ACCEL 73.1071 68.9791 DISP -0.0486 -5.2394 ACCEL 140.9062 143.2994 DISP -0.0938 -10.5507 LEVEL BASE

PROFILES OF TOTAL ACCELERATION AND DISPLACEMENT AT TIME OF MAX BASE SHEARS

MAXIMUM BASE SHEAR IN X DIRECTION TIME : 3.735

ACCEL 1.5830 -70.5990 ACCEL -4.4373 70.9963 ACCEL 69.5194 70.9963 > DISP 5.3075 -4.5405 > > DISP -0.0462 -4.5405 -1.9796 5.1257 DISP MAXIMUM BASE SHEAR IN Y DIRECTION TIME : 3.155 ACCEL - 19.0552 139.0588 ACCEL 0.1722 -104.6614 ACCEL 146.6272 139.0588 2 × × × DISP 22.7904 -9.6144 DISP -0.0975 -9.6144 DISP -0.6457 10.9257 SUPERSTRUCTURE : SUPERSTRUCTURE : SUPERSTRUCTURE : BASE LEVEL BASE BASE LEVEL LEVEL

-69.8758 -70.5990 ACCEL ≻ DISP ACCEL - 105 . 4977 - 104 . 66 14 2 DISP 0.0704 10.9257 SUPERSTRUCTURE : LEVEL

0.0465 5.1257 BASE

MAX OVERTURNING MOMENT X DIRECTION

OVERTURNING MOMENT -377512.6513

SUPR/STURE TIME 1 9.880

9.880 TIME

MAX STRUCTURAL SHEAR X DIRECTION

MAX STUCTURAL SHEAR -1327.4003

FORCE AT C.M. OF ENTIRE BASE		FORCE AT C.M. OF ENTIRE BASE	DIRECTION		FORCE AT C.M. OF ENTIRE BASE		FORCE AT C.M. OF ENTIRE BASE					R TIME Y DIR		
INERTIA FORCES -1327.4003 -37.4167	MAX STUCTURAL SHEAR 4555.7605	INERTIA FORCES 4555.7605 1376.8334	MAX STRUCTURAL SHEAR Y	MAX STUCTURAL SHEAR 276.3457	INERTIA FORCES 276.3457 -476.4251	MAX STUCTURAL SHEAR 2271.4638	INERTIA FORCES 2271.4638 682.9684					DIR TIME X DIR		
	TIME 3.735		DIRECTION	TIME 4.770		TIME 3.780		EOR FACH SUPFRSTRUCTURE	:		WITH RESPECT TO MASS CENTER OF BASE	Y DIR TIME X DIR TIME Y 3007E-01		RESPECT TO MASS CENTER OF BASE
INERTIA FORCES -1327.4003 -37.4167	TIME OVERTURNING MOMENT 3.735 958896.4609	INERTIA FORCES 4555.7605 1376.8334	MAX OVERTURNING MOMENT Y D	TIME OVERTURNING MOMENT 4.770 78592.7073	INERTIA FORCES 276.3457 -476.4251	TIME OVERTURNING MOMENT 3.780 478097.6979	INERTIA FORCES 2271.4638 682.9684	MAXIMUM INTEDSTODY DDIFT DATIOS' FOR FAC		CTURE : 1	COORDINATES DF COLUMN LINES WITH C/L : 1 X COOR : -834.960 Y COOR : 0.000	LINES 1 TIME X DIR TIME Y DIR 9.890 0.1419E+00 4.780 0.3007E-01		COORDINATES OF COLUMN LINES WITH RESPECT C/L : 1 X COOR : -14.400 Y COOR : 0.000
FLOOR 1 BASE	SUPR/STURE TIM 2 3.7	FLOOR 1 BASE	MAX C	SUPR/STURE TIME 1 4.770	FLOOR 1 BASE	SUPR/STURE TIME 2 3.780	B-49		MOMIXAM.	SUPERSTRUCTURE	COORDINAT	COLUMN LINES LEVEL TIME 1 9.890	SUPERSTRUCTURE	COORDINAT C/L : 1

COLUMN LINES

-

DIR ≻ TIME X DIR TIME DIR ≻ TIME X DIR TIME LEVEL TIME X DIR TIME Y DIR 1 3.735 0.5066E-03 3.780 0.2252E-03

LOW DAMPING RUBBER BEARING - NONLINEAR VISCOUS FLUID DAMPER SYSTEM

WITHOUT VERTICAL GROUND MOTION AND OVERTURNING MOMENT EFFECTS

INPUT

```
52 LOW DAMPING RUBBER BEAR. + 24 NON-LINEAR DAMPERS
                     Kips/in*sect2
                                          Secs
in
1 2 9 9 386.22
1 3
1 3
0.0002 10 100 500 1
0.5 0.25
2 0.02 2000 0 1
         35.12
         35.12
     3659.0059
0.0
0.0
      42.24794
 1700000.00000
 0.005 0.005 0.005
0.0
0.0
284.40, 18.00
46716.90
      46716.90
    6618227.50
0.0
0.0
        31.07037
   1700000.00000
 0.02 0.02 0.02
0.0
0.0
210.48, 18.00
0.0, 0.0, 0.0, 0.0, 0.0
      9.90109
 427528.00000
7.65 7.65 2374932. 0 0
 -834.96
           0.00
   0.00 -820.56
  806.16 0.00
   0.00 820.56
  -14.40
          0.00
 -509.88
          0.00
   0.00 -495.48
  481.08 0.00
   0.00 495.48
   3 1
 49.595 49.595
   3 1
 49.595 49.595
  3 1
 49.595 49.595
  3 1
 49.595 49.595
  3 1
 198.38 198.38
 3 2
80 80 0.5
  32
 80 80 0.5
   32
 80 80 0.5
   3 2
 80 80 0.5
 0 50 1
 1 2 3 4 5 6 7 8 9
 1
 -834.96
            0.00
 1
  -14.40
            0.00
```

WAVEX.DAT

-0.30719E+01	-0.10240E+02	-0.48048E+01	0.10200E+02	-0.28356E+01
-0.81918E+01	0.12051E+02	-0.10161E+02	-0.13430E+02	0.14611E+02
-0.11618E+02	-0.13430E+02	0.14257E+02	-0.98065E+01	-0.72072E+01
0.14532E+02	0.12248E+02	-0.56712E+01	-0.27962E+01	0.57500E+01
-0.14178E+01	0.15753E+01	0.20755E+02	0.11027E+01	-0.28159E+02
0.72072E+01	0.40526E+02	0.28317E+02	0.46866E+01	-0.10003E+02
-0.23394E+02	-0.17053E+02	0.63014E+00	0.17014E+02	0.12997E+02
0.90582E+00	0.94520E+00	-0.58681E+01	-0.23630E+01	0.88613E+01
0.16108E+02	-0.15753E+01	-0.32058E+02	-0.88219E+01	0.91764E+01
0.40959E+01	0.36666E+02	0.31389E+02	-0.16738E+02	-0.37139E+02
-0.10318E+02	0.47654E+01	0.83493E+01	0.22764E+02	0.98459E+01

•••••

WAVEY.DAT

0.12878E+02	0.22055E+01	-0.22842E+01	-0.16935E+01	0.88219E+01
0.31507E+01	-0.20361E+02	-0.16502E+02	0.46079E+01	0.78767E+01
0.36233E+01	-0.14572E+02	-0.10122E+02	0.16187E+02	0.19652E+02
-0.66952E+00	-0.94914E+01	-0.26781E+01	-0.24418E+01	-0.13587E+02
-0.14926E+02	0.12721E+02	0.29538E+02	0.83493E+01	-0.24851E+02
-0.17526E+02	0.90582E+01	0.17368E+02	-0.60257E+01	-0.21779E+02
-0.24378E+02	-0.10437E+02	0.26190E+02	0.50686E+02	0.10082E+02
-0.44582E+02	-0.47772E+02	-0.19731E+02	0.87825E+01	0.41234E+02
0.38753E+02	-0.40565E+01	-0.36745E+02	-0.37375E+02	-0.10673E+02
0.21464E+02	0.27017E+02	0.39384E+01	-0.22527E+02	-0.35209E+02
-0.29301E+02	0.67346E+01	0.46236E+02	0.47890E+02	0.44897E+01

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PROGRAM 3D-BASIS-M.....A GENERAL PROGRAM FOR THE NONLINEAR DYNAMIC ANALYSIS OF THREE DIMENSIONAL BASE ISOLATED MULTIPLE BUILDING STRUCTURES

DEVELOPED BY...P. C. TSOPELAS, S. NAGARAJAIAH , M. C. CONSTANTINOU AND A. M. REINHORN DEPARTMENT OF CIVIL ENGINEERING STATE UNIV. OF NEW YORK AT BUFFALO

NATIONAL CENTER FOR EARTHOUAKE ENGINEERING RESEARCH STATE UNIVERSITY OF NEW YORK, BUFFALO VERSION 3D-BASIS-ME, JANUARY 1993

DEVELOPED BY...P. C. TSOPELAS, M. C. CONSTANTINOU AND A. M. REINHORN DEPARTMENT OF CIVIL ENGINEERING STATE UNIV. OF NEW YORK AT BUFFALO

52 LOW DAMPING RUBBER BEAR. + 24 NON-LINEAR DAMPERS

UNITS LENGTH : in MASS : Kips/in*sect2 TIME : secs *************INPUT DATA************

NO. OF BUILDINGS...... NO. OF ISOLATORS...... INDEX FOR SUPERSTRUCTURE STIFFNESS DATA=

N 0 -

თ	0.00020 1		0.50000 0.25000 10.00000 100.00000 500	7		0.02000 2000 1.00000 0.00000
INDEX = 1 FOR 3D SHEAR BUILDING REPRES. INDEX = 2 FOR FULL 3D REPRESENTATION NUMBER OF ISOLATORS, OUTPUT IS DESIRED=	TIME STEP OF INTEGRATION (NEWMARK)= INDEX FOR TYPE OF TIME STEP=	INDEX = 1 FOR CONSTANT TIME STEP INDEX = 2 FOR VARIABLE TIME STEP	GAMA FOR NEWMARKS METHOD BETA FOR NEWMARKS METHOD TOLERANCE FOR FORCE COMPUTATION REFERENCE MOMENT OF CONVERGENCE	INDEX FOR GROUND MOTION INPUT=	INDEX = 1 FOR X DIR. INPUT INDEX = 2 FOR X & Y DIR. INPUT INDEX = 3 FOR X & Z DIR INPUT INDEX = 4 FOR X , Y & Z DIR. INPUT	TIME STEP OF RECORD= LENGTH OF RECORD= LOAD FACTOR= ANGLE OF EARTHQUAKE INCIDENCE=

SUPERSTRUCTURE :

-

.....STIFFNESS DATA.....

STIFFNESS (TH LEVEL	REE DIMENSIONAL STIFF X	STIFFNESS (THREE DIMENSIONAL SHEAR BUILDING) .Evel stiff X stiff y	STIFF R	ECCENT X	ECCENT Y
-	35.12000	35 . 12000	3659.00590	0.00001	0.00000
SUPERSTRUCTURE LEVEL	SUPERSTRUCTURE MASS LEVEL TRANSL. MASS	ROTATIONAL MASS	ECCENT X	ECCENT Y	
-	42.24794	1700000.00000	0.00000	0.00000	
SUPERSTRUCTURE	SUPERSTRUCTURE DAMPING				

SUPERSTRUCTURE DAMPING.....

1 0.00500 2 0.00500

0.00500

ო

HEIGHT.....HEIGHT

284.400 18.000 - 0

EIGENVALUES AND EIGENVECTORS (3D SHEAR BUILDING REPRESENTATION)....

PERIOD	0.135432E+03 0.689137E+01 0.689137E+01
EIGENVALUE	0.215236E-02 0.831283E+00 0.831283E+00
MODE NUMBER	- 0 O

ო 0.0000000 0.1538499 0.0000000 0.0000000 0.0000000 0.1538499 2 MODE SHAPES LEVEL × F ≻

0.0007670 0.0000000 0.0000000 ۲ -

R •• SUPERSTRUCTURE

STIFFNESS DATA	DATA				
STIFFNES'S (THF LEVEL	REE DIMENSIONAL STIFF X	STIFFNES'S (THREE DIMENSIONAL SHEAR BUILDING) Evel stiff x stiff y	STIFF R	ECCENT X	ECCENT Y
.	46716.90000	46716.90000	6618227.50000	0.0001	0.00000
SUPERSTRUCTURE LEVEL	SUPERSTRUCTURE MASS	ROTATIONAL MASS	ECCENT X	ECCENT Y	
÷	31.07037	1700000.00000	0.00000	0.0000	
SUPERSTRUCTURE MODE SHAPE	SUPERSTRUCTURE DAMPING MODE SHAPE DAMPING RATIO				
÷	0.02000				

0.02000 0.02000 0.02000 - N O

HEIGHT HEIGHT.. LEVEL

210.480 18.000 - 0

STIFFNESS DATA FOR LINEAR-ELASTIC ISOLATION SYSTEM.....

0.00000	0.00000	0.00000	0.00000	0.00000
IN X DIR. =	IN Y DIR. =	IN R DIR. =	MASS=	MASS=
STIFFNESS OF LINEAR-ELASTIC SYS. IN X DIR.	STIFFNESS OF LINEAR ELASTIC SYS. IN Y DIR.	STIFFNESS OF LINEAR ELASTIC SYS. IN R DIR.	ECCENT. IN X DIR. FROM CEN. OF MASS=	ECCENT. IN Y DIR. FROM CEN. OF MASS=
STIFFNESS OF	STIFFNESS OF	STIFFNESS OF	ECCENT. IN	ECCENT. IN

MASS AT THE CENTER OF MASS OF THE BASE TRANSL. MASS ROTATIONAL MASS

MASS 9.90109 427528.00000

ECY	0.00000
ECX	0.00000
GLOBAL ISOLATION DAMPING AT THE CENTER OF MASS OF THE BASE X E	7.65000 2374932.00000
THE CENTER OF	7.65000
TION DAMPING AT 1 X	7.65000
GLOBAL ISOLA	DAMPING

ISOLATORS LOCATION INFORMATION.....

B-58

0.000	-820.5600	0.0000	820.5600	0.0000	0.0000	-495.4800	0.0000	495.4800
-834.9600	0.0000	806.1600	0.0000	- 14 . 4000	-509.8800	0.0000	481.0800	0.0000
-	2	ო	4	ŋ	9	7	80	ი

....ELEMENT TYPE =

	STIFFNESS Y	49.59500	49.59500	49.59500	49.59500	198.38000
ELEMENT PARAMETERS	STIFFNESS X	49.59500	49.59500	49.59500	49.59500	198.38000
LINEAR ELASTIC	ISOLATOR	-	2	ო	4	ល

....ELEMENT TYPE =

VISCOUS ELEMENT PARAMETERS..... ISOLATOR DAMP-COEF X DAMP-COEF Y POWER OF VELOCITY

80.00000 0.50000 80.00000 0.50000 80.00000 0.50000 80.00000 0.50000 80.00000 0.50000	H ************************************	OUTPUT PUT HISTORY := 50 ILES OPTION= 1	T Desired	= 1 2 3 4 5 6 7 8 9 Suead Building Deddecentation)		0.318444E+01 0.162038E+00 0.162038E+00	σ	0.0000000	0.1794018	0.0000000	CENTER OF MASS OF LEVELS (WITH RESPECT TO THE BASE)	
6 80.00000 80.00000 8 80.00000 80.00000 9	************ OUTPUT PARAMETERS *** TIME HISTORY OPTION	NO TIME HISTORY TIME HISTORY OUT EPS AT WHICH TIME RED	INDEX = 0 FOR NO PROFILES OUTPUT INDEX = 1 FOR PROFILES OUTPUT FORCE-DISPLACEMENT TIME HISTORY DI	LOULATURS NUMBERED	EIGENVALUE	1 0.389307E+01 0.318 2 0.150358E+04 0.165 3 0.150358E+04 0.165	MODE SHAPES LEVEL 1 2	1 X 0.0000000 0.1794018 0.0	1 Y 0.000000 0.000000 0.	1 R 0.0007670 0.0000000 0.0	MAX. RELATIVE DISPLACEMENTS AT CEN (SUPERSTRUCTURE · 1

B-59

3.825 0.1868E-03

1 10.022 0.3903E+02 11.018 -.5926E+01

3.807 0.2545E-03

3.645 -.5174E-01

3.736 -.9993E-01

-

ROTATION

TIME

DISPL Y

TIME

SUPERSTRUCTURE : 2 LEVEL TIME DISPL X

	ROTATION	35202E+01 3.8251867E-03	
\SE	TIME	3.825	
MASS OF BA	DISPL Y TIME	5202E+01	
NTER OF N		3.843 -	
MAX. DISPLACEMENTS AT CENTER OF MASS OF BASE	LEVEL TIME DISPL X TIME	BASE 3.824 - 1015E+02 3.843	
DISPLAC	TIME	3.82	
MAX.	LEVEL	BASE	

MAXIMUM BEARING DISPLACEMENTS

DISPL. SQRT(DX†2+DY†2)	VELOCITY SQRT(VX†2+VY†2)	FORCE SQRT(FX12+FY12)
MAX RES. DI TIME DISPLACEMENT 3.829 0.1133E+02 3.829 0.1153E+02 3.829 0.11478E+02 3.829 0.1140E+02 3.829 0.1140E+02 3.829 0.1144E+02 3.829 0.1144E+02 3.829 0.1132E+02 3.829 0.1132E+02	MAX RES. VE TIME VELOCITY 3.418 0.5119E+02 3.417 0.5221E+02 3.417 0.5185E+02 3.417 0.5185E+02 3.417 0.5131E+02 3.417 0.5134E+02 3.418 0.5110E+02 3.418 0.5110E+02	MAX RES. FC TIME FURCE
MAX DISPL Y TIME X DIRECT Y DIRECT 3.844 - 1012E+02 - 5046E+01 3.843 - 1012E+02 - 5202E+01 3.843 - 1012E+02 - 5502E+01 3.843 - 1012E+02 - 5199E+01 3.843 - 1012E+02 - 5199E+01 3.843 - 1012E+02 - 5199E+01 3.843 - 1012E+02 - 5107E+01 3.843 - 1012E+02 - 5107E+01 3.843 - 1002E+02 - 5107E+01 3.843 - 1002E+02 - 5202E+01 3.843 - 1003E+02 - 5202E+01 3.843 - 1003E+02 - 5202E+01	MAX VELDCITY Y TIME X DIRECT Y DIRECT 3.5913666E+022381E+02 3.5923707E+022432E+02 3.5923651E+022432E+02 3.5923657E+022432E+02 3.5913653E+022432E+02 3.5923657E+022432E+02 3.5923654E+022432E+02 3.5923654E+022432E+02 3.5923654E+022432E+02 3.5923654E+022432E+02	MAX FORCE Y TIME X DIRECT Y DIRECT
MAX DISPL X ISOLATOR TIME X DIRECT Y DIRECT 1 3.824 1015E+02 5029E+01 2 3.824 1030E+02 5185E+01 3 3.824 1015E+02 5185E+01 3 3.824 1015E+02 5185E+01 4 3.824 1015E+02 5185E+01 5 3.824 1015E+02 5185E+01 6 3.824 1015E+02 5185E+01 7 3.824 1015E+02 5185E+01 8 3.824 1015E+02 5185E+01 8 3.824 1015E+02 5185E+01 9 3.824 1006E+02 5185E+01	MAXIMUM BEARING VELOCITIES MAX VELOCITY X ISOLATOR TIME X DIRECT Y DIRECT 1 3.4544717E+021832E+02 2 3.4514783E+021961E+02 3 3.4544717E+021865E+02 5 3.4544717E+021857E+02 6 3.4544717E+021857E+02 7 3.4524757E+021918E+02 8 3.4544717E+021918E+02 8 3.4544717E+021918E+02 8 3.4544717E+021935E+02 8 3.4574678E+021873E+02	MAXIMUM BEARING FORCES MAX FORCE X ISOLATOR TIME X DIRECT Y DIRECT

SQRT (FX↑
FORCE
MAX RES. TIME FORCE 3.829 0.5618E+03 3.829 0.5618E+03 3.829 0.5581E+03 3.829 0.5585E+03 3.829 0.5585E+03 3.407 0.6618E+03 3.407 0.66618E+03 3.407 0.66618E+03 3.407 0.66618E+03
TIME 3.829 3.829 3.829 3.829 3.407 3.407 3.407 3.407
MAX FORCE Y TIME X DIRECT Y DIRECT 3.8445018E+032503E+03 3.8435094E+032580E+03 3.8435019E+032580E+03 3.8434943E+032580E+03 3.8434943E+032580E+03 3.8432007E+041031E+04 3.5914841E+033920E+03 3.5924835E+033945E+03 3.5524835E+033945E+03 3.5524818E+033945E+03
MAX FORCE Υ TIME X DIRECT Y DIRECT 3.844 5018E+03 2503E+03 3.843 2503E+03 3.843 5094E+03 25580E+03 3.843 2560E+03 3.843 5019E+03 25580E+03 3.25580E+03 3.843 4943E+04 3.843 4943E+03 25580E+03 3.3945E+03 3.3945E+03 3.592 4835E+03 3945E+03 3.3945E+03 3.3945E+03
TIME 3.844 3.843 3.843 3.591 3.592 3.592 3.592 3.592 3.592 3.592 3.592 3.592 3.592 3.592 3.592 3.592 3.592 3.592 3.592 5
X Y DIRECT 2494E+03 2571E+03 2571E+03 2571E+03 3447E+03 3447E+03 3519E+04 3519E+04 3519E+03 3519E+03 3519E+03
MAX FORCE X TIME X DIRECT Y DIRECT 3.824 5033E+03 2494E+03 3.824 5109E+03 2571E+03 3.824 5109E+03 2571E+03 3.824 5033E+03 2646E+03 3.824 4957E+03 2571E+03 3.824 4957E+03 3646E+03 3.824 4957E+03 3646E+03 3.824 5495E+03 3447E+03 3.455 5495E+03 3502E+03 3.455 5495E+03 3519E+03 3.455 5495E+03 3502E+03 3.455 5495E+03 3519E+03 3.455 5495E+03 3462E+03
TIME 3.824 3.824 3.824 3.824 3.824 3.455 3.455 3.455 3.455 3.455
ISOLATOR 1 2 3 3 6 6 8 8 9 8

MAX. TOTAL ACCELERATIONS AT CENTER OF MASS OF LEVELS

SUPERSTRUCTURE : 1 LEVEL TIME ACCEL X TIME ACCEL Y TIME ACCEL

۲

10.011 -.3244E+02 11.009 0.4927E+01 3.663 -.5861E-06 -

ACCEL R TIME TIME ACCEL Y SUPERSTRUCTURE : 2 LEVEL TIME ACCEL X 3.787 -.9945E-03 3.644 0.7780E+02 3.735 0.1503E+03 -

MAX. ACCELERATIONS AT CENTER OF MASS OF BASE LEVEL TIME ACCEL X TIME ACCEL Y TIME ACCEL R BASE 3.699 0.1481E+03 3.679 0.8192E+02 6.081 0.6538E-01

PROFILES OF TOTAL ACCELERATION AND DISPLACEMENT AT TIME OF MAX BASE DISPLACEMENTS

MAXIMUM BASE DISPLACEMENT IN X DIRECTION TIME : 3.824

SUPERSTRUCTURE :

		_		
	×		>	
LEVEL	DISP	ACCEL	DISP	ACCEL
-	22.4005	-18.5478	4.7462	-3.8455
BASE	-10.1484	96.9077	-5.1848	64.8046
SUPFRST	SUPFRSTRUCTURE .	0		
	. ×	4	7	
LEVEL	DISP	ACCEL	DISP	ACCEL
-	-0.0843	126.0661	-0.0437	65.3455
BASE	-10.1484	96.9077	-5.1848	64.8046
MAXIMUN TIME :	MAXIMUM BASE DISPLACEMENT TIME : 3.843	ACEMENT IN	Y DIRECTION	
SUPERST	SUPERSTRUCTURE :	.		

ACCEL -3.6537 41.5514 ACCEL 58.9043 41.5514 > 4.5285 -5.2017 DISP -0.0395 -5.2017 DISP ACCEL -18.3897 105.5173 ACCEL 101.3220 105.5173 2 DISP 22.2317 -10.1186 × × DISP -0.0685 -10.1186 SUPERSTRUCTURE : BASE LEVEL LEVEL -

BASE

PROFILES OF TOTAL ACCELERATION AND DISPLACEMENT AT TIME OF MAX ACCELERATION IN EACH BUILDING

SUPERSTRUCTURE : 1

MAX ACCELERATION IN X DIRECTION TIME : 10.011
 LEVEL
 DISP
 ACCEL
 DISP
 ACCEL

 1
 39.0252
 -32.4439
 -4.9032
 4.0881

 BASE
 2.4990
 -4.3183
 0.1289
 -11.0347

MAX ACCELERATION IN Y DIRECTION TIME : 11.009 LEVEL DISP ACCEL DISP ACCEL 1 23.3868 -19.1713 -5.9254 4.9270 BASE 2.6353 -10.5967 -0.5949 -7.9287

SUPERSTRUCTURE : 2

MAX ACCELERATION IN X DIRECTION TIME : 3.735 LEVEL DISP ACCEL DISP ACCEL 1 -0.0999 150.2559 -0.0511 76.8194 BASE -9.2473 140.0605 -4.6020 75.2855

MAX ACCELERATION IN Y DIRECTION TIME : 3.644
 LEVEL
 DISP
 ACCEL
 DISP
 ACCEL

 1
 -0.0893
 134.6504
 -0.0517
 77.8041

 BASE
 -6.8886
 129.1985
 -3.2762
 66.4948

.MAXIMUM STRUCTURAL SHEARS.....

SUPERST. No TIME FORCE X TIME FORCE Y TIME Z MOMENT 1 10.011 -.1371E+04 11.009 0.2082E+03 3.663 -.9963E+00 2 3.735 0.4669E+04 3.644 0.2417E+04 3.787 -.1691E+04 .MAXIMUM STORY SHEARS......

TIME RES. SHEAR SQRT(FX12+FY12) Z MOMENT TIME > FORCE TIME SUPERSTRUCTURE : 1 LEVEL TIME FORCE X

3.663 -.9963E+00 10.016 0.1382E+04 10.011 -.1371E+04 11.009 0.2082E+03 -

Z MOMENT TIME RES. SHEAR SQRT(FX+2+FY+2) TIME FORCE Y TIME SUPERSTRUCTURE : 2 LEVEL TIME FORCE X

3.732 0.5245E+04 3.787 -.1691E+04 3.644 0.2417E+04 3.735 0.4669E+04 - PROFILES OF TOTAL ACCELERATION AND DISPLACEMENT AT TIME OF MAX STRUCT SHEAR IN EACH BUILDING

SUPERSTRUCTURE : 1

MAX STRUC SHEAR IN X DIRECTION TIME : 10.011 LEVEL DISP ACCEL DISP ACCEL 1 39.0252 -32.4439 -4.9032 4.0881 BASE 2.4990 -4.3183 0.1289 -11.0347

MAX STRUC SHEAR IN Y DIRECTION TIME : 11.009 LEVEL DISP ACCEL DISP ACCEL 1 23.3868 -19.1713 -5.9254 4.9270 BASE 2.6353 -10.5967 -0.5949 -7.9287

SUPERSTRUCTURE : 2

MAX STRUC SHEAR IN X DIRECTION TIME : 3.735

MAX STRUCTURAL SHEAR X DIRECTION

ACCEL 76.3190 76.9049

DISP -0.0508 -4.6699

ACCEL 149.9164 140.3449

DISP -0.0998 -9.3761

LEVEL 1 BASE

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×

MAX OVERTURNING MOMENT X DIRECTION

PROFILES OF TOTAL ACCELERATION AND DISPLACEMENT AT TIME OF MAX BASE SHEARS

ACCEL 76.8194 75.2855

DISP -0.0511 -4.6020

ACCEL 150.2559 140.0605

DISP -0.0999 -9.2473

BASE

LEVEL

MAX STRUC SHEAR IN Y DIRECTION TIME : 3.644

ACCEL 77.8041 66.4948

DISP -0.0517 -3.2762

ACCEL 134.6504 129.1985

DISP -0.0893 -6.8886

LEVEL 1

BASE

		ACCEL -4.4900 74.7148		ACCEL 76.9759	74.7148			ACCEL - 4 . 45 19 76 . 9049	
DIRECTION	:	۲ DISP 5.4449 -4.5787	;	DISP -0.0512	-4.5787	DIRECTION	>		
×	-	ACCEL - 18.7662 140.1913	2	ACCEL 150.2219	140.1913	>	-	ACCEL - 18 . 7936 140 . 3449	c
MAXIMUM BASE SHEAR IN TIME : 3.732	SUPERSTRUCTURE :	A DISP 22.5072 -9.2024	SUPERSTRUCTURE :	A DISP -0.0999	-9.2024	MAXIMUM BASE SHEAR IN TIME : 3.741	SUPERSTRUCTURE :	DISP 22.5554 -9.3761	. 101050570105105
MAXIMUM TIME :	SUPERSI	LEVEL 1 BASE	SUPERS1	LEVEL 1	BASE	MAXIMUN TIME :	SUPERST	LEVEL 1 BASE	CIDEDCT

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2 SUPERSTRUCTURE :

	FORCE AT C.M. OF ENTIRE BASE		FORCE AT C.M. OF ENTIRE BASE	DIRECTION		FORCE AT C.M. OF ENTIRE BASE		FORCE AT C.M. OF ENTIRE BASE					R TIME Y DIR	
MAX STUCTURAL SHEAR -1370.6860	INERTIA FORCES -1370.6860 -42.7563	MAX STUCTURAL SHEAR 4668.5055	INERTIA FORCES 4668.5055 1386.7519	MAX STRUCTURAL SHEAR Y	MAX STUCTURAL SHEAR 208.1539	INERTIA FORCES 208.1539 -78.5025	MAX STUCTURAL SHEAR 2417.4009	INERTIA FORCES 2417.4009 658.3715					DIR TIME X DIR	
TIME 10.011		TIME 3.735		DIRECTION	TIME 11.009		TIME 3.644		FOR EACH SUPERSTRUCTURE		WITH RESPECT TO MASS CENTER OF BASE		Y DIR TIME X DIR TIME Y 2232E-01	
OVERTURNING MOMENT -389823.1014	INERTIA FORCES -1370.6860 -42.7563	DVERTURNING MOMENT 982627.0317	INERTIA FORCES 4668.5055 1386.7519	MAX DVERTURNING MOMENT Y D:	DVERTURNING MOMENT 59198.9824	INERTIA FORCES 208.1539 -78.5025	DVERTURNING MOMENT 508814.5347	INERTIA FORCES 2417.4009 658.3715	.MAXIMUM INTERSTORY DRIFT RATIOS' FOR EACH	E . 1	CDLUMN LINES : -834.960 : 0.000	·	X DIR TIME Y DIR 0.1465E+00 11.008 0.2232E-01	Е: 2
SUPR/STURE TIME 1 10.011	FLOOR 1 BASE	SUPR/STURE TIME 2 3.735	FLOOR 1 BASE	MAX OVER	SUPR/STURE TIME 1 11.009	FLOOR 1 BASE	SUPR/STURE TIME E 3.644	ELOOR BASE 92	.MAXIMUM INTE	SUPERSTRUCTURE	COORDINATES DF C/L : 1 X COOR Y COOR	COLUMN LINES	LEVEL TIME 1 10.022	SUPERSTRUCTURE

COORDINATES OF COLUMN LINES WITH RESPECT TO MASS CENTER OF BASE

C/L : 1 X COOR : -14.400 Y COOR : 0.000

DIR ≻ TIME DIR × TIME DIR ≻ TIME X DIR TIME COLUMN LINES 1 IME 1 DIR TIME Y DIR 1 3.736 0.5192E-03 3.722 0.2854E-03 FPS SYSTEM

WITHOUT VERTICAL GROUND MOTION AND OVERTURNING MOMENT EFFECTS

52 FPS BEARINGS - CONSTANT NORMAL LOAD in Kips/in*sect2 secs 1 2 5 5 386.22 1 3 1 3 0.001 50 1 500 1 0.5 0.25 2 0.02 2000 0 1 35.12 35.12 3659.0059 0.0 0.0 42.24794 1700000.00000 0.005 0.005 0.005 0.0 0.0 284.40, 18.00 46716.90 46716.90 6618227.50 0.0 0.0 31.07037 1700000.00000 0.02 0.02 0.02 0.0 0.0 210.48, 18.00 390.06, 390.06, 121188853.0, -14.40, 0.00 9.90109 427528.00000 00000 -834.96 0.00 0.00 -820.56 806.16 0.00 0.00 820.56 -14.40 0.00 3 4 0.045 0.045 0.03 0.03 0.8 0.8 0.02 0.02 4017.625 3 4 0.045 0.045 0.03 0.03 0.8 0.8 0.02 0.02 4017.625 34 0.045 0.045 0.03 0.03 0.8 0.8 0.02 0.02 4017.625 3 4 0.045 0.045 0.03 0.03 0.8 0.8 0.02 0.02 4017.625 3 4 0.045 0.045 0.03 0.03 0.8 0.8 0.02 0.02 16070.5 0 10 1 1 2 3 4 5 1 -834.96 0.00 1 -14.40 0.00

WAVEX.DAT

-0.30719E+01	-0.10240E+02	-0.48048E+01	0.10200E+02	-0.28356E+01
-0.81918E+01	0.12051E+02	-0.10161E+02	-0.13430E+02	0.14611E+02
-0.11618E+02	-0.13430E+02	0.14257E+02	-0.98065E+01	-0.72072E+01
0.14532E+02	0.12248E+02	-0.56712E+01	-0.27962E+01	0.57500E+01
-0.14178E+01	0.15753E+01	0.20755E+02	0.11027E+01	-0.28159E+02
0.72072E+01	0.40526E+02	0.28317E+02	0.46866E+01	-0.10003E+02
-0.23394E+02	-0.17053E+02	0.63014E+00	0.17014E+02	0.12997E+02
0.90582E+00	0.94520E+00	-0.58681E+01	-0.23630E+01	0.88613E+01
0.16108E+02	-0.15753E+01	-0.32058E+02	-0.88219E+01	0.91764E+01
0.40959E+01	0.36666E+02	0.31389E+02	-0.16738E+02	-0.37139E+02
-0.10318E+02	0.47654E+01	0.83493E+01	0.22764E+02	0.98459E+01

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WAVEY.DAT

0.12878E+02	0.22055E+01	-0.22842E+01	-0.16935E+01	0.88219E+01
0.31507E+01	-0.20361E+02	-0.16502E+02	0.46079E+01	0.78767E+01
0.36233E+01	-0.14572E+02	-0.10122E+02	0.16187E+02	0.19652E+02
-0.66952E+00	-0.94914E+01	-0.26781E+01	-0.24418E+01	-0.13587E+02
-0.14926E+02	0.12721E+02	0.29538E+02	0.83493E+01	-0.24851E+02
-0.17526E+02	0.90582E+01	0.17368E+02	-0.60257E+01	-0.21779E+02
-0.24378E+02	-0.10437E+02	0.26190E+02	0.50686E+02	0.10082E+02
-0.44582E+02	-0.47772E+02	-0.19731E+02	0.87825E+01	0.41234E+02
0.38753E+02	-0.40565E+01	-0.36745E+02	-0.37375E+02	-0.10673E+02
0.21464E+02	0.27017E+02	0.39384E+01	-0.22527E+02	-0.35209E+02
-0.29301E+02	0.67346E+01	0.46236E+02	0.47890E+02	0.44897E+01

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3D-BASIS-M......A GENERAL PROGRAM FOR THE NONLINEAR DYNAMIC ANALYSIS OF THREE DIMENSIONAL BASE ISOLATED MULTIPLE BUILDING STRUCTURES DEVELOPED BY...P. C. TSOPELAS, S. NAGARAJAIAH , M. C. CONSTANTINOU AND A. M. REINHORN DEPARTMENT OF CIVIL ENGINEERING STATE UNIV. OF NEW YORK AT BUFFALO NATIONAL CENTER FOR EARTHQUAKE ENGINEERING RESEARCH STATE UNIVERSITY OF NEW YORK, BUFFALO 3D-BASIS-ME, JANUARY 1993 VERSION PROGRAM

A. M. REINHORN DEVELOPED BY...P. C. TSOPELAS, M. C. CONSTANTINOU AND DEPARTMENT OF CIVIL ENGINEERING STATE UNIV. OF NEW YORK AT BUFFALO FPS BEARINGS - CONSTANT NORMAL LOAD 52

Kips/in*sect2 secs 5 UNITS LENGTH : MASS : TIME : *************INPUT DATA************

4 10 +

ى ا	0.00100		0.50000 0.25000 50.00000 1.00000	2		0.02000 2000 1.00000 0.00000
INDEX = 1 FOR 3D SHEAR BUILDING REPRES. INDEX = 2 FOR FULL 3D REPRESENTATION NUMBER OF ISOLATORS, OUTPUT IS DESIRED=	TIME STEP OF INTEGRATION (NEWMARK)= INDEX FOR TYPE OF TIME STEP=	INDEX = 1 FOR CONSTANT TIME STEP INDEX = 2 FOR VARIABLE TIME STEP	GAMA FOR NEWMARKS METHOD= BETA FOR NEWMARKS METHOD= TOLERANCE FOR FORCE COMPUTATION= REFERENCE MOMENT OF CONVERGENCE= MAX NUMBER OF ITERATIONS WITHIN T.S=	INDEX FOR GROUND MOTION INPUT=	INDEX1 FORXDIR.INDUTINDEX2 FORX& YDIR.INPUTINDEX3 FORX& ZDIRINPUTINDEX4 FORXY& ZDIRINPUT	TIME STEP OF RECORD

SUPERSTRUCTURE : 1

.....STIFFNESS DATA.....

ECCENT Y	0.00000		
ECCENT X	0.00001	ECCENT Y	0.00000
STIFF R	3659 . 00590	ECCENT X	0.00000
STIFFNESS (THREE DIMENSIONAL SHEAR BUILDING) Level stiff x stiff y	35.12000	ROTATIONAL MASS	1700000.00000
REE DIMENSIONAL STIFF X	35.12000	SUPERSTRUCTURE MASS LEVEL TRANSL. MASS	42.24794
STIFFNESS (THI LEVEL	-	SUPERSTRUCTURE LEVEL	÷

SUPERSTRUCTURE DAMPING.....

1 0.00500 2 0.00500

0.00500

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HEIGHT.....HEIGHT

284.400 18.000 -0

EIGENVALUES AND EIGENVECTORS (3D SHEAR BUILDING REPRESENTATION)....

PERIOD	0.135432E+03 0.689137E+01 0.689137E+01
EIGENVALUE	0.215236E-02 0.831283E+00 0.831283E+00
MODE NUMBER	- 0 W

0.0000000 0.1538499 0.0000000 2 -MODE SHAPES LEVEL ×

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0.0000000 0.0000000 0.1538499 ≻

0.0007670 0.0000000 0.0000000 ۵ -

2 SUPERSTRUCTURE :

· · · · STIFFNES	STIFFNESS DATA				
STIFFNESS (TH LEVEL	HREE DIMENSIONAL STIFF X	STIFFNESS (THREE DIMENSIONAL SHEAR BUILDING) Evel stiff x stiff y	STIFF R	ECCENT X	ECCENT Y
-	46716.90000	46716.90000	6618227.50000	0.00001	0.00000
SUPERSTRUCTURE LEVEL	SUPERSTRUCTURE MASS	ROTATIONAL MASS	ECCENT X	ECCENT Y	
-	31.07037	1700000.00000	0.00000	0.00000	
SUPERSTRUCTUR MODE SHAPE	SUPERSTRUCTURE DAMPING MODE SHAPE DAMPING RATIO				
, 0 0 −	0.02000 0.02000 0.02000				

210.480 18.000

- 0

HEIGHT.....HEIGHT

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STIFFNESS DATA FOR LINEAR-ELASTIC ISOLATION SYSTEM.....

390.06000 390.06000 121188853.00000 -14.40000 0.00000
IN X DIR. = IN Y DIR. = IN R DIR. = MASS
×≻×× SS. T SS. T
Z Z Z Z Z Z H H H W W
YS. YS. OF
AST AST OM OM
STIFFNESS OF LINEAR-ELASTIC SYS. IN X DIR. = STIFFNESS OF LINEAR ELASTIC SYS. IN Y DIR. = STIFFNESS OF LINEAR ELASTIC SYS. IN R DIR. = ECCENT. IN X DIR. FROM CEN. OF MASS ECCENT. IN Y DIR. FROM CEN. OF MASS
ĒĒĒ → → → → × ≻
STIFI STIFI STIFI ECCEN

MASS AT THE CENTER OF MASS OF THE BASE TRANSL. MASS ROTATIONAL MASS

MASS 9.90109 427528.00000

	ECΥ
Ε	ECX
GLOBAL ISOLATION DAMPING AT THE CENTER OF MASS OF THE BASE	æ
THE CENTER OF	۶
ON DAMPING AT	×
GLOBAL ISOLATI	

0.00000	
0.00000	
0.00000	
0.00000	
0.00000	
DAMPING	

ISOLATORS LOCATION INFORMATION.....

0.0000	-820.5600	0.0000	820.5600	0.0000
-834.9600	0.0000	806.1600	0.0000	- 14 . 4000
-	2	e	4	വ

....ELEMENT TYPE =

	NORMAL FORCE		4011.62900	4017.62500	4017 62500		00029.1104	16070.50000
	ΥΙΕLD DISPL. Υ		0.02000	0.02000	0.02000		0.020.0	0.02000
DNSTANT NORMAL FORCE & FMAX)	YIELD DISPL. X			0.02000	0.02000		00030.0	0.02000
· · · ·	PA Y	0.800		0.00	0.800	0.800		0.800
& FMAX)	PA X	0.800			0.800	0.800		0.800
DRMAL FORCE	FMIN Y	0.03000	000000		0.03000	0.03000		0.03000
CONSTANT NC	FMIN X	0.03000	0.03000		0.030000	0.03000		0.0000.0
AMETERS(FMAX Y	0.04500	0.04500		0.04000	0.04500	0,04500	0.0420.0
PA	FMAX X	0.04500	0.04500	0.04500	000000	0.04500	0,015,00	0000000
SLIDING B	ISULATUK	-	7	r) .	4	Ľ)

TIME HISTORY OPTION=

0

INDEX = 0 FOR NO TIME HISTORY OUTPUT INDEX = 1 FOR TIME HISTORY OUTPUT

																	MAX RES TIME DISPLACEMEN 3.866 0.1534E+02 3.866 0.1579E+02 3.667 0.1579E+02
NO. OF TIME STEPS AT WHICH TIME HISTORY OUTPUT IS DESIRED	INDEX = O FOR NO PROFILES OUTPUT INDEX = 1 FOR PROFILES OUTPUT	FORCE-DISPLACEMENT TIME HISTORY DESIRED AT ISOLATORS NUMBERED	EIGENVALUES AND EIGENVECTORS (3D SHEAR BUILDING REPRESENTATION)	MODE NUMBER EIGENVALUE PERIOD	1 0.389307E+01 0.318444E+01 2 0.150358E+04 0.162038E+00 3 0.150358E+04 0.162038E+00	MODE SHAPES LEVEL 1 2 3	1 X 0.000000 0.1794018 0.0000000	1 Y 0.0000000 0.0000000 0.1794018	1 R 0.0007670 0.0000000 0.0000000	MAX. RELATIVE DISPLACEMENTS AT CENTER OF MASS OF LEVELS (WITH RESPECT TO THE BASE)	SUPERSTRUCTURE : 1 LEVEL TIME DISPL X TIME DISPL Y TIME ROTATION	1 10.044 0.4053E+02 4.7539815E+01 3.887 0.3999E-03	SUPERSTRUCTURE : 2 LEVEL TIME DISPL V TIME ROTATION	1 3.809 - 1136E+00 3.872 - 8195E-01 4.755 - 5577E-03	MAX. DISPLACEMENTS AT CENTER OF MASS OF BASE LEVEL TIME DISPL X TIME DISPL Y TIME ROTATION BASE 3.8571317E+O2 3.8968243E+O1 3.8873998E-O3	MAXIMUM BEARING DISPLACEMENTS	MAX DISPL X MAX DISPL Y ISOLATOR TIME X DIRECT Y DIRECT TIME X DIRECT Y DIRECT 1 3.8571317E+027837E+01 3.8971303E+027909E+01 2 3.8571350E+028168E+01 3.8961336E+028243E+01 3 3 857 - 1317E+02 - 8168E+01 3 80613036E+02856EE+01

ES. DISPL. SQRT(DX†2+DY†2) ENT 3.867 0.1568E+02 3.866 0.1523E+02 3.896 -.1303E+02 -.8565E+01 3.896 -.1271E+02 -.8243E+01 3.857 - 1317E+02 - 8488E+01 3.856 - 1285E+02 - 8164E+01 ω4

	MAX RES. VELOCITY SQRT(VX†2+VY†2) OCITY 02E+02 14E+02 86E+02 75E+02 44E+02	SQRT (FX†2+FY†2)
	VELOCITY	FORCE
	MAX RES. TIME VELOCITY 3.554 0.5902E+02 3.555 0.5986E+02 3.555 0.5986E+02 3.554 0.5875E+02 3.554 0.5944E+02 3.554 0.5944E+02	MAX RES. TIME FORCE 3.149 0.1808E+03 2.593 0.1808E+03 9.172 0.1808E+03 3.146 0.1808E+03 3.146 0.1808E+03 2.593 0.7232E+03
	MAX VELOCITY Y TIME X DIRECT Y DIRECT 3.5944573E+023295E+02 3.5954655E+023393E+02 3.5974533E+023393E+02 3.5954464E+023393E+02 3.5954559E+023391E+02	MAX FORCE Y TIME X DIRECT Y DIRECT 3.1493453E+01 0.1808E+03 9.167 0.6027E+00 0.1808E+03 9.172 0.1792E+01 0.1808E+03 3.1462252E+01 0.1808E+03 3.1462252E+01 0.7231E+03 9.172 0.6640E+01 0.7231E+03
MAXIMUM BEARING VELOCITIES	MAX VELOCITY X ISOLATOR TIME X DIRECT Y DIRECT 1 3.4755359E+022342E+02 2 3.4745468E+022456E+02 3 3.4755359E+022456E+02 4 3.4775251E+022445E+02 5 3.4755359E+022451E+02	MAXIMUM BEARING FORCES ISOLATOR TIME X DIRECT Y DIRECT 1 3.213 - 1808E+03 - 2562E+01 2 2.593 0.1808E+03 0.1507E+01 3 6.024 0.1808E+03 0.1507E+01 4 2.593 0.1808E+03 0.1513E+01 5 2.593 0.7232E+03 0.6053E+01

3.866 0.1551E+02

3.896 -.1303E+02 -.8237E+01

3.857 -.1317E+02 -.8162E+01

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MAX. TOTAL ACCELERATIONS AT CENTER OF MASS OF LEVELS

	~	1E - 05	2
	ACCEL	1224	ACCEL
	TIME	3.706	TIME
	≻	+01	~
	ACCEL Y	10.0333370E+02 4.742 0.8163E+01 3.7061224E-05	ACCEL Y
	TIME	4.742	TIME
	×	+02	×
т ш	ACCEL	3370E	RE: 2 ACCEL
SUPERSTRUCTURE :	TIME	10.033	SUPERSTRUCTURE : 2 LEVEL TIME ACCEL
SUPERS	LEVEL	-	SUPERS

MAX. ACCELERATIONS AT CENTER OF MASS OF BASE LEVEL TIME ACCEL X TIME ACCEL Y TIME ACCEL R BASE 7.398 -.2271E+03 6.247 -.1482E+03 6.155 0.2386E-01

3.808 0.1707E+03 3.871 0.1232E+03 4.739 0.2174E-02

-

PROFILES OF TOTAL ACCELERATION AND DISPLACEMENT AT TIME OF MAX BASE DISPLACEMENTS

MAXIMUM BASE DISPLACEMENT IN X DIRECTION TIME : 3.857

SUPERSTRUCTURE : 1 X

≻

ACCEL -6.3355 117.3894	ACCEL 119.2784 117.3894	A ACCEL -6.0140 63.7978
DISP 7.7007 -8.1682	Y DISP -0.0790 -8.1682	Y DIRECTION Y DISP -8.2425
ACCEL -21.4382 126.2531	2 ACCEL 160.7382 126.2531	ACEMENT IN 1 ACCEL 21.1124 177.4479
DISP 25.8562 -13.1743	SUPERSTRUCTURE : X LEVEL DISP 1 -0.1074 BASE -13.1743	MAXIMUM BASE DISPLACEMENT TIME : 3.896 SUPERSTRUCTURE : 1 X ACCEL 1 25.5221 -21.11 BASE -13.0336 177.44
LEVEL 1 BASE	SUPERST LEVEL 1 BASE	MAXIMUM TIME : SUPERSTR LEVEL BASE SUDEPORTD

ACCEL 94.5614 63.7978

DISP -0.0644 -8.2425

ACCEL 89.4051 177.4479

DISP -0.0606 -13.0336

LEVEL 1 BASE

≻

×

PROFILES OF TOTAL ACCELERATION AND DISPLACEMENT AT TIME OF MAX ACCELERATION IN EACH BUILDING

SUPERSTRUCTURE : 1

MAX ACCELERATION IN X DIRECTION TIME : 10.033

ACCEL 3.7966 13.2791
DISP -4.5752 0.5174
ACCEL -33.6970 25.5478
DISP 40.5288 1.9910
LEVEL 1 BASE

MAX ACCELERATION IN Y DIRECTION TIME : 4.742

ACCEL	8.1633	-80.1861
DISP	-9.8099	5.6894
ACCEL		-113.8913
DISP	0.8322	9.8686
LEVEL	-	BASE

SUPERSTRUCTURE : 2

MAX ACCELERATION IN X DIRECTION

TIMÉ : 3.808

ACCEL 104.9407	106.3026	
S O L	- / - 000	TION
ACCEL 170.7426	168.8/33	IN Y DIRECTION
DISP -0.1	- 12 . 09.00	ACCELERATION E : 3.871
	ВАЗЕ	MAX A0 TIME

ACCEL	123.2494	87.1164
DISP	-0.0819	-8.2144
ACCEL	143.3974	87.5046
DISP	-0.0965	-13.1541
LEVEL	-	BASE

.MAXIMUM STRUCTURAL SHEARS.....

SUPERST. No TIME FORCE X TIME FORCE Y TIME Z MOMENT 1 10.033 -.1424E+04 4.742 0.3449E+03 3.706 -.2080E+01 2 3.808 0.5305E+04 3.871 0.3829E+04 4.739 0.3697E+04 MAXIMUM STORY SHEARS.....

RES. SHEAR SQRT(FX12+FY12) 3.706 -.2080E+01 10.032 0.1433E+04 Z MOMENT TIME TIME 4.742 0.3449E+03 > FORCE TIME 10.033 -.1424E+04 SUPERSTRUCTURE : 1 LEVEL TIME FORCE X -

RES. SHEAR SQRT(FX12+FY12) TIME Z MOMENT TIME ≻ FORCE TIME SUPERSTRUCTURE : 2 LEVEL TIME FORCE X

3.841 0.6287E+04 4.739 0.3697E+04 3.871 0.3829E+04 3.808 0.5305E+04 -

PROFILES OF TOTAL ACCELERATION AND DISPLACEMENT AT TIME OF MAX STRUCT SHEAR IN EACH BUILDING

SUPERSTRUCTURE : 1

MAX STRUC SHEAR IN X DIRECTION TIME : 10.033

ACCEL	3.7966	13.2791
DISP	-4.5752	0.5174
ACCEL	-33.6970	ഗ
DISP	40.5288	1.9910
LEVEL	-	BASE

MAX STRUC SHEAR IN Y DIRECTION TIME : 4.742

ACCEL	8.1633	-80.1861
DISP	-9.8099	5.6894
ACCEL	-0.6490	-113.8913
DISP	0.8322	9.8686
LEVEL	-	BASE

SUPERSTRUCTURE : 2

MAX STRUC SHEAR IN X DIRECTION TIME : 3.808

ACCEL	104.9407	106.5026
DISP	-0.0697	-7.7689
ACCEL	170.7426	168.8793
DISP	-0.1136	-12.8939
LEVEL	-	BASE

MAX STRUC SHEAR IN Y DIRECTION TIME : 3.871

ACCEL	123.2494	87.1164
DISP	-0.0819	-8.2144
ACCEL	143.3974	87.5046
DISP	-0.0965	-13.1541
LEVEL	-	BASE

PROFILES OF TOTAL ACCELERATION AND DISPLACEMENT AT TIME OF MAX BASE SHEARS

MAXIMUM BASE SHEAR IN X DIRECTION TIME : 3.808

SUPERSTRUCTURE : 1

							MAX STRUCTURAL SHEAR X DIRECTION	TIME MAX STUCTURAL SHEAR 10.033 - 1423.6289	INERTIA FORCES -1423.6289 252.9515 FORCE AT C.M. OF ENTIRE BASE	TIME MAX STUCTURAL SHEAR 3.808 5305.0348	INERTIA FORCES 5305.0348 1672.0887 FORCE AT C.M. OF ENTIRE BASE	MAX STRUCTURAL SHEAR Y DIRECTION	TIME MAX STUCTURAL SHEAR 4.742 344.8809	INERTIA FORCES 344.8809 -793.9299 FORCE AT C.M. OF ENTIRE BASE
ACCEL -6.5616 106.5026	ACCEL 104 .9407 106 .5026		ACCEL -6.3062 113.1370		ACCEL 120.8517 113.1370	<pre>CE PROFILES***********</pre>								
۲.9075 7.9075 -7.7689	۲ DISP -0.0697 -7.7689	DIRECTION	۲ DISP 7.6708 -8.1843	>	DISP -0.0801 -8.1843	*************F0RCE	DIRECTION	ENT 690	S 289 515	ENT 337	S 348 887	DIRECTION	ENT 321	0 0 0 0 0 0
ACCEL -21.5934 168.8793	2 ACCEL 170.7426 168.8793	, ×	1 ACCEL -21.4117 114.8752	5	ACCEL 157.2186 114.8752	* * *	NG MOMENT X	DVERTURNING MOMENT -404880.0690	INERTIA FORCES - 1423.6289 252.9515	OVERTURNING MOMENT 1116603.7337	INERTIA FORCES 5305.0348 1672.0887	VG MOMENT Y	OVERTURNING MOMENT 98084.1321	INERTIA FORCES 344.8809 -793.9299
X DISP 25.9461 -12.8939	SUPERSTRUCTURE : X LEVEL DISP 1 -0.1136 BASE -12.8939	M BASE SHEAR 3.861	SUPERSTRUCTURE : X LEVEL DISP 1 25.8306 BASE -13.1722	SUPERSTRUCTURE :	DISP -0.1052 -13.1722		MAX OVERTURNING MOMENT	TIME DVE 10.033	INE	TIME 0VE 3.808	INE	MAX OVERTURNING MOMENT	TIME OVEF 4.742	INE
LEVEL 1 BASE	SUPERS LEVEL BASE	MAXIMUM TIME :	SUPERS LEVEL 1 BASE	SUPERS	LEVEL 1 BASE	B-		SUPR/STURE	FLOOR 1 BASE	SUPR/STURE 2	FLOOR 1 BASE	MA	SUPR/STURE	FLOOR 1 BASE

B-80

STUCTURAL SHEAR 3829.4054 ERTIA FORCES 3829.4054 B62.5474 FORCE AT C.M. DF ENTIRE BASE			X DIR TIME Y DIR		X DIR TIME Y DIR
TIME MAX STUCTU 3.871 INERTIA			Y DIR TIME		Y DIR TIME
SUPR/STURE TIME OVERTURNING MOMENT 2 3.871 806013.2396 FLOOR INERTIA FORCES 1 3829.4054 BASE 862.5474	.MAXIMUM INTERSTORY DRIFT RATIOS' FOR EACH SUPERSTRUCTURE	SUPERSTRUCTURE : 1 COORDINATES OF COLUMN LINES WITH RESPECT TO MASS CENTER OF BASE C/L : 1 X COOR : -834.960 Y COOR : 0.000 COLUMN LINES	LEVEL TIME X DIR TIME Y DIR TIME X DIR TIME 1 10.044 0.1521E+00 4.752 0.3598E-01 SUPERSTRUCTURE : 2	COORDINATES DF COLUMN LINES WITH RESPECT TO MASS CENTER DF BASE C/L : 1 X CODR : -14.400 Y CODR : 0.000 COLUMN LINES	LEVEL TIME X DIR TIME Y DIR TIME X DIR TIME 1 3.809 0.5900E-03 3.872 0.4664E-03

FPS SYSTEM

WITH EFFECT OF VERTICAL GROUND MOTION AND OVERTURNING MOMENTS

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INPUT
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52 FPS BEARINGS - VARIABLE NORMAL LOAD in Kips/in*sect2 secs 1 2 5 5 386.22 1 3 1 3 0.001 50 1 500 1 0.5 0.25 4 0.02 2000 0 1 35.12 35.12 3659.0059 0.0 0.0 42.24794 1700000.00000 0.005 0.005 0.005 0.0 0.0 284.40, 18.00 46716.90 46716.90 6618227.50 0.0 0.0 31.07037 1700000.00000 0.02 0.02 0.02 0.0 0.0 210.48, 18.00 00000 9.90109 427528.00000 00000 -834.96 0.00 0.00 -820.56 806.16 0.00 0.00 820.56 -14.40 0.00 36 82.4 0.105 0.105 0.03 0.03 0.8 0.8 0.02 0.02 4017.625 36 82.4 0.105 0.105 0.03 0.03 0.8 0.8 0.02 0.02 4017.625 36 82.4 0.105 0.105 0.03 0.03 0.8 0.8 0.02 0.02 4017.625 3 6 82.4 0.105 0.105 0.03 0.03 0.8 0.8 0.02 0.02 4017.625 3 6 82.4 0.105 0.105 0.03 0.03 0.8 0.8 0.02 0.02 16070.5 0 10 1 1 2 3 4 5 1 -834.96 0.00 1 -14.40 0.00

WAVEX.DAT

-0.30719E+01	-0.10240E+02	-0.48048E+01	0.10200E+02	-0.28356E+01
-0.81918E+01	0.12051E+02	-0.10161E+02	-0.13430E+02	0.14611E+02
-0.11618E+02	-0.13430E+02	0.14257E+02	-0.98065E+01	-0.72072E+01
0.14532E+02	0.12248E+02	-0.56712E+01	-0.27962E+01	0.57500E+01
-0.14178E+01	0.15753E+01	0.20755E+02	0.11027E+01	-0.28159E+02
0.72072E+01	0.40526E+02	0.28317E+02	0.46866E+01	-0.10003E+02
-0.23394E+02	-0.17053E+02	0.63014E+00	0.17014E+02	0.12997E+02
0.90582E+00	0.94520E+00	-0.58681E+01	-0.23630E+01	0.88613E+01
0.16108E+02	-0.15753E+01	-0.32058E+02	-0.88219E+01	0.91764E+01
0.40959E+01	0.36666E+02	0.31389E+02	-0.16738E+02	-0.37139E+02
-0.10318E+02	0.47654E+01	0.83493E+01	0.22764E+02	0.98459E+01
•••••				

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WAVEY.DAT

0.12878E+02	0.22055E+01	-0.22842E+01	-0.16935E+01	0.88219E+01
0.31507E+01	-0.20361E+02	-0.16502E+02	0.46079E+01	0.78767E+01
0.36233E+01	-0.14572E+02	-0.10122E+02	0.16187E+02	0.19652E+02
-0.66952E+00	-0.94914E+01	-0.26781E+01	-0.24418E+01	-0.13587E+02
-0.14926E+02	0.12721E+02	0.29538E+02	0.83493E+01	-0.24851E+02
-0.17526E+02	0.90582E+01	0.17368E+02	-0.60257E+01	-0.21779E+02
-0.24378E+02	-0.10437E+02	0.26190E+02	0.50686E+02	0.10082E+02
-0.44582E+02	-0.47772E+02	-0.19731E+02	0.87825E+01	0.41234E+02
0.38753E+02	-0.40565E+01	-0.36745E+02	-0.37375E+02	-0.10673E+02
0.21464E+02	0.27017E+02	0.39384E+01	-0.22527E+02	-0.35209E+02
-0.29301E+02	0.67346E+01	0.46236E+02	0.47890E+02	0.44897E+01
		0.102502102	0.470502402	0.4489/E+01

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WAVEZ.DAT

0 150005 00				
0.17092E+02	0.23354E+02	0.32688E+01	-0.31625E+02	0.59075E+00
0.33240E+02	-0.16344E+02	-0.15675E+02	0.68921E+01	-0.24969E+02
0.14099E+02	0.30522E+02	-0.24536E+02	0.20479E+01	0.40211E+02
-0.18510E+02	-0.53365E+02	0.19810E+02	0.42219E+02	-0.23433E+02
-0.42810E+02	-0.59075E+01	0.29538E+02	0.48757E+02	0.33870E+02
-0.37966E+02	-0.93851E+02	-0.26781E+02	0.48481E+02	0.63447E+02
0.47260E+02	0.12800E+02	-0.28553E+02	-0.48481E+02	-0.10988E+02
0.21622E+02	0.35445E+00	-0.70496E+01	0.22055E+01	0.68527E+01
-0.14178E+02	0.12681E+02	0.56397E+02	0.74435E+01	-0.15045E+02
0.43322E+01	-0.41471E+02	-0.47497E+02	0.32216E+02	0.68960E+02
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FUNCTIONS

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FUNCTION FOVM ( OVMX, OVMY, XP, YP, I )
С
   CALCULATING AXIAL FORCES IN THE BEARINGS FROM OVERTURNING MOMENTS
С
   DEVELOPED BY..... PANAGIOTIS TSOPELAS.... FEB 1993
С
С
IMPLICIT REAL*8(A-H,O-Z)
   COMMON /MAIN1 /NB, NP, MNF, MNE, NFE, MXF
    COMMON /BEARAREA/AREA(500)
    DIMENSION OVMX (NB+1,2), OVMY (NB+1,2), XP (NP), YP (NP)
    IF(I.EQ.1) THEN
    OMX = 0.0
    OMY=0.0
    DO 10 K=1,NB+1
    OMX=OMX+OVMX(K, 1)
    OMY=OMY+OVMY(K, 1)
10 CONTINUE
    ENDIF
    IF(I.EQ.5) FOVM=0.0
    IF(I.EQ.4) FOVM= OMY/1641.12
    IF(I.EQ.3) FOVM= OMX/1641.12
    IF(I.EQ.2) FOVM=-OMY/1641.12
    IF(I.EQ.1) FOVM=-OMX/1641.12
    RETURN
    END
FUNCTION FFMAX (FRMAX, FRMIN, FNOR, I)
С
С
    CALCULATING MAXIMUM FRICTION COEFFICIENT AS FUNC OF PRESSURE
С
    DEVELOPED BY.....FEB 1993
С
IMPLICIT REAL*8(A-H,O-Z)
    COMMON /MAIN1 /NB, NP, MNF, MNE, NFE, MXF
    COMMON /BEARAREA/AREA(500)
    EXTERNAL WGTCF
    FACT=FNOR
    PRESS=15.0*FACT
    ALP=0.10729586
    DF=0.065
    FFMAX=FRMAX-(DF) *DTANH(ALP*PRESS)
    RETURN
    END
```

BY...P. C. TSOPELAS, M. C. CONSTANTINOU AND A. M. REINHORN DEPARTMENT OF CIVIL ENGINEERING STATE UNIV. OF NEW YORK AT BUFFALO 3D-BASIS-M..... A GENERAL PROGRAM FOR THE NONLINEAR DYNAMIC ANALYSIS OF THREE DIMENSIONAL BASE ISOLATED MULTIPLE BUILDING STRUCTURES DEVELOPED BY...P. C. TSOPELAS, S. NAGARAJAIAH , M. C. CONSTANTINOU AND A. M. REINHORN DEPARTMENT OF CIVIL ENGINEERING STATE UNIV. OF NEW YORK AT BUFFALO NATIONAL CENTER FOR EARTHQUAKE ENGINEERING RESEARCH STATE UNIVERSITY OF NEW YORK, BUFFALO 3D-BASIS-ME, JANUARY 1993 **FPS BEARINGS - VARIABLE NORMAL LOAD** in Kips/in*sect2 secs VERSION UNITS LENGTH : MASS : | TIME : DEVELOPED PROGRAM 52

NO: OF BUILDINGS......= NO: OF ISOLATORS......= INDEX FOR SUPERSTRUCTURE STIFFNESS DATA=

4 10 -

OUTPUT

ى س	0.00100		0.50000 0.25000 50.00000 1.00000	4		0.02000 2000 1.00000 0.00000
INDEX = 1 FOR 3D SHEAR BUILDING REPRES. INDEX = 2 FOR FULL 3D REPRESENTATION NUMBER OF ISOLATORS, OUTPUT IS DESIRED=	TIME STEP OF INTEGRATION (NEWMARK)= INDEX FOR TYPE OF TIME STEP=	INDEX = 1 FOR CONSTANT TIME STEP INDEX = 2 FOR VARIABLE TIME STEP	GAMA FOR NEWMARKS METHOD BETA FOR NEWMARKS METHOD TOLERANCE FOR FORCE COMPUTATION REFERENCE MOMENT OF CONVERGENCE MAX NUMBER OF ITERATIONS WITHIN T.S=	INDEX FOR GROUND MOTION INPUT=	INDEX = 1 FOR X DIR. INPUT INDEX = 2 FOR X & Y DIR. INPUT INDEX = 3 FOR X & Z DIR INPUT INDEX = 4 FOR X , Y & Z DIR. INPUT	TIME STEP OF RECORD

-SUPERSTRUCTURE :STIFFNESS DATA.....

STIFFNESS LEVEL	(THREE DIMENSIONAL STIFF X	STIFFNESS (THREE DIMENSIONAL SHEAR BUILDING) .evel stiff x stiff y	STIFF R	ECCENT X	ECCENT Y
÷	35.12000	35.12000	3659.00590	0.00001	0.00000
SUPERSTRUCTI LEVEL	SUPERSTRUCTURE MASS LEVEL TRANSL. MASS	ROTATIONAL MASS	ECCENT X	ECCENT Y	
-	42.24794	1700000.00000	0.00000	0.00000	
SUDEDSTRUCT	SUDEDSTRUCTURE DAMPING				

- N

HEIGHT	284.400 18.000
HEIGHT LEVEL	-0

EIGENVALUES AND EIGENVECTORS (3D SHEAR BUILDING REPRESENTATION)....

PERIOD	0.135432E+03 0.689137E+01 0.689137E+01	Э
EIGENVALUE	0.215236E-02 0.831283E+00 0.831283E+00	-
MODE NUMBER	- N Ø	MODE SHAPES LEVEL

×

0.0000000 0.1538499 0.0000000 0.0000000 0.0000000 0.1538499 +

0.0007670 0.0000000 0.0000000 т 2

SUPERSTRUCTURE	URE : 2				
· · · · · STIFFN	STIFFNESS DATA				
STIFFNESS LEVEL	STIFFNESS (THREE DIMENSIONAL SHEAR BUILDING) .EVEL STIFF X STIFF Y	SHEAR BUILDING) STIFF Y	STIFF R	ECCENT X	ECCENT Y
÷	46716.90000	46716.90000	6618227.50000	0.00001	0.00000
SUPERSTRUCT LEVEL	SUPERSTRUCTURE MASS LEVEL TRANSL. MASS	ROTATIONAL MASS	ECCENT X	ECCENT Y	
÷	31.07037	1700000.00000	0.00000	0.00000	
SUPERSTRUCT MODE SHAPE	SUPERSTRUCTURE DAMPING MODE SHAPE DAMPING RATIO				
- 0 G	0.02000 0.02000 0.02000				
HEIGHT LEVEL					
- 0	210.480 18.000				

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STIFFNESS DATA FOR LINEAR-ELASTIC ISOLATION SYSTEM.....

0.00000	0.00000	0.00000	0.00000	0.00000
IN X DIR. =	IN Y DIR. =	IN R DIR. =	MASS=	MASS=
STIFFNESS OF LINEAR-ELASTIC SYS. IN X DIR.	STIFFNESS OF LINEAR ELASTIC SYS. IN Y DIR.	STIFFNESS OF LINEAR ELASTIC SYS. IN R DIR.	ECCENT. IN X DIR. FROM CEN. OF MASS=	ECCENT. IN Y DIR. FROM CEN. OF MASS=
STIFFNESS OF LI	STIFFNESS OF LI	STIFFNESS OF L	ECCENT. IN X	ECCENT. IN Y

OF THE BASE	ROTATIONAL MASS
0	. MASS
CENTER	TRANSL
THE	
AT	
MASS	

427528.00000	
9.90109	
MASS	

-	ECX
THE BASE.	
THE	~
OF 1	
MASS	
Ч	
THE CENTER OF MASS	۲
THE	
F	
N DAMPING A	×
SOLATIO	
GLOBAL I	

ECY	0.00000
ECX	0.00000
α.	0.00000
7	0.00000
×	0.00000
	DAMPING

ISOLATORS LOCATION INFORMATION...... ISOLATOR X Y

0.0000	-820.5600	0.0000	820.5600	0.0000
-834.9600	0.0000	806.1600	0.0000	- 14 . 4000
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.....ELEMENT TYPE =

	NORMAL FORCE	4017.62500 4017.62500 4017.62500 4017.62500 4017.62500 16070.50000
	YIELD DISPL. Y	0.02000 0.02000 0.02000 0.02000 0.02000
	YIELD DISPL. X	0.02000 0.02000 0.02000 0.02000 0.02000
	PA Y	0.800 0.800 0.800 0.800 0.800 0.800 0.800
	PA X	0.800 800 800 800 800 800 800 800 800 80
	FMIN Y	0.03000 0.030000 0.0300000000
	FMIN X	0.03000 0.030000 0.0300000000
	FMAX Y	0.10500 0.10500 0.10500 0.10500 0.10500 0.10500
AMETERS	S FMAX X	0.10500 0.10500 0.10500 0.10500 0.10500
F. P. S. BEARING PARAMETERS	RADIUS	82.4000 82.4000 82.4000 82.4000 82.4000 82.4000
F. P. S. B	I SOLATOR	- იღ4 ი

TIME HISTORY OPTION=

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INDEX = 0 FOR NO TIME HISTORY OUTPUT INDEX = 1 FOR TIME HISTORY OUTPUT

																	MAX RES. DISPL. SQRT(DX12+DY12) TIME DISPLACEMENT 3.864 0.1549E+02 3.865 0.1589E+02 3.865 0.1580E+02 3.864 0.1539E+02 3.864 0.1539E+02
= 10 1=		= 1 2 3 4 5	BUILDING REPRESENTATION)		- 0.0	σ				: MASS OF LEVELS Respect to the Base)	TIME ROTATION	I 3.918 0.3829E-03	TIME ROTATION	l 4.7174879E-03	BASE / TIME ROTATION)1 3.9183828E-03		MAX DISPL Y TIME X DIRECT Y DIRECT 3.8881319E+027994E+01 3.8931346E+028303E+01 3.8971312E+028604E+01 3.8971285E+028303E+01
NO. OF TIME STEPS AT WHICH TIME HISTORY OUTPUT IS DESIRED	INDEX = O FOR NO PROFILES OUTPUT INDEX = 1 FOR PROFILES OUTPUT	FORCE-DISPLACEMENT TIME HISTORY DESIRED AT ISOLATORS NUMBERED=	EIGENVALUES AND EIGENVECTORS (3D SHEAR BU	MODE NUMBER EIGENVALUE PERIOD	1 0.389307E+01 0.318444E+01 2 0.150358E+04 0.162038E+00 3 0.150358E+04 0.162038E+00	MODE SHAPES LEVEL 1 2	1 X 0.000000 0.1794018 0.0000000	1 Y 0.000000 0.000000 0.1794018	1 R 0.0007670 0.0000000 0.0000000	MAX. RELATIVE DISPLACEMENTS AT CENTER OF (WITH RE	SUPERSTRUCTURE : 1 LEVEL TIME DISPL X TIME DISPL Y	1 10.042 0.4072E+02 4.7559973E+01	SUPERSTRUCTURE : 2 LEVEL TIME DISPL Y	1 3.8371281E+00 3.8488832E-01	MAX. DISPLACEMENTS AT CENTER OF MASS OF B LEVEL TIME DISPL X TIME DISPL Y BASE 3.8541329E+O2 3.8938303E+O1	MAXIMUM BEARING DISPLACEMENTS	MAX DISPL X ISOLATOR TIME X DIRECT Y DIRECT 1 3.8541329E+027924E+01 2 3.8561359E+028230E+01 3 3.8541329E+028509E+01 4 3.8531300E+028217E+01

MAXIMUM BEARING VELOCITIES	G VELOCITIES		
ISOLATOR TIME	X DIRECT Y DIRECT	TIME X DIRECT Y DIRECT	MAX RES. VELUCITY SQRT(VXT2+VYT2) TIME VELOCITY
1 3.47	3.4745490E+022441E+02	3.5984465E+023288E+02	3.470 0.6012E+02
2 3.47	3 - 5576E+02 - 2533E+02	3.6004528E+023371E+02	3.469 0.6128E+02
3 3.47	45490E+022612E+02		3.469 0.6085E+02
4 3.47	45404E+022528E+02	3.6004363E+023371E+02	3.470 0.5970E+02
5 3.47	45490E+022527E+02	3.6004446E+023370E+02	3.469 0.6048E+02
MAXIMUM BEARING FORCES	G FORCES		
	MAX FORCE X	MAX FORCE V	MAY DES ENDCE SODI(EY+3+EV+3)
ISOLATOR TIME	X DIRECT Y	TIME X DIRECT Y DIRECT	
1 4.72	4.720 0.7791E+03 0.4426E+03	3.8555955E+035061E+03	5 0.8
2 3.80	3.8037639E+034950E+03	3.8407032E+035281E+03	3.812 0.9130E+03
3 3.81	3.8199104E+036537E+03	3.8418701E+036887E+03	3.827 0.1124E+04
4 3.81	38656E+036059E+03	3.8487896E+O36577E+O3	3.828 0.1066E+04
5 3.80	3.8043263E+042167E+04	3.8472940E+042367E+04	3.820 0.3954E+O4

MAX. TOTAL ACCELERATIONS AT CENTER DF MASS OF LEVELS

۲	-05	æ	-02
ACCEL R	1 10.0313385E+02 4.743 0.8294E+01 3.6691131E-05	ACCEL R	3.836 0.1927E+03 3.847 0.1328E+03 4.697 0.1911E-02
A	i.	Ă	o.
ME	669	ME	697
F	, m	Ē	4
ACCEL Y TIME	1 01	ACCEL Y TIME	+03
CEL	294E	CEL	328E
AC	0.8	ACC	÷.
	743		847
TIME	4.	TIME	с. С
×	+02	×	۴03
EL -	385E	2 CEL	927E
но ВСС	Э.	Е : АС(0.19
CTUR ME	031	CTUR ME	336
TRUG	10.0	TRUG	ω. Θ
SUPERSTRUCTURE : 1 LEVEL TIME ACCEL X	-	SUPERSTRUCTURE : 2 LEVEL TIME ACCEL X	-
LE SU		LE ^I	

MAX. ACCELERATIONS AT CENTER OF MASS OF BASE LEVEL TIME ACCEL X TIME ACCEL Y TIME ACCEL R BASE 7.398 -.2344E+03 8.493 0.1943E+03 3.923 0.5560E-01 PROFILES OF TOTAL ACCELERATION AND DISPLACEMENT AT TIME OF MAX BASE DISPLACEMENTS

MAXIMUM BASE DISPLACEMENT IN X DIRECTION TIME : 3.854

SUPERSTRUCTURE : 1 X

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3.854 -.1329E+02 -.8217E+01

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ACCEL -6.3118 92.9336	ACCEL 131.1268 92.9336	ACCEL -5.9874 78.3010
DISP 7.6727 -8.2217	Y DISP -0.0875 -8.2217	Y DIRECTION DISP 7.3187 -8.3031
ACCEL -21.4288 104.6639	2 ACCEL 173.5615 104.6639	ACEMENT IN 1 ACCEL -21.0847 191.1480
DISP 25.8472 -13.2932	SUPERSTRUCTURE : X LEVEL DISP 1 -0.1168 BASE -13.2932	MAXIMUM BASE DISPLACEMENT TIME : 3.893 SUPERSTRUCTURE : 1 LEVEL DISP ACCEL 1 25.4976 -21.084 BASE -13.1509 191.148
LEVEL 1 BASE	SUPERST LEVEL 1 BASE	MAXIMUM TIME : SUPERST LEVEL BASE

	ACCEL	-5.9874	78.3010			ACCEL	75.8248	78.3010
7	DISP	7.3187	-8.3031		>	DISP	-0.0520	-8.3031
	ACCEL	-21.0847	191.1480	2		ACCEL	68.7997	191.1480
· ×	DISP	25.4976	-13.1509	SUPERSTRUCTURE :	×	DISP	-0.0468	-13.1509
	LEVEL	-	BASE	SUPERST		LEVEL	-	BASE

PROFILES OF TOTAL ACCELERATION AND DISPLACEMENT AT TIME OF MAX ACCELERATION IN EACH BUILDING

SUPERSTRUCTURE : 1

MAX ACCELERATION IN X DIRECTION TIME : 10.031

ACCEL	3.6887	-11.7807
DISP	-4.4418	0.5464
ACCEL	-33.8494	18.3669
DISP	40.7131	1.8965
LEVEL	-	BASE

MAX ACCELERATION IN Y DIRECTION TIME : 4.743

ACCEL	8.2944	-77.2029
DISP	-9.9671	5.6604
ACCEL	-0.3804	- 109 . 0376
DISP	0.5085	9.9173
LEVEL	-	BASE

SUPERSTRUCTURE : 2

MAX ACCELERATION IN X DIRECTION

3.836 •• TIMÉ

ACCEL 129.2239 96.2306	
DISP -0.0855 -8.1198	DIRECTION
ACCEL 192.6852 117.2983	≻
DISP -0.1280 -13.2615	ACCELERATION IN E : 3.847
LEVEL 1 BASE	MAX A TIME

1004	ACCEL	132.8276	89.1733
	UISP	-0.0883	-8.1870
	ACCEL	185.7066	102.3710
	DISP	-0.1243	-13.2881
	LEVEL		BASE

MAXIMUM STRUCTURAL SHEARS.....

TIME Z MOMENT 3.669 -.1923E+01 4.697 0.3248E+04 TIME FORCE Y 4.743 0.3504E+03 3.847 0.4127E+04 SUPERST. No TIME FORCE X 1 10.031 -.1430E+04 2 3.836 0.5987E+04

MAXIMUM STORY SHEARS.....

Z MOMENT TIME RES. SHEAR SQRT(FX+2+FY+2) 3.669 -.1923E+01 10.030 0.1439E+04 TIME 4.743 0.3504E+03 FORCE Y TIME 10.031 -.1430E+04 FORCEX SUPERSTRUCTURE : LEVEL TIME FORCE -

MOMENT TIME RES. SHEAR SQRT(FX+2+FY+2) Ν TIME FORCE Y TIME SUPERSTRUCTURE : 2 LEVEL TIME FORCE X

3.839 0.7228E+04 4.697 0.3248E+04 3.847 0.4127E+04 3.836 0.5987E+04 -

							DIRECTION		FORCE AT C.M. OF ENTIRE RASE		FORCE AT C.M. OF ENTIRE BASE	DIRECTION		FORCE AT C.M. OF ENTIRE BASE
							MAX STRUCTURAL SHEAR X	TIME MAX STUCTURAL SHEAR 10.031 - 1430.0659	INERTIA FORCES - 1430.0659 181.8523	TIME MAX STUCTURAL SHEAR 3.836 5986.7993	INERTIA FORCES 5986.7993 1161.3813	MAX STRUCTURAL SHEAR Y I	TIME MAX STUCTURAL SHEAR 4.743 350.4192	INERTIA FORCES 350.4192 -764.3927
ACCEL -6.5571 144.3394	ACCEL 102.6127 144.3394		ACCEL -6.3584 89.1733		ACCEL 132.8276 89.1733	RCE								
ACCEL DISP -21.6165 7.9027 231.3975 -7.8217	2 ACCEL DISP 165.6228 -0.0681 231.3975 -7.8217	IN Y DIRECTION	- ACCEL DISP -21.4709 7.7213 102.3710 -8.1870		Y ACCEL DISP 185.7066 -0.0883 102.3710 -8.1870	**************	G MOMENT X DIRECTION	DVERTURNING MOMENT -406710.7518	TIA FORCES -1430.0659 181.8523	DVERTURNING MOMENT 1260101.5233	TIA FORCES 5986.7993 1161.3813	3 MOMENT Y DIRECTION	DVERTURNING MOMENT 99659.2330	FIA FORCES 350.4192 -764.3927
X LEVEL DISP 1 25.9779 BASE -13.0243	SUPERSTRUCTURE : 2 X LEVEL DISP 1 -0.1100 BASE -13.0243	MAXIMUM BASE SHEAR TIME : 3.847	SUPERSTRUCTURE : 1 X LEVEL DISP 1 25.8884 BASE -13.2881	SUPERSTRUCTURE : 2	LEVEL DISP 1 -0.1243 BASE -13.2881	B-	MAX OVERTURNING MOMENT	SUPR/STURE TIME OVER 1 10.031	FLOOR 1 BASE	SUPR/STURE TIME OVER 2 3.836	FLOOR INERTIA 1 BASE	MAX OVERTURNING MOMENT	SUPR/STURE TIME OVERT 1 4.743	FLOOR INERTIA 1 BASE

PROFILES OF TOTAL ACCELERATION AND DISPLACEMENT AT TIME OF MAX STRUCT SHEAR IN EACH BUILDING

SUPERSTRUCTURE : 1

MAX STRUC SHEAR IN X DIRECTION TIME : 10.031

ACCEL	3.6887	-11.7807	
DISP	-4.4418	0.5464	
ACCEL	-33.8494	18.3669	
DISP	40.7131	1.8965	
LEVEL	-	BASE	

MAX STRUC SHEAR IN Y DIRECTION TIME : 4.743

ACCEL	8.2944	-77.2029
DISP	-9.9671	5.6604
ACCEL	-0.3804	- 109 . 0376
DISP	0.5085	9.9173
LEVEL	-	BASE

SUPERSTRUCTURE : 2

MAX STRUC SHEAR IN X DIRECTION TIME : 3.836

ACCEL	129.2239	96.2306
DISP	-0.0855	-8.1198
ACCEL	192.6852	117.2983
DISP	-0.1280	-13.2615
LEVEL	-	BASE

MAX STRUC SHEAR IN Y DIRECTION TIME : 3.847

ACCEL	132.8276	89.1733
DISP	-0.0883	-8.1870
ACCEL	185.7066	102.3710
DISP	-0.1243	-13.2881
LEVEL	-	BASE

PROFILES OF TOTAL ACCELERATION AND DISPLACEMENT AT TIME OF MAX BASE SHEARS

MAXIMUM BASE SHEAR IN X DIRECTION TIME : 3.804

SUPERSTRUCTURE : 1

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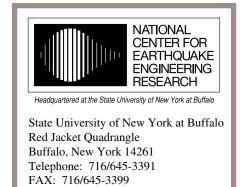
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