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Shake Table Testing of Triple Friction Pendulum Isolators under Extreme Conditions

by Apostolos A. Sarlis, Michael C. Constantinou and Andrei M. Reinhorn



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Apostolos A. Sarlis¹ Michael Constantinou² and Andrei M. Reinhorn³

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Preface

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This report describes an experimental program of a 3-story seismically isolated structure in which Triple Friction Pendulum (TFP) isolators were tested under extreme conditions, including uplift. This report presents information on (a) the performance of the TFP isolators and the isolated superstructure under strong excitation where the TFP isolators operate in all five regimes, including stiffening and deformation up to the displacement capacity, (b) the effect of the vertical component of earthquakes on the isolation system and superstructure response, (c) the behavior of TFP bearings of unusual configurations of which the behavior cannot be predicted by conventional models of TFP behavior, and (d) comparison of experimental results to analytical predictions of programs SAP2000 and 3pleANI in order to investigate the degree of accuracy of existing analysis models and newly developed formulations.

ABSTRACT

This report describes an experimental program of a 3-story seismically isolated structure in which Triple Friction Pendulum (TFP) isolators were tested under extreme conditions, including uplift.

This report presents information on (a) the performance of the TFP isolators and the isolated superstructure under strong excitation where the TFP isolators operate in all five regimes, including stiffening and deformation up to the displacement capacity, (b) the effect of the vertical component of earthquakes on the isolation system and superstructure response, (c) the behavior of TFP bearings of unusual configurations of which the behavior cannot be predicted by conventional models of TFP behavior, and (d) comparison of experimental results to analytical predictions of programs SAP2000 and 3pleANI in order to investigate the degree of accuracy of existing analysis models and newly developed formulations.

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SECTION 1 INTRODUCTION

The behavior of the Triple Friction Pendulum (TFP) isolator has been previously described by Fenz and Constantinou (2008a to 2008e) and Morgan (2007). The TFP isolator exhibits multiple changes in stiffness and strength with increasing amplitude of displacement. The construction of the force-displacement loop is complex as it may contain several transition points which depend on the geometric and frictional properties. Figure 1-1 shows the geometry of a Triple FP bearing and its parameters. Its behavior is characterized by radii R_1 , R_2 , R_3 and R_4 (typically $R_1=R_4$ and $R_2=R_3$), heights h_1 , h_2 , h_3 and h_4 (typically $h_1=h_4$ and $h_2=h_3$, distances (related to displacement capacities) d_1 , d_2 , d_3 and d_4 (typically $d_2=d_3$ and $d_1=d_4$) and friction coefficients μ_1 , μ_2 , μ_3 and μ_4 (typically $\mu_2 = \mu_3 < \mu_1 \le \mu_4$).



The lateral force-displacement relation of the isolation system is illustrated in Figure 1-2. Five different loops are shown in Figure 1-2, each one valid in one of five different regimes of displacement. The parameters in the loops relate to the geometry of the bearing, the friction coefficient values and total weight W carried by the isolation system as described in Fenz and Constantinou (2008a and 2008b). Triple FP isolators are designed to operate in regimes I to IV, whereas regime V is reserved for providing displacement restraint in earthquakes beyond the

maximum considered earthquake. In regime V, the isolator has consumed its displacement capacities d_1 and d_4 and only slides on surfaces 2 and 3 (see Figure 1-1).

Figure 1-2: Force-displacement loops of Triple FP bearing (Fenz and Constantinou, 2008)

For response history analysis, the TFP can be modeled using the Series Model described in Fenz and Constantinou (2008d) provided that $d_2=d_3$, $d_1=d_4$ and $\mu_2 = \mu_3 < \mu_1 \le \mu_4$. A simpler model (Parallel Model) for the special case of $\mu_2 = \mu_3 < \mu_1 = \mu_4$ and provided that stiffening does not occur was presented by Sarlis et al. (2010). Recently, Becker and Mahin (2011), Ray et al. (2013) and Dao et al. (2013) have developed formulations that can model the TFP behavior. All the formulations are based on satisfaction of horizontal force equilibrium and are restricted to the same constraints as the Series Model: $d_2=d_3$ and $d_1=d_4$, $\mu_2=\mu_3 < \mu_1 \le \mu_4$. Under such conditions, these models produce nearly identical results.

The TFP behavior for any random combination of geometric and frictional properties is described on the basis of a more advanced theory in Sarlis and Constantinou (2013). The theory is based on consideration of equilibrium of moments in addition to equilibrium of forces and requires use of eight degrees-of-freedom to describe the displacements and rotations of the parts of the bearing in each principal direction. The new model was implemented in the newly developed program 3pleANI that calculates and animates the TFP motion under extreme conditions, including uplift, landing and impact of components.

The frictional parameters that describe the behavior of the Triple FP bearing in the models of Fenz and Constantinou (2008a to 2008e) (denoted now as $\overline{\mu}_1, \overline{\mu}_2, \overline{\mu}_3, \overline{\mu}_4$, with the following constraints $\overline{\mu}_2 = \overline{\mu}_3 \leq \overline{\mu}_1 \leq \overline{\mu}_4$) utilize the values extracted from experiments of the Triple FP bearings and are not fundamental properties of the interfaces. Sarlis and Constantinou (2013)

have shown that the true frictional values ($\mu_1, \mu_2, \mu_3, \mu_4$ without any constraints) are related to those in the models of Fenz and Constantinou (2008a to 2008e) by the following equations:

$$\begin{split} \overline{\mu}_{2} &= \mu_{2} \frac{R_{2}}{R_{eff2}} \\ \overline{\mu}_{1} &= \frac{\mu_{1}R_{1} - \mu_{2}R_{2}}{R_{eff1} - R_{eff2}} \\ \overline{\mu}_{4} &= \frac{\mu_{4}R_{4} - \mu_{2}R_{2}}{R_{eff1} - R_{eff2}} \end{split}$$
(1-1)

Program 3pleANI makes use of friction values $\mu_1, \mu_2, \mu_3, \mu_4$.

This report describes an experimental program of a 3-story seismically isolated structure in which TFP isolators were tested under extreme conditions, including uplift. Analytical predictions of the response of the tested structure are made using the advanced theory model of Sarlis and Constantinou (2013).

The 3-story model structure is a modification of the six-story structure extensively used in the past at the University at Buffalo (Reinhorn et al., 1989; Mokha et al., 1990; Wolff and Constantinou, 2004; Fenz and Constantinou, 2008e). The structure was isolated using three different configurations of TFP bearings, including one in which the frictional properties were such that it could not be modeled by any existing models based on horizontal force equilibrium alone. The main purpose of these tests was to:

- 1. Study the performance of the TFP isolators and the isolated superstructure under strong excitation where the TFP isolators operate in all five regimes, including stiffening and up to their displacement capacity. Earlier studies of Fenz and Constantinou (2008e) presented shake table results of testing of a six-story structure in which the TFP isolators reached displacements in Regime IV (see Figure 1-2) but were further limited due to uplift. Morgan (2007) presented experimental results where TFP isolators displaced in all five regimes of operation but the tests were conducted with sinusoidal excitation rather than random seismic motions. Also, the isolators uplifted prior to reaching their displacement capacity in similarity to the Fenz and Constantinou (2008e) tests. More recently, Becker and Mahin (2011) presented experimental results of TFP bearings in all five regimes of operation in the testing of an isolated rigid block. The tests presented in this report add to the body of experimental results on the TFP isolators by extending to flexible structure in which the TFP bearings are displaced to their displacement capacity and simultaneously undergo uplift.
- 2. Study the effect of the vertical component of earthquakes on the isolation system and superstructure response. Previous experimental work (Fenz and Constantinou, 2008e, Morgan, 2007 and Becker and Mahin, 2011) also reported on this issue and generally have shown small effect of the vertical component on the horizontal global response. The study

of vertical earthquake effects in this report adds to the existing body of knowledge and includes some data where the effects are important.

- 3. Study the re-centering capability and the effect of initial offsets on the response.
- 4. Study the behavior of TFP bearings of unusual configurations such as cases having higher friction in the inner sliding surfaces, which cannot be predicted by conventional models of TFP behavior. The behavior of these bearings is described in Sarlis and Constantinou (2013).
- 5. Collect data on response to compare with analytical predictions in programs SAP2000 and 3pleANI (Sarlis and Constantinou, 2013) in order to investigate the degree of accuracy of existing analysis models and newly developed formulations.

SECTION 2 EXPERIMENTAL SETUP

2.1 Specimen Description

The model structure used in the shake table testing is shown in the photographs of Figure 2-1 and Figure 2-2. Figure 2-3 shows schematics of the model structure on the shake table. The model structure is a quarter length scale three-story steel model. The superstructure is a portion of the 6story model last used by Fenz and Constantinou (2008e) in testing of TFP isolators. The superstructure consists of moment resisting frames in the longitudinal direction and consists of braced frames in the transverse direction. Five concrete blocks, each weighing 8.9kN, were installed at each floor and two more at the base in order to achieve mass similitude. The total weight of the model (frame, base and added weight) on top of the isolators was 196kN (distributed as 53.2kN at the base and 47.6kN at each floor). All beams and columns are S3×5.7 (SI designation S75×8.5) and all braces are $L1\frac{1}{2}\times1\frac{1}{2}\times\frac{1}{4}$ (SI designation L38×38×6.4). The beam to column connections are fully welded and stiffened so that they are rigid. Horizontal bracing of all floors at all bays achieves, together with the concrete blocks, rigid diaphragm behavior. The 3-story structure seats on a base-mat that consists of a grid of two longitudinal W14x90 (SI designation W360x134) beams and four transverse W12x35 (SI designation W310x52) beams which are located at the superstructure's column locations. Also, the model features two HSS16x8x5/16 (SI designation HSS406.4x203.2x7.9) beams in the transverse direction that are connected on the top of the W14 x90 beams.

Four isolators were placed below the W14x90 beams on a 122cmx244cm footprint as shown in Figure 2-2 and Figure 2-3. The yellow plates seen at the bottom of the isolator-load cell assembly in Figure 2-2 and Figure 2-3 were used to first level the bearings and then to raise them so that the gravity loads on each isolator were approximately equal. The leveling plates were also used for bearing alignment.

Testing was conducted with earthquake shaking in the longitudinal (or E-W direction in Figure 2-3) and vertical directions.

Figure 2-1: Front view of the tested structure

Figure 2-2: View of TFP isolators installed at the base of the tested structure

The techniques for the installation, leveling and alignment of the isolators have been described in Fenz and Constantinou (2008e). At the conclusion of a test, triple FP bearings may exhibit permanent displacements; particularly in the cases where the coefficient of friction at the inner sliding surfaces (2 and 3 in Figure 1-1) is large. Figure 2-4 shows a photograph of the TFP obtained at the end of a shake table test which shows two types of permanent displacements that may be exhibited by triple FP bearings:

- a) Isolation system permanent displacement, which is the offset between the top and bottom concave plates of the bearings.
- b) Internal component permanent displacements. These permanent displacements always occur even in the absence of isolation system permanent displacements.

Permanent displacements of either type affect the behavior of the bearings in subsequent earthquakes. This complicates the comparison of experimental results for various tested configurations as the initial conditions are different. Moreover, analytical prediction of the experimental response would have required measurement of the permanent displacements of the internal components, which is complex. Accordingly, the tested structure was re-centered when needed by use of the following procedure. First, a hydraulic jack was placed inclined with one edge supported on the shake table and the other on the base of the structure (see Figure 2-4(b)) in order to bring the structure to its zero position. Next, a hydraulic jack was placed vertically in order to remove the normal load from one bearing at a time and re-center the internal components.

(a) Example of permanent displacements (b) Re-centering procedure Figure 2-4: Permanent displacements and TFP re-centering

2.2 Instrumentation

The complete list of the instruments used in the tests is presented in Table 2-1.

Table 2-1. List of misti unrentation notation, location and unrection of measurement					
Name	Туре	Location	Direction		
1N	Load Cell	SE LC Normal Force	V		
1SY	Load Cell	SE LC Shear Force	Т		
1SX	Load Cell	SE LC Shear Force	L		
1MY	Load Cell	SE LC Moment	Т		
1MX	Load Cell	SE LC Moment	L		
2N	Load Cell	NE LC Normal Force	V		
2SY	Load Cell	NE LC Shear Force	Т		
2SX	Load Cell	NE LC Shear Force	L		
2MY	Load Cell	NE LC Moment	Т		

Table 2	2-1:	List	of inst	trument	ation 1	notation.	location	and	direction	of	measurer	nent
I abit	-1.	LISU	OI III.5	ii umenta	ation	notation,	location	anu	uncenon	UI	measurer	nunu

2MX	Load Cell	NE LC Moment	L
3N Load Cell		NW LC Normal Force	V
3SY	Load Cell	NW LC Shear Force	Т
3SX	Load Cell	NW LC Shear Force	L
3MY	Load Cell	NW LC Moment	Т
3MX	Load Cell	NW LC Moment	L
4N	Load Cell	SW LC Normal Force	V
4SY	Load Cell	SW LC Shear Force	Т
4SX	Load Cell	SW LC Shear Force	L
4MY	Load Cell	SW LC Moment	Т
4MX	Load Cell	SE LC Moment	L
SPSE-SL	String Pot	SE Table	L
SPSW-SL	String Pot	SW Table	L
SPSE-1L	String Pot	SE 1st floor	L
SPSW-1L	String Pot	SW 1st floor	L
SPSE-2L	String Pot	SE 2nd floor	L
SPSW-2L	String Pot	SW 2nd floor	L
SPSE-3L	String Pot	SE 3rdfloor	L
SPSW-3L	String Pot	SW 3rd floor	L
SPSE-BL	String Pot	SE Base	L
SPSW-BL	String Pot	SW Base	L
SPNE-BT	String Pot	NE Base	Т
SPSE-TR-TC	String Pot	SE Bearing Top Concave plate	L
SPSE-TR-TS	String Pot	SE Bearing Top Slide plate	L
SPSE-TR-TR	String Pot	SE Bearing Rigid Slider	L
SPSE-TR-BS	String Pot	SE Bearing Bottom Slide plate	L
ASE-SL	Accelerometer	SE Table	L
ASW-SL	Accelerometer	SW Table	L
ASE-1L	Accelerometer	SE 1st floor	L
ASW-1L	Accelerometer	SW 1st floor	L
ASE-2L	Accelerometer	SE 2nd floor	L
ASW-2L	Accelerometer	SW 2nd floor	L
ASE-3L	Accelerometer	SE 3rdfloor	L
ASW-3L	Accelerometer	SW 3rd floor	L
ASE-BL	Accelerometer	SE Base	L
ASW-BL	Accelerometer	SW Base	L
ANE-BT	Accelerometer	NE Base	<u> </u>
ANE-1T	Accelerometer	NE 1st floor	<u> </u>
ANE-3T	Accelerometer	NE 3rd floor	Т
ASW-BT	Accelerometer	SW Base	Т
ASW-1T	Accelerometer	SW 1st floor	Т
ASW-3T	Accelerometer	SW 3rd floor	T
AN-SV	Accelerometer	NW Table	V
AN-SV-2	Accelerometer	NE Table	V
AS-SV	Accelerometer	SW Table	V
AS-SV-2	Accelerometer	SE Table	V
AN-BV	Accelerometer	NW Base	V
AN-BV-2	Accelerometer	NE Base	V
AS-BV	Accelerometer	SW Base	V

AS-BV-2	Accelerometer	SE Base	V			
ALC1-BV	Accelerometer	SE Load Cell 1	V			
ALC1-BL	BL Accelerometer SE Load Cell 1		L			
ALC2-BV	Accelerometer	NE Load Cell 2	V			
ALC2-BL	Accelerometer	NE Load Cell 2	L			
ALC3-BL Accelerometer		NW Load Cell 3	L			
ALC4-BL Accelerometer		SW Load Cell 4	L			
L=Longitudinal direction, V= vertical direction, LC=load cell, SE=South-East, SW=South-West,						
	NE=North-East, NW=North-West					

Figure 2-5 and Figure 2-6 show the location of the potentiometers (displacement transducers) and accelerometers installed on the superstructure and shake table. Two accelerometers and two displacement transducers were installed at each floor, base and the shake table in order to have redundancy in the measurements and to also measure torsional motion. Vertical accelerometers were installed on the shake table and the base at four opposite corners. Transverse accelerometers were also installed on the 1st and 3rd floors and at the base at the NE and SW corners of the model.

Figure 2-5: Location of displacement transducers on superstructure and shake table

Figure 2-6: Location of accelerometers on superstructure and shake table

The TFP isolators were installed on top of four 5-component load cells. The load cells measured axial, shear forces in two orthogonal directions and moments about two horizontal axes. Details about the load cells and how they are calibrated can be found in Bracci et al. (1992). The list of all measured components (channels) is shown in Figure 2-7. The TFP isolator on Load Cell 1 was also instrumented with displacement transducers as shown in Figure 2-8 in order to measure the displacements of the inner components. It should be noted that the inner parts of the TFP isolators occasionally experience torsional motions due to uneven distribution of friction tractions. This leads to erroneous measurements by the string pots so that the displacements of the inner parts could not be measured.

Figure 2-7: Five-component load cell channels

Figure 2-8: Displacement transducers installed at TFP inner components

An important part of any experimental study is to have redundancy in the measurements so that (a) the accuracy of measurements can be checked, and (b) sufficient data are acquired in case of failure of instrumentation. Although rarely reported, load cells often have measurement errors due to calibration errors (particularly for complex multichannel cells in which there is channel "cross-talk"), manufacturing errors (e.g., due imperfect placement of strain gages), installation errors in the test arrangement (e.g., leveling), condition of other supporting equipment (e.g., conditioners) and effects of the environmental conditions (e.g., temperature and humidity). Deviations of measured force of up to 20% of the actual forces are not uncommon. Figure 2-9 compares results for the base shear in shake tests of the tested isolated model obtained by direct measurement of the shear force (force F_{lc}) and by processing of the acceleration records obtained at each floor and the base-mat of the structure (force F_{acc}). Force F_{lc} was obtained as the sum of the shear forces recorded by the load cells supporting the isolators (sum of 1SX+2SX+3SX+4SX in Figure 2-7) and force F_{acc} was calculated as the sum of the floor and base-mat inertia forces:

$$F_{acc} = m_b \ddot{u}_b + m_f \left(\ddot{u}_1 + \ddot{u}_2 + \ddot{u}_3 \right)$$
(2-1)

where m_b is the mass of the base-mat (weight equal to 53.2kN), m_f is the mass of one floor (weight equal to 47.6kN), \ddot{u}_b is the longitudinal acceleration of the center of mass of the basemat and \ddot{u}_1, \ddot{u}_2 and \ddot{u}_3 are the center of mass accelerations of the 1st, 2nd and 3rd floors, respectively. The center of mass accelerations were calculated as the average of the two accelerometers recording on each floor. For example, \ddot{u}_b is the average of the recordings of instruments ASE-BL and ASW-BL (see Figure 2-6), \ddot{u}_1 is the average of the recordings of instruments ASE-1L and ASW-1L, etc.

The two sets of results in Figure 2-9 are in very good agreement. However, to obtain this good agreement, the load cell measurement was multiplied by a correction factor of 1.055-a factor found to be needed as the load cell measurements were systematically lower than the results obtained from processing of the acceleration records, which were presumed to be accurate. It was discovered that the difference was due to load cell calibration. The load cell calibration procedure followed is described in Bracci et al. (1992) and utilizes the fixture shown in Figure 2-10. The load cells are bolted together and placed on top of two rollers at the edges of the two outermost load cells. A loading beam is placed on top of the load cells supported by two rollers

placed on two of the load cells. A reference load cell is placed at the center and on top of the loading beam and load is applied on top of the reference load cell. The two outermost load cells are calibrated for half the load measured by the reference load cell. This however ignores the weight of the loading beam and the weight of the load cells. Each load cell has a weight of about 1.8kN and the loading beam, reference load cell and other features weigh another 1.8kN for a total of about 9kN additional unaccountable load. The distribution of this load gives rise to shear forces of 4.5kN for the two outer cells which are calibrated for shear force. Given that load cells were calibrated to a shear of about 90kN, this leads to a calibration error of the order of 5%.

Figure 2-9: Comparison of base shear-base displacement loops obtained from processing of acceleration records (force F_{acc}) and directly measured by load cells (force F_{ld})

Figure 2-10: Load cell calibration fixture

An additional problem encountered in the tests was significant drifting of the load cell values with time due to environment temperature changes that affected the temperature of the load cell conditioners. The sensitivity of the conditioners is shown in Figure 2-11(a) for the vertical load on each of the four load cells when a fan was used to cool the conditioners. Load cell drifts of about 15kN can be observed when no additional load was applied on the structure. Figure 2-11(b) shows load cell drifting over a 12 hour period without the use of a fan to cool down the conditioners. In the latter case, the drift in measured load in two of the load cells is 50kN and 400kN which indicates the severity of the problem.

Figure 2-11: Load cell normal load variation for (a) with fan cooling the conditioners and (b) without fan cooling

It was determined that the problem of drifting values was negligible for short times of the order of one minute so that it did not affect the measurements in single dynamic tests. Accordingly, the procedure followed to obtain values of load on each isolator at the start of each test was as follows:

- 1. At the first test and when re-centering of the isolators was needed or whenever the isolators were replaced, the load cells were balanced by jacking the structure up and removing the normal load from each isolator at a time (see Section 2.1). The normal load values were recorded after normal load was reinstated at all isolators. These values then served as the initial normal load values for the subsequent test.
- 2. For subsequent tests, the changes in the normal loads from the beginning to the end of each test were added to the initial values until the next time the load cells were balanced.

Application of the procedure described above resulted in the evolution of the value of the sum of the measured normal load on the four isolators for the duration of testing (the value of the measurement should be constant and equal to weight of the structure). The evolution of the measured total load is shown in Figure 2-12. While the measured load still exhibits some small drift (by less than 3% in over 100 tests), the drift is far less than that depicted in Figure 2-11 and it does not affect the fidelity of the measured forces in the testing.

Figure 2-12: Evolution of measured total vertical load on four isolators during testing

SECTION 3 INDIVIDUAL TESTING OF TRIPLE FP ISOLATORS

This section presents experimental results on the behavior of the isolators that were used for the shake table testing. The isolators were tested in the single bearing testing machine at the University at Buffalo (Kasalanati and Constantinou, 1999). For the shake table tests, TFP isolators of three different configurations were used with the geometric characteristics presented in Table 3-1. From the three configurations, A and C had the exact same geometry while Configuration B had slightly different geometry (the rigid slider was slightly shorter). The values of the friction coefficients for Configurations A and B satisfied the condition $\mu_2 = \mu_3 < \mu_1 < \mu_4$ for which standard models of TFP isolator behavior are valid. Configuration C satisfied the condition $\mu_1 = \mu_4 < \mu_2 = \mu_3$, for which the behavior cannot be predicted with standard models. The tests revealed the frictional properties of the isolators.

Geometric	Configuration	Configuration
Properties	A and C	В
$R_1 = R_4 \text{ (mm)}$	473	473
$R_2 = R_3 (\text{mm})$	76	76
$h_2 = h_3 (\text{mm})$	23	18
$h_1 = h_4 ({\rm mm})$	38	33
$R_{eff1} = R_{eff4} \ (\text{mm})$	435	440
$R_{eff2} = R_{eff3} \ (\text{mm})$	55	58
$d_1 = d_4 (\text{mm})$	64	64
$d_2 = d_3 \text{ (mm)}$	19	19
$b_1 = b_4 \pmod{2}$	101	101
$b_2 = b_3 \text{ (mm)}$	51	51

 Table 3-1: Geometric properties of Triple FP used in shake table tests (with reference to Figure 1-1)

3.1 Equipment and Instrumentation Used

A detailed description of the bearing testing machine can be found in Kasalanati and Constantinou (1999). The machine is depicted in Figure 3-1. The bearing sits on top of a five component load cell (the particular type of load cell used is denoted "5D-LC-12-BLU" in the University at Buffalo Structural Engineering and Earthquake Simulation Laboratory manual, <u>http://nees.buffalo.edu/docs/labmanual/HTML/Chapter%203.htm#_Toc145756944</u>). This load cell records forces in three directions and moments about two axes. The horizontal actuator shown in Figure 3-1 is also equipped with an axial-only load cell which allows for direct

verification of force measurements for slow tests and indirect for dynamic tests (correction is needed for the inertia force effects of the loading beam in Figure 3-1).

Figure 3-1: Schematic of single bearing testing machine (Kasalanati et al., 1999)

Figure 3-2 shows the instrumentation used to monitor the motion of the three internal components of the bearing. Note the two instruments that were needed for each component as the parts also exhibited torsion (Sarlis and Constantinou, 2013). This apparatus was used for the testing of only Configurations B and C.

Figure 3-2 shows string pots (potentiometers) SP-1, SP-2, SP-5 and SP-6 attached to the tip of the interior restrainer ring of the corresponding slide plates which are located at a distance z_{sp} from surface 1 for SP-1 and SP-2 and surface 4 for SP-5 and SP-6. SP-3 and SP-4 are attached at the mid-height of the rigid slider and directly measure the displacement of its center of mass (see Figure 1-1 for terminology). The measured displacements of the parts required post-processing on the basis of the geometry of the components and the location of the instruments in order to calculate displacements at each surface.

Figure 3-2: String-pot instrumentation of internal components for tested Triple FP bearing Configurations B and C
3.2 Configuration A Testing

Configuration A consisted of the isolators used for testing by Fenz and Constantinou (2008e). Surfaces 1, 2 and 3 of these isolators consisted of a material labeled as M1 and shown in Figure 3-3 on the right and Figure 3-4. Friction on surfaces 2 and 3 was much lower than on surface 1 due to the combination of higher pressure and the effect of some lubrication that was introduced in the 2008 tests. Surface 4 consisted of a high friction material labeled as M8 and shown in Figure 3-3. Experimental results for the isolators of Configuration A are presented in Figure 3-5. The left column of the graphs shows results for tests with imposed lateral motion at 0.01Hz frequency and the right column shows results for 0.3Hz frequency. The results are in the form of loops of the horizontal force normalized by the vertical force versus the displacement of the top concave plate with respect to the bottom concave plate. The notation used in the graphs is as follows for one of the four tested bearings: M1LC1-M1LC4 denotes the isolator that consisted of the bottom slide plate with material M1 used in the shake table testing in the bearing placed on top of load cell 1 (Figure 2-3), together with the bottom slide plate with material M1 used in the shake table testing in the bearing placed on top of load cell 4. These tests were conducted one month prior to the shake table tests.



Figure 3-3: Views of slide plates with material M8 (left row) and material M1 (right row) used in Configuration A isolators



Figure 3-4: Top view of rigid sliders with material M1 used for surfaces 2 and 3 in Configuration A isolators



Figure 3-5: Normalized force-displacement loops of Configuration A TFP isolators

3.3 Configuration B Testing

The bearings for Configuration B consisted of the same materials as those of Configuration A and have not been tested prior to the shake table testing. Rather, they were tested five months after the completion of the shake table tests. The bearings have not been cleaned or conditioned in the period between the shake table and the bearing machine testing. Results are presented in Figure 3-6. For these tests, the bearings were placed in the bearing testing machine at an offset that led to un-symmetric displacement input. TSB1 denotes the isolator that was located on top of load cell 1 (see Figure 2-7), TSB2 on top of load cell 2, etc. in the shake table tests. Observations in the results of Configuration B are:

- 1. During testing, bearing TSB3 exhibited stick-slip phenomena on surface 4 that were pronounced in low velocity tests (f=0.02Hz). While the phenomenon of stick-slip may be artificial and created by the test apparatus and/or any corrections of errors due to inertia effects (see Section 4 of Constantinou et al., 2007), it is believed that it was real and the result of high breakaway friction coefficient. The phenomenon was not observed in the shake table testing and it was barely observed in the faster test machine tests because frictional heating eliminated the difference between breakaway and sliding friction values.
- 2. The internal bearing parts (BSP; bottom slide plate, RS; rigid slider and TSP; top slide plate) exhibited significant torsional rotations as indicated by the results of Figure 3-7 which presents recorded values of the torsion angle during the tests for which the loops are presented in Figure 3-6. The torsion angle was calculated from the difference between the measurements of the two displacement transducers of each part shown in Figure 3-2 and divided by the distance between the attachments of the two transducers to the parts. It can be seen that some components exhibited up to 70° angle of rotation about the vertical axis. This behavior was also occasionally observed but not directly measured in the shake table tests in all configurations. It is caused by uneven distribution of traction forces on the sliding surfaces. The motion resulted in changes in the displacement capacities of the internal parts.
- 3. In test TSB4-f=0.1Hz, the tested bearings exhibited uplift so that the normalized force could not be obtained (division by zero). For this test the lateral force-displacement loop and the history of the vertical force on the bearing are shown in Figure 3-8. Uplift can be recognized when the lateral force is zero over a range of displacements. Uplift occurred because of inability to control the axial load on the tested bearings, particularly at high speed motion.



Figure 3-6: Normalized force-displacement loops of Configuration B TFP isolators



Figure 3-7: Torsion angle of internal components of TFP isolators of Configuration B



3.4 Configuration C Testing

Experimental results for the isolators of Configuration C are presented in Figure 3-9. Testing was conducted after the shake table testing. The isolators were assembled with friction on surfaces 1 and 4 having a value that is much smaller than the value of friction on the inner surfaces 2 and 3. This is an unusual configuration of which the behavior cannot be predicted by the conventional models of TFP bearings. Rather, the more advanced theory in Sarlis and Constantinou (2013) is capable of describing their behavior. For these isolators, motion initiates simultaneously on surfaces 1 and 4 when the lateral force becomes equal to the highest friction force among the two surfaces 1 and 4. In theory, motion on surfaces 2 and 3 will not initiate until the following two incidents occur: a) the displacement capacity of surfaces 1 and 4 is consumed, and b) the lateral force becomes equal to the highest friction force among surfaces 2 and 3. Between incidents a) and b) above there is an abrupt increase in the isolator lateral force. Actually, the displacement capacity of surfaces 1 and 4 cannot be simultaneously consumed as a result of initial offsets of the TFP surfaces caused by misalignments in the top concave plate. Such complex cases can be analyzed using the theory presented in Sarlis and Constantinou (2013). For the results in Figure 3-9, the measurements of both the isolator and actuator load cells are shown as some small differences were observed in the two independent measurements. Also, Figure 3-10 presents results on the torsional motion exhibited by the internal components of the bearings of Configuration C.



Figure 3-9: Normalized force-displacement loops of Configuration C TFP isolators



Figure 3-10: Torsion angle of internal TFP bearing components exhibited of the isolators of Configuration C

3.5 Additional Topics on the Testing of TFP Isolators

Figure 3-11 presents a comparison of experimental normalized force-displacement loops for a bearing of Configuration A under different axial loads. Figure 3-11 on the left shows results for an isolator having material M8 on surfaces 1 and 4 and subjected to a normal load of a) N=107kN and b) N=44kN. Figure 3-11 on the right shows results for the case in which the high friction material M8 was replaced by the low friction material M1. For both isolators, the inner surfaces 2 and 3 consist of material M1. Note that there is small effect of load on the behavior of the bearings due to the effect of pressure on the coefficient of friction at surface 1 and 4 where the apparent bearing pressure varies between 6.4 and 14.0MPa. In contrast, there is no effect on the friction coefficient of surfaces 2 and 3 where pressure varies between 25 and 55MPa-already large values for which pressure does not have significant effects (Constantinou et al, 2007).



Figure 3-11: Normalized force-displacement loops of TFP isolators at different loads

A subject investigated in the testing arises when isolators exhibit differences in vertical displacements. This situation occurs when isolators of different geometric and frictional properties are combined in the isolation system. It also occurs when identical isolators are used but natural variability in frictional properties causes differential vertical displacements of the isolator parts. Differential vertical motion results in redistribution of the axial load on the isolators in addition to variations due to overturning moment and vertical earthquake effects. Figure 3-12 shows results from tests conducted at the single bearing machine for two isolators having the same geometry but different friction properties: a) one with material M8 (high friction) on surfaces 1 and 4 and b) one with material M1 (low friction) on surfaces 1 and 4 and subjected to identical displacement inputs. The two isolators exhibit different vertical displacements. This occurs because the isolator with the higher friction M8 material exhibits larger displacements on surfaces 2 and 3 prior to initiation of motion on surfaces 1 and 4 than the isolator with the lower friction M1 material (note that surfaces 2 and 3 have small radius of

curvature which affects the vertical motion). Note in Figure 3-12 that there is permanent vertical displacement for the bearing of the higher friction material M8 despite the fact that the bearing has no permanent horizontal displacement. This is due to the fact that there is permanent displacement of the internal components of the bearing.



Figure 3-12: Vertical displacement histories of isolators with a) low friction material M1 on surfaces 1 and 4 and b) high friction material M8 on surfaces 1 and 4

Permanent vertical displacements such as those shown in Figure 3-12 can cause redistribution of the axial loads on the isolators after the seismic shaking ends. As an example, Figure 3-13 shows the vertical load on the four isolators (calculated using the procedure described in Section 2.2) at the start of consecutive shake table tests conducted for the three-story model structure. There is vertical load re-distribution at the conclusion of tests 2 and 13, 14 and again at 15, when the bearings returned to their original condition. Note that the load shifts so that more load is carried by the two bearings along the diagonal NE-SW and less load by the bearings on the diagonal NW-SE. Such shift in the load can easily occur due to the large vertical stiffness of the bearings and the large stiffness on the base-mat supporting the structure on top of the bearings. Under such conditions, small differences in height are due to misalignments, small differences in friction values even for otherwise identical bearings and slightly different initial conditions for the bearings.



Figure 3-13: Normal load values recorded at the start of consecutive shake table tests

SECTION 4 SHAKE TABLE TESTING RESULTS

4.1 Introduction

This section presents experimental results of the shake table testing of the 3-story structure shown in Figure 2-1 to Figure 2-3 as follows:

- 1. Section 4.2 presents results for the fixed structure that are used to identify the superstructure properties.
- 2. Section 4.3 presents a testing summary and description of the ground motions used for the shake table tests of the isolated structure.
- 3. Section 4.4 presents results for the isolated structure for low (displacements <50mm) and moderate (displacements <100mm) amplitude ground motions.
- 4. Section 4.5 presents results for the isolated structure subjected to strong ground motions that result in stiffening of the isolators and in some cases contact with the restrainers.
- 5. Section 4.6 presents results that investigate the effect of the vertical component of ground motions on the horizontal response of the isolation system and superstructure.

4.2 Fixed-base Structure

Prior to testing the isolated structure, the superstructure was identified by directly connecting the base, without the isolators, on the load cells (see Figure 2-3) and subjecting it to shake table motion. For the identification of the superstructure properties, the shake table was driven in white noise motion with frequency content of 0 to 50Hz, amplitude of 0.1g and 60 second duration. The transfer functions were obtained (see Bracci et al., 1992 for a description of the process) using records of acceleration recorded at each floor and the shake table. They are shown in Figure 4-1. The mode shape, period and damping ratio of each of the three translational (testing direction) modes of the superstructure were derived from the transfer functions (see Bracci et al., 1992) and are presented in Table 4-1.

	superstructure obtained in 1617 amplitude white hoise testing													
ĺ	Mode	Period	Damping	Mode Shape										
	No.	(sec)	Ratio	1 st floor	2 nd floor	3 rd floor								
	1^{st}	0.299	0.0862	0.415	0.753	1.000								
	2^{nd}	0.077	0.0137	1.216	0.816	-1.000								
	3 rd	0.046	0.0078	2.364	-2.199	1.000								

 Table 4-1: Modal shape, period and damping ratio for three modes of vibration of superstructure obtained in low amplitude white noise testing



Figure 4-1: Amplitude of transfer functions of superstructure obtained in low amplitude white noise testing

The structure has a high damping ratio in the first mode, something also observed in previous identification of the complete 6-story model (Fenz and Constantinou, 2008e and Wolff and Constantinou, 2004). This is attributed to slippage in the connections of the concrete blocks to the steel frame. The damping is dependent on the amplitude of motion, hence excitation too. It is largest at small amplitude vibration with rich frequency content. The structure was also identified in low amplitude (to prevent yielding) seismic excitation using motion ATL 270 (see Table 4-3). Results are presented in Table 4-2 and Figure 4-2. There is some difference between the two sets of results, which is typical of the difficulties in the identification of models that are not exactly linear elastic and linear viscous.

Mode	Period	Damping)e	
No.	(sec)	Ratio	1 st floor	2 nd floor	3 rd floor
1^{st}	0.277	0.0597	0.385	0.746	1.000
2^{nd}	0.077	0.0135	1.217	0.803	-1.000
3 rd	0.045	0.0060	2.528	-2.328	1.000

 Table 4-2: Modal shape, period and damping ratio for three modes of vibration of superstructure obtained in low amplitude seismic testing with motion ATL 270



Figure 4-2: Amplitude of transfer functions of superstructure obtained in low amplitude seismic testing with motion ATL 270

4.3 Testing Summary and Selection of Ground Motions

Table 4-3 presents characteristics of the ground motions that were used for the shake table tests. The majority of the ground motions selected for this study have near fault characteristics since these typically impose large displacement demands on isolated structures. Due to similitude requirements, all the ground motions had to be scaled in time by a factor of 0.5. This scale factor alone was not sufficient to cause displacements in the stiffening regimes of the TFP bearings. In order to amplify the effect of the utilized ground motions and excite the structure in Regime V, scales in time larger than 0.5 and scales in accelerations larger than 1.0 were used, which distorted similitude.

The following should be noted about the results that are presented in Section 4:

- 1. Displacements and accelerations were directly measured by string pots and accelerometers, respectively. Relative displacements were calculated by subtracting the records of displacements at two points.
- 2. All results presented here are un-processed with the exception of a digital 50Hz low pass filter that was applied directly by the data acquisition system.
- 3. The vertical acceleration of the base-mat and shake table was calculated using the average of the measurements of four accelerometers that were located at the four opposite corners of the base-mat and shake table.

4. The normalized base shear was calculated from records of acceleration after multiplication by the effective masses and addition over the height of the model (F_{acc} as given by Equation (2-1)) and dividing by the sum of the instantaneous vertical load measured by the four load cells. Small fluctuations in the normalized base shear loops occur because of a) errors in measurements of the vertical force by the load cells, and b) variations of friction due to pressure changes. Large fluctuations typically occur because of uplift of the structure, which was observed in tests that included vertical excitation and are presented in Section 4.6. Note also that the normalized base shear loops are less accurate than the non-normalized loops since they are divided by the sum of the load cell measurements and thus are susceptible to load cell error measurements.

Earthquake/ Date	Station	Component Notation	M _w	PGA (g)	PGV (cm/sec)	PGD (cm)
San Fernando 2/1971	CDMG 279 Pacoima Dam, Upper Left Abutment	PUL-164	6.6	1.16	75.6	18.1
Northridge-01 1/1994	CDMG 24514 Sylmar - Olive View Med FF	SYL-360	6.7	0.70	95.4	21.9
Northridge-01 1/1994	USGS/VA 637 LA - Sepulveda VA Hospital	0637-270	6.7	0.80	74.1	16.3
Chi-Chi, Taiwan 9/1999	CWB 9999936 TCU129	ТСИ-129-Е	7.6	0.79	47.3	38.7
Kobe 1/1995	JMA 99999 KJMA	KJM-000	6.9	0.71	77.8	18.9
Northridge-01 1/1994	CDMG 24279 Newhall - Fire Station	NWH-360	6.7	0.70	81.8	26.1
N. Palm Springs 07/1986	USGS 5231 Anza - Tule Canyon	ATL-270	6.06	0.10	7.27	0.73
M _w : Moment Magnit	ude. PGA: Peak Ground Acceleration	on. PGV: Peak Grou	und Velo	city. PGD:	Peak Ground	l Displ.

Table 4-3: Ground motions used for the Triple FP testing

4.4 Isolated Structure Results for Low and Moderate Amplitude Excitations

This section presents results for configurations tested with selected ground motions of low and moderate amplitude so that the isolators did not exhibit stiffening. A complete set of results is presented in Appendix A. For this set of tests, there was no vertical component of earthquake applied to the structure apart from some unintentional high frequency vertical excitation that existed in all tests. Table 4-4 presents the recorded peak response quantities for selected ground motions and for Configurations A, B and C. The response quantities are: (a) Base (or Isolator) displacement, (b) Inter-story drift as percentage of story height, (c) Floor and base acceleration, and (d) Base shear force (BS).

Figure 4-3 to Figure 4-8 presents results on base shear-base displacement loops, drift histories and floor 5%-damped acceleration spectra for selected ground motions for each of the tested configurations. Additional results are presented in the appendices. Note that the time scale

factors reported herein are applied to the time step of the model scaled ground motion. For example, the test designation SYL360 (0.5/1.3) denotes that ground motion SYL360 was applied in the longitudinal direction, the original acceleration values were multiplied by factor 0.5 and the duration of the motion scaled for similitude was additionally multiplied by factor 1.3 (that is, the original motion was first compressed in time by factor 2 for similitude and then the duration was further multiplied by factor 1.3).

Configuration A														
Ground Motion	Mult	iplier	Base displacement (mm) ³			Story	drift height	(% of)	Floo	BS				
WIOTION	A ¹	t^2	In.	Max	Res	Ch.	1	2	3	Base	1	2	3	
SYL360	1.0	1.0	2	86	-2	88	0.31	0.39	0.54	0.51	0.41	0.42	0.48	56
SYL360	0.5	1.3	-4	38	-12	33	0.22	0.33	0.32	0.36	0.39	0.42	0.43	34
PUL164	0.5	1.0	-7	26	-8	18	0.20	0.25	0.19	0.40	0.37	0.38	0.38	28
PUL164	1.0	1.0	-8	62	-9	57	0.28	0.41	0.27	0.48	0.44	0.50	0.55	42
PUL164	0.8	1.4	-8	95	10	103	0.32	0.54	0.29	0.47	0.46	0.45	0.54	63
NWH360	0.5	1.0	-8	19	0	27	0.20	0.23	0.18	0.21	0.26	0.23	0.28	34
NWH360	1.0	1.0	1	45	-5	46	0.24	0.38	0.27	0.34	0.33	0.33	0.38	38
NWH360	1.5	1.0	-2	82	-1	85	0.37	0.43	0.36	0.50	0.46	0.52	0.45	53
KJM000	0.5	1.0	-4	17	-2	15	0.26	0.26	0.19	0.27	0.28	0.33	0.35	31
KJM000	1.0	1.0	-2	40	0	38	0.33	0.41	0.24	0.35	0.34	0.39	0.43	37
KJM000	1.5	1.0	3	71	1	67	0.32	0.55	0.35	0.42	0.45	0.44	0.56	48
637270	0.5	1.0	0	22	5	22	0.23	0.33	0.16	0.24	0.26	0.27	0.32	35
637270	1.0	1.0	5	45	5	41	0.26	0.41	0.21	0.32	0.33	0.36	0.39	41
TCU129E	1.0	2.0	1	68	2	69	0.33	0.48	0.41	0.44	0.47	0.46	0.63	46
					(Config	gurati	on B						
SYL360	0.5	1.0	0	25	1	25	0.21	0.27	0.16	0.29	0.27	0.25	0.35	23
SYL360	1.0	1.0	1	76	-2	77	0.24	0.39	0.30	0.36	0.33	0.32	0.39	41
SYL360	0.5	1.3	-3	36	0	33	0.20	0.27	0.21	0.26	0.27	0.30	0.30	33
PUL164	1.0	1.0	-2	66	-3	64	0.24	0.36	0.25	0.36	0.37	0.36	0.48	39
PUL164	0.5	1.4	23	87	22	65	0.20	0.35	0.25	0.30	0.36	0.34	0.32	40
KJM000	0.5	1.5	0	42	1	42	0.21	0.28	0.17	0.26	0.23	0.24	0.28	35
KJM000	1.0	1.5	-5	88	-6	85	0.32	0.46	0.31	0.34	0.31	0.36	0.41	46
KJM000	1.0	1.0	15	52	18	56	0.26	0.36	0.23	0.36	0.29	0.31	0.42	39
TCU129E	1.0	1.0	22	46	24	23	0.26	0.36	0.22	0.41	0.42	0.36	0.47	28
NWH360	0.5	1.0	14	36	16	22	0.22	0.26	0.21	0.20	0.19	0.22	0.25	31
NWH360	1.0	1.0	16	64	9	50	0.22	0.35	0.24	0.31	0.27	0.27	0.34	39
637270	0.5	1.8	-1	82	5	83	0.19	0.40	0.19	0.27	0.26	0.29	0.31	44
637270	0.5	1.0	5	28	10	23	0.15	0.27	0.13	0.19	0.21	0.19	0.25	28
637270	1.0	1.0	10	55	14	45	0.22	0.39	0.17	0.26	0.28	0.35	0.31	38
					(Config	gurati	on C						
SYL360	0.5	1.0	0	19	4	19	0.26	0.40	0.26	0.58	0.31	0.42	0.53	28
SYL360	1.0	1.0	4	55	-1	55	0.27	0.36	0.29	0.50	0.40	0.40	0.61	38
SYL360	1.0	1.2	7	92	12	98	0.29	0.39	0.34	0.58	0.38	0.44	0.58	42
PUL164	0.5	1.4	-2	26	2	28	0.30	0.37	0.28	0.56	0.57	0.55	0.62	33
637270	0.5	1.8	9	66	3	56	0.23	0.43	0.24	0.32	0.27	0.30	0.42	37
NWH360	0.5	1.5	-3	34	4	37	0.29	0.36	0.25	0.31	0.28	0.34	0.43	34
KJM000	0.5	1.5	0	40	1	40	0.30	0.44	0.33	0.40	0.36	0.38	0.59	37
TCU129E	0.5	2.0	5	19	-2	20	0.40	0.41	0.31	0.45	0.37	0.42	0.64	32
			-											

Table 4-4: Peak response quantities in tests with low and moderate amplitude ground motions

A multiplies accelerations of original ground motion
 t multiplies time step of the ground motion in addition to the 0.5 factor that is applied due to similitude

3. In. is the initial, Res is the residual and Ch. is the maximum change of the base displacement (with respect to In.)

4. Base Shear



Figure 4-3: Experimental results for Configuration A and ground motion 0637-270 scaled by factors 1.0 in acceleration and 1.0 in time



Figure 4-4: Experimental results for Configuration A and ground motion TCU-129-E scaled by factors 1.0 in acceleration and 2.0 in time



Figure 4-5: Experimental results for Configuration B and ground motion SYL-360 scaled by factors 0.5 in acceleration and 1.3 in time



Figure 4-6: Experimental results for Configuration B and ground motion KJM-000 scaled by factors 1.0 in acceleration and 1.5 in time



Figure 4-7: Experimental results for Configuration C and ground motion SYL-360 scaled by factors 0.5 in acceleration and 1.0 in time



Figure 4-8: Experimental results for Configuration C and ground motion KJM-000 scaled by factors 0.5 in acceleration and 1.8 in time

4.5 **Isolated Structure Results for High Amplitude Excitations**

This section presents results of shake table tests with strong excitations so that the isolators exhibited stiffening (Regimes IV and V) and contact with the restrainer rings. There was no vertical excitation in this group of tests apart from some very high frequency parasitic vertical excitation. Table 4-5 presents experimental results of peak response quantities for all tested configurations for which the isolators exhibited stiffening.

	1		D	see dier	laaam	ont	Stor	v drift	(0/					
Ground	Mult	iplier	(mm) ³					of height)			Floor acceleration (g)			
Motion	A ¹	t ²	In.	Max	Res	Ch.	1	2	3	Base	1	2	3	(kN)
SYL360	1.1	1.3	-6	143	5	137	0.34	0.54	0.49	0.71	0.58	0.74	0.75	83
637270	1.0	1.5	9	111	25	102	0.33	0.52	0.31	0.53	0.49	0.50	0.60	70
TCU129E	1.2	2.0	2	102	-17	103	0.53	0.49	0.75	0.50	0.47	0.42	0.46	60
PUL164	1.0	1.4	10	154	31	143	0.44	0.70	0.58	0.66	0.63	0.64	0.72	103
Configuration B														
SYL360	1.0	1.3	-2	138	8	136	0.27	0.43	0.49	0.61	0.56	0.64	0.53	64
PUL164	1.0	1.3	1	135	17	134	0.43	0.57	0.35	0.44	0.41	0.45	0.51	62
PUL164	0.8	1.4	20	143	25	123	0.37	0.52	0.49	0.58	0.50	0.47	0.51	71
NWH360	0.7	1.8	4	117	13	113	0.38	0.49	0.33	0.35	0.38	0.39	0.40	58
NWH360	0.8	1.6	-1	115	24	117	0.33	0.52	0.32	0.47	0.45	0.54	0.56	61
KJM000	0.9	1.8	-3	123	-7	120	0.43	0.51	0.42	0.57	0.53	0.52	0.54	61
637270	1.0	1.5	10	124	22	114	0.30	0.46	0.26	0.48	0.40	0.46	0.46	60
637270	0.8	1.8	21	158	5	137	0.72	0.94	0.84	1.23	0.93	0.91	0.80	131
					(Config	guratio	n C						
637270	0.8	1.8	-4	140	-3	144	0.32	0.68	0.54	0.92	0.74	0.76	0.82	93
NWH360	1.2	1.5	2	113	0	115	0.38	0.44	0.38	0.37	0.42	0.44	0.56	47
KJM000	1.5	1.3	-7	104	-8	111	0.40	0.60	0.40	0.50	0.51	0.56	0.72	45
PUL164	0.5	2.0	1	107	5	107	0.34	0.44	0.37	0.64	0.64	0.68	0.77	47
PUL164	1.1	1.4	7	122	40	115	0.35	0.39	0.38	0.45	0.61	0.58	0.44	52
1 A multipli		laration	s of or	iginal g	cound n	notion								

Table 4-5	: Peak res	ponse qua	antities for	high am	nlitude	motions
	• I can i co	ponse que	antitics for	men am	philude	motions

1. A multiplies accelerations of original ground motion

2. t multiplies time step of the ground motion in addition to the 0.5 factor that is applied due to similitude

3. In. is the initial, Res is the residual and Ch. is the maximum change of the base displacement (with respect to In.)

4. Base Shear

Figure 4-9 to Figure 4-11 present results on base shear-base displacement loops, drift histories, floor acceleration histories, bearing axial load histories and floor 5%-damped acceleration spectra for one ground motion for each of the tested configurations. Additional results are presented in the appendices. Note that the time scale factors reported herein are applied to the time step of the model scaled ground motion. For example, the test designation SYL360 (0.5/1.3)denotes that ground motion SYL360 was applied in the longitudinal direction, the original acceleration values were multiplied by factor 0.5 and the duration of the motion scaled for similitude was additionally multiplied by factor 1.3 (that is, the original motion was first compressed in time by factor 2 for similitude and then the duration was further multiplied by

factor 1.3). Also, Figure 4-12 presents frames captured from the video of the motion of isolator TFP-1 (located on the SE corner; see Figure 2-7) where the isolator exhibits its maximum displacement for the test designated 0637270(0.8/1.8). Graphs of results for this test are shown in Figure 4-10. A complete set of results is presented in Appendix A.



Figure 4-9: Experimental results for Configuration A and ground motion PUL-164 scaled by factors 1.0 in acceleration and 1.4 in time



Figure 4-9 (cont'd): Experimental results for Configuration A and ground motion PUL-164 scaled by factors 1.0 in acceleration and 1.4 in time



Figure 4-10: Experimental results for Configuration B and ground motion 0637-270 scaled by factors 0.8 in acceleration and 1.8 in time



Figure 4-10 (cont'd): Experimental results for Configuration B and ground motion 0637-270 scaled by factors 0.8 in acceleration and 1.8 in time



Figure 4-11: Experimental results for Configuration C and motion 0637-270 scaled factors 0.8 in acceleration and 1.8 in time



Figure 4-11 (cont'd): Experimental results for Configuration C and ground motion 0637-270 scaled factors 0.8 in acceleration and 1.8 in time





(d) t=0.12sec (e) t=0.16 (f) t=0.20 Figure 4-12: Captured frames of TFP isolator motion during maximum deformation for Configuration B and ground motion 0637-270 (results presented in Figure 4-10)

4.6 Isolated Structure Results for Tests with Vertical Component of Ground Motion

This section presents comparisons of experimental results with only horizontal excitation applied in the longitudinal direction (case L) and with combined horizontal and vertical excitation (case L+V).

Peak response results for all tested configurations are presented in Table 4-6. Figure 4-13 to Figure 4-15 present comparison of results (base shear loops, normalized base shear loops, drift histories, acceleration histories and floor spectra) for tests selected from Table 4-6 for one ground motion for each configuration. In Figure 4-13 to Figure 4-15, tests without a vertical component are denoted as "L" and tests with a vertical component are denoted as "L+V". Graphical results from the remaining tests of Table 4-6 are presented in the Appendices.

Configuration A														
Gr. Motion	Sc	ale	Ba	ise Disp (mi	olacem m) ³	ent	Stor of	y Drif f heigh	t (% t)	Floo	BS ⁴			
	A ¹	t ²	In.	Max	Res	Ch.	1	2	3	Base	1	2	3	(KN)
L=PUL164			-8	62	-9	57	0.28	0.41	0.27	0.48	0.44	0.50	0.55	42
L=PUL164 V=PULUP	1.0	1.0	-4	61	-8	57	0.33	0.51	0.39	0.90	0.93	0.82	0.87	55
L=NWH360			1	45	-5	46	0.24	0.38	0.27	0.34	0.33	0.33	0.38	38
L=NWH360 V=NWHUP	1.0	1.0	3	45	-4	48	0.26	0.43	0.31	0.65	0.82	0.79	0.75	62
L=KJM000			-2	40	0	38	0.33	0.41	0.24	0.35	0.34	0.39	0.43	37
L=KJM000 V=KJMUP	1.0	1.0	-1	41	0	40	0.40	0.47	0.30	0.52	0.64	0.68	0.57	58
L=0637270			5	45	5	41	0.26	0.41	0.21	0.32	0.33	0.36	0.39	41
L=0637270 V=0637UP	1.0	1.0	7	45	6	38	0.32	0.51	0.27	0.59	0.62	0.62	0.65	49
					(Config	guratio	n B	•		•	•		
L=NWH360			16	64	9	50	0.22	0.35	0.24	0.31	0.27	0.27	0.34	39
NWH360 V=NWHUP	1.0	1.0	1	52	1	51	0.30	0.43	0.28	0.62	0.86	1.04	0.73	54
L=KJM000			15	52	18	56	0.26	0.36	0.23	0.36	0.29	0.31	0.42	39
L=KJM000 V=KJMUP	1.0	1.0	8	47	6	55	0.28	0.42	0.28	0.54	0.51	0.64	0.53	46
L=TCU129			22	46	24	23	0.26	0.36	0.22	0.41	0.42	0.36	0.47	28
L=TCU129 V=TCUUP	1.0	1.0	7	29	3	23	0.23	0.38	0.33	0.40	0.41	0.35	0.48	34
L=637270			10	55	14	45	0.22	0.39	0.17	0.26	0.28	0.35	0.31	38
L=637270 V=0637UP	1.0	1.0	9	54	10	44	0.25	0.35	0.26	0.47	0.76	0.73	0.59	49
L=637270			5	129	21	124	0.31	0.50	0.32	0.46	0.39	0.46	0.54	64
L=637270 V=0637UP	0.7	1.8	15	138	17	123	0.39	0.46	0.43	0.66	0.64	0.77	0.59	70
						Config	guratio	n C						
L=637270			3	112	-3	109	0.30	0.49	0.32	0.40	0.34	0.36	0.54	46
L=637270 V=0637UP	0.7	1.8	1	98	-2	97	0.35	0.51	0.32	0.69	0.50	0.58	0.56	53
L=KJM000			1	68	-3	67	0.32	0.53	0.31	0.42	0.44	0.48	0.59	39
KJM000 V=KJMUP	1.0	1.5	0	61	-5	61	0.39	0.59	0.35	0.50	0.54	0.52	0.54	50

 Table 4-6: Peak response quantities obtained in horizontal and combined horizontal-vertical excitation

1. A multiplies accelerations of original ground motion

2. t multiplies time step of the ground motion in addition to the 0.5 factor that is applied due to similitude

3. In. is the initial, **Res** is the residual and **Ch.** is the maximum change of the base displacement (with respect to In.)

4. Base Shear



Figure 4-13: Experimental results for Configuration A and ground motion PUL-164 scaled by factors 1.0 in acceleration and 1.0 in time



Figure 4-13 (cont'd): Experimental results for Configuration A and ground motion PUL-164 scaled by factors 1.0 in acceleration and 1.0 in time


Figure 4-14: Experimental results for Configuration B and ground motion 0637-270 scaled by factors 0.7 in acceleration and 1.8 in time



Figure 4-14 (cont'd): Experimental results for Configuration B and ground motion 0637-270 scaled by factors 0.7 in acceleration and 1.8 in time



Figure 4-15: Experimental results for Configuration C and ground motion KJM-000 scaled by factors 1.0 in acceleration and 1.5 in time



Figure 4-15 (cont'd): Experimental results for Configuration C and ground motion KJM-000 scaled by factors 1.0 in acceleration and 1.5 in time

4.7 Comments on Experimental Results of Sections 4.4 to 4.6

In discussing the experimental results of Sections 4.4 to 4.6, it is important to first comment on the behavior of the three tested Triple FP configurations. Section 5 presents details on the frictional properties of the configurations. In summary, the four isolators exhibited different frictional properties. However, for the discussion herein, the weighted average values for the entire isolation system were used. Table 4-7 presents representative weighted friction values at high velocity for the four sliding surfaces of each of the three tested configurations. They are based on the data in Table 5-1 of Section 5.

Friction coefficient	Configuration A	Configuration B	Configuration C
μ_1	0.102	0.108	0.128
μ2	0.038	0.033	0.228
μ3	0.038	0.033	0.228
μ_4	0.173	0.155	0.128

Table 4-7: Weighted average friction coefficient values for tested configurations

Evidently, Configurations A and B are very similar in frictional and geometric properties (see also Table 3-1), characterized by capability to exhibit all five regimes, and to take advantage of the adaptive nature of the Triple FP isolator. Configuration C lacks these attributes and behaves as a high friction (value of 0.128) single FP or a double FP (with equal friction on the two sliding surfaces) isolator but with a final stiffening regime. With this background, the following observations are made from the results reported in Sections 4.4 to 4.6:

- 1. The results of Section 4.4 show the advantages of the adaptive Configurations A and B over the non-adaptive Configuration C. The advantages are particularly obvious for ground motions KJM000(0.5/1.5), NWH360(0.5/1.0), SYL360(1.0/1.0) and0637270(0.5/1.8) in Table 4-4 where Configurations B and C have the same base shear and base displacements but Configuration B has much less inter-story drifts and floor accelerations. The advantages of adaptive systems over non-adaptive systems have been discussed by Fenz and Constantinou (2008a to e), Morgan (2007) and Morgan and Mahin (2010).
- 2. In the results of Section 4.5, the isolators experience impact on all restrainer rings (surfaces 1, 2, 3 and 4) in Figure 4-9 and Figure 4-10. Also, as seen in Figure 4-11, the isolators are excited in their Regime V. In all cases, the isolators exhibited stable behavior under these extreme conditions. Also, note that the large shear forces reached in these experiments depend on the strength of the restrainer rings of the triple FP bearings. In the tests, the rings had very high strength that is unlikely to be achieved (or is desirable) in full size bearings. Full size bearings will have limited strength of the rings so that impact, like those experienced in the tests, would have most likely resulted in fracture of the impacted rings of surfaces 2 or 3, which would have limited the shear

force, allowed for some additional displacement and, likely, cause some damage to the sliding material.

As evident in the figures, upon reaching the isolator displacement capacity, the south side of the model uplifted for a short duration as indicated by the vanishing axial load record for the south side. This caused rocking in the structure, which in turn limited the floor accelerations. Moreover, in Figure 4-10 and Figure 4-11, the isolator maximum displacement values are larger than the theoretical displacement capacity of the isolators. As discussed in Section 3.3, Item 2, this is most likely caused by uneven distribution of traction forces on the sliding surfaces that results in torsional motion of the inner parts, which in turn causes changes in the displacement capacities of the internal parts.

3. In some tests and particularly in tests of configuration TSA with motion PUL-164 (1.0/1.0) and TSB with motion 0637-270 (0.7/1.8) (see graphs in Figure 4-13 to Figure 4-15), the vertical ground excitation had an important effect on floor accelerations but insignificant effect on base displacements (consistent with the results of Fenz et al. 2008e, Morgan, 2007 and Becker and Mahin, 2011) and some small effect on inter-story drifts. The floor spectra were also affected but over a limited range of frequencies, larger than about 10Hz in the time scale of the tests. It should also be noted that in the tests with (L+V) and without (L), the vertical ground excitation is not directly comparable due to large differences in parasitic rocking shake table motion in the tests. The rocking motion of the shake table was systematically larger in tests with vertical motion than in tests without it due to limited ability to control the shake table.

SECTION 5 ANALYTICAL PREDICTION OF RESPONSE

5.1 Introduction

This section presents the data on the friction properties of the Triple FP isolators used in the shake table testing, and presents comparisons of experimental results of the tested 3-story model structure to analytical results obtained with programs 3pleANI (Sarlis and Constantinou, 2013) and SAP2000 (Computers and Structures, 2007).

5.2 Identification of Friction Properties

The friction coefficient values for each sliding interface of each bearing used in the testing have been identified. The coefficient of friction is considered to be velocity-dependent and assumed to follow the relation (Constantinou et al., 1990):

$$\mu_{i} = \mu_{fi} - \left(\mu_{fi} - \mu_{si}\right)e^{-a_{i}|v_{i}|}$$
(5-1)

In Equation (5-1), μ_{fi} and μ_{si} are the values of the friction coefficient at large velocities (called *fast* herein) and at zero velocity (called *slow* herein), respectively, α_i is a rate parameter that controls the variation with velocity and v_i is the sliding velocity of the *i-th* surface. For identification of the three parameters needed to describe the model of friction of Equation (5-1), the recorded force-displacement loops had to be decomposed to loops of force versus sliding displacement for each sliding interface for at least three different velocities. This enabled the identification of values of the friction coefficients, which were then used to construct analytical force-displacement loops for comparison to the experimentally recorded loops.

For example, Figure 5-1 shows comparisons of experimental results for the isolators of Configuration A and analytical results obtained by the model of Fenz and Constantinou (2008a to e) following identification of the friction coefficient values. The isolators consist of particular interfaces as identified in Section 3.2 of this report. Results are presented for a slow test at frequency of 0.01Hz on the left column and for a fast test at frequency of 0.3Hz on the right column. The friction coefficient values used to construct the analytical loops are presented in each graph. In these graphs, μ_1 denotes the least friction value and μ_4 denotes the largest friction value among the two main sliding interfaces.



Figure 5-1: Comparison of experimental results obtained from testing of individual isolators and analytical results for Configuration A. Left column presents results at frequency of 0.01Hz; right column for frequency of 0.3Hz

Figure 5-2 presents theoretical force-displacement loops of the isolators of Configuration A using the identified slow and fast friction coefficients values in Figure 5-1. The four isolators are identified by the load cell number (LC1, LC2, LC3 and LC4) with reference to Figure 2-3 for the location of each load cell. The construction of these loops also requires values of the friction coefficient for the sliding interfaces 2 and 3 (see Figure 1-1), which were assumed to be equal. Values of the friction coefficient $\mu_2 = \mu_3$ were also identified from the experimental loops. The rate parameter could not be identified for the Configuration A bearings as there were insufficient test data at intermediate velocities for individual bearings. Rather, shake table test data were utilized. Values of the parameter are presented later in this report.



Figure 5-2: Analytical loops for Triple FP isolators of Configuration A

Figure 5-3 presents comparisons of experimental and analytical results for the isolators of Configuration B. TSB1 denotes the isolator that was placed on load cell LC1 (Figure 2-3), TSB2 is the isolator placed on load LC2, etc. Tests were conducted at 0.02Hz frequency (left column of graphs) for the identification of the slow friction coefficient values and at frequency of 0.5Hz (right column of graphs) for the identification of the fast friction coefficient values. The rate parameters were identified using an additional test conducted at 0.1Hz and shown in the center column of the graphs in Figure 5-3. All friction coefficient values were identified from the decomposed loops (force versus the sliding displacement of each surface) as shown in Figure 5-4 (for the same tests as those shown in Figure 5-3). The identified friction coefficient values are presented in each graph of Figure 5-3. Note that the abnormalities in the experimental loops of

TSB2 and TSB4 at frequency of 0.5Hz are caused by uplift of the bearings and therefore division by zero load in the normalization of the lateral force by the vertical load. The identification procedure was based on the decomposed loops of Figure 5-4 and the following considerations:

- 1. The slow friction coefficient was identified by matching the analytical loop with the minimum width of the experimental loop of each of sliding interfaces 1 and 4, measured at maximum displacement for the tests conducted at 0.02Hz frequency (essentially zero velocity).
- 2. The fast friction coefficient was identified by matching the analytical loop with the experimental loop at the zero displacement force intercept of each of sliding interfaces 1 and 4 (velocity is maximum) for the tests conducted at 0.5Hz frequency.
- 3. The slow friction of surfaces 2 and 3 was obtained from the isolator force-displacement loops of Figure 5-3 when velocity reverses sign. On unloading, the drop in force equals to twice the friction force on surfaces 2 and 3 at essentially zero velocity. The fast friction coefficient of surfaces 2 and 3 was difficult to determine so that approximate values were assigned based on a study of the loops of Figure 5-3.

The following are noted in the results of Figure 5-3:

- 1. The calibrated analytical model cannot capture the experimental behavior well during initial loading as a result of initial offsets of the Triple FP inner parts which existed in most of the tests.
- 2. Some of the isolators have different properties although they are composed from essentially the same materials. The only possible explanation for this behavior is the effect of contamination of the sliding interfaces with dust and lubricants during the numerous interchanges of parts in the conduction of testing.

Figure 5-5 shows comparisons of the analytical friction coefficient versus sliding velocity graphs to experimental results for surfaces 1 and 4 of the isolators of Configuration B. The experimental data on velocity were obtained by numerical differentiation of the surface displacement histories acquired by the instruments shown in Figure 3-2. The tests utilized in collecting the data in Figure 5-5 are those presented in Figure 5-4 and Figure 5-3. Note that the graphs of Figure 5-5 include information on the rate parameter a in units of sec/mm.

Figure 5-6 presents comparisons of experimental and analytical force-displacement loops for the isolators of Configuration C. Figure 5-7 shows the decomposed loops for the same tests whereas Figure 5-8 shows graphs of the friction coefficient as a function of the surface velocity. For these isolators, the friction coefficient for surfaces 1 and 4 had essentially the same value so that the calibrated model is based on the assumption of equal friction values. Note that in Figure 5-7 the displacements of surfaces 1 and 4 are not exactly equal as they should have been if the friction coefficient values were equal for the two surfaces. However, in this case, the difference is due to initial offsets in the internal components of the isolators at the start of each test. Note that the offset occurs naturally at the conclusion of a test even when the parts are centered at the start of the test. Accordingly, all tests but the first one started with initial offsets.



Figure 5-3: Comparison of experimental results obtained from testing of individual isolators and analytical results for Configuration B. Left column presents results for frequency of 0.02Hz; center column for 0.1Hz and right column for 0.5Hz



isolators of Configuration B at three different excitation frequencies



two isolators of Configuration B at three different excitation frequencies



Figure 5-5: Friction coefficient as function of velocity for surfaces 1 and 4 of isolators of Configuration B (parameter *a* in units of sec/mm)



Figure 5-6: Comparison of experimental results obtained from testing of individual isolators and analytical results for Configuration C. Left column presents results for frequency of 0.02Hz; center column for 0.1Hz and right column for 0.5Hz



isolators of Configuration C at three different excitation frequencies



Figure 5-7 (cont'd): Decomposed normalized force versus sliding displacement loops for two isolators of Configuration C at three different excitation frequencies



Figure 5-8: Friction coefficient as function of velocity for surfaces 1 and 4 of isolators of Configuration C (surfaces 1 and 4 are assumed to be identical) (parameter *a* in units of sec/mm)

5.3 Analytical Prediction Using Program 3pleANI

Analysis of the tested isolated model structure have been conducted in program 3pleANI (Sarlis and Constantinou, 2013). This program allows for:

- 1. Explicit modeling of the superstructure,
- 2. Use of an advanced model of the Triple FP isolator with unrestricted geometric and frictional parameters, and
- 3. Consideration of non-zero initial conditions.

Table 5-1 presents the identified fast and slow coefficient of friction values for the fours isolators in each of the three tested configurations on the shake table. These friction values are those used in program 3pleANI (denoted as $\mu_1, \mu_2, \mu_3, \mu_4$) which differ from the values in the theory of Fenz and Constantinou (2008a to 2008e) (denoted as $\overline{\mu}_1, \overline{\mu}_2, \overline{\mu}_3, \overline{\mu}_4$) as discussed in Section 1 herein. The two sets of friction coefficient values are related through Equation (1-1).

It should be noted that for the isolators of Configuration A, the friction coefficient values used for the analytical prediction of the experimental results are somewhat higher from the ones identified in the bearing machine tests (Section 5.2), whereas for Configurations B and C they

are identical. This was actual behavior and was likely caused by contamination of the interfaces during multiple disassembly and reassembly of the bearings in the course of the test program.

	I l. 4	Friction coefficient				
	Isolator		Configuration A (TSA)			
	ID	Surface 1	Surface 2	Surface 3	Surface 4	
Fast	TSA1	0.104	0.042	0.042	0.120	
	TSA2	0.103	0.038	0.038	0.200	
	TSA3	0.084	0.045	0.045	0.201	
	TSA4	0.118	0.028	0.028	0.170	
Slow	TSA1	0.047	0.014	0.014	0.091	
	TSA2	0.056	0.014	0.014	0.164	
	TSA3	0.042	0.021	0.021	0.157	
	TSA4	0.063	0.014	0.014	0.119	
		Configuration B (TSB)				
Fast	TSB1	0.142	0.053	0.053	0.178	
	TSB2	0.099	0.015	0.015	0.184	
	TSB3	0.074	0.031	0.031	0.090	
	TSB4	0.118	0.031	0.031	0.166	
Slow	TSB1	0.078	0.031	0.031	0.134	
	TSB2	0.052	0.008	0.008	0.101	
	TSB3	0.030	0.011	0.011	0.079	
	TSB4	0.039	0.008	0.008	0.102	
		Configuration C (TSC)				
Fast	TSC1	0.124	0.258	0.258	0.124	
	TSC2	0.138	0.248	0.248	0.138	
	TSC3	0.124	0.212	0.212	0.124	
	TSC4	0.124	0.193	0.193	0.124	
Slow	TSC1	0.064	0.184	0.184	0.064	
	TSC2	0.055	0.178	0.178	0.055	
	TSC3	0.064	0.150	0.150	0.064	
	TSC4	0.064	0.136	0.136	0.064	

Table 5-1: Friction coefficients values ($\mu_1, \mu_2, \mu_3, \mu_4$) used in program 3PLEANI for analytical prediction of response

Comparisons of analytical results produced by 3pleANI and experimental results are presented in Figure 5-10 for Configuration B and Figure 5-11 for Configuration C with experimental results obtained in the testing of the isolators in the single bearing testing machine. For the simulation results in program 3pleANI, the top concave plate (TCP) was subjected to a prescribed displacement and varying axial load which were the ones recorded in the experiments. Also in the simulations, the initial offsets of the inner parts of the bearings as measured at the beginning of each test by the instrumentation shown in Figure 3-2 were included in the analysis. Isolator TSB4 in Figure 5-10 underwent uplift so that its normalized loop is not defined during the uplift duration. For this test, the non-normalized loop is shown in Figure 5-12. Note that program

3pleANI can analyze isolators exhibiting uplift. It can be seen that analytical and experimental results are in good agreement except for minor differences attributed to:

- Load cell error measurements which were obvious in the comparisons between the actuator load cell and isolator load cells in Figure 3-9. In fact, the isolator load cell had to be repaired after these tests. In testing friction pendulum isolators, the measurements of the load cells can easily be verified by comparing the analytically predicted stiffness with the experimentally measured stiffness. Such comparisons led to the requirement of multiplying the experimental results with different scaled factors for each test
- 2. The presence of the rubber seal which was not accounted for in the analyses. The rubber seal has a more pronounced effect in the reduced size tested bearings than in full size isolators. The seal affects the stiffness of Regimes I, II, IV and V for bearings with $\mu_2 = \mu_3 < \mu_1 < \mu_4$ and the stiffness of Regime V for bearings with $\mu_1 = \mu_4 < \mu_2 = \mu_3$ (see Figure 1-2). It is noted that the seal was omitted here for simplicity. However, an example of an analysis in 3pleANI that shows the effect of the seal is shown in Figure 5-9. Note that in Figure 5-9 the analysis that includes the seal over-predicts the stiffness because the exact seal properties for the analysis were unknown. For more details the reader is referred to Sarlis and Constantinou (2013).



Figure 5-9: Rubber seal effect in the reduced size tested isolators



Figure 5-10: Comparison of experimental (bearing test machine) and analytical (red lineprogram 3pleANI) normalized force-displacement loops for Configuration B isolators



Figure 5-11: Comparison of experimental (bearing test machine) and analytical (red lineprogram 3pleANI) normalized force-displacement loops for Configuration C isolators



Figure 5-12: Comparison of experimental (bearing test machine) and analytical (red lineprogram 3pleANI) force-displacement loops of isolator TSB4 (with uplift)

5.3 Structural Model and Analytical Results of Fixed-base Superstructure using Program 3pleANI

The stiffness and damping matrices of the superstructure of the isolated model, fixed at the base, were constructed using the procedures presented in Bracci et al. (1992) and the identified mode shapes and periods (shown in Table 4-1 and Table 4-2). These matrices are presented in Table 5-2. Note that two sets of matrices are presented as based on data obtained in white noise and in seismic motion identification tests. In the analysis programs 3pleANI (and later program SAP2000), the stiffness matrix was derived from the white noise data. For the damping matrix, the mode shapes identified in the white noise tests were used (consistent with the construction of the stiffness matrix). However, the damping ratios obtained in the seismic identification tests were used as these tests resulted in more realistic values.

Analysis of the fixed-base superstructure with seismic motion ATL 270 at its base was conducted in 3pleANI and results are compared to experimental results in Figure 5-13 to 5-14. The figures show histories of inter-story drift and floor accelerations, and the 5%-damped floor acceleration spectra. The analytical results are in good agreement with the experimental results except for the peak values of response which are occasionally over-estimated or under-estimated by the analytical model. There are two reasons for this: a) the experimental response has not been filtered (except for a filter at 50Hz) so that it contains noise, and b) the analytical model is based on linear elastic and linear viscous behavior, whereas the fixed superstructure exhibits nonlinear behavior due to flexing and slipping of the concrete block connections.

Test No.	Stiffness matrix (kN/cm)	Damping Matrix (kN-sec/cm)						
White Noise 0.1g, 0-50Hz	$K = \begin{bmatrix} 555.5 & -333.2 & 26.6 \\ -333.2 & 515.8 & -233.2 \\ 26.6 & -233.2 & 185.9 \end{bmatrix}$	$C = \begin{bmatrix} 0.118 & 0.018 & 0.021 \\ 0.018 & 0.123 & 0.028 \\ 0.021 & 0.028 & 0.143 \end{bmatrix}$						
Ground motion ATL-270	$K = \begin{bmatrix} 558.9 & -330.2 & 40.3 \\ -330.2 & 520.9 & -242.5 \\ 40.3 & -242.5 & 190.1 \end{bmatrix}$	$C = \begin{bmatrix} 0.101 & 0.018 & 0.004 \\ 0.018 & 0.097 & 0.015 \\ 0.004 & 0.015 & 0.116 \end{bmatrix}$						

 Table 5-2: Stiffness and damping matrices constructed from identified mode shapes and damping ratios



Figure 5-13: Comparison of analytical (program 3pleANI) and experimental results for inner-story drift of fixed structure for ground motion ATL-270



Figure 5-14: Comparison of analytical (program 3pleANI) and experimental results for floor acceleration of fixed structure for ground motion ATL-270



Figure 5-15: Comparison of analytical (program 3pleANI) and experimental results for 5%-damped floor response spectra of fixed structure for ground motion ATL-270

5.4 Analytical Results of Isolated Structure using Program 3pleANI

Comparisons of experimental and analytical results for the isolated structure are presented in Figure 5-16 to Figure 5-18. For each isolation system configuration, six ground motions are shown; two of small amplitude (displacements less than 50mm) two of moderate amplitude (displacements less than 100mm) and two of high amplitude (displacements greater than 100mm) ground motions. Additional comparisons of results are presented in Appendix B.

The superstructure was described in program 3PLEANI using the stiffness and damping matrices shown in Table 5-2 (case of seismic test identification) and the isolators using the friction coefficient values of Table 5-1. In the analysis, non-zero initial isolator displacements were used as the measured permanent isolator displacements at the conclusion of the preceding test. However, the isolator internal part offsets could not be accurately measured in the experiments since only one potentiometer was used for each part. Also, the internal bearing parts exhibited torsion (see Section 3), further complicating the extraction of data on the motion of the parts. Accordingly, analysis was used to approximately calculate the internal parts offsets and then use them as initial conditions, together with the experimentally measured permanent isolator displacement, for the analysis in the subsequent test. Note that the initial conditions so determined may contain errors and thus violate equilibrium and compatibility. To correct for this, analysis at the first integration step results in the calculation of the internal parts sliding displacements that satisfy equilibrium and compatibility.

The test results presented in Figure 5-16 to Figure 5-21 and in Appendix B do not include a vertical component of excitation. In all tests, however, there was parasitic vertical excitation, which was included in the analysis. In program 3pleANI, the vertical excitation is included by varying the axial load on each isolator as:

$$N = W(1 + \ddot{u}_{vg} / g) \tag{5-2}$$

In Equation (5-2), W is the starting (at time t=0) value of load on the isolator and \ddot{u}_{vg} is the history of vertical acceleration taken as positive if in the downward direction. W was obtained at the beginning of each test for each isolator using the procedure described in Section 2.2 (see also Figure 3-13). In all simulations, the vertical excitation was imported directly into the program as obtained from the average of the measurements of the four shake table vertical accelerometer recordings (ASSV, ASSV2 ANSV, and ANSV2 as shown in Figure 2-6) after filtering them using a low pass 30Hz filter.

Note that the approach in program 3pleANI to account for the vertical acceleration effects ignores the damping and flexibility of the structure in the vertical direction. An approach for accounting for flexibility and damping in program 3pleANI (but not used in the analyses presented herein) is:

- 1. Analyze a Single-Degree-of-Freedom (SDOF) system with damping and stiffness representing the structure in the vertical direction and subjected to ground excitation \ddot{u}_{yg} .
- 2. Use the calculated total acceleration response history of the SDOF as input \ddot{u}_{vg} in program 3pleANI.

Note that the procedure described above is similar to the procedure followed when analyzing a structure in program SAP2000. Accordingly, results that include the vertical excitation component (L+V tests) are presented only when analysis is performed in program SAP2000 in Section 5.4.

Figure 5-16 to Figure 5-21 demonstrate that the analytical model in program 3pleANI predicts well the experimental response in terms of frequency content of the response and shape of the loops but it occasionally over-predicts or under-predicts the experimental peak response. However, the predicted peak base displacement and peak base shear force are in very good agreement with the experimental peak values. The occasional over- or under-prediction of the peak structural response was also observed in the analysis of the structure without the isolation system (see Figure 5-13 to Figure 5-15). It is believed that this is due to inability of the analytical model of the superstructure to capture sliding and minor impact in the connections of the masses of the model to the floors and in the connections of the braces to beams and columns during strong shaking. Additional reasons for differences between analytical and experimental results are:

- 1. Uncertainty in the friction coefficient values, which certainly changed during testing due primarily to heating effects as the bearings were extensively tested without pausing to allow for return to ambient temperature conditions. Note also that the identification of friction coefficients was done under different heating conditions than those that existed in the shake table tests.
- 2. Effect of rubber seal in the tested reduced size bearings (see Figure 5-9).
- 3. Anisotropy in friction. Note that in some tests, the friction of surfaces 2 and 3 is direction dependent, with different values at positive displacements than at negative displacements. This is explained by the fact that the contact forces are applied away from the center of the sliding surface (Sarlis and Constantinou, 2013), resulting in uneven wear and variability in friction.
- 4. The accuracy of the analytical prediction deteriorates at small amplitude motions, due to inaccuracies in the friction-velocity relation at small velocities.
- 5. For the isolation system normalized force-displacement loops, the division by the instantaneous vertical load introduces error in the experimental results due to the addition of axial loads from four load cells, with the measurement of each one of these load cells containing some error.

In some tests, the analytically predicted floor response spectra are substantially higher than the experimental ones and in some others the analytically predicted floor response spectra are substantially lesser than the experimental ones. A notable example of the former case is test TSB 0637-270 (0.8/1/8) (see Figure 5-17 and Figure 5-20) where the analysis under-predicts the experimental floor spectra. In this case, there was impact on the restrainer rings which was not well captured in the analysis. Under conditions of impact with large restrainer stiffness and strength, small differences in the prediction of displacement result in large differences in force prediction, and thus floor response spectra as well.

Test TSC NWH-360 (1.2/1.5) (see Figure 5-18 and Figure 5-21) is an example of over-prediction of floor response spectral values by analysis. In this case, analysis predicted response in the stiffening isolator range, which did not occur. The result was over-prediction of acceleration response and floor spectral values.



Figure 5-16: Comparison of analytical (program 3pleANI) and experimental results for isolated structure with Configuration A isolators



Figure 5-16 (cont'd): Comparison of analytical (program 3pleANI) and experimental results for isolated structure with Configuration A isolators



Figure 5-16 (cont'd): Comparison of analytical (program 3pleANI) and experimental results for isolated structure with Configuration A isolators



Figure 5-17: Comparison of analytical (program 3pleANI) and experimental results for isolated structure with Configuration B isolators



Figure 5-17 (cont'd): Comparison of analytical (program 3pleANI) and experimental results for isolated structure with Configuration B isolators



Figure 5-17 (cont'd): Comparison of analytical (program 3pleANI) and experimental results for isolated structure with Configuration B isolators



Figure 5-18: Comparison of analytical (program 3pleANI) and experimental results for isolated structure with Configuration C isolators



Figure 5-18 (cont'd): Comparison of analytical (program 3pleANI) and experimental results for isolated structure with Configuration C isolators


Figure 5-18 (cont'd): Comparison of analytical (program 3pleANI) and experimental results for isolated structure with Configuration C isolators



Figure 5-19: Comparison of analytical (program 3pleANI) and experimental results for isolated structure with Configuration A isolators for drift and acceleration histories and 5%-damped floor response spectra



Figure 5-19 (cont'd): Comparison of analytical (program 3pleANI) and experimental results for isolated structure with Configuration A isolators for drift and acceleration histories and 5%-damped floor response spectra



Figure 5-19 (cont'd): Comparison of analytical (program 3pleANI) and experimental results for isolated structure with Configuration A isolators for drift and acceleration histories and 5%-damped floor response spectra



Figure 5-19 (cont'd): Comparison of analytical (program 3pleANI) and experimental results for isolated structure with Configuration A isolators for drift and acceleration histories and 5%-damped floor response spectra



Figure 5-19 (cont'd): Comparison of analytical (program 3pleANI) and experimental results for isolated structure with Configuration A isolators for drift and acceleration histories and 5%-damped floor response spectra



Figure 5-19 (cont'd): Comparison of analytical (program 3pleANI) and experimental results for isolated structure with Configuration A isolators for drift and acceleration histories and 5%-damped floor response spectra



Figure 5-20: Comparison of analytical (program 3pleANI) and experimental results for isolated structure with Configuration A isolators for drift and acceleration histories and 5%-damped floor response spectra



Figure 5-20 (cont'd): Comparison of analytical (program 3pleANI) and experimental results for isolated structure with Configuration B isolators for drift and acceleration histories and 5%-damped floor response spectra



Figure 5-20 (cont'd): Comparison of analytical (program 3pleANI) and experimental results for isolated structure with Configuration B isolators for drift and acceleration histories and 5%-damped floor response spectra



Figure 5-20 (cont'd): Comparison of analytical (program 3pleANI) and experimental results for isolated structure with Configuration B isolators for drift and acceleration histories and 5%-damped floor response spectra



Figure 5-20 (cont'd): Comparison of analytical (program 3pleANI) and experimental results for isolated structure with Configuration B isolators for drift and acceleration histories and 5%-damped floor response spectra



Figure 5-20 (cont'd): Comparison of analytical (program 3pleANI) and experimental results for isolated structure with Configuration B isolators for drift and acceleration histories and 5%-damped floor response spectra



Figure 5-21: Comparison of analytical (program 3pleANI) and experimental results for isolated structure with Configuration C isolators for drift and acceleration histories and 5%-damped floor response spectra



Figure 5-21 (cont'd): Comparison of analytical (program 3pleANI) and experimental results for isolated structure with Configuration C isolators for drift and acceleration histories and 5%-damped floor response spectra



Figure 5-21 (cont'd): Comparison of analytical (program 3pleANI) and experimental results for isolated structure with Configuration C isolators for drift and acceleration histories and 5%-damped floor response spectra



Figure 5-21 (cont'd): Comparison of analytical (program 3pleANI) and experimental results for isolated structure with Configuration C isolators for drift and acceleration histories and 5%-damped floor response spectra



Figure 5-21 (cont'd): Comparison of analytical (program 3pleANI) and experimental results for isolated structure with Configuration C isolators for drift and acceleration histories and 5%-damped floor response spectra



Figure 5-21 (cont'd): Comparison of analytical (program 3pleANI) and experimental results for isolated structure with Configuration C isolators for drift and acceleration histories and 5%-damped floor response spectra

5.5 Analytical Prediction Using Program SAP2000

The superstructure, fixed at its base, was modeled in program SAP2000 using linear elastic frame elements for all beams, columns and braces. The diaphragm bracing of the superstructure was explicitly modeled and no diaphragm constraints were assigned. The concrete blocks were modeled as lumped masses without mass moment of inertia. The self-weight of the frame was explicitly captured using the steel density value for the material in SAP2000. Additional small masses were added at the base-mat to capture the difference in the total weight calculated by the program and the one obtained from measurement by the load cells. This additional weight was contributed by the elements not accounted for in the model, such as steel connecting plates, stiffeners, bolts and connection angles. Due to the large dimensions of the base-mat beams compared to the superstructure elements, rigid beam elements have been used to connect the bottom of the columns to the centerline of the W360 beams of the base-mat. Rigid offsets have not been used for the beam-to-column connections of the structure. Table 5-3 presents results for the modal properties of the model, fixed at the base, as obtained by program SAP2000 for the first three modes. The damping ratio is the value assigned for each mode in SAP2000 for the construction of the inherent damping matrix. Note that the assigned damping ratio values are between the values identified in the two sets of experiments and presented in Table 4-1 and Table 4-2. There is good agreement between the mode shapes and period values obtained in the experimental identification (Table 4-1 and Table 4-2) and the results of the modal analysis in SAP2000.

Analysis of the fixed-base superstructure with seismic motion at its base was conducted and results are compared to experimental results in Figure 5-22 to Figure 5-24 in terms of histories of inter-story drift and floor acceleration, and 5%-damped floor acceleration spectra. Results are in good agreement but, as in the case of analysis with program 3pleANI, the peak values of response are occasionally over-estimated or under-estimated by the analytical model.

Mode	Period (sec)	Assigned	Mode Shape				
		Damping Ratio	1 st floor	2 nd floor	3 rd floor		
1^{st}	0.292	0.0650	0.331	0.741	1.000		
2^{nd}	0.092	0.0100	1.176	0.808	-1.000		
3 rd	0.053	0.0078	2.286	-2.397	1.000		

Table 5-3: Modal characteristics of analytical model in SAP2000



Figure 5-22: Comparison of analytical (program SAP2000) and experimental results for inner-story drift of fixed structure for ground motion ATL-270



Figure 5-23: Comparison of analytical (program SAP2000) and experimental results for floor acceleration of fixed structure for ground motion ATL-270



Figure 5-24: Comparison of analytical (program SAP2000) and experimental results for 5%-damped floor response spectra of fixed structure for ground motion ATL-270

In SAP2000, the Triple FP isolators were modeled using the series model described in Fenz and Constantinou (2008d and e). The series model consists of three friction pendulum elements arranged in series and denoted as FP1, FP2 and FP3. Gap elements are connected between the top and bottom joints of the FP2 element and the top and bottom joints of the FP3 element. The properties specified in program SAP2000 for the series model representation of the isolators are shown in Table 5-4. For more details on how these properties are selected, the reader is referred to Fenz and Constantinou (2008d and e) and Sarlis and Constantinou (2010). Note that the effective stiffness of element FP1 is assigned a small value so that "damping leakage" is minimized (Sarlis and Constantinou, 2010). Also, the effective stiffness of elements FP2 and FP3 is assigned a large value in order to reduce the execution time. Given that elements FP1, FP2 and FP3 are arranged in series, the high effective stiffness of elements FP2 and FP3 does not affect the total effective stiffness of the assembly (so that damping leakage is minimized). Also the vertical stiffness of the elements is selected such that the dominant mode of the structure in the vertical direction has the same frequency as the one measured in the experiments (as obtained from transfer functions of the base of the structure in the vertical direction from records of accelerations and for tests that included a vertical component).

SAP2000 element ID	FP1			FP2			FP3					
Configuration	Α	B	С	Α	B	С	Α	В	С			
Element Height (mm)		25.4			38.1			38.1				
Shear Deformation Location from bottom (mm)	0		0		0							
Element Mass (kN-s ² /mm)	0.00002			0.1		0.1						
Yield Displacement (mm)	0.1		0.1		0.1							
Vertical Stiffness (kN/mm)	235		235		235							
Rotational/Torsional Stiffness (R1,R2,R3)	0		Fixed		Fixed							
Rotational Moment of Inertia (kN-mm-sec ²)	0.00113		0		0							
	r		Isol	ator 1	r	[[1	1			
Effect. Stiffness (kN/mm)		0.0175	1	130	150	120	130	310	120			
Elastic Stiffness (kN/mm)	30	60	528	130	150	120	240	310	120			
Radius (mm)	106	116	106	382	382	382	382	382	382			
Friction Slow	0.02	0.04	0.264	0.055	0.09	0.07	0.11	0.16	0.07			
Friction Fast	0.06	0.07	0.28	0.12	0.165	0.135	0.14	0.21	0.135			
Rate Parameter (sec/mm)	0.03	0.015	0.01	0.102	0.046	0.0175	0.0239	0.052	0.0175			
Isolator 2									1			
Effect. Stiffness (kN/ mm)		0.0175	I	130	126	120	130	208	120			
Elastic Stiffness(kN/ mm)	40	20	510	154	126	120	420	208	120			
Radius (mm)	106	106	106	382	382	382	382	382	382			
Friction Slow	0.02	0.01	0.255	0.067	0.063	0.06	0.20	0.124	0.06			
Friction Fast	0.055	0.02	0.27	0.12	0.12	0.15	0.24	0.225	0.15			
Rate Parameter (sec/mm)	0.03	0.015	0.01	0.102	0.035	0.0175	0.0239	0.081	0.0175			
Isolator 3												
Effect. Stiffness (kN/mm)		0.0175	I	130	70	140	130	190	140			
Elastic Stiffness (kN/mm)	40	30	430	136	70	140	400	190	140			
Radius (mm)	106	116	106	382	382	382	382	382	382			
Friction Slow	0.03	0.015	0.215	0.048	0.035	0.07	0.19	0.095	0.07			
Friction Fast	0.065	0.04	0.23	0.095	0.085	0.135	0.24	0.105	0.135			
Rate Parameter (sec/mm)	0.03	0.015	0.01	0.102	0.029	0.02	0.0239	0.1036	0.02			
	1		Isol	ator 4	1	1		1	1			
Effect. Stiffness (kN/mm)		0.0175	I	130	94	140	130	250	140			
Elastic Stiffness (kN/mm)	50	20	390	130	94	140	330	250	140			
Radius (mm)	106	116	106	382	382	382	382	382	382			
Friction Slow	0.02	0.01	0.195	0.075	0.047	0.07	0.145	0.125	0.07			
Friction Fast	0.04	0.04	0.21	0.14	0.14	0.135	0.205	0.20	0.135			
Rate Parameter (sec/mm)	0.03	0.015	0.01	0.102	0.029	0.0175	0.0239	0.1036	0.0175			

Table 5-4: Series model properties of Triple FP isolators in program SAP2000

Comparisons of analytical results obtained in SAP2000 to experimental results are presented in Figure 5-25 to Figure 5-30 for tests without vertical component of excitation and in Figure 5-25 and 5-32 for tests with a vertical component of excitation. Six ground motions for each configuration are shown: two for small amplitudes (isolator displacement<50mm), two for moderate amplitude (isolator displacement<100mm) and two for high amplitudes (isolator displacement>100mm). The ground motions presented are the same as those presented in Section 5.3 for the analysis with program 3pleANI. Additional results are presented in Appendix C. The effect of the initial base displacement in the SAP2000 analysis was included by creating an additional analysis case where a force was applied at the base and then removed. The force value was such that a permanent displacement was achieved in the analytical model equal to the one measured in the experiments (the process required a trial and error approach in order to find the force vector).

An immediate observation in the results presented in Figure 5-25 to Figure 5-30 is that the fidelity of the analytical prediction by SAP2000 is similar to that of program 3pleANI (presented in Section 5.3), although the program 3pleANI results appear slightly better than those of SAP2000 likely due to (a) better modeling of the velocity dependence of the friction coefficient (it is approximate in the series model of the Triple FP), and (b) more accurate consideration of the non-zero initial conditions.

The comparison of analytical and experimental results in Figure 5-31 and Figure 5-32 with combined horizontal and vertical excitation is less favorable that in Figure 5-25 to Figure 5-30 without the vertical excitation. An important contributor to this problem is the effect of the specified vertical stiffness of the isolators in the analysis. Incorrect specification of the vertical stiffness, combined with vertical excitation, often results in numerical problems and incorrect or premature prediction of isolator uplift and/or incorrect fluctuation of vertical load on the isolators. These difficulties are evident in the results of Figure 5-31 and Figure 5-32.



Figure 5-25: Comparison of analytical (program SAP2000) and experimental results for isolated structure with Configuration A isolators



Figure 5-25 (cont'd): Comparison of analytical (program SAP2000) and experimental results for isolated structure with Configuration A isolators



Figure 5-25 (cont'd): Comparison of analytical (program SAP2000) and experimental results for isolated structure with Configuration A isolators



Figure 5-26: Comparison of analytical (program SAP2000) and experimental results for isolated structure with Configuration B isolators



Figure 5-26 (cont'd): Comparison of analytical (program SAP2000) and experimental results for isolated structure with Configuration B isolators



Figure 5-26 (cont'd): Comparison of analytical (program SAP2000) and experimental results for isolated structure with Configuration B isolators



Figure 5-27: Comparison of analytical (program SAP2000) and experimental results for isolated structure with Configuration C isolators



Figure 5-27 (cont'd): Comparison of analytical (program SAP2000) and experimental results for isolated structure with Configuration C isolators



Figure 5-27 (cont'd): Comparison of analytical (program SAP2000) and experimental results for isolated structure with Configuration C isolators



Figure 5-28: Comparison of analytical (program SAP2000) and experimental results for isolated structure with Configuration A isolators for drift and acceleration histories and 5%-damped floor response spectra



Figure 5-28 (cont'd): Comparison of analytical (program SAP2000) and experimental results for isolated structure with Configuration A isolators for drift and acceleration histories and 5%-damped floor response spectra


Figure 5-28 (cont'd): Comparison of analytical (program SAP2000) and experimental results for isolated structure with Configuration A isolators for drift and acceleration histories and 5%-damped floor response spectra



Figure 5-28 (cont'd): Comparison of analytical (program SAP2000) and experimental results for isolated structure with Configuration A isolators for drift and acceleration histories and 5%-damped floor response spectra



Figure 5-28 (cont'd): Comparison of analytical (program SAP2000) and experimental results for isolated structure with Configuration A isolators for drift and acceleration histories and 5%-damped floor response spectra



Figure 5-28 (cont'd): Comparison of analytical (program SAP2000) and experimental results for isolated structure with Configuration A isolators for drift and acceleration histories and 5%-damped floor response spectra



Figure 5-29: Comparison of analytical (program SAP2000) and experimental results for isolated structure with Configuration B isolators for drift and acceleration histories and 5%-damped floor response spectra



Figure 5-29 (cont'd): Comparison of analytical (program SAP2000) and experimental results for isolated structure with Configuration B isolators for drift and acceleration histories and 5%-damped floor response spectra



Figure 5-29 (cont'd): Comparison of analytical (program SAP2000) and experimental results for isolated structure with Configuration B isolators for drift and acceleration histories and 5%-damped floor response spectra



Figure 5-29 (cont'd): Comparison of analytical (program SAP2000) and experimental results for isolated structure with Configuration B isolators for drift and acceleration histories and 5%-damped floor response spectra



Figure 5-29 (cont'd): Comparison of analytical (program SAP2000) and experimental results for isolated structure with Configuration B isolators for drift and acceleration histories and 5%-damped floor response spectra



Figure 5-29 (cont'd): Comparison of analytical (program SAP2000) and experimental results for isolated structure with Configuration B isolators for drift and acceleration histories and 5%-damped floor response spectra



Figure 5-30: Comparison of analytical (program SAP2000) and experimental results for isolated structure with Configuration C isolators for drift and acceleration histories and 5%-damped floor response spectra



Figure 5-30 (cont'd): Comparison of analytical (program SAP2000) and experimental results for isolated structure with Configuration C isolators for drift and acceleration histories and 5%-damped floor response spectra



Figure 5-30 (cont'd): Comparison of analytical (program SAP2000) and experimental results for isolated structure with Configuration C isolators for drift and acceleration histories and 5%-damped floor response spectra



Figure 5-30 (cont'd): Comparison of analytical (program SAP2000) and experimental results for isolated structure with Configuration C isolators for drift and acceleration histories and 5%-damped floor response spectra



Figure 5-30 (cont'd): Comparison of analytical (program SAP2000) and experimental results for isolated structure with Configuration C isolators for drift and acceleration histories and 5%-damped floor response spectra



Figure 5-30 (cont'd): Comparison of analytical (program SAP2000) and experimental results for isolated structure with Configuration C isolators for drift and acceleration histories and 5%-damped floor response spectra



Figure 5-31: Comparison of analytical (program SAP2000) and experimental results for isolated structure in combined horizontal and vertical excitation



Figure 5-31 (cont'd): Comparison of analytical (program SAP2000) and experimental results for isolated structure in combined horizontal and vertical excitation



Figure 5-32: Comparison of analytical (program SAP2000) and experimental results of histories of drift and acceleration and 5%-damped floor spectra for isolated structure in combined horizontal and vertical excitation



Figure 5-32 (cont'd): Comparison of analytical (program SAP2000) and experimental results of histories of drift and acceleration and 5%-damped floor spectra for isolated structure in combined horizontal and vertical excitation



Figure 5-32 (cont'd): Comparison of analytical (program SAP2000) and experimental results of histories of drift and acceleration and 5%-damped floor spectra for isolated structure in combined horizontal and vertical excitation

SECTION 6 CONCLUSIONS

The results of a testing program of an isolated three-story structure supported by Triple FP isolators of three different configurations have been reported. The isolator configurations included two highly adaptive ones that exhibited all five regimes of operation of the isolator. The third configuration lacked adaptability and resembled in behavior the single FP. Testing consisted of horizontal excitation and combined horizontal and vertical excitation in a variety of time and acceleration scales so that the isolators operated in all five regimes of operation, and in some tests experienced uplift and impact on their restrainer rings. In general, the conditions of testing may be characterized as extreme.

Also, the response of the tested structure was predicted by numerical simulation in the commercial program SAP2000 and in the newly developed more advanced program 3pleANI and compared to the experimental results. It was concluded that:

- 1. The isolators exhibited stable behavior under the extreme conditions of testing. Certain aspects of the measured response were, however, unrealistic due to the very high stiffness and strength of the restrainer rings of the model isolators by comparison to those of full size isolators.
- 2. The vertical component of excitation had no or insignificant effect on the isolator displacement demand, had some minor effect on structural drifts and had an important effect on the floor accelerations. The effect of the vertical acceleration was enhanced in the testing by large parasitic rocking motion of the shake table.
- 3. The response of the structure could be predicted accurately in terms of isolator displacement, base shear, drift and acceleration histories, although the peak values (particularly of acceleration) were occasionally under-predicted or over-predicted. It is believed that this was primarily due to incomplete modeling of the superstructure.
- 4. Programs 3pleANI and SAP2000 provided comparable prediction of response but program 3pleANI has slightly better predictions due to better description of the velocity dependence of friction at each sliding interface, and more accurate consideration of the non-zero initial conditions.
- 5. Prediction of the response under combined horizontal and vertical excitation was less accurate that when only horizontal excitation was considered. The difference is likely the result of inaccurate modeling of the vertical stiffness of the isolators that may result in numerical errors and affect the prediction of the history of the vertical load on the isolator, and may predict incorrectly or prematurely isolator uplift.

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Appendix A Experimental Data






















































































































































































Appendix B 3pleANI2







































































































































Appendix C

SAP 2000

































































































































































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