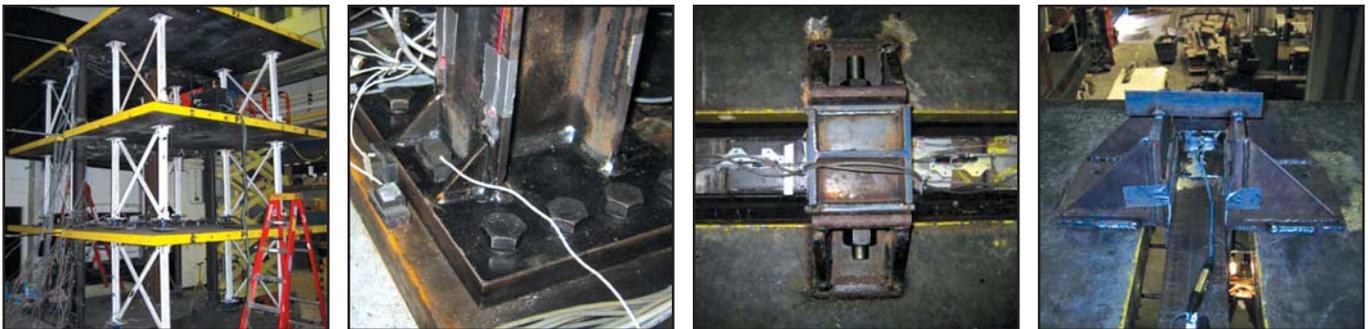


# Numerical and Experimental Studies of Self-Centering Post-Tensioned Steel Frames

by  
Dong Wang and Andre Filiatrault



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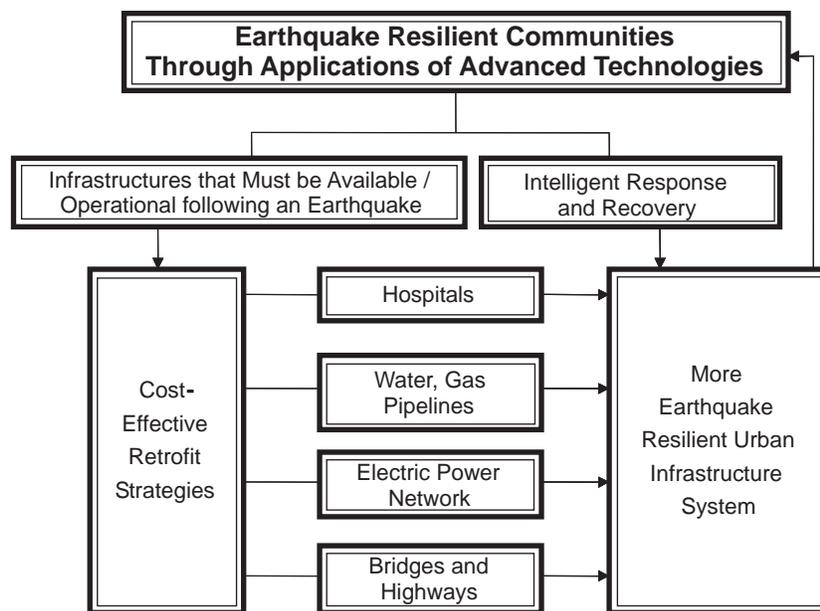
## Preface

The Multidisciplinary Center for Earthquake Engineering Research (MCEER) is a national center of excellence in advanced technology applications that is dedicated to the reduction of earthquake losses nationwide. Headquartered at the University at Buffalo, State University of New York, the Center was originally established by the National Science Foundation in 1986, as the National Center for Earthquake Engineering Research (NCEER).

Comprising a consortium of researchers from numerous disciplines and institutions throughout the United States, the Center's mission is to reduce earthquake losses through research and the application of advanced technologies that improve engineering, pre-earthquake planning and post-earthquake recovery strategies. Toward this end, the Center coordinates a nationwide program of multidisciplinary team research, education and outreach activities.

MCEER's research is conducted under the sponsorship of two major federal agencies: the National Science Foundation (NSF) and the Federal Highway Administration (FHWA), and the State of New York. Significant support is derived from the Federal Emergency Management Agency (FEMA), other state governments, academic institutions, foreign governments and private industry.

MCEER's NSF-sponsored research objectives are twofold: to increase resilience by developing seismic evaluation and rehabilitation strategies for the post-disaster facilities and systems (hospitals, electrical and water lifelines, and bridges and highways) that society expects to be operational following an earthquake; and to further enhance resilience by developing improved emergency management capabilities to ensure an effective response and recovery following the earthquake (see the figure below).



A cross-program activity focuses on the establishment of an effective experimental and analytical network to facilitate the exchange of information between researchers located in various institutions across the country. These are complemented by, and integrated with, other MCEER activities in education, outreach, technology transfer, and industry partnerships.

*This report presents a numerical and experimental study to evaluate and compare the seismic performance of Self-Centering Post-Tensioned (SCPT) steel frames and conventional Steel Moment Resisting Frames (SMRF). A methodology for designing SCPT structures is developed and used to retrofit an existing four-story SMRF medical facility located in Northridge, California. A Relative Performance Index (RPI) is proposed to compare the enhancement in the seismic response of SCPT's to the SMRF's. Numerical simulations and shake table tests were carried out on two scaled 3-story 2-bay SCPT and SMRF building prototypes based on the medical facility. The study indicates that the seismic floor displacements of SCPT and SMRF are alike, while absolute floor accelerations are lower when selfcentering systems are implemented. Moreover, while yielding in the beam-to-column connections of the SMRF are observed, only yielding of the sacrificial devices in the self centering system of the SCPT occurred during the strongest ground motions considered.*

## ABSTRACT

In this research, numerical and experimental studies were conducted to evaluate the seismic performance of Self-Centering Post-Tensioned (SCPT) steel frames and Steel Moment Resisting Frames (SMRF).

Numerical analyses of Single-Degree-of-Freedom (SDOF) and Multiple-Degree-of-Freedom (MDOF) Self-Centering Systems (SCS) and Elasto-Plastic Systems (EPS) were conducted under simulated strong ground motions. A Relative Performance Index (RPI) was developed as an evaluation criterion of the seismic performance of both systems. A design procedure for SCPT frames was also developed. The results obtained from the numerical analyses indicated better seismic performances of the SCS/SCPT systems when compared to the EPS/SMRF systems. The MCEER Demonstration Hospital building was used as a prototype of a SMRF building in the numerical studies.

An experimental study was carried out on the five degrees-of-freedom shake table at the University at Buffalo. Two 3-story, 2-bay, steel plane frame models incorporating SCPT connections and conventional fully welded moment-resisting connections were used in this test. The SCPT frame, unlike traditional welded steel frames, incorporates high strength post-tensioned strands along with sacrificial yielding elements in each beam-to-column connection and is particularly appealing for hospital buildings from an initial investment stand point. These two test models were subjected to various ground motions of increasing intensities. The results of the tests indicated that the displacement response of the SCPT frame was very similar to the fully-welded SMRF and the acceleration response was reduced. While the beams of the SMRF yielded under the largest seismic excitations, the energy dissipation mechanism of the SCPT frame was limited to the Energy Dissipating (ED) bars without inelastic deformations of the beams and columns. The experimental results suggest that only the ED bars would need to be replaced in the SCPT frame as compared to the costly replacement of the damaged beams or columns sections in the SMRF building after a significant earthquake.



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# **CHAPTER I**

## **INTRODUCTION**

In the history of human kind, the construction technology has always been one of the most important and most practical technologies because any social, natural or engineering science cannot be developed without a safe and comfortable housing environment. Several thousands years ago, our ancestors used the natural construction materials for their houses. At that time, these primitive types of construction could only supply basic housing needs. The most obvious evidence of the evolution of the human civilization is the increased sophistication associated with housing structures. Before the 20<sup>th</sup> century, wood and stones were widely used for the construction of mostly low-rise buildings not exceeding 3 to 4 stories. Nevertheless, these simple buildings did supply an adequate environment for the progress of technologies but were not constructed to resist some natural hazards like floods or earthquakes. In the last century, the engineering community has developed optimized construction materials, such as concrete and steel, which allowed the construction of much larger and higher buildings. But still due to the lack of knowledge about the effects of earthquakes on structures, these early engineered buildings could only survive small seismic events. This lack of proper earthquake-resistant design in buildings remained a significant design flaw until approximately fifty years ago. In the last thirty years, the development of computers and mechanics as well as the observations of the actual behavior of structures following large earthquakes in urban areas has allowed a much better understanding of the effects of earthquakes on structures. This improved understanding has been embodied by more stringent seismic design requirements in building codes. Engineers could then begin to

explicitly implement seismic requirements and detailing at the design stage. As a result, the seismic performance of newly constructed buildings was increased gradually. These traditional seismic design requirements that are still included in current building codes, concentrate mainly on maintaining the structural integrity of the structural system during a major seismic event in order to prevent loss of lives. Therefore, according to this life-safety design requirement approach, buildings may still be damaged beyond repair and may need to be demolished or rebuilt.

Because of the increased needs of space by expanding urban areas, an increase in mid- to high-rise buildings has been observed in the last thirty years. Steel Moment Resisting Frame (SMRF) constructions are most widely used in mid- and high-rise buildings in North America. Because of their widespread use, the seismic behavior of SMRF buildings has become one of the most important concerns in the mind of designers and owners. In this chapter the seismic behavior of SMRF and the applicability of the concept of Self-Centering Systems (SCS) to SMRF are discussed and the research objectives and organizations of this report are also presented.

## **1.1 Seismic Behavior of Steel Moment Resisting Frames (SMRF)**

Prior to the 1994 Northridge earthquake in California, it was believed that SMRF represented one of the most adequate forms of earthquake-resistant construction, as evidenced by very few failures or severe damaged observed in SMRFs and steel structures in general after earthquakes. In fact, prior to 1994, the only severe damage reported in steel

structures, occurred in the Pio Suarez building in the 1985 Mexico City Earthquake (SAC 1995). Because of this apparent good seismic behavior of steel structures in general and SMRFs in particular, designers and owners believed that this type of building was one of the most seismic-resisting structures.

The other reason for the good reputation that SMRFs enjoyed is that SMRFs were considered to be a ductile framing system due to the ductile property along with the strength and reliability of structural steel. These apparent good material properties motivated engineers to push the envelope by designing larger and taller buildings. Indeed, many practicing engineers believed for years, albeit incorrectly, that steel structures were immune to earthquake-induced damage as a consequence of the material's inherent ductile properties (Bruneau et al. 1998). Welded SMRF connections were mostly applied in SMRF structures. Over the years, two major types of welded SMRF connections were developed. The first and earliest (pre-1970) type of connections involved welding the whole beam section to the column, as shown in Fig. 1-1. The other and more recent type of connections incorporate a shear tab welded to the flange of the column and bolted to the web of beams, while the flange of the beam is welded to the column with a full penetration grooved weld held in place by a backing bar, as shown in Fig. 1-2. Several experimental investigations on these two types of connections conducted by several investigators (Popov & Pinkney, 1969; Popov & Bertero, 1973; Popov et al. 1975) have concluded that both types of connections appear to possess a very good ductile behavior. It should be noted, however, that rare unacceptable behaviors like sudden fractures at the weld-to-column interface at the beam

bottom flange were also reported (Engelhardt & Husain, 1993). Based on these early experimental results, conducted mostly on small scale connection specimens, the earthquake engineering community considered SMRFs as excellent ductile lateral load-resisting systems, which lead to an increased numbers of SMRFs buildings, particularly in California.

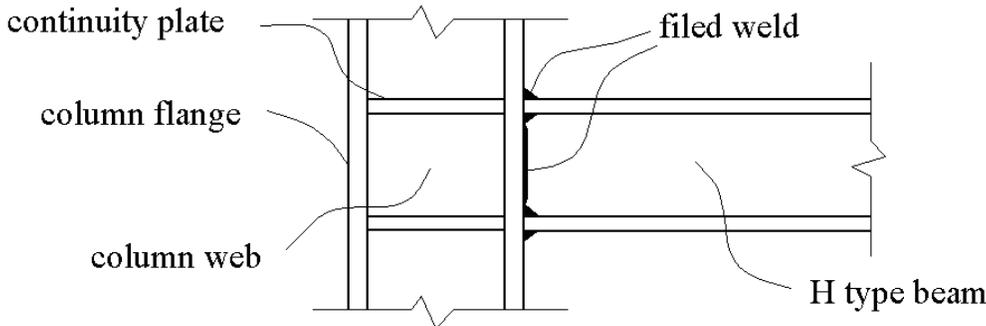


Figure 1-1 Earlier (pre-1970) Type of Welded SMRF Connection

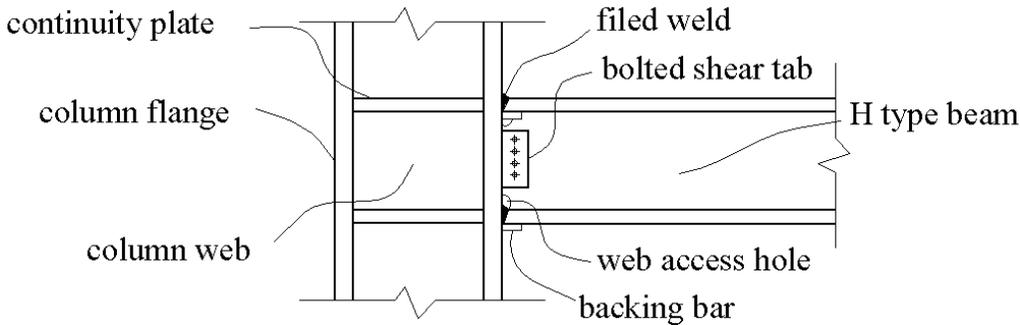


Figure 1-2 Recent (post-1970) Type of Welded SMRF Connection

However, the January 17<sup>th</sup>, 1994 Northridge Earthquake in Los Angeles, completely changed the perspective of the earthquake engineering community towards SMRFs. Over 100 SMRF structures experienced brittle beam-to-column connection fractures as a result of this Richter magnitude 6.8 seismic event (SAC 1995). The height of damaged structures covered a wide

range from one to 26 stories. The age of damaged buildings spanned from as old as 30 years to new. A typical connection damage observed in a SMRF after the Northridge earthquake is shown in Fig. 1-3, in which a crack was observed in the column flange connected to the weld of the beam bottom flange and propagated to the web of the column.

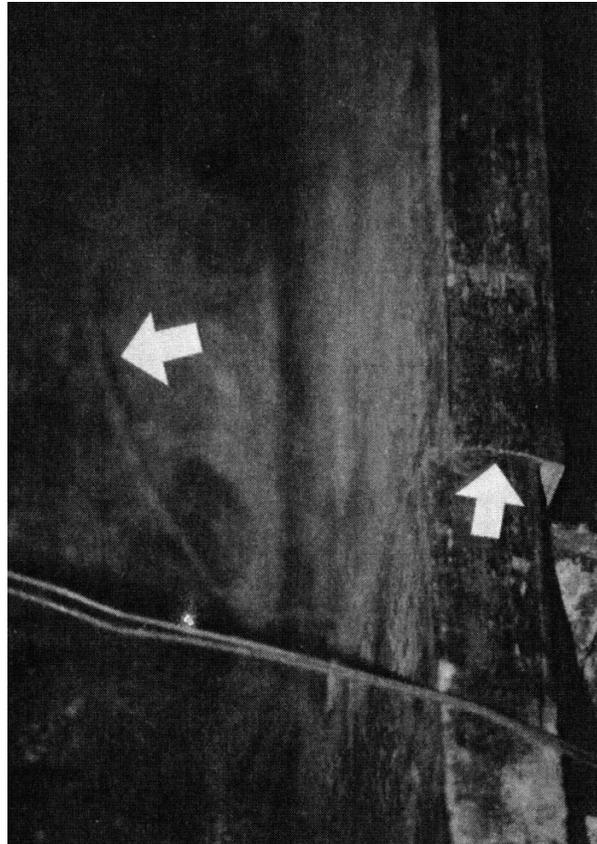


Figure 1-3 Typical Connection Damage in the Northridge Earthquake (Bruneau et al. 1998)

Figure 1-4 illustrates eight types of fractures observed in beam-to-column connections during the Northridge earthquake. Fractures and cracks were found to propagate around the fused zone in the flange and web of columns and beams. Causes of failures are complicated and include many aspects: inspection quality, weld design, fracture mechanics, base metal

elevated yield stress, welds stress condition, stress concentrations, effect of triaxial stress conditions, loading rate and presence of composite floor slab. More details on the cause of these observed fractures can be found in Bruneau et al. (1998).

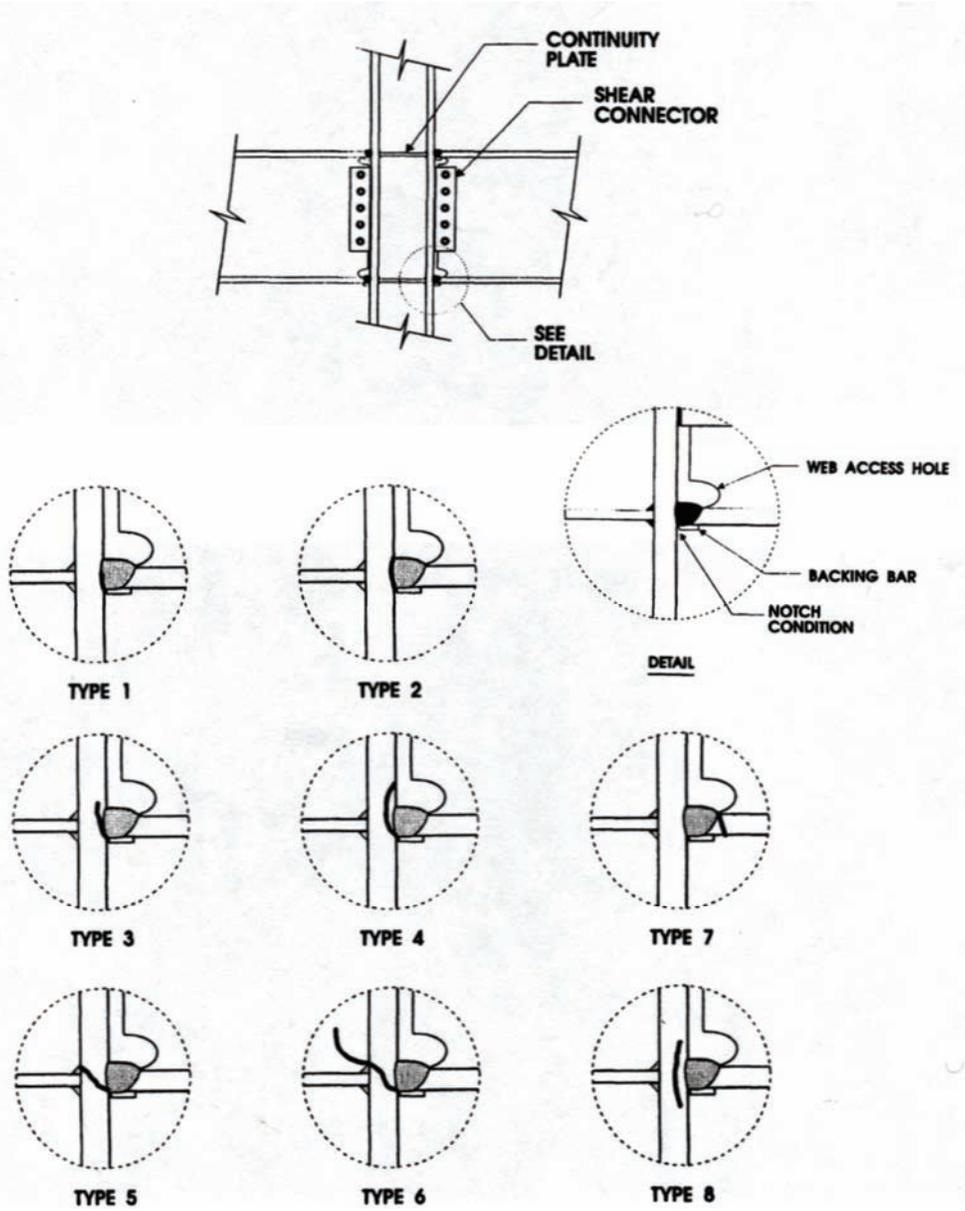


Figure 1-4 Eight Types of Fractures in Beam-Column Connections Identified as a Result of the Northridge Earthquake (Tremblay et al. 1995)

After the 1994 Northridge earthquake, a significant research effort was conducted by the SAC joint venture, which is a partnership of Structural Engineers Association of California (SEAOC), the Applied Technology Council (ATC) and the California Universities for Research in Earthquake Engineering (CUREe). Another parallel research was initiated by the Federal Emergency Management Agency (FEMA), which published several instructional documents: FEMA-350, FEMA-351, FEMA-352, FEMA-353, FEMA-354, FEMA-355 and FEMA-356. Those documents provided new design procedures and guidelines to insure a good seismic performance of SMRFs. As a result of this research effort, it is believed that the seismic performance of SMRF buildings has been increased to an acceptable level compared to that observed in the Northridge earthquake.

Although brittle fractures in beam-to-column connections were widespread in SMRF buildings during the Northridge earthquake, no SMRF structures collapsed and it can be assumed that SMRFs reached the life safety performance level. However, severe economic loss resulted. During the 1994 Northridge Earthquake, with the loss of approximately \$18.5 billion due to building damage, nonstructural damage accounted for about 50% of this total (Kircher, 2003). From an economic point of view, not only the seismic properties of connections (inelastic response, yielding sections of beams, plastic rotations etc.) in earthquakes should be considered but also the seismic response (displacement, velocity and acceleration response) of buildings, local buckling, residual deformation and corresponding cost for recovering functions of post-earthquake buildings should be given more

considerations. Further discussions and potential solutions on these aspects will be presented in the following chapters.

## 1.2 Concepts of Elasto-Plastic Systems (EPS) and Self-Centering Systems (SCS)

Conventional buildings designed according to current seismic design requirements should have capacities for ductile inelastic response and for dissipating energy during earthquake shaking. Such structural systems are referred to as Elasto-Plastic Systems (EPS). Fig. 1-5 illustrates the idealized inelastic response of EPS during earthquake excitations.

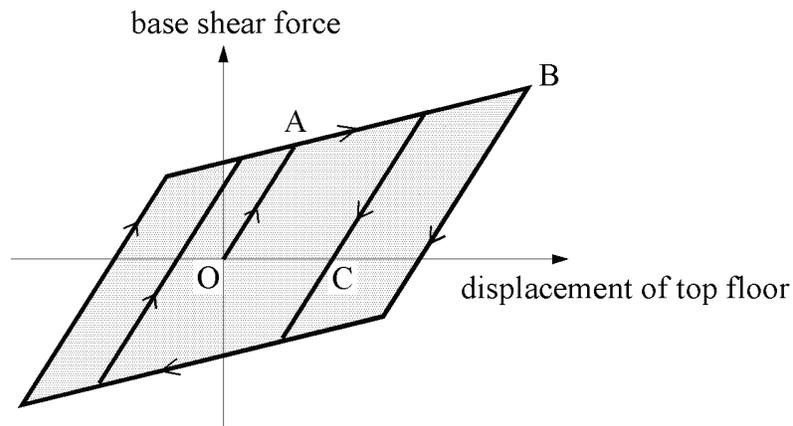


Figure 1-5 Ideal Inelastic Seismic Response of Elasto-Plastic Systems (EPS)

The portion OA in the hysteretic loop in Fig. 1-5 represents the elastic response of EPS. The structure yields at point A and reaches its maximum displacement at point B. The shaded area illustrates the energy dissipated through hysteretic yielding of the main structural elements: mainly the end of beams. A larger shaded area represents more energy dissipated,

an increase of the effective damping in the structural system and an increased probability that the EPS structure survives in the earthquakes. However, the hysteretic energy dissipated by EPS is also directly associated with structural damage to the structure and with potential large permanent inelastic deformations. The residual drift of the structure can be relatively large due to the large plastic deformation, as illustrated by point C in Fig. 1-5. Most SMRFs can be modeled as EPS systems. In SMRFs, the sections at the end of the beams develop plastic rotations or local buckling, which can lead to significant repair to the structure and associated costs to recover its normal operations. If a severe residual drift occurred, the cost of repairing the structure may be more than that of building a new one.

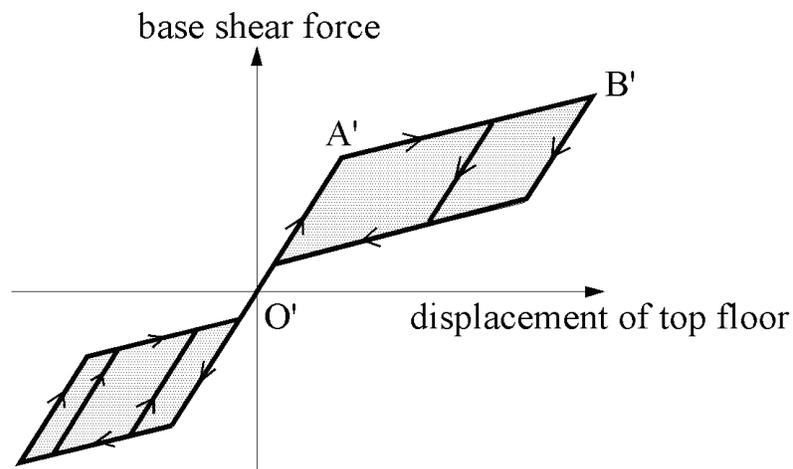


Figure 1-6 Ideal Seismic Response of Self-Centering Systems (SCS)

The idealized seismic response of a Self-Centering System (SCS) is shown in Fig. 1-6. There are two major differences between the seismic response of the EPS shown in Fig. 1-5 and that of the SCS shown in Fig. 1-6. First, the hysteretic loops are different which indicates that the energy dissipated in the SCS is less than that of the EPS. The reduced energy absorbed by the SCS “flows” into special devices rather than being dissipated hysteretically

by the main structural elements. Therefore, the damage to the structure is diminished or eliminated. Second, in Fig. 1-6, there is no point in the zero force axis except the original point, which means that the residual drift of the SCS is zero. The zero residual drift in the SCS can eliminate the cost to “return” the structure to its original position. After comparing the different properties, it can be argued that the SCS exhibits much better hysteretic properties than EPS, although special energy absorbing devices can be more costly in the initial construction. Self-centering properties in structural systems can be achieved using special dampers, control materials (like shape memory alloy) or special connections. More details on practical means of achieving self-centering properties are presented in Chapter II of this report. When compared to the cost of repairing a structure and recovering its operations, it is apparent that potentially the SCS can achieve a higher seismic performance at a reduced cost.

### **1.3 Research Objectives and Organizations**

The objectives of this research are to numerically and experimentally investigate the properties and seismic performance of Self-Centering Systems (SCS) and apply the Self-Centering Post-Tensioned (SCPT) connections to the design of SMRF buildings.

Chapter I introduces the seismic performance of SMRF, the concepts of Elasto-Plastic Systems (EPS) and Self-Centering Systems, research objectives, and organizations.

The previous research on Self-Centering Systems is presented in Chapter II.

An analysis tool for Single-Degree-of-Freedom (SDOF) SCS excited by strong ground motions is developed in Chapter III. A full parametric dynamic analysis of SDOF EPS and SCS is conducted and the seismic performance is compared.

In Chapter IV, the redesign of the Multidisciplinary Center for Earthquake Engineering Research (MCEER) Demonstration Hospital incorporating SCPT connections is presented. Different numerical models of the structure are developed. Push-over analysis, seismic analysis and fragility analysis are also conducted.

Chapter V presents the design and scaling of the prototype structure and the design of the SCPT model used in the shake table testing.

In Chapter VI, the experimental procedure for the SMRF and SCPT models are presented. The selection of the ground motion used as the seismic excitation and the instrumentations are also discussed.

The experimental results are presented and compared in Chapter VII.

Chapter VIII describes the comparison of the results obtained from the initial numerical models, calibrated numerical models and experimental results.

The conclusions of both the numerical and experimental studies and suggestions for future work are presented in Chapter IX.

Through this report the SCS/SCPT systems are proved to have a better seismic performance and a reduced cost of repair after earthquakes than the most popular EPS/SMRF systems.

## CHAPTER 2

### PREVIOUS RESEARCH AND APPLICATIONS ON SELF-CENTERING SYSTEMS

Self-Centering Systems (SCS) have been studied analytically and experimentally mostly in the last fifteen years. As introduced in Chapter I, these systems, involving post-tensioned and energy dissipating elements or other special devices with self-centering property, also exhibit nonlinear softening behavior, ductility and energy dissipation. Re-centering forces are provided by post-tensioned systems or other special devices to return the structure to its original position and eliminate or diminish residual deformations.

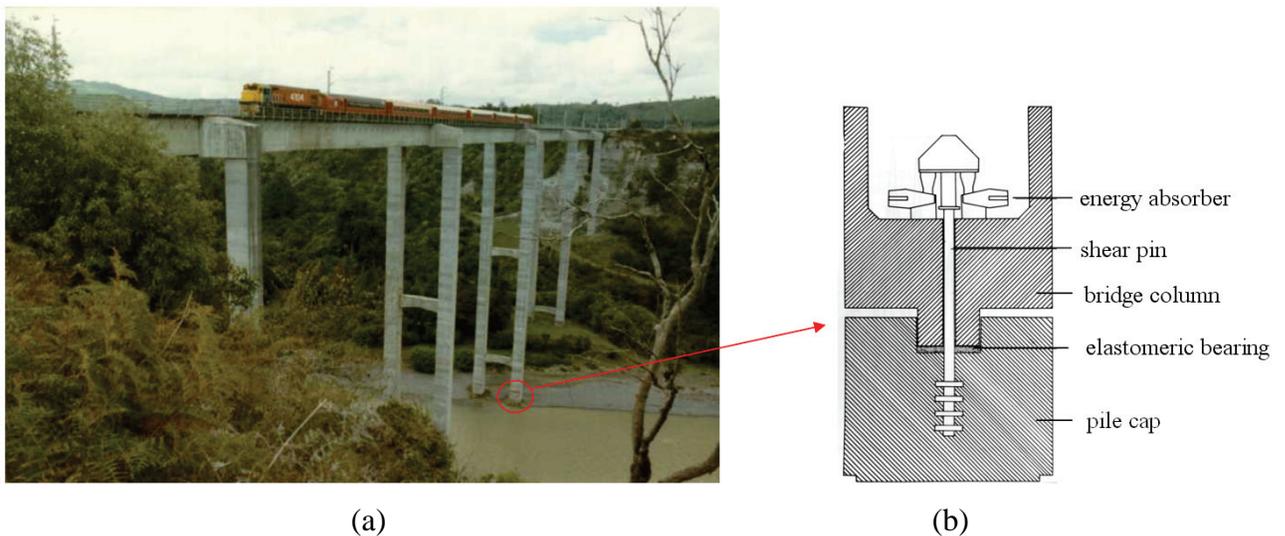


Figure 2-1 Rangitikei Railway Bridge (a) Overall View (b) Rocking Pad (Cormack 1988)

The first practical application of SCS was the design of the “stepping” Rangitikei rail bridge in New Zealand in 1981. As shown in Fig. 2-1(a), this 70 m (230 ft) tall 315 m (1033 ft) long bridge was constructed with six spans of prestressed concrete hollow-box girders

(Cormack 1988). The self-centering property is exhibited by the rocking foundation system as shown in Fig. 2-1(b). The weight of the bridge allows the re-centering of the structure when there is a transverse displacement. Therefore, this bridge behaves as an inverted pendulum with its own weight providing the self-centering effect.

After this first application of SCS, more research efforts were conducted. To achieve the re-centering capability, two major methods were developed: self-centering dampers and special self-centering connections. These two methods are discussed in this chapter in order to present previous and recent research and applications on SCS.

## **2.1 Self-Centering Dampers**

Structures designed with conventional dampers, such as viscous dampers, friction dampers or viscoelastic dampers, do not prevent residual drifts to occur after earthquakes. To eliminate or decrease these residual drifts, dampers can be improved to incorporate re-centering capabilities. These dampers with SCS properties are referred herein as “Self-Centering Dampers”.

### **2.1.1 Shape Memory Alloys (SMA) Dampers**

Shape Memory Alloys (SMA) dampers are metallic dampers incorporating a special class of alloys, which produce the so-called super elastic behavior as shown in Fig. 2-2.

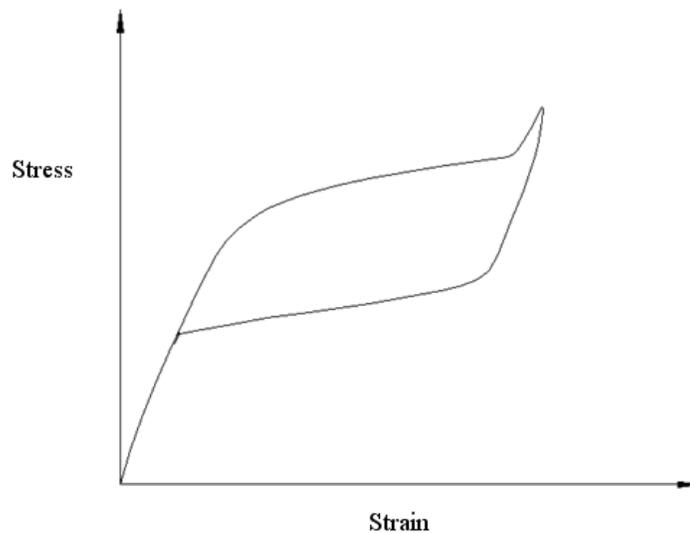


Figure 2-2 Super Elastic Hysteresis

There are five primary SMA systems that have been developed: the Nickel-Titanium family (often called Nitinol); two copper-based systems, Cu-Zn-Al and Cu-Al-Ni; Fe-Mn-Si alloys; and some special stainless steel compositions (Whittaker et al. 1995). The stress-strain relationship of SMAs depends on the alloying temperature. A high alloying temperature produces a fully austenite phase, which exhibits a linear elastic stress-strain relationship. A fully martensite phase occurs in a low alloying temperature and exhibits a viscoelastic stress-strain relationship. An intermediate alloying temperature produces a hybrid phase involving both phases, which displays the super elastic behavior similar to the SCS. Aiken et al. (1992) conducted shake table tests on a small-scale three-story steel frame using SMA (Nitinol) dampers, which incorporated Nitinol wires at the end of the cross-bracing systems. Due to the large pre-load in the braces, the Nitinol wires were cycled in the range of martensite phase and lost the self-centering property. Another SMA (Cu-Zn-Al) damper test in a 2/5- scale five-story frame was conducted by Witting and Cozzarelli (1992). This test

result indicated that the SMA (Cu-Zn-Al) damper had a super elastic stress-strain relation for a very limited number of cycles and after a few cycles the damper performed an elastoplastic hysteretic behavior due to the increased internal friction.

Another experimental work of SMA in structures was conducted by Ocel et al. (2004). In this investigation, the SMA was used in two full scale steel beam-column connections as shown in Fig. 2-3. Results from cyclic loading and dynamic loading indicated that the connections with SMA exhibited a high level of energy dissipation, large ductility and no strength degradation. The recovery of residual shape by heating the SMA demonstrated the self-centering capability in this system.

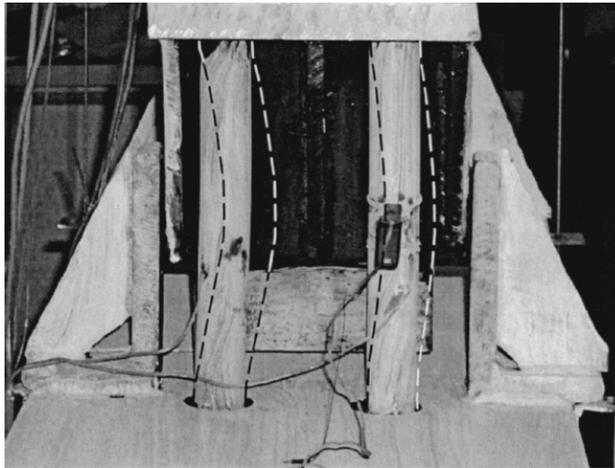


Figure 2-3 Recovered-deformation of Connections with SMA after Heating: dashed lines represent residual shape before heating (Ocel et al. 2004)

### **2.1.2 Energy Dissipating Restraint (EDR)**

The Energy Dissipating Restraint (EDR) was originally designed as a seismic restraint device for the support of piping systems in nuclear power plants. The EDR is a damper

device, indicated in Fig. 2-4, which is based on a friction mechanism. In this device, as a result of the combination of wedges, internal stops and springs, a frictional force is produced and determined proportional to the relative displacement of two device ends. More details of the mechanism in this device can be found in Nims (1993).

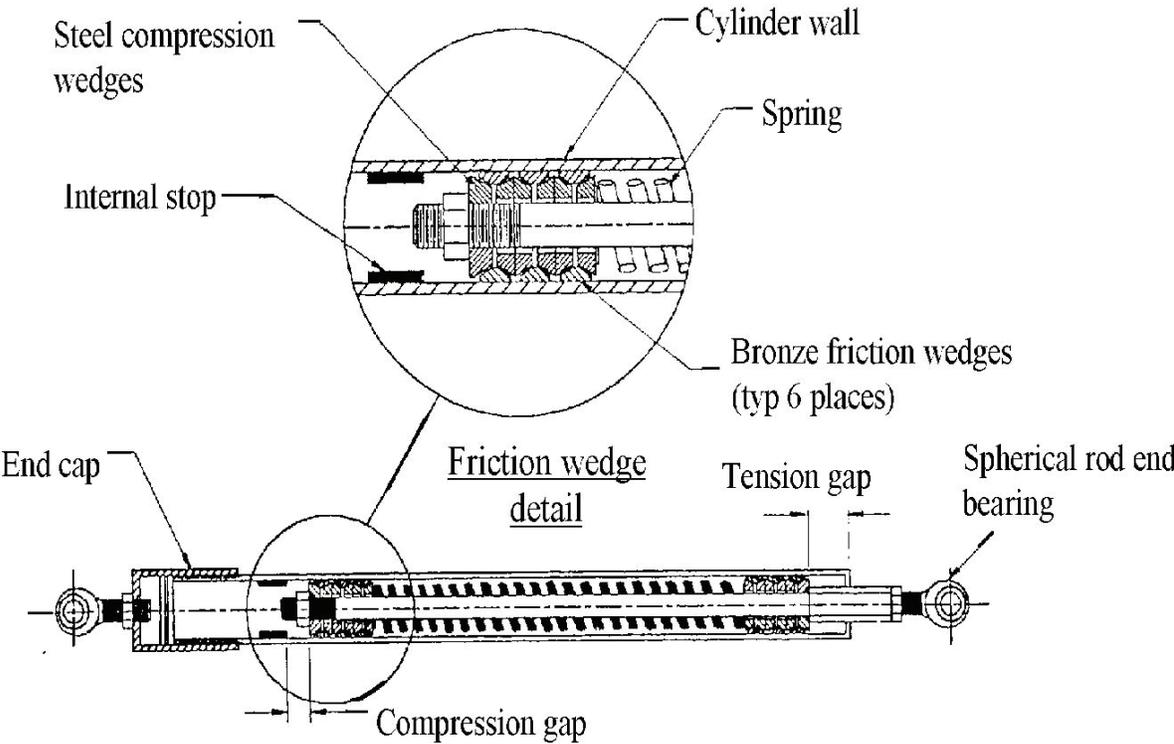


Figure 2-4 Internal View of Energy Dissipating Restraint (Nims et al. 1993)

The EDR can provide various self-centering hysteretic responses as shown in Fig. 2-5. An experimental shake table study of a small-scale 3-story EDR frame was conducted by Aiken et al. (1993), as shown in Fig. 2-6. The results of this test indicated that the effect of the EDR reduced the model deformations and interstory drifts.

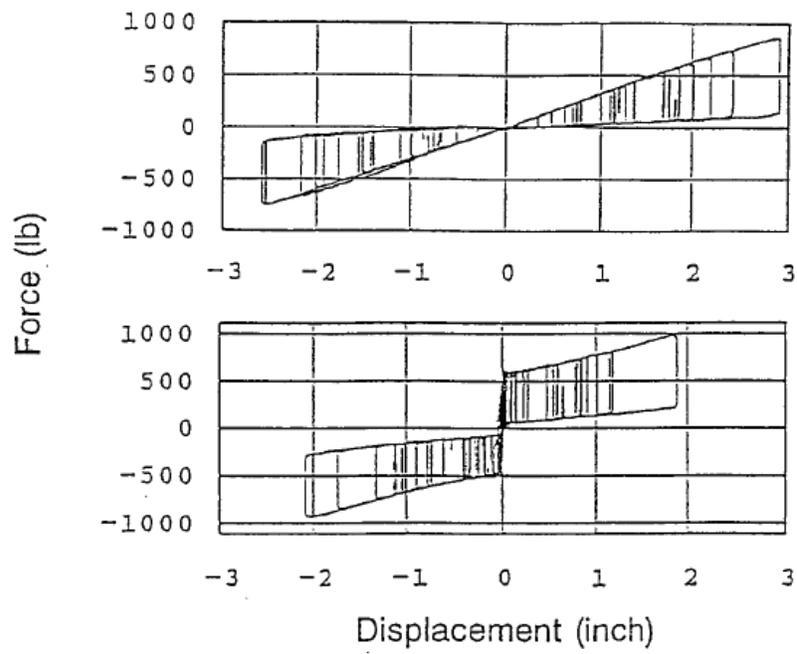


Figure 2-5 Various Self-Centering Hysteresis Loops Provided by the EDR Device (Aiken et al. 1993)

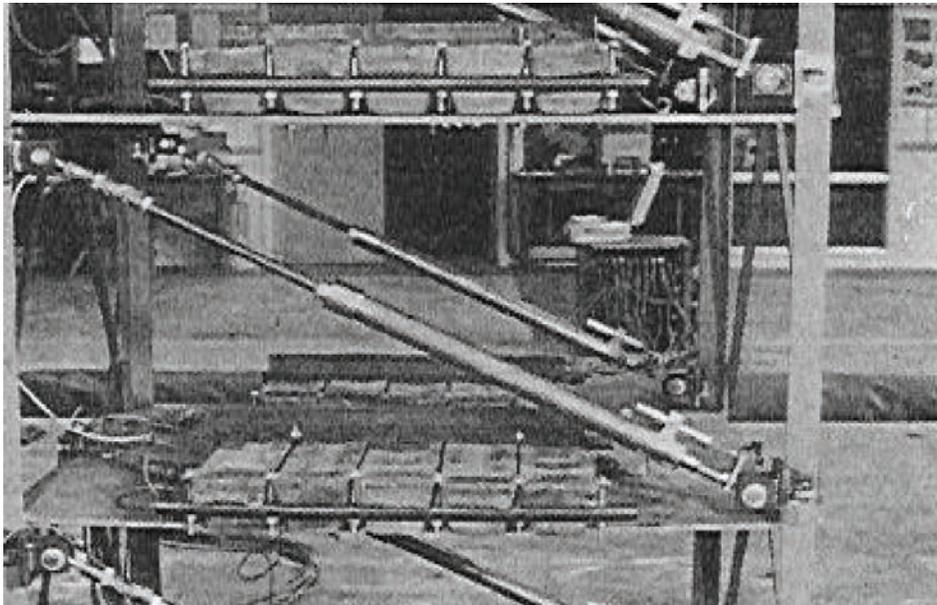


Figure 2-6 View of EDRs Installed at the Second Level of Test Frame (Aiken et al. 1993)

### 2.1.3 SHAPIA Damper

The SHAPIA damper is another damping device that exhibits the self-centering behavior. This damper uses a friction-based ring spring as the key component for the energy dissipation as shown in Fig. 2-7. Under compression load, the friction springs are compressed by the left and right cup while the tie bar moves toward the right. Under the tension load, the right cup is pulled by the tie bar toward to the left while the left cup is hold by the right sliding sleeve so that the friction springs are also compressed. This simple and effective mechanism makes the friction springs compressing and dissipating energy under both compression and tension external load.

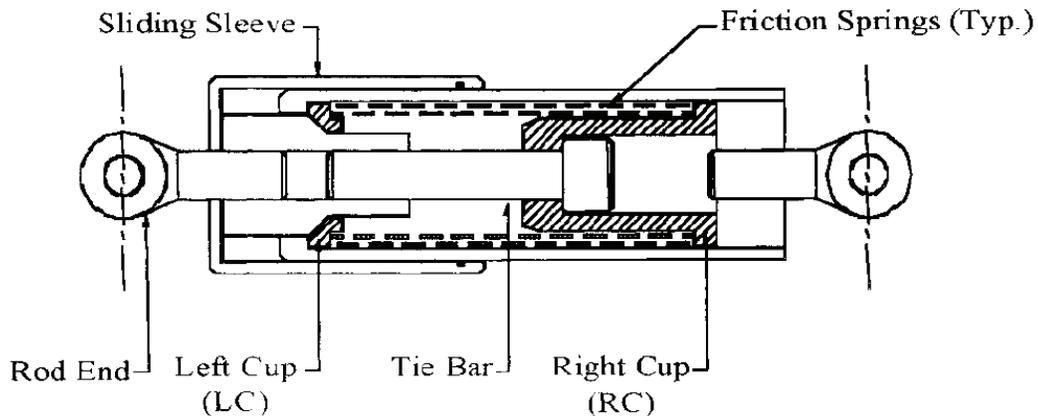


Figure 2-7 Diagrammatic View of the SHAPIA Damper (Filiatrault et al. 2000)

A seismic test on a single-story steel frame incorporating a SHAPIA damper was conducted by Filiatrault et al. (2000). The test demonstrated that the SHAPIA damper reduced the lateral displacement of the test structure through its self-centering characteristics as shown in Fig. 2-8.

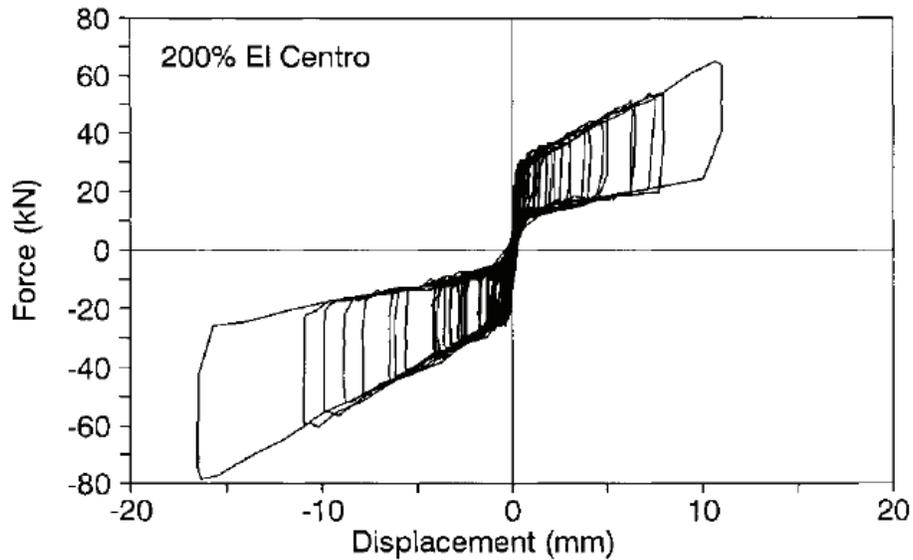


Figure 2-8 Force-Displacement Hysteresis loop of SHAPIA Damper Recorded during the Seismic Test (Filiatrault et al. 2000)

## 2.2 Self-Centering Connections

Self-Centering Systems (SCS) can also be achieved by special self-centering connections. Most of these connections use post-tensioned bars and tendons to provide the re-centering capability along with metallic devices, usually mild steel, to absorb the energy incurred by the excitation of earthquakes. The advantages of the self-centering connections are not only the re-centering capability but also the lower cost compared to the self-centering dampers. Depending on their orientation, self-centering connections in structures can be classified into two types: horizontal connections (beam to column) and vertical connections (wall to foundation or column to foundation).

### 2.2.1 Rocking Self-Centering Connections

Rocking self-centering connections can be implemented in bridge piers and in unbonded post-tensioned precast concrete walls in buildings. As introduced in the beginning of this chapter, the first application of these connections was the south Rangitikei Railway Bridge in New Zealand. Since then, several experimental investigations have been conducted in the United States using a similar concept. Mander and Cheng (1997) investigated the performance of unbonded post-tensioning in bridge piers under seismic loading.

Fig. 2-9(a) shows an unbonded post-tensioned precast concrete wall. The post-tensioned steel is anchored at the top of the wall and the foundation. A gap between the wall and the foundation develops due to the lateral deformation or the overturning moment. When unloading, the post-tensioned steel and the weight of the wall return the wall to its original position, thereby exhibiting a self-centering property as shown in Fig. 2-9(b).

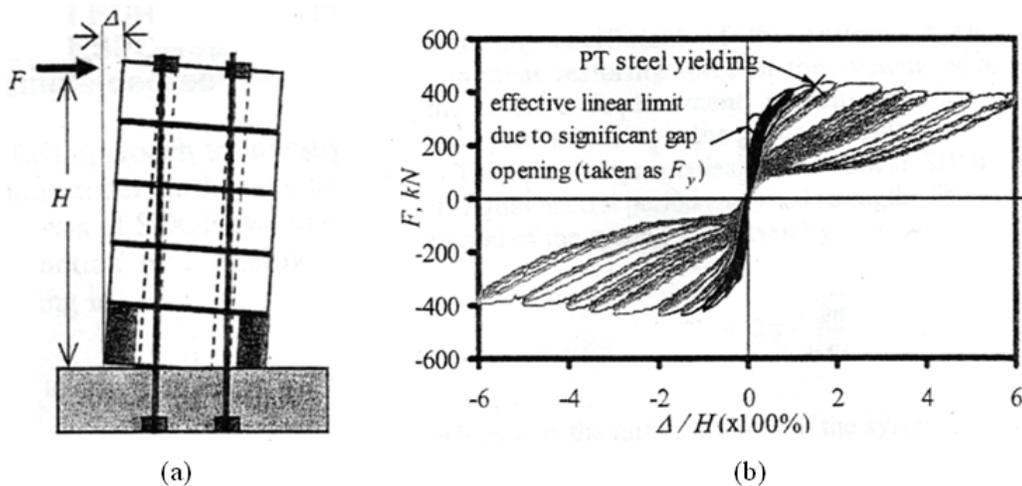


Figure 2-9 Post-Tensioned Precast Concrete Wall (a) Deformed Configuration of Unbonded PT Precast Concrete Wall (b) Experimental Cyclic lateral Load-Drift Behavior of Unbonded PT Precast Concrete Wall (Perez 2004)

Kurama et al. (1996, 1997, 1999 and 2004) and Shen et al. (2002) analytically investigated the seismic behavior and design of similar unbonded post-tensioned precast concrete walls. As a result of the analysis, unbonded post-tensioned precast walls had larger displacements than comparable monolithic cast-in-place reinforced concrete walls and had significantly smaller maximum residual displacements than the monolithic walls, which verified the self-centering capability of the walls. Stanton and Nakaki (2002) developed a post-tensioned split rocking wall system, in which wall panels are split to allow rocking of the individual panels. The re-centering force is provided mainly by post-tensioned tendons. Restrepo (2002) further developed these walls by adding conventional tapered mild steel bars between the foundation and the wall as energy dissipating devices. The advantage of the added steel is to absorb more energy and delay the yielding of post-tensioned tendons, which provide the large displacement and re-centering capability observed in this shake table tests. Figure 2-10 shows the experimental test of a rocking wall conducted by Holden et al. (2003). Results of conventional walls and rocking walls from this quasi-static test were compared and indicated the benefit of this hybrid wall system. Another rocking shear wall system was developed by Ajrab et al. (2004), who conducted also a parametric study. This analytical research verified that rocking wall structures with supplemental damping systems provided viable alternatives to conventional fixed-base ones.

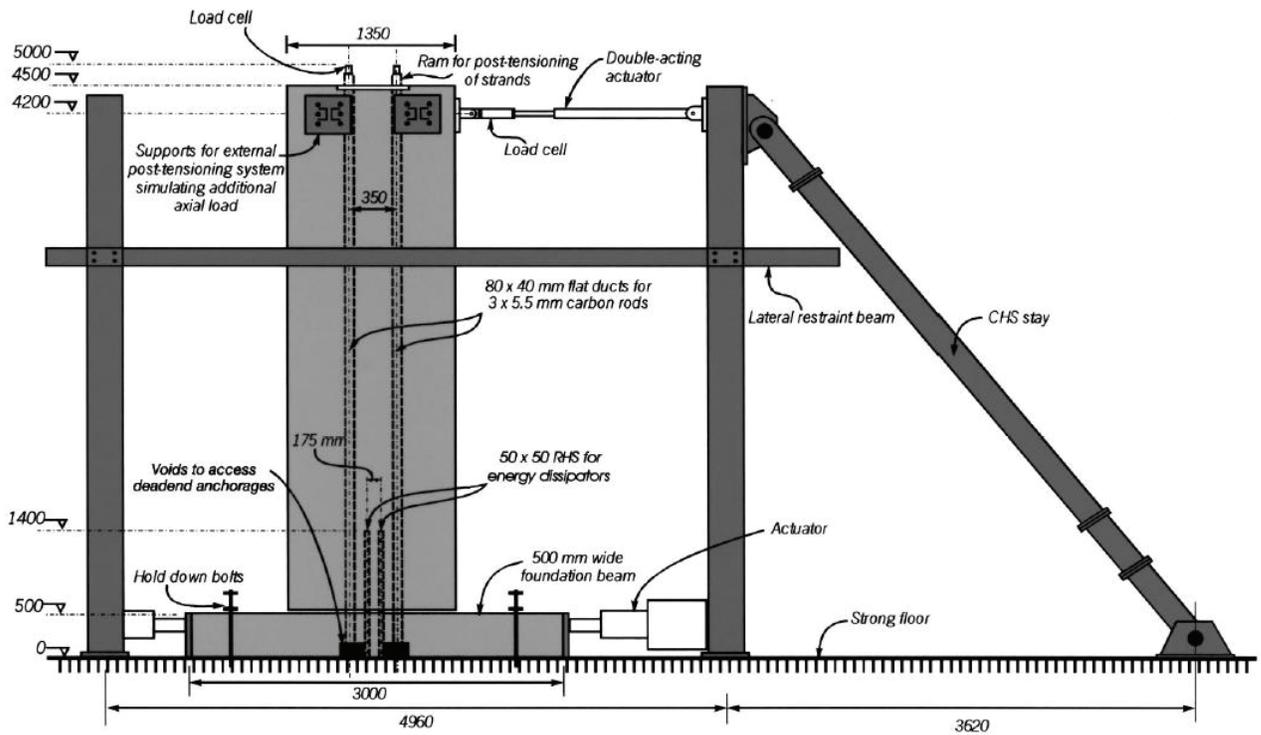


Figure 2-10 Loading Frame and Rocking Wall (Holden et al. 2003)

### 2.2.2 Self-Centering Connections in Concrete Structures

Priestley and Tao (1993) first proposed the use of partially unbonded post-tensioned tendons through beam-column joints as a design for precast concrete ductile connections. Figure 2-11 indicates the shear transfer mechanism in a typical partially unbonded post-tensioned beam-to-column connections. The horizontal shear force is transferred by a diagonal compression strut and the vertical shear force is transferred by the contact friction between the beam and column in the joint. The clamping force, provided by the unbonded post-tensioned tendons, produces a self-centering characteristic.



used for the test using several different earthquakes records. Under the excitation of the scaled earthquake records, which is equivalent to 1.5 UBC Zone 4 loading, the behavior of the structure was extremely satisfactory. The self-centering connections of the structure led to a significant reduction of residual drifts. There was only minimal damage observed in the shear wall direction. Among the different connections used in the structure, the hybrid connections with energy dissipating bars exhibited the best performance including the capacities for large deformations and re-centering characteristics. A detailed report of the performance of self-centering connections and the analysis procedure used in this test is given by Pampanin et al. (2000).

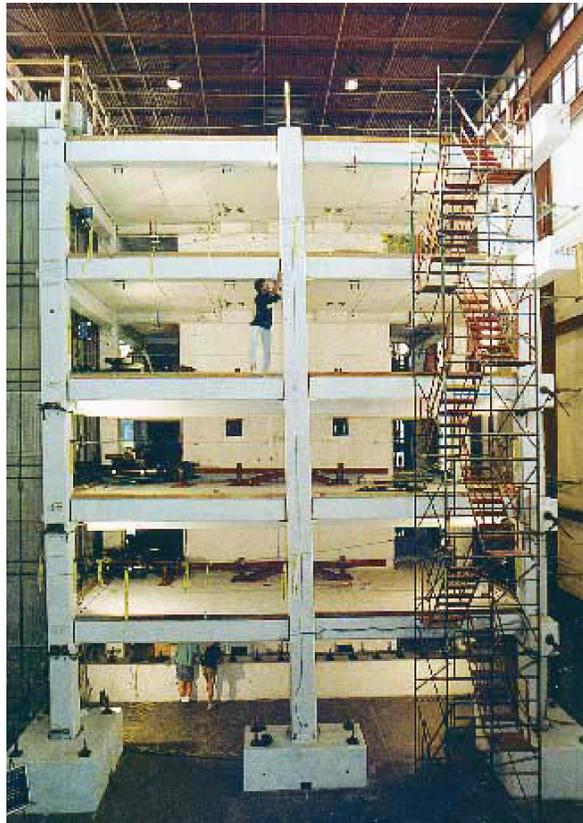


Figure 2-12 Overall View of Five-Story Building under Test (Priestley et al. 1999)

### **2.2.3 Self-Centering Connections in Steel Structures**

Self-centering connections in steel structures have been studied mostly in the last five years. A Post-Tensioned (PT) connection was developed by Ricles et al. (2001, 2002). Another one was the Post-Tensioned Energy Dissipating (PTED) connection proposed by Christopoulos et al. (2002a, 2002b). An analytical research of another type of self-centering connections, the Post-tensioned Friction Damped Connection (PFDC), was conducted by Rojas et al. (2005). In this section, these three types of connections are discussed.

#### **2.2.3.1 Post-Tensioned (PT) Connections**

As shown in Fig. 2-13(a), the Post-Tensioned (PT) beam-column steel connection proposed by Ricles et al. (2001) uses post-tensioned strands, distributed along the depth of the beam to provide the self-centering capability, and bolted angles between the top/bottom flange of the beams and columns to provide the energy dissipation. Shear forces are transferred by the friction at the beam-column interface and bolted angles. One of the advantages in this configuration is no requirement for field welding.

A large-scale test program was conducted by Ricles et al. (2002). Fig. 2-13(b) illustrates the test setup. Loading was applied by displacing the top of the column through a series of increasing symmetric displacement cycles. As shown in Fig. 2-13(c), the test results indicated the several advantages of this PT connection: (1) the similar initial stiffness to conventional welded steel connections; (2) the self-centering capability; and (3) the energy dissipation by the bolted angles without evident damage to the beam and column. Another

full-scale PT connection was tested using inelastic cyclic loading by Garlock et al. (2005) and once again it demonstrated the benefits of this type of connection. A performance-based seismic design approach for this PT frame was outlined and evaluated by Garlock et al. (2007).

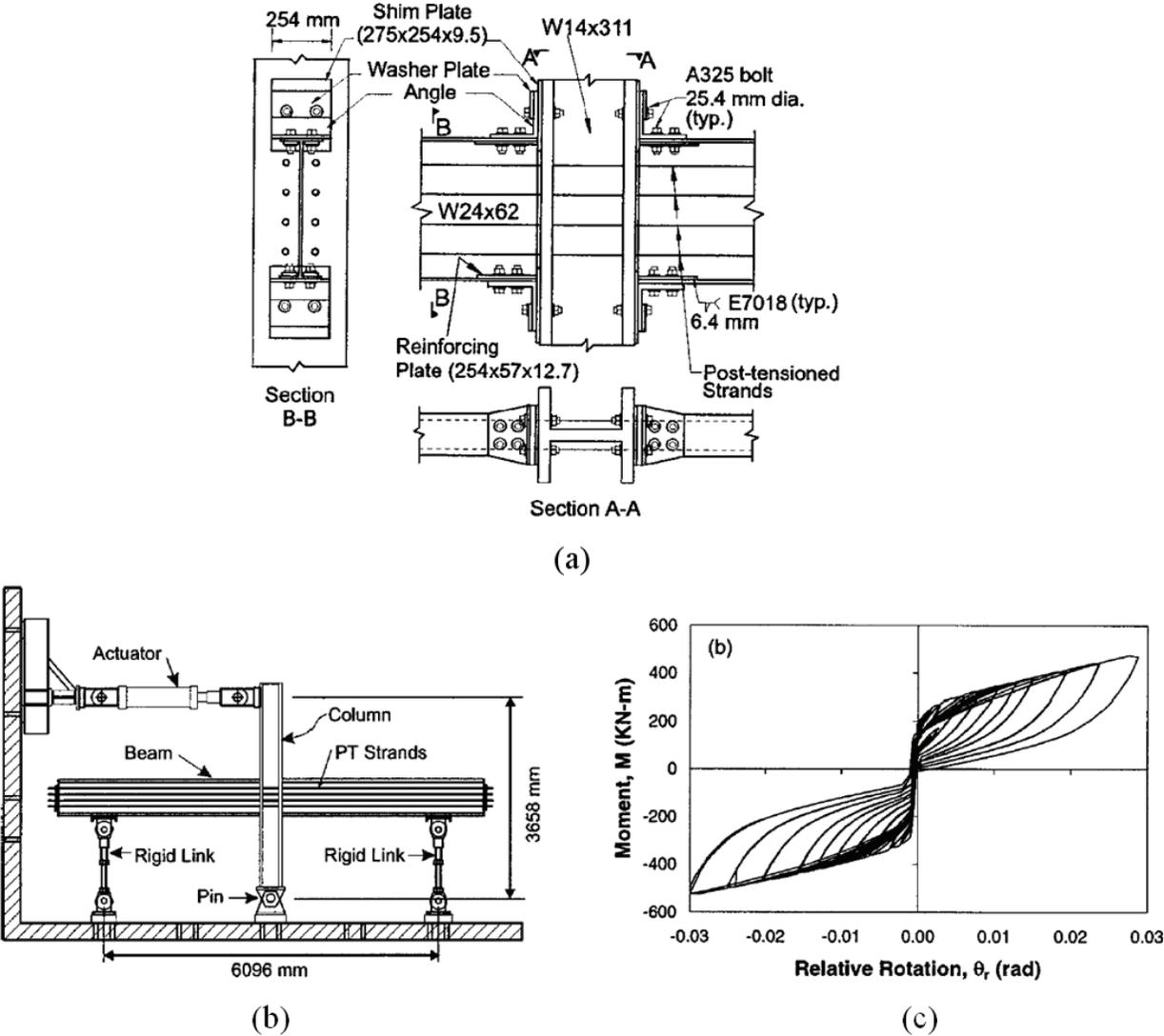


Figure 2-13 Post-Tensioned Connections Test (a) Post-Tensioned Specimen Connection (b) Post-Tensioned Connection Subassembly Setup (c) Moment-Rotation Response of a PT Connection (Ricles, et al. 2002)

### 2.2.3.2 Post-Tensioned Energy Dissipating (PTED) Connections

Another self-centering steel connection: the Post-Tensioned Energy Dissipating (PTED) connection was proposed by Christopoulos et al. (2002a, b). As shown in Fig. 2-14, the PTED connection incorporates two symmetric high strength Post-Tensioned (PT) tendons or bars at the mid-depth of the beam as re-centering components and four symmetric

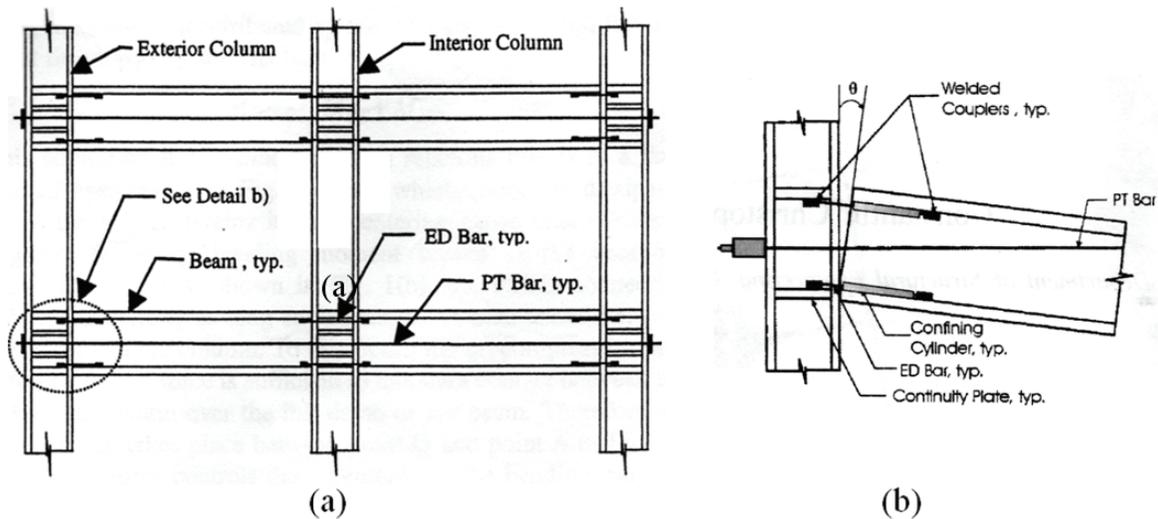


Figure 2-14 Concept of PTED connections (a) Steel Frame with PTED Connections  
(b) Geometric Configuration of Exterior PTED Connections (Christopoulos et al, 2002a)

Energy Dissipating (ED) bars per connection for dissipating energy. The PT bars are anchored at the flange of the exterior columns of the lateral load-resisting frame and the ED bars are welded to the inside face of the beam flanges and the continuity plates of the exterior columns. To prevent the buckling of the ED bars in compression, confining steel sleeves welded to the flanges of beams or the continuity plates of columns confine the ED bars. As shown in Fig. 2-15(a), cyclic testing of a PTED beam-to-column connection was conducted by using inelastic cyclic loading at the mid-length of the beam. Fig. 2-15(b)

indicates that the PTED connection exhibited very good self-centering characteristics and eliminated the residual drift without any damage in the beam and column. The experimental results were in good conformity to the predictions provided by a numerical study developed by the same authors.

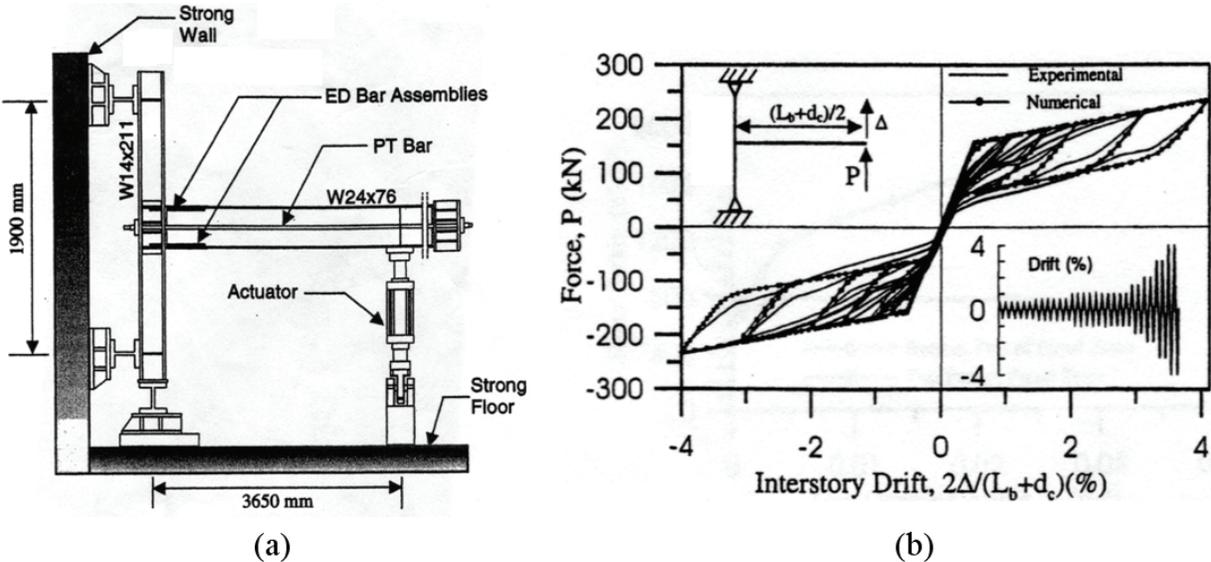


Figure 2-15 Cyclic Testing of a PTED connection (a) Experimental Setup (b) Experimental and Numerical Force-Interstory Drift Response (Christopoulos et al., 2002b)

Another experimental study was also conducted by Collins and Filiatrault (2003). The test model was a half-scale steel moment-resisting frame involving PTED connections along with a concrete floor slab. The test results demonstrated that the steel frame with PTED connections had the capacity for large deformations and self-centering capability without evident damage to the primary steel components. Although the concrete floor slab had uniform crack patterns at a 3% inter-story deformation, it didn't limit the performance of the PTED connections.

**2.2.3.3 Post-tensioned Friction Damped Connections (PFDC)**

Rojas et al. (2005) analytically investigated another self-centering steel connection: the Post-tensioned Friction Damped Connection (PFDC). As shown in Fig. 2-16, this type of connection includes friction plates bolted in the flanges of beams and columns for energy dissipation and high strength post-tensioned strands to provide the self-centering property. The shear force is transferred by the combination of the bolted shear tab, the friction between beams and columns, and the friction plate. Results from an inelastic analysis of a six-story, four-bay, Steel Moment-Resisting Frame (SMRF) with PFDC indicated that the system has good energy dissipation and self-centering capability and its seismic performance could exceed that of the SMRF with conventional welded connections.

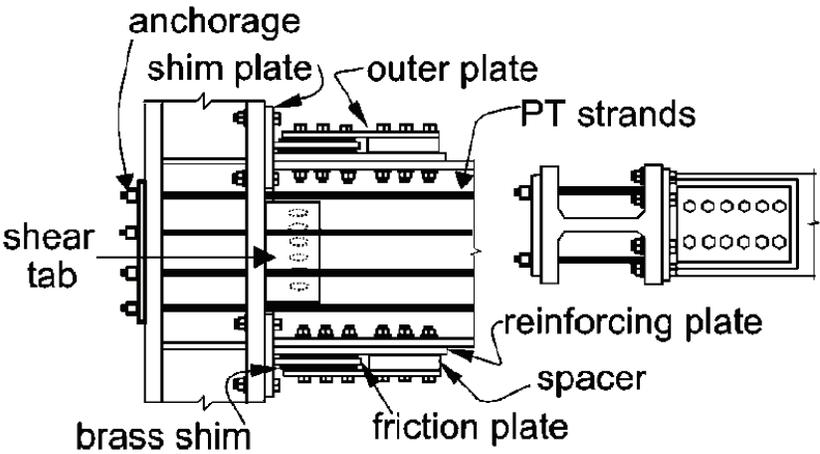


Figure 2-16 Post-tensioned Friction Damped Connection (PFDC) Details (Rojas et al. 2005)

## **2.3 Summary and Research Motivation**

In this chapter, various Self-Centering Systems (SCS) were reviewed. The self-centering capability (i.e. no permanent deformation) and energy dissipation mechanism without major damage to structural components are the main benefits of SCS.

Up to now, the main applications of SCS have been as rocking systems in bridge structures. Although there have been several experimental studies of SCS on building models or subassemblies, there is lack of practical applications to real building structures. More research efforts focusing on the actual implementation of SCS in complete building systems are necessary for the development of these systems. To address this issue, this report investigates the use of SCS for the re-design of a real structure: the MCEER Demonstrated Hospital. To experimentally validate the analytical research, shake table testing was conducted on two 1/3 scale models of steel SCPT and SMRF structures.

The objectives of this study are to provide more experiences on the actual design methods of SCS in building structures and demonstrate the feasibility of SCS as a practical design alternative in complete structural systems.



## **CHAPTER 3**

### **SEISMIC RESPONSE OF SINGLE-DEGREE OF-FREEDOM SELF-CENTERING SYSTEMS**

In order to apply the concept of Self-Centering Systems (SCS) for the seismic design of actual Multi-Degree-of-Freedom (MDOF) structural systems, it is useful to consider the seismic response of Single-Degree-of-Freedom (SDOF) systems under strong ground motion excitations. This chapter describes the development of a Matlab-based computer code used to evaluate the seismic response of SDOF SCS and Elasto-Plastic Systems (EPS), which properties were introduced in Chapter I (see Fig. 1-5). The seismic performances of these two systems are also discussed. The main objective of this chapter is to develop insight and design aides to be used for the seismic design of MDOF equipped with SCS.

This chapter builds on similar analyses performed by Christopoulos et al. (2002a) under a more restricted set of ground motions. The approach taken herein is somewhat similar to the one used by Christopoulos but the computer codes are independently developed and the evaluation method is more complex and more integrated so that the results can be used more directly as a preliminary guideline for the seismic design of more complex structures equipped with SCS. This improved analysis of the results coupled with a more elaborated ensembles of ground motion excitations is the main advantage over the results obtained by Christopoulos et al. (2002a).

### 3.1 Hysteretic Models of Single-Degree-of-Freedom Self-Centering Systems and Elasto-Plastic Systems

A SDOF model with either SCS or EPS is considered in the numerical analysis as shown in Fig. 3-1 (a). An equivalent damping ratio of 5% of critical is introduced to consider other sources of energy dissipation in the structure that are not associated with hysteretic behavior of the structural elements after yielding. The global hysteretic characteristic of the structural system is considered by a nonlinear spring with initial stiffness  $k_0$ , as shown in Fig. 3-1 (b) and (c).

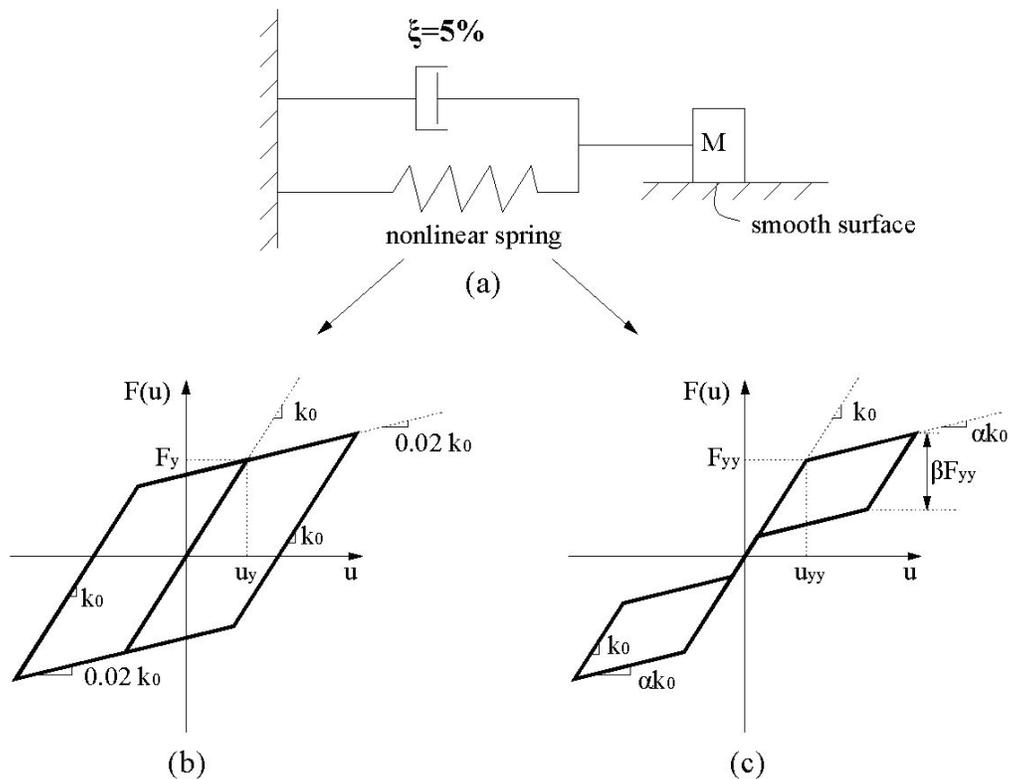


Figure 3-1 SDOF Numerical Model (a) SDOF system (b) Hysteretic Loop of Elasto-Plastic Systems (EPS) (c) Hysteretic Loop of Self-Centering Systems (SCS)

The bilinear elasto-plastic hysteretic behavior shown in Fig. 3-1 (b) for the nonlinear spring represents an idealized response of an EPS, which is representative of Steel Moment-Resisting Frames (SMRF) designed with post-Northridge welded connections (Gross et al. 1999). In this idealized loop, it is assumed that there is no strength degradation. The EPS yields at displacement  $u_Y$  under the yielding force  $F_Y$  and a post-yielding stiffness is assumed to be equal to 2% of the initial stiffness. The second nonlinear spring considered exhibits a self-centering behavior as shown in Fig. 3-1 (c). This model represents the SMRF incorporating the self-centering components. The initial stiffness is the same as that of the EPS and the post-yielding stiffness is variable and is controlled by the factor  $\alpha$ , which ranges from 2% to 35%. Another controlling parameter is the energy dissipating factor  $\beta$ , which ranges from 0 to 1. The SCS model yields at the displacement  $u_{YY}$  under the force  $F_{YY}$ , which indicates that the yielding point in these two models can be different. This difference in yielding force between the SCS and EPS is controlled by the strength ratio factor  $\eta$  and yielding ratio factor  $\psi$ . These controlling parameters will be presented and discussed in details in Section 3.2.2.

### **3.2 Seismic Parametric Study of Single-Degree-of-Freedom (SDOF) Self-Centering Systems (SCS)**

A numerical parametric study of the seismic response of SDOF SCS is conducted in this section using two ensembles of MCEER simulated earthquake ground motions, in which the 25 motions are strong ground motions having a probability of exceeding of 2% in 50 years in California and the other 25 motions have a 10% probability of exceedance in 50 years in CA.

The seismic responses of SDOF SCS and EPS are compared and the results are plotted in the form of design charts to estimate the optimum SCS parameters to guide the design of more complex MDOF structures equipped with SCS.

**3.2.1 Definition of Earthquake Ground Motions**

The MCEER simulated ground motions, developed by Wanitkorkul and Filiatrault (2005), were used as input in this parametric study. These ground motions were generated at an assumed hospital site located on the west-coast of the United States. A near-field site model was used in this simulation because the west-coast in the U.S. was considered to be in a near-fault region. The simulated ground motions were scaled to match the uniform hazard spectra in Northridge, CA, provided by the U.S. Geological Survey (USGS) as shown in Fig. 3-2. There were four different hazard levels (2%, 5%, 10%, 20% probability of exceedance

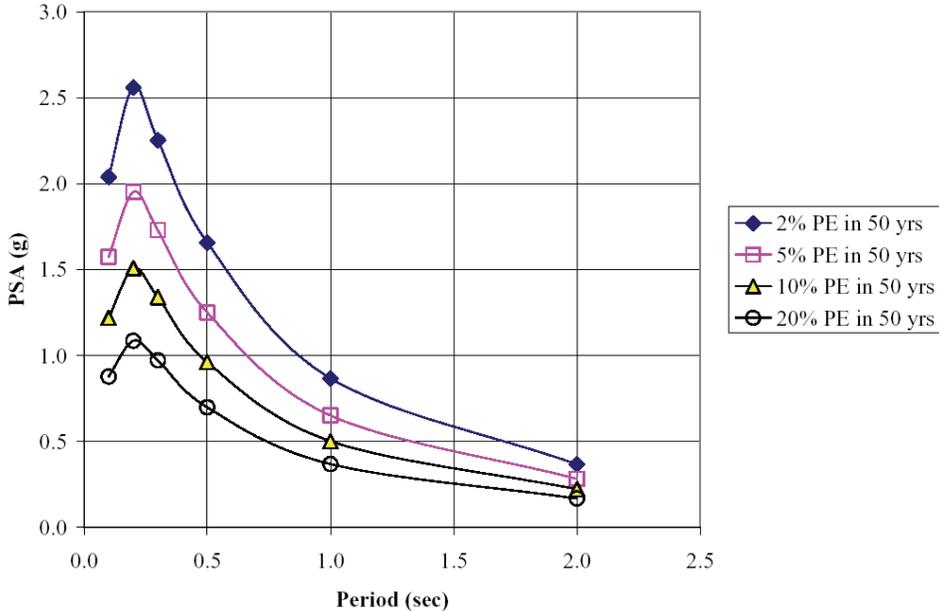


Figure 3-2 USGS Uniform Hazard Spectra: Northridge, CA (Wanitkorkul and Filiatrault, 2005)

in 50 years) considered in this simulation. An ensemble of 25 earthquake motions for each probability level was generated by using a simulation methodology proposed by Mavroeidis and Papageorgiou (2003). More details about the simulation procedure and ground motions can be found in the report by Wanitkorkul and Filiatrault (2005). In order to analyze the seismic response of SDOF SCS and EPS under the excitations of earthquakes in severe and medium hazard levels, the two ensembles of 25 MCEER ground motions with 2% probability and 10% probability of exceedance in 50 years are selected.

### 3.2.2 Equations of Motion and Controlling Parameters

The equation of motion for a nonlinear SDOF system is given by:

$$m\ddot{u} + c\dot{u} + F(u, \dot{u}) = -m\ddot{u}_g \quad (3.1)$$

where  $m$  is the mass,  $c$  is the damping constant,  $u$  is the relative displacement,  $\dot{u}$  is the relative velocity,  $\ddot{u}$  is the relative acceleration of SDOF model and  $F(u, \dot{u})$  is the nonlinear restoring force, which are also shown in Fig. 3-1. The restoring force  $F(u, \dot{u})$  is dependent on not only the displacement but also the velocity due to the nonlinear properties of the SCS and EPS. The term  $\ddot{u}_g$  is the ground acceleration and the mass,  $m$ , is representative of the total mass of a structure. The damping constant is defined by:

$$c = 2\xi\sqrt{mk_0} \quad (3.2)$$

where the damping ratio  $\xi$  is assumed to be 5% of critical and  $k_0$  is the initial stiffness of the model, which can be calculated from:

$$k_0 = \frac{4\pi^2 m}{T_0^2} \quad (3.3)$$

where  $T_0$  is the initial fundamental period of the structure. According to the prestandard and commentary for the seismic rehabilitation of buildings developed by the Federal Emergency Management Agency (FEMA 356), the fundamental period  $T_0$  can be estimated by:

$$T_0 = C_t h_n^\lambda \quad (3.3)$$

where:

$C_t = 0.035$  for Steel Moment Resisting Frame (SMRF) systems

$h_n =$  Height (in feet) above the base to roof level

$\lambda = 0.8$  for SMRF systems

It is assumed that the story height of typical SMRFs is 11ft and the number of stories ranges from one to fifteen. From Eq. (3.3) and the above assumption, the range of fundamental periods can be calculated as:

$$0.23\text{sec} < T_0 < 2.08\text{sec} \quad (3.4)$$

In order to consider the full range response of short-period systems, a lower bound value for  $T_0$  is assumed at 0.01 second.

Another controlling factor is the strength factor  $\eta$ , which is defined by:

$$\eta = \frac{F_y}{W} \quad (3.5)$$

where  $F_y$  is the yielding strength and  $W$  is the weight of the structure. This factor controls the yielding force, which is also the design shear, with the specific weight of the building.

This factor is assumed to be less than 1. In order to compare the results from EPS and SCS with different yielding strength, a yielding strength ratio factor  $\psi$  is given by:

$$\psi = \frac{(Fy)_{SCS}}{(Fy)_{EPS}} = \frac{\frac{(Fy)_{SCS}}{W}}{\frac{(Fy)_{EPS}}{W}} = \frac{\eta_{SCS}}{\eta_{EPS}} \quad (3.6)$$

To fully define the hysteretic properties of SCS, two supplemental factors,  $\alpha$  and  $\beta$ , are incorporated as shown in Fig. 3-1. The range of post-yielding stiffness factor  $\alpha$  is assumed to be from 0.02 to 0.35, which is mainly dependent on the initial PT force in the PT components of SCS. The energy dissipating factor  $\beta$  is assumed to range from 0 to 1. When  $\beta$  equals to 0, the SCS behaves as a bi-linear elastic system without energy dissipation. A value of  $\beta$  equal to 1 corresponds to the largest energy dissipating capacity that a SCS can exhibit while maintaining its self-centering characteristics. If  $\beta$  is larger than 1, the SCS loses its self-centering capability and a residual displacement occurs at the end of each cycle. Table 3-1 shows these controlling parameters, in which  $\alpha$  and  $\beta$  are used for modeling the SDOF SCS, while in SDOF EPS model,  $\alpha$  is assumed to be 0.02 and  $\beta$  actually equals to 2. After the permutation and combination of these four factors, there are 576 SDOF SCS models and 36 SDOF EPS models included in the parametric analysis.

**Table 3-1 Controlling Parameters in SDOF SCS and EPS models**

$\alpha$	$\beta$	$\eta$	$T_0$ (sec)
0.02	0.0	0.05	0.01
0.10	0.3	0.10	0.25
0.20	0.6	0.20	0.50
0.35	1.0	0.30	1.00
		0.50	1.50
		1.00	2.00

### 3.2.3 Integration Algorithm for Nonlinear Time-History Dynamic Analysis

The Newmark's method (Newmark 1959) is used for integrating the second-order equation of motion in the time domain. This integration algorithm is based on solving the following equations at the end of each time-step:

$$m\ddot{u}_{i+1} + c\dot{u}_{i+1} + (f_s)_{i+1} = p_{i+1} \quad (3.7)$$

$$\dot{u}_{i+1} = \dot{u}_i + [(1 - \gamma)\Delta t]\ddot{u}_i + (\gamma\Delta t)\ddot{u}_{i+1} \quad (3.8)$$

$$u_{i+1} = u_i + (\Delta t)\dot{u}_i + [(0.5 - \Gamma)(\Delta t)^2]\ddot{u}_i + [\Gamma(\Delta t)^2]\ddot{u}_{i+1} \quad (3.9)$$

where  $\gamma$  and  $\Gamma$  determine the variation of acceleration over a time step and the stability and accuracy of this integration. These three equations (3.7, 3.8 and 3.9) are combined to compute the unknown displacement  $u_{i+1}$ , velocity  $\dot{u}_{i+1}$  and acceleration  $\ddot{u}_{i+1}$  at the time-step  $i+1$  from the known  $u_i$ ,  $\dot{u}_i$  and  $\ddot{u}_i$  at the time-step  $i$  through an iteration process;  $f_s$  is the restoring force and  $p$  is the external force such as earthquake excitations.

The Newmark's method applied to nonlinear systems introduces numerical errors, as shown in Fig. 3-3 (a). Those numerical errors are the results of using the tangent stiffness  $k_{ab}$  instead of the unknown secant stiffness  $k_{ab'}$  in the numerical calculations. To minimize these errors, the Newton-Raphson iteration can be added to the Newmark's method as illustrated in Fig. 3-3 (b). This iteration is performed within one time step to make the numerical result at the end of a first iteration (point  $B$  in Fig. 3-3 (b)) closer to the exact point  $B'$  to an acceptable error level after 3 iterations. A full description of the Newmark's method with the Newton-Raphson iteration can be found in Chopra (2000).

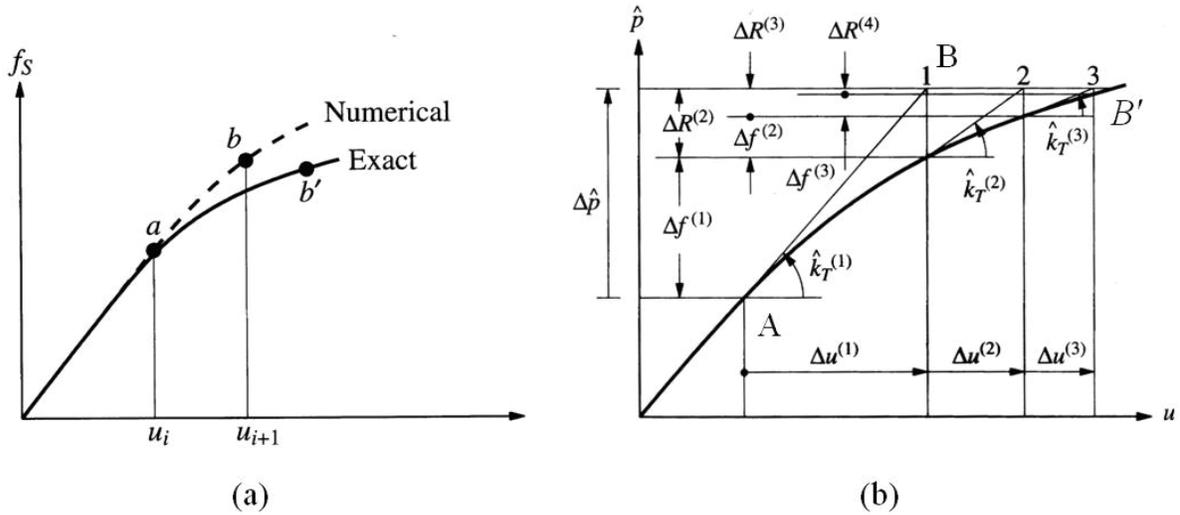


Figure 3-3 Error in Newmark's Method (a) Numerical Error in Newmark's method (b) Newton-Raphson Iteration (Chopra 2000)

The Newmark's method, assuming a constant average acceleration with  $\gamma = \frac{1}{2}$  and  $\Gamma = \frac{1}{4}$  in Eqs. 3.8 and 3.9, is used herein to develop a Matlab-based computer code to evaluate the seismic response of SDOF SCS and EPS. The Newton-Raphson iteration is implemented to decrease the error in the computation. The time step is set to about 0.005 second and 20 seconds of free vibration are added at the end of each earthquake ground motion in order to determine the residual displacement of these systems.

### 3.2.4 Performance Indices

The seismic responses of SDOF SCS and EPS can be evaluated by various performance indices including the mean displacement ductility, the mean maximum absolute acceleration and a new Relative Performance Index (RPI) defined herein.

### 3.2.4.1 Mean Displacement Ductility

The displacement ductility ratio is defined by:

$$\mu = \left| \frac{u_{\max}}{u_y} \right| \quad (3.10)$$

where  $u_{\max}$  is the maximum relative displacement of the SDOF system and  $u_y$  is the corresponding yielding displacement. Based on the numerical results obtained from an ensemble of 25 ground motions, the mean displacement ductility  $\bar{\mu}$  of a specific model is defined by:

$$\bar{\mu} = \frac{\sum_{i=1}^{25} \mu_i}{25} \quad (3.11)$$

where  $\mu_i$  is the displacement ductility of a specific model under the  $i^{\text{th}}$  earthquake in an ensemble of 25 ground motions.

The mean displacement ductility is an indicator of the deformation capacity in structures. It is also an indicator of the potential damage that a structure would experience under an ensemble of earthquakes.

### 3.2.4.2 Mean Maximum Absolute Acceleration

The mean maximum absolute acceleration  $\bar{a}_{\max}$  is defined by:

$$\bar{a}_{\max} = \frac{\sum_{i=1}^{25} \max(|(\ddot{u} + \ddot{u}_g)_i|)}{25} \quad (3.12)$$

where  $\ddot{u}$  is the relative acceleration response of the systems and  $\ddot{u}_g$  is the ground acceleration. The maximum absolute acceleration is an indicator of the damage potential to acceleration-sensitive non-structural components. It also represents the maximum transient shear force transferred to the structure by the ground motion.

### 3.2.4.3 Relative Performance Index (RPI)

In this study, a new Relative Performance Index (RPI) is defined as follows:

$$RPI = b \cdot \frac{(\bar{u}_{\max})_{SCS}}{(\bar{u}_{\max})_{EPS}} + (1 - b) \cdot \frac{(\bar{a}_{\max})_{SCS}}{(\bar{a}_{\max})_{EPS}} \quad (3.13)$$

where  $b$  is a coefficient with values between 0 and 1 (the values of 0, 0.3, 0.5, 0.7, 1.0 are used in this study);  $(\bar{u}_{\max})_{SCS}$  and  $(\bar{a}_{\max})_{SCS}$  are the mean response values of maximum displacement and absolute acceleration in Self-Centering Systems (SCS);  $(\bar{u}_{\max})_{EPS}$  and  $(\bar{a}_{\max})_{EPS}$  are the mean response values of maximum displacement and absolute acceleration in Elasto-Plastic Systems (EPS)

The RPI is a weighted index, which provides an indicator of the combined effect of the displacement and acceleration responses on the structure. When the RPI equals to 1, the seismic performances of the SCS and EPS are equivalent. When the RPI is less than 1, the seismic performance of the SCS is “better” than that of the EPS. Conversely, when the RPI is larger than 1, the seismic performance of the SCS is “worse” than that of the EPS. Therefore, the RPI can serve as a single numerical parameter to guide the choice of optimum SCS parameters in the design of more complex structures.

When  $b$  is less than 0.5 in Eq. (3.13), the numerical value of the RPI is mainly controlled by acceleration response. This range of values of  $b$  could be used for the design of structures incorporating acceleration sensitive nonstructural components. When  $b$  is larger than 0.5 in Eq. (3.13), the numerical value of the RPI is mainly controlled by displacement response. This range of values of  $b$  could be used for the design of structures in which damage to the structural components is of prime interest and/or for the design of structures incorporating drift-sensitive nonstructural components.

### **3.2.5 Numerical Evaluation of Seismic Response of SDOF SCS and EPS**

In this section the results of the numerical parametric study conducted on the seismic response of SDOF SCS and EPS are presented along with the observed trends

#### **3.2.5.1 Seismic Response of SDOF SCS**

The mean displacement ductility ratios for the SCS excited by the ground motions corresponding to a 2% and 10% probability of exceedance in 50 years are shown in Figs. 3-4 and 3-5, respectively. The ductility ratio is reduced for increasing values of  $\alpha$  and  $\beta$  especially in lower period systems ( $T_0 \leq 0.5$ sec) or for increasing the value of strength ratio  $\eta$ . The values of  $\alpha$  and  $\beta$  have no impact on the ductility for  $\eta=1.0$ , where most of the responses remains in the elastic ranges and consequently  $\mu_\Delta$  remains under unity. As expected, the mean displacement ductility is larger for 2%-50 years ground motions than for 10%-50 years ground motions.

The mean maximum absolute acceleration for the SCS excited by the ground motions corresponding to a 2% and 10% probability of exceedence in 50 years are shown in Figs. 3-6 and 3-7, respectively. When  $\alpha$  is increasing, the mean maximum absolute acceleration is also increasing. This increase in acceleration response is more significant for lower period systems ( $T_0 \leq 1.0\text{sec}$ ). As  $\alpha$  is increasing, the mean maximum absolute acceleration is increasing more quickly for lower values of  $\eta$ . When  $\beta$  is increasing, the acceleration response is reduced for  $T_0 \leq 1.0\text{sec}$  and higher values of  $\eta$ , except  $\eta=1.0$ .

For other cases, the mean maximum acceleration is insensitive to values of  $\beta$ . For  $\eta=1.0$ , the values of  $\bar{a}_{\max}$  have no change for different values of  $\alpha$  and  $\beta$  with same period ( $T_0$ ) because the relative high yielding strength and the acceleration responses tend to be in the elastic ranges.

It is noted again that the earthquakes in the highest hazard level (2% probability) causes the larger acceleration response by comparing the Fig. 3-6 and 3-7.

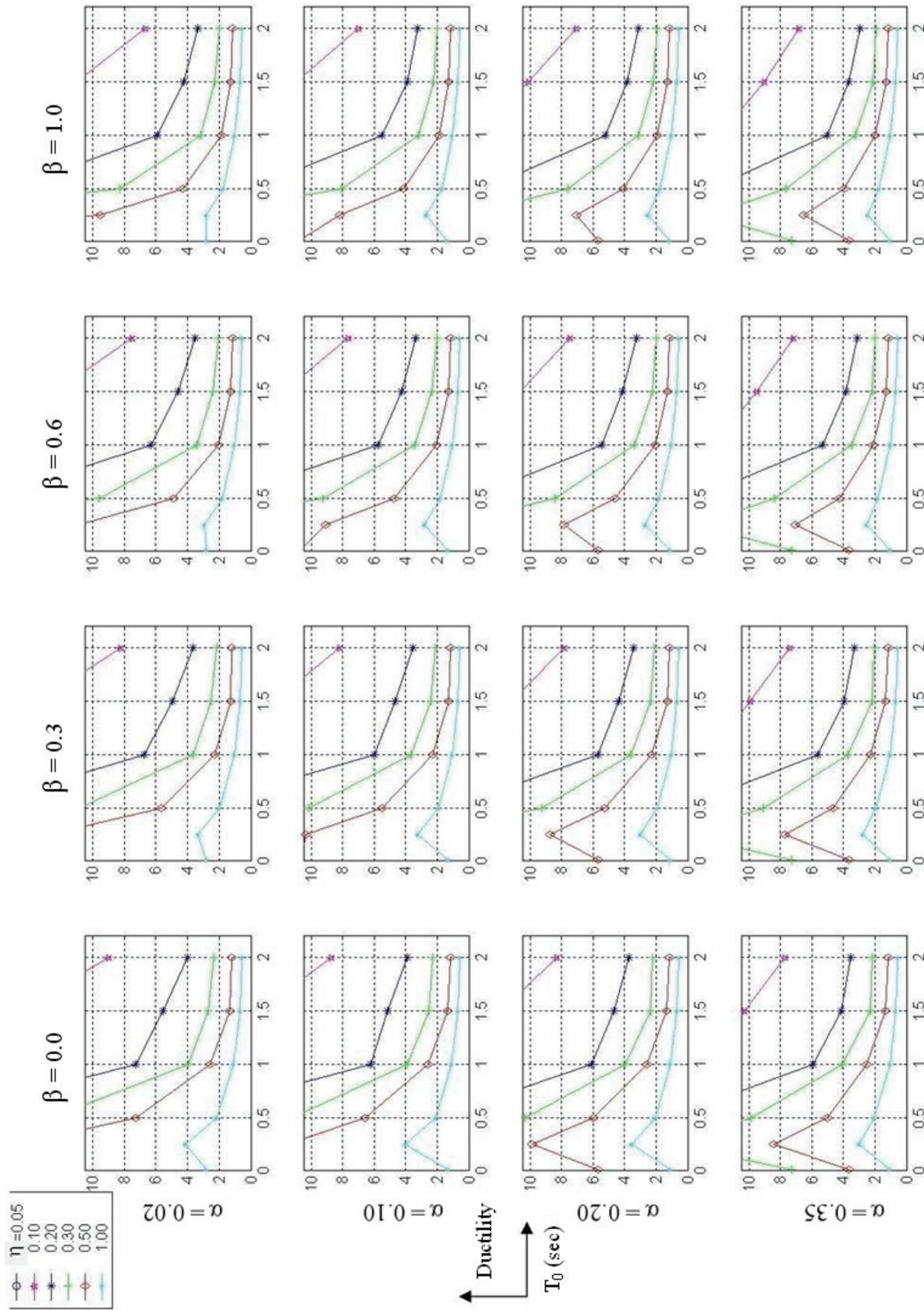


Figure 3-4 Mean Displacement Ductility of Self-Centering System (SCS) under Earthquakes with 2% probability of exceedance in 50 years

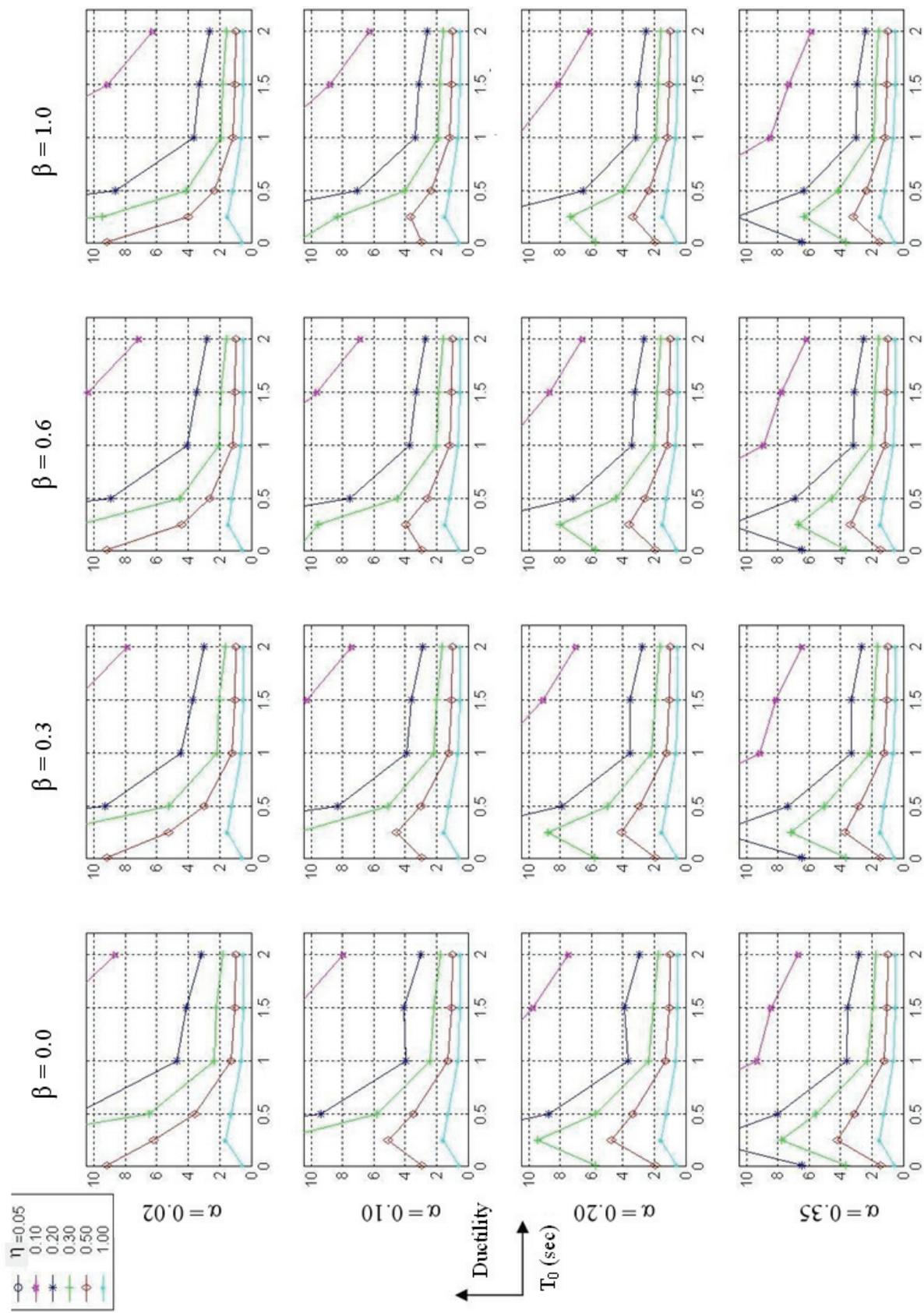


Figure 3-5 Mean Displacement Ductility of SCS under Earthquakes with 10% probability of exceedance in 50 years

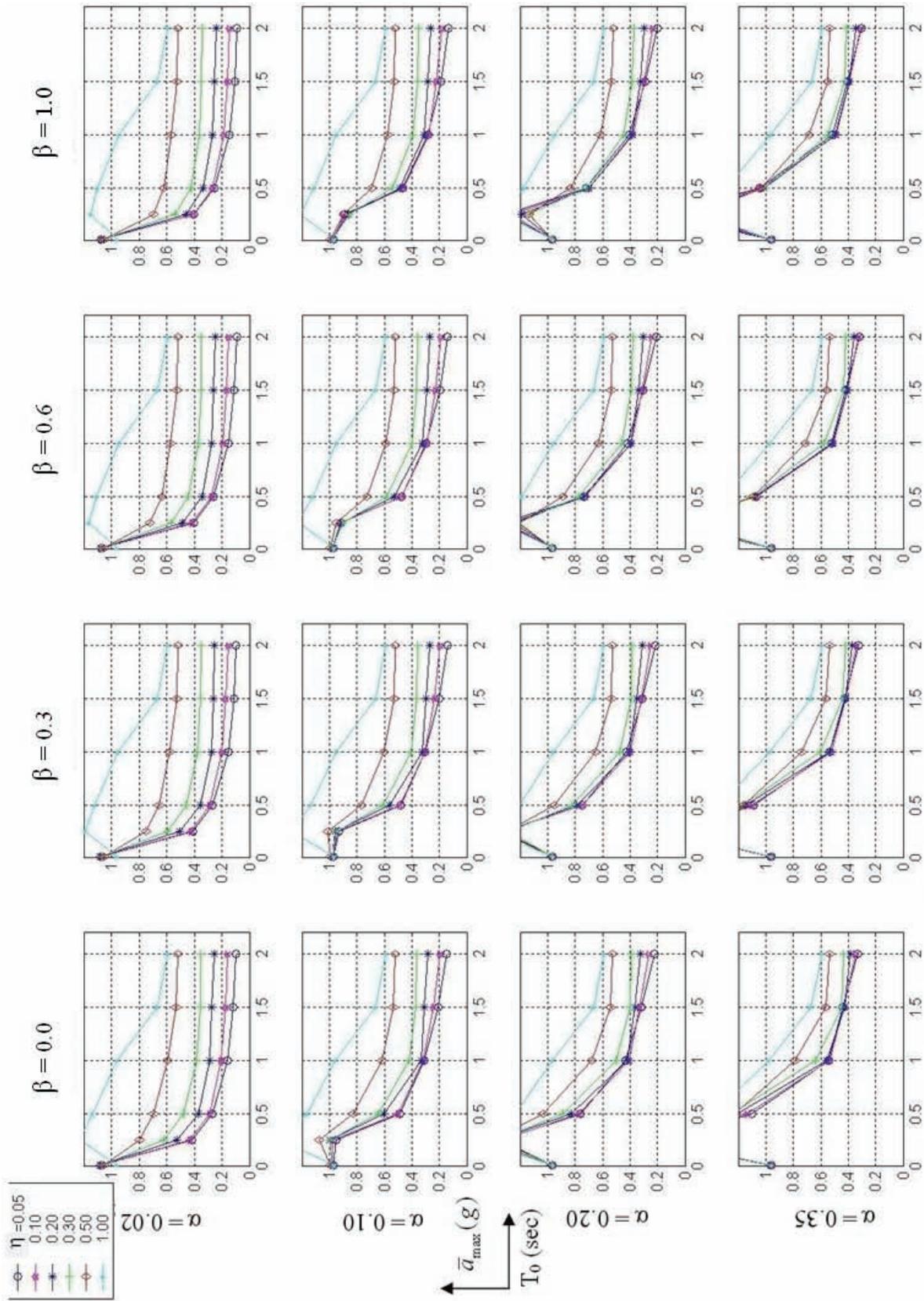


Figure 3-6 Mean Maximum Absolute Acceleration of SCS under Earthquakes with 2% probability of exceedance in 50 years

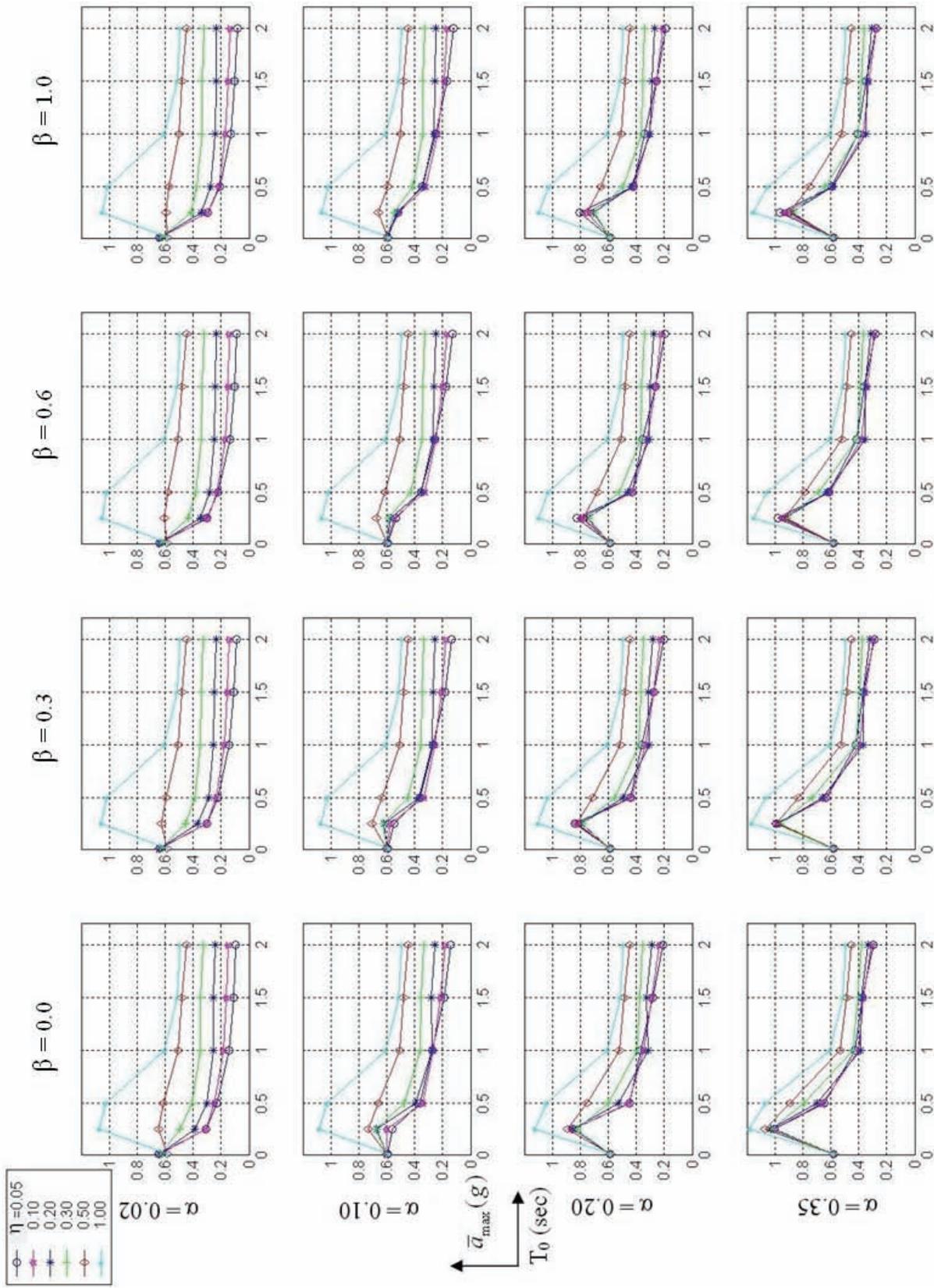


Figure 3-7 Mean Maximum Absolute Acceleration of SCS under Earthquakes with 10% probability of exceedance in 50 years

### 3.2.5.2 Seismic Response of SDOF EPS

The seismic responses of SDOF EPS excited by the ground motions corresponding to a 2% and 10% probability of exceedance in 50 years are shown in Figs. 3-8 and 3-9, respectively.

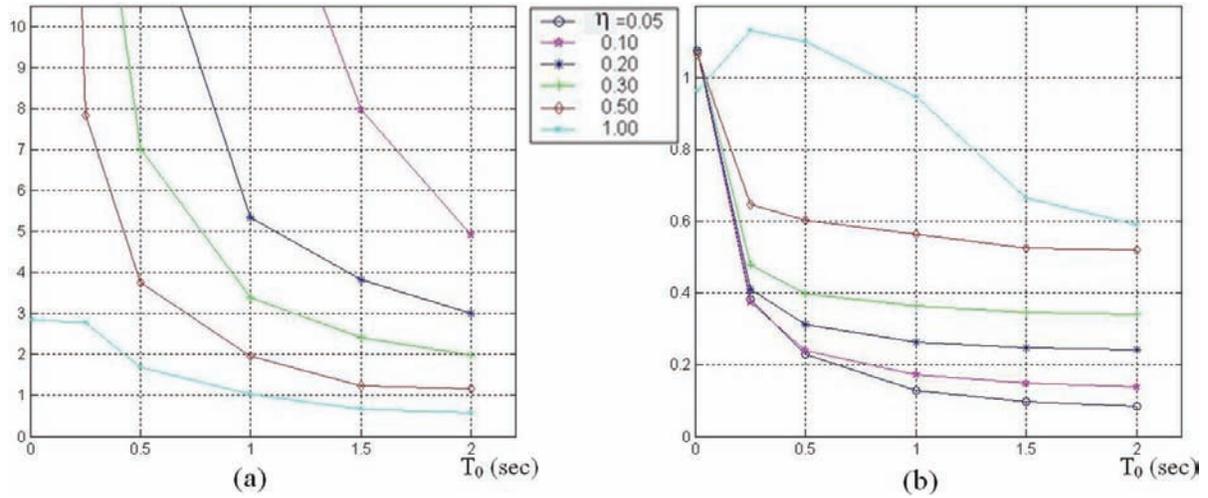


Figure 3-8 Seismic Responses of EPS under Ground Motions having 2% Probability of Exceedance in 50 years (a) Mean Displacement Ductility (b) Mean Maximum Absolute Acceleration (g)

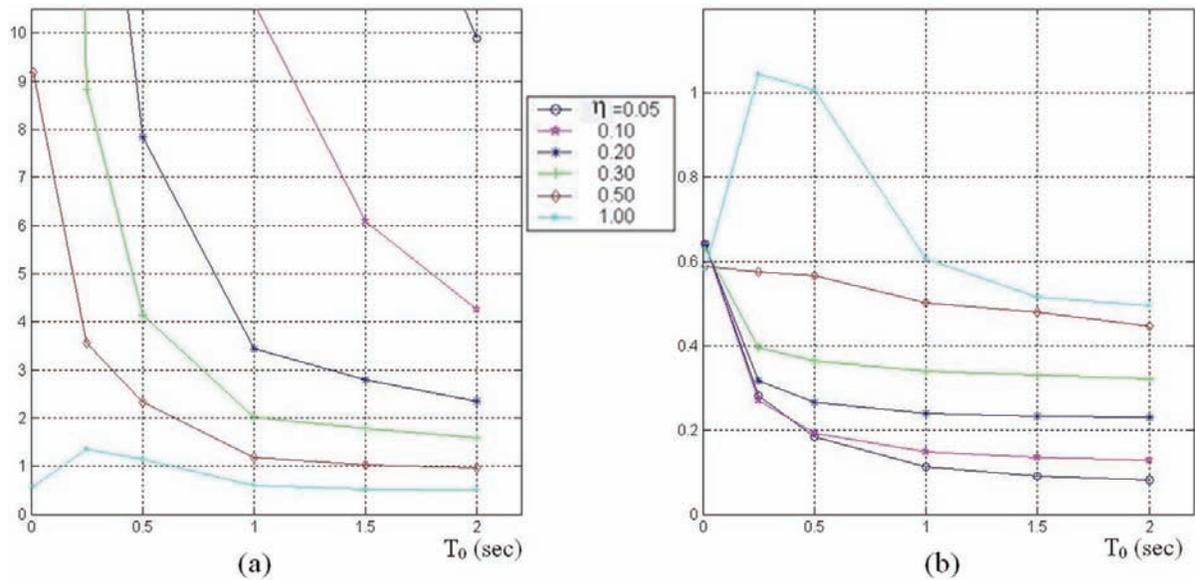


Figure 3-9 Seismic Responses of EPS under Ground Motions having 10% Probability of Exceedance in 50 years (a) Mean Displacement Ductility (b) Mean Maximum Absolute Acceleration (g)

From Fig. 3-8 (a) and 3-9 (a), the mean values of the displacement ductility ratio are increasing when the fundamental period or the strength factor is reduced. When the initial period of systems is less than one second, the mean ductility is increasing significantly when the strength factor is decreased. As shown in Fig. 3-8 (b) and 3-9 (b), the mean maximum absolute acceleration is increasing when the strength factor  $\eta$  increases. For short period systems ( $T_0 < 0.5 \text{ sec}$ ), the acceleration values are increasing rapidly compared to that of long period systems ( $T_0 > 1 \text{ sec}$ ).

Also, all responses of EPS are increased from lower hazard level to higher one. For the long period systems ( $T_0 > 1 \text{ sec}$ ) with larger values of  $\eta$  ( $> 0.3$ ), these responses are similar for both hazard levels.

### **3.2.6 Comparative Seismic Responses between SDOF Self-Centering Systems (SCS) and Elasto-Plastic Systems (EPS)**

In order to compare the seismic responses of SDOF SCS and EPS, the Relative Performance Index (RPI) introduced in Section 3.2.4.4 is used to generate a series of total 70 comparative figures under two ensembles of 25 earthquakes records with 2% and 5% exceedance of 50 years, which are shown in Appendix A and B.

From Appendix A and B, it is found that, when RPI is completely based on maximum displacement response (i.e. the weighted factor of displacement ratio equals to 1), if  $\alpha$  is increasing, the value of RPI decreases especially for shorter period systems ( $T_0 \leq 1.0 \text{ sec}$ )

and if  $\beta$  decreases, the value of RPI is increasing. But for longer period systems ( $T_0 > 1.0\text{sec}$ ) with  $\alpha \geq 0.1$ , the RPI is insensitive to  $\beta$ .

When RPI is completely based on maximum absolute acceleration responses (i.e. the weighted factor of acceleration ratio is 1), if  $\alpha$  is increasing, the value of RPI is increasing especially for lower  $\psi$  values and if  $\beta$  is increasing, the RPI is reduced especially for lower period systems  $T_0 (\leq 1.0\text{sec})$  with lower  $\psi$  values. For  $T_0 \geq 1.0$ , the RPI is insensitive to  $\beta$ .

When the weighted factor is not equal to 0 or 1, the RPI represents a combined effect of the above two trends. When the weighted factor of the displacement ratio is larger than that of the acceleration ratio, the RPI tends to be close to the first one (i.e. weight factor of displacement ratio equals to 1). The reverse effect is also observed.

A final conclusion for the comparative response of the two systems is that to any reasonable specific EPS, there will be at least one SCS, in which the displacement response is less than that of EPS or the acceleration response is less than that of EPS or both of these two responses of SCS are reduced compared to that of EPS. In other words, the seismic performance of SCS can exceed that of EPS in respect of displacement responses or acceleration responses or both.

### **3.2.7 Recommended Use of Design Aid Charts**

The Relative Performance Index (RPI) shown in Appendix A and B can be used as design aid charts to design SCS based on the existing EPS.

The weight factor in the design aid charts ranges from 0 to 1 and the values of SCS factors such as  $\alpha$ ,  $\beta$ ,  $\eta$ ,  $\psi$  and  $T_0$  are shown in Table 3-1 (see Sec. 3.2.2). Based on the previous discussion in Sec. 3.2, the ranges of values of those factors in the design aid charts were determined by the physical condition and design code such as FEMA 356. Therefore, those values are rational so that an equivalent EPS of a real structure can be found and then the corresponding SCS with better seismic performance can also be determined. Based on the discussion in Sec. 3.2.5 and the observations for these design aid charts in Appendix A and B, it is believed that the RPI can be interpolated with the selected factors within the range of the existed values.

For the recommendations of the selection of the individual factors in SCS, the post-yielding factor  $\alpha$  could be selected lower than 0.2, since to achieve higher value  $\alpha$ , much more re-centering force (i.e. post-tensioned force) are needed which may lead to the yielding of the beam or other components. The energy dissipating factor  $\beta$  should be selected less than 1 to maintain the self-centering properties without residual drifts and may be selected around 0.6~0.8 since the larger  $\beta$  can lead to reduced seismic responses as a result of dissipating more energy while too large  $\beta$  (i.e. close to 1) may lead to lose part of the capability of eliminating the residual drifts in the condition that not all the connections are SCS (e.g. beam-to-column joints are SCS while the base column connections are EPS). The strength ratio factor  $\psi$  may be selected as low as 0.5 (i.e.  $\eta_{SCS} = 0.5\eta_{EPS}$ ) since when a value less than 0.5 for  $\psi$  is selected, the displacement response of SCS is higher than that of EPS.

If a building has more displacement-sensitive non-structural components, the weight factor  $b$  should be selected larger than 50% while  $b$  should be considered less than 50% (i.e.  $1-b > 50\%$ ) to account for more acceleration-sensitive non-structural components existing in other types of buildings such as hospitals.

Since the design aid charts (i.e. RPI) in Appendix A were calculated based on the MCEER simulated earthquakes with 2% probability of exceedance in 50 years, it should be used when designing for the Maximum Considered Earthquake (MCE). When the Design Basis Earthquake (DBE) is considered, the design aid charts in Appendix B are recommended.

### **3.3 Summary**

In this chapter, a numerical seismic nonlinear analysis of Single-Degree-of-Freedom (SDOF) Self-Centering Systems (SCS) and Elasto-Plastic Systems (EPS) was conducted by using the Newmark's integration method with the Newton-Raphson iteration. The comparative evaluation of the seismic responses of two systems was presented, which indicated that the seismic performance of SCS is better than that of EPS. The main advantage of SCS is that no permanent deformation and no inelastic action associated with damage to the main structural elements occur after earthquakes. The parametric comparative results in Appendix A and B were given and discussed, which can be used as design aid charts to the seismic design of actual structures incorporating SCS. The use of design aid charts (i.e. RPI) obtained will be discussed further in Chapter IV.

## **CHAPTER 4**

### **SEISMIC PERFORMANCE OF MCEER DEMONSTRATION HOSPITAL WITH SELF-CENTERING SYSTEMS**

As discussed in previous chapters, there has not been yet any practical implementation of Self-Centering Systems (SCS) in Steel Moment-Resisting Frames (SMRF). Before such implementation can take place, the seismic performance of SMRF equipped with SCS needs to be investigated. In this chapter, numerical investigations were carried out to obtain the seismic response of one of the steel buildings part of the MCEER West Coast Demonstration Hospital located in Southern California. This facility was designed in the 1970s and the WC70 design version of the building (Yang and Whittaker 2002) was considered in this study.

Two numerical studies are discussed in this Chapter. In the first study, the response of the original WC70 building model is compared to that of the same building in which ideal self-centering hysteretic properties are introduced at each beam-to-column connection. In the second numerical study, actual Self-Centering Post-Tensioned (SCPT) connections are designed and analyzed.

The structural properties and the numerical model of the WC70 hospital building are first introduced. Thereafter, based on the seismic response of the Single-Degree-of-Freedom (SDOF) SCS presented in Chapter II, the hospital was redesigned with ideal SCS and a comparison between the seismic performance of the original building and the redesigned

one is presented in terms of push-over analysis under cyclic loading, seismic analysis and fragility analysis under the MCEER strong ground motions (see Chapter III). A complete seismic design procedure for SCPT connections is then developed. Finally, this procedure is used to design a SCS with SCPT connections for the hospital building and the results from the push-over analysis, the seismic analysis and the fragility analysis are compared again to that of the original building.

**4.1 Numerical Model of MCEER Demonstration Hospital**

The steel building considered in this study is part of the MCEER West Coast Demonstration Hospital. This facility is located in Southern California and was constructed in the 1970s (hence the designation WC70). A conventional Steel Moment-Resisting Frame (SMRF) system was used in this 4-story building. As shown in Fig. 4-1, the lateral load-resisting system of this building is composed of 4 SMRFs in the North-South direction (lines B, F, J

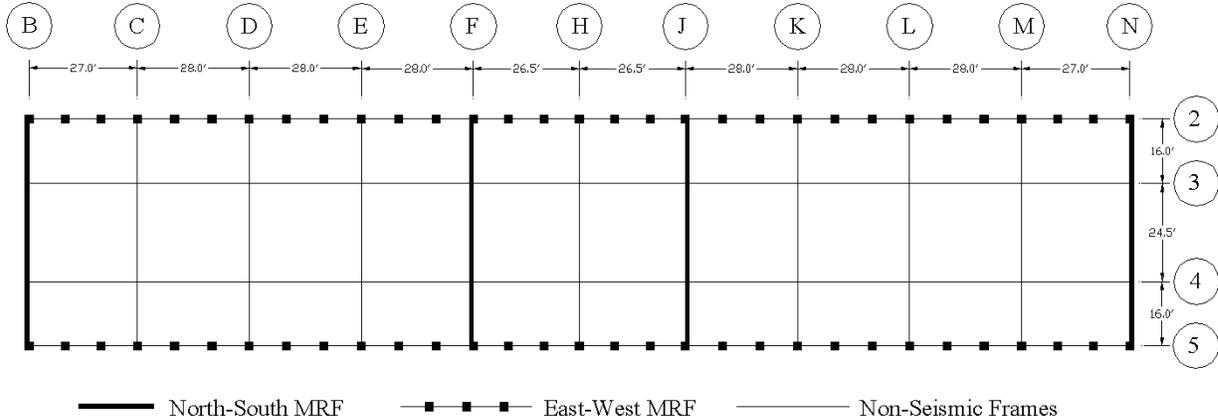


Figure 4-1 Plan View of WC70 Hospital Building Considered

and N) and 2 MRFs in the East-West direction (lines 2 and 5). The dimensions and sections of moment resisting frames on lines B, N, F and J in Fig. 4-1 are shown in Figs. 4-2 and 4-3.

The other frames in the structure were designed only to resist the gravity loads and are assumed not to contribute in resisting the lateral forces.

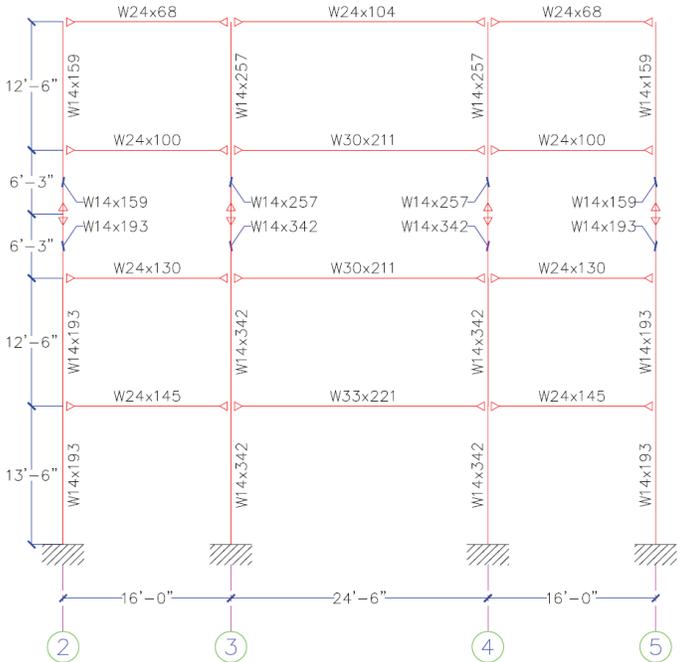


Figure 4-2 Elevation View of Exterior Moment Resisting Frame on lines B and N of WC70 (Yang and Whittaker 2002)

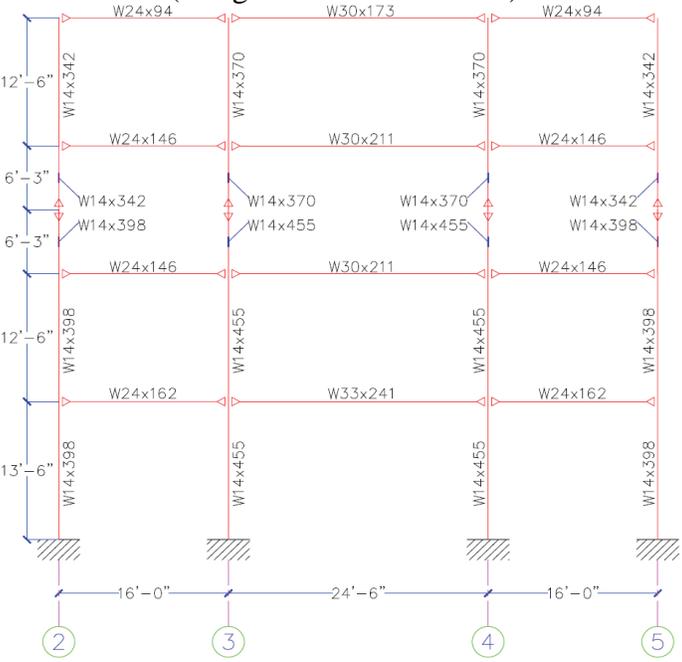


Figure 4-3 Elevation View of Interior Moment Resisting Frame on lines F and J of WC70 (Yang and Whittaker 2002)

In order to evaluate the seismic performance of this building, a two-dimension numerical model is developed in the RUAUMOKO-2D software (Carr 2004) to model the North-South MRFs as shown in Fig. 4-4. Due to the symmetry, only half of these 4 MRFs were used in this model: the exterior frames on line B or N and the interior frames on lines F or J. A pinned gravity column is introduced in the model to account for the second order (P- $\Delta$ ) effect

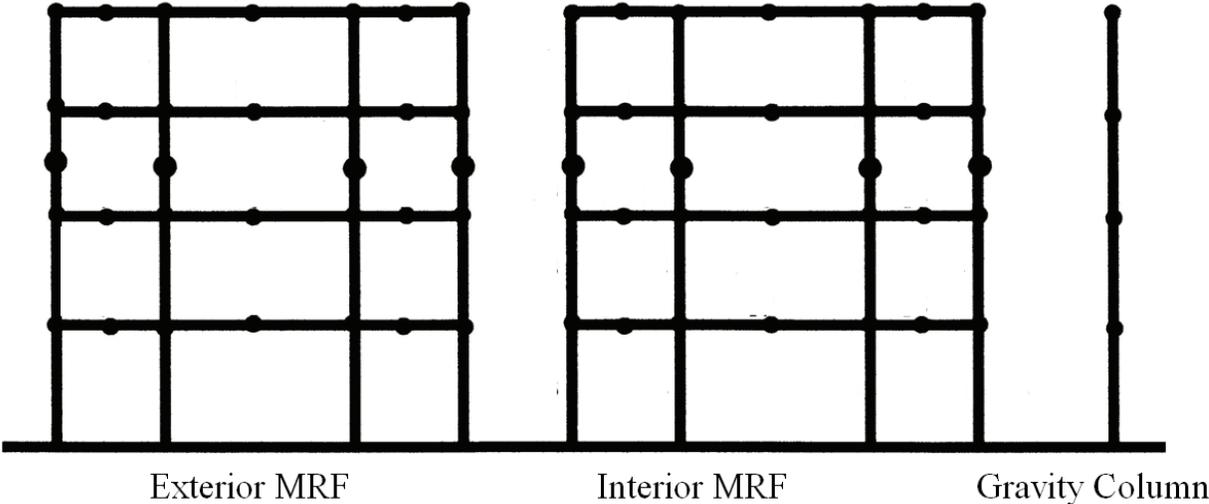


Figure 4-4 Numerical Model of WC Hospital

generated by those non-seismic frames on the lateral load-resisting frames. A frame element is utilized to represent all beams and columns. The inelastic response is assumed to be concentrated in the plastic hinges forming at the end of frame members. A bilinear moment-curvature hysteresis similar to that of Elasto-Plastic Systems (EPS) with a post-yielding stiffness equal to 0.02 is assigned to all frame numbers, as indicated in Fig. 4-5, where the  $k_0$  is the initial flexural stiffness of the frame elements. All slab contributions are neglected. This model is defined as the original model of the building. The model the retrofitted with

ideal SCS and the model re-designed with SCPT systems are discussed in the following sections.

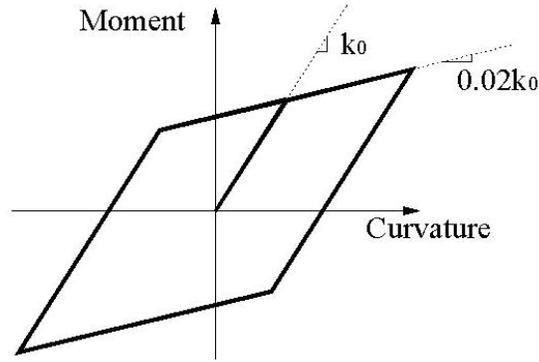


Figure 4-5 Bilinear Moment-Curvature Hysteresis

## 4.2 Re-Design Procedure and Numerical Model for Steel Moment Resisting Frames (SMRF) with Ideal Self-Centering Systems (SCS)

The ideal SCS represent structural components that exhibit complete self-centering properties by changing the elasto-plastic hysteretic properties to self-centering hysteresis at every beam-to-column connections but without considerations of installing actual self-centering devices such as those mentioned in Chapter II. The re-design objectives are to reduce the displacement and/or acceleration responses and to decrease or eliminate the residual drifts in the SCS frame compared to the SMRF structures so that the damage on the non-structural and structural components can be reduced after earthquakes. Based on the seismic performance of SDOF SCS in Chapter III, the redesign procedure for SMRF with ideal SCS can be conducted through the following steps:

- (1) Determine the structural properties of the SMRF including the fundamental period, seismic weights, mode shapes, yielding strength.
- (2) According to the modal contributions, the Multi-Degree-of-Freedom (MDOF) SMRF is reduced to be an equivalent SDOF system. For regular SMRF buildings, the first mode contribution to seismic response is considered to be dominant while the contributions from higher modes are neglected. In fact, it is assumed that all SCS joints in the structure experience the same rotation, thereby implying a linear (straight line) first modal shape. The parameters of the equivalent SDOF system can be determined by the first mode properties of the MDOF system. For particular conditions, where higher modal contributions need to be considered, the method for generating an equivalent SDOF system can be found in Chopra, 2000.
- (3) Determine the original strength ratio factor  $\eta_{\text{original}}$  of the SMRF based on its beam and column properties.
- (4) According to the Relative Performance Index (RPI) values given in Appendix A or B, determine the parameters of the SCS including the post-yielding factor  $\alpha$ , energy dissipating factor  $\beta$ , yielding ratio factor  $\psi$  and required strength ratio factor  $\eta_{\text{required}}$ . For different redesign objectives, the relative importance of displacement and acceleration response can be taken into account by the specific weighting factor  $b$  introduced in the RPI calculation, as described in Chapter III. If one is concerned with displacement-sensitive structural and nonstructural components, more weight

should be given to the displacement response than that of acceleration response in the RPI calculation, while more weight should be given to the acceleration response if one is concerned with acceleration-sensitive components.

- (5) Establish a numerical model for the original SMRF buildings and implement the SCS with selected parameters to change the original model to a new one with SCS.
- (6) Evaluate and compare the seismic performance of these two models.
- (7) If the performance of the SCS can not reach a satisfactory level, repeat the step (4), (5) and (6) until a good performance is achieved. Usually with the initial parameters, a relative good retrofit result can be obtained. In order to get better results, the procedure can be repeated by modifying the controlling parameters of the SCS.

Those steps in re-design procedure are also summarized in Fig. 4-6.

Since the WC70 building is a welded SMRF, the above procedure can also be used in the re-design process. The structural properties of this building are provided in Section 4.1. The total seismic weight of the WC70 building is 20881 kN and its modal properties are shown in Table 4-1.

**Table 4-1 Modal Properties of the Original Model for WC70 Building**

Mode	Period (sec)	Cumulative Mass
1 <sup>st</sup>	0.760	85%
2 <sup>nd</sup>	0.257	96%
3 <sup>rd</sup>	0.148	99%
4 <sup>th</sup>	0.100	100%

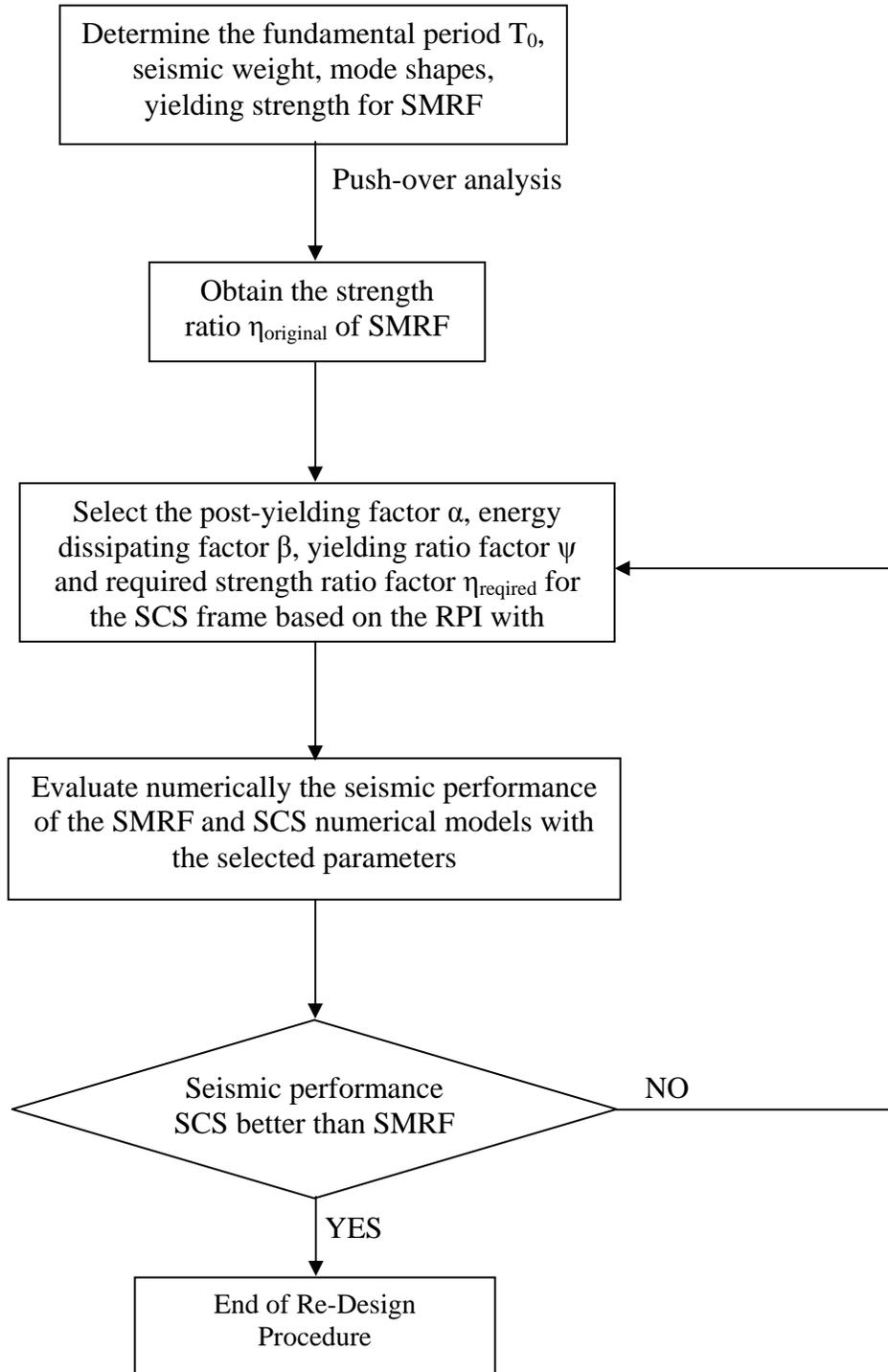


Figure 4-6 Flowchart of the Re-Design Procedure for Ideal SCS Frame

The results shown in Table 4-1 indicate that 85% of the total participating mass is associated with the 1<sup>st</sup> mode of vibration. Therefore, the seismic response of the building is mostly dominated by the first mode and the weight of the equivalent SDOF system used to model the building is made equal to the first mode weight:

$$W_{eq} = 85\% W_{total} = 85\% * 20881 = 17749kN \tag{4.1}$$

Where  $W_{eq}$  is the weight of equivalent SDOF system and  $W_{total}$  is the total weight. In order to determine the strength ratio of the original building,  $\eta_{original}$ , a push-over analysis on the original building model is conducted using a set of lateral forces proportional to the 1<sup>st</sup> mode shape, as shown in Fig. 4-7. The structure yields at point A with a top displacement of 114 mm and a base shear of 10690 kN, as seen in Fig. 4-7 (a). The strength ratio  $\eta_{original}$  is given by:

$$\eta_{original} = \frac{F_Y}{W_{eq}} = \frac{10690.1}{17749} = 0.60 \tag{4.2}$$

where  $F_Y$  is the yielding force.

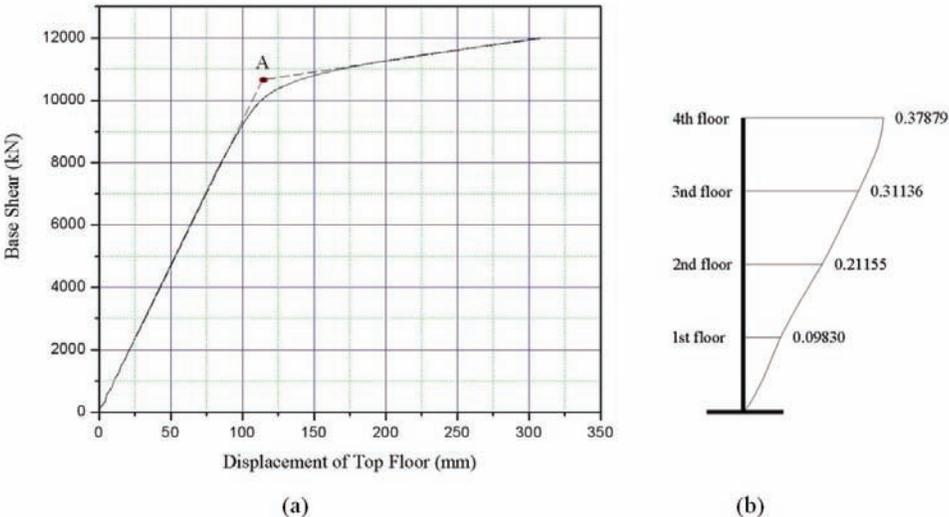


Figure 4-7 Pushover Analysis of the SMRF structure  
 (a) Pushover Curve (b) First Mode Shape

Because hospital buildings contain mainly sensitive medical equipments that are acceleration-sensitive, the calculations of the RPI should be based mainly on minimizing acceleration response. With the original strength ratio  $\eta_{\text{original}}=0.6$  and fundamental period  $T=0.7\text{sec}$ , the parameters of  $\alpha$ ,  $\beta$  and  $\eta_{\text{scs}}$  can be selected in Appendix A or B with RPI values based on  $b=0.3$  (70% acceleration response and 30% displacement response). Although the results presented in Appendix A and B were not made specifically for  $\eta_{\text{original}}$  equal to 0.6,  $\eta_{\text{original}}=0.5$  is selected as the closest value. The initial values of  $\alpha$ ,  $\beta$  and  $\psi$  selected in Appendix B in page B-13 are listed in Table 4-2. With these three parameters, the RPI of the corresponding SCS is less than 1, which indicates that the performance of WC70 hospital with SCS should be better than that of original building.

**Table 4-2 Parameters in self-centering systems**

$\alpha$	$\beta$	$\psi = \eta_{\text{scs}} / \eta_{\text{original}}$
0.2	0.6	0.6

The original numerical model of the WC70 building described in Section 4.1 is then re-designed with ideal self-centering system incorporating the parameters shown in Table 4-2. For this purpose, the Elasto-Plastic (EP) properties of the moment-curvature relationship at the end of all beam elements in the original model are changed to incorporate the characteristics of SCS to generate a new SCS model. This process is illustrated in Fig. 4-8. The values of  $\alpha$  and  $\beta$  in Fig. 4-8(c) are taken in Table 4-2. The yield moments in Fig. 4-8(b) and (c) have the following relation:

$$\frac{M_{Y_{scs}}}{M_{Y_{original}}} = \frac{\frac{(F_Y)_{scs}}{W_{eq}}}{\frac{(F_Y)_{original}}{W_{eq}}} = \frac{\eta_{scs}}{\eta_{original}} = \psi = 0.60 \quad (4.3)$$

where:

$F_Y$ : yielding force for the equivalent SDOF system

$M_Y$ : yielding moment for the beams or columns

$W_{eq}$ : the weight of equivalent SDOF system.

$\eta$ : the strength ratio

$\psi$ : the yielding ratio, the value of 0.60 is taken from Table 4-2.

In order to evaluate the complete performance of SCS, the base connections (i.e. the ends of columns connected to the foundation) of the SCS model are also changed to three types: fixed-base connections, pin-base connections and SCS-base connections. As shown in Fig. 4-8, the fixed-base connections are similar to that of the original model with the EP property and the ideal SCS-base connections lead base columns to have the ability of SCS, while the pin-based connections do not supply any moment resistance.

Therefore, a total of four numerical models are considered, as shown in Table 4-3. The first model represents the original WC70 building and the other three represent a re-design of the same building with ideal SCS incorporating three different base connections.

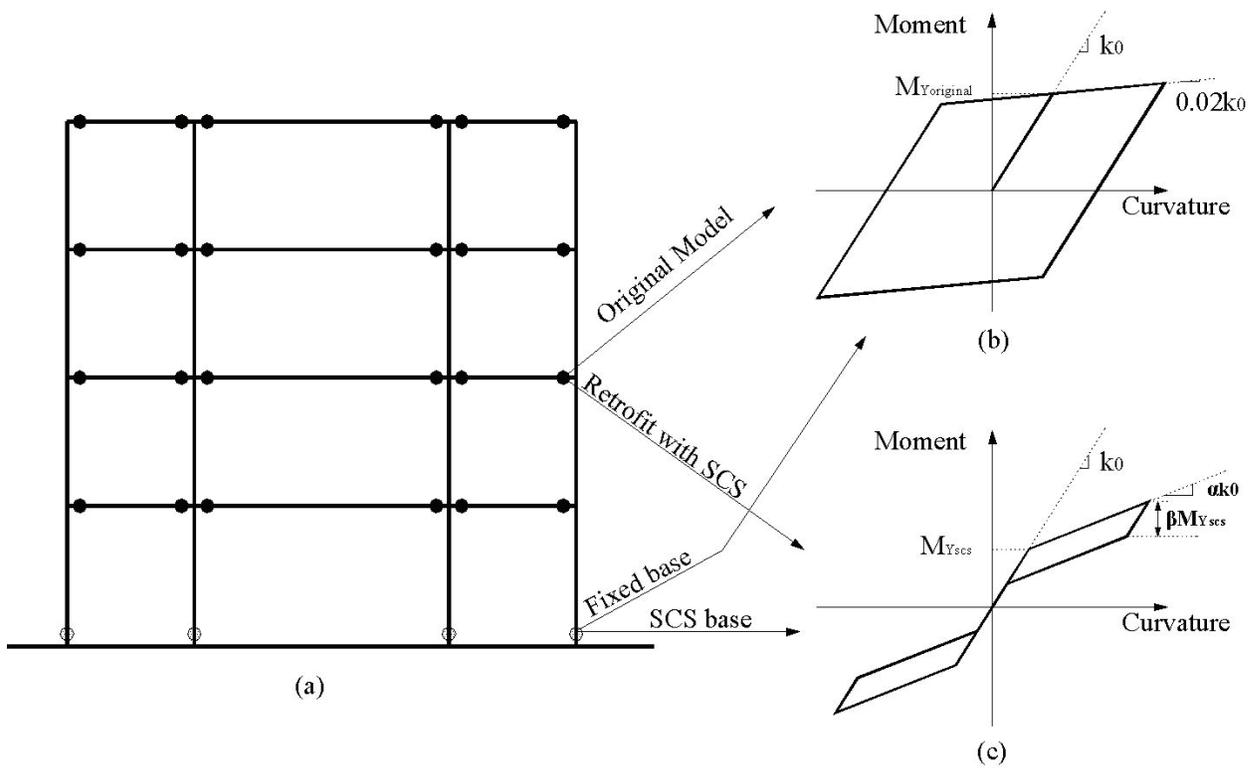


Figure 4-8 Numerical Models of EPS and SCS

(a) Frame in original model of WC70 hospital (b) Bilinear Hysteresis (c) SCS Hysteresis

**Table 4-3 Moment-curvature Properties of the Original and SCS models**

	Original Model (Elasto-Plastic System, EPS)	Self-Centering System (SCS) Model		
		Fixed-base	SCS-base	Pin-base
Moment-Curvature Property (MCP) of beam elements	EPS Hysteresis	SCS Hysteresis	SCS Hysteresis	SCS Hysteresis
MCP of column elements	EPS Hysteresis	EPS Hysteresis	EPS Hysteresis	EPS Hysteresis
MCP of base column elements	EPS Hysteresis	EPS Hysteresis	SCS Hysteresis	No moment resistance

### **4.3 Performance of the Original WC70 Hospital Model and the Models Re-Designed with Ideal Self-Centering Systems**

In this section, the seismic performances of the original and re-designed WC70 building models are investigated using Cyclic nonlinear quasi-static and seismic analyses. The results obtained are also expressed in terms of fragility curves.

#### **4.3.1 Cyclic Nonlinear Quasi-Static Analyses**

Cyclic nonlinear quasi-static analyses were conducted under one cycle of loading to evaluate the global seismic performance of the four models described above. The results of these Cyclic nonlinear quasi-static analyses are shown in Fig. 4-9. Lateral loads are added at each floor and scaled to the first mode shape of the building (see Fig. 4-7(a)) until a total lateral force (base shear) of 12650kN is reached. This base shear level corresponds for the original building to a global top floor displacement ductility factor equal to 4. At this base shear level, the maximum top floor displacement is about 1000 mm for the SCS model with pin base. Obviously, this large displacement indicates that the building would have collapsed because the stiffness of the first floor is too low due to the pin base. As shown in Fig. 4-9, after retrofit with SCS with fixed and SCS base connections, the initial stiffness of the building is not changed substantially compared to the initial building but the first-yield force level is reduced significantly, as expected since the moment capacity of all SCS beam-to-column connections is only 60% of that of the original building. At the same base shear level, the maximum top floor displacements of the SCS models with fixed or SCS base connections are reduced substantially compared to that of the original building. The points A, B and C

identified in Fig. 4-9 are the residual drifts associated with the different building models, which indicates that the SCS model with SCS base connections possess the best performance. The residual drift of the SCS model with fixed base is also reduced largely compared to that of the original model. The possible residual drifts (point B and C) of SCS models are not zero as that of SDOF SCS discussed in the previous Chapter. In MDOF systems, the SCS are only added into the beam-to-column connections, which can not prevent the column ends to yield. From the results of these analyses, it is clear that the performance of SCS models exceeds that of original model except for the pin-base SCS model.

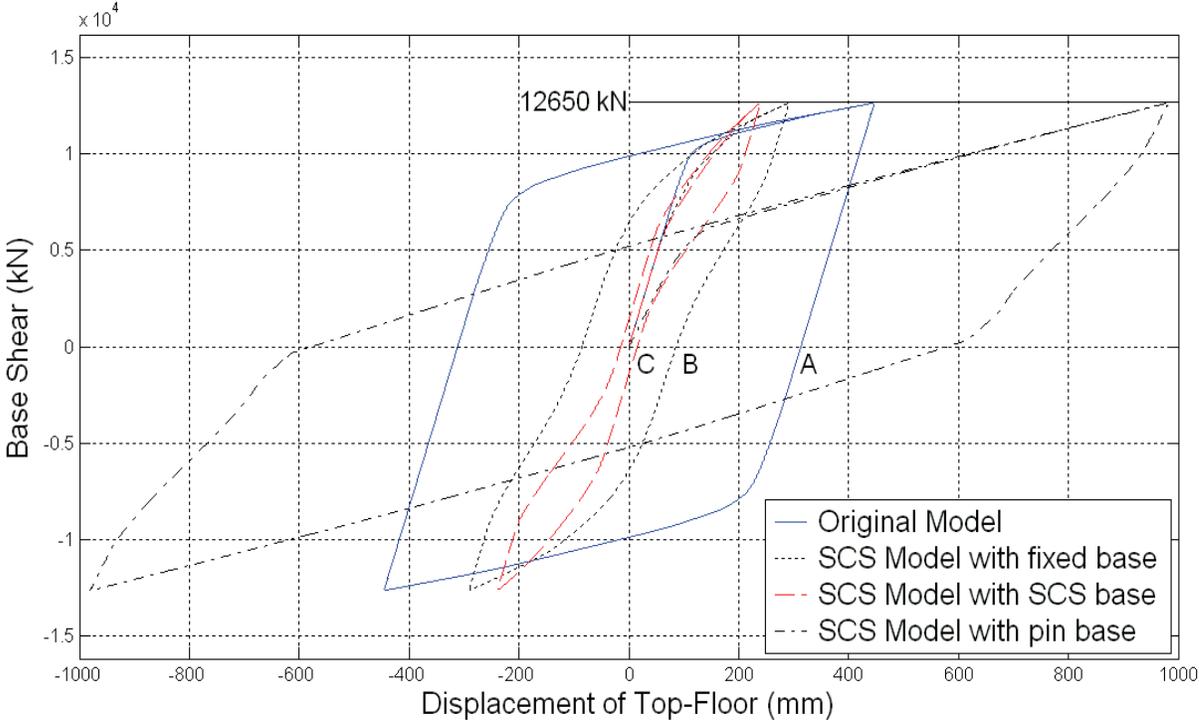


Figure 4-9 Hysteretic Loop in Push-over Analysis

#### 4.3.2 Seismic Analyses under MCEER Ground Motions

Seismic analyses were conducted by using the 4 ensembles of 25 MCEER ground motions having 2%, 5%, 10% and 20% probabilities of exceedance in 50 years, as introduced in Section 3.2.1. Analysis results for mean values of various responses from each ensemble of ground motion are shown in Tables 4-4 to 4-7. Through comparing the results, it is found that:

- (i) From these four tables, it is apparent that although the acceleration response of the SCS model with pin base is the lowest, the maximum first interstory displacement and column ductility are very large, which indicates that this pin-base model can not resist the strong ground motions due to the low stiffness of the first floor. Therefore, in the following discussion, the SCS model with pin base is neglected and the SCS model is referred to as the SCS model with fixed or SCS base.
- (ii) By comparing the mean values of top floor maximum displacements and interstory drifts, it is found that the response of the SCS model is similar or slightly larger than that of the original model. The reason for this increase is that the determination of SCS parameters is mainly based on the acceleration response with a 70% weight and the non-SCS connections such as base columns also have a negative impact on the displacement response of the SCS model.

**Table 4-4 Mean Values of response under 25 MCEER ground motions with 20% probability of exceedance in 50 years**

Mean Values	unit	Original Model		SCS Model with Fixed Base		SCS Model with SCS Base		SCS Model with Pin Base	
		mm	%	mm	%	mm	%	mm	%
<b>Maximum Displacement</b>	top floor	130.14	0.84	136.63	0.88	142.71	0.92	186.82	1.20
<b>Residual Displacement</b>	top floor	12.48	0.08	3.00	0.02	0.59	0.00	26.55	0.17
<b>Maximum Interstory Drift</b>	1 <sup>st</sup> floor	40.21	0.98	40.22	0.98	43.16	1.05	131.77	3.20*
	2 <sup>nd</sup>	40.90	1.07*	42.72	1.12*	44.11	1.16*	32.92	0.86
	3 <sup>rd</sup>	36.06	0.95	38.63	1.01	38.46	1.01	21.08	0.55
	4 <sup>th</sup>	25.30	0.66	24.91	0.65	26.08	0.68	14.15	0.37
<b>Residual Interstory Drift</b>	1 <sup>st</sup>	5.06	0.12*	2.53	0.06*	0.17	0.00	26.38	0.64*
	2 <sup>nd</sup>	4.22	0.11	0.37	0.01	0.29	0.01*	0.19	0.00
	3 <sup>rd</sup>	2.82	0.07	0.22	0.01	0.23	0.01	0.03	0.00
	4 <sup>th</sup>	0.52	0.01	0.02	0.00	0.02	0.00	0.01	0.00
<b>Acceleration Unit</b>		g		g		g		g	
<b>Maximum Absolute Acceleration</b>	1 <sup>st</sup>	0.68		0.64		0.65		0.46	
	2 <sup>nd</sup>	0.74		0.70		0.69		0.42	
	3 <sup>rd</sup>	0.79		0.81		0.73		0.44	
	4 <sup>th</sup>	1.06*		0.93*		0.95*		0.59*	
<b>Maximum Ductility of Beams</b>	2.90		4.77 / 1.0**		4.93 / 1.0**		5.41 / 1.0**		
<b>Maximum Ductility of Columns</b>	2.03		1.90		4.17 / 1.0		4.65		

\* Shaded values represent maximum and residual interstory drifts and maximum absolute accelerations.

\*\*Ductility values in SCS energy dissipating components / ductility in structural members.

**Table 4-5 Mean Values of response under 25 MCEER ground motions with 10% probability of exceedance in 50 years**

Mean Values	unit	Original Model		SCS Model with Fixed Base		SCS Model with SCS Base		SCS Model with Pin Base	
		mm	%	mm	%	mm	%	mm	%
<b>Maximum Displacement</b>	top floor	143.38	0.92	156.99	1.01	169.95	1.09	203.55	1.31
<b>Residual Displacement</b>	top floor	13.46	0.09	4.37	0.03	1.65	0.01	32.43	0.21
<b>Maximum Interstory Drift</b>	1 <sup>st</sup> floor	45.63	1.11	48.92	1.19	51.33	1.25	147.33	3.58*
	2 <sup>nd</sup>	45.03	1.18*	48.36	1.27*	53.11	1.39*	35.19	0.92
	3 <sup>rd</sup>	41.10	1.08	42.86	1.12	45.63	1.20	23.20	0.61
	4 <sup>th</sup>	29.86	0.78	29.41	0.77	30.71	0.81	15.60	0.41
<b>Residual Interstory Drift</b>	1 <sup>st</sup>	5.27	0.13*	3.65	0.09*	0.41	0.01	32.22	0.78*
	2 <sup>nd</sup>	4.79	0.13	0.74	0.02	0.87	0.02*	0.19	0.01
	3 <sup>rd</sup>	3.60	0.09	0.29	0.01	0.46	0.01	0.04	0.00
	4 <sup>th</sup>	0.91	0.02	0.03	0.00	0.04	0.00	0.01	0.00
<b>Acceleration Unit</b>		g		g		g		g	
<b>Maximum Absolute Acceleration</b>	1 <sup>st</sup>	0.83		0.83		0.78		0.51	
	2 <sup>nd</sup>	0.88		0.85		0.83		0.46	
	3 <sup>rd</sup>	0.88		0.88		0.80		0.48	
	4 <sup>th</sup>	1.23*		1.07*		1.07*		0.65*	
<b>Maximum Ductility of Beams</b>		3.36		5.59 / 1.0**		5.97 / 1.0**		5.70 / 1.0**	
<b>Maximum Ductility of Columns</b>		2.81		2.78		5.05 / 1.0**		5.86	

\* Shaded values represent maximum and residual interstory drifts and maximum absolute accelerations.

\*\*Ductility values in SCS energy dissipating components / ductility in structural members.

**Table 4-6 Mean Values of response under 25 MCEER ground motions with 5% probability of exceedance in 50 years**

Mean Values	unit	Original Model		SCS Model with Fixed Base		SCS Model with SCS Base		SCS Model with Pin Base	
		mm	%	mm	%	mm	%	mm	%
<b>Maximum Displacement</b>	top floor	205.46	1.32	223.38	1.44	253.95	1.63	244.66	1.57
<b>Residual Displacement</b>	top floor	42.47	0.27	12.04	0.08	6.68	0.04	40.46	0.26
<b>Maximum Interstory Drift</b>	1 <sup>st</sup> floor	73.52	1.78*	79.10	1.92*	74.12	1.80	178.96	4.34*
	2 <sup>nd</sup>	67.14	1.76	68.40	1.80	84.63	2.22*	41.18	1.08
	3 <sup>rd</sup>	53.66	1.41	58.28	1.53	68.12	1.79	29.14	0.76
	4 <sup>th</sup>	34.55	0.91	40.10	1.05	42.59	1.12	19.28	0.51
<b>Residual Interstory Drift</b>	1 <sup>st</sup>	16.34	0.40*	8.35	0.20*	1.27	0.03	40.27	0.98*
	2 <sup>nd</sup>	14.15	0.37	3.29	0.09	3.79	0.10*	0.16	0.00
	3 <sup>rd</sup>	10.30	0.27	0.98	0.03	1.72	0.05	0.05	0.00
	4 <sup>th</sup>	2.56	0.07	0.11	0.00	0.28	0.01	0.01	0.00
<b>Acceleration Unit</b>		g		g		g		g	
<b>Maximum Absolute Acceleration</b>	1 <sup>st</sup>	1.05		0.93		0.96		0.63	
	2 <sup>nd</sup>	1.01		0.94		0.97		0.55	
	3 <sup>rd</sup>	1.04		1.00		0.99		0.57	
	4 <sup>th</sup>	1.41*		1.27*		1.25*		0.78*	
<b>Maximum Ductility of Beams</b>		5.42		7.97 / 1.0**		9.20 / 1.0**		6.35 / 1.0**	
<b>Maximum Ductility of Columns</b>		5.23		5.83		7.38 / 1.0**		7.96	

\* Shaded values represent maximum and residual interstory drifts and maximum absolute accelerations.

\*\*Ductility values in SCS energy dissipating components / ductility in structural members.

**Table 4-7 Mean Values of response under 25 MCEER ground motions with 2% probability of exceedance in 50 years**

Mean Values	unit	Original Model		SCS Model with Fixed Base		SCS Model with SCS Base		SCS Model with Pin Base	
		mm	%	mm	%	mm	%	mm	%
<b>Maximum Displacement</b>	top floor	235.06	1.51	246.25	1.58	281.77	1.81	308.82	1.99
<b>Residual Displacement</b>	top floor	51.98	0.33	17.00	0.11	10.19	0.07	45.89	0.30
<b>Maximum Interstory Drift</b>	1 <sup>st</sup> floor	85.57	2.08*	92.72	2.25*	87.26	2.12	243.44	5.91*
	2 <sup>nd</sup>	77.25	2.03	75.02	1.97	97.61	2.56*	44.52	1.17
	3 <sup>rd</sup>	62.34	1.64	61.20	1.61	74.90	1.97	32.27	0.85
	4 <sup>th</sup>	38.72	1.02	45.93	1.21	48.12	1.26	22.72	0.60
<b>Residual Interstory Drift</b>	1 <sup>st</sup>	19.87	0.48*	12.22	0.30*	1.85	0.04	45.65	1.11*
	2 <sup>nd</sup>	16.41	0.43	4.07	0.11	7.49	0.20*	0.31	0.01
	3 <sup>rd</sup>	11.72	0.31	1.40	0.04	2.76	0.07	0.10	0.00
	4 <sup>th</sup>	4.28	0.11	0.51	0.01	0.71	0.02	0.02	0.00
<b>Acceleration Unit</b>		g		g		g		g	
<b>Maximum Absolute Acceleration</b>	1 <sup>st</sup>	1.18		1.10		1.20		0.72	
	2 <sup>nd</sup>	1.13		1.14		1.12		0.63	
	3 <sup>rd</sup>	1.15		1.09		1.06		0.66	
	4 <sup>th</sup>	1.47*		1.41*		1.42*		0.90*	
<b>Maximum Ductility of Beams</b>		6.17		8.80 / 1.0**		10.33 / 1.0**		6.86 / 1.0**	
<b>Maximum Ductility of Columns</b>		6.38		7.12		8.63 / 1.0		11.57	

\* Shaded values represent maximum and residual interstory drifts and maximum absolute accelerations.

\*\*Ductility values in SCS energy dissipating components / ductility in structural members.

- (iii) For the residual displacement and residual interstory drift, the SCS model shows a very good ability to largely reduce or even eliminate residual drifts. The SCS model with SCS base has extremely small residual displacements and residual inter-story drifts, which is because the self-centering ability of the SCS base is accounted for more reduction of residual displacements. This reduction of residual drifts in SCS model can reduce the cost to return the building to its own position after earthquakes.
- (iv) The maximum floor acceleration is reduced by about 10% for the SCS building compared to the original building under the ground motions with 20%, 10% or 5% exceedance probability in 50 years. This indicates a better seismic performance of the SCS Model compared to that of the original model. As shown in Table 4-7, the 4% reduction in acceleration response indicates that under the ground motions with the high hazard level (2% exceedance probability in 50 years), the building re-designed with SCS has less impact on acceleration than that in lower hazard levels (5%, 10% or 20% exceedance in 50 years).
- (v) For the ductility of columns and beams, the ductility associated with the value “1.0” (i.e. no yielding) in SCS model means that no yielding and no damage occurred in the beams or columns while those yieldings happened in the energy dissipating components of SCS. Under the ground motions with the lower hazard levels, the ductility of columns in the SCS model is similar to that of the original model while for the higher hazard level the ductility is increasing for the SCS model compared to the original model, which indicates that the SCS model with fixed base can not

protect the columns from yielding if the columns of the SMRF building yielded in severe earthquakes. Therefore, reinforcing the base columns in the SCS model with fixed base or changing them to SCS base connections is necessary if the original design of the SMRF building can not save the base column connections.

- (vi) By comparing the responses between the SCS model with fixed base and SCS base, it is observed that the displacement and acceleration responses are similar in these two models. The SCS model with SCS base, however, has less residual drift than the SCS model with fixed base. Also, the residual drifts of these two models are much lower compared to that of the original model. Therefore, from this point of view, it is believed that the seismic performance of the SCS model with fixed base is similar to that of the SCS model with SCS base.

### 4.3.3 Fragility Analyses

Fragility is defined as the probability that a system exceeds a limit state as a function of some measures of seismic intensity. In this section, the mean return period corresponding to the different exceedance probabilities of MCEER ground motions is used as the measure of seismic intensity. In Section C1.6.1.2 of FEMA 356, the mean return period is defined by:

$$P_R = \frac{-Y}{\ln(1 - P_{EY})} \quad (4.4)$$

where  $P_R$ : Mean return period (years)

$Y$ : Time (years) for the desired earthquake hazard level

$P_{EY}$ : Probability of exceedance for the desired earthquake hazard level

Using the Eq. (4.4), the mean return period of the MCEER ground motions with 4 hazard levels having 20%, 10%, 5% and 2% exceedance probabilities in 50 years, is obtained as shown in Table 4-8.

**Table 4-8 Mean Return Period**

Hazard Level	20%/50years	10%/50years	5%/50years	2%/50years
Mean Return Period $P_R$ (year)	224.1	474.6	974.8	2474.9

The structural performance levels are used as the limit state in the fragility analyses. This limit state is defined by table C1-3 in FEMA 356, part of which is shown in the following table 4-9. In this table, the 0.2% residual drift for immediate occupancy is added by the authors in order to determine the fragility in the low performance level.

**Table 4-9 Limit State for Fragility Analysis (FEMA 356)**

Structural Performance Level		Immediate Occupancy	Life Safety	Collapse Prevention
Steel Moment Frames	Maximum Interstory Drift (transient)	0.7%	2.5%	5%
	Residual Drift (permanent)	0.2%	1%	5%

A regression process was conducted to obtain the approximate function between the fragility probability and the mean return period using the data obtained from the seismic analyses. The regression equation is defined as:

$$P_{FRAGILITY} = a * \exp\left(c * \frac{-Y}{P_R}\right) \quad (4.5)$$

where:

a and c: regression coefficients

$P_{FRAGILITY}$ : probability of exceeding

Y and  $P_R$  are defined in Eq. (4.4).

The results of the fragility analyses are shown in Figs. 4-10, 4-11, 4-12, 4-13 and 4-14. Note that the result for the probability of fragility in collapse prevention level with the criteria of 5% residual (permanent) drift is not presented since probabilities of the three models are all zero. In the structural performance of immediate occupancy, similar results are achieved in all three models as shown in Fig. 4-10, while as indicated in Fig. 4-11, the better results are obtained in two SCS models compared to those in the original model with the criteria of 0.7% residual drift. In the medium level (life safety), the similar or better performance are obtained in SCS models compared to those of original model as shown in Fig. 4-12 and 4-13. In collapse prevention level, as shown in Fig. 4-14, all 3 models can easily pass through the criteria of 5% transient drift, which indicates that no large transient drift occurred in the building after severe earthquakes. Since the better performance can be achieved in the SCS models based on the residual drift criteria, it is believed that SCS possesses the good ability to reduce or eliminate the residual drifts leading to save much in the cost of recovering the normal function of the hospital buildings.

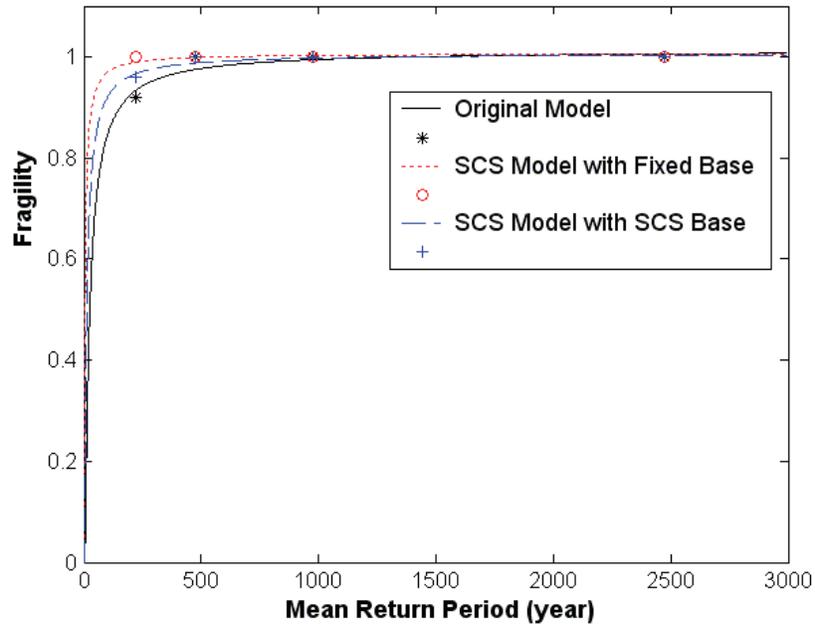


Figure 4-10 Fragility of Immediate Occupancy with 0.7% transient drift criteria

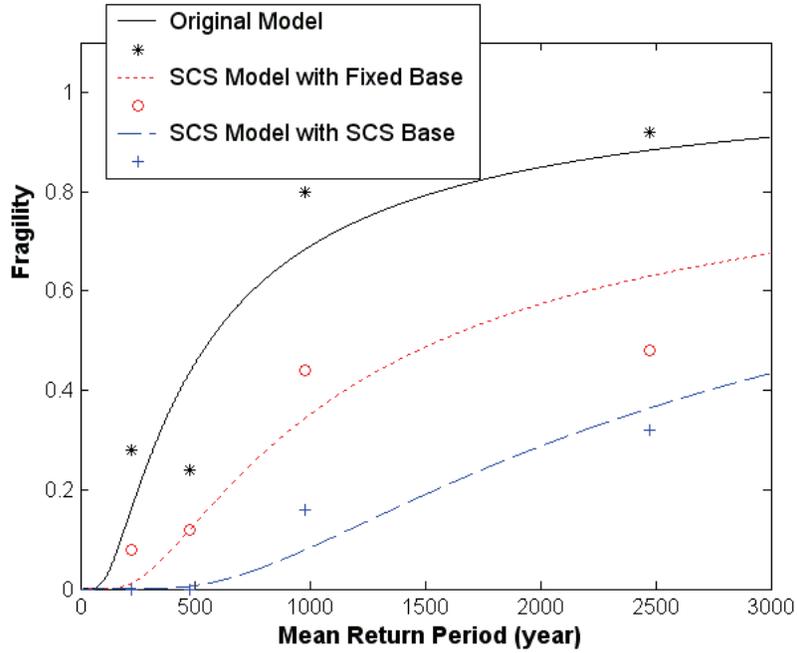


Figure 4-11 Fragility of Immediate Occupancy with 0.2% permanent drift criteria

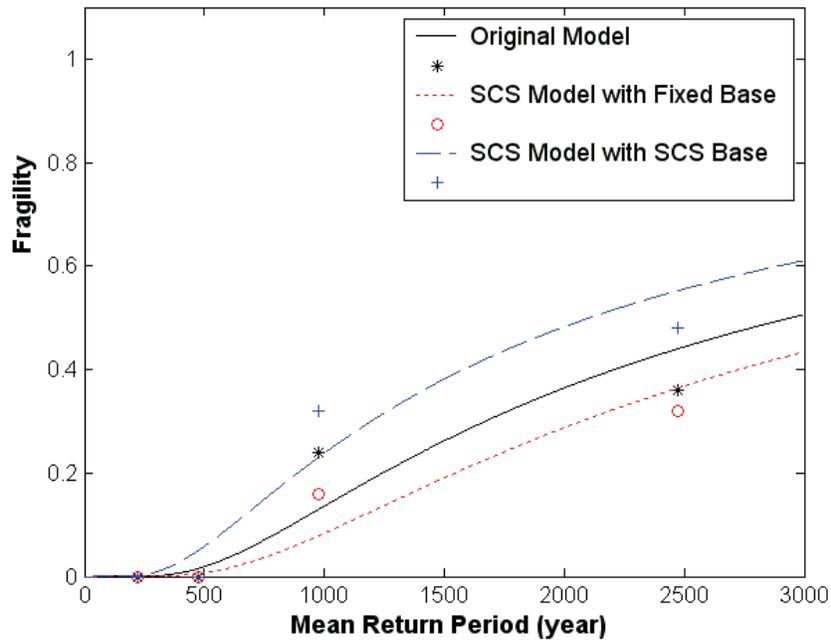


Figure 4-12 Fragility of Life Safety with 2.5% transient drift criteria

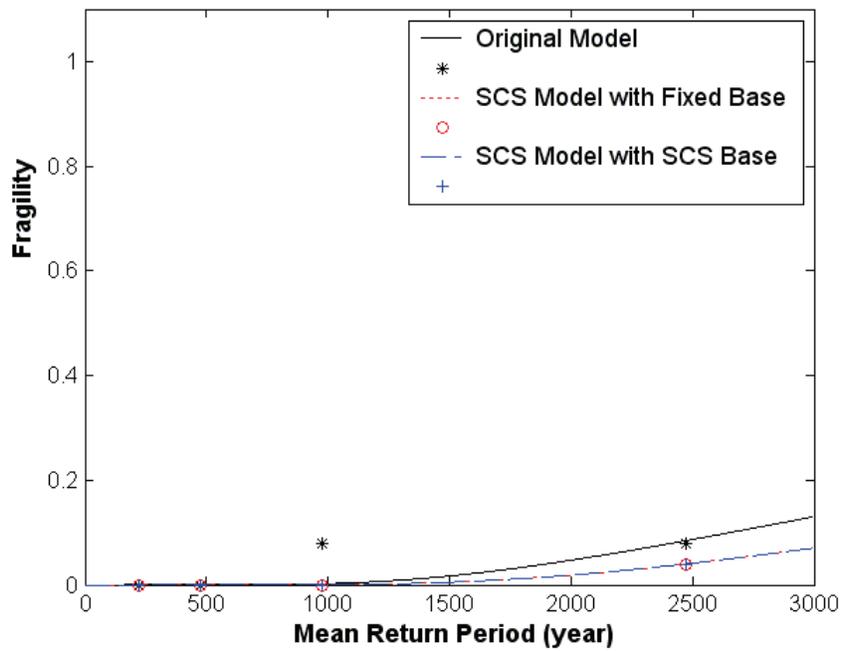


Figure 4-13 Fragility in Life Safety with 1% permanent drift criteria

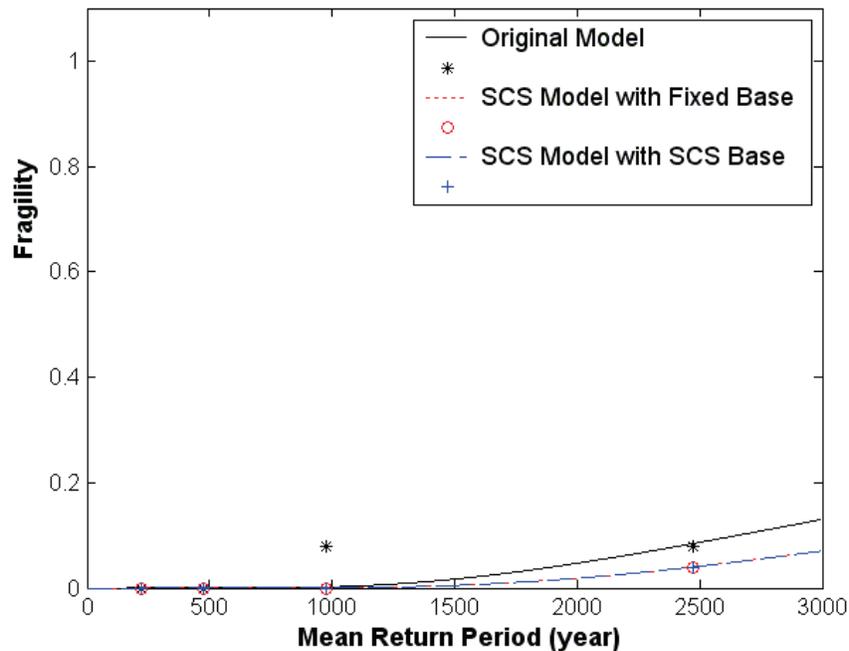


Figure 4-14 Fragility in Collapse Prevention with 5% transient drift criteria

#### 4.4 Practical Re-Design Procedure for SCS with SCPT connections in SMRF Buildings

In order to implement the SCS behavior, it is necessary to use practical SCS devices in real buildings. Such devices should possess good self-centering abilities, relatively low cost and easy installation and replacement. Based on these criteria, the Self-Centering Post-Tensioned (SCPT) connections (i.e. PTED connections as introduced in Chapter II) were used for the practical re-design of the WC70 building. In this section, the relationship between the moment-curvature property and SCS parameters is presented and then the re-design procedure of SCS with SCPT connections is developed.

**4.4.1 Relationship between Parameters of SCS and SCPT Connections**

Following the procedure presented in Section 4.2, the SCS parameters  $\alpha$ ,  $\beta$ ,  $\psi$  and  $\eta$  can be determined. Having these SCS parameters, the next step is to determine the physical properties of SCPT connections. As shown in Fig. 4-15, four parameters in the SCPT connections are required: the cross area of Energy Dissipating (ED) elements  $A_{ED}$ , the cross area of Post-Tensioned (PT) elements  $A_{PT}$ , the thickness of the Reinforcing Plates (R-Plate) welded to the outside of the upper and lower flanges of the beams  $t_R$  and the initial PT force in the PT bars  $F_{PTin}$ . These four parameters can not be directly determined only by the SCS parameters but need to be estimated based from the properties of the beam-to-column connections. Once the physical parameters of the SCPT connections are estimated, the moment-curvature relationship of the SCPT connections can be determined to check whether the SCS parameters of the SCPT connections are consistent with the estimated  $A_{ED}$ ,  $A_{PT}$ ,  $t_R$  and  $F_{PTin}$ . If a large difference exists between the original set of SCS parameters and the ones calculated from the estimated  $A_{ED}$ ,  $A_{PT}$ ,  $t_R$  and  $F_{PTin}$ , those four SCS parameters need to be adjusted and the iteration should be continued until similar SCS parameters are achieved.

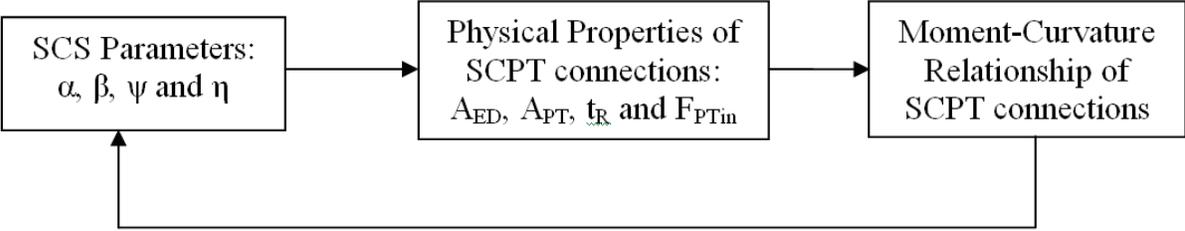


Figure 4-15 Relationship between parameters in SCS and SCPT connections

#### 4.4.1.1 First Estimate of Design Parameters ( $A_{ED}$ , $A_{PT}$ , $t_R$ and $F_{PTin}$ ) of SCPT Connections

Parametric studies showed that a first estimate for the initial PT force  $F_{PTin}$  can be taken as 20% of axial yielding force of beams. The stress in the PT elements should remain in the elastic range, therefore:

$$F_{PTmax} < F_{yPT} \quad (4.6)$$

where  $F_{yPT}$  is the yielding force of the PT elements and  $F_{PTmax}$  is the max force developed in the PT elements, which occurs when the gap-opening angle of SCPT connections reaches the maximum designed value (assumed to equal to 0.03 rad):

$$F_{PTmax} = F_{PTin} + A_{PT}(\Delta\epsilon_{PT})E \quad (4.7)$$

Where E is the Youngs' Modulus and  $\Delta\epsilon_{PT}$  is the increase in strain in the PT bars at the target gap-opening angle, which can be estimated by:

$$\Delta\epsilon_{PT} = \frac{\Delta L_{PT}}{L_{PT}} \approx \frac{n(\theta_{target}d_c)}{L_{PT}} \quad (4.8)$$

where  $d_c$  is the distance from the position of the PT elements to the neutral axial (approximately assumed to be equal to half of the depth of the beams),  $\theta_{target}$  is the target gap-opening angle,  $L_{PT}$  is the length of the PT elements and n is the number of gaps in the frame. Each gap opening will result in a similar strain increase in the PT elements.

The yielding force of the PT elements in Eq. (4.6) is given by:

$$F_{yPT} = A_{PT} f_{yPT} \quad (4.9)$$

where  $f_{yPT}$  is the yield strength of the PT elements.

From Eq. (4.6) to (4.9),  $A_{PT}$  should be meet the following criteria:

$$A_{PT} > \frac{F_{PTin}}{f_{yPT} - \frac{n(\theta_{target} d_c)}{L_{PT}} E} \quad (4.10)$$

As proposed by Christopoulos et al. (2002c), the energy dissipating factor  $\beta$  can be estimated by:

$$\beta \approx \frac{2F_{ED}(d_b - t_f)}{F_{PTin}(d_b / 2) + F_{ED}(d_b - t_f)} \quad (4.12)$$

where  $d_b$  is the depth of the beam,  $t_f$  is the thickness of the flange of beams and  $F_{ED}$  is the force developed in the ED elements at the target gap-opening angle, which can be estimated as 1.2 times the yielding force of the ED. The factor 1.2 is included to take into account the strain hardening effect in the ED bars. Therefore, the force in the ED elements is given by:

$$F_{ED} \approx 1.2 A_{ED} f_{yED} \quad (4.13)$$

where  $f_{yED}$  is the yield strength of ED bars. Substituting Eq (4.13) into Eq (4.12), a first estimate of the area of the ED elements  $A_{ED}$  can be obtained:

$$A_{ED} \approx \frac{F_{PTin}(d_b/2)}{\left(\frac{2}{\beta} - 1\right)1.2f_{yED}(d_b - t_f)} \quad (4.14)$$

R-plates are used to reinforce the flange of the beams in order to avoid compression yielding when the gaps open. For this purpose, high strength (100ksi yielding stress) steel are used for the R-plates. A first estimate of the thickness of R-plates  $t_R$  can be taken in the range from 50% to 100% of the thickness of beam flanges.

#### 4.4.1.2 Moment-Curvature Relationship of SCPT Connections

Since the designed parameters of SCPT connections are estimated, it is necessary to verify that the corresponding SCS parameters calculated for the SCPT connections with the designed parameters are in accordance with the originally selected SCS parameters. This verification is achieved by determining the moment-curvature relationship of each SCPT connection with the estimated SCS parameters.

As shown in Fig. 4-16, when the moment  $M$  applied to the connection reaches a critical value, the gap begins to open ( $\theta > 0$ ) and the contact point  $C$  in Fig. 4-16 (b), which moves from  $A$  to  $B$ , is assumed to be the neutral axis in the beam section. In order to determine the moment-curvature relationship, the force in the ED bars and the PT bars must be determined. The solution for these forces can be obtained for five distinct phases: (i) before gap opening ( $h_C = d_b + 2t_R$ ); (ii) point  $C$  is in the upper beam flange ( $d_b + 2t_R > h_C \geq d_b + t_R - t_f$ ); (iii) point  $C$  is between the upper flange and the upper half depth of the beam ( $d_b + t_R - t_f > h_C \geq d_b/2 + t_R$ ); (iv)

point C is between the lower half depth of the beam and the lower beam flange ( $d_b/2+t_R > h_C \geq t_R+t_f$ ); (v) point C in the lower beam flange ( $h_C < t_R+t_f$ ).

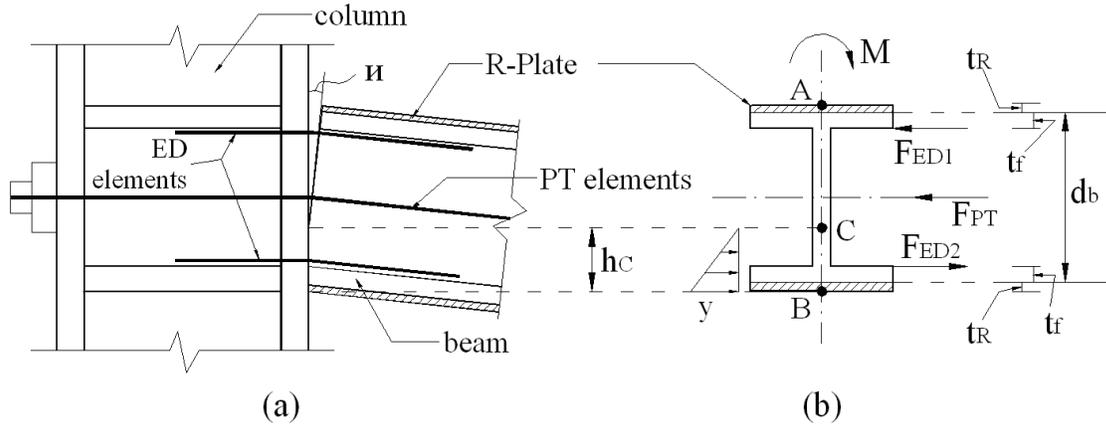


Figure 4-16 SCPT connection (a) beam-column connection (b) cross section of beam

**Phase (i)**  $h_C = d_b + 2t_R$

The beam section of the SCPT connection is in its initial position, as shown in Fig. 4-17(a). The parameters  $F_{ED1}$  and  $F_{ED2}$  are the upper and lower forces in the ED elements,  $F_{PT}$  is the force in the PT elements and  $\varepsilon_{b-in}$  is the initial strain in the beam section. Since the ED elements are installed after the installation and post-tension of the PT elements, there is no force in the ED bars. As shown in Fig. 4-17(b), the gap is beginning to open right after the critical moment is reached and the neutral point C will begin to move down. Through the horizontal force equilibrium in Fig. 4-17(a), the initial strain in the beam is given by:

$$\varepsilon_{b-in} = \frac{F_{PTin}}{E(A_b + 2A_R)} \quad (4.15)$$

where  $A_b$  is the area of beam section and  $A_R$  is the area of one R-plate section.

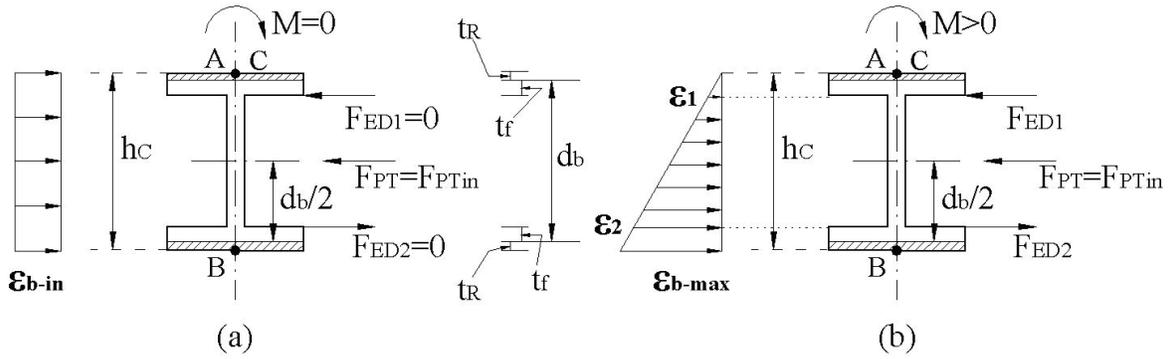


Figure 4-17 Onset of Gap-Opening, Phase i (a) beam cross section in initial position  
(b) beam cross section at the onset of gap-opening

From the Fig. 4-17 (a) to (b), the distribution of strains in the beam section is changed, which results in changes of the strains in the ED and PT elements without any gap-opening. Because the length of the PT elements is more than 10 to 20 times of the depth of the beam, which depends on the number of spans in one story, the change in the PT strain can be neglected and the force  $F_{PT}$  is assumed to be equal to the initial post-tensioned force  $F_{PTin}$ . For the same reason, as long as the neutral point C is higher than the position of the PT element (i.e.  $h_c > d_b/2 + t_r$ ),  $F_{PT}$  is assumed to be the same as the initial PT force. However, since the length of the ED elements is similar to the depth of the beam, the changes of strains in ED bars can not be neglected. The change in strains in the ED bars can be approximated to the change of normal strains in the beam section at the same height:

$$\epsilon_{ED1} = \epsilon_{b-in} - \epsilon_1 \quad (4.16)$$

$$\epsilon_{ED2} = \epsilon_2 - \epsilon_{b-in} \quad (4.17)$$

where  $\varepsilon_{ED1}$  and  $\varepsilon_{ED2}$  are the strains in the upper and lower ED elements;  $\varepsilon_1$  and  $\varepsilon_2$  are the strains in the beam section at the same height as the upper and lower ED bars as shown in Fig. 4-17(b) and can be defined by:

$$\varepsilon_1 = \frac{h_C - (d_b + t_R - t_f)}{h_C} \varepsilon_{b-\max} \quad (4.18)$$

$$\varepsilon_2 = \frac{h_C - (t_R + t_f)}{h_C} \varepsilon_{b-\max} \quad (4.19)$$

where  $h_C$  is the distance between the neural point C and the bottom of the lower R-plate,  $d_b$  is the depth of beam,  $t_R$  is the thickness of the R-plate,  $t_f$  is the thickness of the beam flange and  $\varepsilon_{b-\max}$  is the maximum strain in the beam section. Through horizontal force equilibrium,  $\varepsilon_{b-\max}$  is given by:

$$\varepsilon_{b-\max} = \frac{F_{PTin} + 2A_{ED}E\varepsilon_{b-in}}{\frac{E}{2} [2A_{ED} + h_C b_f - (b_f - t_w)(d_b - 2t_f)]} \quad (4.20)$$

where  $b_f$  is the width of beam flange and  $t_w$  is the thickness of beam web.

From Eq. (4.15) to (4.20), the force in the ED and PT elements can be determined and the critical gap opening moment can also be obtained by moment equilibrium in Fig. 4-17(b).

**Phase (ii)**  $d_b + 2t_R > h_C \geq d_b + t_R - t_f$

In this phase the gap begins to open and the neutral point C is located in the upper beam flange, as shown in Fig. 4-18. Due to the gap opening, there is no strain in the upper beam flange located higher than the neutral point C. The height of the neutral axis  $h_C$  is

undetermined, which is the major difference compared to Phase (i). Therefore,  $\varepsilon_{b-\max}$  can not be obtained from Eq. (4-20) but can be estimated by:

$$\varepsilon_{b-\max} = h_c \frac{\theta}{d_b + 2t_R} \quad (4.21)$$

where  $\theta$  is the gap-opening angle. More details on this approximation can be found in Pampanin et al. (2000).

With Eqs. (4.15) to (4.21), the position of the neutral axis  $h_c$  can be determined by horizontal force equilibrium in Fig. 4-18. The moment corresponding to the gap opening angle  $\theta$  can also be obtained by moment equilibrium.

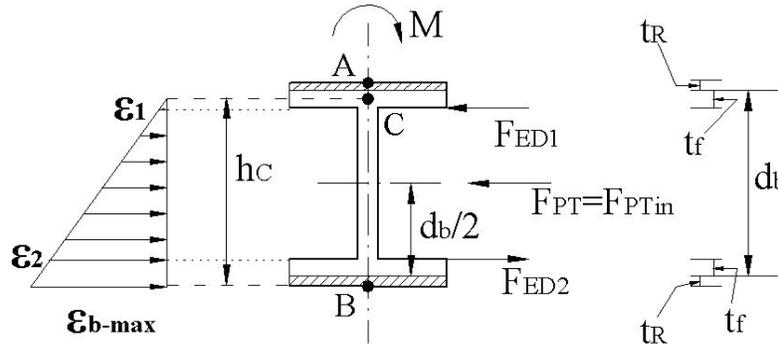


Figure 4-18 Beginning of Gap-Opening, Phase ii

**Phase (iii)**  $d_b + t_R - t_f > h_c \geq d_b/2 + t_R$

In this phase, the neutral axis is moving into the upper half of the beam web, as shown in Fig. 4-19. Comparing Fig. 4-19 with Fig. 4-18, the major difference is the disappearance of the strain  $\varepsilon_1$  in the upper part of the beam, which is due to the increasing of gap-opening angle.

Therefore the strain in the upper ED elements is given by:

$$\varepsilon_{ED1} = \frac{\theta(d_b - t_f - h_C)}{L_{ED}} + \varepsilon_{b-in} \quad (4.22)$$

where  $L_{ED}$  is the length of ED bars and assumed to be equal to the depth of beams and  $\varepsilon_{b-in}$  is defined in Eq. (4.15).

In Eq. (4.22), the first term represents the increased in strains due to the gap-opening and the second term indicates the effect of the initial post-tensioned force. The replacement of Eq.(4.16) by Eq. (4.22) is the only different step between Phase (ii) and (iii). Thereafter, the moment can be determined through moment equilibrium.

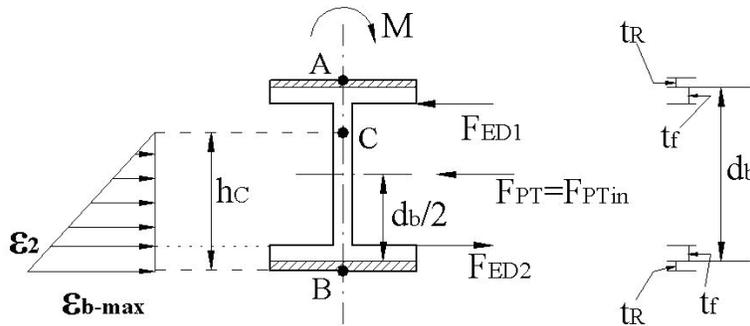


Figure 4-19 Gap-Opening Phase iii

**Phase (iv)**  $d_b/2 + t_R > h_C \geq t_R + t_f$

In this phase, the neutral axis moves under the PT elements, as shown in Fig. 4-20, which results in a relatively large increase in the PT force  $F_{PT}$ . In the previous three phases, the PT force did not change. Therefore, the strain in the PT elements  $\varepsilon_{PT}$  is given by:

$$\varepsilon_{PT} = \frac{\theta\left(\frac{d_b}{2} + t_R - h_C\right)}{L_{PT}} + \varepsilon_{PT-in} \quad (4.23)$$

where  $L_{PT}$  is the length of the PT elements and  $\varepsilon_{PT-in}$  is the initial post-tensioned strain.

With Eq. (4.23), the PT force  $F_{PT}$  can be obtained. The other steps to obtain the forces in the ED elements and moment  $M$  are the same as those in Phase (iii)..

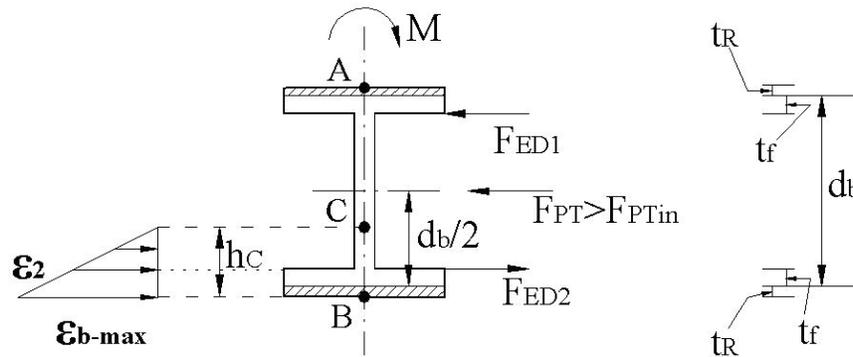


Figure 4-20 Gap-Opening, Phase iv

**Phase (v)**  $h_C < t_R + t_f$

As shown in Fig. 4-21, in this phase, the neutral point C moves down to the lower beam flange due to the large gap-opening angle, which should reach or exceed the target angle of 0.03rad. The lower ED elements transit from compression in the previous phases to tension and the strain in the lower ED elements  $\varepsilon_{ED2}$  is given by:

$$\varepsilon_{ED2} = \frac{\theta(t_R + t_f - h_C)}{L_{ED}} + \varepsilon_{b-in} \quad (4.24)$$

From Eqs. 4.15, 21, 22, 23 and 24, the height of the neutral axis can be determined through horizontal force equilibrium in Fig. 4-21 and the forces in the PT and ED elements can also be obtained. Again through moment equilibrium, the moment  $M$  can be determined.

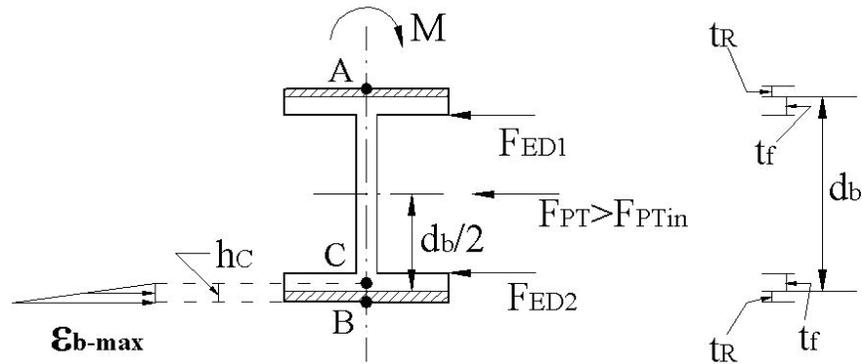


Figure 4-21 Gap-Opening, Phase v

For any specified gap-opening angle, the moment can be obtained in one of the five phases described above and the whole moment-curvature relationship can be established to verify whether the corresponding SCS parameters with the designed SCPT connections are similar to the SCS parameters. If there is an unacceptable difference between these two sets of SCS parameters, the physical parameters ( $A_{ED}$ ,  $A_{PT}$ ,  $t_R$  and  $F_{PTin}$ ) of the SCPT connections should be modified and the iteration needs to be continued until the verification of conformity with the SCS parameters. Note also that at the target gap-opening angle (i.e. maximum designed angle 0.03) the beam and R-plates should remain elastic, which is another criteria to modify the designed parameters of the SCPT connections.

Based on a series of numerical iterations on the physical parameters ( $A_{ED}$ ,  $A_{PT}$ ,  $t_R$  and  $F_{PTin}$ ) of SCPT connections, the following conclusions can be made:

- 1) It is difficult for the post-yielding factor  $\alpha$  to reach values larger than 0.2 by increasing the area of the ED elements  $A_{ED}$  or the area of the PT elements  $A_{PT}$ .

- 2) It is relatively easy to get the normal range (0.5 to 1) of the energy dissipating factor  $\beta$  by adjusting  $A_{ED}$ .
- 3) Increasing the initial PT force  $F_{PTin}$  can increase the strength ratio  $\eta$  and yielding ratio factor  $\psi$ , but at the same time it may increase the possibility of yielding in the beam flange or R-plate before or at the target gap-opening angle.
- 4) When  $A_{ED}$  is increasing, the factors  $\beta$ ,  $\psi$  and  $\eta$  are also increasing.
- 5) An increase of  $A_{PT}$  causes also an increase of the post-yielding factor  $\alpha$ .
- 6) Increasing the thickness of the R-plates  $t_R$  can reduce the yielding in the beam or the R-plates.

When a iteration process for adjusting the physical parameters of the SCPT connections is conducted, the above conclusions can be used as a guideline to accelerate the iteration process.

#### **4.4.2 Practical Re-Design Procedure of SMRF Buildings with SCPT Connections**

There are three distinct steps in the practical re-design procedure of SMRF buildings with SCPT connections: preliminary design, intermediate design and final design. In the preliminary design, all the SCS parameters ( $\alpha$ ,  $\beta$ ,  $\eta$  and  $\psi$ ) and physical parameters ( $A_{ED}$ ,  $A_{PT}$ ,  $t_R$  and  $F_{PTin}$ ) of the SCPT connections are first determined. Thereafter, in the intermediate design, the physical parameters ( $A_{ED}$ ,  $A_{PT}$ ,  $t_R$  and  $F_{PTin}$ ) are modified in the same story and same joint. Finally, in the final design, the sections of R-plates, ED and PT elements are selected according to the modified parameters and available steel products.

#### 4.4.2.1 Preliminary Design

The preliminary design phase described can be summarized as followed:

- i) Determine the desirable SCS parameters ( $\alpha$ ,  $\beta$ ,  $\eta$  and  $\psi$ ) for the SMRF building by implementing the redesign procedure proposed in Chapter 4.2.
- ii) Provide a first estimate of the physical parameters ( $A_{ED}$ ,  $A_{PT}$ ,  $t_R$  and  $F_{PTin}$ ) to each beam-column joint in the whole buildings according to Section 4.4.1.1.
- iii) Obtain the moment-curvature relationship of each joint to verify the desirable SCS parameters according to Section 4.4.1.2.
- iv) Iterate on the values of the physical parameters until verification of the target SCS parameters according to Section 4.4.1.
- v) Finally, form a preliminary set of physical parameters ( $A_{ED}$ ,  $A_{PT}$ ,  $t_R$  and  $F_{PTin}$ ) for each beam-column joint.

#### 4.4.2.2 Intermediate Design

In a real building, it is common that different beam sections exist in adjacent spans of a given story. However, following the preliminary design procedure, the different sets of physical SCPT parameters are selected for different beam sections. Therefore, a conflict arises because in one frame of a given story, the area of the PT elements and the initial PT force should be the same and so should the area of the ED elements in a joint connected to different beam sections. Therefore, the aim of the intermediate design step is to resolve this

conflict. The approach is to modify the parameters to an interpolated value so that the performance of the whole building approaches that of the ideal SCS frame with the target SCS parameters, although for each single joint the SCS property will deviate from the target one. The procedure is summarized as follows:

i) Modify the initial PT force  $F_{PTin}$  in the same frame of the same story.

for  $(F_{PTin})_{\max} > \frac{3}{2}(F_{PTin})_{\min}$

$$(F_{PTin})_M = (F_{PTin})_{\min} + \frac{1}{4}[(F_{PTin})_{\max} - (F_{PTin})_{\min}] \quad (4.25)$$

for  $(F_{PTin})_{\max} \leq \frac{3}{2}(F_{PTin})_{\min}$

$$(F_{PTin})_M = (F_{PTin})_{\min} + \frac{1}{3}[(F_{PTin})_{\max} - (F_{PTin})_{\min}] \quad (4.26)$$

where  $(F_{PTin})_{\max}$  and  $(F_{PTin})_{\min}$  are the maximum and minimum of values of the initial PT force of one frame in the same story, respectively;  $(F_{PTin})_M$  is the modified initial PT force. If there is a large difference between the maximum and minimum initial PT force, the modified one is set closer (1/4) to the minimum one in Eq. (4.25) than that (1/3) in Eq. (4.26). The criteria is to set a relatively low interpolated value since the large value of the initial PT force may result in yielding in the small beam sections at the target gap-opening angle.

ii) Modify the area of the PT elements  $A_{PT}$  in the same story.

$$(A_{PT})_M = (A_{PT})_{\min} + \frac{2}{3}[(A_{PT})_{\max} - (A_{PT})_{\min}] \quad (4.27)$$

where  $(A_{PT})_{max}$  and  $(A_{PT})_{min}$  are the maximum and minimum of values of the area of PT elements in one frame of the same story, respectively;  $(A_{PT})_M$  is the modified area of the PT elements. The modified area is selected to be closer to the larger value ( $(A_{PT})_{max}$ ), which represents the safety design idea.

iii) Re-calculate the area of the ED elements and the thickness of the R-plates for each joint according to Section 4.4.1.

iv) Finally, a new set of physical SCPT parameters is developed for each beam-column connection.

#### **4.4.2.3 Final Design**

Based on the modified SCPT parameters from the intermediate design step, the selection of real steel products is conducted in the final design.

i) Selection for PT elements.

Considering cost, the PT elements should be selected with readily used steel products such as the product manufactured by DYWIDAG Systems International (DSI). DYWIDAG mono-strand tendons are typically made from cold-drawn, low relaxation 7-wire strand conforming to ASTM A416, gr270 [1860MPa] and there are two sizes available: 0.5 and 0.6 in diameter, which are suitable for PT elements. High strength DSI bars can also be used as PT elements.

ii) Selection of ED elements

The ED elements can also be selected from DSI Thread Bar products, which conform to ASTM A615 (Grade 60 & 75) CAN/CSA(G3018-M1982) with 60 or 75 ksi yielding stress. The designation of DSI product is including #6, #7, #8, #9, #10, #11, #14 and #18, which has different cross section area. For a design value of  $A_{ED}$ , the DSI product can be machined down to the corresponding area.

iii) Selection of R-plates

R-plates are selected with high yield stress steel products conforming to A514 with 90 or 100 ksi yield stress. The size can be selected from Table 1-19 in LRFD 3<sup>rd</sup> Edition (LRFD, 2001).

#### **4.5 Re-Design and Numerical Modeling of WC70 Hospital Building with SCPT connections**

The re-design of the WC70 hospital building incorporating SCPT connections was conducted by following the practical re-design procedure given in Section 4.4.2. The SCPT connections were designed for each beam-column joint without changing the properties of the base column connection. The results of the preliminary design, intermediate design and final design are shown in Table 4-10, 4-11 and 4-12, respectively. To improve the performance of the Ideal SCS hospital model with  $\alpha=0.2$ ,  $\beta=0.6$  and  $\psi = \eta_{SCS} / \eta_{Original}=0.6$ , these SCS parameters were modified as a target set:  $\alpha=0.05$ ,  $\beta=0.8$  and  $\psi = \eta_{SCS} / \eta_{Original}=0.5$  as shown in Table 4-10. It is noted that in these three tables, the maximum stresses in the beam flanges and R-plates are listed to verify that the beams and R-plates remain elastic.

Table 4-10 Preliminary Design for WC70 Hospital with SCPT Connections

\* Target SCS parameters:  $\alpha=0.05$ ;  $\beta=0.8$ ;  $\psi=\eta_{SCS}/\eta_{Original}=0.5$ ;

	Beam Section	Initial PT force $F_{PTin}$ (kips)	Area of PT bar $A_{PT}$ (in <sup>2</sup> )	Area of 2 ED bars $A_{ED}$ (in <sup>2</sup> )	Thickness of R plate $t_R$ (in)	$\alpha$	$\beta$	$\psi$ ( $\eta_{SCS}/\eta_{Original}$ )	Max Stress of R plate (ksi)	Max Stress of beam Flange (ksi)
1 <sup>st</sup> floor (interior)	w33x241	759.70	7.43	3.01	0.98	0.04	0.80	0.50	78.97	49.72
	w24x162	505.62	5.93	2.04	0.98	0.05	0.80	0.50	85.83	48.05
1 <sup>st</sup> floor (exterior)	w33x221	700.90	6.72	2.76	0.89	0.04	0.80	0.50	76.78	49.61
	w24x146	462.25	5.17	1.85	0.82	0.04	0.80	0.50	82.82	49.91
2 <sup>nd</sup> floor (interior)	w30x211	681.09	5.87	2.70	0.86	0.03	0.80	0.50	78.32	49.60
	w24x146	462.25	5.17	1.85	0.82	0.04	0.80	0.50	82.82	49.91
2 <sup>nd</sup> floor (exterior)	w30x211	681.09	5.87	2.70	0.86	0.03	0.80	0.50	78.32	49.60
	w24x131	413.88	4.53	1.64	0.72	0.04	0.80	0.50	79.09	49.34
3 <sup>rd</sup> floor (interior)	w30x211	681.09	5.87	2.70	0.86	0.03	0.80	0.50	78.32	49.60
	w24x146	462.25	5.17	1.85	0.82	0.04	0.80	0.50	82.82	49.91
3 <sup>rd</sup> floor (exterior)	w30x211	681.09	5.87	2.70	0.86	0.03	0.80	0.50	78.32	49.60
	w24x103	307.55	3.61	1.22	0.83	0.04	0.80	0.50	81.52	48.08
4 <sup>th</sup> floor (interior)	w30x173	561.00	4.70	2.20	0.70	0.03	0.80	0.50	72.79	48.37
	w24x94	282.54	3.22	1.11	0.70	0.04	0.80	0.50	78.78	49.79
4 <sup>th</sup> floor (exterior)	w24x104	341.19	2.38	1.33	0.45	0.02	0.80	0.50	67.00	46.19
	w24x68	202.01	2.22	0.78	0.50	0.04	0.80	0.50	70.51	48.52

Table 4-11 Intermediate Design for WC70 Hospital with SCPT Connections

\* Target SCS parameters:  $\alpha=0.05$ ;  $\beta=0.8$ ;  $\psi = \eta_{SCS} / \eta_{Original}=0.5$ ;

	Position	Beam Section	Initial PT force $F_{PTin}$ (kips)	Area of PT bar $A_{PT}$ (in <sup>2</sup> )	Area of 2 ED bars $A_{ED}$ (in <sup>2</sup> )	Thickness of R plate $t_R$ (in)	$\alpha$	$\beta$	$\psi$ ( $\eta_{SCS} / \eta_{Original}$ )	Max Stress of R plate (ksi)	Max Stress of beam Flange (ksi)
1 <sup>st</sup> floor (interior)	End	w24x162	590.31	6.93	2.39	1.10	0.05	0.80	0.59	92.30	49.78
	Internal	w33x241									
	Joint	w24x162									
1 <sup>st</sup> floor (exterior)	End	w24x146	541.80	6.20	2.18	1.04	0.05	0.80	0.59	89.20	48.10
	Internal	w33x221									
	Joint	w24x146									
2 <sup>nd</sup> floor (interior)	End	w24x146	535.20	5.64	2.15	0.98	0.05	0.80	0.58	87.80	48.51
	Internal	w30x211									
	Joint	w24x146									
2 <sup>nd</sup> floor (exterior)	End	w24x131	502.95	5.42	2.01	0.91	0.05	0.80	0.61	86.30	48.86
	Internal	w30x211									
	Joint	w24x131									
3 <sup>rd</sup> floor (interior)	End	w24x146	535.20	5.64	2.15	0.98	0.05	0.80	0.58	87.80	48.51
	Internal	w30x211									
	Joint	w24x146									
3 <sup>rd</sup> floor (exterior)	End	w24x103	400.94	5.12	1.61	1.13	0.06	0.80	0.66	93.09	48.50
	Internal	w30x211									
	Joint	w24x103									
4 <sup>th</sup> floor (interior)	End	w24x94	352.16	4.21	1.40	0.92	0.05	0.80	0.63	87.66	49.98
	Internal	w30x173									
	Joint	w24x94									
4 <sup>th</sup> floor (exterior)	End	w24x68	248.40	2.33	0.97	0.59	0.04	0.80	0.61	76.05	49.75
	Internal	w24x104									
	Joint	w24x68									

Table 4-12 Final Design for WC70 Hospital with SCPT Connections

\* Target SCS parameters:  $\alpha=0.05$ ;  $\beta=0.8$ ;  $\psi=\eta_{SCS}/\eta_{Original}=0.5$ ;

	Position	Beam Section	Initial PT force $F_{PTin}$ (kips)	Area of PT bar $A_{PT}$ (in <sup>2</sup> )	Area of 2 ED bars $A_{ED}$ (in <sup>2</sup> )	Thickness of R plate $t_R$ (in)	$\alpha$	$\beta$	$\psi$ ( $\eta_{SCS}/\eta_{Original}$ )	Max Stress of R plate (ksi)	Max Stress of beam Flange (ksi)
1 <sup>st</sup> floor (interior)	internal	w33x241	590	32@0.6in. 6.944	#10 @1.234 in. 2.39	0.875	0.033	0.809	0.38	71.53	45.99
	external	w24x162									
1 <sup>st</sup> floor (exterior)	internal	w33x221	540	4@0.5 in. 26@0.6 in. 6.254	#10 @1.178 in. 2.18	0.75	0.032	0.811	0.37	69.35	46.88
	external	w24x146									
2 <sup>nd</sup> floor (interior)	internal	w30x211	535	26@0.6in. 5.642	#10 @1.170 in. 2.15	0.875	0.029	0.804	0.38	70.87	42.63
	external	w24x146									
2 <sup>nd</sup> floor (exterior)	internal	w30x211	500	36@0.5in. 5.508	#9 @1.128 in. 2.00	0.875	0.028	0.801	0.36	68.87	40.86
	external	w24x131									
3 <sup>rd</sup> floor (interior)	internal	w30x211	535	26@0.6in. 5.642	#10 @1.170 in. 2.15	0.875	0.029	0.804	0.38	70.87	42.63
	external	w24x146									
3 <sup>rd</sup> floor (exterior)	internal	w30x211	400	24@0.6in. 5.208	#9 @1.012 in. 1.61	0.875	0.026	0.805	0.29	63.17	35.88
	external	w24x103									
4 <sup>th</sup> floor (interior)	internal	w30x173	350	28@0.5in. 4.284	#8 @0.944 in. 1.40	0.625	0.025	0.809	0.30	59.80	39.07
	external	w24x94									
4 <sup>th</sup> floor (exterior)	internal	w24x104	250	16@0.5in. 2.448	#7 @0.786 in. 0.97	0.5	0.020	0.797	0.36	57.88	36.13
	external	w24x68									

A numerical model of the WC70 Hospital Building with SCPT connections was developed with the RUAUMOKO software (Carr 2004), as shown in Fig. 4-22. The gap-opening characteristic is modeled by a multi-spring gap element, which yields with zero tension force and remains rigid in compression. Each beam-to-column joint in the WC70 moment-resistant frame was modified with SCPT connections. The effect of the R-plates was added by increasing the beam section between the rigid bar and gap element. In the internal beam-column joint, the ED bars were connected to the beams but not to the columns.

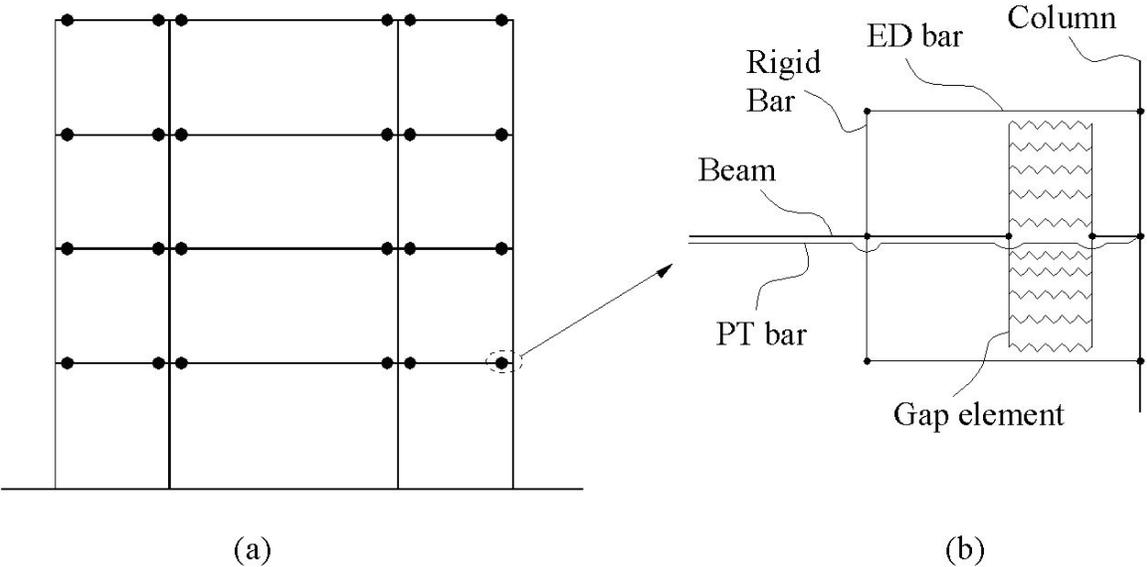


Figure 4-22 Numerical Model of WC70 Hospital Building with SCPT Connections  
(a) WC70 Frame (b) Detail of External SCPT Connections

### 4.6 Performance of SCS WC70 Hospital Model with SCPT Connections

In this section, the seismic performances of the SCS Hospital Model equipped with SCPT connections is investigated through Cyclic nonlinear quasi-static analyses, seismic analyses and fragility analyses. The calculation methods used for these analyses are same as those in Section 4.3. Therefore, the methods are not discussed in the following presentation.

#### 4.6.1 Cyclic nonlinear quasi-static Analyses

The result of the Cyclic nonlinear quasi-static analyses is shown in Fig. 4-23. In this figures, it is observed that the initial stiffness of the SCS model equipped with SCPT connections does not change, while the yield strength is reduced and the post-yielding stiffness is a slightly higher than that of the original model. Those observations validate the design procedure. The possible maximum residual displacement (point B in Fig. 4-23) in the SCS

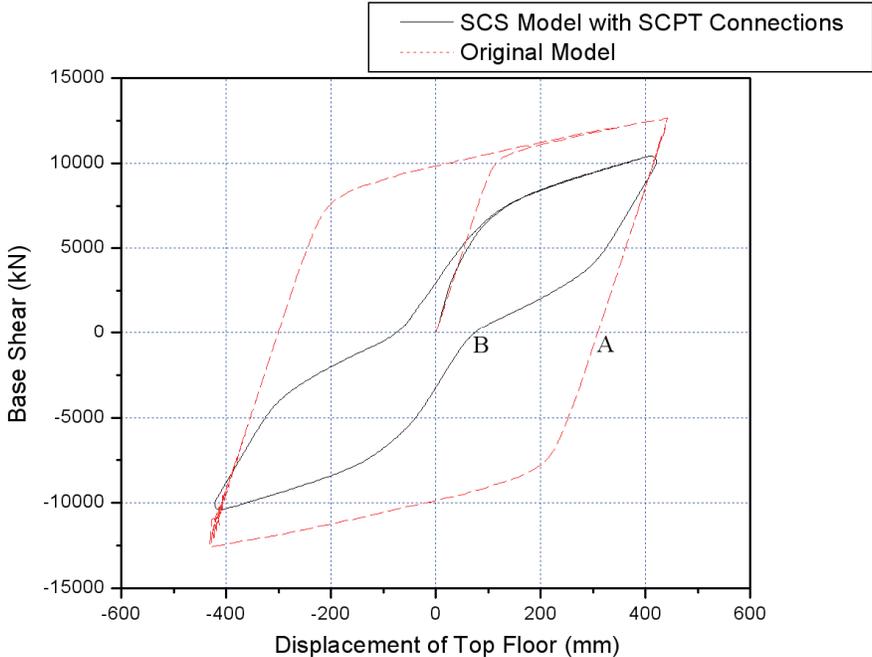


Figure 4-23 Push-Over Analysis for two models

model is reduced compared to that (point A) in the original model, which shows the good self-centering ability. The reason for non-zero residual drift is that column connections did not incorporate SCPT connections with self-centering properties.

#### **4.6.2 Seismic Analyses**

The seismic analyses were conducted under the same MCEER ground motion as mentioned in Section 4.3. In order to compare the performance of the different numerical models, the results obtained from the SCS model equipped with SCPT connections along with those of the original and ideal SCS models are listed in Table 4-13 to 4-16. The SCS model with SCPT connections is termed herein as new model.

From these tables, it is observed that the displacements and residual drifts of the new model are reduced, which shows that the performance of new model exceeds not only that of the original model but also that of the ideal SCS model. The acceleration responses of the new model are reduced by 15% to 30% compared to those of the original model, which largely exceeds the performance of the ideal SCS model. The ductility in the columns in new model is also reduced compared to the reduction in the ideal model, which indicates that the damage to the columns is reduced in the new model. All of the above observations show that the new model incorporating adjusted SCS parameters with SCPT connections can achieve much better performance than the previous ideal model, reduce the displacement responses and the acceleration responses at the same time and decrease or diminish the residual drifts, all of which shows the new re-designed SCS model with SCPT connections has a complete ability to improve the seismic performance.

**Table 4-13 Mean Values of response under 25 MCEER ground motions having 20% probability of exceedance in 50 years**

Mean Values	unit	Original Model		SCS Model with SCPT connections (New Model) $\alpha = 0.05, \beta = 0.8$ $\psi = \eta_{SCS} / \eta_{Original} = 0.5$		Ideal SCS Model with fixed base $\alpha = 0.2, \beta = 0.6$ $\psi = \eta_{SCS} / \eta_{Original} = 0.6$	
		mm	%	mm	%	mm	%
<b>Maximum Displacement</b>	top floor	130.14	0.84	112.64	0.72	136.63	0.88
<b>Residual Displacement</b>	top floor	12.48	0.08	1.49	0.01	3.00	0.02
<b>Maximum Interstory Drift</b>	1 <sup>st</sup> floor	40.21	0.98	30.99	0.75	40.22	0.98
	2 <sup>nd</sup>	40.90	1.07*	36.45	0.96*	42.72	1.12*
	3 <sup>rd</sup>	36.06	0.95	30.26	0.79	38.63	1.01
	4 <sup>th</sup>	25.30	0.66	18.61	0.49	24.91	0.65
<b>Residual Interstory Drift</b>	1 <sup>st</sup>	5.06	0.12*	1.30	0.03*	2.53	0.06*
	2 <sup>nd</sup>	4.22	0.11	0.15	0.00	0.37	0.01
	3 <sup>rd</sup>	2.82	0.07	0.04	0.00	0.22	0.01
	4 <sup>th</sup>	0.52	0.01	0.02	0.00	0.02	0.00
<b>Acceleration Unit</b>		g		g		g	
<b>Maximum Absolute Acceleration</b>	1 <sup>st</sup>	0.68		0.55		0.64	
	2 <sup>nd</sup>	0.74		0.56		0.70	
	3 <sup>rd</sup>	0.79		0.60		0.81	
	4 <sup>th</sup>	1.06*		0.75*		0.93*	
<b>Maximum Ductility of Beams</b>		2.90		1.0**		1.0**	
<b>Maximum Ductility of Columns</b>		2.03		0.78		1.90	

\* Shaded values represent maximum and residual interstory drifts and maximum absolute accelerations.

\*\*No yielding in beams of SCS Models.

**Table 4-14 Mean Values of response under 25 MCEER ground motions having 10% probability of exceedance in 50 years**

Mean Values	unit	Original Model		SCS Model with SCPT connections (New Model) $\alpha = 0.05, \beta = 0.8$ $\psi = \eta_{SCS} / \eta_{Original} = 0.5$		Ideal SCS Model with fixed base $\alpha = 0.2, \beta = 0.6$ $\psi = \eta_{SCS} / \eta_{Original} = 0.6$	
		mm	%	mm	%	mm	%
<b>Maximum Displacement</b>	top floor	143.38	0.92	133.81	0.86	156.99	1.01
<b>Residual Displacement</b>	top floor	13.46	0.09	2.44	0.02	4.37	0.03
<b>Maximum Interstory Drift</b>	1 <sup>st</sup> floor	45.63	1.11	37.55	0.91	48.92	1.19
	2 <sup>nd</sup>	45.03	1.18*	42.76	1.12*	48.36	1.27*
	3 <sup>rd</sup>	41.10	1.08	35.81	0.94	42.86	1.12
	4 <sup>th</sup>	29.86	0.78	21.50	0.56	29.41	0.77
<b>Residual Interstory Drift</b>	1 <sup>st</sup>	5.27	0.13*	2.12	0.05*	3.65	0.09*
	2 <sup>nd</sup>	4.79	0.13	0.24	0.01	0.74	0.02
	3 <sup>rd</sup>	3.60	0.09	0.08	0.00	0.29	0.01
	4 <sup>th</sup>	0.91	0.02	0.03	0.00	0.03	0.00
<b>Acceleration Unit</b>		g		g		g	
<b>Maximum Absolute Acceleration</b>	1 <sup>st</sup>	0.83		0.64		0.83	
	2 <sup>nd</sup>	0.88		0.67		0.85	
	3 <sup>rd</sup>	0.88		0.67		0.88	
	4 <sup>th</sup>	1.23*		0.85*		1.07*	
<b>Maximum Ductility of Beams</b>		3.36		1.0**		1.0**	
<b>Maximum Ductility of Columns</b>		2.81		1.18		2.78	

\* Shaded values represent maximum and residual interstory drifts and maximum absolute accelerations.

\*\*No yielding in beams of SCS Models.

**Table 4-15 Mean Values of response under 25 MCEER ground motions having 5% probability of exceedance in 50 years**

Mean Values	unit	Original Model		SCS Model with SCPT connections (New Model) $\alpha = 0.05, \beta = 0.8$ $\psi = \eta_{SCS} / \eta_{Original} = 0.5$		Ideal SCS Model with fixed base $\alpha = 0.2, \beta = 0.6$ $\psi = \eta_{SCS} / \eta_{Original} = 0.6$	
		mm	%	mm	%	mm	%
<b>Maximum Displacement</b>	top floor	205.46	1.32	189.14	1.22	223.38	1.44
<b>Residual Displacement</b>	top floor	42.47	0.27	6.17	0.04	12.04	0.08
<b>Maximum Interstory Drift</b>	1 <sup>st</sup> floor	73.52	1.78*	59.35	1.44	79.10	1.92*
	2 <sup>nd</sup>	67.14	1.76	58.71	1.54*	68.40	1.80
	3 <sup>rd</sup>	53.66	1.41	47.70	1.25	58.28	1.53
	4 <sup>th</sup>	34.55	0.91	29.10	0.76	40.10	1.05
<b>Residual Interstory Drift</b>	1 <sup>st</sup>	16.34	0.40*	4.73	0.11*	8.35	0.20*
	2 <sup>nd</sup>	14.15	0.37	0.88	0.02	3.29	0.09
	3 <sup>rd</sup>	10.30	0.27	0.68	0.02	0.98	0.03
	4 <sup>th</sup>	2.56	0.07	0.04	0.00	0.11	0.00
<b>Acceleration Unit</b>		g		g		g	
<b>Maximum Absolute Acceleration</b>	1 <sup>st</sup>	1.05		0.76		0.93	
	2 <sup>nd</sup>	1.01		0.76		0.94	
	3 <sup>rd</sup>	1.04		0.82		1.00	
	4 <sup>th</sup>	1.41*		1.06*		1.27*	
<b>Maximum Ductility of Beams</b>		5.42		1.0**		1.0**	
<b>Maximum Ductility of Columns</b>		5.23		3.69		5.83	

\* Shaded values represent maximum and residual interstory drifts and maximum absolute accelerations.

\*\*No yielding in beams of SCS Models.

**Table 4-16 Mean Values of response under 25 MCEER ground motions having 2% probability of exceedance in 50 years**

Mean Values	unit	Original Model		SCS Model with SCPT connections (New Model) $\alpha = 0.05, \beta = 0.8$ $\psi = \eta_{SCS} / \eta_{Original} = 0.5$		Ideal SCS Model with fixed base $\alpha = 0.2, \beta = 0.6$ $\psi = \eta_{SCS} / \eta_{Original} = 0.6$	
		mm	%	mm	%	mm	%
<b>Maximum Displacement</b>	top floor	235.06	1.51	208.47	1.34	246.25	1.58
<b>Residual Displacement</b>	top floor	51.98	0.33	8.22	0.05	17.00	0.11
<b>Maximum Interstory Drift</b>	1 <sup>st</sup> floor	85.57	2.08*	71.30	1.73*	92.72	2.25
	2 <sup>nd</sup>	77.25	2.03	64.31	1.69	75.02	1.97
	3 <sup>rd</sup>	62.34	1.64	50.93	1.34	61.20	1.61
	4 <sup>th</sup>	38.72	1.02	31.94	0.84	45.93	1.21
<b>Residual Interstory Drift</b>	1 <sup>st</sup>	19.87	0.48*	7.04	0.17*	12.22	0.30*
	2 <sup>nd</sup>	16.41	0.43	0.99	0.03	4.07	0.11
	3 <sup>rd</sup>	11.72	0.31	0.48	0.01	1.40	0.04
	4 <sup>th</sup>	4.28	0.11	0.05	0.00	0.51	0.01
<b>Acceleration Unit</b>		g		g		g	
<b>Maximum Absolute Acceleration</b>	1 <sup>st</sup>	1.18		0.95		1.10	
	2 <sup>nd</sup>	1.13		0.87		1.14	
	3 <sup>rd</sup>	1.15		0.89		1.09	
	4 <sup>th</sup>	1.47*		1.20*		1.41*	
<b>Maximum Ductility of Beams</b>		6.17		1.0**		1.0**	
<b>Maximum Ductility of Columns</b>		6.38		4.86		7.12	

\* Shaded values represent maximum and residual interstory drifts and maximum absolute accelerations.

\*\*No yielding in beams of SCS Models.

### 4.6.3 Seismic Fragility Analysis

As shown in Figs. 4-24 to 4-27, it is observed that for each performance level, the fragility of the SCS model equipped with SCPT connections is lower than that of the original model and ideal SCS model, especially with respect to permanent drifts (residual drifts). It is again apparent that the new model with the adjusted set of SCS parameters incorporating SCPT connections achieved a better performance than those of the original model and of the ideal SCS model.

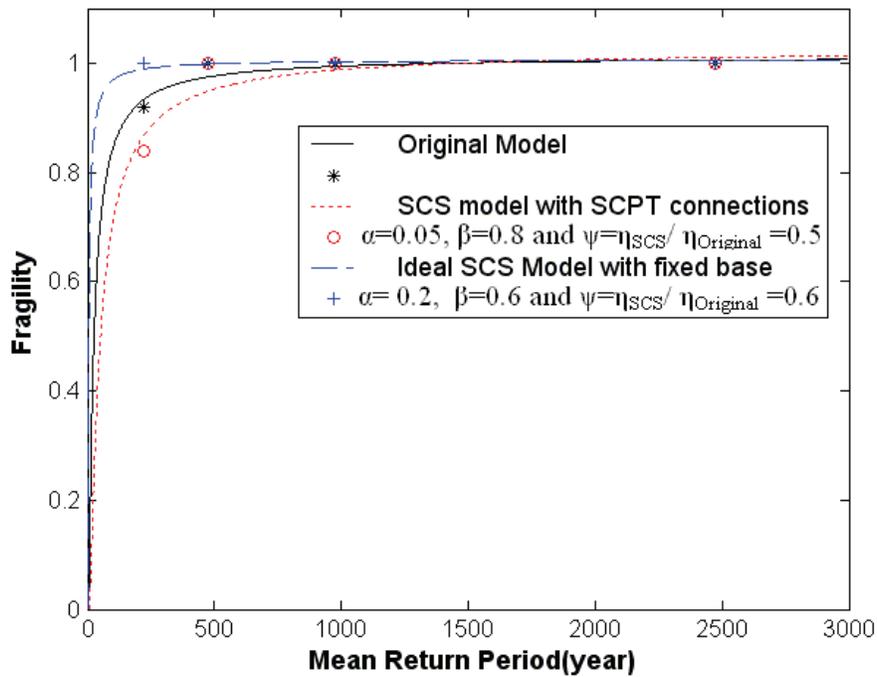


Figure 4-24 Fragility of Immediate Occupancy with the criteria of 0.7% transient drift

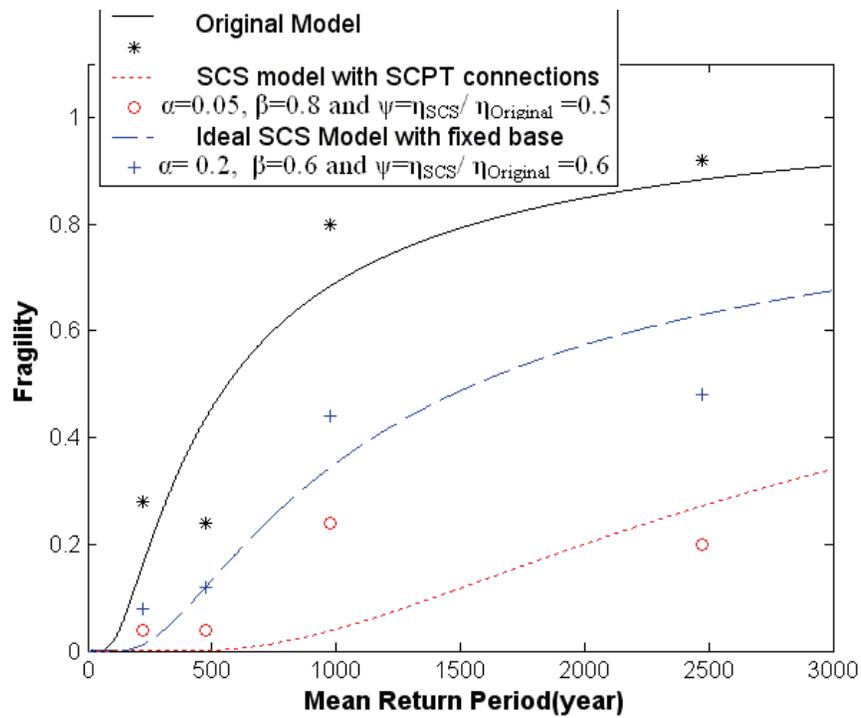


Figure 4-25 Fragility of Immediate Occupancy with the criteria of 0.2% permanent drift

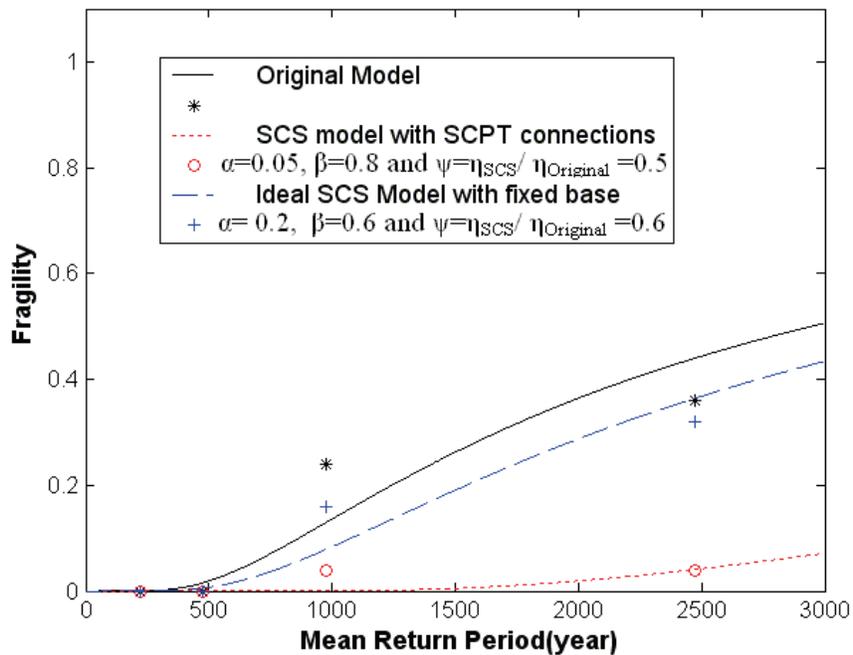


Figure 4-26 Fragility of Life Safety with the criteria of 2.5% transient drift

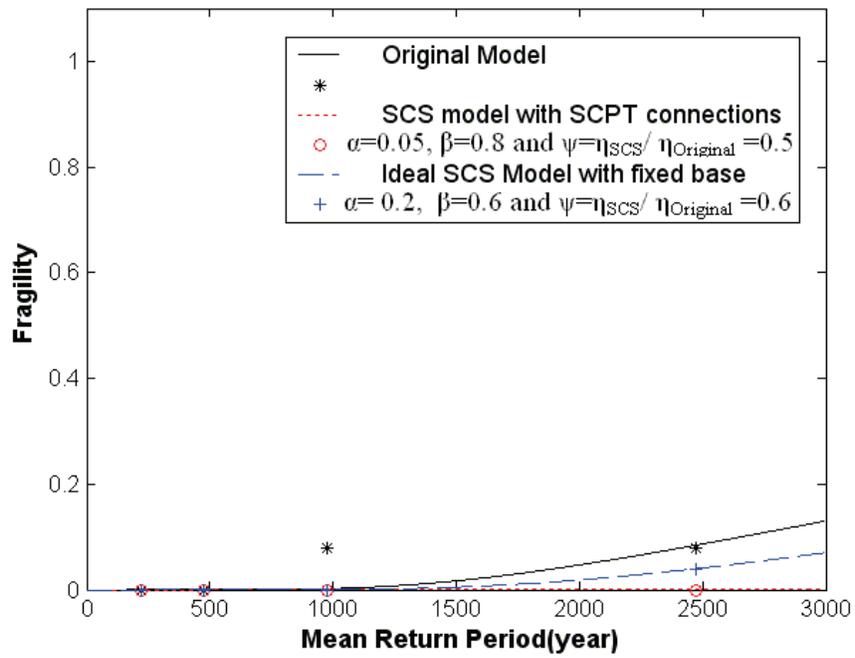


Figure 4-27 Fragility of Life Safety with the criteria of 1% permanent drift



## **CHAPTER 5**

### **PROTOTYPES AND MODELS FOR SHAKE TABLE TESTING**

The numerical research on the application of Self-Centering Systems (SCS) to steel framed structures was presented in previous parts of this report. From this chapter to the end of the report, the experimental testing of two steel frame specimens is reported.

To the knowledge of the authors, the experimental research of SCS implemented in an entire Steel Moment Resisting Frame (SMRF) has never been conducted. All of the previous experimental works associated with SCS in SMRF have been concentrated on subassemblies such as beam-column joints or base-column joints. There is a need to experimentally investigate the seismic performance of a complete steel frame using SCS. Therefore, the shake table testing with two steel frames: SMRF and SCS frame, is carried out to evaluate the seismic performance of each system and to validate the numerical study conducted in former chapters.

This chapter describes the procedure for designing the prototypes considered to define the models used in the shake table testing, in which the SAP 2000 computer program and the 2003 National Earthquake Hazards Reduction Program (NEHRP) Recommended Provisions (also called as FEMA 450) were utilized. Thereafter, the scaling of the prototypes leading to the test models is presented and verified numerically. Finally, according to the seismic

design procedure for Self-Centering Post-Tensioned (SCPT) frame developed in Chapter IV, the SCPT frame model is designed and detailed.

### 5.1 Prototype Design

To experimentally investigate the seismic performance of Steel Moment Resisting Frames (SMRF) with and without Self-Centering Systems (SCS), the prototype is considered to be the MCEER West Coast Demonstration Hospital termed as WC70 (see Section 4.1). However, due to the limitations of dimensions and gravity capacities of the shake table, it is difficult to design a whole-frame assembly model scaled exactly from the WC70 structure. Therefore, the prototype was re-designed.

To achieve the similar seismic behavior of the WC70 frame, the re-designed prototype building is assumed to be a small outpatient building with the same structural characteristics as the WC70 frame and located at the same site (Northridge, CA). The density of gravity load in the new prototype structure is identical to that of the WC70 frame. As shown in Fig. 5-1, the prototype is a three-story SMRF building.

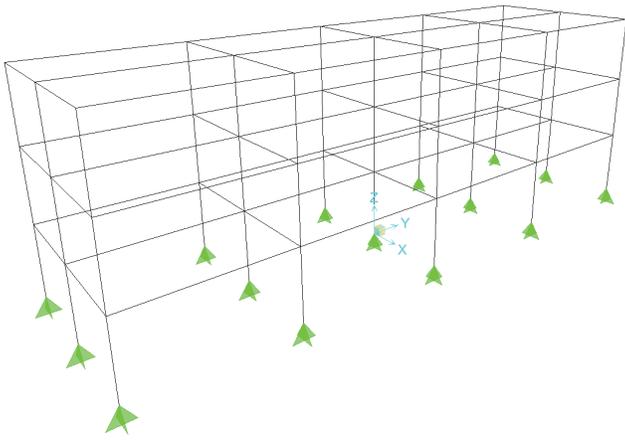


Figure 5-1 Re-Designed Prototype

The plan view of the prototype building is presented in Fig. 5-2. The surrounding Moment Resisting Frames (MRF) are represented by bold lines, which supply the lateral resistance and part of the gravity load, and the inside gravity frames are represented by simple lines, which only provide the gravity support but no lateral resistance.

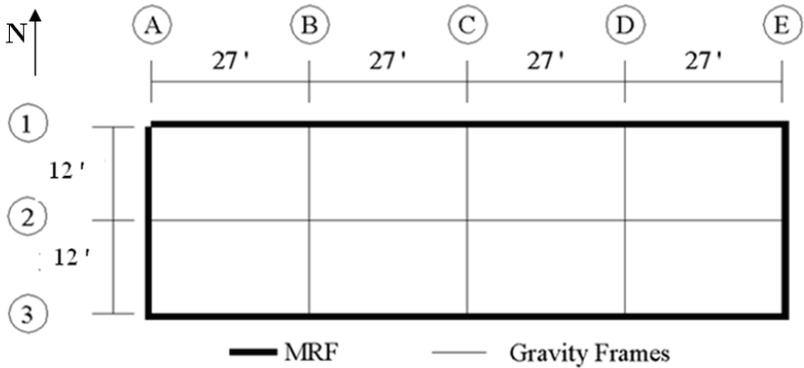


Figure 5-2 Plan View of Prototype Building

The re-designed prototype building is symmetric. Therefore, the frame in line A is selected as the prototype frame in this study, which represents half of the building. To simplify the prototype design, the seismic mass of typical floors and roof is assumed to be the same as the WC70 structure ( $0.5194 \text{ kip-sec}^2/\text{in.}$ ) and the corresponding weight of the half structure is 601.83 kips. According to FEMA450, the occupancy importance factor  $I$  is selected as 1.5 for a hospital building type and the site class is assumed as site class D due to the insufficient detail to determine the site class in accordance with Sec. 3.5.1 of FEMA 450. Design coefficients and factors for the prototype building are shown in Table 5-1, which is from Table 4.3-1 of FEMA 450, where  $R$  is the response modification coefficient,

**Table 5-1 FEMA 450 Design Coefficients and Factors for SMRF**

Basic Seismic-Force-Resisting System	Detailing Reference Section	$R^a$	$\Omega_0^b$	$C_d^c$
Special steel moment frames	AISC Seismic, Part I, Sec. 9	8	3	5/2

<sup>a</sup> Response modification factor

<sup>b</sup> System overstrength factor

<sup>c</sup> Deflection amplification factor

Following the Equivalent Lateral Force Procedure in Sec. 5.2 of FEMA 450, the seismic coefficient  $C_s$  and design seismic base shear  $V$  are, respectively:

$$C_s = \frac{S_{DS}}{R/I} = \frac{1.2578}{8/1.5} = 0.2358 \quad (5-1)$$

$$V = C_s W = 0.2358 * 601.83 = 141.91 \text{ kips} \quad (5-2)$$

where:

$S_{DS}$  = the design spectra response acceleration parameter in the short period range

$W$  = the weight of the half prototype structure

The equivalent lateral force,  $F_x$ , at any level can be determined by:

$$F_x = C_{vx} V \quad (5-3)$$

$$C_{vx} = \frac{w_x h_x^k}{\sum_{i=1}^n w_i h_i^k} \quad (5-4)$$

where the values for all the coefficients in Eq. (5-3) and (5-4) can be found in Section 5.2.3 of FEMA450.

Therefore, the equivalent force for the 1<sup>st</sup>, 2<sup>nd</sup> and 3<sup>rd</sup> level is 23.51 kips, 47.27 kips and 71.12 kips, respectively. The SAP 2000 computer program was used to calculate the response of the frame under those equivalent lateral forces with story drift control and structural over-strength control. The resulting beam and column sections are shown in Fig. 5-3.

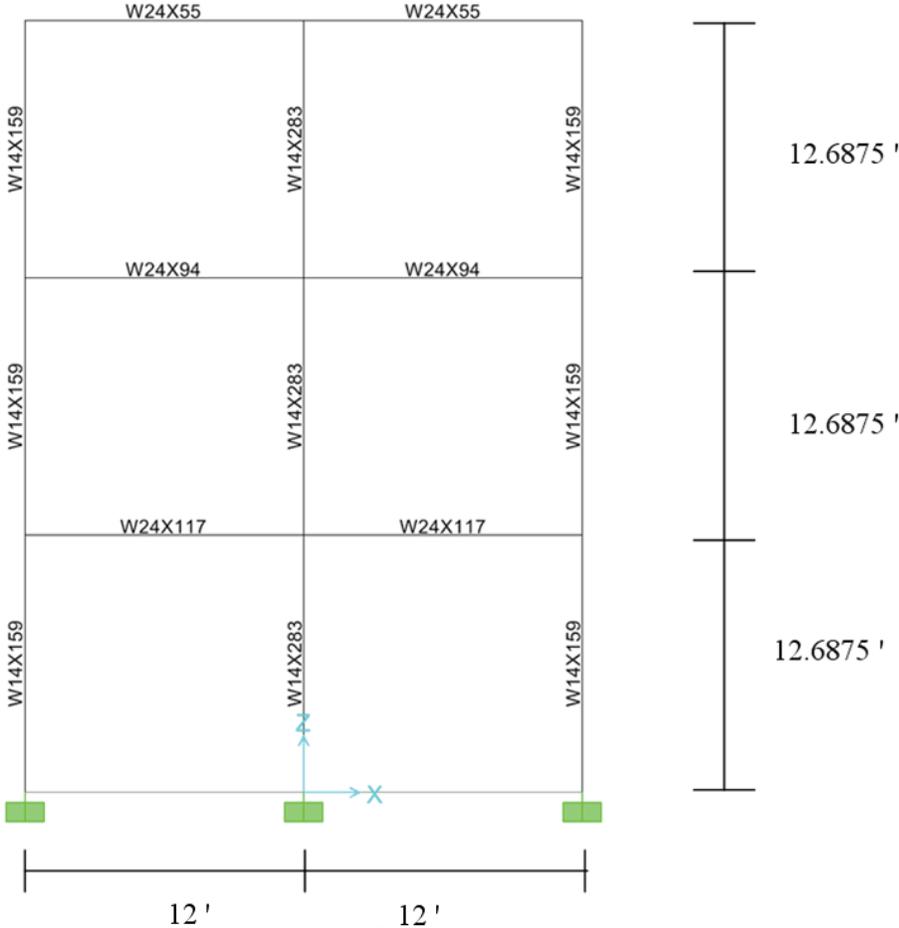


Figure 5-3 Prototype Frame

## 5.2 Scaling of Prototype

To experimentally investigate the seismic performance of the prototype building, the structure needs to be scaled as a model compatible with the requirements of the shake table. There are always some constraints in every shake table. The one used in this study is one of the three shake tables located in the Structural Engineering and Earthquake Simulation Laboratory (SEESL) at the University at Buffalo. The dimension of the concrete testing platform attached to this shake table is 10 ft. by 20 ft. and the acceleration limitation is  $\pm 1.15$  g for a 44-kip rigid specimen. The displacement limitation is  $\pm 6$  in. and the over-turning moment capacity is 333 kips-ft. Considering those limitations of the shake table as well as the gravity column frame (mass simulator), which will be presented in the next Chapter, the two fundamental scaling factors are calculated as:

$$\text{Scaling Factor of Linear Dimension: } S_l = 152.5in / 50.75in = 3 \quad (5-5)$$

$$\text{Scaling Factor of Mass: } S_M = 601.83 / 51.36 = 11.718 \quad (5-6)$$

where the scaling factor of mass is the mass ratio of prototype building to the gravity column frame.

Using the same material (structural steel) for both of the prototype and the model buildings, the scaling factor for the elastic modulus is

$$S_E = 1 \quad (5-7)$$

Therefore, the scaling factor of strain is:  $S_\varepsilon = S_{\Delta l} / S_l = 1$  (5-8)

where  $S_{\Delta l}$  is the scaling factor for the elongation with respect to the original length  $l$ .

All other scaling factors can be derived from scaling factors of the mass and linear dimension. From Eq. (5-7) and (5-8), the scaling factor for stress can be computed as:

$$S_\sigma = S_E S_\varepsilon = 1 \quad (5-9)$$

So, 
$$S_\sigma = \frac{S_{Force}}{S_{Area}} = \frac{S_M S_a}{S_{Area}} = \frac{S_M S_l S_T^{-2}}{S_l^2} = S_M S_l^{-1} S_T^{-2} = 1 \quad (5-10)$$

where  $S_a$  and  $S_T$  are the scaling factors for acceleration and time.

From Eq. (5-10), the time scaling factor can be determined as:

$$S_T = S_M^{1/2} S_l^{-1/2} \quad (5-11)$$

Therefore, from Eq (5-5) to (5-11), the following scaling factors can be defined as:

Frequency:  $S_\omega = S_T^{-1} = S_M^{-1/2} S_l^{1/2} \quad (5-12)$

Velocity:  $S_v = S_l S_T^{-1} = S_M^{-1/2} S_l^{3/2} \quad (5-13)$

Acceleration:  $S_a = S_l S_T^{-2} = S_M^{-1} S_l^2 \quad (5-14)$

Force:  $S_F = S_M S_a = S_l^2 \quad (5-15)$

Moment of Inertia:  $S_I = S_l^4 \quad (5-16)$

Energy:  $S_{EN} = S_F S_l = S_l^3 \quad (5-17)$

All the scaling factors along with their numerical values are presented in the Table 5-2.

**Table 5-2 Scaling Factors for Shake Table Testing**

		Dimension	Scale Factors	
			Artificial Mass Simulation	Value
Loading	Force	$MLT^{-2}$	$S_l^2$	9
	Pressure	$ML^{-1}T^{-2}$	1	1
	Horizontal Acceleration	$LT^{-2}$	$S_M^{-1}S_l^2$	0.768
	Velocity	$LT^{-1}$	$S_M^{-1/2}S_l^{3/2}$	1.518
	Time	T	$S_M^{1/2}S_l^{-1/2}$	1.976
Geometry	Linear dimension	L	$S_l$	3
	Displacement	L	$S_l$	3
	Elastic Section Modulus	$L^3$	$S_l^3$	27
	Moment of Inertia	$L^4$	$S_l^4$	81
	Frequency	$T^{-1}$	$S_M^{-1/2}S_l^{1/2}$	0.506
Material Properties	Modulus of Elasticity of Steel	-	1	1
	Stress	$ML^{-1}T^{-2}$	1	1
	Strain	-	1	1
	Poisson ratio	-	1	1
	Mass	M	$S_M$	11.718
	Energy	$ML^2T^{-2}$	$S_l^3$	81

Based on the scaling factors for elastic section modulus and moment of inertia in Table 5-2, the SMRF model for shake table testing is designed as shown in Fig. 5-4. The actual scaling factors in terms of the properties of the beams and the columns are shown in Table 5-3. The differences between the target and actual scaling factors are due to the limitation of structural steel products for the beam and column sections.

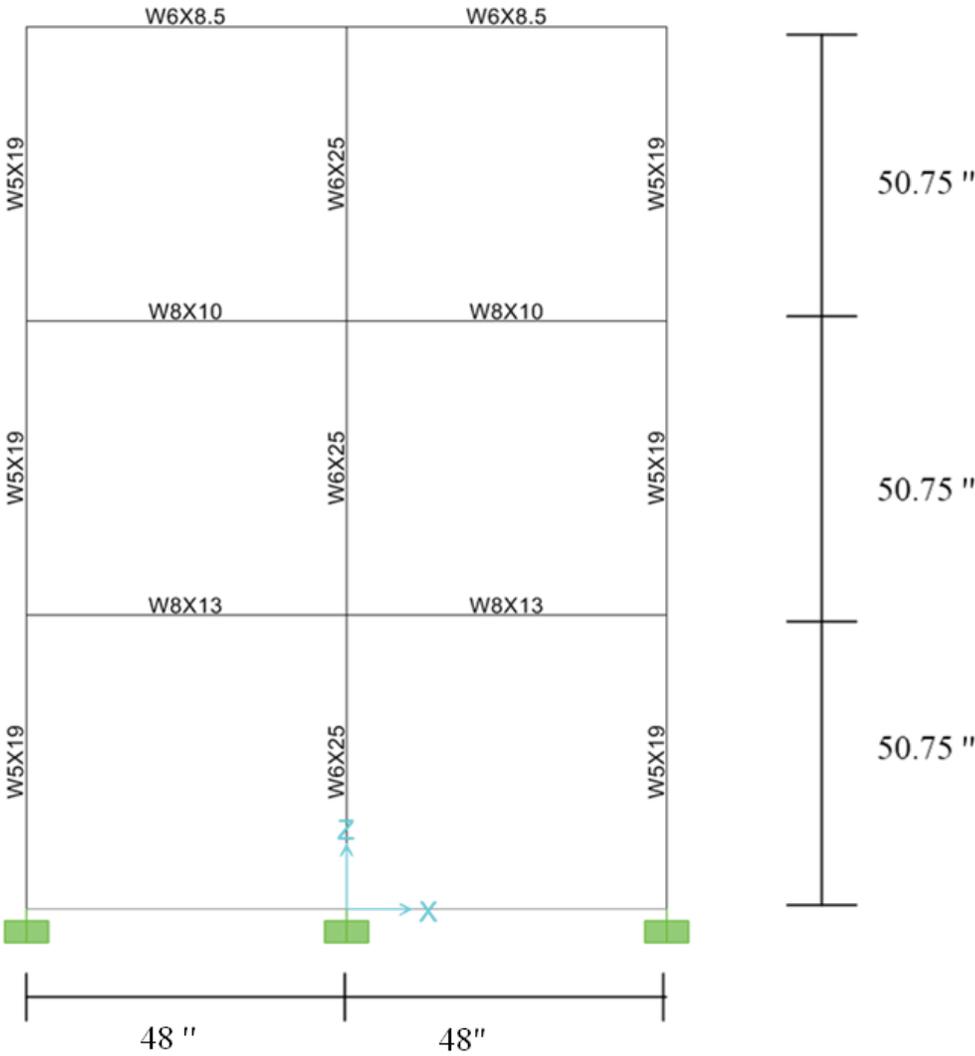


Figure 5-4 SMRF Model Configuration

**Table 5-3 Actual Scaling Factors of Beams and Columns Properties**

Sections			Elastic Section Modulus (S <sub>x</sub> )		Moment of Inertia (I <sub>x</sub> )	
			in <sup>3</sup>	Scaling Factor (Target value 27)	in <sup>4</sup>	Scaling Factor (Target value 81)
Beams	Prototype	w24x117	291	29.4	3540	89.4
	Model	w8x13	9.91		39.6	
	Prototype	w24x94	222	28.4	2700	87.7
	Model	w8x10	7.81		30.8	
	Prototype	w24x55	115	22.6	1360	91.9
	Model	w6x8.5	5.08		14.8	
Columns	Prototype	w14x159	254	24.9	1900	72.2
	Model	w5x19	10.2		26.3	
	Prototype	w14x283	459	27.3	3840	71.6
	Model	w6x25	16.8		53.6	

### 5.3 Analytical Verification of Scaling Factors

In order to verify the scaling factors obtained in the previous section, the seismic analysis of the 2D SMRF prototype and model (Fig. 5-3 and 5-4) was carried out using the SAP 2000 computer program. The seismic excitation considered was the first 3 seconds of the MCEER simulated earthquake in Scenario 1, Event 1, with a 2% probability of exceedance in 50 years. The joint information for the numerical analysis is found in Fig. 5-5.

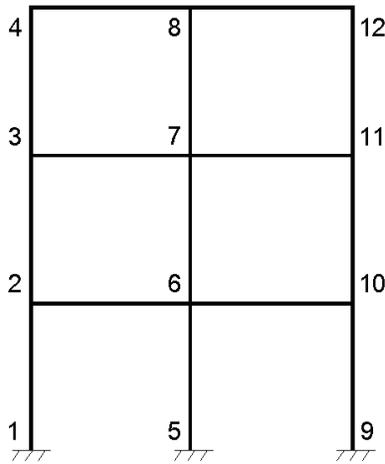


Figure 5-5 Nodal Information of Models in SAP

The displacement and acceleration responses of both the SMRF prototype and model are shown in Table 5-4. The target scaling factors for displacements and accelerations are 3 and 0.768, respectively, to which the corresponding average values calculated in Table 5-4 are approximately equal. There are small differences between the values of acceleration ratios and the target acceleration scaling factor as described in the previous section, when the sections of beams and columns in the scaled model are selected. Due to the available structural steel products, it is impossible to find the exact section with the exact scaling factor for depths and widths as well as elastic modulus and moment inertia. Finally, it leads to the difference between the target scaling factor and the ones calculated with the model implementing actual sections. However, the differences are small such that the scaling procedure can be considered to be in accordance with the simulation principles.

**Table 5-4 Verification of Scaling Factors**

Joint*		Relative Displacement		Scaling Factor	Absolute Acceleration		Scaling Factor
		Prototype	Model	ratio	Prototype	Model	ratio
		in	in	( $S_I=3$ )	g	g	( $S_a=0.768$ )
1	Max	0.000	0.000	--	0.318	0.413	0.770
1	Min	0.000	0.000	--	-0.252	-0.329	0.768
2	Max	0.548	0.177	3.096	0.352	0.501	0.702
2	Min	-0.515	-0.166	3.102	-0.448	-0.475	0.944
3	Max	1.296	0.416	3.115	0.518	0.648	0.800
3	Min	-1.054	-0.335	3.146	-0.505	-0.619	0.816
4	Max	1.840	0.601	3.062	0.460	0.653	0.704
4	Min	-1.363	-0.447	3.049	-0.612	-0.733	0.835
5	Max	0.000	0.000	--	0.318	0.413	0.770
5	Min	0.000	0.000	--	-0.252	-0.329	0.768
6	Max	0.548	0.177	3.096	0.349	0.500	0.699
6	Min	-0.514	-0.165	3.115	-0.444	-0.469	0.946
7	Max	1.294	0.415	3.118	0.515	0.644	0.799
7	Min	-1.052	-0.334	3.150	-0.501	-0.614	0.816
8	Max	1.837	0.600	3.062	0.457	0.651	0.703
8	Min	-1.361	-0.446	3.052	-0.609	-0.731	0.832
9	Max	0.000	0.000	--	0.318	0.413	0.770
9	Min	0.000	0.000	--	-0.252	-0.329	0.768
10	Max	0.548	0.177	3.096	0.352	0.501	0.702
10	Min	-0.515	-0.166	3.102	-0.448	-0.475	0.944
11	Max	1.296	0.416	3.115	0.518	0.648	0.800
11	Min	-1.054	-0.335	3.146	-0.505	-0.619	0.816
12	Max	1.840	0.601	3.062	0.460	0.653	0.704
12	Min	-1.363	-0.447	3.049	-0.612	-0.733	0.835
Average				3.096			0.792

\*see Figure 5-5

## 5.4 Design of Self-Centering Post-Tensioned (SCPT) Frame Model

To compare the seismic performance of the SMRF and SCS, two models are considered for shake table testing: a conventional Steel Moment-Resisting Frame (SMRF) model as shown in Fig. 5-4 and a Self-Centering Post-Tensioned (SCPT) frame model in which the Post-Tensioned Energy-Dissipating (PTED) connections are implemented in the beam-column connections as discussed in Chapter IV. Those PTED connections are designed according to the procedure proposed in Section 4.4.2

### 5.4.1 Preliminary Design

Following the procedure of Section 4.4.2, the modal and push-over analyses of the SMRF prototype are conducted using SAP 2000. As a result of those analyses, the fundamental period and yielding strength can be determined as shown in Table 5-5 and Fig. 5-6, respectively. Therefore,

$$\text{Prototype Natural Period:} \quad T_0 = 0.581 \text{ sec} \quad (5-18)$$

Prototype Effective Weight in 1<sup>st</sup> mode:

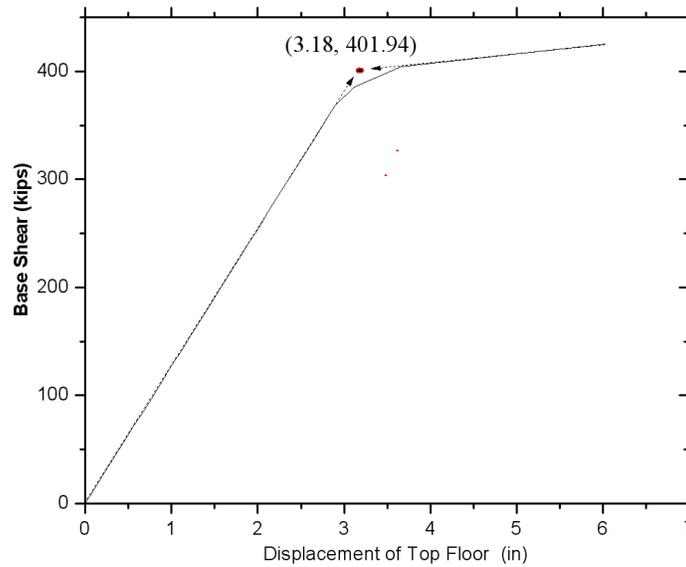
$$W_{effective} = 85.15\% W_{total} = 85.15\% * 200.61 * 3 = 512.46 \text{ kips} \quad (5-19)$$

$$\text{Strength Ratio:} \quad \eta_{prototype} = \frac{F_Y}{W_{effective}} = \frac{401.94}{512.46} = 0.78 \quad (5-20)$$

where  $F_Y$  is yielding strength of the SMRF prototype as shown in Fig. 5-6.

**Table 5-5 Modal Information of SMRF Prototype**

Modal Number	Period Second	Modal Participating Mass Ratios		
		Horizontal Direction	Vertical Direction	Sum (horizontal direction)
1 <sup>st</sup> Mode	0.58089	0.8515	0	85.15%
2 <sup>nd</sup> Mode	0.557528	0	0.7007	85.15%
3 <sup>rd</sup> Mode	0.53076	0	0	85.15%
4 <sup>th</sup> Mode	0.213143	0	0.00002376	85.15%
5 <sup>th</sup> Mode	0.193317	0.1208	0	97.23%



**Figure 5-6 Push-Over Analysis of SMRF Prototype**

Considering that the prototype building is a hospital, there are more acceleration-sensitive non-structural components than displacement- or velocity-sensitive ones. Therefore, the Relative Performance Index (RPI, see Chapter III) is concentrated on 70% acceleration response and 30% displacement response when the controlling parameters of SCS are selected. As shown in Fig. 5-7, the dash circle represents the desirable RPI. According to the page B13 and B14 in appendix B, Eq. (5-14) and (5-16), the controlling parameters are

determined by interpolation, as seen in table 5-6, because the strength ratio of original SMRF in page B13 and B14 is 0.5 and 1, respectively, while the strength ratio of the SMRF prototype is 0.78.

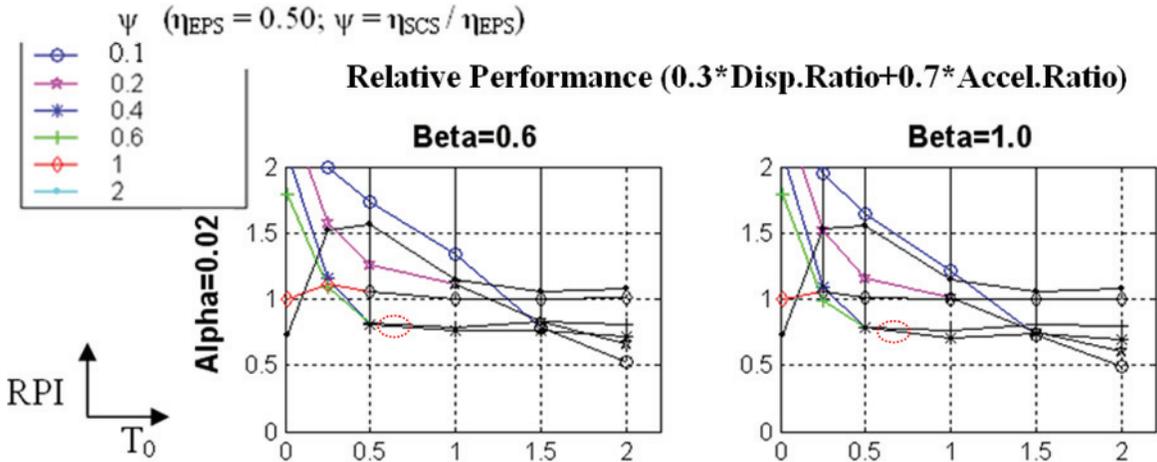


Figure 5-7 Relative Performance Index (RPI, part of page B13 in Appendix B)

**Table 5-6 Design Parameters of SCPT Frame**

$\alpha$	$\beta$	$\psi = \eta_{SCPT} / \eta_{prototype}$
0.03	0.8	0.5

Based on the selected SCS parameters and the practical design procedure, the preliminary design of the SCPT frame model are presented in Table 5-7, in which the real controlling parameters ( $\alpha$ ,  $\beta$  and  $\psi$ ) are calculated based on the selected SCPT parameters (i.e. initial PT force, area of PT strands and ED bars, and thickness of R-plate) in order to be compared with the desirable SCS parameters. The maximum stresses in the R-plates and the beam flanges are also listed whereas the yielding stresses of these two parts are 100 ksi and 50 ksi, respectively, such that the beams remain elastic under seismic loading.

**Table 5-7 Preliminary Design of SCPT Frame Model**

	1 <sup>st</sup> floor	2 <sup>nd</sup> floor	3 <sup>rd</sup> floor
Beam Section	w8x13	w8x10	w6x8.5
Initial force in PT bars (kips)	37.06	29.9	27.48
Area of PT strands (in <sup>2</sup> )	0.57	0.42	0.24
Area of ED (2_EDbar) (in <sup>2</sup> )	0.15	0.12	0.11
Thickness of R plate (in)	0.268	0.205	0.126
$\alpha$	0.051	0.046	0.026
$\beta$	0.798	0.799	0.799
$\psi = \eta_{SCPT} / \eta_{prototype}$	0.5	0.502	0.502
Max Stress of R plate (ksi)	81.26	75.07	72.82
Max Stress of Flange (ksi)	48.8	48.81	49.58

#### 5.4.2 Final Design

The intermediate design is neglected since the beams of the two spans at the same level in the SCPT model are identical. In the final design, according to the results of the preliminary design and the products supplied by DYWIDAG Systems International (DSI), the real sections of Post-Tensioned (PT) strands and Energy-Dissipating (ED) bars are selected based on the procedure proposed in Section 4.4.2.3. The final design results are shown in Table 5-8. The cross sectional area of 0.6 and 0.5 inch PT strands are 0.217 and 0.153 in<sup>2</sup>, respectively. The yield and ultimate stress of PT strands are 243 and 270 ksi, respectively. The original

diameter of #6 ED bars used in this model is 0.86 in while it was machined to 5/16 and 9/32 inch for the 1<sup>st</sup> and 2<sup>nd</sup>/3<sup>rd</sup> floors as shown in Table 5-8 and the yield stress of ED bars is 75 ksi. The steel of Reinforcing Plate (R-Plate) implemented in this model is ASTM A514 with a 100 ksi yield stress. The controlling parameters ( $\alpha$ ,  $\beta$  and  $\psi$ ), as calculated from the selected PT strands and ED bars, are slightly different than the desirable ones, which does not lead to much influence on the relative performance of SCPT frame model as compared to the SMRF model. The maximum stress in the beam flanges is reduced slightly such that the beam stresses remain much lower than those in the preliminary design, in which they are close to the yielding stress.

**Table 5-8 Final Design of SCPT Frame Model**

	1st floor	2nd floor	3rd floor
Beam Section	w8x13	w8x10	w6x8.5
Initial force in PT bars (kips)	37	30	27
Area of PT strands (in <sup>2</sup> )	2@0.6in. (0.434)	2@0.6in. (0.434)	2@0.5in. (0.306)
Area of ED (2_EDbar) (in <sup>2</sup> )	#6 2@5/16 in. (0.153 in <sup>2</sup> )	#6 2@9/32 in. (0.124 in <sup>2</sup> )	#6 2@9/32 in. (0.124 in <sup>2</sup> )
Thickness of R plate (in)	5/16	1/4	3/16
$\alpha$	0.047	0.053	0.034
$\beta$	0.820	0.829	0.877
$\psi = \eta_{SCPT} / \eta_{SMRF}$	0.48	0.48	0.51
Max Stress of R plate (ksi)	78.59	77.13	75.66
Max Stress of Flange (ksi)	41.24	46.08	43.30



## **CHAPTER 6**

### **EXPERIMENTAL PROCEDURES**

To experimentally investigate the seismic performance of the Steel Moment Resisting Frame (SMRF) and Self-Centering Post-Tensioned (SCPT) frame designed in Chapter V, a shake table testing program was conducted. This chapter describes the experimental procedures used in this testing program.

First, the scope of the testing program is discussed in terms of the two models, the floor mass simulator and the shake table used in the tests. Thereafter, based on the numerical predictions of the seismic performance of the designed SCPT frame, the ground motion used in the shake table tests is selected. Finally, the test protocol and instrumentations are presented.

#### **6.1 Scope of Testing**

To compare the seismic performance of the SMRF and SCPT frame, two different frame models were used in the shake table testing program. An existing floor mass simulator was implemented to provide the scaled mass added to the frame models. This floor mass simulator supported only the gravity loading used to simulate the mass of the models but did not contribute any lateral stiffness or strength to the test specimen. Also, the characteristics of the shake table system are detailed.

### 6.1.1 Description of Frame Models

Two frame models (SMRF and SCPT frame) were considered in the shake table testing program. These frame models were designed in Chapter V. More details on these two models are presented in this section.

#### 6.1.1.1 Steel Moment Resisting Frame (SMRF) Model

As shown in Fig. 6-1 and Table 6-1, the SMRF model is a 2-bay, 3-story frame, made of ASTM A572 Grade 50 steel. The prototype of the SMRF model is a hospital building constructed in the 1970s incorporating fully welded beam-to-column connections (beam flanges and webs welded to the column flanges). Similar welded connections were used in the SMRF model in order to simulate the similar performance of the prototype building. The

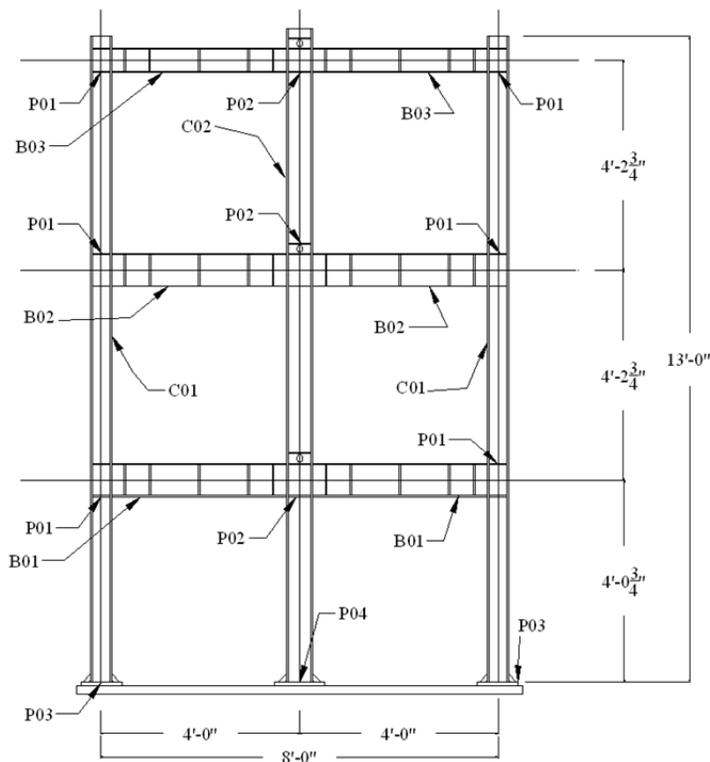


Figure 6-1 SMRF model

Table 6-1 Part Description for SMRF model

Part Code	Part Description	QTY
C01	Exterior Column, 13-ft length W5x16	2
C02	Interior Column, 13-ft 2-in length W6x25	1
B01	Beam in 1 <sup>st</sup> floor, 3ft 6 1/4 in W8x13	2
B02	Beam in 2 <sup>nd</sup> floor, 3ft 6 1/4 in W8x10	2
B03	Beam in 3 <sup>rd</sup> floor, 3ft 6 1/4 in W6x8.5	2
P01	3/8 in thick Continuity Plate 4 9/32 x 2 3/8 in	24
P02	3/8 in thick Continuity Plate 5 15/32 x 2 7/8 in	12
P03	1 in thick Exterior Base Plate 12 x 9 3/4	2
P04	1 in thick Interior Base Plate 12 x 12	1

two exterior columns (i.e. part code C01 in Table 6-1) were designed as w5x19 in Chapter V. However, a w5x16 section was used in the model since the w5x19 was not available from local manufacturers. Although the w5x16 section is slightly lighter than the w5x19 section, both the moment of inertia and the elastic section modulus are similar, which did not affect significantly the fundamental period and yielding strength of the model. As shown in Fig 6-2, the base columns were welded to stiffened base plates. These steel base plates were then bolted on the shake table. Therefore, a fixed-base connection was assumed for the model.

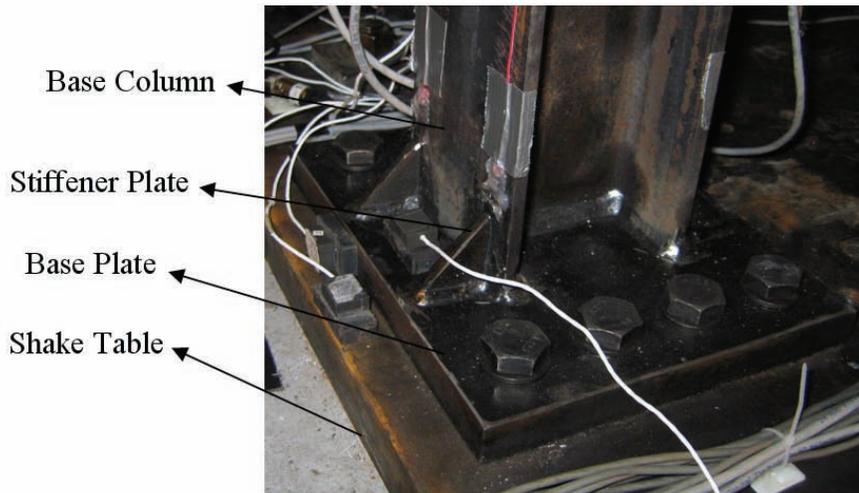


Figure 6-2 Base Column Connection of SMRF Model

As seen in Fig. 6-3, continuity plates were welded to the columns to reinforce the panel zone and stiffener plates were welded to the beams to prevent local buckling in accordance with the requirements of the Load & Resistance Factor Design (LRFD 3<sup>rd</sup> edition, 2001). All the construction drawings for the SMRF model are included in Appendix C.

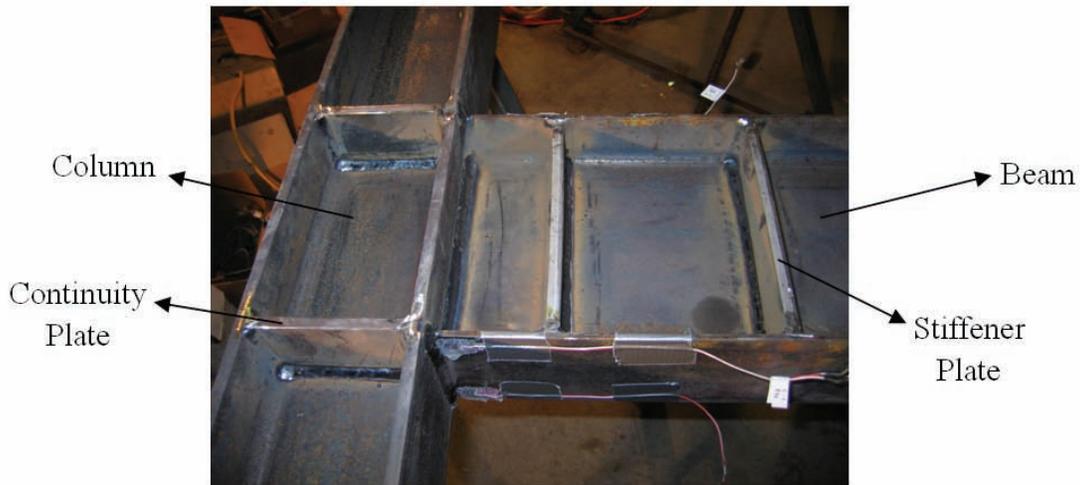


Figure 6-3 Beam-Column Connection in SMRF Mode

#### 6.1.1.2 Self-Centering Post-Tensioned (SCPT) Frame Model

A photograph of the SCPT frame model is shown in Fig. 6-4. This model incorporated the same beam and column sections as those used in the SMRF model. However, no welding is introduced between the beams and columns. Instead, Post-Tensioned (PT) strands were installed along the beam to provide a “clamping” force for the beam-column connections. The PT strands also provided a re-centering capability to the structural system under earthquakes. As shown in Fig. 6-5, two PT strands were used for each level along with Energy Dissipating (ED) bars. Four ED bars were welded to each beam-column connection. Each ED bar is connected to a threaded mechanic connector (i.e. coupler), which is welded to the beam flange and continuity plate in the adjacent column. The ED bars were designed to yield in tension and compression in order to absorb energy during seismic shaking.

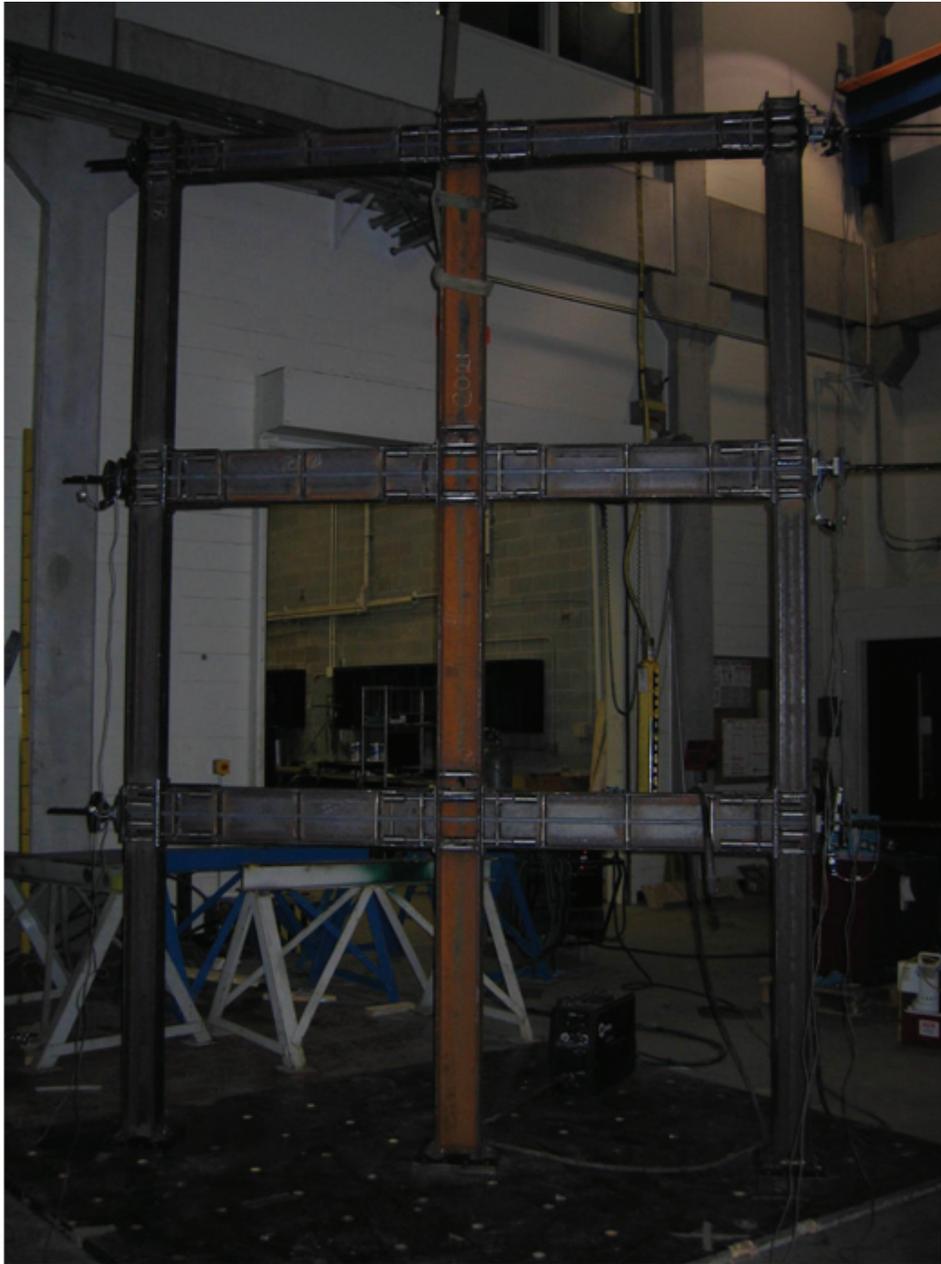


Figure 6-4 SCPT Frame Model

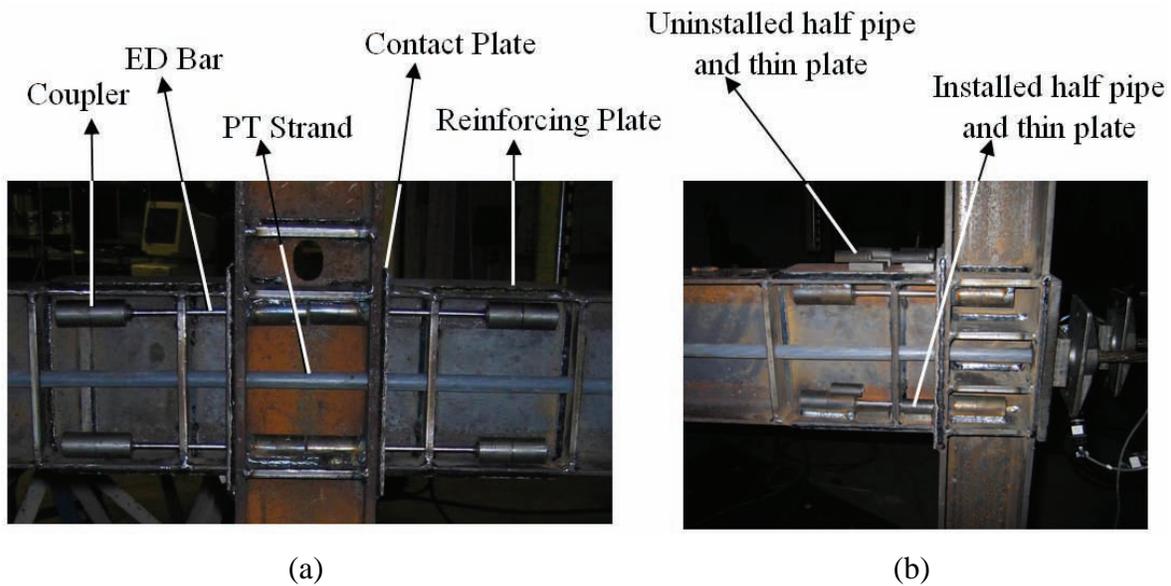


Figure 6-5 Beam-Column Connections in the SCPT Frame1 Model

(a) Central Connection in the 1<sup>st</sup> level (b) Right Exterior Connection in the 1<sup>st</sup> level

To prevent buckling in compression of the ED bars, thin plates and half pipes (see DWG ED103 in Appendix D) were welded to the beam flanges surrounding the ED bars as shown in Fig. 6-5(b). The PT strands, ED bars and couplers were donated by the DYWIDAG-System International (DSI). To prevent the beam from yielding, reinforcing plates, made of ASTM A514 steel with a 100ksi yielding strength, were welded to the beam flanges. Due to the large expected compressive stress in the beam-column connections when a gap opened, contact plates (also made of ASTM A514) were welded to the column flanges to avoid local yielding. The holes in the central column, as seen in Fig. 6-5 (a), are used to connect the two mass floors. The order of assembling the SCPT frame model is to prestress the PT strands at first and then to weld the ED bars. All the construction drawings for the SCPT frame model are included in Appendix D.

**6.1.2 Floor Mass Simulator**

The Floor Mass Simulator (FMS) was designed in an earlier research (Kusumastuti 2005) and has been utilized for several seismic projects in SEESL. As shown in Fig. 6-6 (a), the main purpose of the FMS is to simulate the floor mass of the prototype structure. The FMS is composed of two adjacent frames supporting six steel plates, each weighting about 8.5 kips. As seen in Fig. 6-6 (b) and (c), due to the rocking support design, the FMS performs as a pin-based structure in the shaking direction providing no lateral stiffness. The bracing



(a)



(b)



(c)

Figure 6-6 Floor Mass Simulator (a) FMS installed with SCPT Frame Model (b) Rocking Support at the top of the column (c) Rocking Support at the base of the column

system resists the deformations in the transverse direction. Each frame model (SMRF and SCPT frame) was installed between two frames of the FMS and was connected to the FMS by circular steel bars through the central column at each level, as shown in Fig. 6-7 (a). Connecting the frame model only through its central column allows the axial deformation of the beams of the frame model. Axial elongation of the beams must not be prevented for the SCPT frame in order for the gap openings to occur freely. Through the connections shown in Fig. 6-7 (b), the FMS prevents the out-of-plane deformations of the frame model and the torsion deformations of its beams.



(a)



(b)

Figure 6-7 Connections between FMS and Frame Model  
(a) Connection of the floor mass      (b) Transverse support

### 6.1.3 Shaking Table System

The shaking table system used in this experimental investigation is a 5-degree-of-freedom system located in the Structural Engineering and Earthquake Simulation Laboratory (SEESL) at University at Buffalo, as shown in Fig. 6-8. The 12 ft. by 12 ft. shaking table is usually covered by a reinforced concrete testing platform with plan dimensions of 10ft. by 20ft.,

which extends its testing area. The payload of the shaking table is 110 kips without the concrete platform and 85 kips with it. In this experimental study, only the longitudinal (horizontal) movement of the shake table was utilized. The shake table limitations for displacement, velocity and acceleration in this direction are  $\pm 6$  in., 30 in./sec and 1.15g, respectively, at a payload of 44 kips.

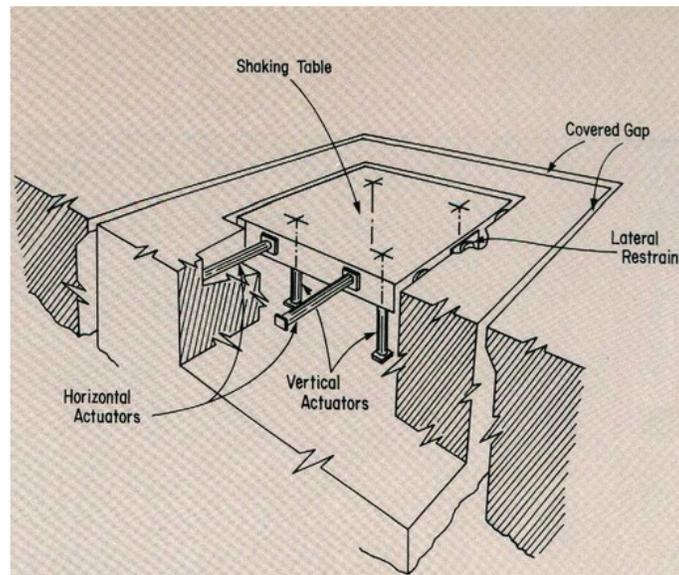


Figure 6-8 Perspective View of Shaking Table and Foundation  
(From <http://Nees.Buffalo.edu>)

## 6.2 Selection of Ground Motions

Since the prototype structure was designed for a life safety performance in Chapter V, the ground motion used in the shaking table testing was selected among an ensemble of 25 synthetic MCEER earthquake records with a 10% probability of exceedance in 50 years (see the Section 3.2.1). A numerical seismic analysis was conducted to evaluate the seismic

performance of the SCPT frame model under this ensemble of scaled 25 records in order to select the ground motion to be used in the testing program within the limitations of the shake table. For this purpose, the maximum inter-story drift, maximum displacement of the top floor and maximum acceleration in each floor for each of these input motion were calculated and are shown in Figs. 6-9 to 6-15. In these figures, the Earthquake No. Label represents each of the 25 earthquake records. The values between mean plus one positive and one negative standard deviation (top and bottom horizontal lines in Figs. 6-9 to 6-15) are considered to be statistically representative responses and thus, these earthquake records causing those responses are candidates as the excitation for the seismic testing. Considering the displacements and accelerations limitations of the shaking table, the Earthquake No. 7 (Event 7 of Scenario 1) was selected as the input ground motion for the seismic tests. According to the scaling factors in Table 5-2 in Chapter V, the amplitude of this selected ground motion is divided by 0.768 and its duration is divided by 1.976.

**Note** (for Figs. 6-9 to 6-15): The three horizontal lines in each figure represents the standard deviations and mean of the bar values. The horizontal axis in regard to Earthquake No. means the different earthquake records: No. 1 to 12 for the Event 1 to 12 of Scenario 1; No. 13 to 20 for the Event 1 to 8 of Scenario 2; No. 21 to 23 for the Event 1 to 3 of Scenario 3; No. 24 to 25 for the Event 1 to 2 of Scenario 4 in the ensemble of 25 synthetic MCEER earthquake records with 10% probability of exceedance in 50 years.

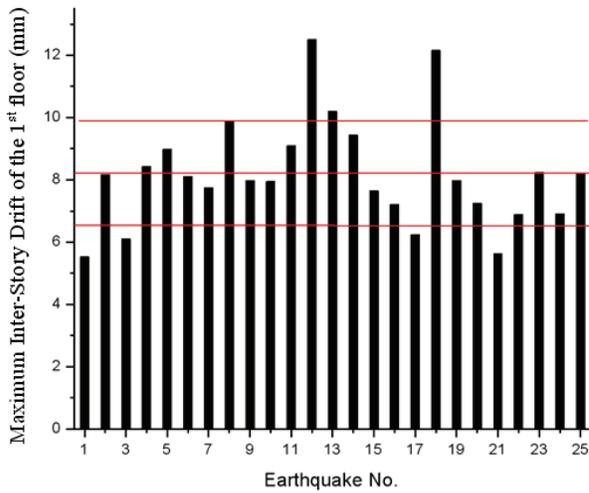


Figure 6-9 Maximum Inter-story Drift in the 1<sup>st</sup> level

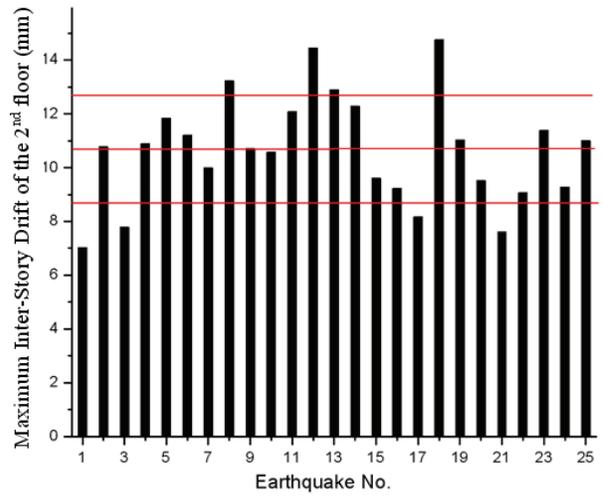


Figure 6-10 Maximum Inter-story Drift in the 2<sup>nd</sup> level

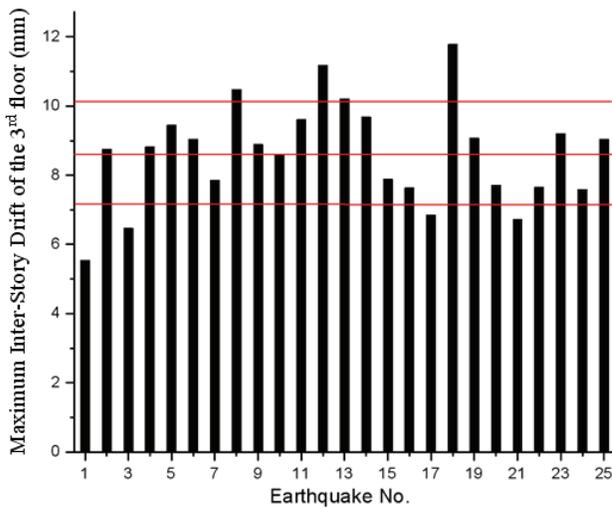


Figure 6-11 Maximum Inter-story Drift in the 3<sup>rd</sup> level

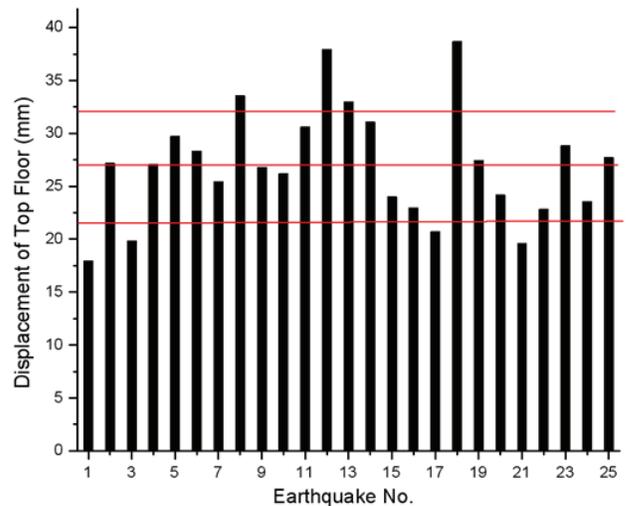


Figure 6-12 Maximum Displacement of Top Floor

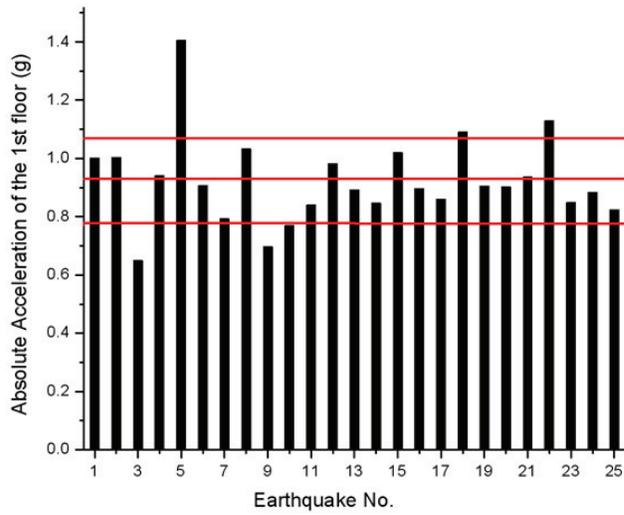


Figure 6-13 Maximum Acceleration  
in the 1<sup>st</sup> level

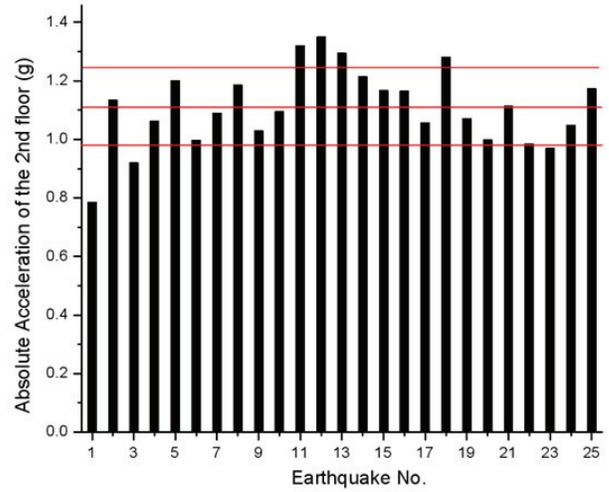


Figure 6-14 Maximum Acceleration  
in the 2<sup>nd</sup> level

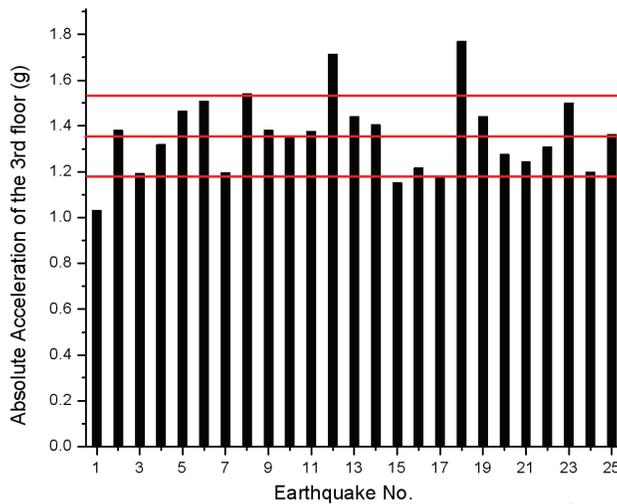


Figure 6-15 Maximum Acceleration in the 3<sup>rd</sup> level

As shown in Fig. 6-16, the Peak Ground Acceleration (PGA) of the selected ground motion is 0.68g, which is below the acceleration limit of 1.15g of the shake table. Therefore, the maximum possible amplitude of the acceleration input is 150%.

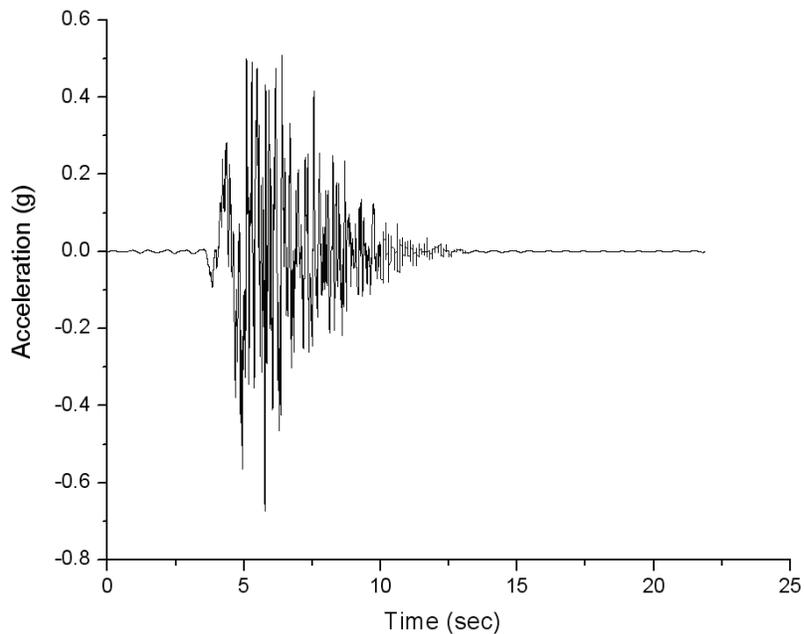


Figure 6-16 Selected Ground Motion

(Scaled MCEER ground motion for event 7 of Scenario 1 with 10% probability of exceedance in 50 years)

### 6.3 Test Protocols

Two series of shake table tests were conducted for both the Steel Moment Resisting Frame (SMRF) and Self-Centering Post-Tensioned (SCPT) frame model. In the first series, the selected ground motion shown in Fig. 6-16 was used. In the second series, the direction of the ground motion was reversed (polarity). These two series were conducted in order to observe maximum gap opening under positive and negative moments at each beam-column joint of the SCPT frame. Different amplitudes of the selected ground motion were used during the tests. The amplitude of the input excitation was expressed as a percentage of the full-scale record shown in Fig. 6-15 and varied from 25% to 150%. Low level white noise tests were also conducted after each seismic test and used to monitor the changes in dynamic

characteristic of the test frames such as their natural frequencies and mode shapes. This white noise excitation has 0.05g amplitude and a wide frequency band (0.5 - 50Hz). Pull-Back tests were also conducted to double check the system information.

The test protocols for both the SCPT and SMRF models are shown in Table 6-2 and 6-3, respectively. Due to some malfunction of the test equipment such as the failure of the video capture and miscalibration of the instrumentation, some tests were repeated. Some beams of the SCPT frame model slid down vertically along the face of the columns in the large amplitude seismic tests during gap openings, which was not expected. Therefore, small round bars were welded to the columns under the beam flanges to provide a vertical support and then the tests were repeated (Tests no. 42 to 78 in Table 6-2). To better limit the vertical movement of the beams, round bars were also installed on the top of the beams and the same tests were repeated again (Tests no. 79 to 108 in Table 6-2). Therefore, tests of No. 1 to 41 are named as the 1<sup>st</sup> test phase, No. 42 to 78 as the 2<sup>nd</sup> phase and No. 79 to 108 as the 3<sup>rd</sup> phase in the SCPT model. Theoretically, the friction between the beams and columns in the SCPT frame model should have been sufficient to resist the shear and prevent the vertical movement. However, it is believed that due to the fact that the ends of the beams were not manufactured perfectly flat, and then grinded, the friction coefficient was lower and the area of contact surface was reduced. Therefore, in the large amplitude seismic tests (e.g. 150% amplitude), the beams moved vertically. Presumably, this movement could have been prevented by sand blasting the contact surfaces between the beams and columns. The notes included in Table 6-2 and 6-3 are predictions from the numerical results, where the term

“elastic range” means the beams and columns remain elastic and the term “gap” refers to the possible gap between beams and columns in the SCPT model.

**Table 6-2 Test Protocol for the SCPT Frame Model**

Test No.	Test Label	Excitation	Direction	PGA		Note
				g	%	
1	PWN1	White Noise	N/A	0.05	N/A	System Identification
2	PWN1A	White Noise	N/A	0.05	N/A	System Identification
3	PE1	S1E7*	Positive	0.17	25	Elastic Range
4	PWN2	White Noise	N/A	0.05	N/A	System Identification
5	PWN1B	White Noise	N/A	0.10	N/A	System Identification
6	PWN1C	White Noise	N/A	0.15	N/A	System Identification
7	PE1A	S1E7	Positive	0.17	25	Elastic Range
8	PWN1D	White Noise	N/A	0.05	N/A	System Identification
9	PE1B	S1E7	Positive	0.17	25	Elastic Range
10	PD1	Sinusoidal Wave	N/A	N/A	N/A	System Identification
11	PWN2A	White Noise	N/A	0.05	N/A	System Identification
12	PE2	S1E7	Positive	0.34	50	Elastic Range
13	PWN3	White Noise	N/A	0.05	N/A	System Identification
14	PE3	S1E7	Positive	0.48	70	Gap begins opening
15	PWN4	White Noise	N/A	0.05	N/A	System Identification
16	PER1	S1E7	Negative	0.17	25	Elastic Range

**Table 6-2 Test Protocol for the SCPT Frame Model (continued)**

Test No.	Test Label	Excitation	Direction	PGA		Note
				g	%	
17	PER2	S1E7	Negative	0.34	50	Elastic Range
18	PER3	S1E7	Negative	0.48	70	Gap begins opening
19	PWN4A	White Noise	N/A	0.05	N/A	System Identification
20	PE4	S1E7	Positive	0.58	85	Most Gaps Opening
21	PWN5	White Noise	N/A	0.05	N/A	System Identification
22	PER4	S1E7	Negative	0.58	85	Most Gaps Opening
23	PWN5A	White Noise	N/A	0.05	N/A	System Identification
24	PE5	S1E7	Positive	0.68	100	Most Gaps Opening
25	PWN6	White Noise	N/A	0.05	N/A	System Identification
26	PE5A	S1E7	Positive	0.68	100	Most Gaps Opening
27	PWN6A	White Noise	N/A	0.05	N/A	System Identification
28	PER5	S1E7	Negative	0.68	100	Most Gaps Opening
29	PWN6B	White Noise	N/A	0.05	N/A	System Identification
30	PE6	S1E7	Positive	0.85	125	Most Gaps Opening
31	PWN7	White Noise	N/A	0.05	N/A	System Identification
32	PWN7A	White Noise	N/A	0.05	N/A	System Identification
33	PE6A	S1E7	Positive	0.85	125	Most Gaps Opening
34	PWN7B	White Noise	N/A	0.05	N/A	System Identification
35	PER6	S1E7	Negative	0.85	125	Most Gaps Opening

**Table 6-2 Test Protocol for the SCPT Frame Model (continued)**

Test No.	Test Label	Excitation	Direction	PGA		Note
				g	%	
36	PWN7C	White Noise	N/A	0.05	N/A	System Identification
37	PE7	S1E7	Positive	1.02	150	Most Gaps Opening
38	PE7A	S1E7	Positive	1.02	150	Most Gaps Opening
39	PWN8	White Noise	N/A	0.05	N/A	System Identification
40	PER7	S1E7	Negative	1.02	150	Most Gaps Opening
41	PWN8A	White Noise	N/A	0.05	N/A	System Identification
Round bars were installed under the beam flanges after Test no. 41.						
42	PWN9	White Noise	N/A	0.05	N/A	System Identification
43	PE8	S1E7	Positive	0.17	25	Elastic Range
44	PWN10	White Noise	N/A	0.05	N/A	System Identification
45	PER8	S1E7	Negative	0.17	25	Elastic Range
46	PWN11	White Noise	N/A	0.05	N/A	System Identification
47	PE9	S1E7	Positive	0.34	50	Elastic Range
48	PWN12	White Noise	N/A	0.05	N/A	System Identification
49	PER9	S1E7	Negative	0.34	50	Elastic Range
50	PWN13	White Noise	N/A	0.05	N/A	System Identification
51	PE10	S1E7	Positive	0.48	70	Gap begins opening
52	PWN14	White Noise	N/A	0.05	N/A	System Identification
53	PER10	S1E7	Negative	0.48	70	Gap begins opening

**Table 6-2 Test Protocol for the SCPT Frame Model (continued)**

Test No.	Test Label	Excitation	Direction	PGA		Note
				g	%	
54	PWN15	White Noise	N/A	0.05	N/A	System Identification
55	PE11	S1E7	Positive	0.58	85	Most Gaps Opening
56	PWN16	White Noise	N/A	0.05	N/A	System Identification
57	PER11	S1E7	Negative	0.58	85	Most Gaps Opening
58	PWN17	White Noise	N/A	0.05	N/A	System Identification
59	PE12	S1E7	Positive	0.68	100	Most Gaps Opening
60	PWN18	White Noise	N/A	0.05	N/A	System Identification
61	PER12	S1E7	Negative	0.68	100	Most Gaps Opening
62	PWN19	White Noise	N/A	0.05	N/A	System Identification
63	PE13	S1E7	Positive	0.85	125	Most Gaps Opening
64	PE13A	S1E7	Positive	0.85	125	Most Gaps Opening
65	PWN20	White Noise	N/A	0.05	N/A	System Identification
66	PER13	S1E7	Negative	0.85	125	Most Gaps Opening
67	PWN21	White Noise	N/A	0.05	N/A	System Identification
68	PE14	S1E7	Positive	1.02	150	Most Gaps Opening
69	PE15	S1E7	Positive	0.85	125	Most Gaps Opening
70	PWN22	White Noise	N/A	0.05	N/A	System Identification
71	PER15	S1E7	Negative	0.85	125	Most Gaps Opening
72	PWN23	White Noise	N/A	0.05	N/A	System Identification
73	PE16	S1E7	Positive	1.02	150	Most Gaps Opening

**Table 6-2 Test Protocol for the SCPT Frame Model (continued)**

Test No.	Test Label	Excitation	Direction	PGA		Note
				g	%	
74	PWN24	White Noise	N/A	0.05	N/A	System Identification
75	PE16A	S1E7	Positive	1.02	150	Most Gaps Opening
76	PWN25	White Noise	N/A	0.05	N/A	System Identification
77	PER16	S1E7	Negative	1.02	150	Most Gaps Opening
78	PWN26	White Noise	N/A	0.05	N/A	System Identification
Round bars were welded on the top of the beam flanges after Test no. 78.						
79	PWN27	White Noise	N/A	0.05	N/A	System Identification
80	PWN28	White Noise	N/A	0.05	N/A	System Identification
81	PE17	S1E7	Positive	0.17	25	Elastic Range
82	PWN29	White Noise	N/A	0.05	N/A	System Identification
83	PER17	S1E7	Negative	0.17	25	Elastic Range
84	PWN30	White Noise	N/A	0.05	N/A	System Identification
85	PE18	S1E7	Positive	0.34	50	Elastic Range
86	PWN31	White Noise	N/A	0.05	N/A	System Identification
87	PER18	S1E7	Negative	0.34	50	Elastic Range
88	PWN32	White Noise	N/A	0.05	N/A	System Identification
89	PE19	S1E7	Positive	0.48	70	Gap begins opening
90	PWN33	White Noise	N/A	0.05	N/A	System Identification
91	PER19	S1E7	Negative	0.48	70	Gap begins opening

**Table 6-2 Test Protocol for the SCPT Frame Model (continued)**

Test No.	Test Label	Excitation	Direction	PGA		Note
				g	%	
92	PWN34	White Noise	N/A	0.05	N/A	System Identification
93	PE20	S1E7	Positive	0.58	85	Most Gaps Opening
94	PWN35	White Noise	N/A	0.05	N/A	System Identification
95	PER20	S1E7	Negative	0.58	85	Most Gaps Opening
96	PWN36	White Noise	N/A	0.05	N/A	System Identification
97	PE21	S1E7	Positive	0.68	100	Most Gaps Opening
98	PWN37	White Noise	N/A	0.05	N/A	System Identification
99	PER21	S1E7	Negative	0.68	100	Most Gaps Opening
100	PWN38	White Noise	N/A	0.05	N/A	System Identification
101	PE22	S1E7	Positive	0.85	125	Most Gaps Opening
102	PWN39	White Noise	N/A	0.05	N/A	System Identification
103	PER22	S1E7	Negative	0.85	125	Most Gaps Opening
104	PWN40	White Noise	N/A	0.05	N/A	System Identification
105	PE23	S1E7	Positive	1.02	150	Most Gaps Opening
106	PWN41	White Noise	N/A	0.05	N/A	System Identification
107	PER23	S1E7	Negative	1.02	150	Most Gaps Opening
108	PWN42	White Noise	N/A	0.05	N/A	System Identification

\* S1E7 is the scaled MCEER simulated ground motion for event 7 at scenario 1 with 10% probability in 50 years

Note: 1<sup>st</sup> phase: Test No. 1-41; 2<sup>nd</sup> phase: Test No. 42-78; 3<sup>rd</sup> phase: Test No. 79-108

**Table 6-3 Test Protocol for SMRF Model**

Test No.	Test Label	Excitation	Direction	PGA		Note
				(g)	%	
1	MWN01	White Noise	N/A	0.05	N/A	System Identification
2	ME01	S1E7	Positive	0.17	25	Elastic Range
3	MWN02	White Noise	N/A	0.05	N/A	System Identification
4	MER01	S1E7	Negative	0.17	25	Elastic Range
5	MWN03	White Noise	N/A	0.05	N/A	System Identification
6	ME02	S1E7	Positive	0.34	50	Elastic Range
7	MWN04	White Noise	N/A	0.05	N/A	System Identification
8	MER02	S1E7	Negative	0.34	50	Elastic Range
9	MWN05	White Noise	N/A	0.05	N/A	System Identification
10	MER02A	S1E7	Negative	0.34	50	Elastic Range
11	MWN05A	White Noise	N/A	0.05	N/A	System Identification
12	ME02A	S1E7	Positive	0.34	50	Elastic Range
13	MWN05B	White Noise	N/A	0.05	N/A	System Identification
14	ME03	S1E7	Positive	0.48	70	Elastic Range
15	MWN06	White Noise	N/A	0.05	N/A	System Identification
16	MER03	S1E7	Negative	0.48	70	Elastic Range
17	MWN07	White Noise	N/A	0.05	N/A	System Identification
18	ME04	S1E7	Positive	0.58	85	Elastic Range
19	MWN08	White Noise	N/A	0.05	N/A	System Identification

**Table 6-3 Test Protocol for SMRF Model (continued)**

Test No.	Test Label	Excitation	Direction	PGA		Note
				(g)	%	
20	MER04	S1E7	Negative	0.58	85	Elastic Range
21	MWN09	White Noise	N/A	0.05	N/A	System Identification
22	ME05	S1E7	Positive	0.68	100	Elastic Range
23	MWN10	White Noise	N/A	0.05	N/A	System Identification
24	MER05	S1E7	Negative	0.68	100	Elastic Range
25	MWN11	White Noise	N/A	0.05	N/A	System Identification
26	ME06	S1E7	Positive	0.85	125	Beams Yielding
27	ME06A	S1E7	Positive	0.85	125	Beams Yielding
28	MWN12	White Noise	N/A	0.05	N/A	System Identification
29	MER06	S1E7	Negative	0.85	125	Beams Yielding
30	MWN13	White Noise	N/A	0.05	N/A	System Identification
31	ME07	S1E7	Positive	1.02	150	Beams Yielding
32	MWN14	White Noise	N/A	0.05	N/A	System Identification
33	MER07	S1E7	Negative	1.02	150	Beams Yielding
34	MWN15	White Noise	N/A	0.05	N/A	System Identification

## 6.4 Instrumentation

### 6.4.1 Instrumentation for the Self-Centering Post-Tensioned (SCPT) Frame Model

As shown in Table 6-4, a total of 126 channels of data acquisition were used to measure the global and local responses of the SCPT frame model during the shake table testing. The accelerometers were installed at each floor and base plate to measure the horizontal and vertical acceleration responses. The displacement responses at each level were captured by string potentiometers and tempsonic transducers. The potentiometers were attached to the top and bottom of the beam flanges to measure the gap opening in the beam-column connections of the SCPT frame. Load cells were installed at the end of the post-tensioned strands in order to measure the changes in prestress forces. Since both models are symmetric, the strain gages were attached to the beams and columns in only one span to measure the local response of the beams and columns. As shown in Fig. 6-17, the load cells were fabricated by drilling round steel bars and installing strain gages to form a four-element Wheatstone bridge circuit. The position of all the instrumentations used on the SCPT frame model is shown in Fig. 6-18.



Figure 6-17 Load Cell

**Table 6-4 Instrumentation List for SCPT Frame Model**

Tag Name	Channel No.	Sensor Type	Position & Function
AC0-1	1	Accelerometer	Base Plate, Acceleration
AC1-1	2	Accelerometer	Model frame in the 1 <sup>st</sup> floor, Acceleration
AC1-2	3	Accelerometer	Front Mass plate in the 1 <sup>st</sup> floor, Acceleration
AC1-3	4	Accelerometer	Back Mass plate in the 1 <sup>st</sup> floor, Acceleration
AC2-1	5	Accelerometer	Model frame in the 2 <sup>nd</sup> floor, Acceleration
AC2-2	6	Accelerometer	Front Mass plate in the 2 <sup>nd</sup> floor, Acceleration
AC2-3	7	Accelerometer	Back Mass plate in the 2 <sup>nd</sup> floor, Acceleration
AC3-1	8	Accelerometer	Model frame in the 3 <sup>rd</sup> floor, Acceleration
AC3-2	9	Accelerometer	Front Mass plate in the 3 <sup>rd</sup> floor, Acceleration
AC3-3	10	Accelerometer	Back Mass plate in the 3 <sup>rd</sup> floor, Acceleration
AC0-2	11	Accelerometer	Left end of Base Plate, Vertical Acceleration
AC0-3	12	Accelerometer	Left end of Base Concrete Boat, Vertical Acceleration
AC0-4	13	Accelerometer	Right end of Base Plate, Vertical Acceleration
AC0-5	14	Accelerometer	Right end of Base Concrete Boat, Vertical Acceleration
SP0-1	15	String Pot	Base Plate, Displacement
TE1-1	16	Temposonic	Model frame in the 1 <sup>st</sup> floor, Displacement
TE1-2	17	Temposonic	Front Mass plate in the 1 <sup>st</sup> floor, Displacement
TE1-3	18	Temposonic	Back Mass plate in the 1 <sup>st</sup> floor, Displacement
TE2-1	19	Temposonic	Model frame in the 2 <sup>nd</sup> floor, Displacement

**Table 6-4 Instrumentation List for SCPT Frame Model (continued)**

Tag Name	Channel No.	Sensor Type	Position & Function
TE2-2	20	Temposonic	Front Mass plate in the 2 <sup>nd</sup> floor, Displacement
TE2-3	21	Temposonic	Back Mass plate in the 2 <sup>nd</sup> floor, Displacement
TE3-1	22	Temposonic	Model frame in the 3 <sup>rd</sup> floor, Displacement
TE3-2	23	Temposonic	Front Mass plate in the 3 <sup>rd</sup> floor, Displacement
TE3-3	24	Temposonic	Back Mass plate in the 3 <sup>rd</sup> floor, Displacement
PM1-1	25	Potentiometer	Left beam, 1 <sup>st</sup> floor, left end, top beam flange
PM1-2	26	Potentiometer	Left beam, 1 <sup>st</sup> floor, left end, bottom beam flange
PM1-3	27	Potentiometer	Left beam, 1 <sup>st</sup> floor, right end, top beam flange
PM1-4	28	Potentiometer	Left beam, 1 <sup>st</sup> floor, right end, bottom beam flange
PM1-5	29	Potentiometer	Right beam, 1 <sup>st</sup> floor, left end, top beam flange
PM1-6	30	Potentiometer	Right beam, 1 <sup>st</sup> floor, left end, bottom beam flange
PM1-7	31	Potentiometer	Right beam, 1 <sup>st</sup> floor, right end, top beam flange
PM1-8	32	Potentiometer	Right beam, 1 <sup>st</sup> floor, right end, bottom beam flange
PM2-1	33	Potentiometer	Left beam, 2nd floor, left end, top beam flange
PM2-2	34	Potentiometer	Left beam, 2nd floor, left end, bottom beam flange
PM2-3	35	Potentiometer	Left beam, 2nd floor, right end, top beam flange
PM2-4	36	Potentiometer	Left beam, 2nd floor, right end, bottom beam flange
PM2-5	37	Potentiometer	Right beam, 2nd floor, left end, top beam flange

**Table 6-4 Instrumentation List for SCPT Frame Model (continued)**

Tag Name	Channel No.	Sensor Type	Position & Function
PM2-6	38	Potentiometer	Right beam, 2nd floor, left end, bottom beam flange
PM2-7	39	Potentiometer	Right beam, 2nd floor, right end, top beam flange
PM2-8	40	Potentiometer	Right beam, 2nd floor, right end, bottom beam flange
PM3-1	41	Potentiometer	Left beam, 3rd floor, left end, top beam flange
PM3-2	42	Potentiometer	Left beam, 3rd floor, left end, bottom beam flange
PM3-3	43	Potentiometer	Left beam, 3rd floor, right end, top beam flange
PM3-4	44	Potentiometer	Left beam, 3rd floor, right end, bottom beam flange
PM3-5	45	Potentiometer	Right beam, 3rd floor, left end, top beam flange
PM3-6	46	Potentiometer	Right beam, 3rd floor, left end, bottom beam flange
PM3-7	47	Potentiometer	Right beam, 3rd floor, right end, top beam flange
PM3-8	48	Potentiometer	Right beam, 3rd floor, right end, bottom beam flange
LC1-1	49	Load Cell	Model frame, 1st floor, front side.
LC1-2	50	Load Cell	Model frame, 1st floor, back side
LC2-1	51	Load Cell	Model frame, 2nd floor, front side
LC2-2	52	Load Cell	Model frame, 2nd floor, back side
LC3-1	53	Load Cell	Model frame, 3rd floor, front side
LC3-2	54	Load Cell	Model frame, 3rd floor, back side
SGB1-1	55	Strain Gauge	Left beam, 1st floor, left end, front side, top beam flange

**Table 6-4 Instrumentation List for SCPT Frame Model (continued)**

Tag Name	Channel No.	Sensor Type	Position & Function
SGB1-2	56	Strain Gauge	Left beam, 1 <sup>st</sup> floor, left end, back side, top beam flange
SGB1-3	57	Strain Gauge	Left beam, 1 <sup>st</sup> floor, left end, front side, bottom beam flange
SGB1-4	58	Strain Gauge	Left beam, 1 <sup>st</sup> floor, left end, back side, bottom beam flange
SGB1-5	59	Strain Gauge	Left beam, 1 <sup>st</sup> floor, right end, front side, top beam flange
SGB1-6	60	Strain Gauge	Left beam, 1 <sup>st</sup> floor, right end, back side, top beam flange
SGB1-7	61	Strain Gauge	Left beam, 1 <sup>st</sup> floor, right end, front side, bottom beam flange
SGB1-8	62	Strain Gauge	Left beam, 1 <sup>st</sup> floor, right end, back side, bottom beam flange
SGB1-9	63	Strain Gauge	Left beam, top beam flange, right end of left reinforcing plate, 1 <sup>st</sup> floor
SGB1-10	64	Strain Gauge	Left beam, top beam flange, right end of left reinforcing plate, 1 <sup>st</sup> floor
SGB1-11	65	Strain Gauge	Left beam, bottom beam flange, right end of left reinforcing plate, 1 <sup>st</sup> floor
SGB1-12	66	Strain Gauge	Left beam, bottom beam flange, right end of left reinforcing plate, 1 <sup>st</sup> floor
SGB1-13	67	Strain Gauge	Left beam, top beam flange, left end of right reinforcing plate, 1 <sup>st</sup> floor
SGB1-14	68	Strain Gauge	Left beam, top beam flange, left end of right reinforcing plate, 1 <sup>st</sup> floor
SGB1-15	69	Strain Gauge	Left beam, bottom beam flange, left end of right reinforcing plate, 1 <sup>st</sup> floor
SGB1-16	70	Strain Gauge	Left beam, bottom beam flange, left end of right reinforcing plate, 1 <sup>st</sup> floor
SGB2-1	71	Strain Gauge	Left beam, 2 <sup>nd</sup> floor, left end, front side, top beam flange
SGB2-2	72	Strain Gauge	Left beam, 2 <sup>nd</sup> floor, left end, back side, top beam flange
SGB2-3	73	Strain Gauge	Left beam, 2 <sup>nd</sup> floor, left end, front side, bottom beam flange

**Table 6-4 Instrumentation List for SCPT Frame Model (continued)**

Tag Name	Channel No.	Sensor Type	Position & Function
SGB2-4	74	Strain Gauge	Left beam, 2 <sup>nd</sup> floor, left end, back side, bottom beam flange
SGB2-5	75	Strain Gauge	Left beam, 2 <sup>nd</sup> floor, right end, front side, top beam flange
SGB2-6	76	Strain Gauge	Left beam, 2 <sup>nd</sup> floor, right end, back side, top beam flange
SGB2-7	77	Strain Gauge	Left beam, 2 <sup>nd</sup> floor, right end, front side, bottom beam flange
SGB2-8	78	Strain Gauge	Left beam, 2 <sup>nd</sup> floor, right end, back side, bottom beam flange
SGB2-9	79	Strain Gauge	Left beam, top beam flange, right end of left reinforcing plate, 2 <sup>nd</sup> floor
SGB2-10	80	Strain Gauge	Left beam, top beam flange, right end of left reinforcing plate, 2 <sup>nd</sup> floor
SGB2-11	81	Strain Gauge	Left beam, bottom beam flange, right end of left reinforcing plate, 2 <sup>nd</sup> floor
SGB2-12	82	Strain Gauge	Left beam, bottom beam flange, right end of left reinforcing plate, 2 <sup>nd</sup> floor
SGB2-13	83	Strain Gauge	Left beam, top beam flange, left end of right reinforcing plate, 2 <sup>nd</sup> floor
SGB2-14	84	Strain Gauge	Left beam, top beam flange, left end of right reinforcing plate, 2 <sup>nd</sup> floor
SGB2-15	85	Strain Gauge	Left beam, bottom beam flange, left end of right reinforcing plate, 2 <sup>nd</sup> floor
SGB2-16	86	Strain Gauge	Left beam, bottom beam flange, left end of right reinforcing plate, 2 <sup>nd</sup> floor
SGB3-1	87	Strain Gauge	Left beam, 3 <sup>rd</sup> floor, left end, front side, top beam flange
SGB3-2	88	Strain Gauge	Left beam, 3 <sup>rd</sup> floor, left end, back side, top beam flange
SGB3-3	89	Strain Gauge	Left beam, 3 <sup>rd</sup> floor, left end, front side, bottom beam flange
SGB3-4	90	Strain Gauge	Left beam, 3 <sup>rd</sup> floor, left end, back side, bottom beam flange
SGB3-5	91	Strain Gauge	Left beam, 3 <sup>rd</sup> floor, right end, front side, top beam flange

**Table 6-4 Instrumentation List for SCPT Frame Model (continued)**

Tag Name	Channel No.	Sensor Type	Position & Function
SGB3-6	92	Strain Gauge	Left beam, 3 <sup>rd</sup> floor, right end, back side, top beam flange
SGB3-7	93	Strain Gauge	Left beam, 3 <sup>rd</sup> floor, right end, front side, bottom beam flange
SGB3-8	94	Strain Gauge	Left beam, 3 <sup>rd</sup> floor, right end, back side, bottom beam flange
SGB3-9	95	Strain Gauge	Left beam, top beam flange, right end of left reinforcing plate, 3 <sup>rd</sup> floor
SGB3-10	96	Strain Gauge	Left beam, top beam flange, right end of left reinforcing plate, 3 <sup>rd</sup> floor
SGB3-11	97	Strain Gauge	Left beam, bottom beam flange, right end of left reinforcing plate, 3 <sup>rd</sup> floor
SGB3-12	98	Strain Gauge	Left beam, bottom beam flange, right end of left reinforcing plate, 3 <sup>rd</sup> floor
SGB3-13	99	Strain Gauge	Left beam, top beam flange, left end of right reinforcing plate, 3 <sup>rd</sup> floor
SGB3-14	100	Strain Gauge	Left beam, top beam flange, left end of right reinforcing plate, 3 <sup>rd</sup> floor
SGB3-15	101	Strain Gauge	Left beam, bottom beam flange, left end of right reinforcing plate, 3 <sup>rd</sup> floor
SGB3-16	102	Strain Gauge	Left beam, bottom beam flange, left end of right reinforcing plate, 3 <sup>rd</sup> floor
SGC0-1	103	Strain Gauge	Left base column, left flange, Strain
SGC0-2	104	Strain Gauge	Left base column, left flange, Strain
SGC0-3	105	Strain Gauge	Left base column, right flange, Strain
SGC0-4	106	Strain Gauge	Left base column, right flange, Strain
SGC0-5	107	Strain Gauge	Middle base column, left flange, Strain
SGC0-6	108	Strain Gauge	Middle base column, left flange, Strain
SGC0-7	109	Strain Gauge	Middle base column, right flange, Strain

**Table 6-4 Instrumentation List for SCPT Frame Model (continued)**

Tag Name	Channel No.	Sensor Type	Position & Function
SGC0-8	110	Strain Gauge	Middle base column, right flange, Strain
SGC0-9	111	Strain Gauge	Right base column, left flange, Strain
SGC0-10	112	Strain Gauge	Right base column, left flange, Strain
SGC0-11	113	Strain Gauge	Right base column, right flange, Strain
SGC0-12	114	Strain Gauge	Right base column, right flange, Strain
SGC1-1	115	Strain Gauge	Left beam-column connection, 1 <sup>st</sup> floor, left lower column flange
SGC1-2	116	Strain Gauge	Left beam-column connection, 1 <sup>st</sup> floor, left lower column flange
SGC1-3	117	Strain Gauge	Left beam-column connection, 1 <sup>st</sup> floor, right lower column flange
SGC1-4	118	Strain Gauge	Left beam-column connection, 1 <sup>st</sup> floor, right lower column flange
SGC1-5	119	Strain Gauge	Left beam-column connection, 1 <sup>st</sup> floor, left upper column flange
SGC1-6	120	Strain Gauge	Left beam-column connection, 1 <sup>st</sup> floor, left upper column flange
SGC1-7	121	Strain Gauge	Left beam-column connection, 1 <sup>st</sup> floor, right upper column flange
SGC1-8	122	Strain Gauge	Left beam-column connection, 1 <sup>st</sup> floor, right upper column flange
SGC1-9	123	Strain Gauge	Middle beam-column connection, 1 <sup>st</sup> floor, left lower column flange
SGC1-10	124	Strain Gauge	Middle beam-column connection, 1 <sup>st</sup> floor, left lower column flange
SGC1-11	125	Strain Gauge	Middle beam-column connection, 1 <sup>st</sup> floor, right lower column flange
SGC1-12	126	Strain Gauge	Middle beam-column connection, 1 <sup>st</sup> floor, right lower column flange

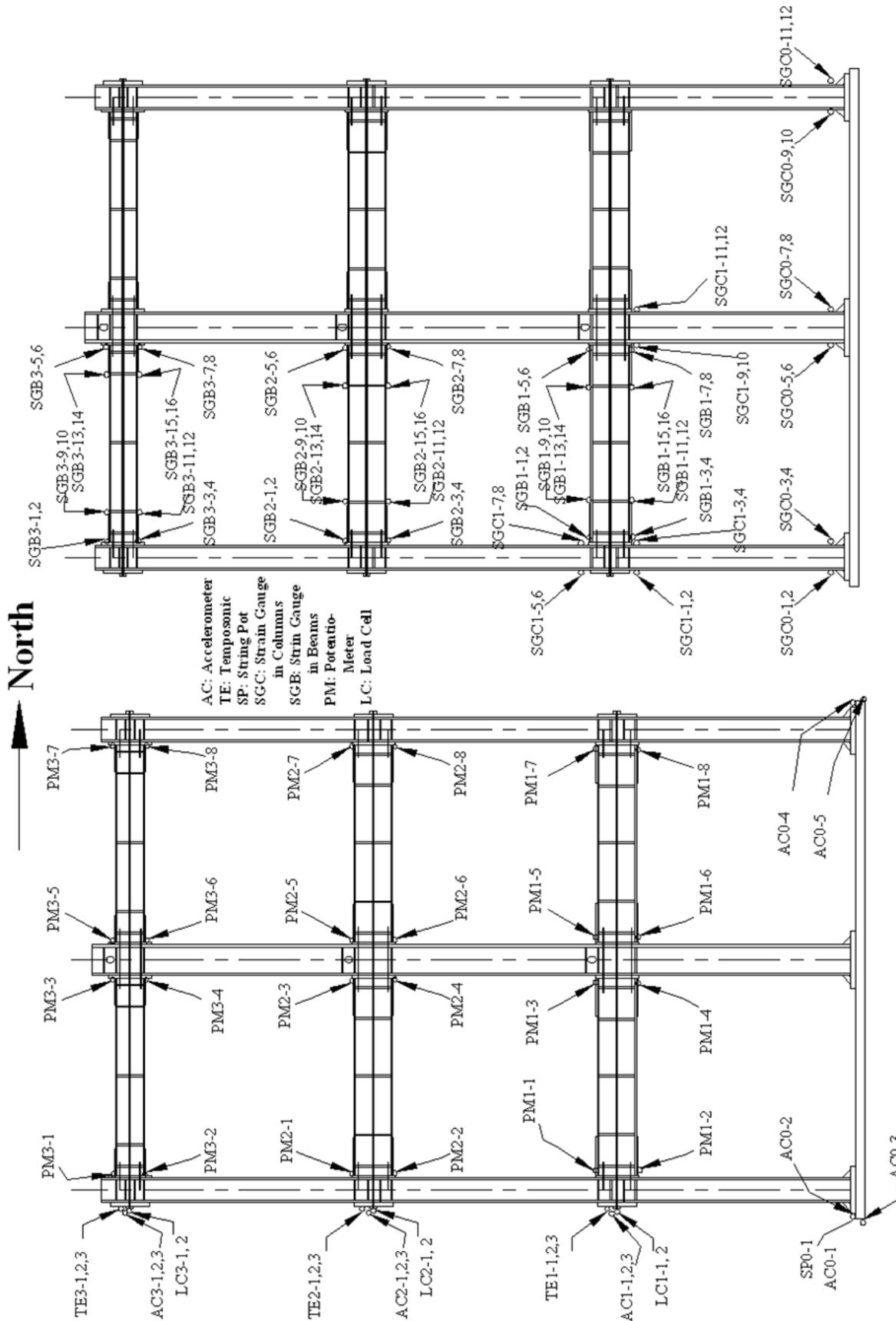


Figure 6-18 Position of Instrumentations for the SCPT Frame Model  
 (Tag names listed in Table 6-4)

#### 6.4.2 Instrumentation for the Steel Moment Resisting Frame (SMRF) Model

As shown in Table 6-5 and Fig. 6-19, the instrumentation for the SMRF model was reduced compared to the one used on the SCPT model. The accelerometers, string potentiometers, temposonic transducers and strain gauges remained in the same position as in the SCPT model. The potentiometers and load cells were removed. The number of strain gages was also reduced.

**Table 6-5 Instrumentation List for SMRF Model**

Tag Name	Channel No.	Sensor Type	Position & Function
SGB1-1	1	Strain Gauge	Left beam, 1 <sup>st</sup> floor, left end, front side, top beam flange
SGB1-2	2	Strain Gauge	Left beam, 1 <sup>st</sup> floor, left end, back side, top beam flange
SGB1-3	3	Strain Gauge	Left beam, 1 <sup>st</sup> floor, left end, front side, bottom beam flange
SGB1-4	4	Strain Gauge	Left beam, 1 <sup>st</sup> floor, left end, back side, bottom beam flange
SGB1-5	5	Strain Gauge	Left beam, 1 <sup>st</sup> floor, right end, front side, top beam flange
SGB1-6	6	Strain Gauge	Left beam, 1 <sup>st</sup> floor, right end, back side, top beam flange
SGB1-7	7	Strain Gauge	Left beam, 1 <sup>st</sup> floor, right end, front side, bottom beam flange
SGB1-8	8	Strain Gauge	Left beam, 1 <sup>st</sup> floor, right end, back side, bottom beam flange
SGB2-1	9	Strain Gauge	Left beam, 2 <sup>nd</sup> floor, left end, front side, top beam flange
SGB2-2	10	Strain Gauge	Left beam, 2 <sup>nd</sup> floor, left end, back side, top beam flange
SGB2-3	11	Strain Gauge	Left beam, 2 <sup>nd</sup> floor, left end, front side, bottom beam flange
SGB2-4	12	Strain Gauge	Left beam, 2 <sup>nd</sup> floor, left end, back side, bottom beam flange

**Table 6-5 Instrumentation List for SMRF Model (continued)**

Tag Name	Channel No.	Sensor Type	Position & Function
SGB2-5	13	Strain Gauge	Left beam, 2 <sup>nd</sup> floor, right end, front side, top beam flange
SGB2-6	14	Strain Gauge	Left beam, 2 <sup>nd</sup> floor, right end, back side, top beam flange
SGB2-7	15	Strain Gauge	Left beam, 2 <sup>nd</sup> floor, right end, front side, bottom beam flange
SGB2-8	16	Strain Gauge	Left beam, 2 <sup>nd</sup> floor, right end, back side, bottom beam flange
SGB3-1	17	Strain Gauge	Left beam, 3 <sup>rd</sup> floor, left end, front side, top beam flange
SGB3-2	18	Strain Gauge	Left beam, 3 <sup>rd</sup> floor, left end, back side, top beam flange
SGB3-3	19	Strain Gauge	Left beam, 3 <sup>rd</sup> floor, left end, front side, bottom beam flange
SGB3-4	20	Strain Gauge	Left beam, 3 <sup>rd</sup> floor, left end, back side, bottom beam flange
SGB3-5	21	Strain Gauge	Left beam, 3 <sup>rd</sup> floor, right end, front side, top beam flange
SGB3-6	22	Strain Gauge	Left beam, 3 <sup>rd</sup> floor, right end, back side, top beam flange
SGB3-7	23	Strain Gauge	Left beam, 3 <sup>rd</sup> floor, right end, front side, bottom beam flange
SGB3-8	24	Strain Gauge	Left beam, 3 <sup>rd</sup> floor, right end, back side, bottom beam flange
SGC0-1	25	Strain Gauge	Left base column, left flange, Strain
SGC0-2	26	Strain Gauge	Left base column, left flange, Strain
SGC0-3	27	Strain Gauge	Left base column, right flange, Strain
SGC0-4	28	Strain Gauge	Left base column, right flange, Strain
SGC0-5	29	Strain Gauge	Middle base column, left flange, Strain
SGC0-6	30	Strain Gauge	Middle base column, left flange, Strain

**Table 6-5 Instrumentation List for SMRF Model (continued)**

Tag Name	Channel No.	Sensor Type	Position & Function
SGC0-7	31	Strain Gauge	Middle base column, right flange, Strain
SGC0-8	32	Strain Gauge	Middle base column, right flange, Strain
SGC0-9	33	Strain Gauge	Right base column, left flange, Strain
SGC0-10	34	Strain Gauge	Right base column, left flange, Strain
SGC0-11	35	Strain Gauge	Right base column, right flange, Strain
SGC0-12	36	Strain Gauge	Right base column, right flange, Strain
SGC1-1	37	Strain Gauge	Left beam-column connection, 1 <sup>st</sup> floor, left lower column flange
SGC1-2	38	Strain Gauge	Left beam-column connection, 1 <sup>st</sup> floor, left lower column flange
SGC1-3	39	Strain Gauge	Left beam-column connection, 1 <sup>st</sup> floor, right lower column flange
SGC1-4	40	Strain Gauge	Left beam-column connection, 1 <sup>st</sup> floor, right lower column flange
SGC1-5	41	Strain Gauge	Left beam-column connection, 1 <sup>st</sup> floor, left upper column flange
SGC1-6	42	Strain Gauge	Left beam-column connection, 1 <sup>st</sup> floor, left upper column flange
SGC1-7	43	Strain Gauge	Left beam-column connection, 1 <sup>st</sup> floor, right upper column flange
SGC1-8	44	Strain Gauge	Left beam-column connection, 1 <sup>st</sup> floor, right upper column flange
SGC1-9	45	Strain Gauge	Middle beam-column connection, 1 <sup>st</sup> floor, left lower column flange
SGC1-10	46	Strain Gauge	Middle beam-column connection, 1 <sup>st</sup> floor, left lower column flange
SGC1-11	47	Strain Gauge	Middle beam-column connection, 1 <sup>st</sup> floor, right lower column flange
SGC1-12	48	Strain Gauge	Middle beam-column connection, 1 <sup>st</sup> floor, right lower column flange

**Table 6-5 Instrumentation List for SMRF Model (continued)**

Tag Name	Channel No.	Sensor Type	Position & Function
SP0-1	49	String Pot	Base Plate, Displacement
TE1-1	50	Temposonic	Model frame in the 1 <sup>st</sup> floor, Displacement
TE1-2	51	Temposonic	Front Mass plate in the 1 <sup>st</sup> floor, Displacement
TE1-3	52	Temposonic	Back Mass plate in the 1 <sup>st</sup> floor, Displacement
TE2-1	53	Temposonic	Model frame in the 2 <sup>nd</sup> floor, Displacement
TE2-2	54	Temposonic	Front Mass plate in the 2 <sup>nd</sup> floor, Displacement
TE2-3	55	Temposonic	Back Mass plate in the 2 <sup>nd</sup> floor, Displacement
TE3-1	56	Temposonic	Model frame in the 3 <sup>rd</sup> floor, Displacement
TE3-2	57	Temposonic	Front Mass plate in the 3 <sup>rd</sup> floor, Displacement
TE3-3	58	Temposonic	Back Mass plate in the 3 <sup>rd</sup> floor, Displacement
AC0-1	59	Accelerometer	Base Plate, Acceleration
AC1-1	60	Accelerometer	Model frame in the 1 <sup>st</sup> floor, Acceleration
AC1-2	61	Accelerometer	Front Mass plate in the 1 <sup>st</sup> floor, Acceleration
AC1-3	62	Accelerometer	Back Mass plate in the 1 <sup>st</sup> floor, Acceleration
AC2-1	63	Accelerometer	Model frame in the 2 <sup>nd</sup> floor, Acceleration
AC2-2	64	Accelerometer	Front Mass plate in the 2 <sup>nd</sup> floor, Acceleration
AC2-3	65	Accelerometer	Back Mass plate in the 2 <sup>nd</sup> floor, Acceleration
AC3-1	66	Accelerometer	Model frame in the 3 <sup>rd</sup> floor, Acceleration

**Table 6-5 Instrumentation List for SMRF Model (continued)**

Tag Name	Channel No.	Sensor Type	Position & Function
AC3-2	67	Accelerometer	Front Mass plate in the 3 <sup>rd</sup> floor, Acceleration
AC3-3	68	Accelerometer	Back Mass plate in the 3 <sup>rd</sup> floor, Acceleration
AC0-2	69	Accelerometer	Left end of Base Plate, <b>Vertical</b> Acceleration
AC0-3	70	Accelerometer	Left end of Base Concrete Boat, <b>Vertical</b> Acceleration
AC0-4	71	Accelerometer	Right end of Base Plate, <b>Vertical</b> Acceleration
AC0-5	72	Accelerometer	Right end of Base Concrete Boat, <b>Vertical</b> Acceleration

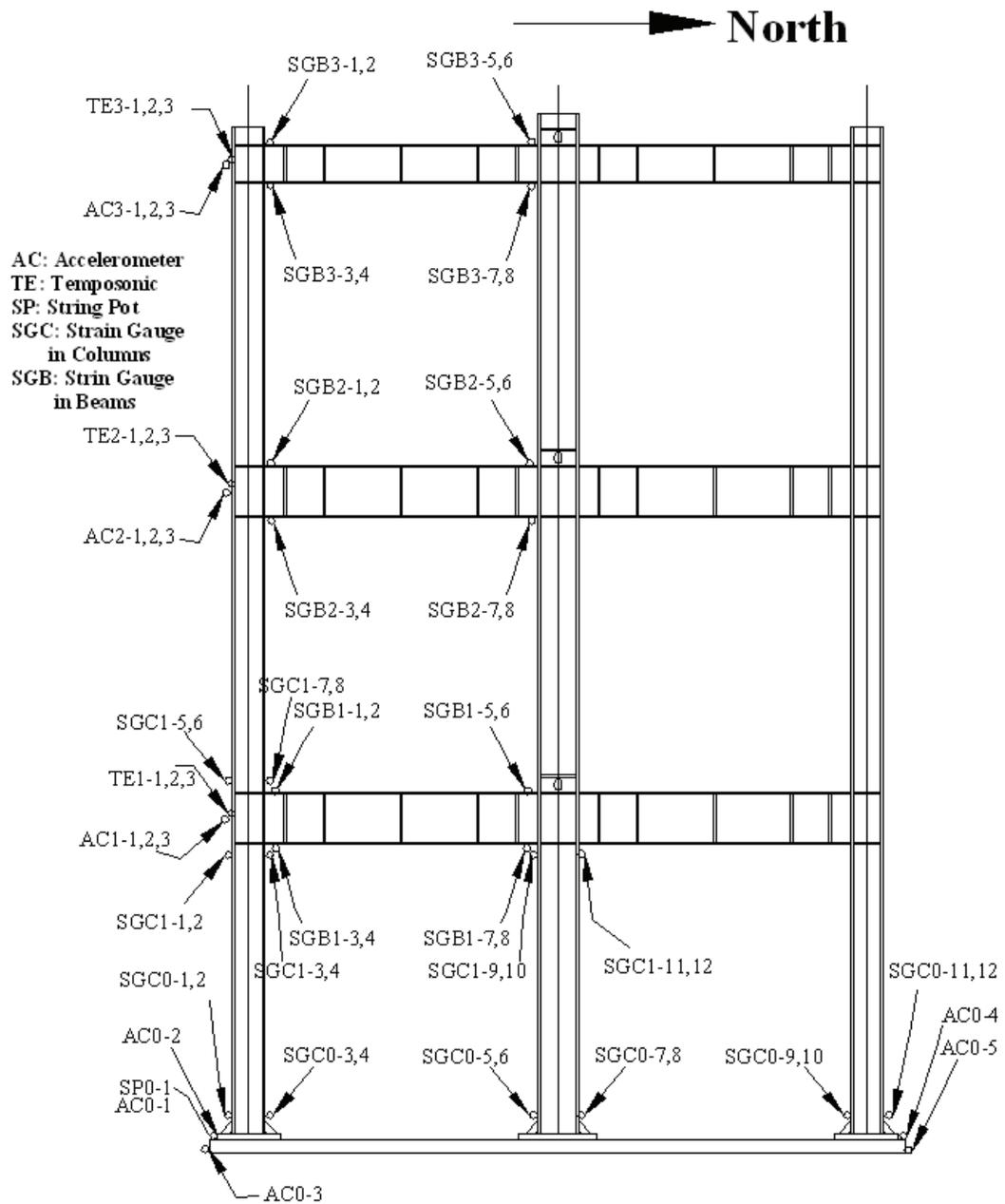


Figure 6-19 Position of Instrumentations for the SMRF Model  
 (Tag names listed in Table 6-5)



## **CHAPTER 7**

### **EXPERIMENTAL RESULTS**

This chapter discusses the experimental results of two series of seismic tests conducted according to the testing program described in Chapter VI. The Steel Moment Resisting Frame (SMRF) and Self-Centering Post-Tensioned (SCPT) frame models were used in these two series of tests under different seismic excitations. In order to identify the dynamic characteristics of those two models, white noise, snap-back and sine sweep tests were carried out and the results are compared. Then, the shake table fidelity is evaluated. Thereafter, the global response in terms of the acceleration and displacement of the model structures and the local response with respect to the individual structural components are presented. Finally, the seismic energy dissipation is evaluated.

Throughout this chapter, the experimental results are discussed by comparing the responses between the SCPT and SMRF models. This comparison is used to experimentally verify the conclusions obtained numerically in previous chapters. Note that the test nomenclature used in this chapter can be found in the test protocols presented in Section 6.3.

#### **7.1 Identification of Dynamic Characteristics**

The dynamic characteristics of a structure are considered to be the natural period/frequency, stiffness, mode shape and damping. To identify these properties for the frame models, different types of tests including white noise, snap-back and sine-sweep tests were conducted. Based on the results of those tests, the dynamic properties of the SCPT and SMRF models

are calculated. Finally, a comparison of the structural properties obtained from different tests is presented and discussed at the end of this section.

### **7.1.1 White Noise Tests**

White noise tests were conducted before the seismic tests to identify the initial dynamic properties of the frame models and then carried out after every seismic test to verify any change of the structural characteristics. These white noise excitations had an amplitude of 0.05 g with a flat frequency band from 0.5 to 50 Hz. Two different amplitudes (0.1g and 0.15g) were also used to see if the structural properties changed with the excitation amplitude.

Using the experimental results in terms of the ground acceleration and the accelerations of the three stories of the frame models, transfer functions were calculated from the ratio of the Fourier transform amplitudes between the output (floor accelerations) and the input (ground acceleration) in the frequency domain. Based on the transfer functions, the natural frequencies of the structures in various modes were determined by the frequencies with peak magnitudes. The mode shapes were also obtained by comparing the amplitudes of the transfer functions of different floors. The directions of the various mode shape components were determined from the phase values between the different transfer functions. The well known half-power (bandwidth) method (Chopra, 2000) was used to calculate the equivalent viscous damping ratios.

The transfer functions in the frequency domain obtained from the white noise tests PWN1B (0.1g), PWN1C (0.15g), PWN1D (0.1g) and PWN2A for the SCPT model and from the white noise tests MWN01, MWN02 and MWN03 for the SMRF model, are shown in Figs. 7-1 to 7-7. Among these tests, tests PWN1D, PWN2A, MWN02 and MWN03 were conducted after a seismic test with 25% intensity (see Section 6.3). Those tests were selected to calculate the initial dynamic properties since both the SCPT and SMRF structures remained elastic without any change in terms of dynamic properties under low intensity seismic tests. Based on those transfer functions, the natural frequencies, mode shapes and damping ratios were determined as shown in Table 7-1 and 7-2.

Table 7-1 indicates that the natural frequencies of the SMRF model are identical for all three white noise tests conducted initially. As seen in Table 7-2, the natural frequencies of the SCPT model in the 1<sup>st</sup> mode are slightly different across the various tests, which may be due to the different amplitude of the white noise tests. Also, the mode shapes and damping ratios, obtained from those white noise tests, are similar for all tests. Basically, those calculated properties are similar under the same type excitation (white noise). Therefore, the average values in terms of natural frequencies, mode shapes and damping ratios were determined to represent the initial dynamic characteristics of the two models as a result of the white noise tests. Further evaluation of these results will be presented in the summary of Sec. 7.1.4.

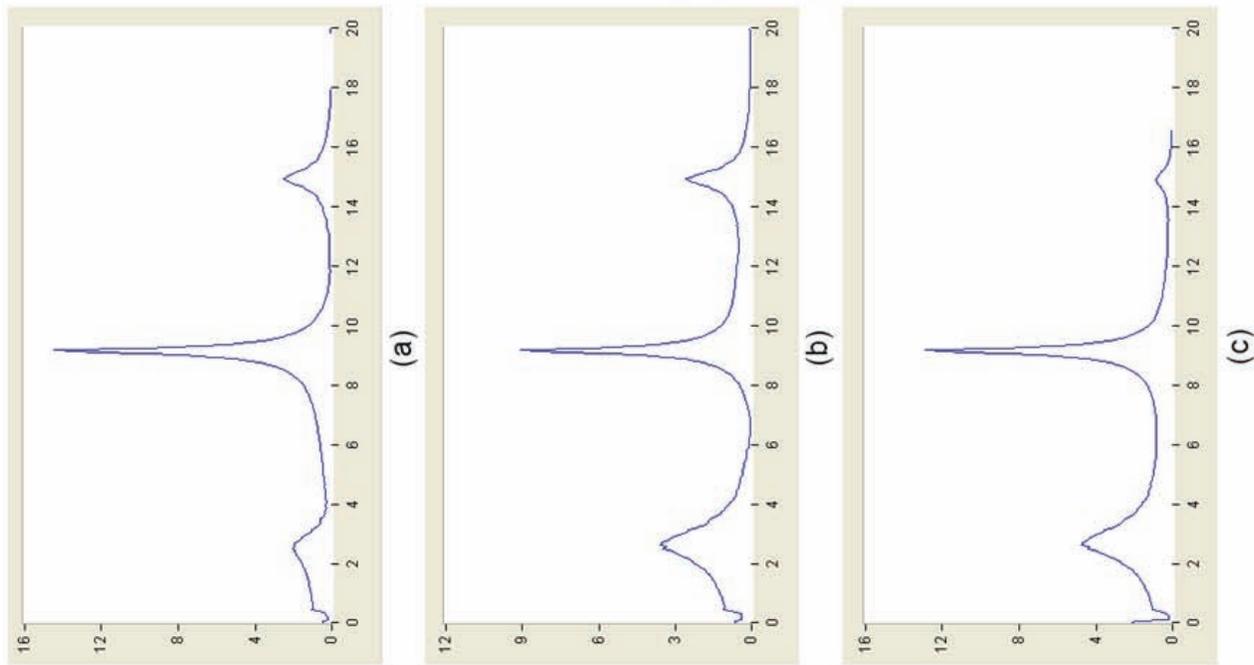


Figure 7-1 Transfer Function of SCPT model for Test PWN1B  
 (a) 1<sup>st</sup> floor (b) 2<sup>nd</sup> floor (c) 3<sup>rd</sup> floor

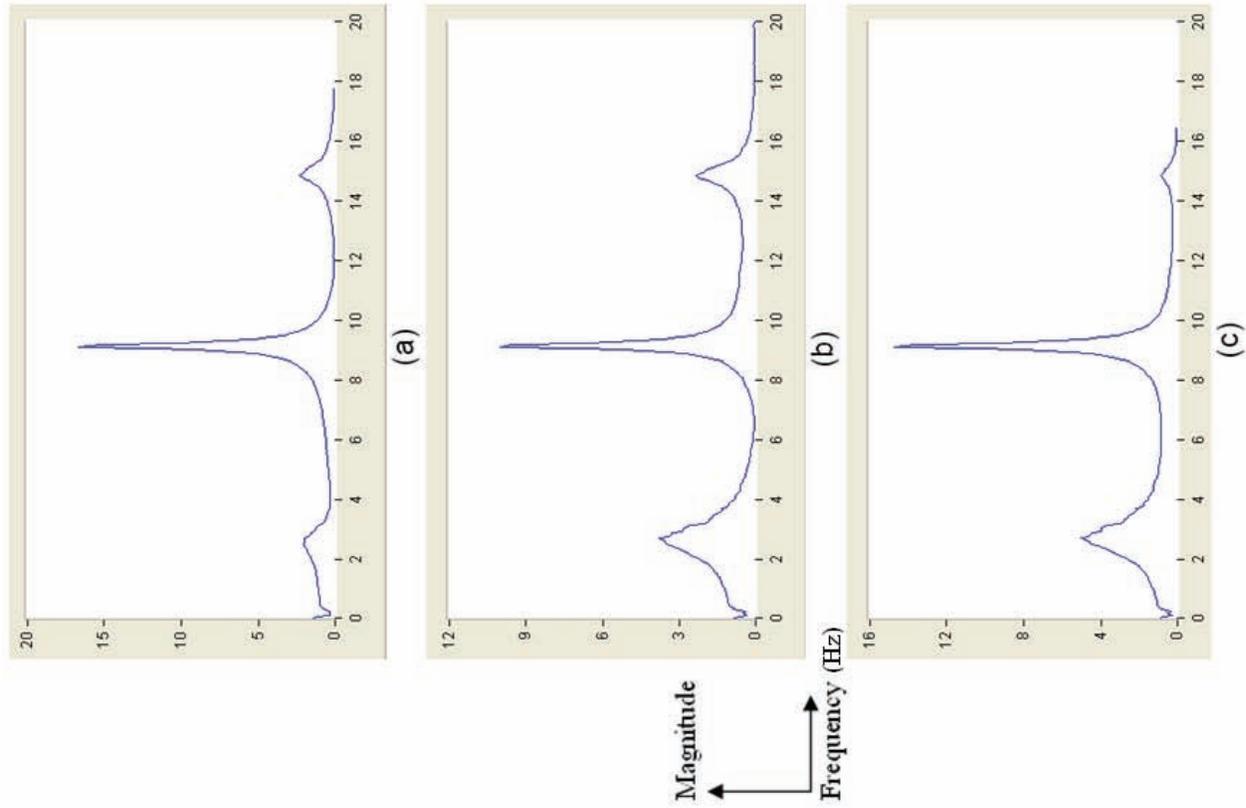


Figure 7-2 Transfer Function of SCPT model for Test PWN1C  
 (a) 1<sup>st</sup> floor (b) 2<sup>nd</sup> floor (c) 3<sup>rd</sup> floor

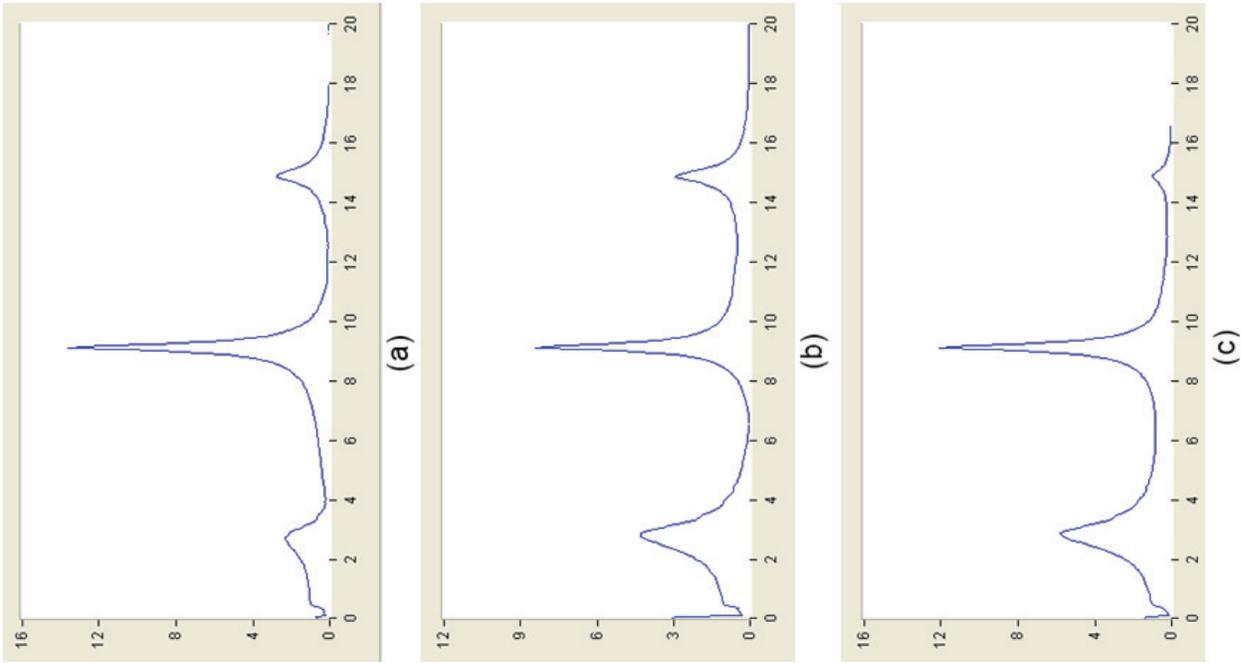


Figure 7-3 Transfer Function of SCPT model for Test PWN1D  
 (a) 1<sup>st</sup> floor (b) 2<sup>nd</sup> floor (c) 3<sup>rd</sup> floor

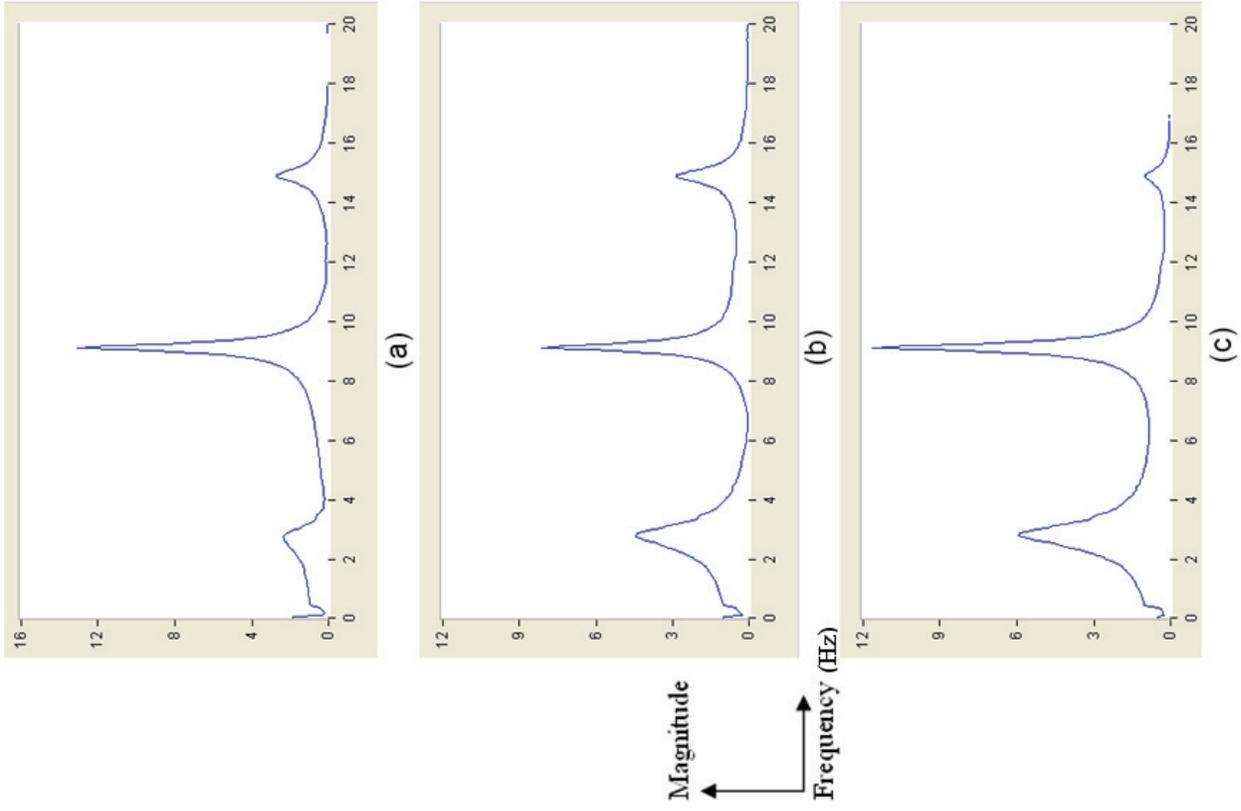


Figure 7-4 Transfer Function of SCPT model for Test PWN2A  
 (a) 1<sup>st</sup> floor (b) 2<sup>nd</sup> floor (c) 3<sup>rd</sup> floor

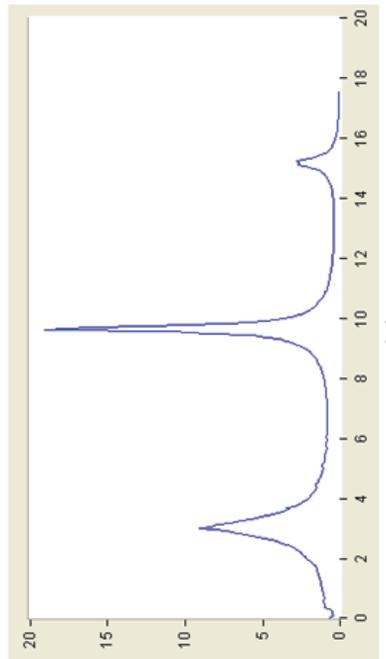
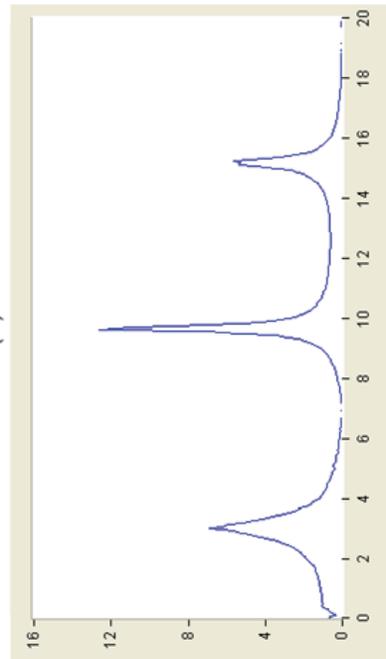
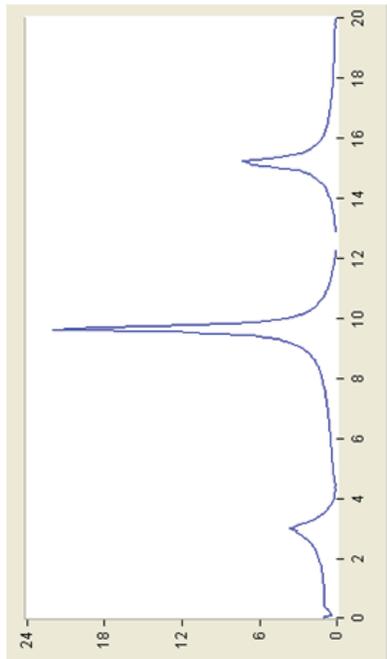


Figure 7-5 Transfer Function of SMRF model for Test MWN01  
 (a) 1<sup>st</sup> floor (b) 2<sup>nd</sup> floor (c) 3<sup>rd</sup> floor

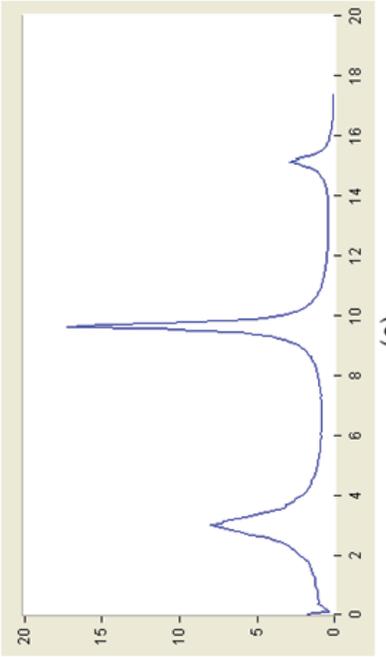
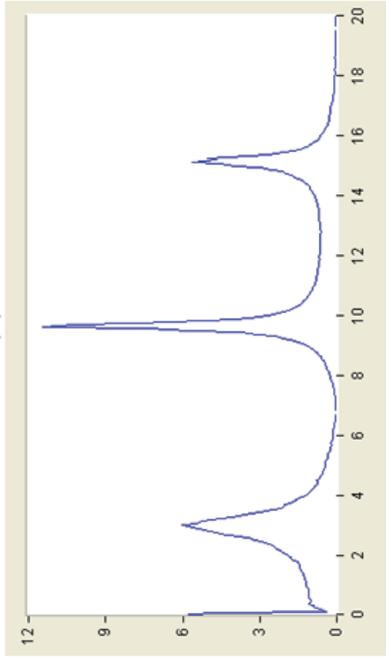
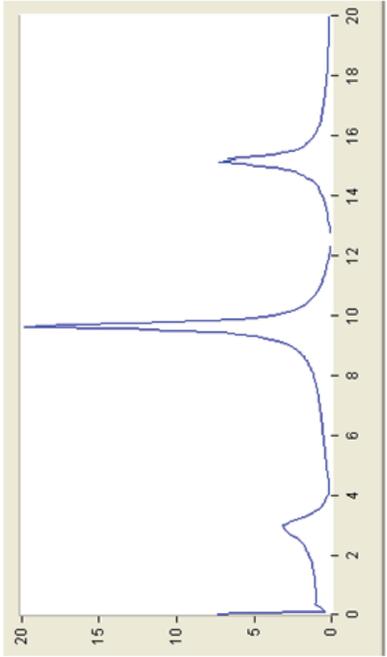


Figure 7-6 Transfer Function of SMRF model for Test MWN02  
 (a) 1<sup>st</sup> floor (b) 2<sup>nd</sup> floor (c) 3<sup>rd</sup> floor

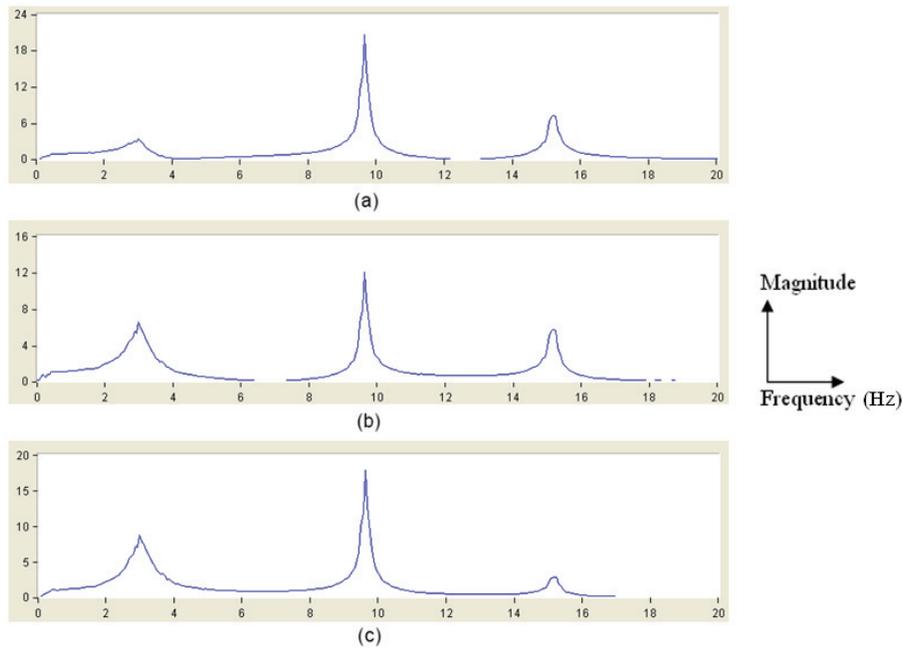


Figure 7-7 Transfer Function of SMRF model for Test MWN03  
 (a) 1<sup>st</sup> floor (b) 2<sup>nd</sup> floor (3) 3<sup>rd</sup> floor

**Table 7-1 Dynamic Properties of the SMRF model obtained from initial white noise tests**

Test No.	Mode	Frequency (Hz)	Mode Shape $\Phi$			Damping Ratio $\xi$ (%)
			1 <sup>st</sup> floor	2 <sup>nd</sup> floor	3 <sup>rd</sup> floor	
MWN01	1 <sup>st</sup>	3.000	0.405	0.756	1.000	14.8
	2 <sup>nd</sup>	9.625	1.000	0.573	-0.870	2.4
	3 <sup>rd</sup>	15.250	1.000	-0.769	0.385	1.8
MWN02	1 <sup>st</sup>	3.000	0.398	0.751	1.000	16.8
	2 <sup>nd</sup>	9.625	1.000	0.576	-0.868	2.6
	3 <sup>rd</sup>	15.125	1.000	-0.774	0.400	1.9
MWN03	1 <sup>st</sup>	3.000	0.399	0.754	1.000	15.4
	2 <sup>nd</sup>	9.625	1.000	0.582	-0.861	2.6
	3 <sup>rd</sup>	15.188	1.000	-0.785	0.389	1.9
Average	1 <sup>st</sup>	3	0.401	0.754	1	15.7
	2 <sup>nd</sup>	9.625	1	0.577	-0.866	2.5
	3 <sup>rd</sup>	15.188	1	-0.776	0.391	1.9

**Table 7-2 Dynamic Properties of the SCPT model obtained from initial white noise tests**

Test No.	Mode	Frequency (Hz)	Mode Shape $\Phi$			Damping Ratio $\xi$ (%)
			1 <sup>st</sup> floor	2 <sup>nd</sup> floor	3 <sup>rd</sup> floor	
PWN1B	1 <sup>st</sup>	2.625	0.424	0.762	1	16.9
	2 <sup>nd</sup>	9.188	1	0.628	-0.892	0.9
	3 <sup>rd</sup>	14.938	-0.974	1	-0.353	1.5
PWN1C	1 <sup>st</sup>	2.688	0.401	0.752	1	15.6
	2 <sup>nd</sup>	9.125	1	0.598	-0.884	1.0
	3 <sup>rd</sup>	14.875	-0.98	1	-0.363	1.7
PWN1D	1 <sup>st</sup>	2.875	0.373	0.742	1	13.7
	2 <sup>nd</sup>	9.125	1	0.617	-0.89	1.2
	3 <sup>rd</sup>	14.875	-0.951	1	-0.354	1.3
PWN2A	1 <sup>st</sup>	2.813	0.389	0.753	1	13.4
	2 <sup>nd</sup>	9.125	1	0.621	-0.893	1.3
	3 <sup>rd</sup>	14.875	-0.943	1	-0.354	1.1
Average	1 <sup>st</sup>	2.75	0.397	0.752	1	14.9
	2 <sup>nd</sup>	9.141	1	0.616	-0.89	1.1
	3 <sup>rd</sup>	14.891	-0.962	1	-0.356	1.4

According to the theory of linear structural dynamics, the stiffness matrix and damping matrix of the frame models can be calculated with the obtained natural frequencies, mode shapes and damping ratios. The stiffness matrix,  $K$ , is defined by:

$$K = M\Phi_n\Omega\Phi_n^T M \quad (7.1)$$

where:

$M$  : Mass matrix. For these two modes with lumped mass in each floor, the mass matrix is diagonal matrix ([45.21, 45.14, 44.81]) lb-sec<sup>2</sup>/in.

$\Phi_n$  : Mass normalized mode shape matrix.  $\Phi_n^T M \Phi_n = I$ , where  $I$  is the unit matrix.

$\Omega$  : Diagonal natural frequency matrix  $[\omega_1^2, \omega_2^2, \dots, \omega_i^2, \dots, \omega_n^2]$ .

$\omega_i$  : Circular natural frequency of the i-th mode (rad/sec)

The damping matrix, C, is defined as follows:

$$C = M\Phi_n \zeta \Phi_n^T M \quad (7.2)$$

where:

$\zeta$  : Diagonal matrix  $[2\xi_1\omega_1, 2\xi_2\omega_2, \dots, 2\xi_i\omega_i, \dots, 2\xi_n\omega_n]$ .

$\xi_i$  : Damping ratio of the i-th mode.

By using Eq. (7.1) and (7.2) as well as the results indicated in Table 7-1 and 7-2, the stiffness matrix and damping matrix of the two model structures were calculated as shown in Table 7-3.

**Table 7-3 Stiffness and Damping Matrices of the SCPT and SMRF models**

	Stiffness Matrix $K$ (kips/in)	Damping Matrix $C$ ( lb-sec/in)
SCPT Model	$\begin{bmatrix} 248.9 & -140.8 & 7.9 \\ -140.8 & 223.0 & -99.6 \\ 7.9 & -99.6 & 85.4 \end{bmatrix}$	$\begin{bmatrix} 101.3 & 1.2 & 49.8 \\ 1.2 & 144.2 & 66.4 \\ 49.8 & 66.4 & 161.2 \end{bmatrix}$
SMRF Model	$\begin{bmatrix} 316.1 & -133.3 & 26.4 \\ -133.3 & 172.7 & -103.0 \\ 26.4 & -103.0 & 103.1 \end{bmatrix}$	$\begin{bmatrix} 184.4 & 12.4 & 41.6 \\ 12.4 & 166.2 & 55.5 \\ 41.6 & 55.5 & 215.4 \end{bmatrix}$

Comparisons of the transfer functions for most of the white noise tests conducted between seismic tests of increasing intensities are shown in Figs. 7-8 to 7-10. The percentage (i.e. intensity of seismic tests) indicated next to the test label indicates that the white noise test was carried out after that seismic test with that normalized intensity.

From these figures, it can be seen that both frame models maintained similar natural frequencies in each mode after all seismic tests, which indicates that no significant yielding occurred during the seismic tests. Major inelastic deformation (yielding) of the structural components would have damaged the structure and caused the large residual drifts or stresses which would decrease the natural frequencies. The natural frequencies for those tests are also listed in Table 7-4 and 7-5.

For the SCPT model, as shown in Table 7-4, it should be noted that in the 1<sup>st</sup> phase of seismic tests (earthquake excitations with 25% to 150% intensity before PWN9), the natural frequencies increased after the 25%-intensity seismic test and decreased after the 70%-intensity test. It is believed that the reason for the first increase is that the contact surface of the beam to the column in the SCPT model was not machined perfectly flat such that not all the area of the beam end was in contact with the column flange. Therefore, after the 25%-intensity seismic test, that contact area may have increased leading to a higher stiffness of the total structure.

The decrease of natural frequency after the 70% seismic test indicates that some gap openings occurred at the interface of some beam-to-column connections, which is consistent

with the numerical predictions obtained in Chapter VI. Although from a theoretical perspective, the natural frequencies should not change in the SCPT structure when only the Energy-Dissipating (ED) bars yield, the changes in the experimental natural frequencies remain relatively small, which can be considered to be in accordance with the theory. Note also that the natural frequencies are higher after tests PWN9 (2<sup>nd</sup> Phase) and PWN28 (3<sup>rd</sup> Phase) compared to those after tests PWN8 and PWN24, which is attributed to the installation of support bars on the bottom and top of the beams to prevent the unexpected vertical movement of the beams observed during the seismic tests (see Section 7.3).

For the SMRF model, as shown in Table 7-5, the natural frequency in the 1<sup>st</sup> mode remains constant until the 100%-intensity seismic test. After the 125%-intensity seismic test, the natural frequency of 1<sup>st</sup> mode is decreased, which indicates that the beams and/or columns yielded. Note that this decrease is very small, which suggests that the ductility of the yielding is low and do not make severe damage on the structure. The natural frequencies of the 2<sup>nd</sup> and 3<sup>rd</sup> modes were also reduced, which may be related to the yielding. No further explanations are given to the change of the higher-mode natural frequencies.

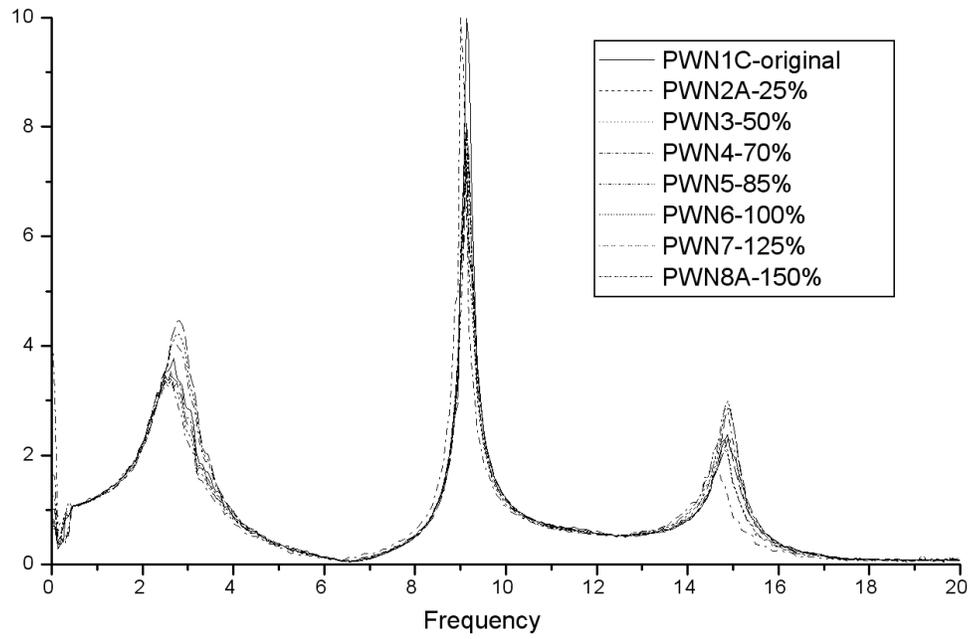


Figure 7-8 Transfer Functions for White Noise Tests, SCPT Model

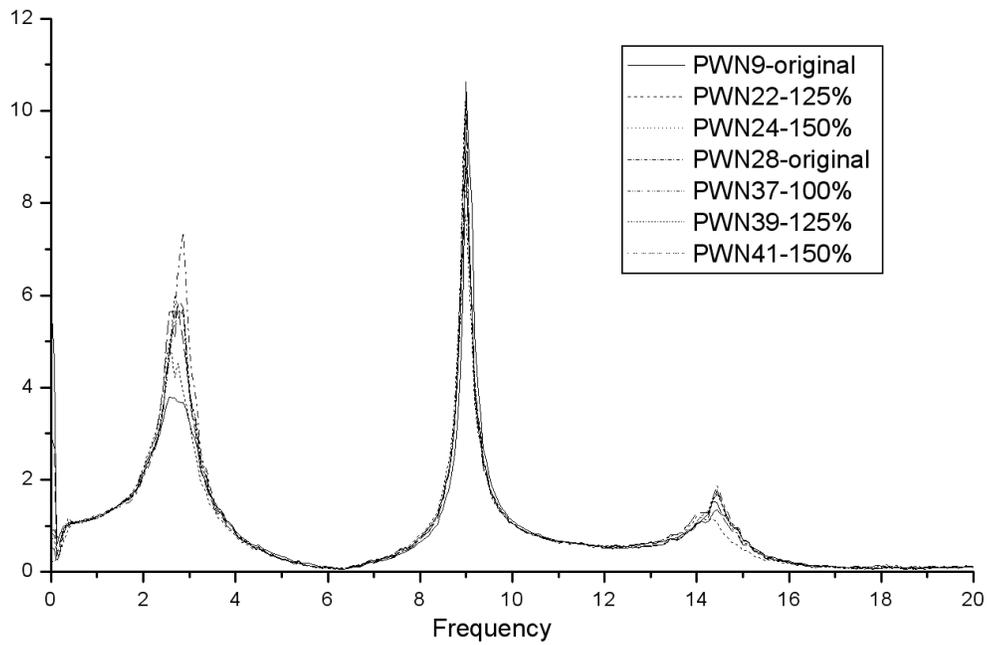


Figure 7-9 Transfer Functions for White Noise Tests, SCPT Model

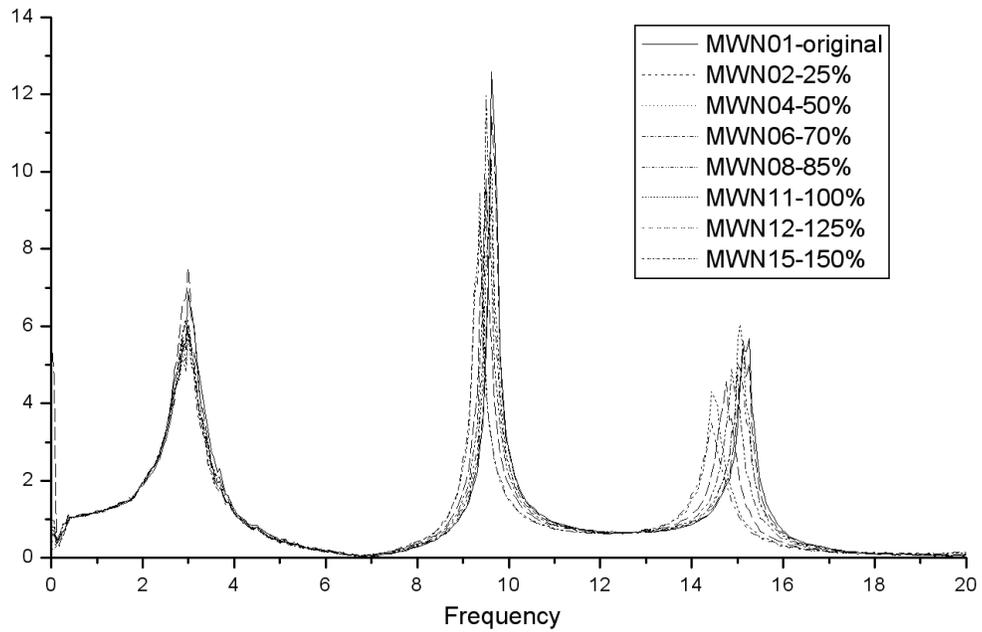


Figure 7-10 Transfer Functions for White Noise Tests, SMRF Model

**Table 7-4 Natural Frequencies of the SCPT Model  
obtained from all white noise tests**

	Natural Frequencies (Hz)							
Test number	PWN1C	PWN2A	PWN3	PWN4	PWN5	PWN6	PWN7	PWN8
1 <sup>st</sup> mode	2.688	2.813	2.813	2.75	2.688	2.625	2.625	2.625
2 <sup>nd</sup> mode	9.125	9.125	9.125	9.125	9.125	9.125	9.125	9
3 <sup>rd</sup> mode	14.875	14.875	14.875	14.813	14.813	14.875	14.75	14.56
Test number	PWN9	PWN22	PWN24	PWN28	PWN37	PWN39	PWN41	
1 <sup>st</sup> mode	2.75	2.75	2.563	2.875	2.875	2.813	2.688	
2 <sup>nd</sup> mode	9	9	9	9	9	9	9	
3 <sup>rd</sup> mode	14.438	14.438	14	14.438	14.438	14.438	14.438	

**Table 7-5 Natural Frequencies of the SMRF Model  
obtained from all white noise tests**

	Natural Frequencies (Hz)							
Test number	MWN01	MWN02	MWN04	MWN06	MWN08	MWN11	MWN12	MWN15
1 <sup>st</sup> mode	3.000	3.000	3.000	3.000	3.000	3.000	2.938	2.938
2 <sup>nd</sup> mode	9.625	9.625	9.625	9.500	9.500	9.500	9.375	9.375
3 <sup>rd</sup> mode	15.250	15.125	15.063	15.000	14.875	14.750	14.438	14.438

### **7.1.2 Snap-Back Tests**

A snap-back test consists in pulling the structure with a relatively low force or to a certain displacement and then quickly releasing it to induce free vibrations. For the two frame models (SCPT & SMRF), snap-back tests were conducted by pulling the structures with a 400~500 lb. force in order to determine their dynamic properties from the recorded acceleration and displacement responses.

By using the same procedure used in the white noise tests, the dynamic characteristics of the two models were calculated through the transfer functions as shown in Fig. 7-11 and 7-12. The calculated results of these dynamic properties will be summarized in Section 7.1.4.

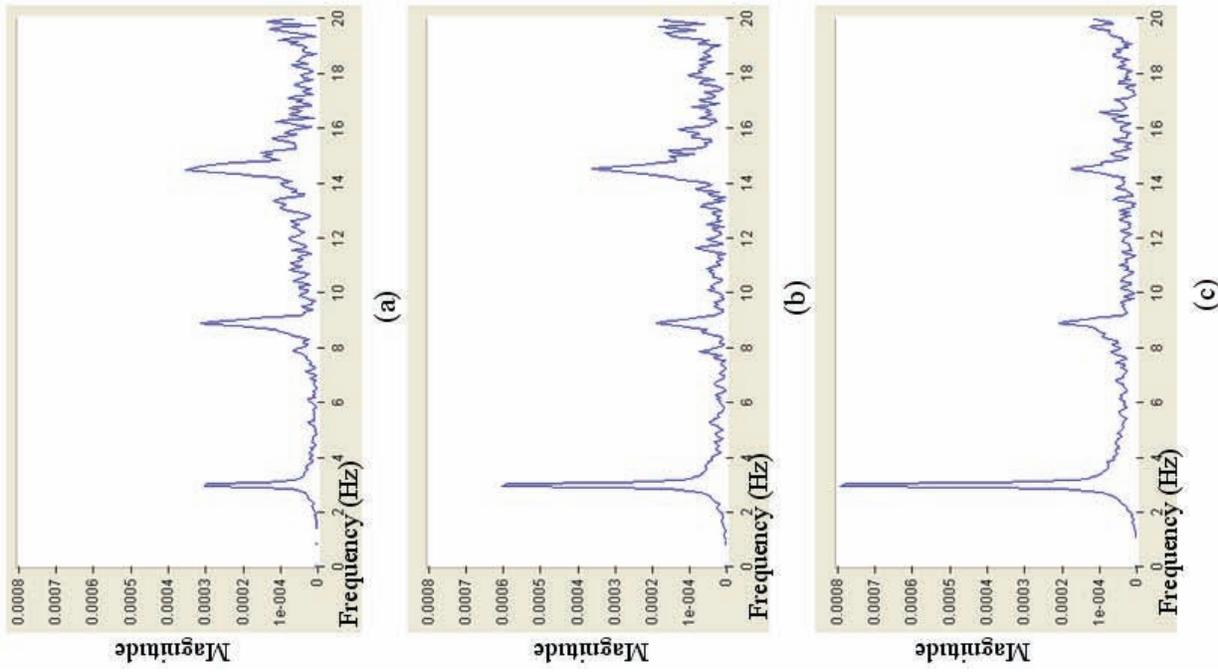


Figure 7-11 Transfer Function of the SCPT model for Snap-Back Test  
 (a) 1<sup>st</sup> floor (b) 2<sup>nd</sup> floor (3) 3<sup>rd</sup> floor

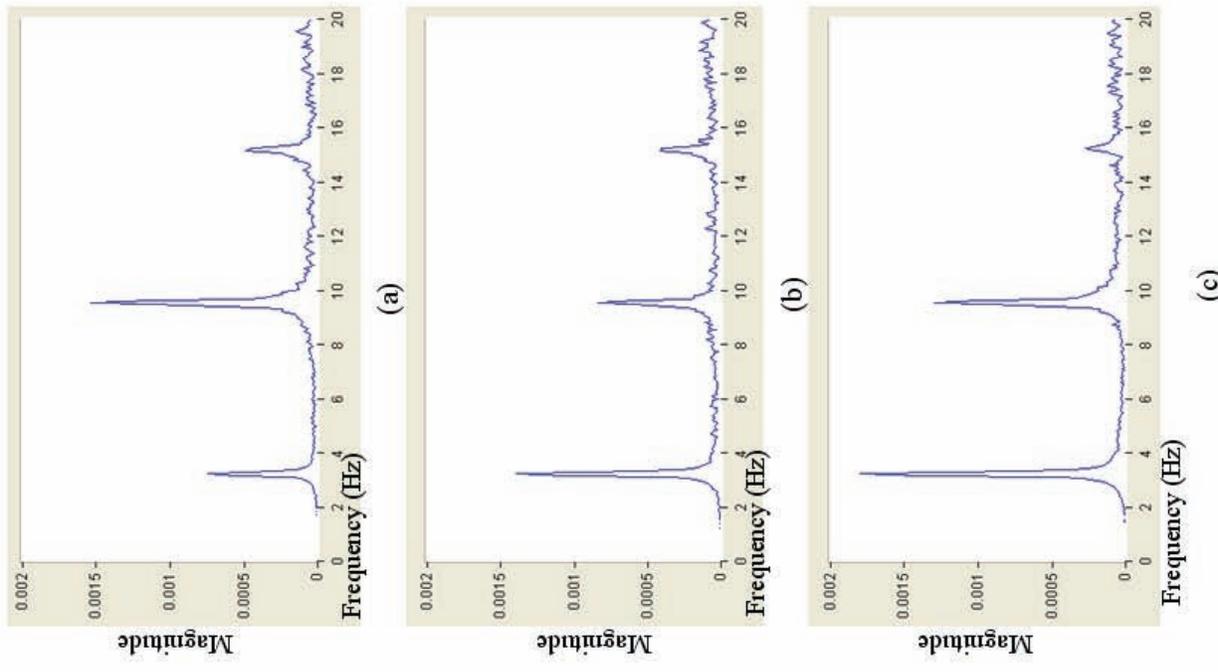
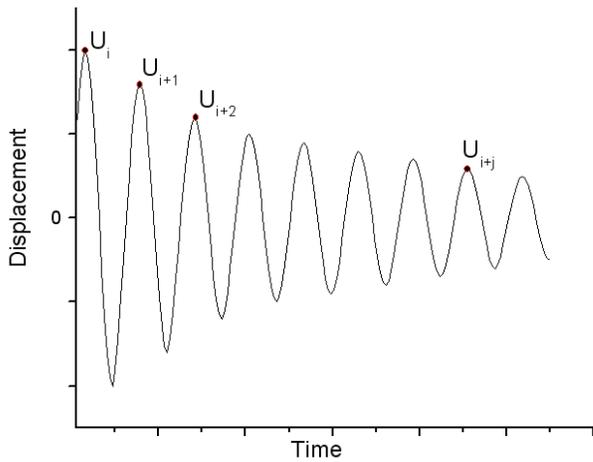


Figure 7-12 Transfer Function of the SMRF model for Snap-Back Test  
 (a) 1<sup>st</sup> floor (b) 2<sup>nd</sup> floor (3) 3<sup>rd</sup> floor

For the damping ratios, except using the half-bandwidth method, another well-known logarithmic decrement decay method (Chopra, 2000) was used. As shown in Fig. 7-13, based on this method, the damping ratio,  $\xi$ , can be estimated as follows:



$$\xi = \frac{1}{2\pi j} \ln \frac{U_i}{U_{i+j}} \quad (7.3)$$

where,  $U_i$  and  $U_{i+j}$  are the displacement amplitudes at the  $i$ -th and  $(i+j)$ -th cycles; and  $j$  is the number of cycles after the  $i$ -th cycle.

Figure 7-13 Displacement time history of a freely vibration system

The displacement time histories of the model structures were notch-filtered around the natural frequencies of the first 3 modes such that only the contribution of a given modal natural frequency remained in the displacement time histories. The selected notch-filtering band is determined by the range of a full peak of the transfer function at a given modal frequency (e.g. as shown in Fig. 7-11, the range 2.6 ~ 3.6 Hz is selected for the 1<sup>st</sup> modal natural frequency 2.973Hz of SCPT modal obtained from the snap-back test as listed in Table 7-6). Figure 7-14 and 7-15 show the displacement responses notch-filtered by the modal frequency for two model structures. By using these filtered displacement time histories, the damping ratios can be estimated by Eq. (7.3). The results of damping ratios calculated from the above method will be also presented in Section 7.1.4.

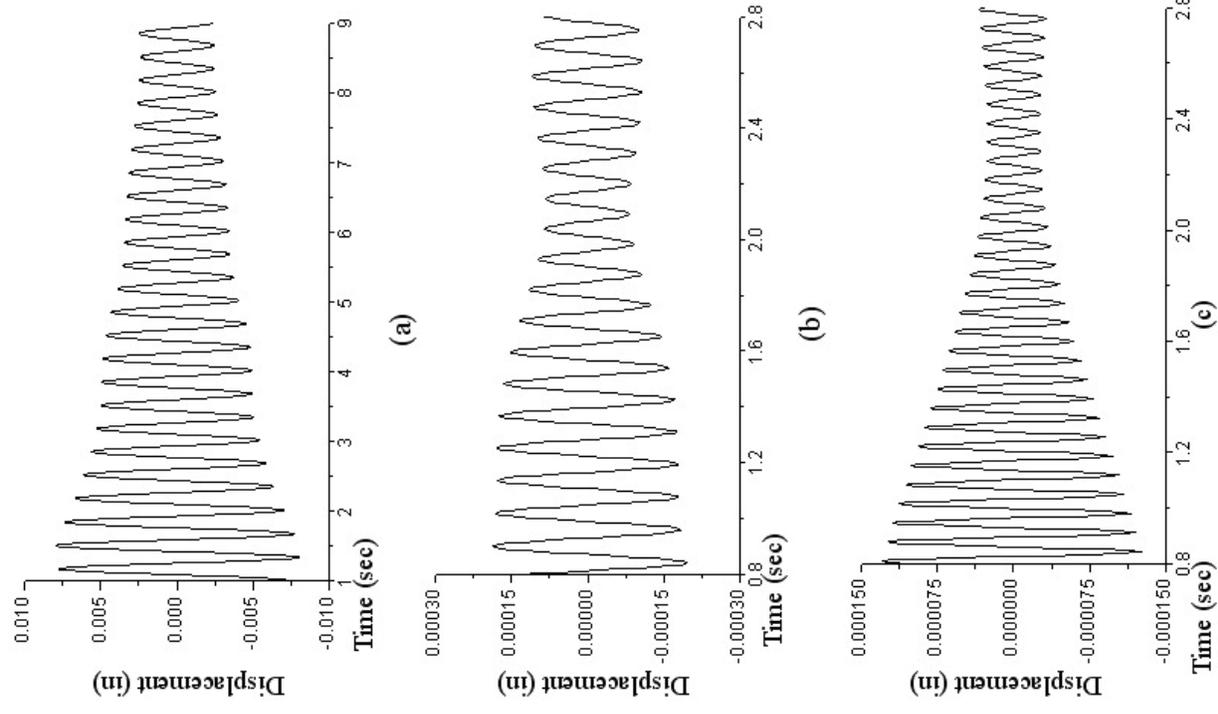


Figure 7-14 Displacement Time History of SCPT Model notch-filtered around the modal frequency of (a) 1<sup>st</sup> mode (b) 2<sup>nd</sup> mode (c) 3<sup>rd</sup> mode

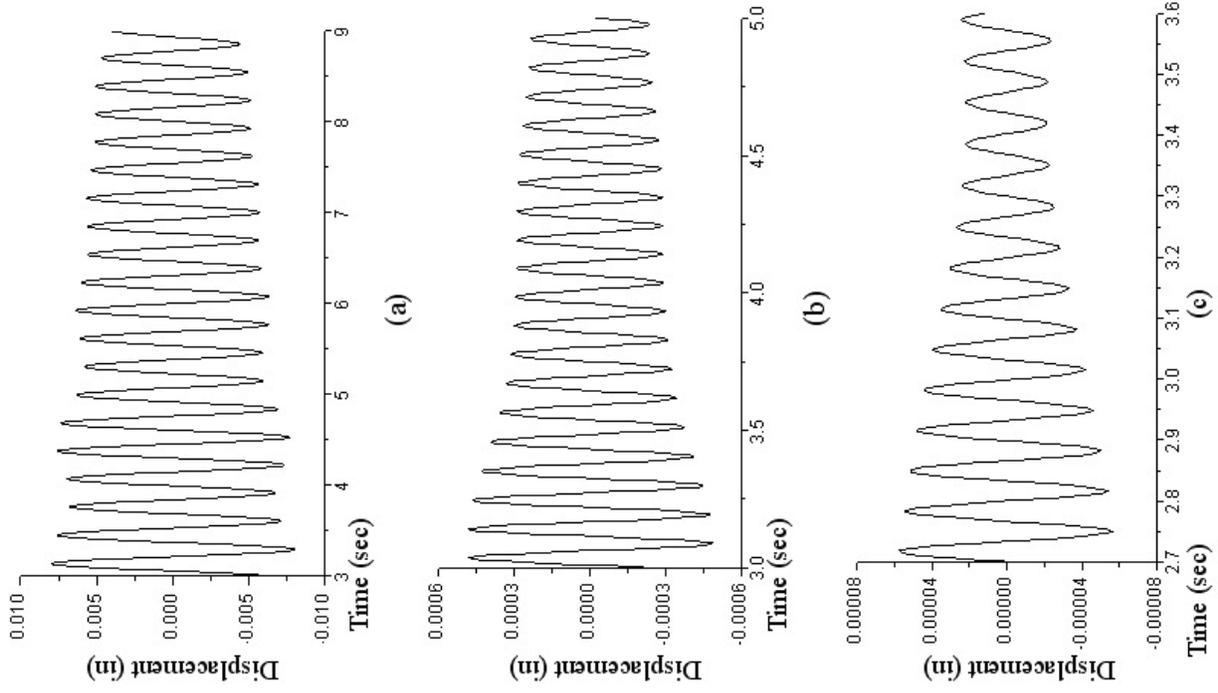


Figure 7-15 Displacement Time History of SMRF Model notch-filtered around the modal frequency of (a) 1<sup>st</sup> mode (b) 2<sup>nd</sup> mode (c) 3<sup>rd</sup> mode

### 7.1.3 Sine-Sweep Test

A sine-sweep test is a dynamic test using a sequence of sinusoidal vibrations as input motion to the shake table. The advantage of a sine-sweep test is that a resonance response occurs when the varying frequency of the sinusoidal signal is close to the natural frequency of the model structure, which leads to a clear peak amplitude in the transfer function. Therefore, the dynamic properties can be easily identified. The disadvantage of this test method is also due to this resonance response, which may damage the model structures. In order to avoid the inelastic deformation of the SMRF model, which would make unrecoverable damage to the structure components, the sine-sweep test was carried out only for the SCPT model.

As shown in Fig. 7-16, the sinusoidal signal used for the sine-sweep test has 5~30 Hz frequency contents and 0.17g acceleration amplitude with a 25 second duration. The transfer functions calculated from the acceleration responses in the frequency domain are shown in Fig. 7-17. The dynamic properties were calculated through the same method used for the white noise tests. All the results are shown in the summary of Section 7.1.4.

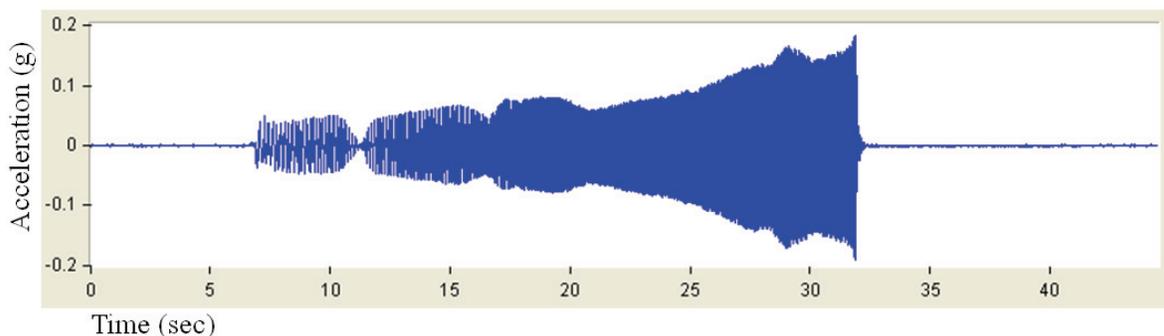


Figure 7-16 Sine-Sweep Excitation used for the SCPT model

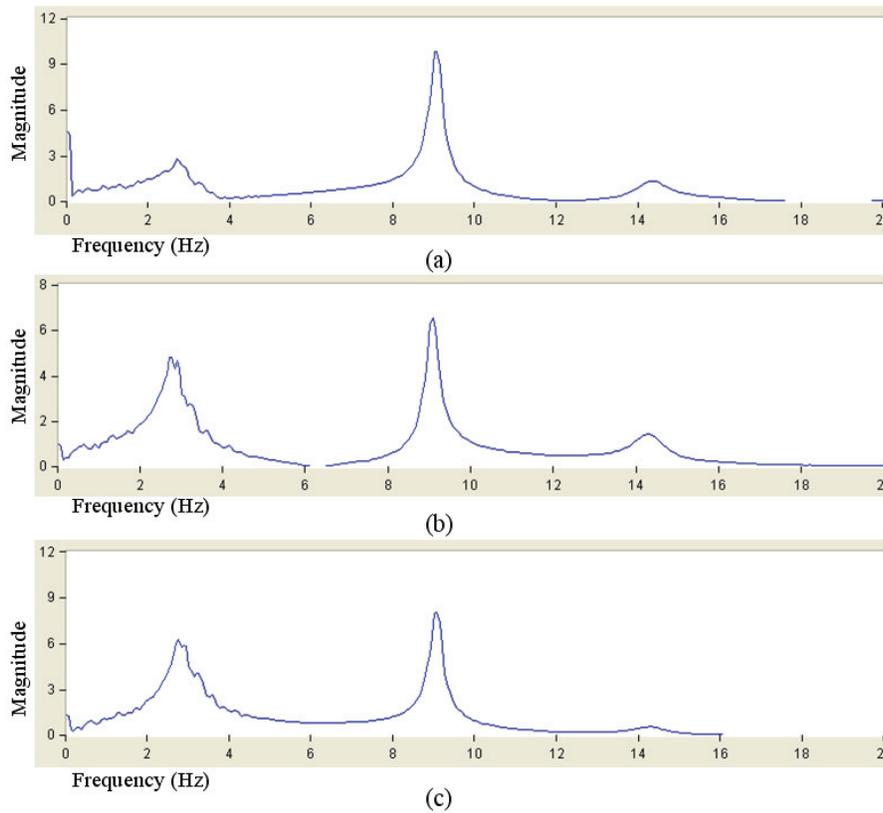


Figure 7-17 Transfer Functions in the SCPT model from the Sine Sweep Test  
 (a) 1<sup>st</sup> floor (b) 2<sup>nd</sup> floor (3) 3<sup>rd</sup> floor

#### 7.1.4 Summary of Identification Tests

To identify the dynamic properties of the SCPT and SMRF model structures, white noise, snap-back and sine-sweep tests were conducted. The dynamic characteristics of the two frame models were calculated based on the transfer functions according to the theory of linear structural dynamics. The results of the identification tests are summarized in Table 7-6 and 7-7, in which the natural frequencies and mode shapes from the white noise tests are average values calculated from Table 7-1 and 7-2. For the snap-back tests, two methods

(half-bandwidth and logarithmic decrement decay methods) were used to calculate the damping ratios.

By comparing the natural frequencies shown in Table 7-6 and 7-7, it is noted that the 1<sup>st</sup> modal natural frequency obtained from the snap-back tests is always slightly higher than those calculated from the white noise and sine-sweep tests for both of two models. The reason for this difference is that for the white noise and sine-sweep tests, the shake table was activated while for the snap-back tests, the shake table was held still and the model structure was pull-released to obtain its free vibration response. Therefore, an additional flexibility was added to the model structure due to the movement of the shake table considering that the shake table was supported by four oil-hydraulic columns (i.e. vertical actuators). Although the stiffness of those supporting columns was very high compared to the low-amplitude white noise and sine-sweep excitations, the structures seems to be slightly “softened” by this effect, leading to the observed reduction of the natural frequency. By comparing the natural frequencies in corresponding tests between the SCPT and SMRF models, the results are always lower for the SCPT model. The reason for this difference is that the beam ends clamped to the column by the Post-Tensioned (PT) strands in the SCPT model were not perfectly flat, thereby reducing the stiffness of the beam-column joints compared to the fully welded beam-column connections in the SMRF model. It is believed that this imperfection of the beam ends led to the slight reduction of the natural frequency for the whole structure. Considering the small difference in natural frequencies between the two models, it should

pointed out that it is consistent with the theoretical assumption that the natural frequencies remains unchanged after redesigning a SMRF using SCPT connections.

As shown in Tables 7-6 and 7-7, the mode shapes calculated from the different identification tests are similar for both SCPT and SMRF models. By comparing the mode shapes of the two models, the 1<sup>st</sup> and 2<sup>nd</sup> modes are essentially identical, while a small difference occurs for the 3<sup>rd</sup> mode. Considering that the seismic response is mainly governed by the lower modes, it can be concluded that both the SCPT and SMRF model structures have similar seismic behavior (i.e. no yielding) in the elastic phase, which is also in accordance with the theory. The stiffness matrix was calculated based on the modal natural frequencies and mass normalized mode shapes. Therefore, the differences among those stiffness matrices are due to the corresponding differences in frequencies and mode shapes.

By comparing the damping ratios in Tables 7-6 and 7-7, it can be observed that the 1<sup>st</sup> modal damping ratios calculated from the white noise and sine-sweep tests are much higher than the ones obtained from the snap-back tests for both models. Considering that the damping ratio for steel structures is generally considered to be 2%~3% of critical (Chopra, 2000), the 14.9% and 8.6% damping ratio values listed in the tables appear to be very high even for the SCPT model. This overestimation of the 1<sup>st</sup> modal damping can be attributed to the accuracy of the calculated transfer functions. By using the half-bandwidth method, as shown in Fig. 7-18(a), the damping ratio can be estimated as  $(f_4-f_3)/2f_k$  for the results (i.e. dash line) calculated from the acceleration response in the experimental tests. However, in that figure, when the peak value  $m_2$  is lower than the real peak  $m_1$  due to the lower frequency resolution

of the transfer function, the real damping ratio (i.e.  $(f_2-f_1)/2f_k$  in solid line) is increased to the calculated one (i.e.  $(f_4-f_3)/2f_k$ ), which is called the missing-peak effect. Another reason for the over-estimation of the damping ratio, as shown in Fig. 7-18 (b), is called the “fatty” effect. When the calculated transfer function is fatter than the real one, the real damping ratio (i.e.  $(f_2-f_1)/2f_k$ ) is estimated to be a higher value (i.e.  $(f_4-f_3)/2f_k$ ). The reason for the fatty effect is that when the shake table was moving along with the above model structures in the white noise and sine-sweep tests, the response represented by the transfer function was not

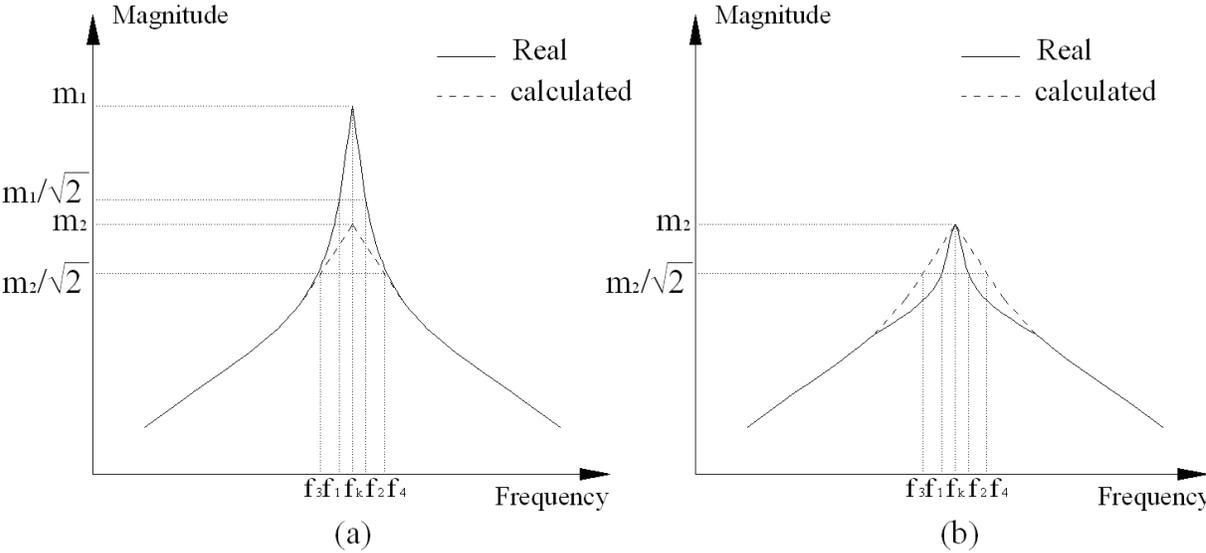


Figure 7-18 Transfer Function for Estimating Damping Ratio  
 (a) missing-peak effect (b) fatty effect

only from the structure itself but also from the movement of the shake table. In another words, the damping ratio is related not only to the structure but also to the foundation, which can be considered as the soil-foundation interaction. However, in the snap-back test the shake table was not moving, so the damping ratios estimated by the transfer functions through the half-bandwidth method are similar to those estimated by the free vibration

response through the logarithmic decrement decay method. Therefore, the damping ratios estimated from the snap-back tests were closer to the real ones of the model structures. Also, the differences in damping matrices are directly due to the differences in modal damping ratios and mode shapes.

**Table 7-6 Dynamic Properties of the SCPT model from identification tests**

Dynamic Properties	White Noise Test	Snap-Back Test		Sine-Sweep Test
Frequencies (Hz)	$\begin{bmatrix} 2.750 \\ 9.141 \\ 14.891 \end{bmatrix}$	$\begin{bmatrix} 2.973 \\ 8.919 \\ 14.534 \end{bmatrix}$		$\begin{bmatrix} 2.750 \\ 9.063 \\ 14.313 \end{bmatrix}$
Mode Shapes	$\begin{bmatrix} 0.397 & 1 & -0.962 \\ 0.752 & 0.616 & 1 \\ 1 & -0.89 & -0.356 \end{bmatrix}$	$\begin{bmatrix} 0.384 & 1 & -0.975 \\ 0.762 & 0.604 & 1 \\ 1 & -0.671 & -0.482 \end{bmatrix}$		$\begin{bmatrix} 0.444 & 1 & -0.952 \\ 0.766 & 0.666 & 1 \\ 1 & -0.82 & -0.417 \end{bmatrix}$
Modal Damping Ratios $\xi$ (%)	$\begin{bmatrix} 14.9 \\ 1.1 \\ 1.4 \end{bmatrix}$	Half-Bandwidth	Decay	$\begin{bmatrix} 8.6 \\ 1.6 \\ 2.5 \end{bmatrix}$
		$\begin{bmatrix} 2.4 \\ 1.3 \\ 1.2 \end{bmatrix}$	$\begin{bmatrix} 1.0 \\ 1.3 \\ 1.2 \end{bmatrix}$	
Stiffness Matrix (kips/in)	$\begin{bmatrix} 248.9 & -140.7 & 7.9 \\ -140.7 & 223 & -99.6 \\ 7.9 & -99.6 & 85.4 \end{bmatrix}$	$\begin{bmatrix} 244.2 & -118.4 & 31.9 \\ -118.4 & 206.3 & -107.1 \\ 31.9 & -107.1 & 83.2 \end{bmatrix}$		$\begin{bmatrix} 230.5 & -118.5 & 16.1 \\ -118.5 & 210.7 & -104.4 \\ 16.1 & -104.4 & 83.5 \end{bmatrix}$
Damping Matrix (lb-sec/in)	$\begin{bmatrix} 101.3 & 1.2 & 49.8 \\ 1.2 & 144.2 & 66.4 \\ 49.8 & 66.4 & 161.2 \end{bmatrix}$	$\begin{bmatrix} 83.1 & -15.4 & 6.0 \\ -15.4 & 72.2 & -18.5 \\ 6.0 & -18.5 & 49.6 \end{bmatrix}$		$\begin{bmatrix} 142.7 & -41.3 & 40.1 \\ -41.3 & 159.1 & -4.1 \\ 40.1 & -4.1 & 116.9 \end{bmatrix}$

**Table 7-7 Dynamic Properties of the SMRF model from identification tests**

Dynamic Properties	White Noise Test	Snap-Back Test	
Frequencies (Hz)	$\begin{bmatrix} 3 \\ 9.625 \\ 15.188 \end{bmatrix}$	$\begin{bmatrix} 3.25 \\ 9.563 \\ 15.188 \end{bmatrix}$	
Mode Shapes	$\begin{bmatrix} 0.401 & 1 & 1 \\ 0.754 & 0.577 & -0.776 \\ 1 & -0.866 & 0.391 \end{bmatrix}$	$\begin{bmatrix} 0.413 & 1 & 1 \\ 0.77 & 0.545 & -0.847 \\ 1 & -0.847 & 0.554 \end{bmatrix}$	
Modal Damping Ratios $\xi$ (%)	$\begin{bmatrix} 15.7 \\ 2.5 \\ 1.9 \end{bmatrix}$	Half-Bandwidth	Decay
		$\begin{bmatrix} 2.1 \\ 0.8 \\ 0.6 \end{bmatrix}$	$\begin{bmatrix} 0.7 \\ 0.8 \\ 1.3 \end{bmatrix}$
Stiffness Matrix (kips/in)	$\begin{bmatrix} 361.1 & -133.3 & 26.4 \\ -133.3 & 172.7 & -103 \\ 26.4 & -103 & 103.1 \end{bmatrix}$	$\begin{bmatrix} 286.9 & -124.7 & 48 \\ -124.7 & 176.2 & -123.6 \\ 48 & -123.6 & 129.3 \end{bmatrix}$	
Damping Matrix (lb-sec/in)	$\begin{bmatrix} 184.4 & 12.4 & 41.6 \\ 12.4 & 166.2 & 55.5 \\ 41.6 & 55.5 & 215.4 \end{bmatrix}$	$\begin{bmatrix} 51 & -2.9 & 4.9 \\ -2.9 & 37.8 & -4.9 \\ 4.9 & -4.9 & 44.7 \end{bmatrix}$	

## 7.2 Shake Table Fidelity

In seismic tests, the earthquake excitation is simulated by the vibration of the shake table. Therefore, the accurate reproduction of the earthquake excitation is a key performance characteristic for a shake table system.

As shown in Figs. 7-19 and 7-20, the shake table fidelity was evaluated by comparing the reference (desired) and feedback (achieved) acceleration response spectra for 50% and 125%-intensity earthquake motions, respectively. The 50%-PGA ground motion is representative of lower-intensity input while the 125% PGA ground motion is representative of higher-intensity motions, and those two tests were conducted for the SCPT model. For the lower-intensity ground motion, the spectrum obtained from the feedback ground acceleration matched well the spectrum calculated from the reference acceleration, which suggests that the shake table reproduced the lower-intensity earthquake motions with high accuracy. For the higher-intensity ground motion, the feedback spectrum matched well the reference spectrum in the lower frequency range but had some deviation in the high frequency range. Considering that the natural frequency of the 1<sup>st</sup> mode in both of the two models is lower than 3.5 Hz, it can be concluded that the shake table reproduced the earthquake excitation effectively and accurately in the natural frequency range of the test structures. The deviation between the reference and feedback spectrum in the higher frequency range can be explained by the fact that when the ground motion input reached near the displacement limit of the shake table, a visible rotation of the shake table was observed especially in the highest-intensity (i.e. 150%) excitation. This rotation effect distorted slightly the feedback spectra in the high frequency range.

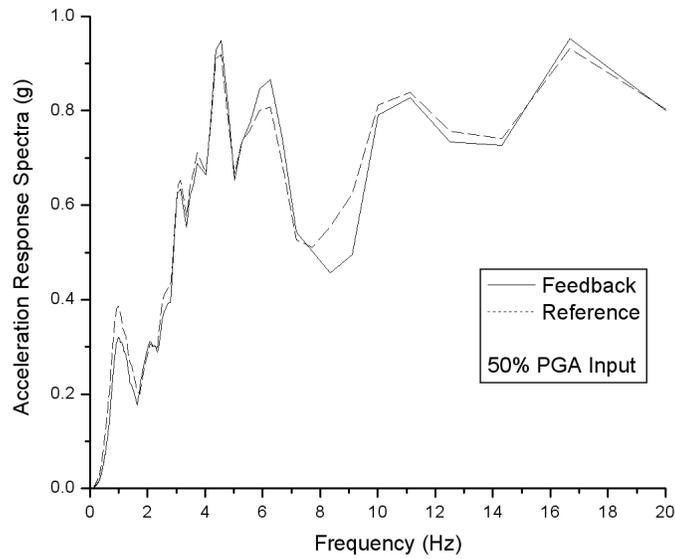


Figure 7-19 Acceleration Spectra for 50%-intensity Earthquake Excitation

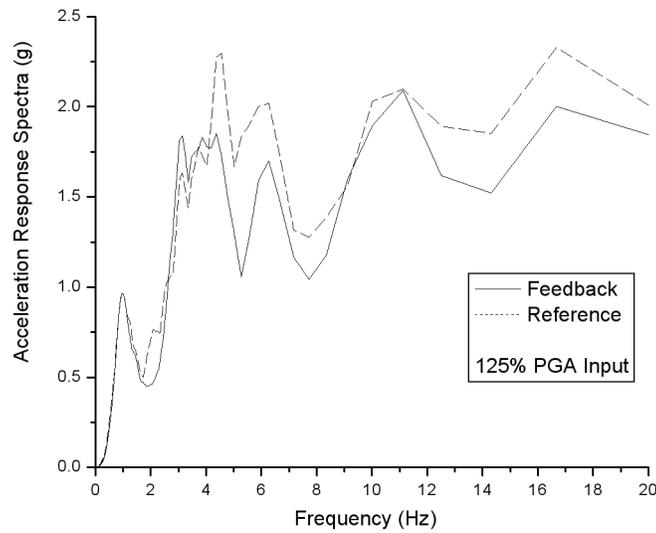


Figure 7-20 Acceleration Spectra for 125%-intensity Earthquake Excitation

### 7.3 Global Response

In this section, the general behavior of both test specimens is discussed. Thereafter, the acceleration and displacement response for the SCPT and SMRF models are presented. The results are discussed by comparing peak displacement and acceleration values using bar

charts. The hysteresis properties are finally discussed. Note that 3 different phases of seismic tests were performed on the SCPT model structure. Each phase included 7 different intensity ground motions from 25% to 150% intensity. Each intensity earthquake excitation was conducted twice; one time in the positive direction and a second time in the negative direction. Details about the testing sequence can be found in the test protocol shown in Section 6.3.

### **7.3.1 General Specimen Behavior**

The seismic tests were conducted on the SMRF and SCPT frame models under the earthquake excitations from 25% to 150% intensity. During those seismic tests, no significant inelastic deformations of the 3 stories were observed. Also, no local buckling or visible local yielding deformations of critical sections in beam-to-column joints were found. These observations are consistent with the results summarized in Section 7.1.4 and verify that both frame specimens exhibited good seismic behaviors as expected.

Under the 150%-intensity earthquake excitation, a visible rotation of the shake table was observed for both SMRF and SCPT models. This rotation may be due to the displacement limit of the shake table and also be due to the limit of axial support capacity of the vertical actuators under the shake table. As discussed in previous sections, this rotation may lead to undesirable impact on the seismic performance of the two models since no such ground rotation effects were considered in the models used for the prediction analysis.

### **7.3.2 Absolute Acceleration Response**

As reported by Kircher (2003), the damage to nonstructural components led to 50% of the \$18.5 billion total loss during the 1994 Northridge earthquake. Recall that the prototype building of the SCPT model was designed as a steel frame hospital, which would have many acceleration-sensitive nonstructural components. Therefore, it is very important to reduce the acceleration response in order to decrease the damage of the acceleration-sensitive nonstructural components and the corresponding economic loss after seismic hazards such as the 1994 Northridge earthquake.

The floor acceleration time histories of the SCPT and SMRF models for all the seismic tests are shown in Appendix E. The peak floor acceleration responses of two models for different-intensity ground motions are shown in Figs. 7-21 to 23 for positive direction excitations and in Figs. 7-24 to 7-25 for negative direction inputs. In those bar charts, the peak accelerations of the SCPT model are always lower than those of the SMRF model. For the low-intensity (25%-100%) ground motions, the acceleration in the SCPT model is reduced by up to 37% compared to that in the SMRF model, while this reduction reaches up to 41.8% after the 100%-intensity excitations. The large reduction in the acceleration response demonstrates the excellent seismic performance of the SCPT structure and also validates the design procedure focused mainly on acceleration issue since the design Relative Performance Index (RPI) had a 70% weighting factor on the acceleration response (see Section 5.4).

From the results shown in Figs. 7-21 to 7-26, it is also observed that, for the SCPT model, the acceleration response for the 2<sup>nd</sup> and 3<sup>rd</sup> phase excitations is increased compared to that of the 1<sup>st</sup> phase excitations. The reason for this increase in acceleration is that after the 1<sup>st</sup>

phase 25%~150% intensity earthquakes, most of the Energy-Dissipating (ED) bars yielded which may reduce the capability of energy dissipation in ED bars since for the post-yielding ED bars with the residual deformation, the yielding force can be increased slightly due to the strain hardening effect although this effect might be very slight. Also, the additional vertically supporting bars installed on the top and bottom of the beams ends stiffened the structure which may lead to a decrease of the natural period of the structure. Those two effects lead to the higher acceleration response in the 2<sup>nd</sup> or 3<sup>rd</sup> phase excitations. Although the acceleration responses are increased in the 2<sup>nd</sup> and 3<sup>rd</sup> phase tests in the SCPT model, they are still lower than those in the SMRF model. Even after one or two 150%-intensity earthquake attacks, the seismic performance of the SCPT model without repairing its structural components is still better than the SMRF model without any earthquake attack history from the point of view of acceleration response.

The trend in acceleration responses observed above is slightly different for the 150%-intensity tests is. The acceleration responses for this highest intensity test in the 2<sup>nd</sup> and 3<sup>rd</sup> phase of the SCPT model and in the SMRF model are similar to the corresponding peak accelerations in the 125%-intensity tests. This is because only the 150%-intensity test in the 1<sup>st</sup> phase excitations of the SCPT model reached the real 150% amplitude of the ground motions while other 150%-intensity tests such as 150% in the 2<sup>nd</sup> or 3<sup>rd</sup> phase of the SCPT model and 150% in the SMRF model were not excited to the real 150% amplitude due to the limitation of the shake table. This conclusion was reached by comparing the actual amplitudes of input ground motions for those 150%-intensity tests.

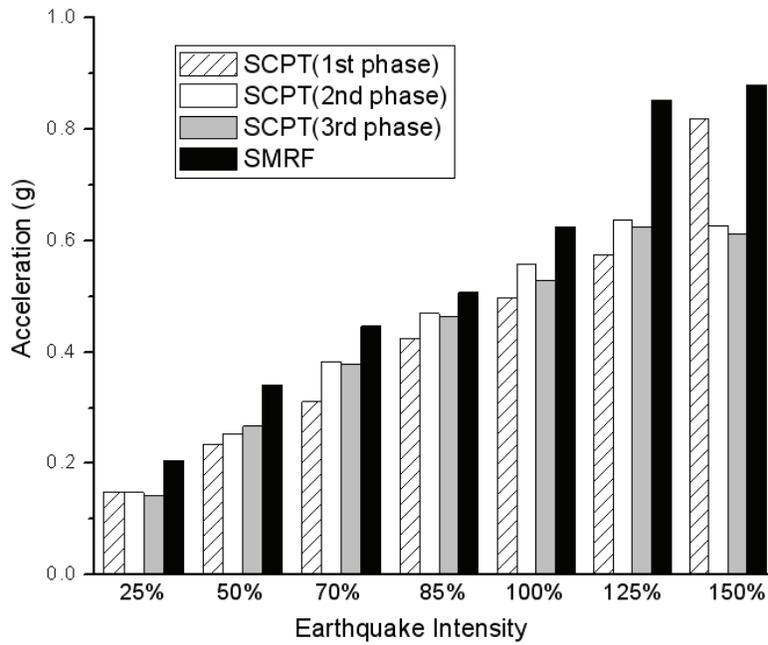


Figure 7-21 Peak Acceleration Response of the 1<sup>st</sup> floor for the SCPT and SMRF Models in Positive Direction Seismic Tests

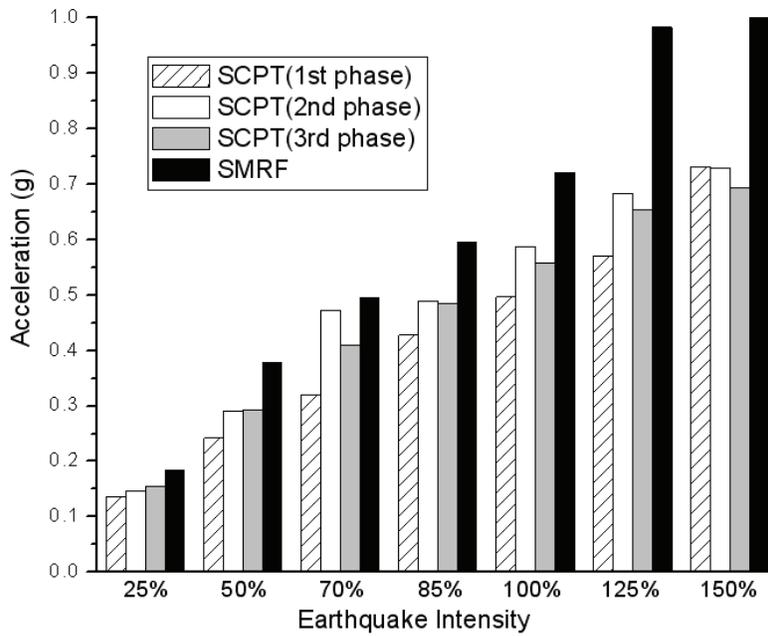


Figure 7-22 Peak Acceleration Response of the 2<sup>nd</sup> floor for the SCPT and SMRF Models in Positive Direction Seismic Tests

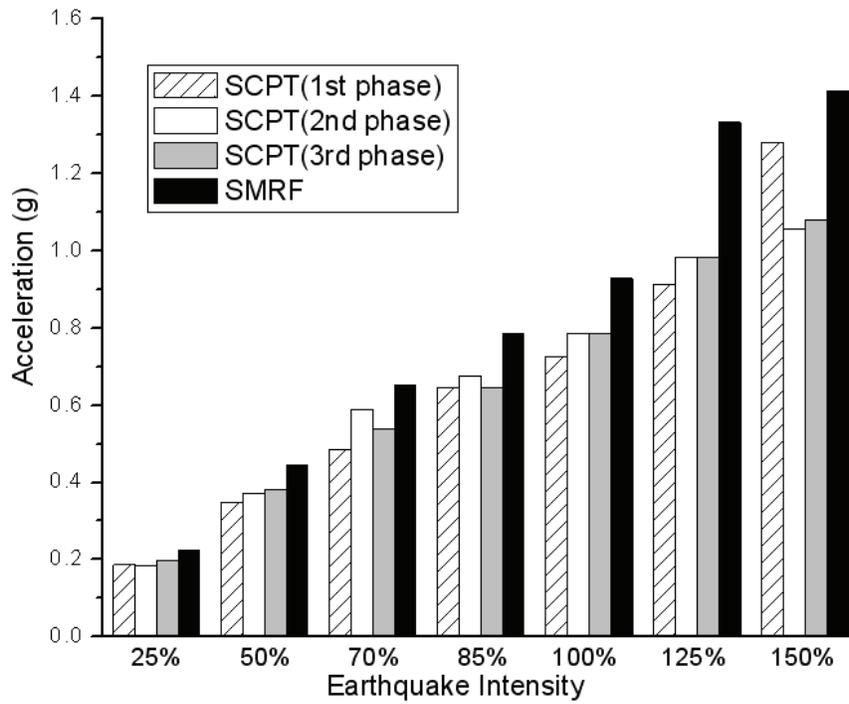


Figure 7-23 Peak Acceleration Response of the 3<sup>rd</sup> floor for the SCPT and SMRF Models in Positive Direction Seismic Tests

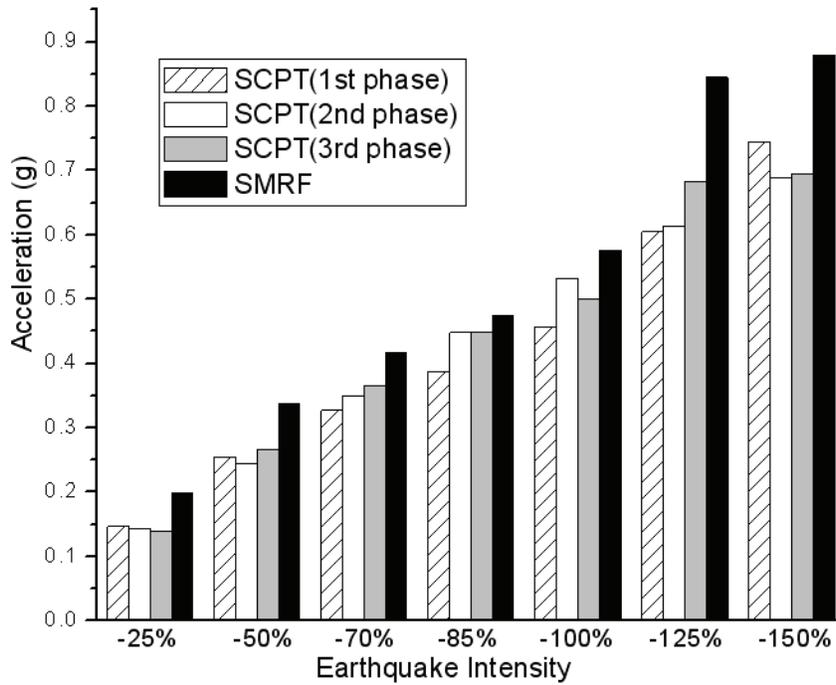


Figure 7-24 Peak Acceleration Response of the 1<sup>st</sup> floor for the SCPT and SMRF Models in Negative Direction Seismic Tests

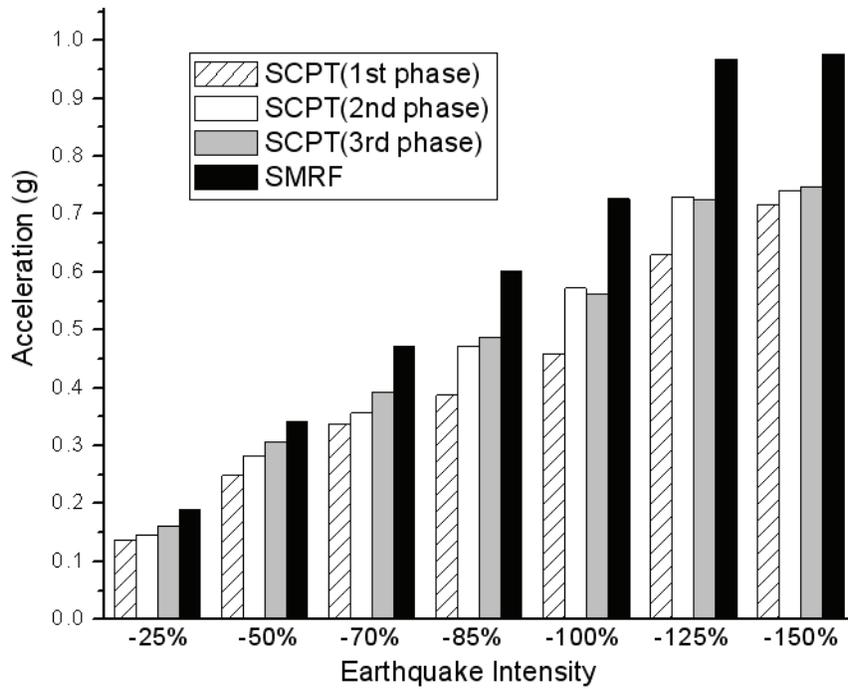


Figure 7-25 Peak Acceleration Response of the 2<sup>nd</sup> floor for the SCPT and SMRF Models in Negative Direction Seismic Tests

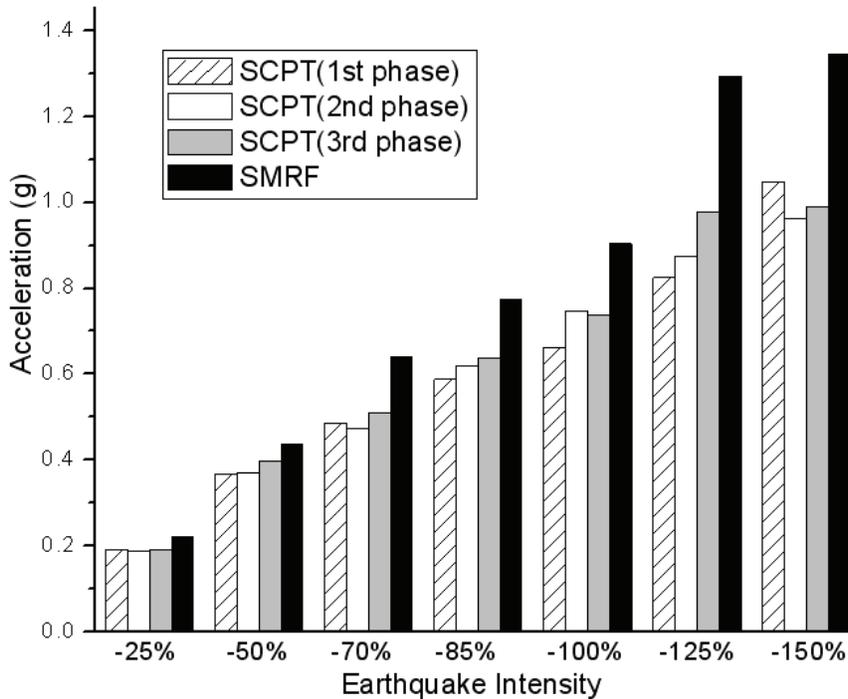


Figure 7-26 Peak Acceleration Response of the 3<sup>rd</sup> floor for the SCPT and SMRF Models in Negative Direction Seismic Tests

### 7.3.3 Inter-Story Drift Response

In this section, the displacement response of the two model frames is presented by comparing their inter-story drifts. The inter-story drift time histories for both models are presented in Appendix F. For comparison purpose, bar charts are used again to compare maximum inter-story drifts, as shown in Figs. 7-27 to 7-29 for positive-direction ground motions and in Figs. 7-30 to 7-32 for negative-direction excitations.

In these figures, it can be seen that the maximum inter-story drifts of the two model structures for the 100%-intensity ground motions are lower than 1%. Considering the excitation with 10% probability of exceedance in 50 years, the corresponding performance level is immediate occupancy with 0.7% maximum inter-story drift according to FEMA 356, while the next level is life safety with 2.5% maximum inter-story drift. Therefore, the performance of both of two frame models can be considered to be in accordance with the immediate occupancy level since the 1% drift is very close to 0.7% and far away from the 2.5%, and no yielding occurred in the beams or columns (see Section 7.4.1).

As indicated in those maximum inter-story drift figures, the drift values of the SCPT models are similar or slightly higher compared to those in the SMRF models in the corresponding intensity seismic tests. Therefore, it can be pointed out that the results are consistent with the design procedure since the RPI (see Section 5.4) had a 30% weighting factor for displacement response and more efforts were made to reduce the acceleration response. Another argument can be made that although the drift response in the SCPT model is

slightly higher than that in the SMRF model, the seismic performance of the SCPT structure is still better since no yielding occurred in the beams or columns while beams and columns yielded in the SMRF model.

It can be also observed from the results shown in Figs 7-27 to 7-32 that the maximum inter-story drifts of the SCPT model for the 2<sup>nd</sup> or 3<sup>rd</sup> phase excitations are higher than those in the 1<sup>st</sup> phase tests. This trend is similar to that observed for the acceleration response. As explained in Section 7.3.2, the reason is also that the capacity of energy dissipation was reduced for the post-yielding ED bars after the 1<sup>st</sup> phase tests. Again, for the 150%-intensity excitations, the trend (e.g. the highest value in the 150%-intensity test is the response from the 1<sup>st</sup> phase test as shown in Fig. 7-28) is different than the above discussions from other lower-intensity earthquakes, which is due to the insufficient input ground motion by the limit of the shake table as explained in Section. 7.3.2.

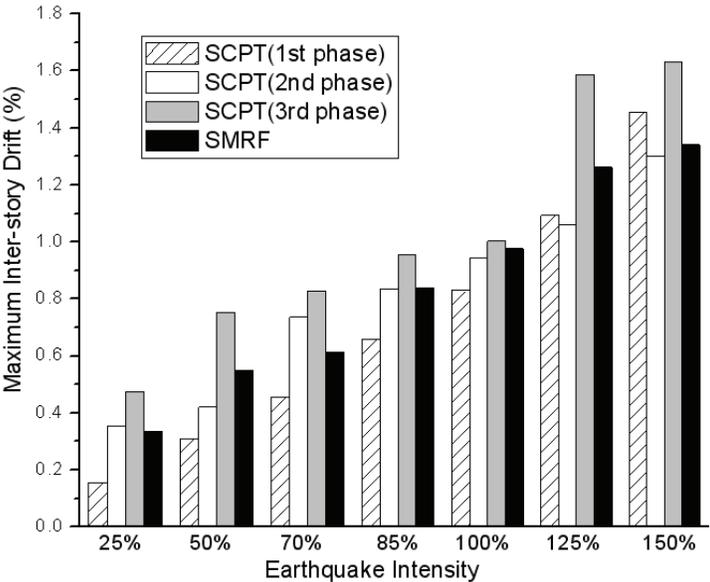


Figure 7-27 Maximum Inter-story Drift of the 1<sup>st</sup> floor for the SCPT and SMRF Models in Positive Direction Seismic Tests

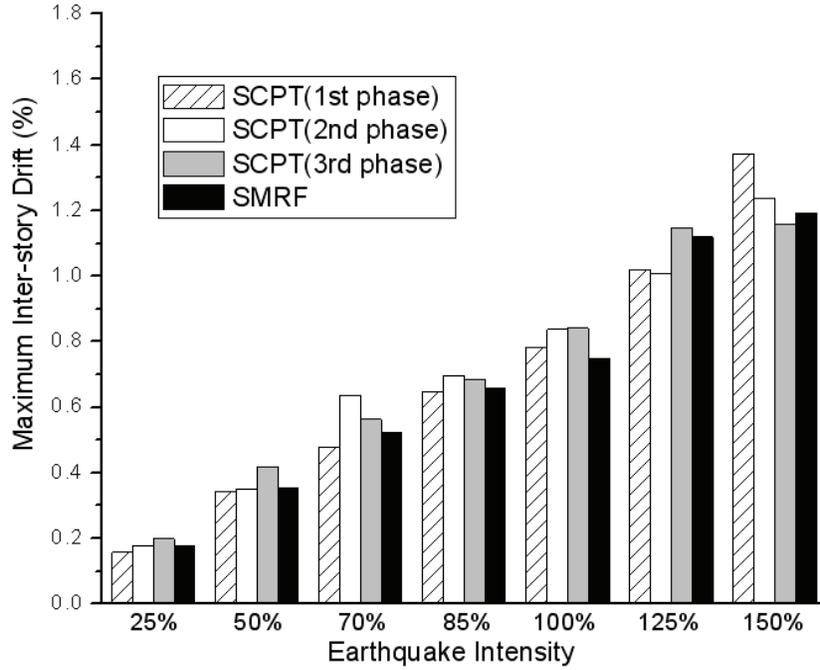


Figure 7-28 Maximum Inter-story Drift of the 2<sup>nd</sup> floor for the SCPT and SMRF Models in Positive Direction Seismic Tests

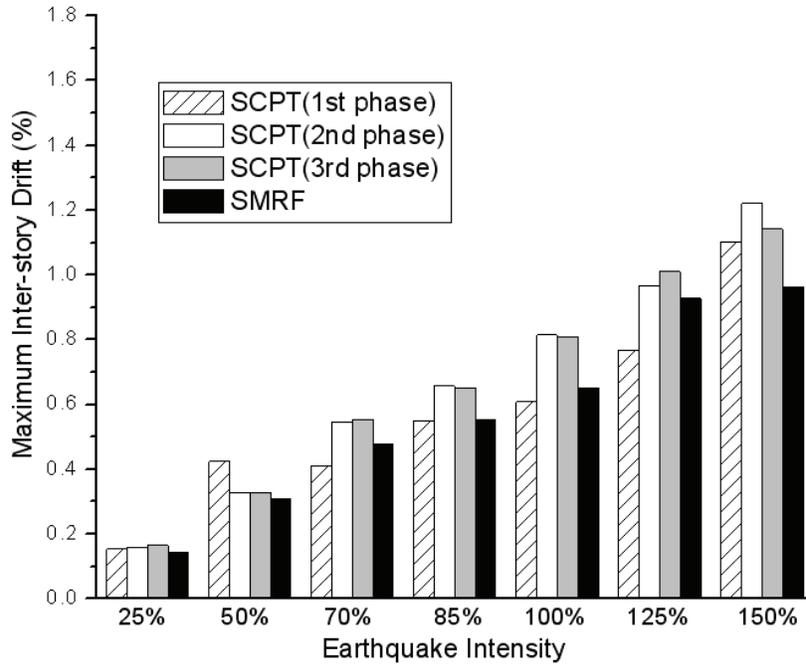


Figure 7-29 Maximum Inter-story Drift of the 3<sup>rd</sup> floor for the SCPT and SMRF Models in Positive Direction Seismic Tests

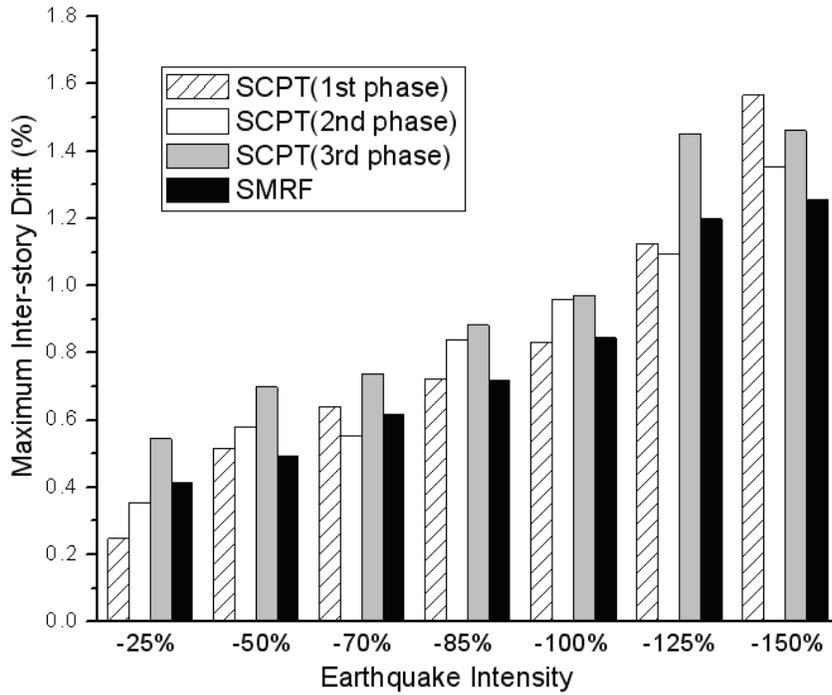


Figure 7-30 Maximum Inter-story Drift of the 1<sup>st</sup> floor for the SCPT and SMRF Models in Negative Direction Seismic Tests

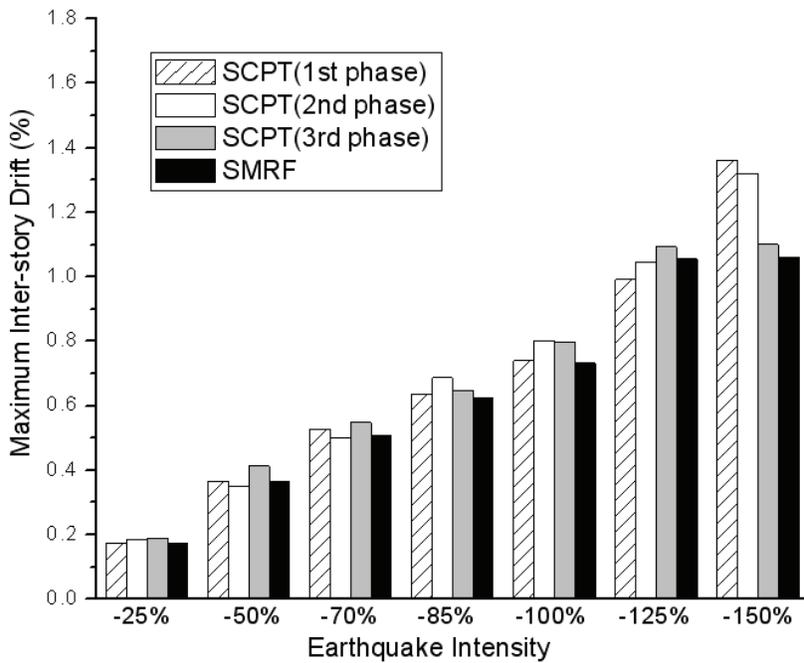


Figure 7-31 Maximum Inter-story Drift of the 2<sup>nd</sup> floor for the SCPT and SMRF Models in Negative Direction Seismic Tests

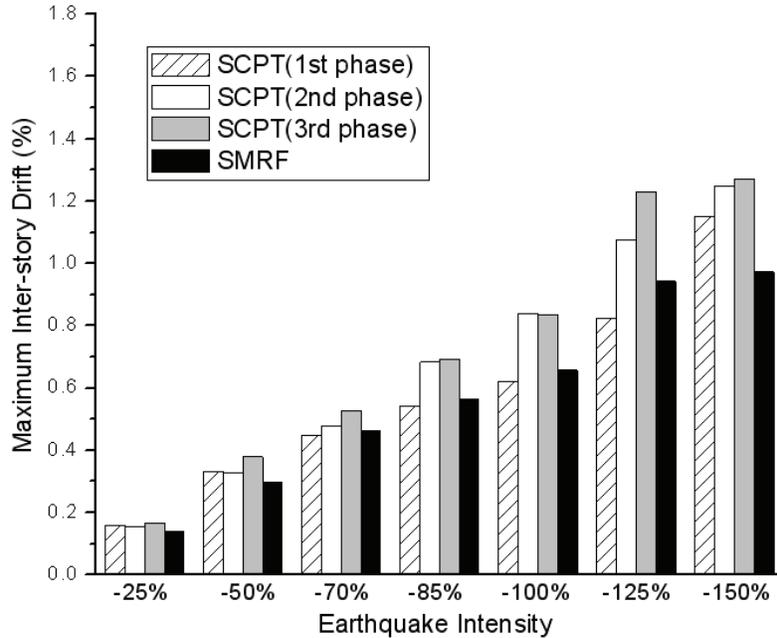


Figure 7-32 Maximum Inter-story Drift of the 3<sup>rd</sup> floor for the SCPT and SMRF Models in Negative Direction Seismic Tests

### 7.3.4 Hysteresis Properties

The hysteretic relationship between the base shear and top floor relative displacement for both the SCPT and SMRF models under the positive 125% intensity seismic excitation are shown in Figs. 7-33, 7-34 and 7-35, respectively, since the ground input was insufficient for the 150% intensity earthquake as mentioned earlier. In these figures, the bold slope lines represent the generalized stiffness for the 1<sup>st</sup> mode, in which  $K_1$  is the generalized stiffness for the 1<sup>st</sup> mode of the SCPT model and is calculated based on the 1<sup>st</sup> mode shape  $\Phi_1$  and stiffness matrix  $K$  (i.e.  $K_1 = \Phi_1^T * K * \Phi_1$ ) obtained from White Noise Tests listed in Table 7-6 (see Section 7.1.4) while  $K_2$  is that of the SMRF model based on the same calculation

method using the structural properties obtained from White Noise Tests as shown in Table 7-7 (see Section 7.1.4). From these figures, it is observed that the initial stiffness of the SCPT model is similar to the  $K_1$  while the initial stiffness of the SMRF model is lower than  $K_2$  but closer to  $K_1$ . These results suggest that the initial stiffness of both models is similar, which is in accordance with the theory (i.e. using SCPT connections have less influence on the initial stiffness of the SMRF structures). Also, it indicates that the responses are mainly contributed from the 1<sup>st</sup> mode response of the structures. The deviation between the initial stiffness of the SMRF model and  $K_2$  is due to the accuracy of the transfer functions, which are used to calculate the modal frequencies and mode shapes in order to obtain stiffness matrix as discussed in Section 7.1. Since the base shear was calculated as the summation of the inertia forces in the three floors, the damping response is included in these hysteretic loops. Also, it is believed that the rotation effect of the shake table and hysteresis properties of vertical supporting actuators (see Section 8.2) made some influence on these hysteretic loops. Therefore, those issues discussed are the reasons that the hysteretic loops obtained experimentally have a little difference with the ideal theoretical ones. It is seen that the loop area in Fig. 7-33 for the SCPT model is larger than that in Fig. 7-34, which is due to the vertical movement of the beams in the 1<sup>st</sup> phase tests while this movement was limited by the additional bars installed on the top and bottom of the beam flanges in the 3<sup>rd</sup> phase tests. It is also found that the maximum base shear of the SCPT model in both the 1<sup>st</sup> and 3<sup>rd</sup> phase +125% intensity seismic tests is lower than that of the SMRF model, which indicates that the SCPT model exhibited better seismic performance than that of the SMRF model. The hysteresis energy issues will be discussed in Section 7.5.

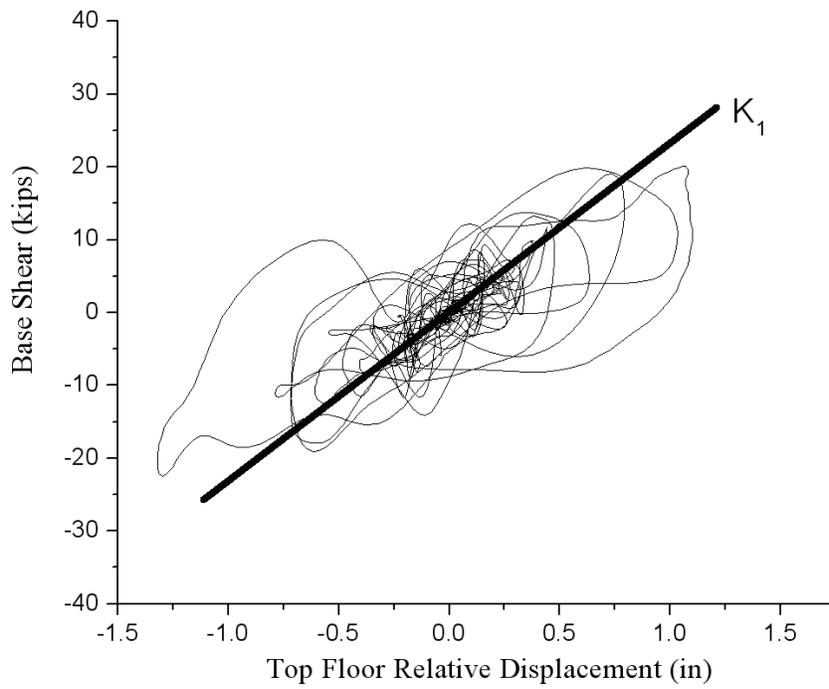


Figure 7-33 Hysteresis for the SCPT Model in test PE6A (1<sup>st</sup> phase, +125% intensity)

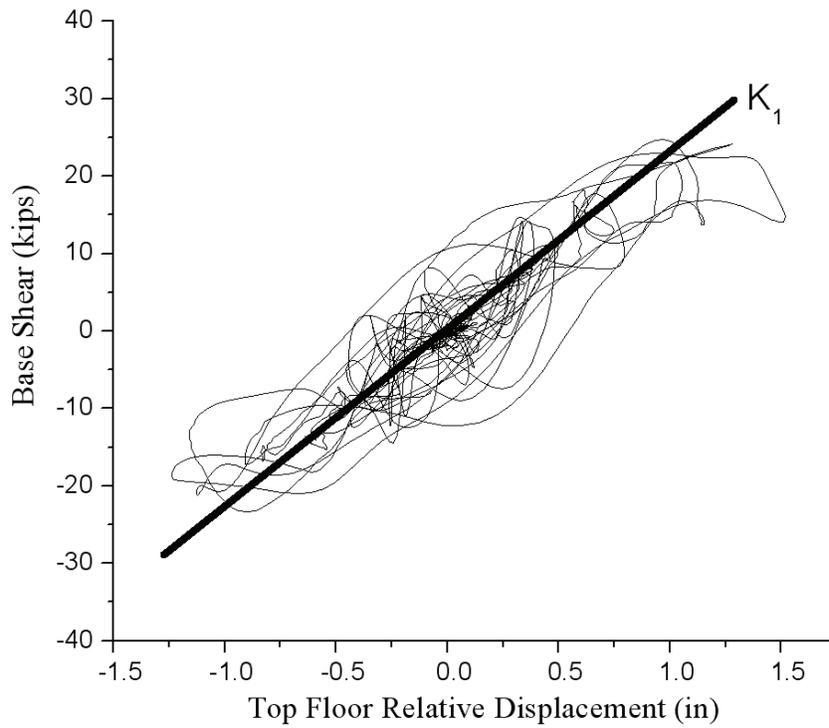


Figure 7-34 Hysteresis for the SCPT Model in test PE22 (3<sup>rd</sup> phase, +125% intensity)

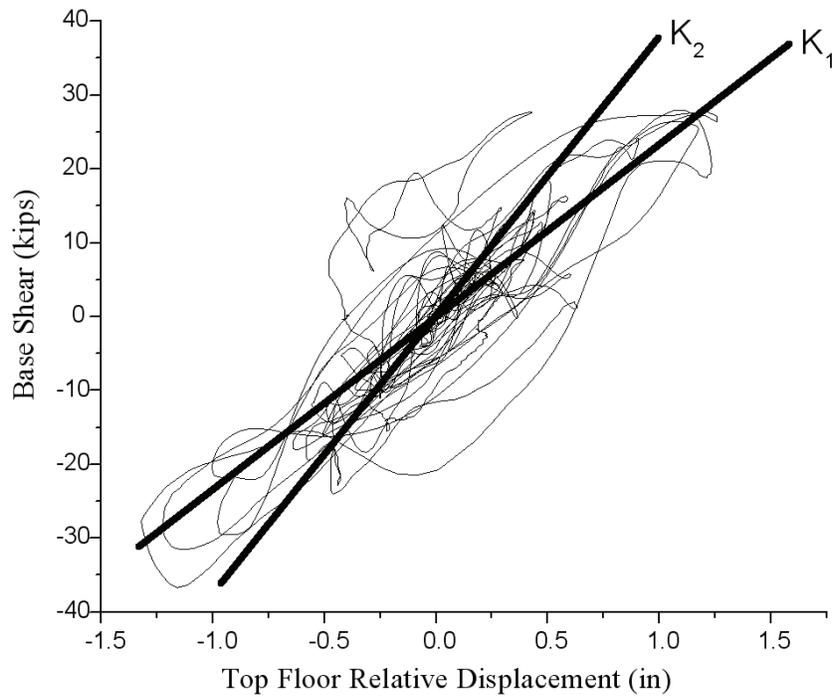


Figure 7-35 Hysteresis for the SMRF Model in test ME06A (+125% intensity)

## 7.4 Local Response

It is very important to evaluate the seismic performance of the individual structural components since the local response presents an index of the damage level and repair cost to the structures after seismic hazards. In this section, the seismic performance of beams and columns as well as the Energy-Dissipating (ED) bars and Post-Tensioned (PT) strands are discussed.

### 7.4.1 Beams and Columns

To investigate the seismic performance of the beams and columns, strain gages were installed at critical sections of the beam-column joints as shown in Section 6.4. According to

the mechanics of material, for structural steel, the strain is the direct index to evaluate if a cross section has yielded. The corresponding moment can be calculated based on the strain and section property. Therefore, the direct index, strain, is used as the only criteria for performance evaluation.

The yielding strain of the steel with 50ksi yielding strength used for the beams and columns in the SCPT and SMRF models was  $1724 \times 10^{-6}$  with the corresponding elastic modulus 29000ksi. If the strain on the exterior surface of the beam or column flanges exceeds the yielding strain, the beam or column can be considered to begin yielding. Through checking the strain time histories in different seismic tests, it was found that no yielding occurred in the beams or columns of the two models for the seismic tests with 25% to 100% intensity, which demonstrates that the seismic performance of the SMRF structure satisfied the design purpose that the structure would remain in the elastic range under the immediate occupancy earthquake level.

From the approximate 50 strain gages installed in each of two model structures, one strain gage was selected for representing each beam in the 1<sup>st</sup>, 2<sup>nd</sup> and 3<sup>rd</sup> floor and each column in the base. The tag names of the selected strain gages were SGB1-3, SGB2-3, SGB3-4, SGC0-2 and SGC0-7 (see Section 6.4). The strain time histories of those strain gages for the 125% and 150% intensity seismic tests are presented in Appendix G, since no yielding occurred in 25%~100% intensity tests. The maximum strain obtained from those time histories were plotted in bar charts, as shown in Figs. 7-36 to 7-41. The dash line in these figures represents the yielding strain.

As seen in Fig. 7-36 and 7-37, the strains in the beams of the 1<sup>st</sup> and 2<sup>nd</sup> floor of the SMRF model exceed the yield limit, while no yielding occurs in those same beams of the SCPT model. As expected, the ED bars yielded instead of the beams as a result of the gap opening in the beam-column joints. These results demonstrate that under the high intensity (125% and 150%) earthquake hazards, the SCPT structure performed better since no damage occurred to the main structural components. The cost associated with repair of the main structural components is much more than the cost of replacing the ED bars. As shown in Fig. 7-38, although no yielding occurred in the 3<sup>rd</sup> floor beams for both models, the strains in the SCPT model is still much lower than those in the SMRF model. This result indicates that that the beams of the SMRF model would yield under higher (>150%) earthquakes.

As shown in Fig. 7-39 and 7-40, the base columns of the SMRF model yielded slightly, while still no yielding occurred in the SCPT structure, which is in accordance with the conclusion based on the performance of the beams. For the weak-beam-strong-column design philosophy of the SMRF building, the columns should not begin yielding before most of the beams yields. However, the two exterior columns were reduced from a w5x19 to w5x16 section due to the material availability, which decrease the yielding strength of the columns. Also, the beams in the 1<sup>st</sup> and 2<sup>nd</sup> floor of the SMRF model had yielded. Considering these two facts, the slight yielding of the columns can be acceptable based on the weak-beam-strong-column approach. Again, as explained in previous sections, the results shown in these figures for the 150%-intensity earthquakes are similar to those for

125%-intensity excitations due to the limit of the shake table leading to the insufficient amplitude of ground motions.

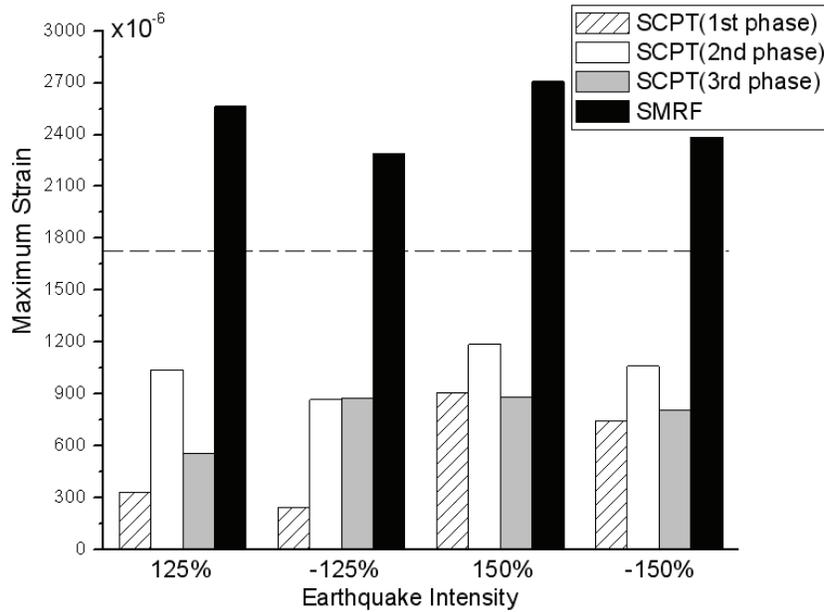


Figure 7-36 Maximum Strains of the beam of the 1<sup>st</sup> floor for the SCPT and SMRF Models

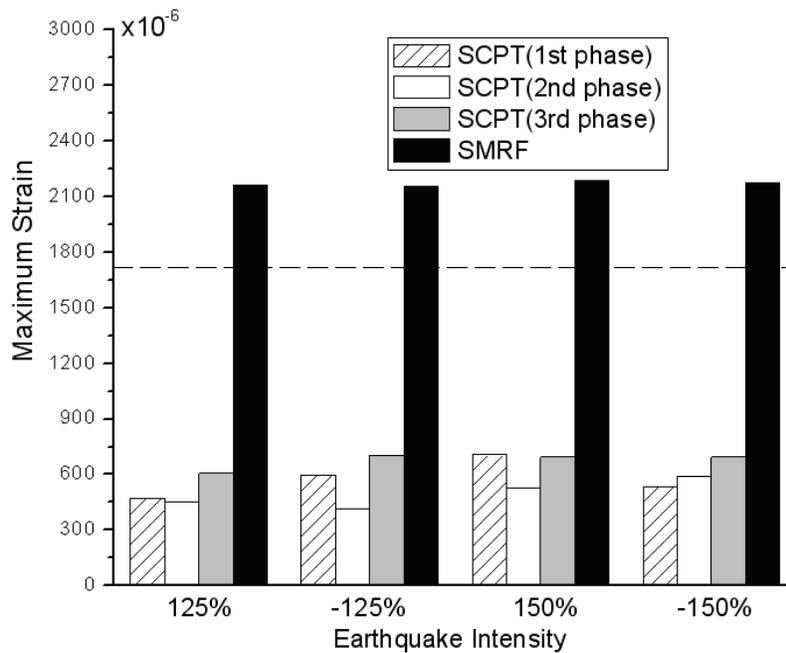


Figure 7-37 Maximum Strains of the beam of the 2<sup>nd</sup> floor for the SCPT and SMRF Models

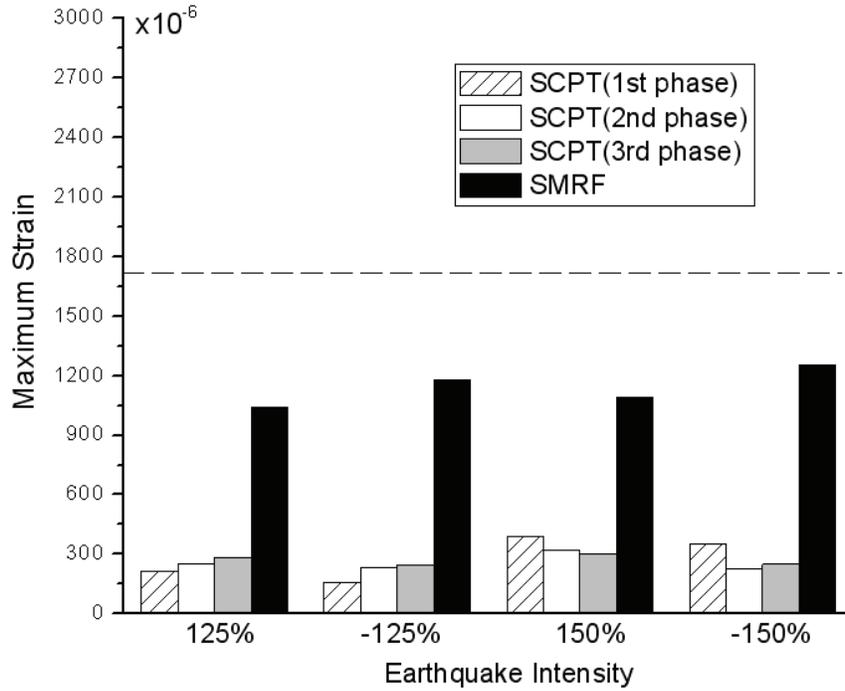


Figure 7-38 Maximum Strains of the beam of the 3<sup>rd</sup> floor for the SCPT and SMRF Models

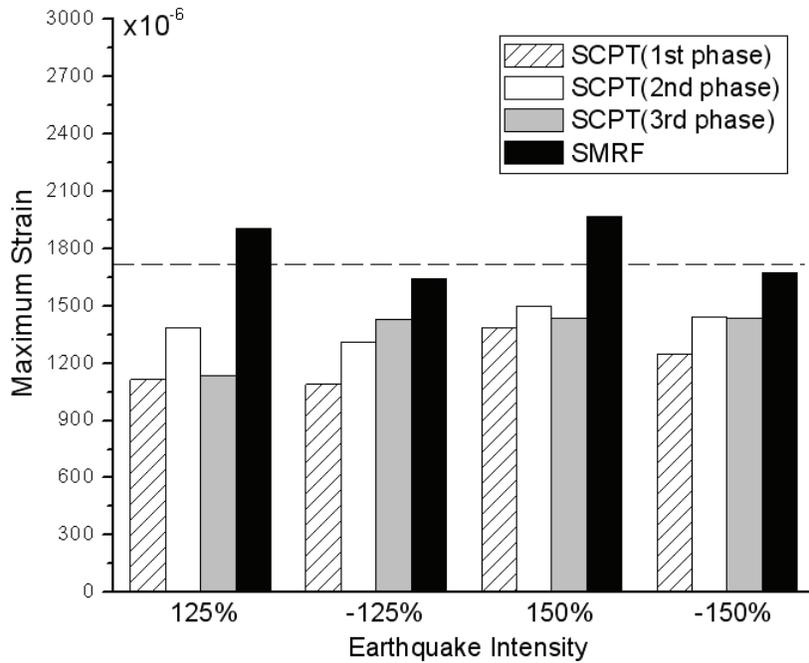


Figure 7-39 Maximum Strains of the exterior column of the base connection for the SCPT and SMRF Models

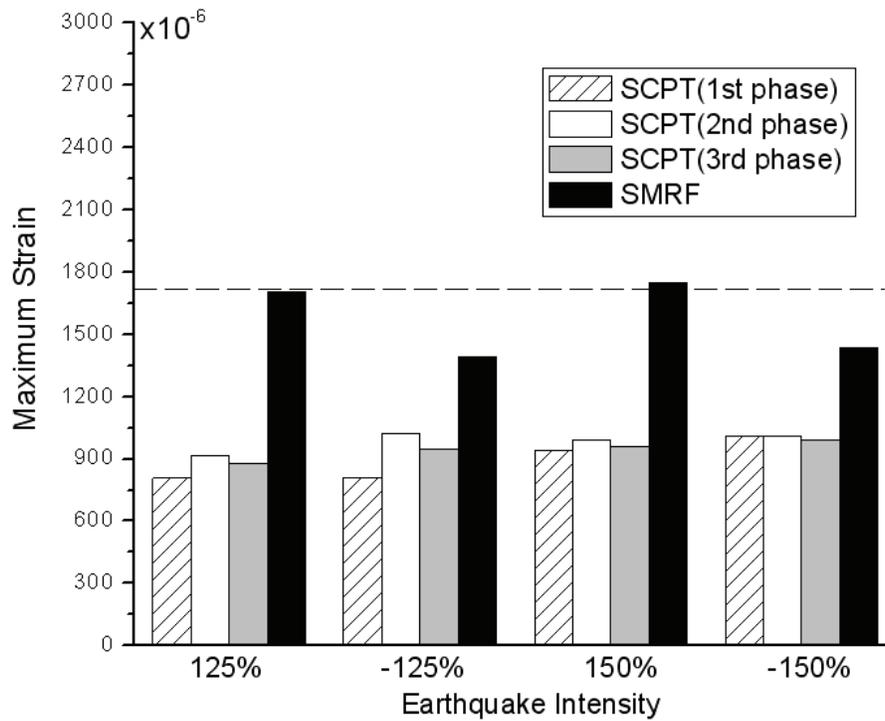


Figure 7-40 Maximum Strains of the center column of the base connection for the SCPT and SMRF Models

#### 7.4.2 Energy Dissipating (ED) Bars

ED bars were used to dissipate energy in the SCPT model as a result of gap opening. The deformation of the ED bars was proportional to the gap opening. In order to measure the gap opening, potentiometers were installed on the top and bottom of the beam flange in each beam-column joint. However, due to unknown circuit problem of the conditioner, which was used to transfer the signal from the potentiometers to the data recording system, the results for the 3<sup>rd</sup> floor were unavailable. For the 1<sup>st</sup> and 2<sup>nd</sup> floor, the deformation of the ED bars is 82% of the gap-opening obtained from the potentiometer since the potentiometer measured the gap-opening on the outside of the beam flange, while the ED bars were welded on the inside of the beam flange. The gap opening angles were calculated as the ratio of the gap-

opening obtained from potentiometers to the depth of the beam. Two potentiometers (PM1-2 and PM2-6) are selected to represent the gap-opening for the 1<sup>st</sup> and 2<sup>nd</sup> floor, respectively. Because no yielding occurred in the ED bars for the 25%~100% intensity seismic tests, only the maximum gap-openings for the 125% and 150%-intensity seismic tests with positive and negative directions are shown in Fig. 7-41 and 7-42. The dash line (0.0015 rad) included in these figures represents the gap opening causing first yield of the ED bars. The gap-opening of the 2<sup>nd</sup> floor for 150% intensity in the 3<sup>rd</sup> phase was unavailable due to the invalid data reading.

In these figures, it can be observed that almost all of the ED bars yielded under the 125% or 150% intensity earthquakes and the largest ductility (0.0039 rad) reached a value of 2.5, which demonstrates the good capability of inelastic deformation and energy dissipation. It is also noted that the gap opening in the 2<sup>nd</sup> and 3<sup>rd</sup> phase is larger than that in the 1<sup>st</sup> phase for the same intensity earthquake. Vertical supports were installed for the 2<sup>nd</sup> and 3<sup>rd</sup> phases which may have limited the rotation of the beam ends. Therefore, the explanation of this phenomenon is that the vertical support stiffened the structure, which led to the higher response compared to those in the SCPT frame without vertical supports. This higher response caused also an increase in the deformations of the ED bars. Such higher response was also observed for the accelerations and inter-story drifts, which is in accordance with this explanation.

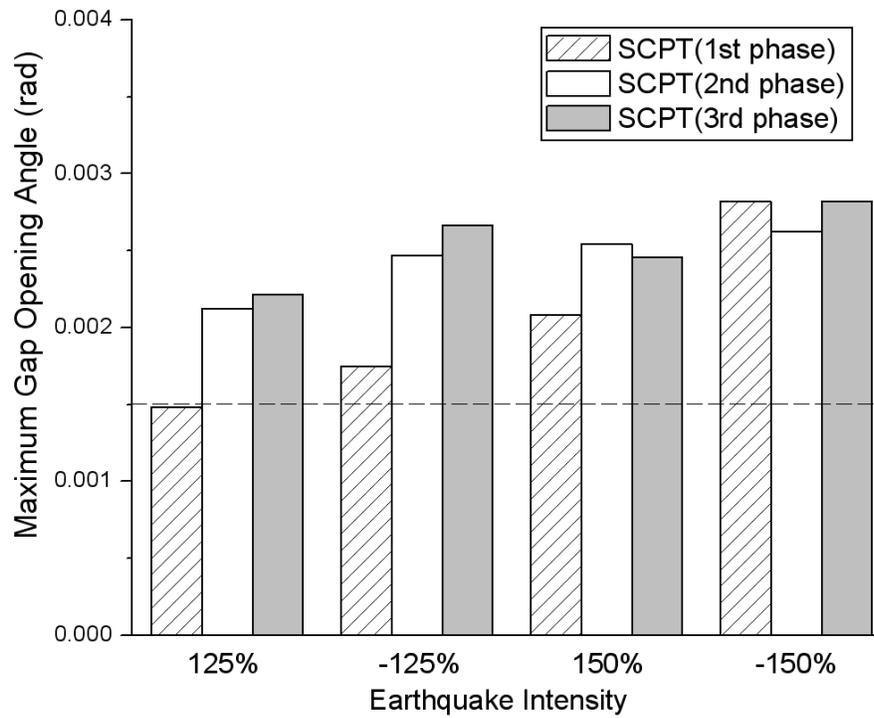


Figure 7-41 Gap Opening of the beam-column joint in 1<sup>st</sup> floor for the SCPT Model

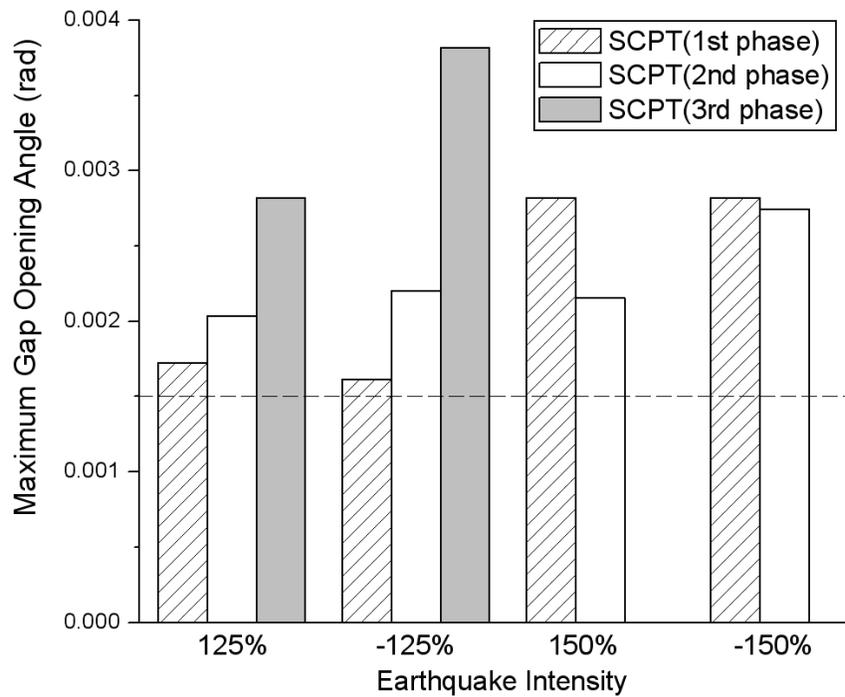


Figure 7-42 Gap Opening of the beam-column joint in 2<sup>nd</sup> floor for the SCPT Model

### 7.4.3 Post-Tensioned (PT) Strands

The PT strands were used to provide the re-centering force in the SCPT model. Maintaining the stress in the PT strands in the elastic range during the seismic tests is a guarantee for the good ability of re-centering.

The deformation in the PT strands was calculated using approximately 50% of the measured gap-opening since the PT strands were on the center of the beam section. The relations between the force and deformation of the PT strands during the high-intensity (125% and 150%) level seismic tests are shown in Figs. 7-43 to 7-54, where the force is the summation of the forces in the two PT strands at the same floor. The negative deformation means the gap opening transfers from the top of the beam to the bottom of the beam while the PT strands is elongated for both directions of gap opening. Note that results for the 3<sup>rd</sup> floor are not shown due to the unavailable data of the gap opening for that floor.

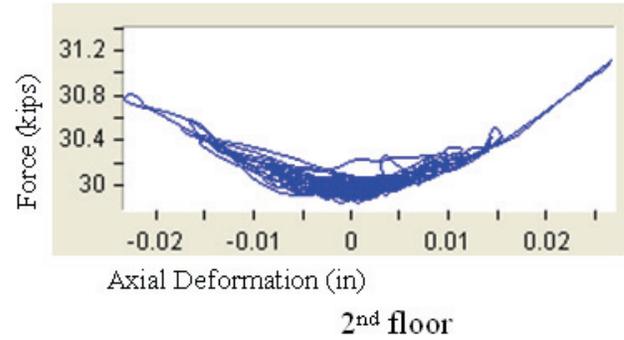
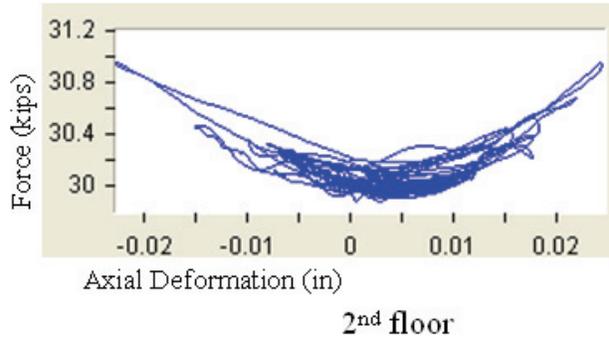
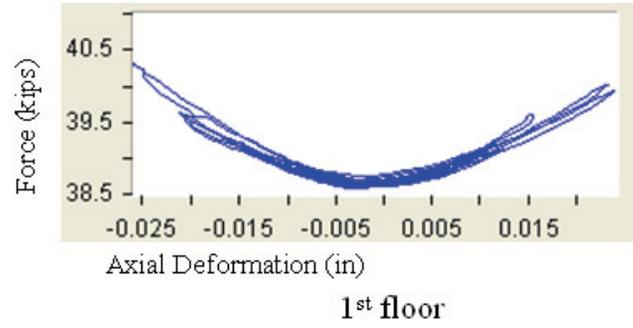
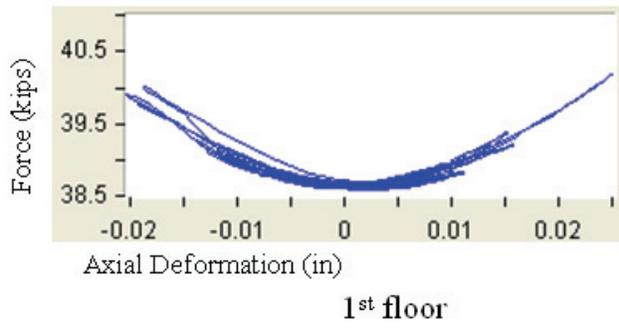
As shown in Fig. 7-54, the maximum force in two PT strands in the 1<sup>st</sup> floor during the seismic tests is about 42 kips. For the 2<sup>nd</sup> and 3<sup>rd</sup> floor, this maximum force is 33.7 kips and 27.6 kips, respectively. The yielding force for two PT strands are 105.4 kips in the 1<sup>st</sup> and 2<sup>nd</sup> floor and 74.4 kips in the 3<sup>rd</sup> floor. Therefore, the PT strands were in the elastic range during those high-intensity seismic tests, which satisfies the design purpose and proves the sufficient ability of PT strands to provide the re-centering force.

The ideal relation between PT force and axial deformation should be like a “V” shape since the PT strands remained in the elastic range. It is observed, however, in Figs. 7-43 to 7-54

that some hysteretic loops occur and the stiffness (the slope) for positive and negative directions are slightly different. It is believed that the potentiometers could not precisely measure the gap opening due to the vertical movement of the beams in 1<sup>st</sup> or 2<sup>nd</sup> phase. Therefore, the axial deformations calculated from the gap opening were not accurate in such condition. The relations become more symmetric in the 3<sup>rd</sup> phase compared to those in the previous phase tests, since the vertical movement of the beam was limited or eliminated by the vertical supports.

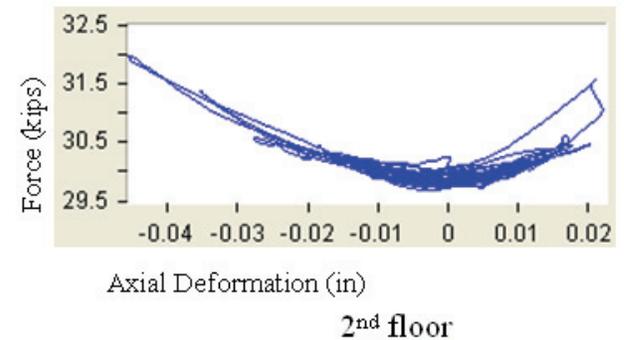
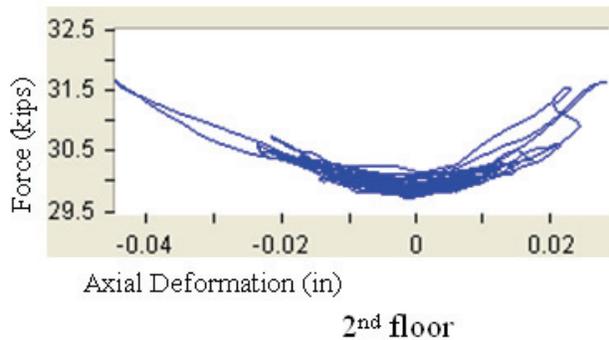
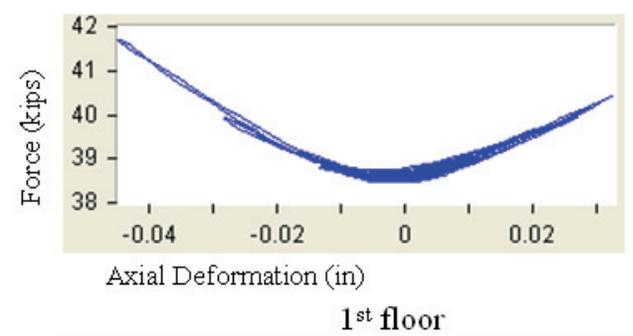
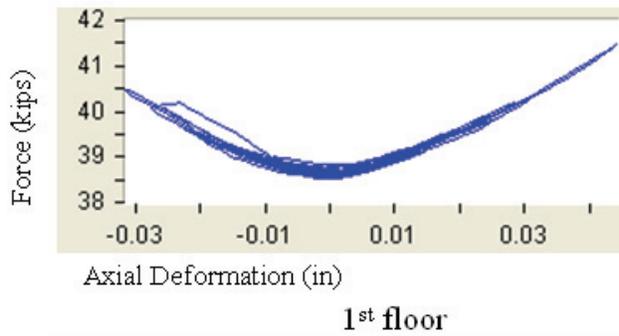
In these figures, it can be found that the maximum force in the 3<sup>rd</sup> phase tests is higher than that in the 1<sup>st</sup> phase tests (e.g. 40.3 kips in +125% of 1<sup>st</sup> phase at PE6A test vs. 41 kips in +125% of 3<sup>rd</sup> phase at PE22 test). This result indicates a higher axial deformation of the PT strands and is consistent with the results obtained from the gap-openings in previous section (i.e. increasing gap-opening from 1<sup>st</sup> phase to 3<sup>rd</sup> phase tests).

Regarding the prestress loss in the PT strands after the seismic tests, no clear loss is observed. The prestress force of the 1<sup>st</sup> floor changed from 38.6 kips in test PE6A (+125% in 1<sup>st</sup> phase) to 38.2 kips in test no. PER23 (-150% in 3<sup>rd</sup> phase), which represents a 1% loss. Considering that the model structure was subjected to 3 phase of 150%-intensity earthquake attacks, this 1% loss is negligible. This result demonstrates again the good seismic performance of the PT strands during the seismic tests.



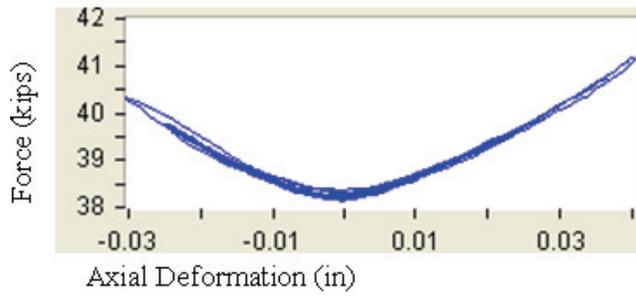
**Figure 7-43 Force vs. Deformation of PT Strands For Test PE6A (+125%)**

**Figure 7-44 Force vs. Deformation of PT Strands For Test PER6 (-125%)**

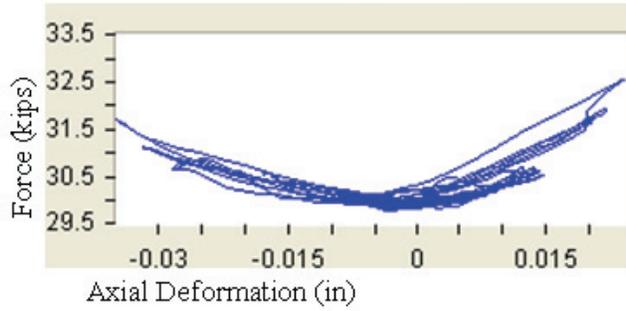


**Figure 7-45 Force vs. Deformation of PT Strands For Test PE7A (+150%)**

**Figure 7-46 Force vs. Deformation of PT Strands For Test PER7 (-150%)**

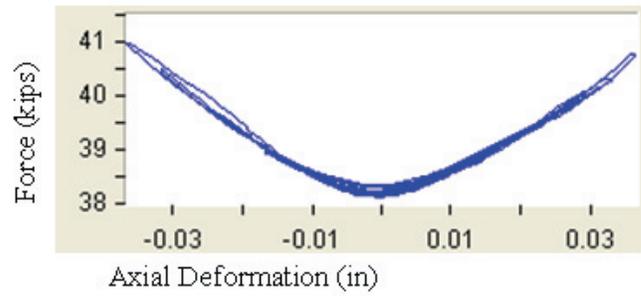


1<sup>st</sup> floor

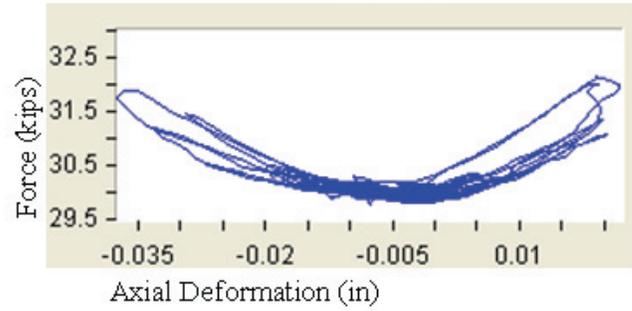


2<sup>nd</sup> floor

Figure 7-47 Force vs. Deformation of PT Strands For Test PE15 (+125%)

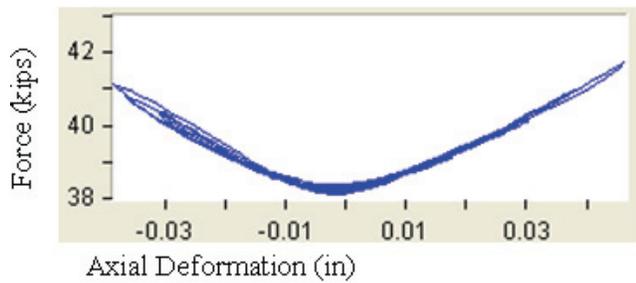


1<sup>st</sup> floor

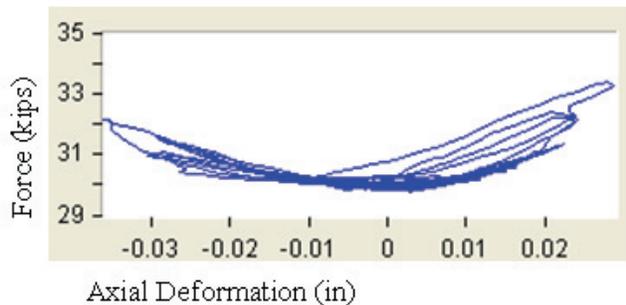


2<sup>nd</sup> floor

Figure 7-48 Force vs. Deformation of PT Strands For Test PER15 (-125%)

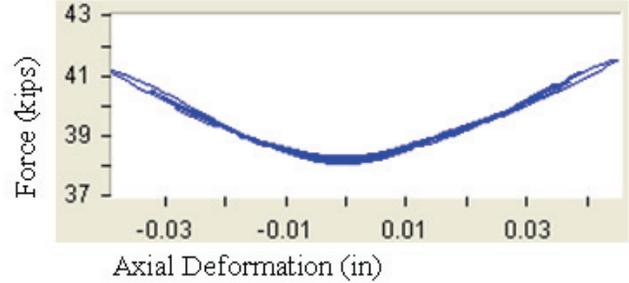


1<sup>st</sup> floor

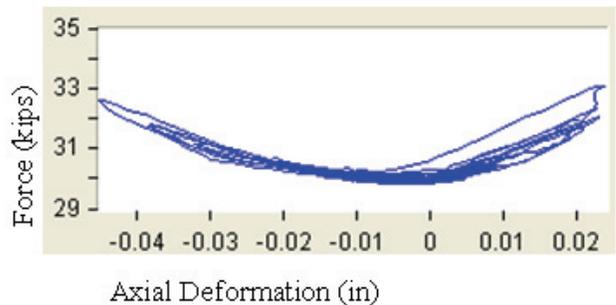


2<sup>nd</sup> floor

Figure 7-49 Force vs. Deformation of PT Strands For Test PE16 (+150%)



1<sup>st</sup> floor



2<sup>nd</sup> floor

Figure 7-50 Force vs. Deformation of PT Strands For Test PER16 (-150%)

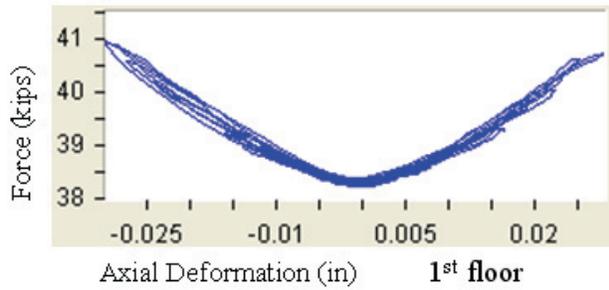


Figure 7-51 Force vs. Deformation of PT Strands For Test PE22 (+125%)

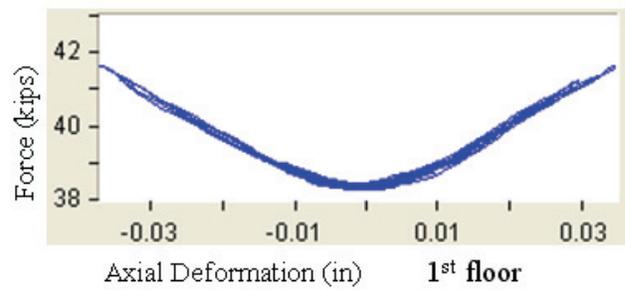


Figure 7-52 Force vs. Deformation of PT Strands For Test PER22 (-125%)

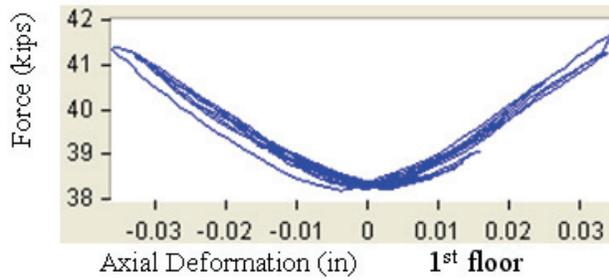


Figure 7-53 Force vs. Deformation of PT Strands For Test PE23 (+150%)

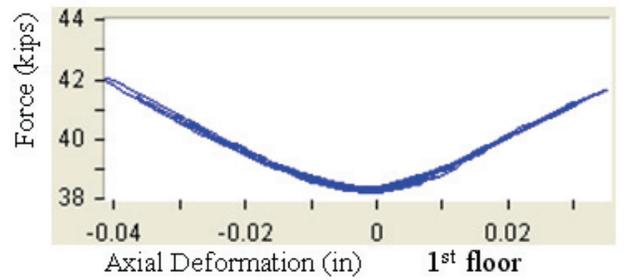


Figure 7-54 Force vs. Deformation of PT Strands For Test PER23 (-150%)

#### 7.4.4 Shear Transfer Issues

Under the large intensity earthquake excitations (e.g. 125% or 150% intensity), vertical movements of the beam ends in the SCPT model were observed through the video cameras installed on the 1<sup>st</sup> floor. These vertical movements of the beam were unexpected since the calculation of the initial model to predict the experimental results indicated that the shear transferred from the beam to column could be resisted by the friction between the beam and column interfaces. Those unexpected vertical movements were the result of insufficient friction. The reduction of this friction is believed to be due to the fact that the contact surface between the beam and column were not sand blasted. This result indicates that the sand blasting of the contact interface or, alternatively, the inclusion of vertical supports such as

shear tab connections may be necessary for SCPT connections. As shown in Fig. 7-55, a slotted shear tab can accommodate the horizontal movement of bolts to avoid the limit to the gap opening. Also, as discussed in Section 6.3, during the experimental tests, the vertical bars were installed on the top and bottom of beam ends in order to limit this vertical movement. After installing the vertical bars, the vertical movement of beams was eliminated and the gap openings in the beam-to-column joints still occurred, which demonstrates that the vertical bars were effective and made little impact on the gap opening.

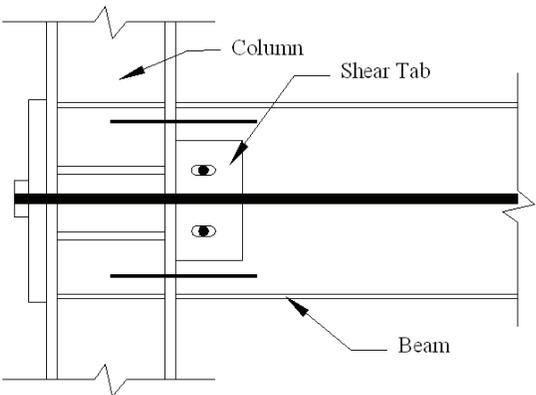


Figure 7-55 Beam-to-Column Connection in SCPT Structures

**7.5 Energy Evaluation**

The energy dissipation capacity is an important index to represent the seismic performance of a structure during earthquakes. Therefore, the energy evaluation for the SCPT and SMRF models is presented in this section. According to the theory of structural dynamics, the energy balance equation is given by:

$$E_K + E_D + E_F = E_I \tag{7.4}$$

where:

- $E_K$ : Kinetic Energy
- $E_D$ : Viscous Damped Energy
- $E_F$ : Hysteretic Energy (Elastic and Plastic)
- $E_I$ : Input Energy

Based on Eq(7.1), the absolute energy equation for the multi-story building subjected to an earthquake excitation was derived as follows (Uang and Bertero, 1990):

$$\frac{1}{2}[\dot{v}_a]^T M[\dot{v}_a] + \int [\dot{v}_r]^T C d[v_r] + \int [f_s]^T d[v_r] = \int \left( \sum_{i=1}^N m_i \ddot{v}_{ai} \right) dv_g \quad (7.5)$$

Where:

- $[\dot{v}_a]$ : Absolute velocity vector
- $M$ : Diagonal mass matrix
- $[\dot{v}_r]$ : Relative velocity vector
- $C$ : Damping matrix
- $d$ : Derivative Sign
- $[v_r]$ : Relative displacement vector
- $[f_s]$ : Restoring force vector
- $m_i$ : Lumped mass of the i-th floor
- $\ddot{v}_{ai}$ : Absolute acceleration at the i-th floor
- $v_g$ : Ground displacement

In Eq. (7.5), the kinetic energy, viscous damped energy and the input energy can be directly calculated by the obtained experimental results while the hysteretic energy may be calculated as the result of  $E_I - E_K - E_D$ . The velocity time histories, used to calculate the kinetic and viscous damped energy, are obtained by derivatives of the corresponding floor displacement time histories measured by tempsonic transducers. The damping matrix used in this calculation for the two models were obtained from the snap-back tests with 2.4% and 2.1% damping ratio of the 1<sup>st</sup> mode for the SCPT and SMRF models, respectively, as shown

in Table 7-6 and 7-7. The energy distribution time history during the 125%-intensity test for the two models is shown on Fig. 7-56 and 7-57, respectively. The label “hysteretic energy – story i” means the hysteretic energy absorbed by the whole structure due to the input energy generated by the motion of the i-the floor mass. The hysteretic energy includes the recoverable and unrecoverable strain energy. At the end of the excitation, it represents only the unrecoverable energy as a result of inelastic deformations of the structures. Comparing these two figures, it is found that the peak kinetic energy in the SMRF model is larger than that in the SCPT model, which suggests that the velocity response of SCPT is lower. The reduction of velocity response can decrease the lost to the velocity-sensitive nonstructural components during earthquakes.

The final energy distribution is presented in the table 7-8. The conventional engineering idea is that the higher energy dissipation the better seismic performance while as shown in this table, it is found that the total hysteretic energy of the SCPT model is lower than that of the SMRF model. However, the seismic performance of SCPT model is better than the SMRF model as shown in the previous sections from global and local responses. The reason is that although lower hysteretic energy is dissipated by the SCPT model compared to the energy dissipated by the SMRF model, the input energy flowing into the SCPT model is also reduced. It should be pointed out that the hysteretic energy absorbed in the SMRF model was due to the inelastic deformation of the beams and columns, which cause damage to the structural components. For the SCPT model, the ED bars played a sacrificial role without any damage to the beams and columns. Again, from the point of view of energy and repair

cost, it demonstrates the better seismic performance of the SCPT structure compared to the SMRF structure.

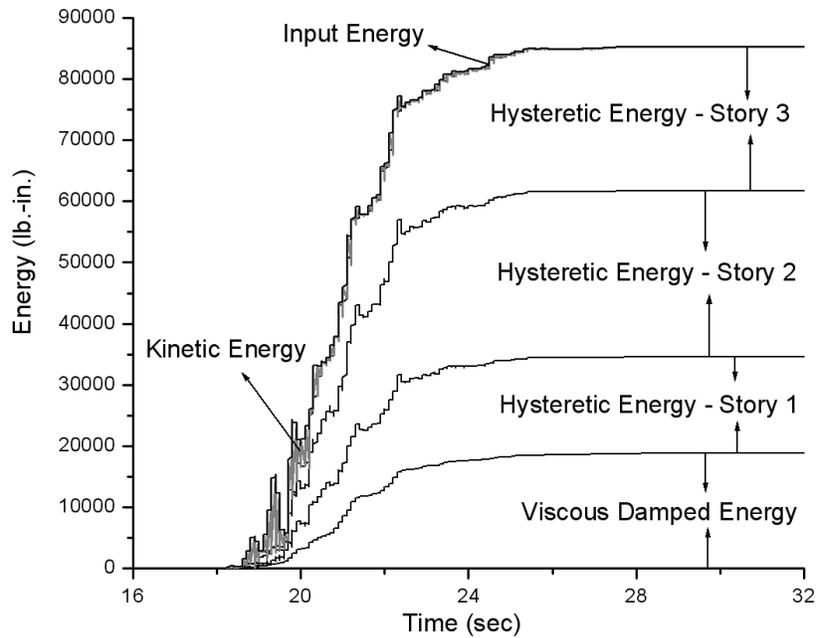


Figure 7-56 Energy Time History of the SCPT Model for Test PE6A (+125% in 1<sup>st</sup> phase)

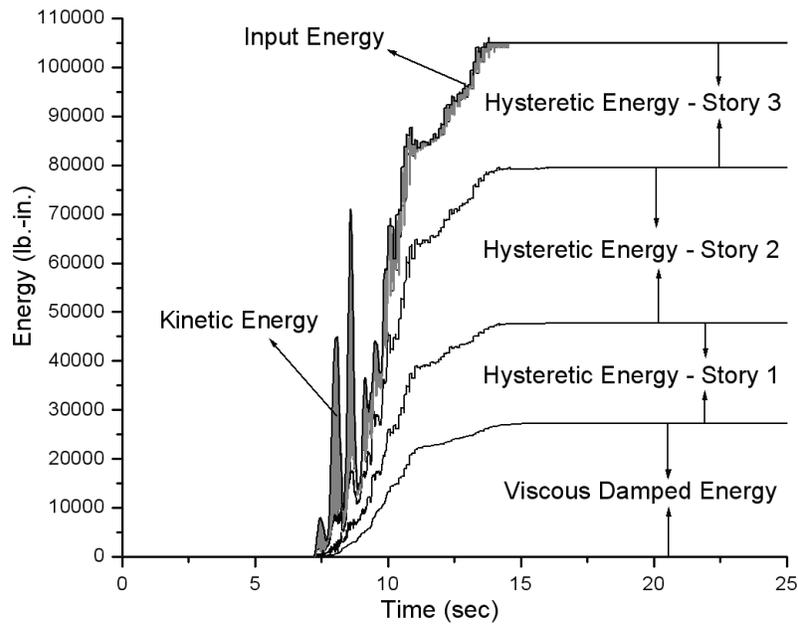


Figure 7-57 Energy Time History of the SMRF Model for Test ME06A (+125%)

**Table 7-8 Energy Distribution (lb-in)**

	Test No.	Input Energy	Viscous Damped Energy	Hysteretic Energy			
				1 <sup>st</sup> floor	2 <sup>nd</sup> floor	3 <sup>rd</sup> floor	total
SCPT Model	PE6A	85155	18757	15798	27148	23452	66398
SMRF Model	ME06A	105008	27323	20404	31674	25607	77685



## **CHAPTER 8**

# **COMPARISON BETWEEN NUMERICAL AND EXPERIMENTAL RESULTS**

Since differences between numerical predictions and experimental results exist in almost all studies, comparisons need to be done to evaluate the performances of the numerical in predicting the experimental results. This chapter presents the comparison between numerical and experimental results for both the SMRF and SCPT models. First, two initial numerical models, one of the SMRF and the other of the SCPT structure, are developed for the prediction of the experimental results obtained from the shake table tests. These two numerical models are then calibrated based on the structural properties measured during the shake table tests. Finally, a comparison between the results obtained from the initial numerical models, calibrated numerical models and experimental testing is discussed.

The purpose of this chapter is to identify the main factors that made an impact on the seismic performance of the test models during the shake table testing and evaluate the reliability of the numerical and experimental results by comparing them. Although the numerical model can almost never be totally identical as the real structures, efforts are made to take those influencing factors as much as possible into account in the numerical models in order to reduce the difference between the numerical predictions and experimental results.

### 8.1 Initial Numerical Models

For the purpose of predicting the seismic response of the test specimens, two different 2D numerical models were developed in the general purpose finite element software RUAUMOKO (Carr 2004).

As shown in Fig. 8-1, the SMRF structure is simulated by a 2D frame consisting of beam and column finite elements. Also in that figure, nodal numbers and element numbers (digits

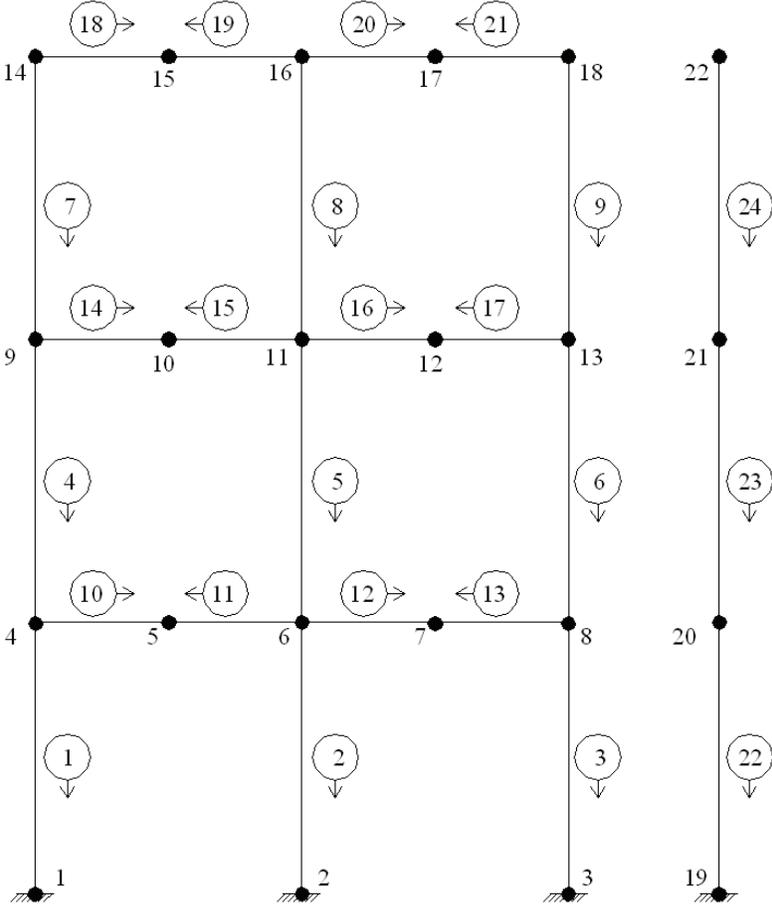


Figure 8-1 Initial Numerical Model of SMRF

with circle) are indicated and the arrow on the circle represents the local direction of each element. The dark points represent the location of the nodes. The super-spring elements on the right hand side (elements 22, 23 and 24) represent the Floor Mass Simulator (FMS) used in the tests. These super-spring elements provided only vertical support without any lateral resistance. The base connections were assumed to be fixed and the floor masses were assumed to be lumped in the central nodes (6, 11 and 16). A Rayleigh type damping model with the first and second modal damping ratios of 2% of critical was considered.

The initial numerical SCPT model is shown in Fig. 8-2. This model incorporated detailed SCPT beam-to-column connections, which can be found in Fig. 4-21(b) (See Section 4.5). The PT strands implemented in each floor are represented by one spring element with two end nodes on the exterior columns (e.g. node 4 and 8 for 1<sup>st</sup> floor). The ED bars (e.g. ED element between node 25 and 27) are located in the top and bottom of the beams and are connected to the beams by rigid bars (e.g. rigid bar element between nodes 27 and 24) since the beam finite element is a line without height in this 2D model. The multi-spring element (e.g. element between nodes 4 and 23) was used for gap elements of the beam-to-column joints. These contact elements had no tension stiffness and the compression stiffness was calculated based on the beam sections. The configurations of the lumped mass and gravity column frame are identical to those in the SMRF model.



experimental results. Since the rotation of the shake table was observed during the seismic tests, the initial models were calibrated mainly for this effect.

As shown in Fig. 6-8 (see Section 6.1), the shake table was supported by four vertical actuators (Actuator Z1, Z2, Z3 and Z4). During the seismic tests, the displacement and force feedbacks of those actuators were recorded. Figure 8-3 shows the relation between the vertical displacement and the force of actuator Z1 under the ME07 excitation (+150% intensity seismic test for the SMRF model). The axial stiffness of actuator,  $K_{actuator}$ , can be estimated as the slope in the figure:

$$K_{actuator} = \frac{(61.6 - (-30))}{(-0.25 - 0.174)} = 216 \text{ kips/in} \tag{8.1}$$

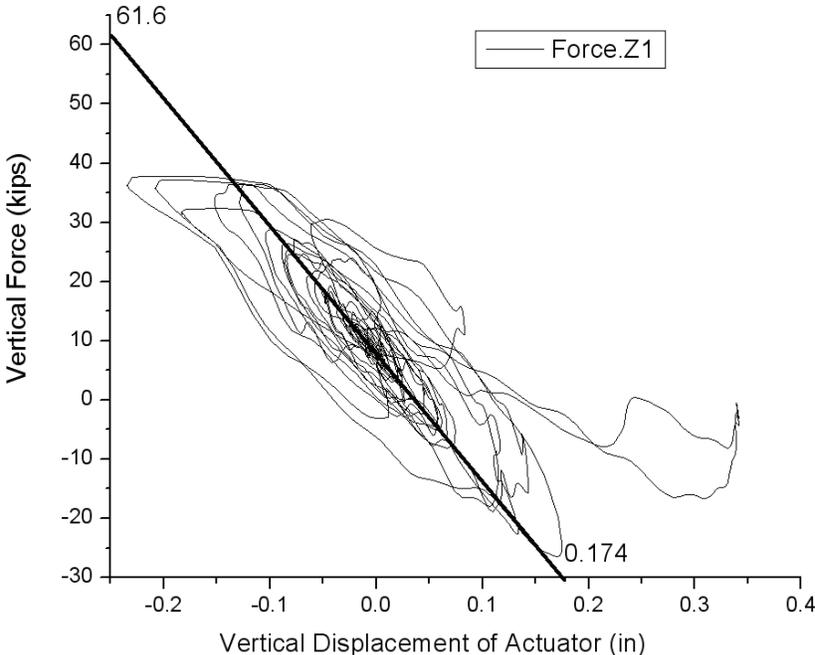


Figure 8-3 Force vs. Displacement of the vertical actuator Z1

Based on Eq. (8.1) and assuming that all four actuators have the identical stiffness, the equivalent rotational stiffness of the shake table,  $K_{rotation}$ , can be determined as follows:

$$K_{rotation} = 2K_{actuator} D_{actuator} \left( \frac{D_{actuator}}{2} \right) = 2 \times 216 \times 100.5 \times \left( \frac{100.5}{2} \right) = 2.182 \times 10^6 \text{ kips-in/rad} \quad (8.3)$$

where  $D_{actuator}$  is the distance between the two vertical actuators in the shaking direction.

To simply simulate this rotational stiffness of the shake table, a rotational spring finite element was added to the initial models, as shown in Fig. 8-4. The nodes 1, 2, 3 and 19 in the figure are the base nodes as shown in Fig. 8-1. These nodes are connected by rigid beam elements (e.g. elements between node 1 and 2) to simulate the stiff structure of the shake table. The new rotational element between nodes 2 and 23 was assumed to be rigid in the vertical and horizontal directions, while the stiffness obtained in Eq. (8.3) was used as rotational stiffness.

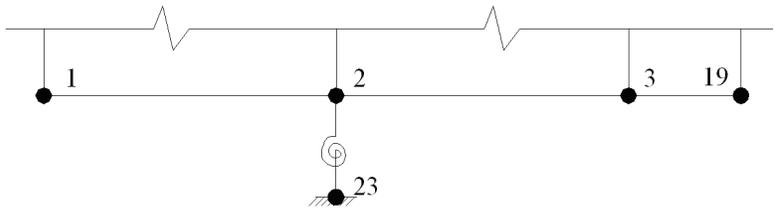


Figure 8-4 Rotational Element in the Numerical Calibrated Model

Both the SMRF and SCPT calibrated numerical models incorporated this rotational element. The calibrated numerical SCPT model was further simplified to a frame with ideal Self-Centering (SC) connections. The reason for this simplification is that in the initial numerical

SCPT model the Post-Tensioned (PT) force was added in the PT spring which was resisted by the axial resistance of the beams and the shear of columns especially in the first floor since the base is fixed, while in the experimental tests the PT strands were tensioned before the installation to the shake table and at that time the base is free. Due to the above reason, the initial PT force would be reduced if the SCPT connections in the initial model were used. Therefore, the simplified ideal SCPT model was used in the calibrated numerical model so that the self-centering property of this model was more similar to that of the experiment model. This ideal SCPT frame had the same node distribution as the SMRF model indicated in Fig. 8-1, but the beam ends connected to the columns incorporated a flag-shape hysteresis moment-curvature relation. The controlling parameters ( $\alpha$ ,  $\beta$  and  $\psi$ ) of this flag shape hysteresis relation were determined by the corresponding values shown in Table 5-8 (see Section 5.4)

### **8.3 Comparison of Results**

In this section, the results obtained from the initial and calibrated numerical models under 25% and 125% intensity seismic excitations are compared to those obtained from the experimental testing for the SMRF and SCPT models. The seismic response of the model structures under 25% and 125% intensity excitations can be seen as representatives of elastic and inelastic responses, respectively. Since the numerical models were simplified (e.g. no actual geometry for the beams or columns, only a line to represent the beam finite elements in the numerical models), the local responses such as strains or gap-openings are neglected

in this comparison. Only the modal natural frequencies and the global responses such as accelerations and displacements are compared.

Table 8-1 and 8-2 compares the modal frequencies obtained from the numerical models with those obtained experimentally. In these two tables, it can be seen that the modal natural frequencies of the initial numerical models are always higher than those measured experimentally for both the SMRF and SCPT structures. However, the modal natural frequencies of the calibrated numerical models are closer to those obtained experimentally. These results indicate that the rotation of the shake table reduced the stiffness of the whole structures leading to a reduction of the modal natural frequencies. It is also noted that the natural frequencies of the higher modes (2<sup>nd</sup> and 3<sup>rd</sup> modes) are not sensitive to the rotation effect.

**Table 8-1 Modal Natural Frequencies of the SMRF Model  
Obtained from Numerical and Experimental Results**

	Modal Natural Frequencies (Hz)			
	Numerical Model		Experimental Model	
Mode	Original	Calibrated	White Noise Test	Snap-Back test
1st	3.44	2.96	3	3.25
2nd	10.35	10.35	9.625	9.563
3rd	17.77	17.7	15.188	15.188

**Table 8-2 Modal Natural Frequencies of the SCPT Model  
Obtained from Numerical and Experimental Results**

	Modal Natural Frequencies (Hz)				
	Numerical Model		Experimental Model		
Mode	Original	Calibrated	White Noise test	Snap-Back test	Sine Sweep test
1st	3.6	2.96	2.75	2.973	2.75
2nd	10.57	10.35	9.141	8.919	9.063
3rd	18.61	17.7	14.891	14.534	14.313

The mode shapes calculated from the initial and calibrated numerical models are shown in Table 8-3. From this table, it is observed that the mode shapes of the calibrated model are similar to the ones of the original model, which indicates that rotation effect of the shake table has little influence on the mode shapes. By comparing those mode shapes with the ones obtained from the experimental testing (see Table 7-6 and 7-7 in Sec. 7.1.4), it can be observed that the mode shapes obtained numerically and experimentally have similar trends, while the observed deviations among them suggests that the numerical models are still different with the experimental models, which will be further discussed in Sec. 8.4.

**Table 8-3 Mode Shapes Obtained from the Initial and Calibrated Numerical Models**

	Original Model	Calibrated Model
Mode Shape of SMRF Model	$\begin{bmatrix} 0.29 & 1 & -1 \\ 0.69 & 0.82 & 0.89 \\ 1 & -0.87 & -0.32 \end{bmatrix}$	$\begin{bmatrix} 0.30 & 1 & -1 \\ 0.68 & 0.82 & 0.90 \\ 1 & -0.87 & -0.31 \end{bmatrix}$
Mode Shape of SCPT Model	$\begin{bmatrix} 0.29 & 1 & -1 \\ 0.67 & 0.86 & 0.86 \\ 1 & -0.88 & -0.30 \end{bmatrix}$	$\begin{bmatrix} 0.30 & 1 & -1 \\ 0.68 & 0.82 & 0.90 \\ 1 & -0.87 & -0.31 \end{bmatrix}$

The comparisons of displacements (inter-story drift) and accelerations between the numerical and experimental models of the SMRF structure are shown in Figs. 8-5 and 8-6. It can be observed that the displacement and acceleration responses predicted by the calibrated numerical models are much closer to those obtained experimentally. This observation verifies that the rotation effect of the shake table is one of the key factors influencing the experiment results since the main difference between the initial and calibrated numerical models is the addition of a rotational spring as discussed earlier. It is also noted that the experiment results show higher inter-story drifts and lower accelerations for the high intensity excitations (+125% intensity excitation) than those of the initial and calibrated models. This result suggests that the test specimens achieved a better seismic performance than that predicted by the numerical analyses from a point of view of larger ductility and lower accelerations. Therefore, the results from the numerical analyses can be considered conservative.

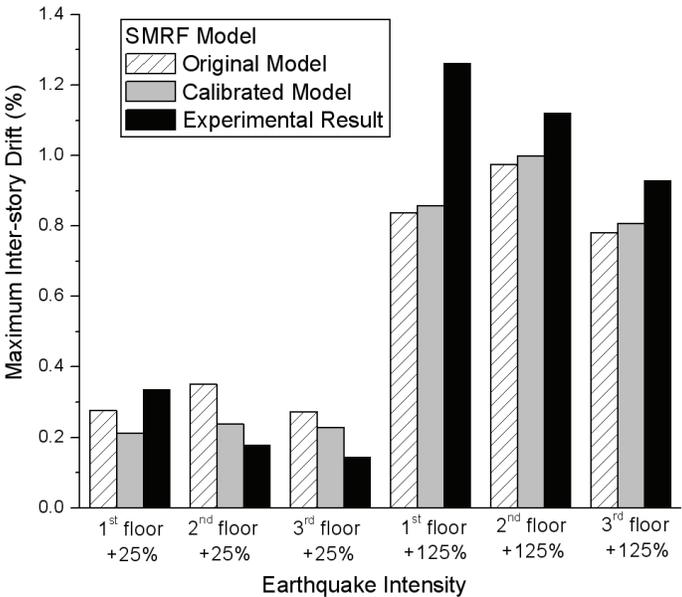


Figure 8-5 Maximum Inter-story Drifts for the SMRF Model

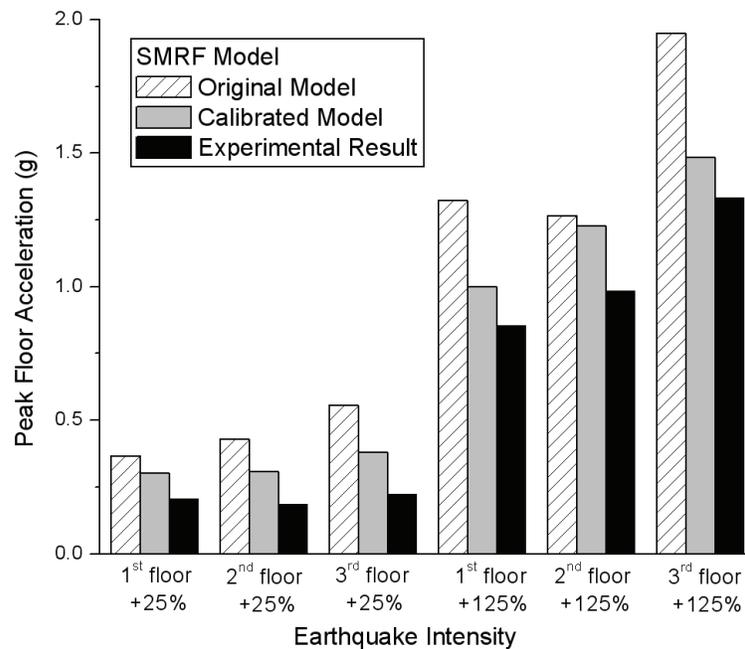


Figure 8-6 Peak Floor Accelerations for the SMRF Model

Figures 8-7 and 8-8 show a comparison of the maximum inter-story drifts and accelerations predicted by the two numerical models with those obtained experimentally for the SCPT structure. Similar to the SMRF structure, the predictions of the calibrated numerical model of the SCPT structure are in better agreement with the experimental results than that of the initial numerical model. It is observed that inter-story drifts predicted by the calibrated model are larger than those of the initial model and experimental results. However, this deviation is as low as about 0.1% of the story height (about 1mm) for the 2<sup>nd</sup> and 3<sup>rd</sup> floors under the 25%-intensity seismic excitation. It is also observed in Fig. 8-8 that a relatively large deviation of the accelerations exists between the calibrated numerical model and the experimental results for higher (+125%) intensity seismic testing. This difference may be due to the vertical movement of the beams of the SCPT structure, which lowered the

stiffness after gap-opening and resulted in lower acceleration responses compared to those predicted by the numerical analysis.

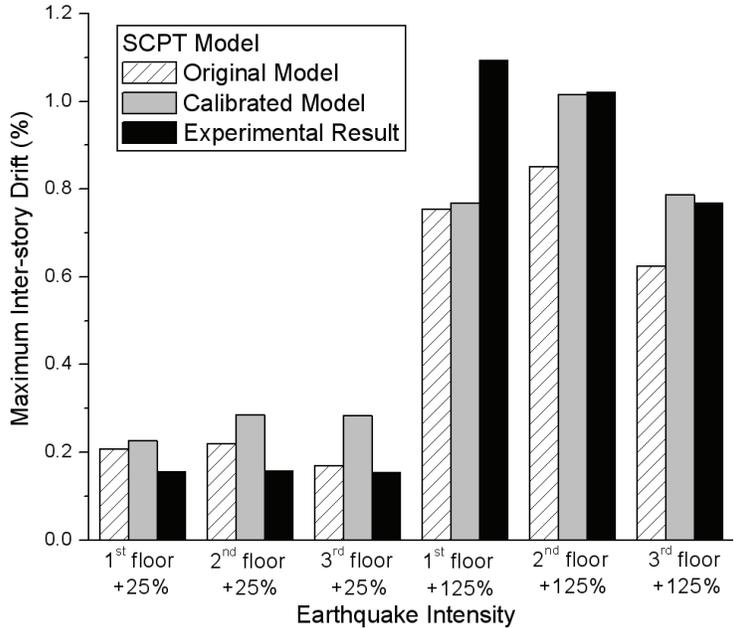


Figure 8-7 Maximum Inter-story Drift of the SCPT Model

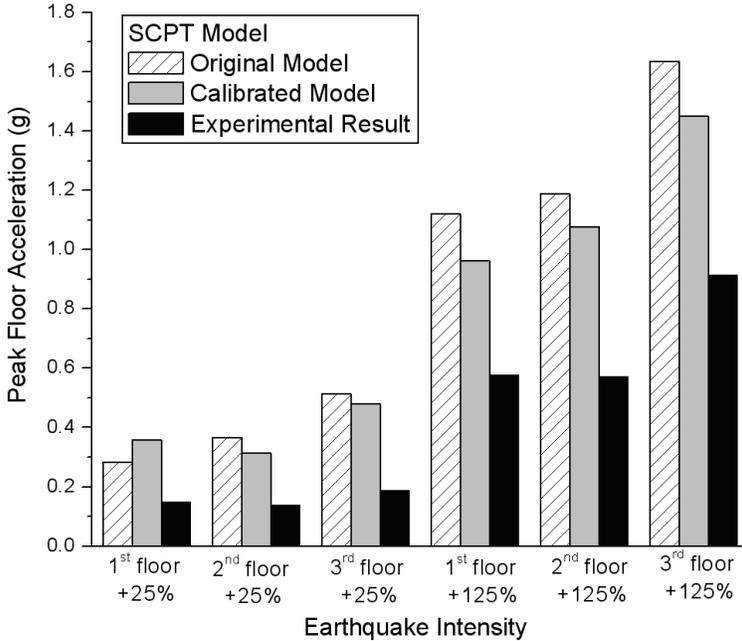


Figure 8-8 Peak Floor Acceleration of the SCPT Model

## 8.4 Summary

As discussed in the previous sections, by considering the effect of the shake table rotational stiffness, the calibrated numerical models for both SMRF and SCPT structures achieved better correlation with the experimental results than the initial models of the same structures. By comparing the responses between the numerical and experimental results, it can be concluded that the predictions of the numerical analysis were more conservative than the actual experimental results.

However, differences between the calibrated numerical model and experimental model still exist. The numerical models are 2D models while the real structures in testing are 3D ones. The slight out-of-plane torsion may have some influence on the response, which can not be simulated in the 2D numerical models. Another issue is that, as shown in Fig. 8-4 (see Section 8.2), the earthquake excitation arises from shaking of the ground (i.e. node 23) in the calibrated numerical models, while the real situation in the seismic testing is through the shaking of the table (i.e. node 1, 2 and 3). As shown in Fig. 8-3 (see Section 8-2), the vertical actuator supporting the shake table exhibited a force-displacement hysteresis loop under seismic excitations. This nonlinear behavior was not simulated by the simple rotational spring added to the calibrated numerical models due to the difficulty to define the property of this irregular hysteresis loop. Therefore, to better match the experimental results, the numerical models need to be refined further to completely reflect the behavior of the test structures.

Although the above differences occurred, both the numerical predictions and experimental results confirmed that the Self-Centering Post-Tensioned (SCPT) frame structure achieved better acceleration response and similar displacement response compared to the Steel Moment Resisting Frame (SMRF) structure under seismic excitations. From this point of view, it is believed that the shake table testing reproduce the same trends predicted by the numerical analysis.

## **CHAPTER 9 CONCLUSIONS**

In this report, numerical and experimental studies of the seismic performance of Self-Centering Post-Tensioned (SCPT) steel frames were conducted. The results were presented mainly in two parts: an analytical study from Chapter I to Chapter IV and an experimental study from Chapter V to Chapter VIII. In this final chapter, the conclusions drawn from these two parts are reviewed and the suggestions for the future studies are discussed.

### **9.1 Summary and Conclusions from Numerical Studies**

In the numerical studies, the seismic performances of Single-Degree-of-Freedom (SDOF) Self-Centering Systems (SCS) and Elasto-Plastic Systems (EPS) were first investigated through nonlinear dynamic analyses. A Relative Performance Index (RPI) was developed to evaluate the performance of SCS in reducing both relative displacements and absolute accelerations in structures under earthquake hazards. Based on this RPI, a parametric analysis was conducted to evaluate the influence of the various hysteretic parameters of SCS on the seismic response of SDOF. The results of this parametric study are included in appendices A and B. Thereafter, numerical models were developed to evaluate the seismic performance of the MCEER Demonstration Hospital building re-designed with SCPT connections. A practical re-design procedure for SCPT connections was proposed. Based on the results of these analyses, the following conclusions can be made:

- (i) The comparison of seismic performance of SDOF SCS and EPS, excited by two ensembles of 25 MCEER ground motions having 2% and 10% probabilities of exceedance in 50 years for Southern California, demonstrated that at least one SDOF SCS with optimum hysteretic parameters ( $\alpha$ ,  $\beta$ ,  $\eta$  and  $\psi$ ) can be found to exhibit better seismic responses than a corresponding SDOF EPS. This conclusion suggests that similar SCS parameters can be found in more complex Multi-Degree-of-Freedom (MODF) systems.
- (ii) The results of pushover analyses, seismic analyses and fragility analyses on the MCEER Demonstration Hospital indicated that the seismic performance of SMRF redesigned with ideal SCS exceeded that of original SMRF. Furthermore, these results also validated the feasibility of using the RPI as a guide for the design of SCS MDOF systems.
- (iii) Further numerical seismic study on SMRF systems re-designed with SCPT connections showed that the seismic performance of the SCPT steel frame exceeds not only that of original SMRF but also that of the model redesigned with ideal SCS. This seismic analysis also validated the practical re-design procedure of SCPT frame buildings.
- (iv) After re-design of the MCEER Demonstration Hospital incorporating the SCPT connections, the seismic results indicated that some base columns yields in severe

earthquakes, such as some of the MCEER ground motions with 2% probability of exceedance in 50 years. Therefore, the base columns should be considered to be re-designed with SCS or to be reinforced.

- (v) The reductions of the seismic responses of SMRF re-designed with SCS also reduce the recovery cost after earthquakes. The materials of SCPT connections are readily available steel products, which can be replaced with relative lower cost compared to the replacement of the structural components such as beams or columns after seismic hazards.
- (vi) The residual drift of structures re-designed with SCS is largely reduced or eliminated and no permanent inelastic deformation associated with the main structural components occurs, which save a lot of cost to return the conventional SMRF structure to its original position and repair the damage of the beams or columns in the SMRF structure after earthquakes.
- (vii) The SCPT frame structure can be designed for different purposes according to the requirements of buildings such as more concern on reducing acceleration responses for buildings with more acceleration-sensitive components or decreasing displacement responses for structures with more displacement-sensitive components. The weighting factor, RPI, can be used for guiding the design of structures with different seismic index sensitive components.

## 9.2 Summary and Conclusions from Experimental Studies

To experimentally validate the seismic performance of SCPT and SMRF structures, two 1/3 scale frames (i.e. SCPT and SMRF) were designed and tested under a selected MCEER ground motion on a shake table at the University at Buffalo. (The shop drawings of these two models are shown in Appendix C and D.) The initial dynamic properties of two models were determined based on identification tests including white noise, snap-back and sine-sweep tests. The global seismic responses such as peak accelerations and displacements, and the local responses from beams, columns, Post-Tensioned (PT) strands and Energy-Dissipating (ED) bars were presented and compared for the SCPT and SMRF structures. (The acceleration, displacement and strain time histories under seismic excitations are shown in Appendix E, F and G.) Also, the energy distribution in the two models during the seismic tests was evaluated. Finally, the predictions of initial and calibrated numerical models were compared to the experimental results. Based on those comparisons, the following conclusions can be drawn:

- (i) The comparisons of the seismic performance between the SCPT and SMRF models demonstrated that the SCPT model performed better than SMRF model under the selected MCEER simulated ground motion.
- (ii) The good seismic behavior of two models under severe seismic excitations (125% and 150% intensity earthquakes) satisfied the design objectives.

- (iii) The reduction in acceleration response and similar displacement response of the SCPT model, compared to those in the SMRF model, validated the design procedure focused mainly on acceleration issue since the Relative Performance Index (RPI) used in the design of the SCPT connections had a 70% weighting factor on the acceleration response and 30% for the displacement response.
- (iv) From the local experimental responses, the PT strands and ED bars exhibited good abilities to provide recentering force and to dissipate energy.
- (v) Unexpected vertical movements of the beam ends in the SCPT model was observed under the severe seismic excitations, which suggested that the beam ending surfaces connected to the columns should be sand blasted to increase the friction or additional shear tabs should be installed to the beam-to-column joints to supply more shear resistance.
- (vi) The better seismic performance of the SCPT model in the 1<sup>st</sup>, 2<sup>nd</sup> and 3<sup>rd</sup> phases of testing indicated that after two time severe earthquake attacks, the SCPT structure still exhibited better behavior than that of SMRF under the 1<sup>st</sup> time earthquake hazard.
- (vii) The strain responses indicated that after severe earthquakes, the beams of the SMRF structure were damaged by yielding while only the ED bars yielded in the SCPT model without any damage to the beams. These results verified that it

would cost much more to repair the beams of the SMRF structure compared to the SCPT building.

- (viii) The energy analysis suggested that although the energy dissipated by the SCPT model was less than that absorbed by the SMRF model, the total energy flowing to the SCPT model was still less than that traveling to the SMRF model. This phenomenon may be associated with the better seismic performance of the SCPT model. The different hysteresis energy distribution (i.e. in yielding beams of SMRF model and in yielding ED bars of SCPT model), again demonstrated the lower repair cost for the SCPT model.
- (ix) The same trends (i.e. the SCPT model exhibited better seismic performance) were obtained from the initial numerical predictions, calibrated numerical models and experimental models. The results of the calibrated numerical models were closer to the experimental results compared to the initial numerical prediction, which verified the impact of the rotation effect of the shake table.

### **9.3 Suggestions for Future Studies**

Since there is lack of research on implementing this new type of SCPT structures to the practical buildings, more efforts should be made to investigate the real applications and resolve the practical problems.

The practical re-design procedure of the SCPT steel frame proposed in this report is slightly complicated, therefore, it need to be further investigated to obtain more practical and simpler design equations without iterations to achieve the optimal and feasible SCPT parameters. Other research efforts on SCS may focus on the application of SCS in the base columns or other column connections in order to prevent the possible yielding of those base columns during severe earthquakes.

The limit to the gap opening of the beam-to-column joints in the SCPT structures due to the floor slab should be considered in the further study, since this limit may leads to the increase of compression in the beams and even yielding of the beams before the design gap opening angles.

The full scale shake table test of SCPT frame may be conducted to obtain more practical performance during earthquakes although it requires much larger shake table to support the structure.

Further research may be conducted to compare the seismic performance of SCPT frames with that of the frames incorporating other passive control methods such as base isolation systems, tuned mass dampers and viscous dampers, although the cost for other passive control methods is apparently higher than that of SCPT frames.

To predict the experimental results more precisely, 3D models should be developed to account for the details of beam-to-column joints and other components such as floor slabs

and gravity support frames. Also, the influence due to the shake table such as rotations and hysteresis properties of vertical supporting actuators should be further investigated in order to achieve much more accurate results compared to the real responses from the practical buildings during earthquakes.

## CHAPTER 10

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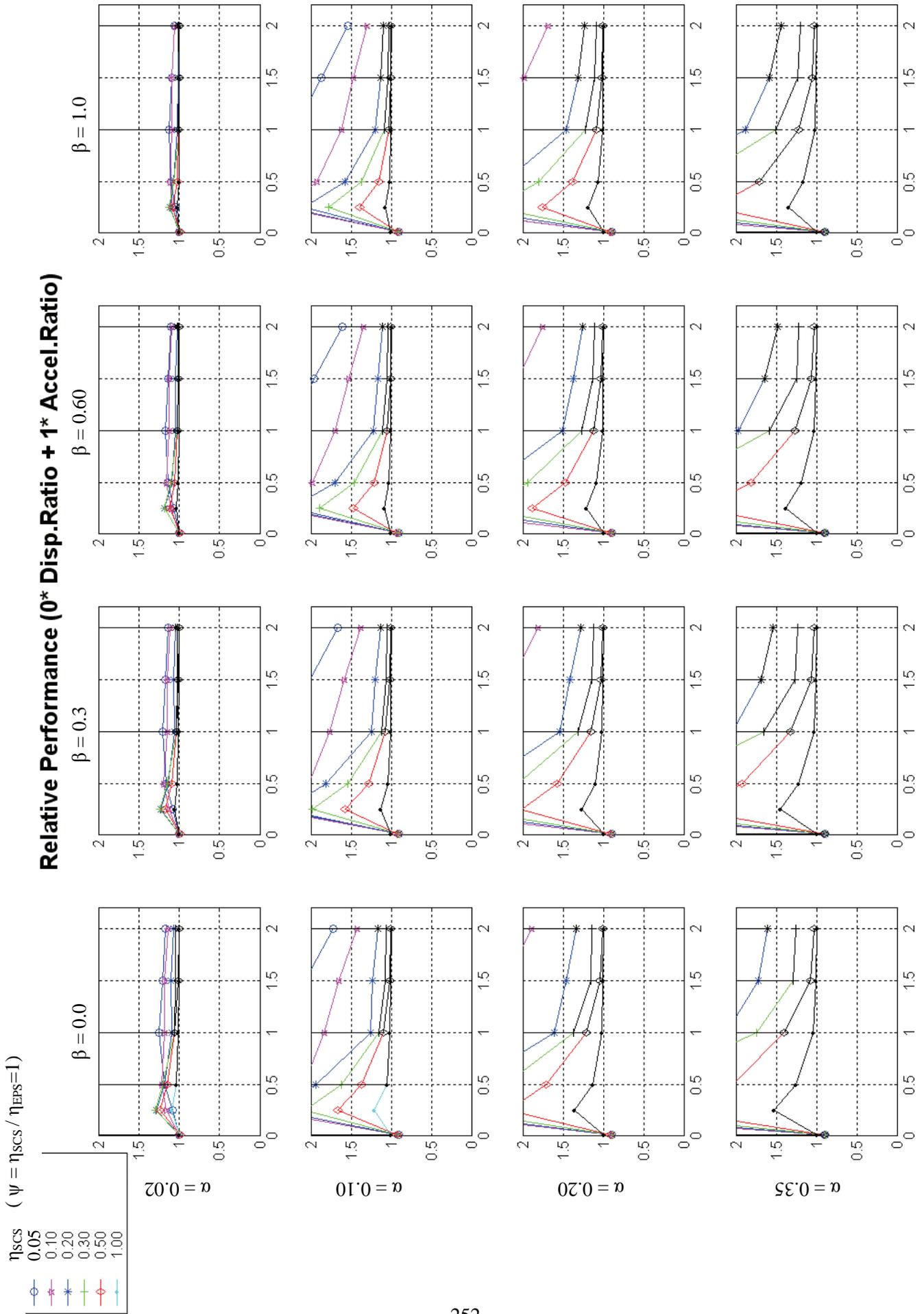
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**Appendix A      Relative Performance Index (RPI) under  
MCEER ground motions having a 2%  
probability of exceedance in 50 years in  
California, U.S.**

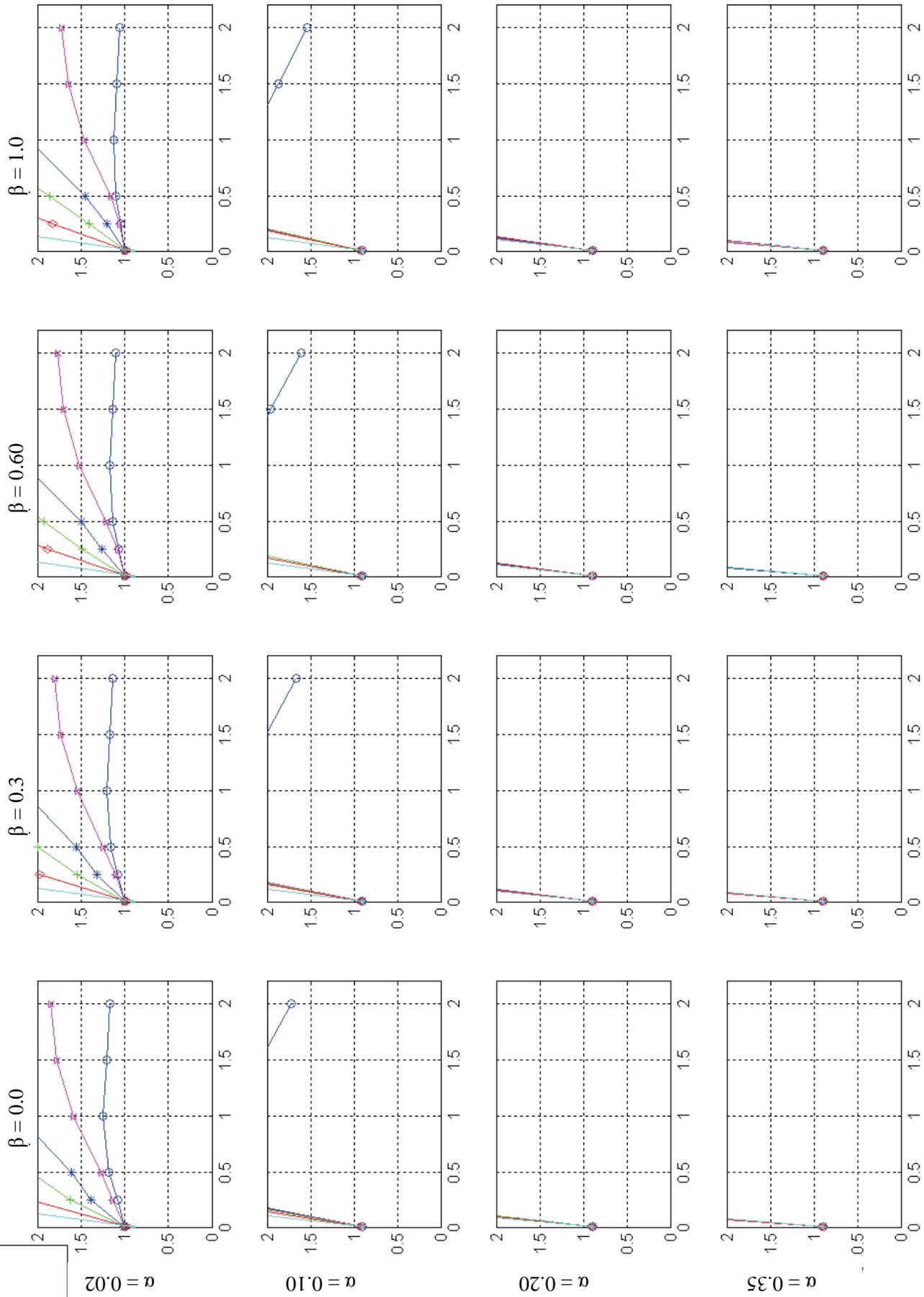
**Relative Performance ( $0^* \text{ Disp.Ratio} + 1^* \text{ Accel.Ratio}$ )**



$\Psi$  ( $\eta_{EPS} = 0.05$ ;  $\psi = \eta_{scs} / \eta_{EPS}$ )



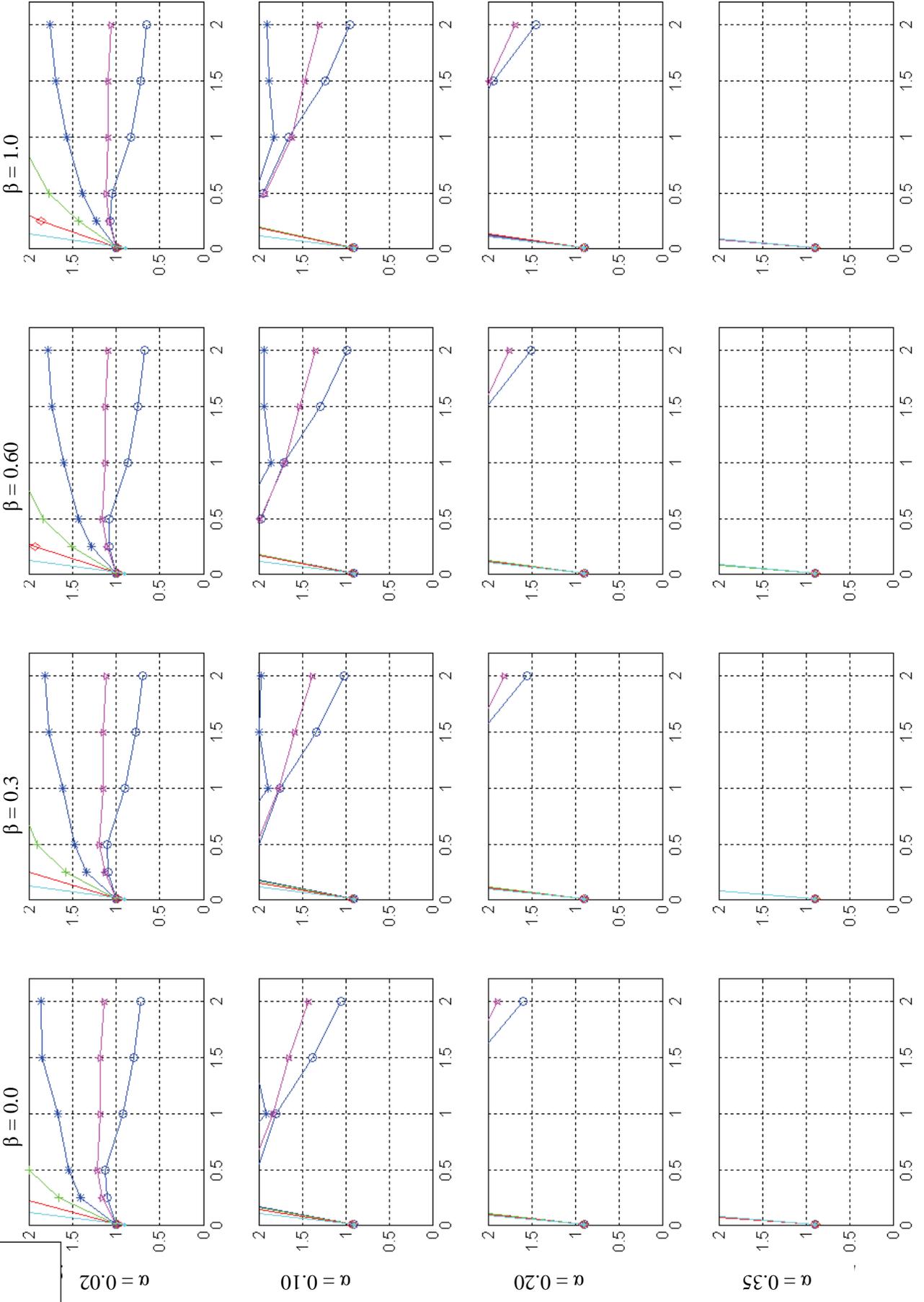
**Relative Performance (0\* Disp.Ratio + 1\* Accel.Ratio)**



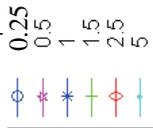
# Relative Performance (0\* Disp.Ratio + 1\* Accel.Ratio)

$\Psi$  ( $\eta_{EPS} = 0.10; \Psi = \eta_{SCS} / \eta_{EPS}$ )

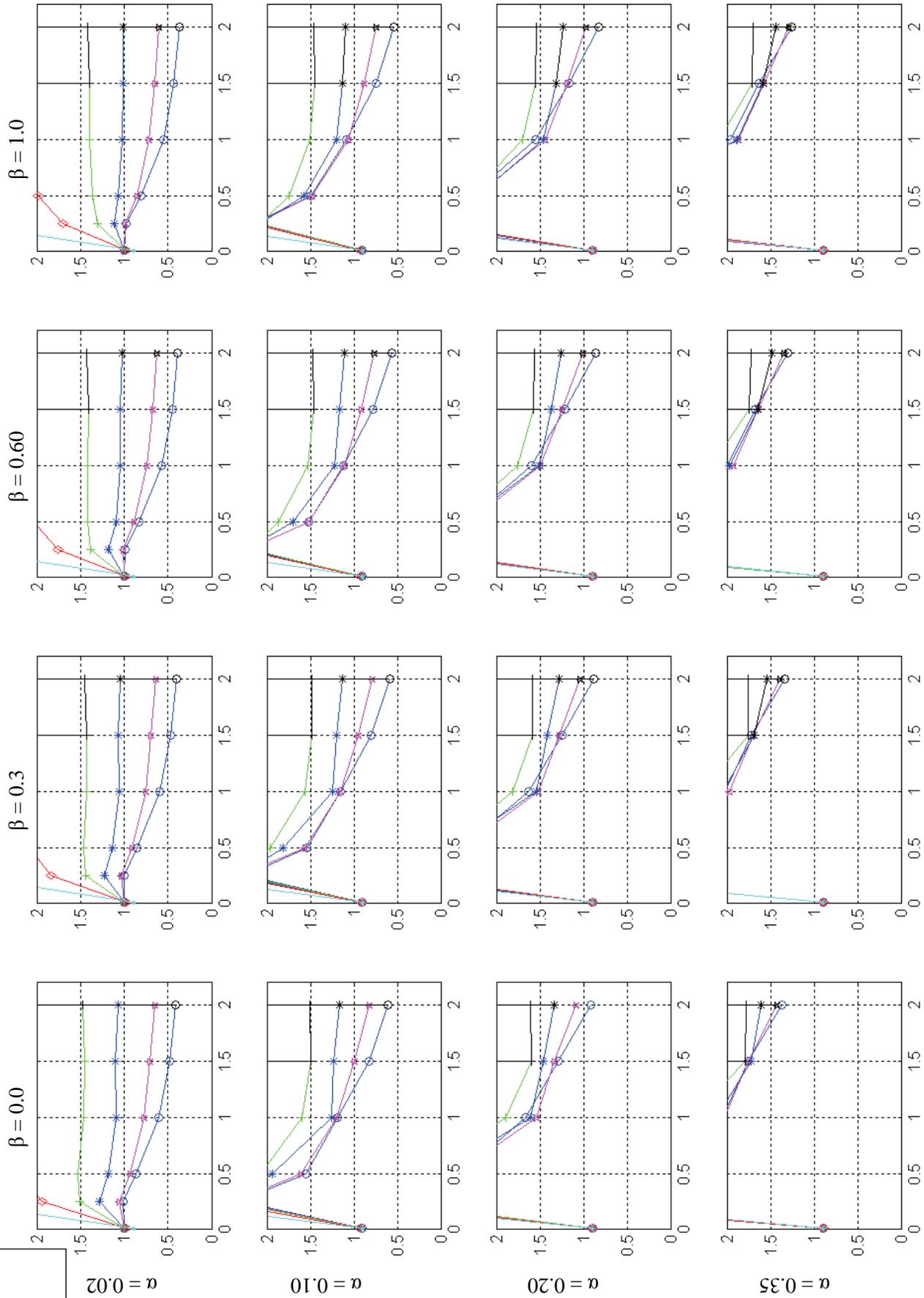
- $\Psi$  0.5
- 1
- 2
- 3
- 5
- 10



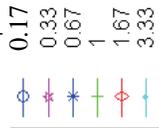
$\Psi$  ( $\eta_{EPS} = 0.20$ ;  $\Psi = \eta_{SCS} / \eta_{EPS}$ )



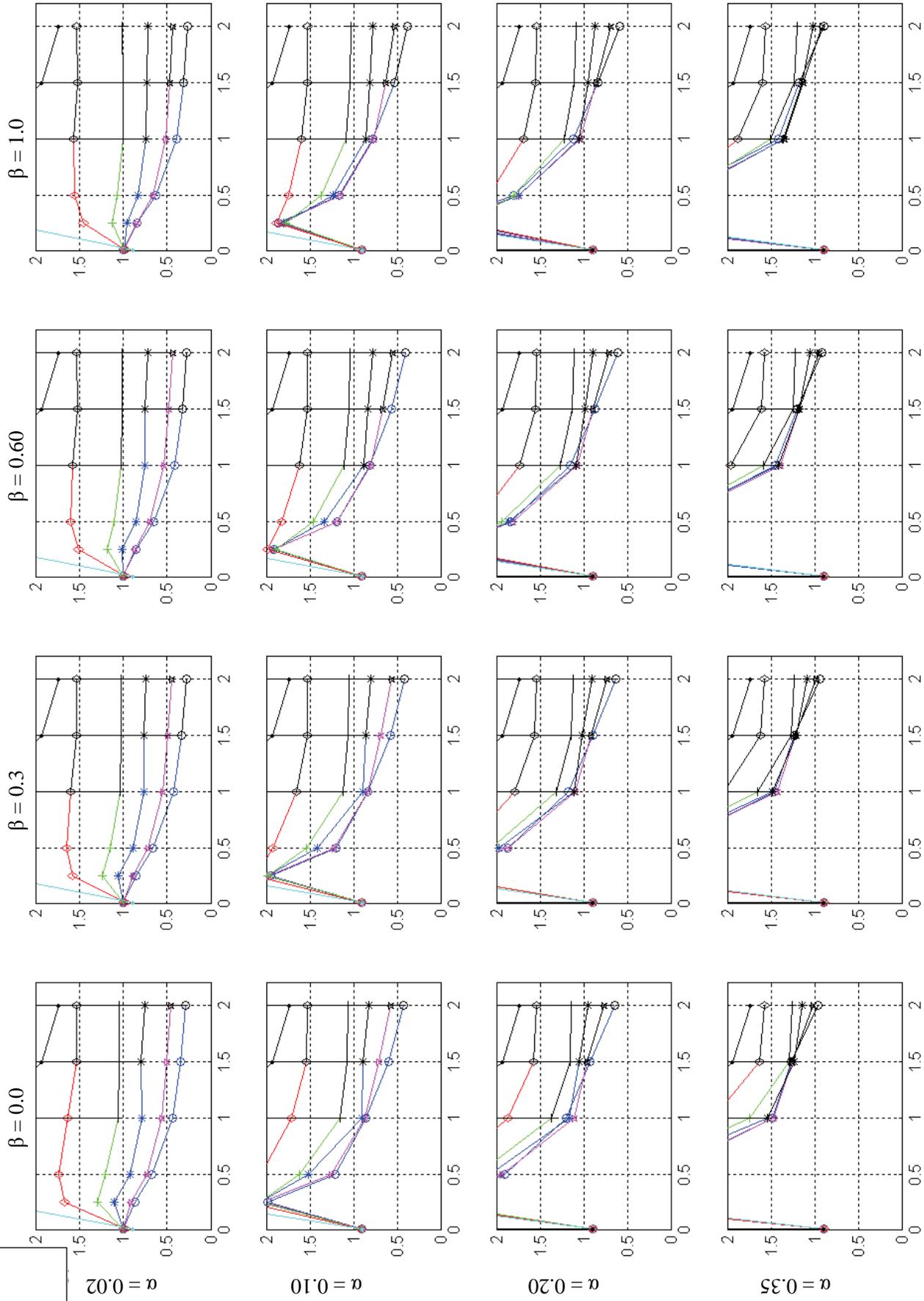
**Relative Performance (0\* Disp.Ratio + 1\* Accel.Ratio)**



$\Psi$  ( $\eta_{\text{EPS}} = 0.30; \psi = \eta_{\text{SCS}} / \eta_{\text{EPS}}$ )



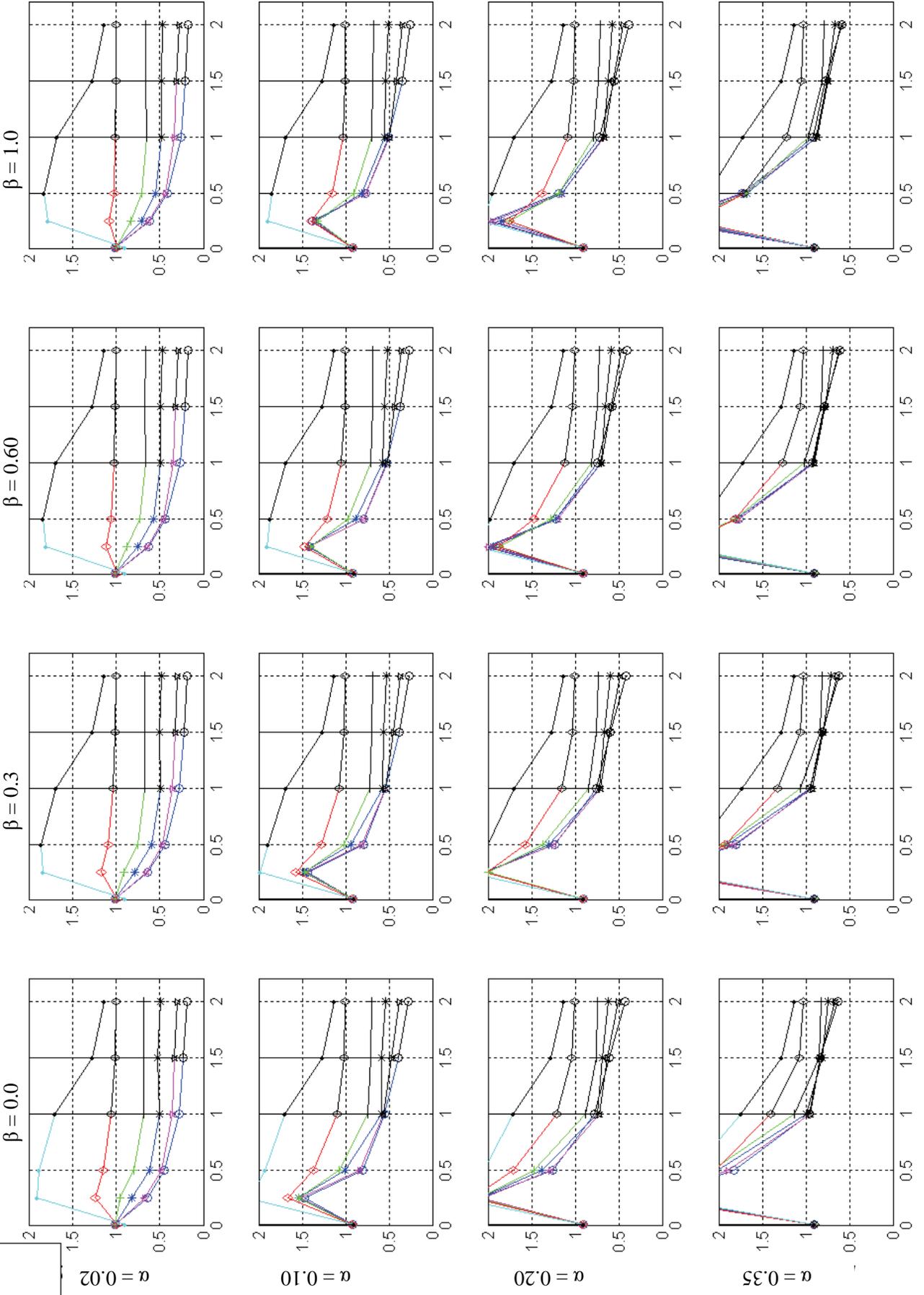
### Relative Performance (0\* Disp.Ratio + 1\* Accel.Ratio)



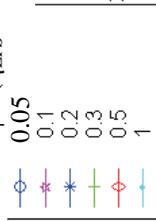
**Relative Performance (0\* Disp.Ratio + 1\* Accel.Ratio)**

$\psi$  ( $\eta_{EPS} = 0.50; \psi = \eta_{scs} / \eta_{EPS}$ )

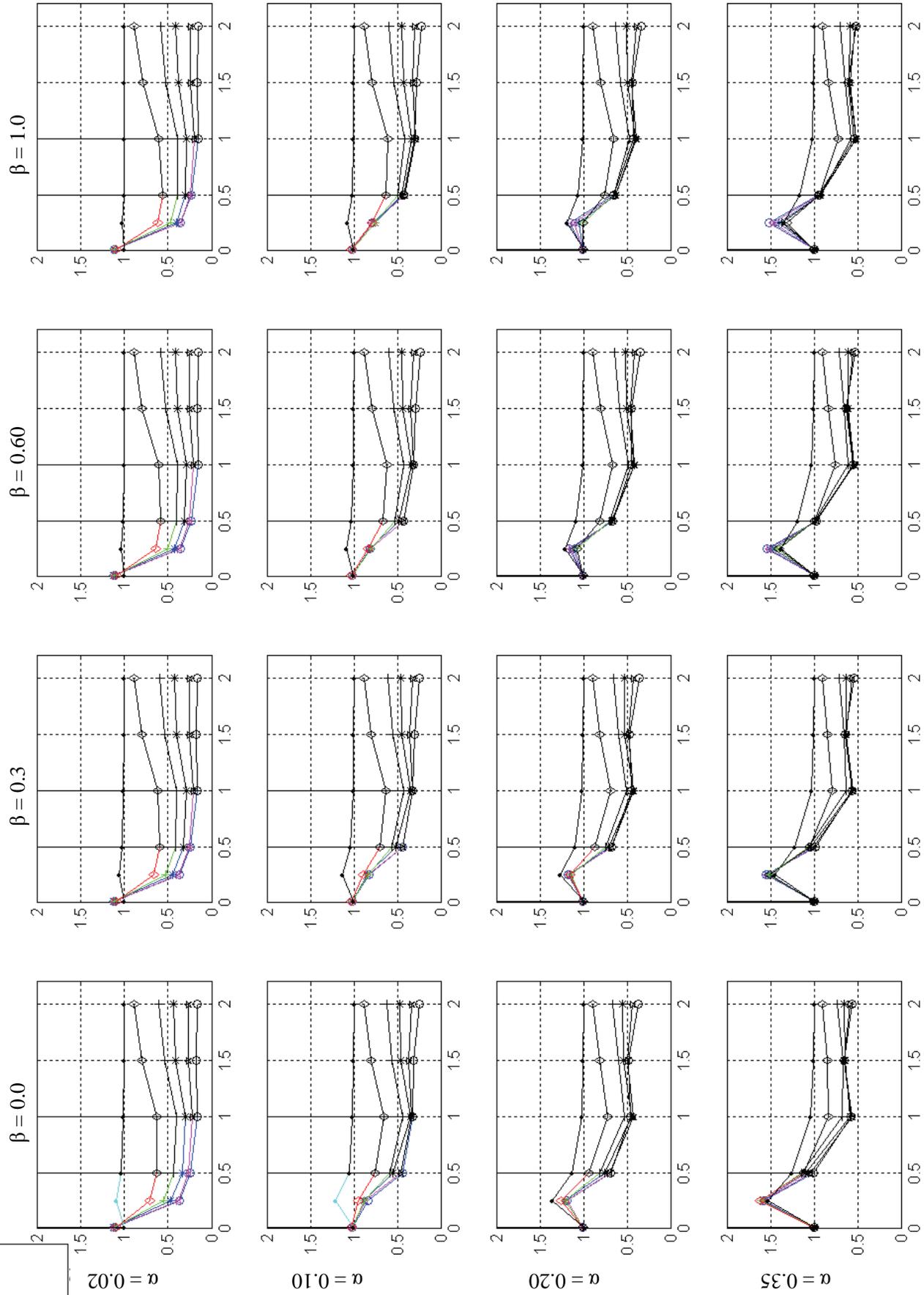
- 0.1
- \* 0.2
- ☆ 0.4
- ◇ 0.6
- ◇ 1
- ◇ 2



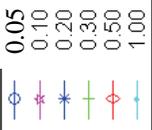
$\Psi$  ( $\eta_{\text{EPS}} = 1.00; \Psi = \eta_{\text{SCS}} / \eta_{\text{EPS}}$ )



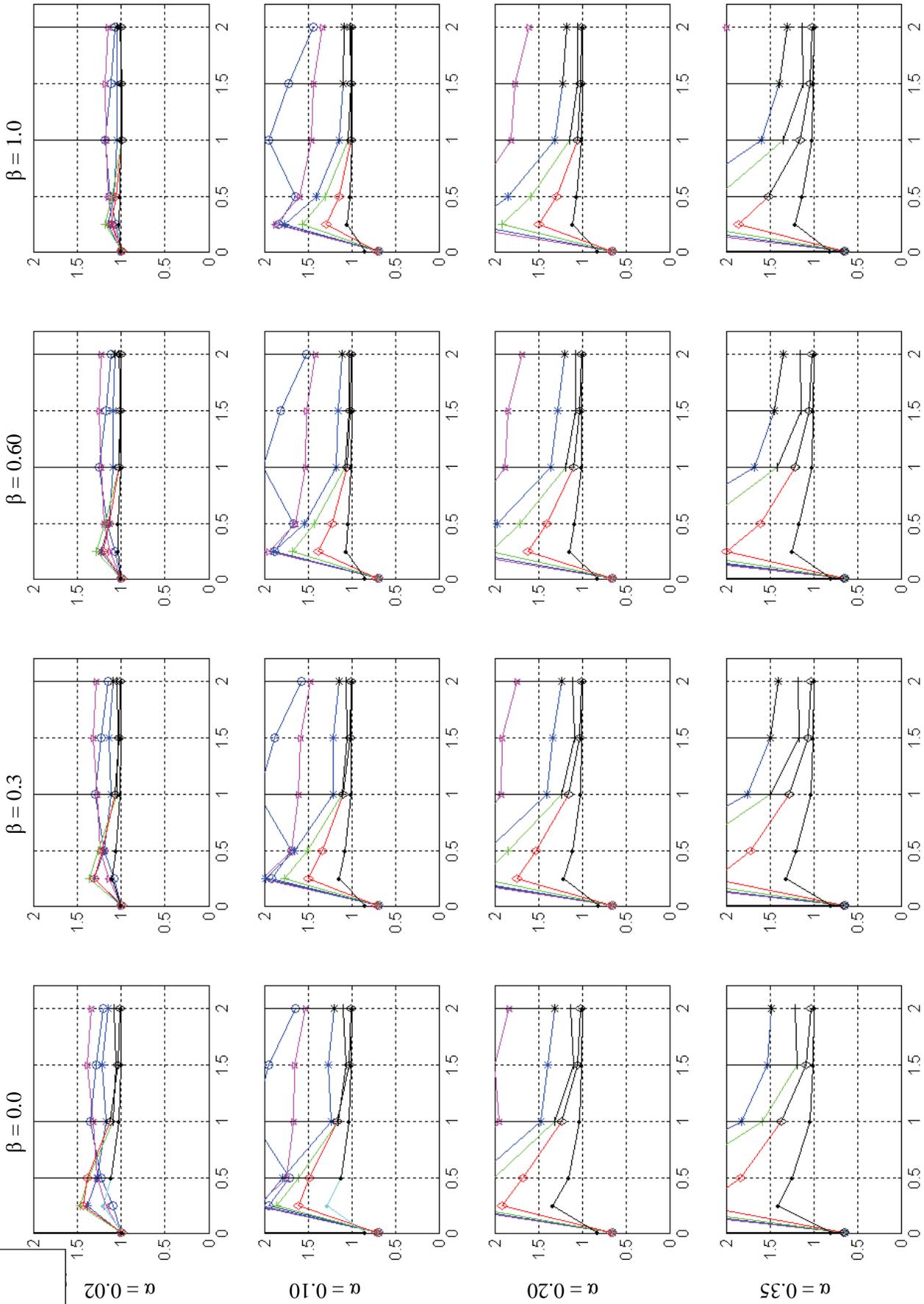
### Relative Performance (0\* Disp.Ratio + 1\* Accel.Ratio)



$\eta_{SCS}$   
( $\Psi = \eta_{SCS} / \eta_{EPS=1}$ )



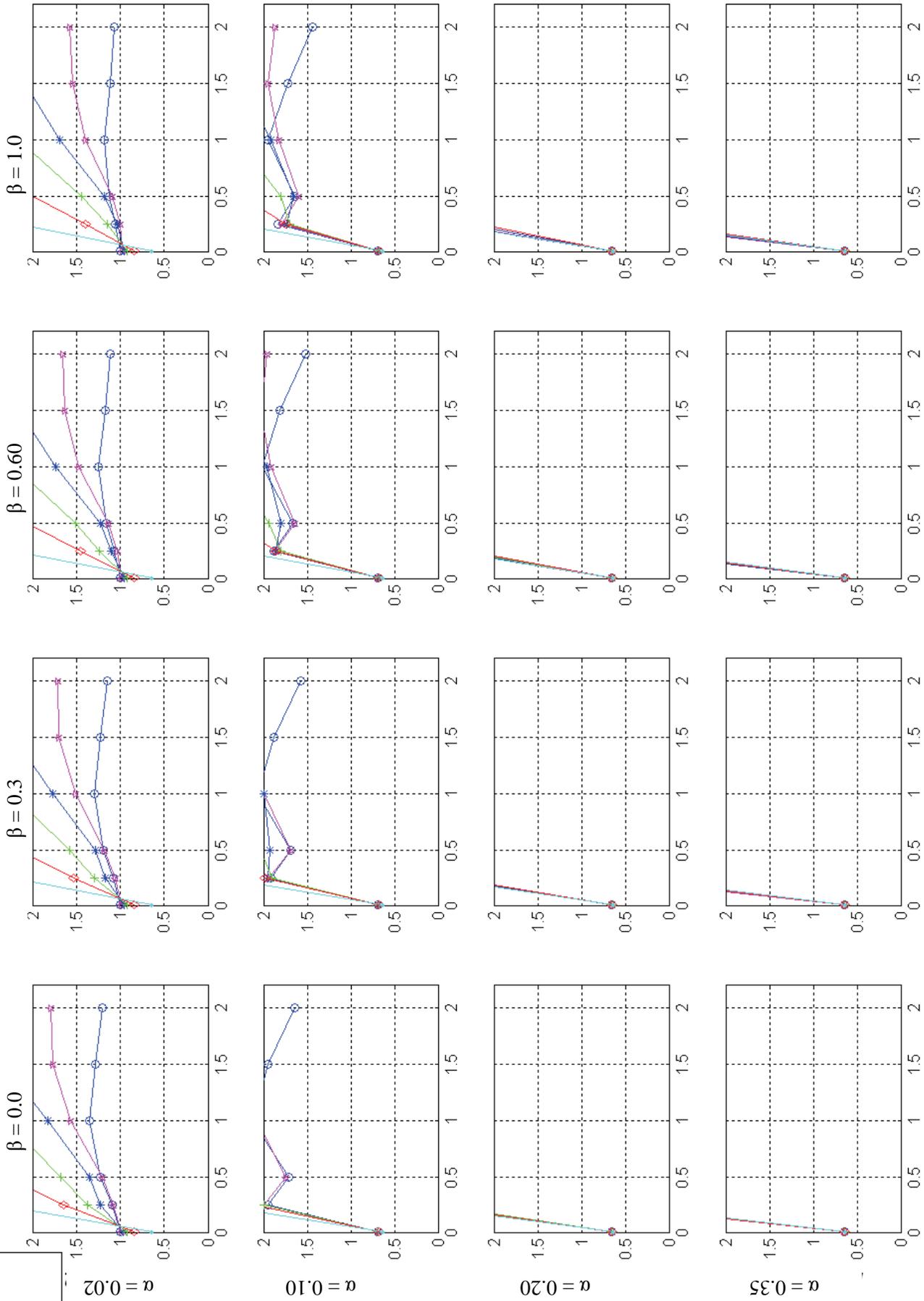
### Relative Performance (0.3\* Disp.Ratio + 0.7\* Accel.Ratio)



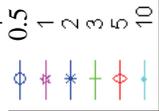
$\Psi$  ( $\eta_{EPS} = 0.05; \Psi = \eta_{SCS} / \eta_{EPS}$ )



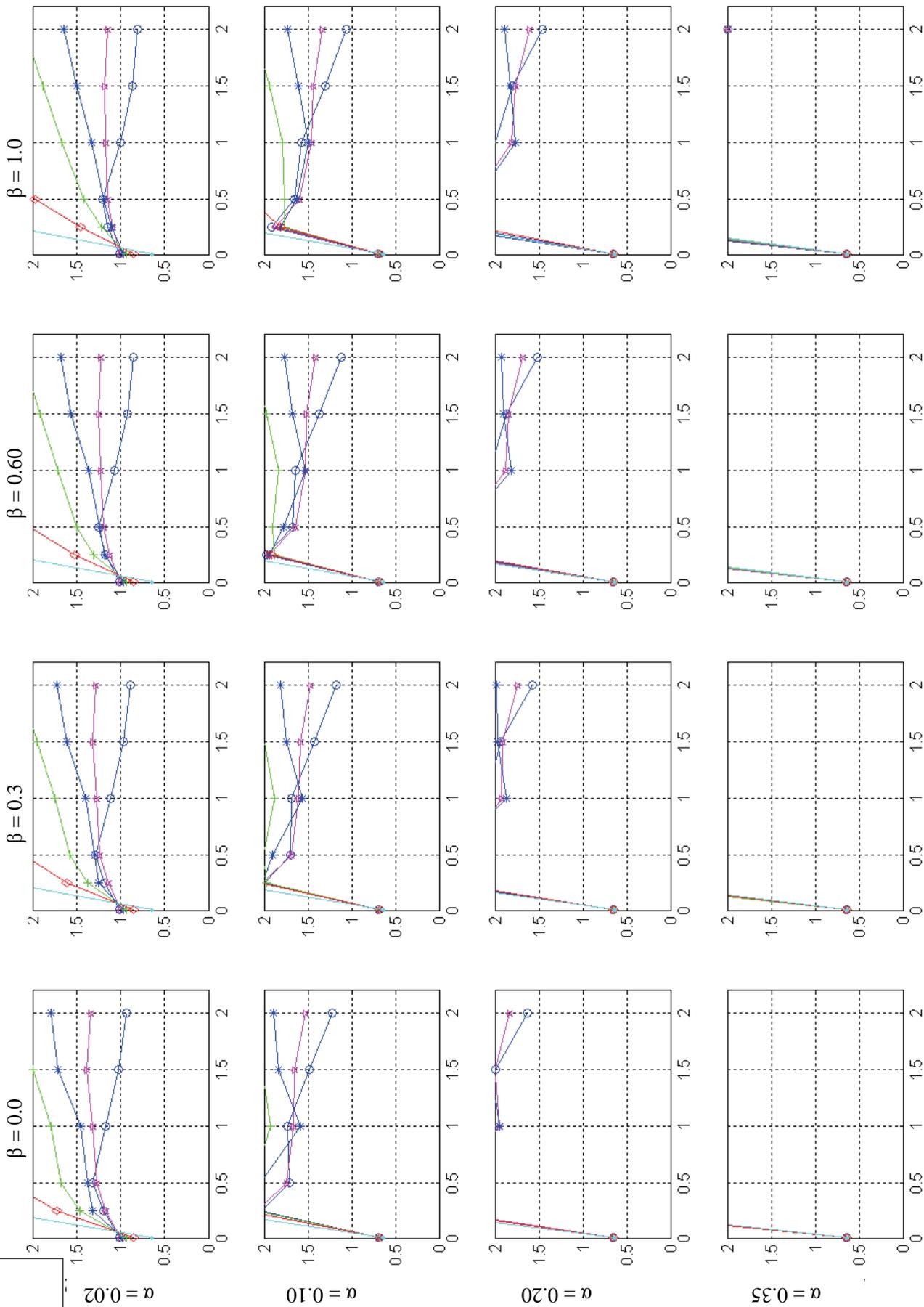
### Relative Performance (0.3\* Disp.Ratio + 0.7\* Accel.Ratio)



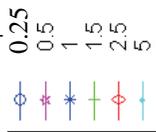
$\Psi$  ( $\eta_{EPS} = 0.10; \Psi = \eta_{SCS} / \eta_{EPS}$ )



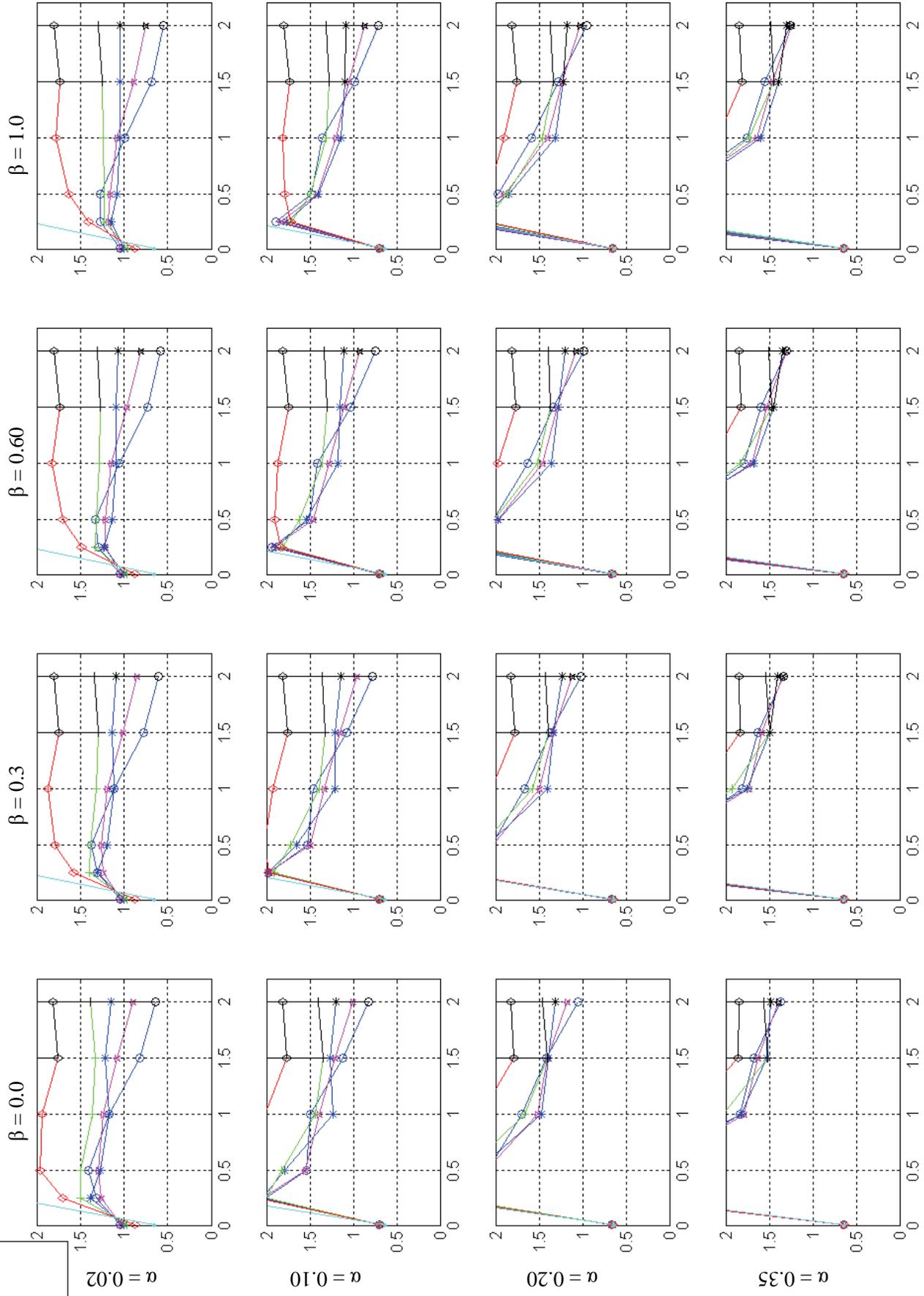
**Relative Performance (0.3\* Disp.Ratio + 0.7\* Accel.Ratio)**



$\Psi$  ( $\eta_{\text{EPS}} = 0.20; \psi = \eta_{\text{SCS}} / \eta_{\text{EPS}}$ )



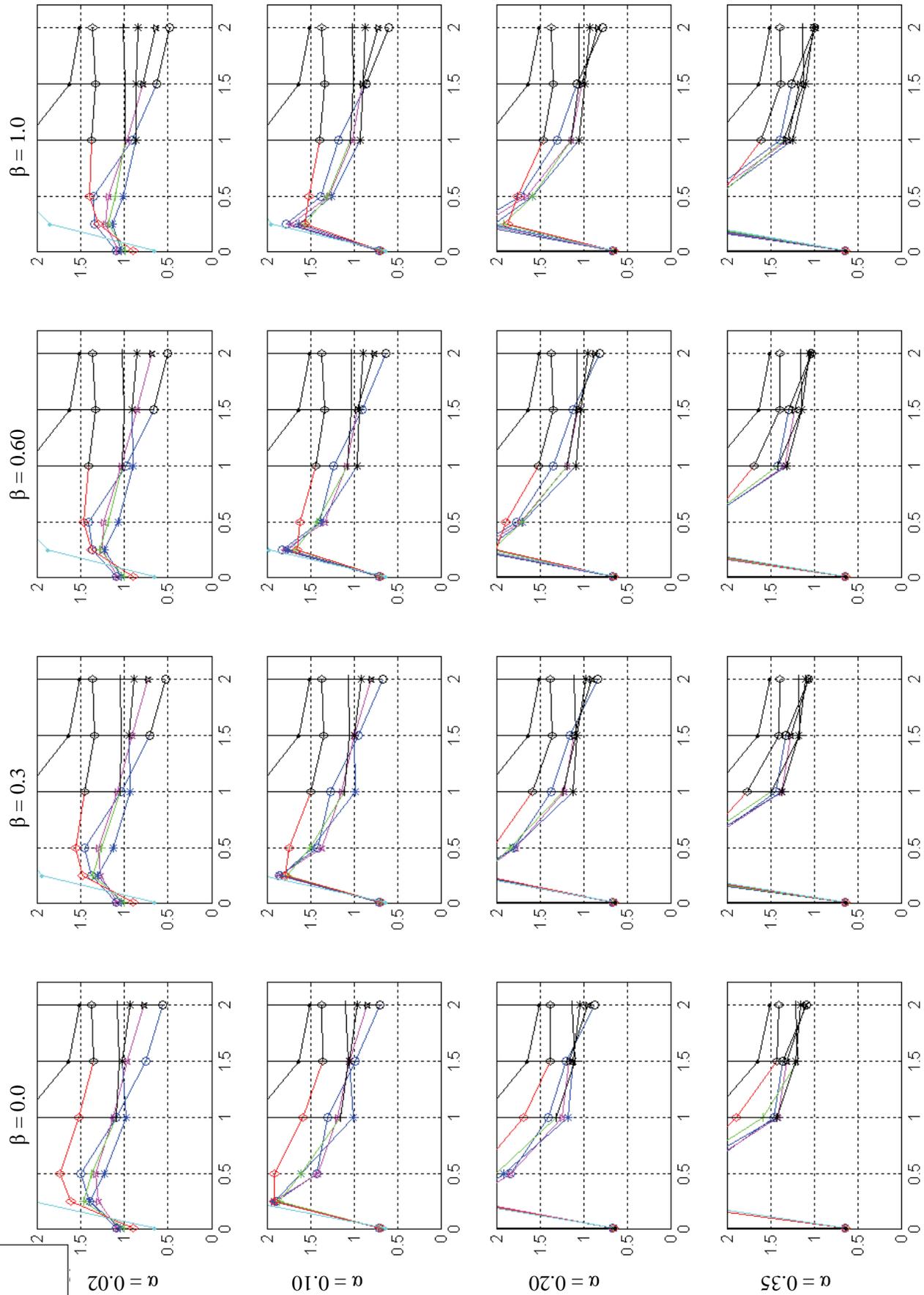
### Relative Performance (0.3\* Disp.Ratio + 0.7\* Accel.Ratio)



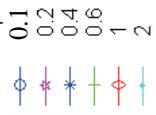
$\Psi$  ( $\eta_{\text{EPS}} = 0.30; \psi = \eta_{\text{SCS}} / \eta_{\text{EPS}}$ )

- 0.17
- 0.33
- 0.67
- 1
- 1.67
- 3.33

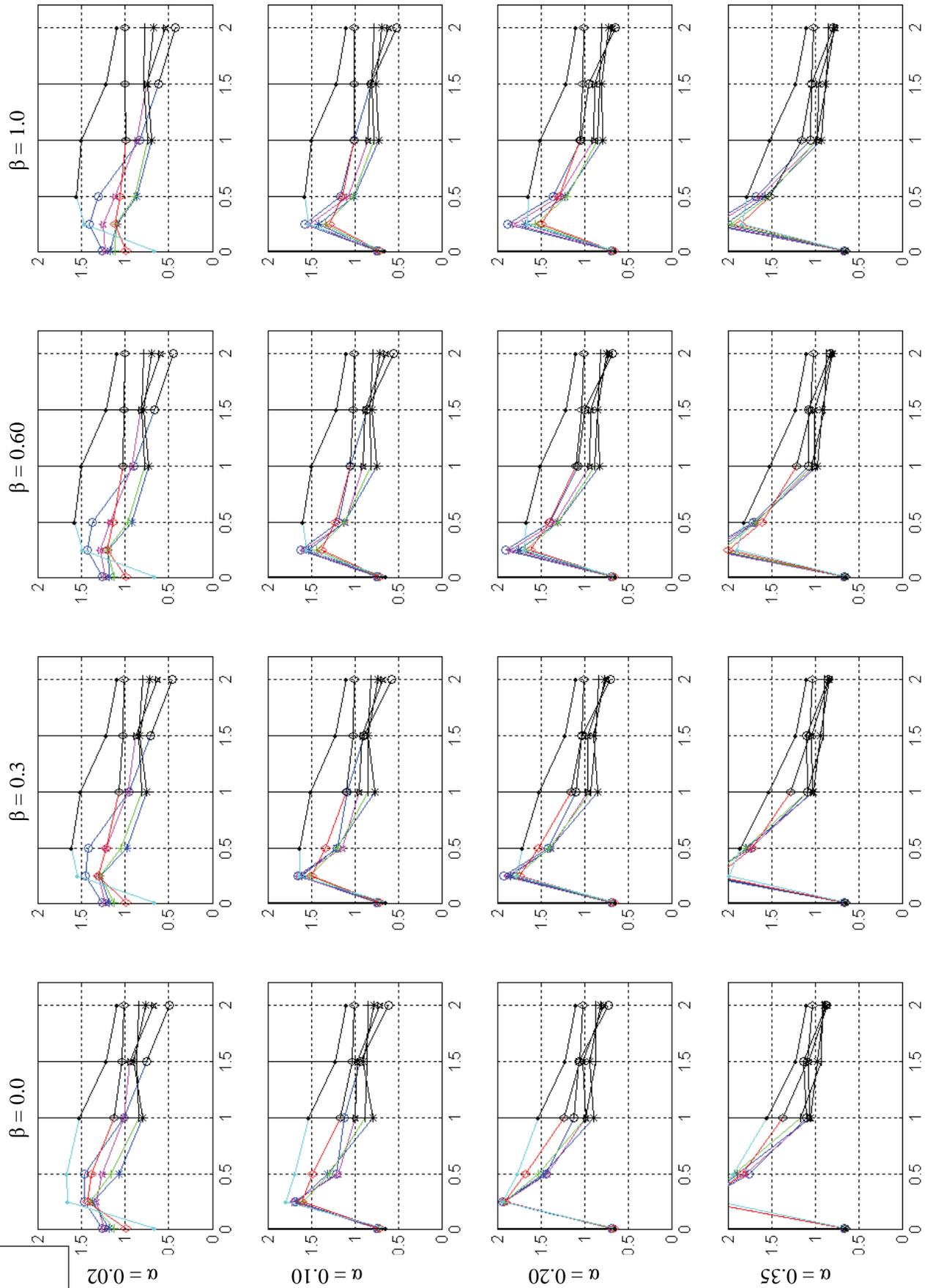
### Relative Performance (0.3\* Disp.Ratio + 0.7\* Accel.Ratio)



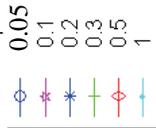
$\Psi$  ( $\eta_{EPS} = 0.50$ ;  $\Psi = \eta_{SCS} / \eta_{EPS}$ )



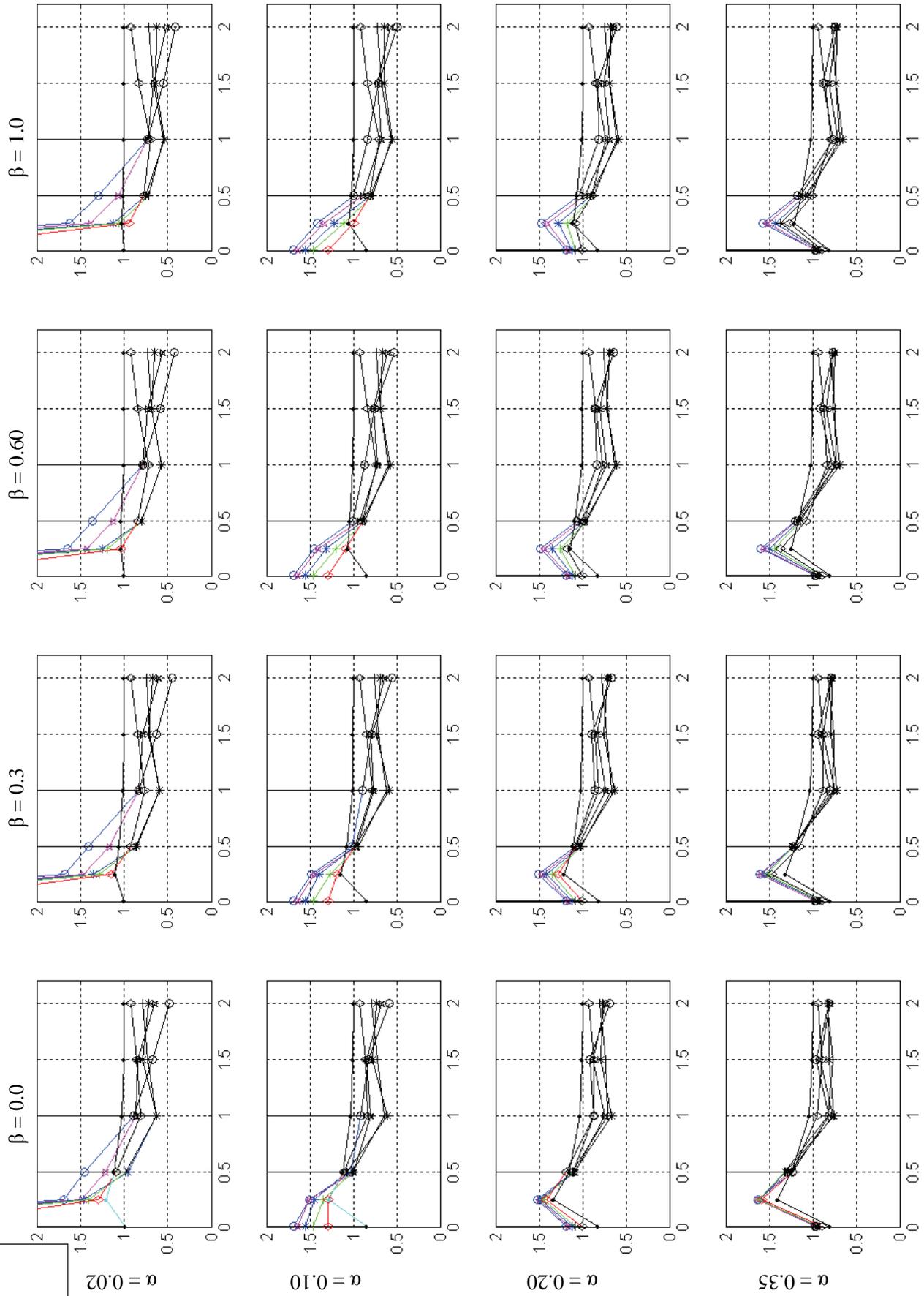
### Relative Performance (0.3\* Disp.Ratio + 0.7\* Accel.Ratio)



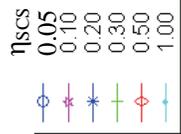
$\Psi$  ( $\eta_{\text{EPS}} = 1.00; \Psi = \eta_{\text{SCS}} / \eta_{\text{EPS}}$ )



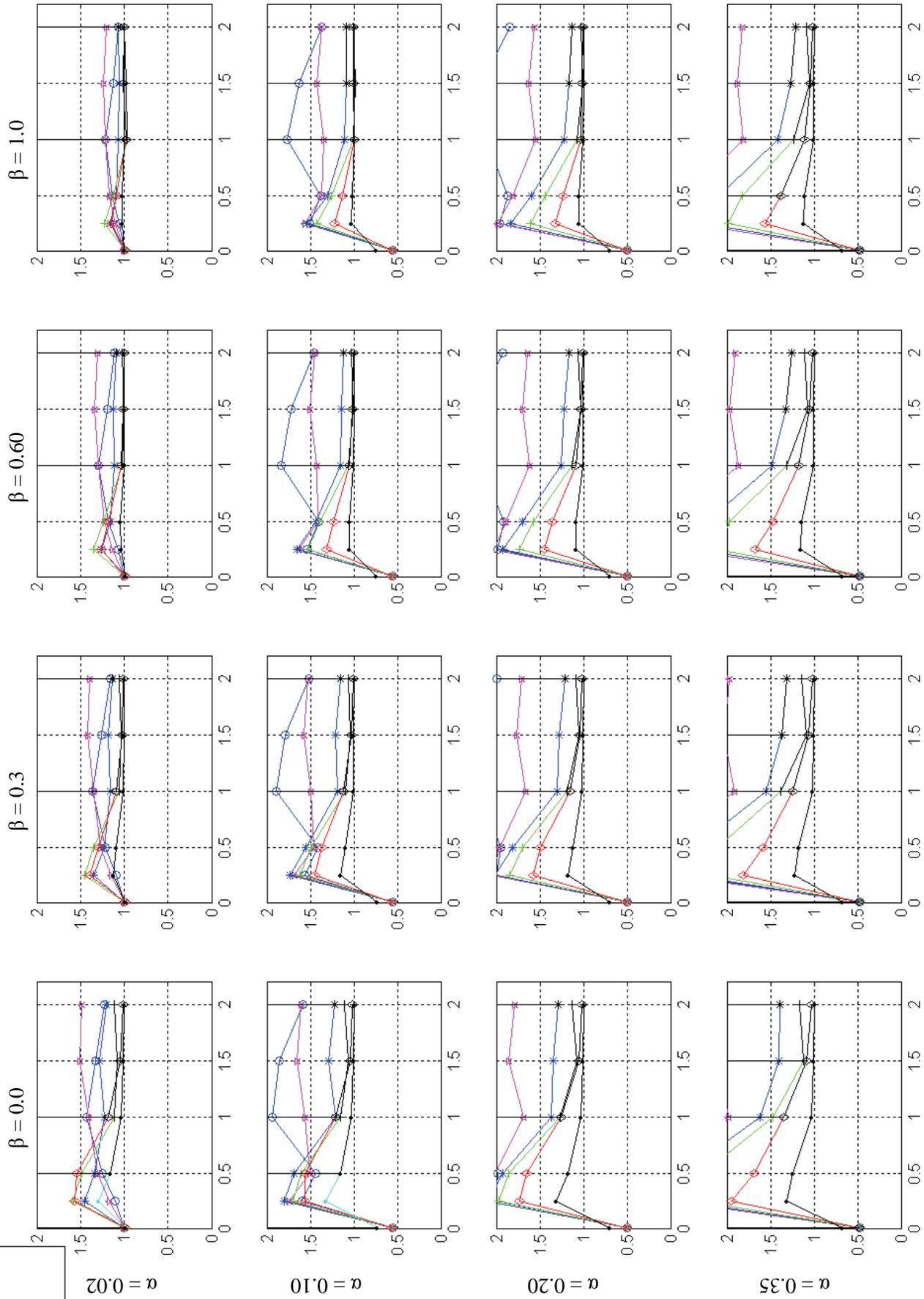
**Relative Performance (0.3\* Disp.Ratio + 0.7\* Accel.Ratio)**



$(\Psi = \eta_{\text{SCS}} / \eta_{\text{EPS}} = 1)$



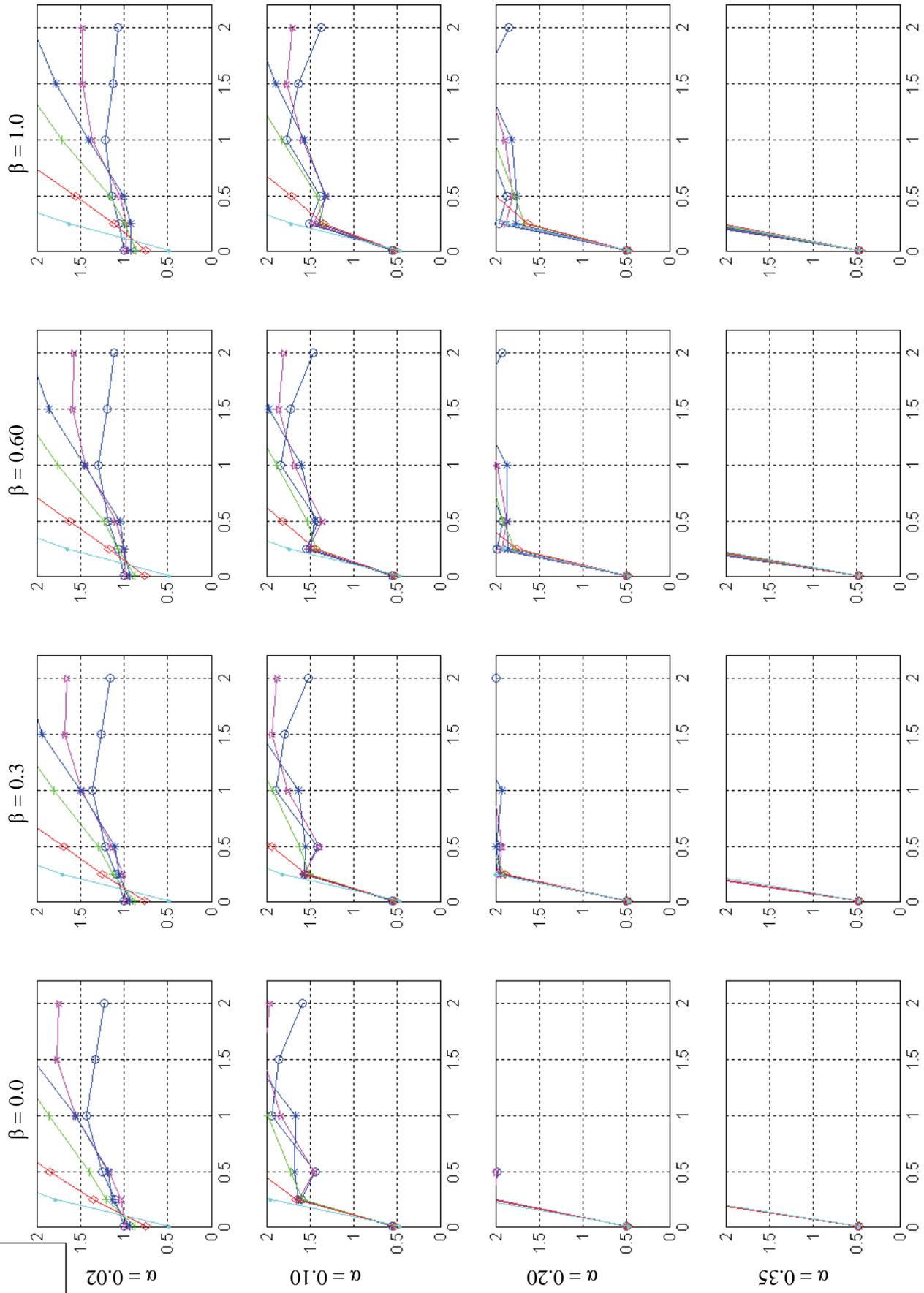
### Relative Performance (0.5\* Disp.Ratio + 0.5\* Accel.Ratio)



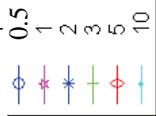
$\Psi$  ( $\eta_{EPS} = 0.05$ ;  $\psi = \eta_{SCS} / \eta_{EPS}$ )



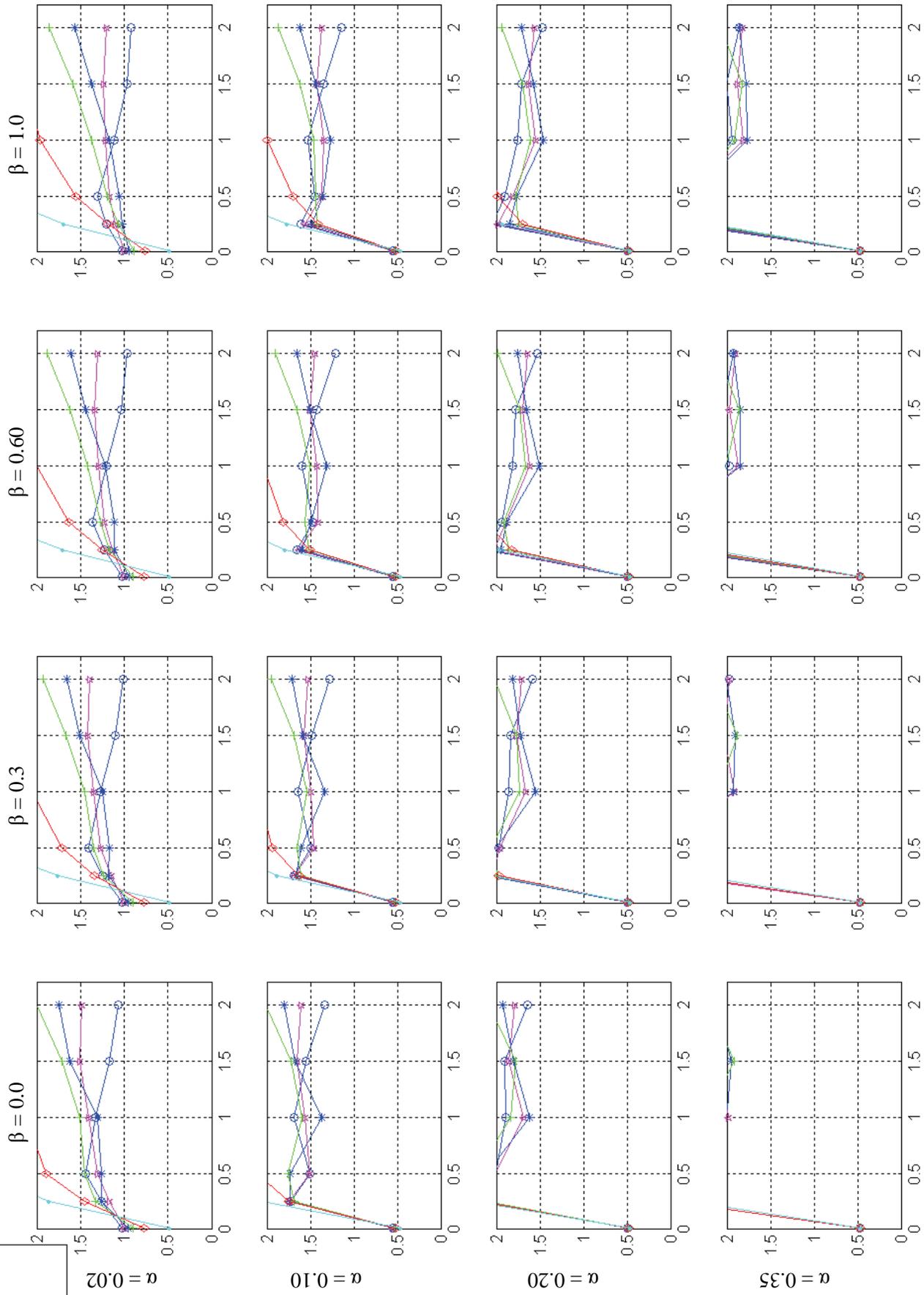
### Relative Performance (0.5\* Disp.Ratio + 0.5\* Accel.Ratio)



$\Psi$  ( $\eta_{EPS} = 0.10; \psi = \eta_{ISCS} / \eta_{EPS}$ )



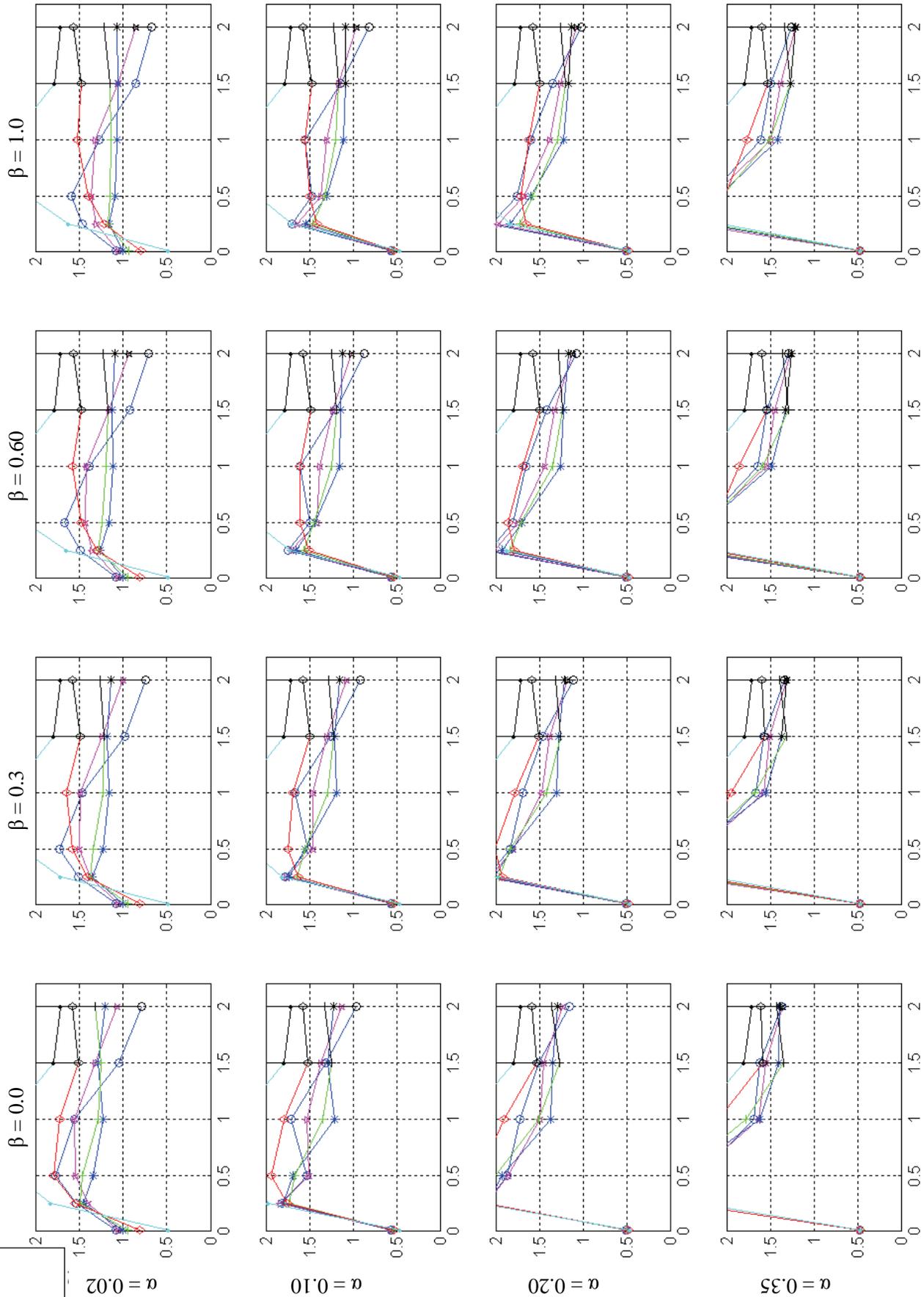
### Relative Performance (0.5\* Disp.Ratio + 0.5\* Accel.Ratio)



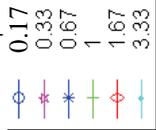
$\Psi$  ( $\eta_{\text{EPS}} = 0.20$ ;  $\Psi = \eta_{\text{SCS}} / \eta_{\text{EPS}}$ )

- 0.25
- 0.5
- 1
- 1.5
- 2.5
- 5

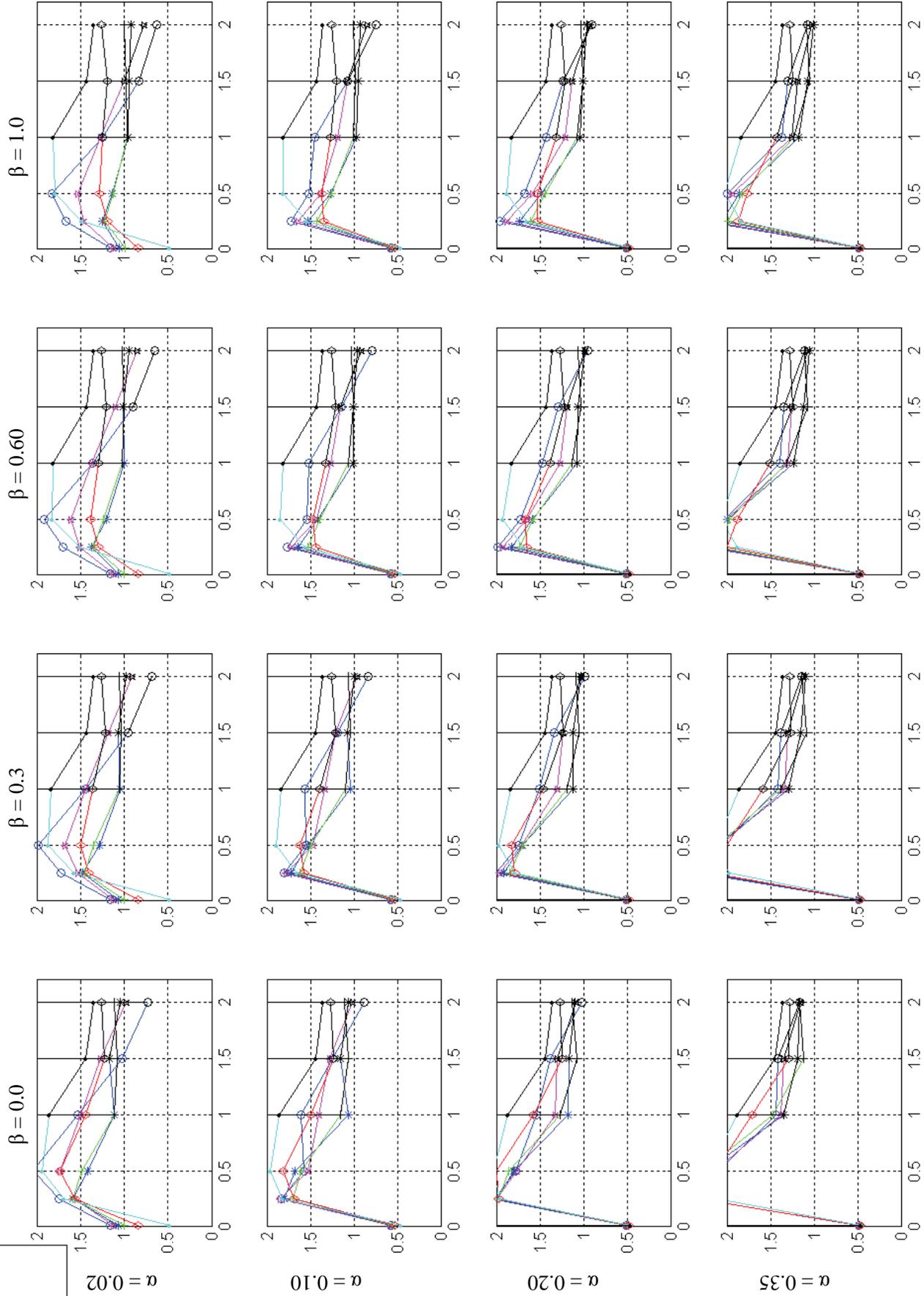
### Relative Performance (0.5\* Disp.Ratio + 0.5\* Accel.Ratio)



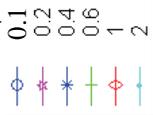
$\Psi$  ( $\eta_{EPS} = 0.30$ ;  $\psi = \eta_{SCS} / \eta_{EPS}$ )



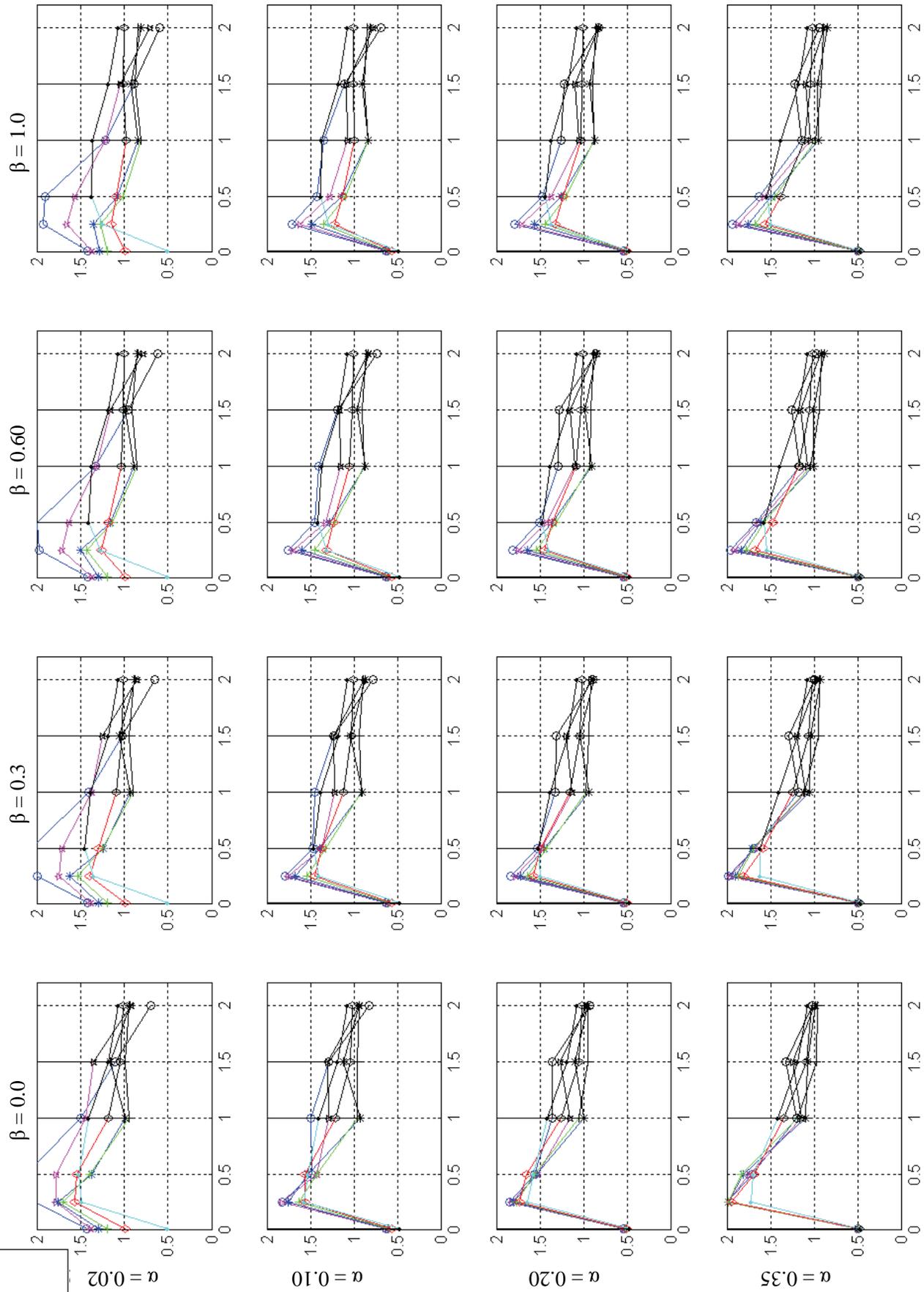
**Relative Performance (0.5\* Disp.Ratio + 0.5\* Accel.Ratio)**



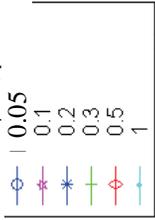
$\Psi$  ( $\eta_{\text{EPS}} = 0.50; \psi = \eta_{\text{SCS}} / \eta_{\text{EPS}}$ )



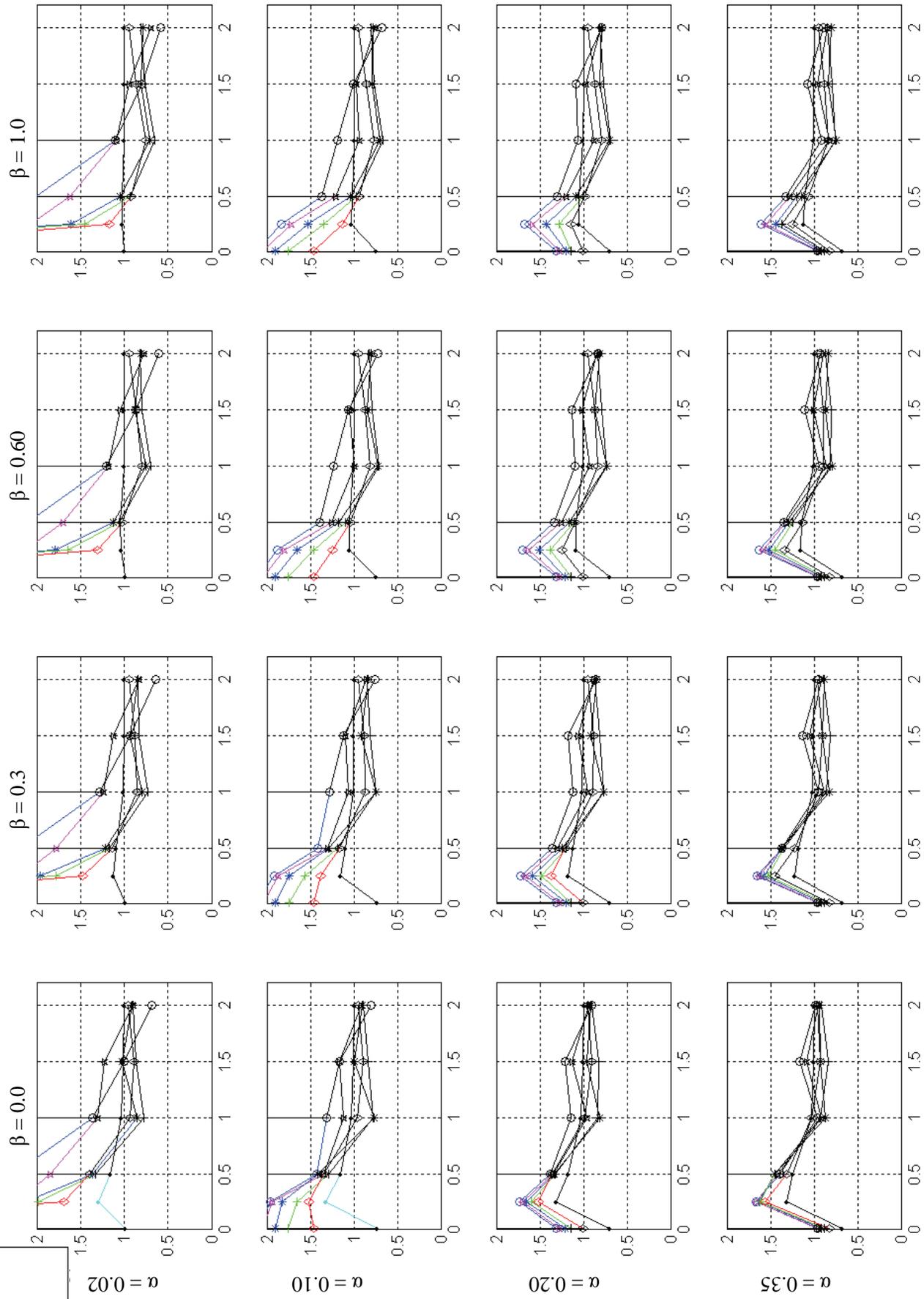
### Relative Performance (0.5\* Disp.Ratio + 0.5\* Accel.Ratio)



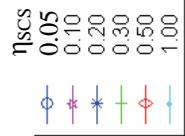
$\Psi$  ( $\eta_{\text{EPS}} = 1.00; \Psi = \eta_{\text{SCS}} / \eta_{\text{EPS}}$ )



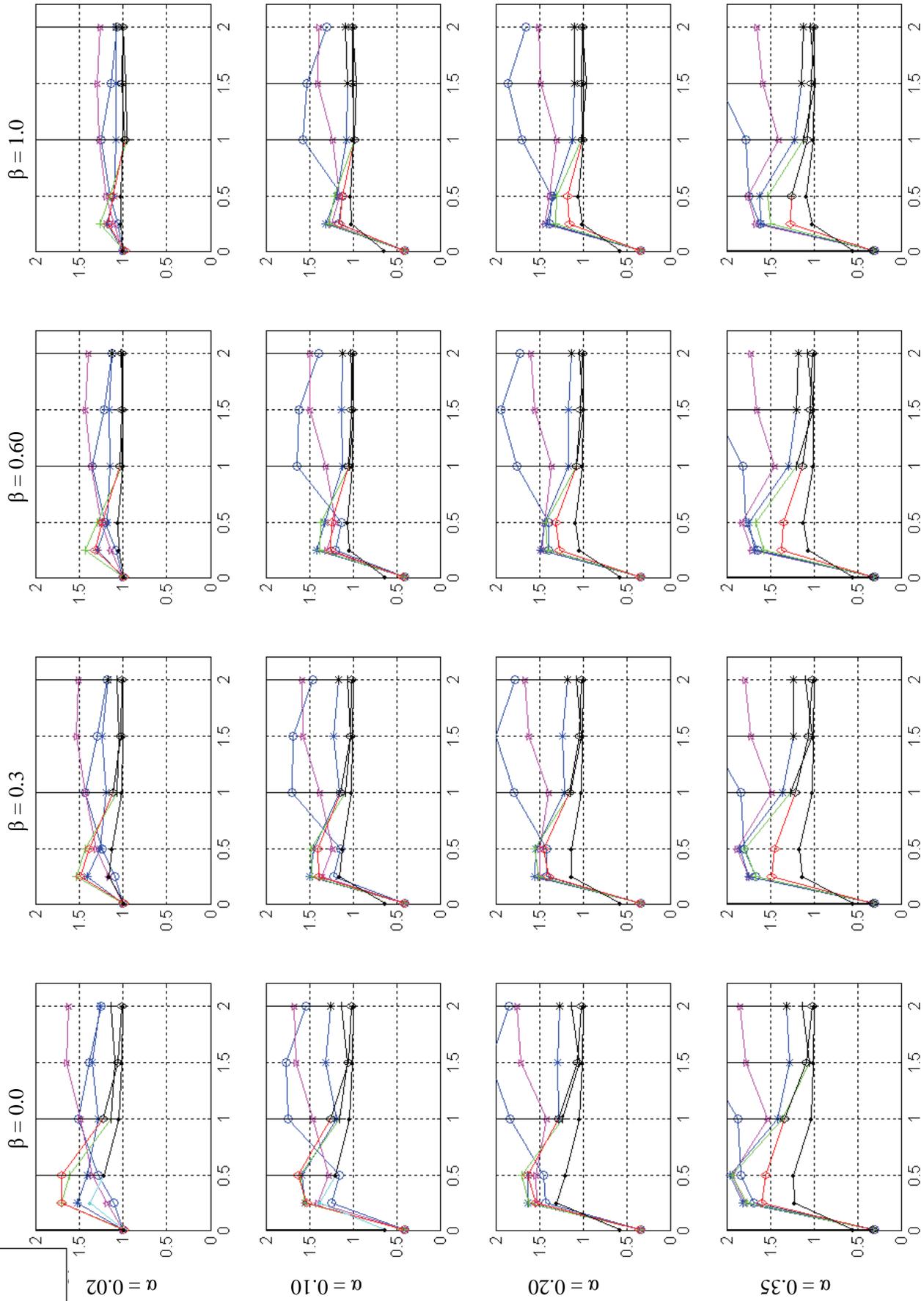
### Relative Performance (0.5\* Disp.Ratio + 0.5\* Accel.Ratio)



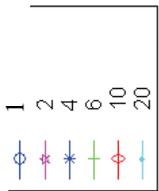
$\eta_{\text{SCS}} / \eta_{\text{EPS}} = 1$



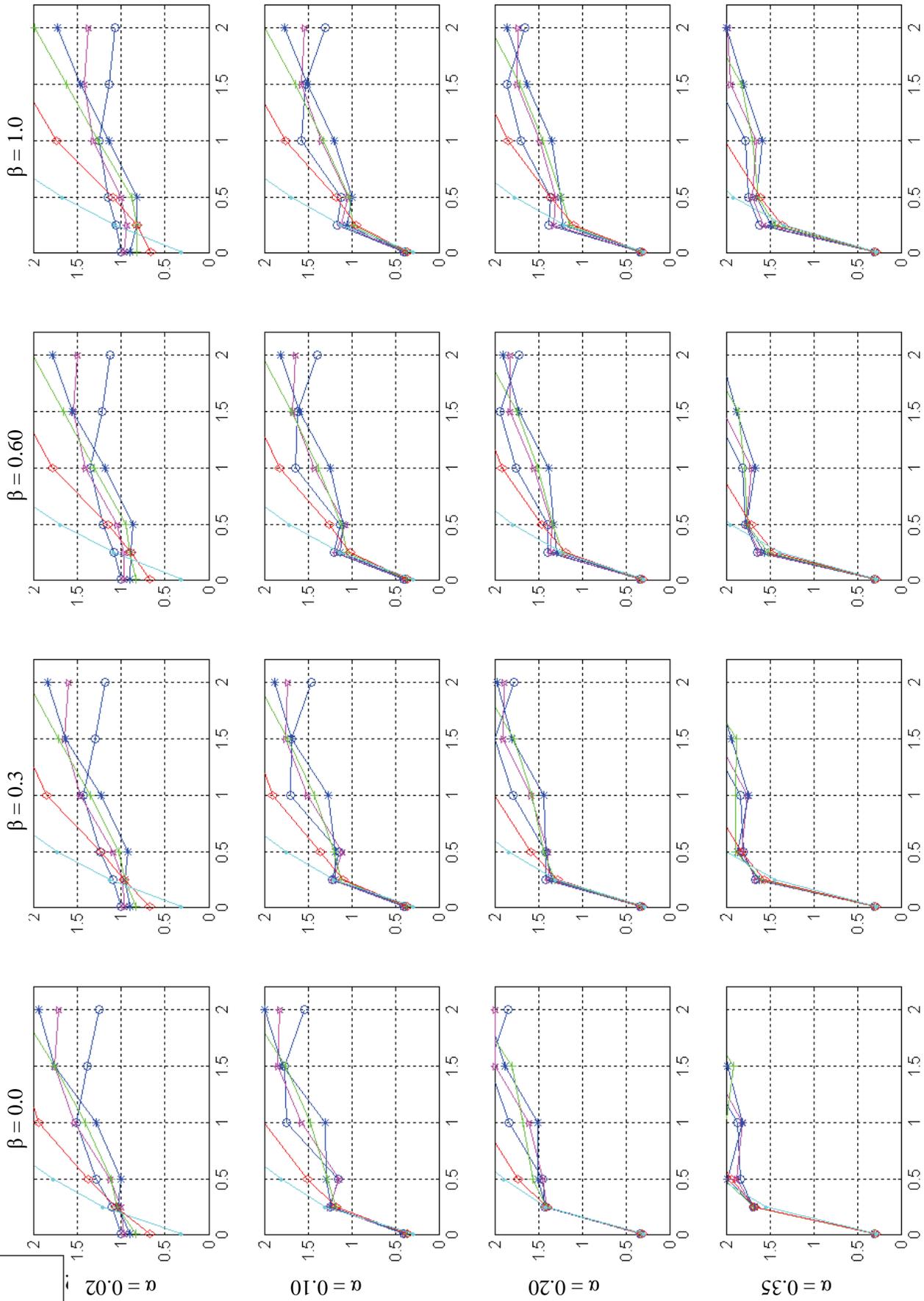
### Relative Performance (0.7\* Disp.Ratio + 0.3\* Accel.Ratio)



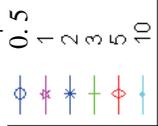
$\Psi$  ( $\eta_{EPS} = 0.05$ ;  $\Psi = \eta_{SCS} / \eta_{EPS}$ )



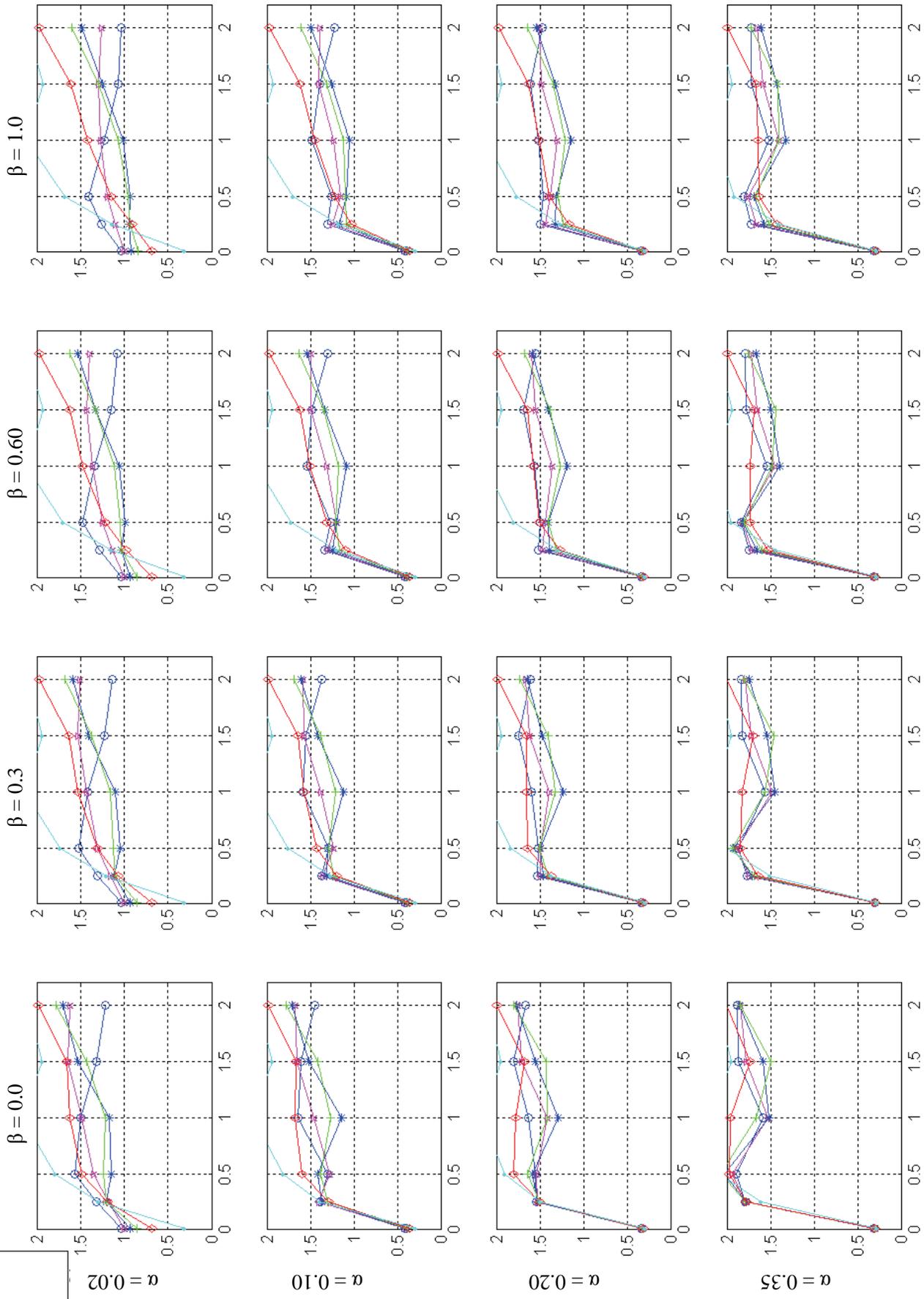
### Relative Performance (0.7\* Disp.Ratio + 0.3\* Accel.Ratio)



$\Psi$  ( $\eta_{EPS} = 0.10; \Psi = \eta_{ISCS} / \eta_{EPS}$ )



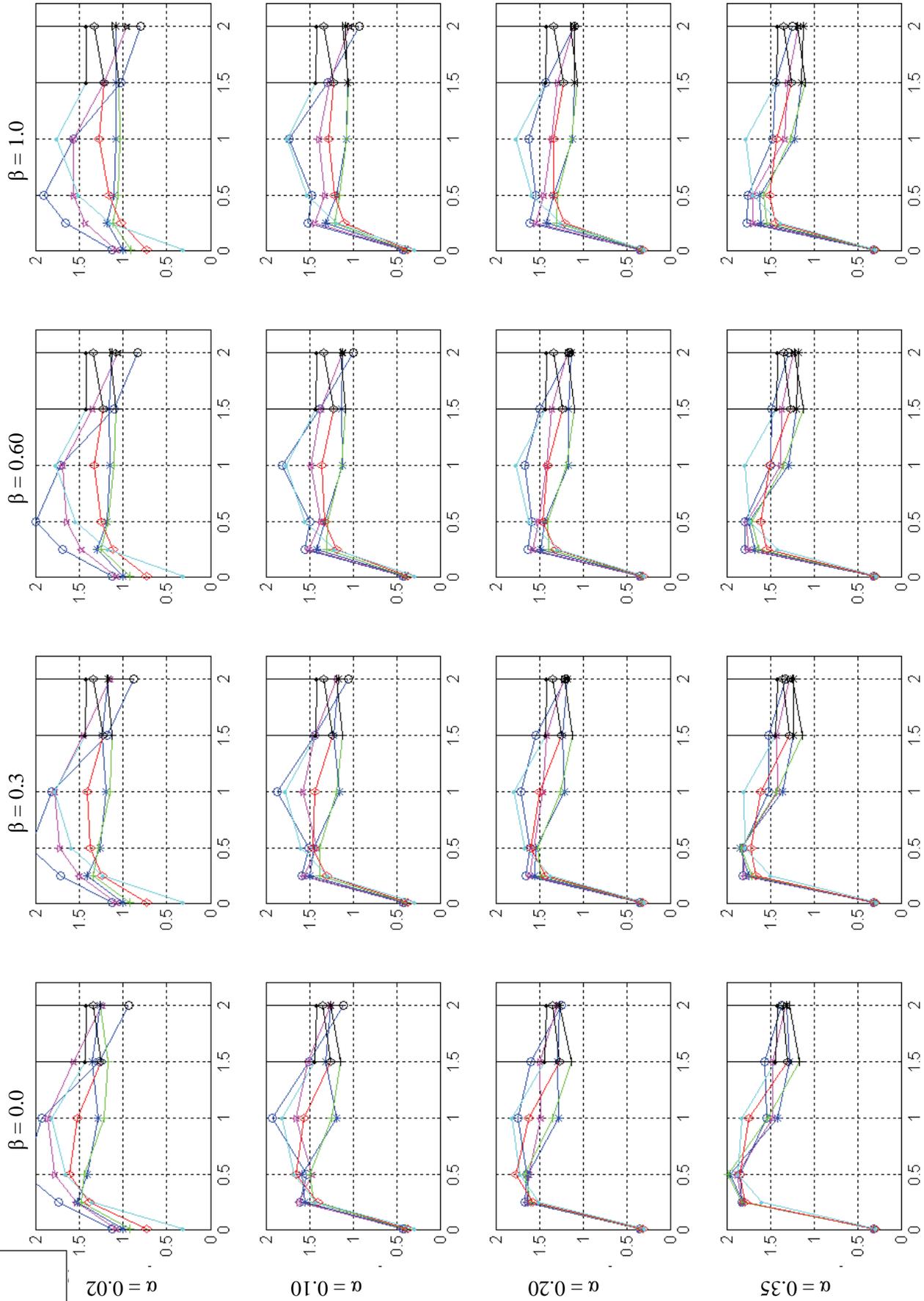
**Relative Performance (0.7\* Disp.Ratio + 0.3\* Accel.Ratio)**



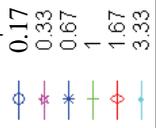
$\Psi$  ( $\eta_{EPS} = 0.20$ ;  $\Psi = \eta_{SCS} / \eta_{EPS}$ )

- 0.25
- \* 0.5
- ▲ 1
- ◇ 1.5
- ◇ 2.5
- ◇ 5

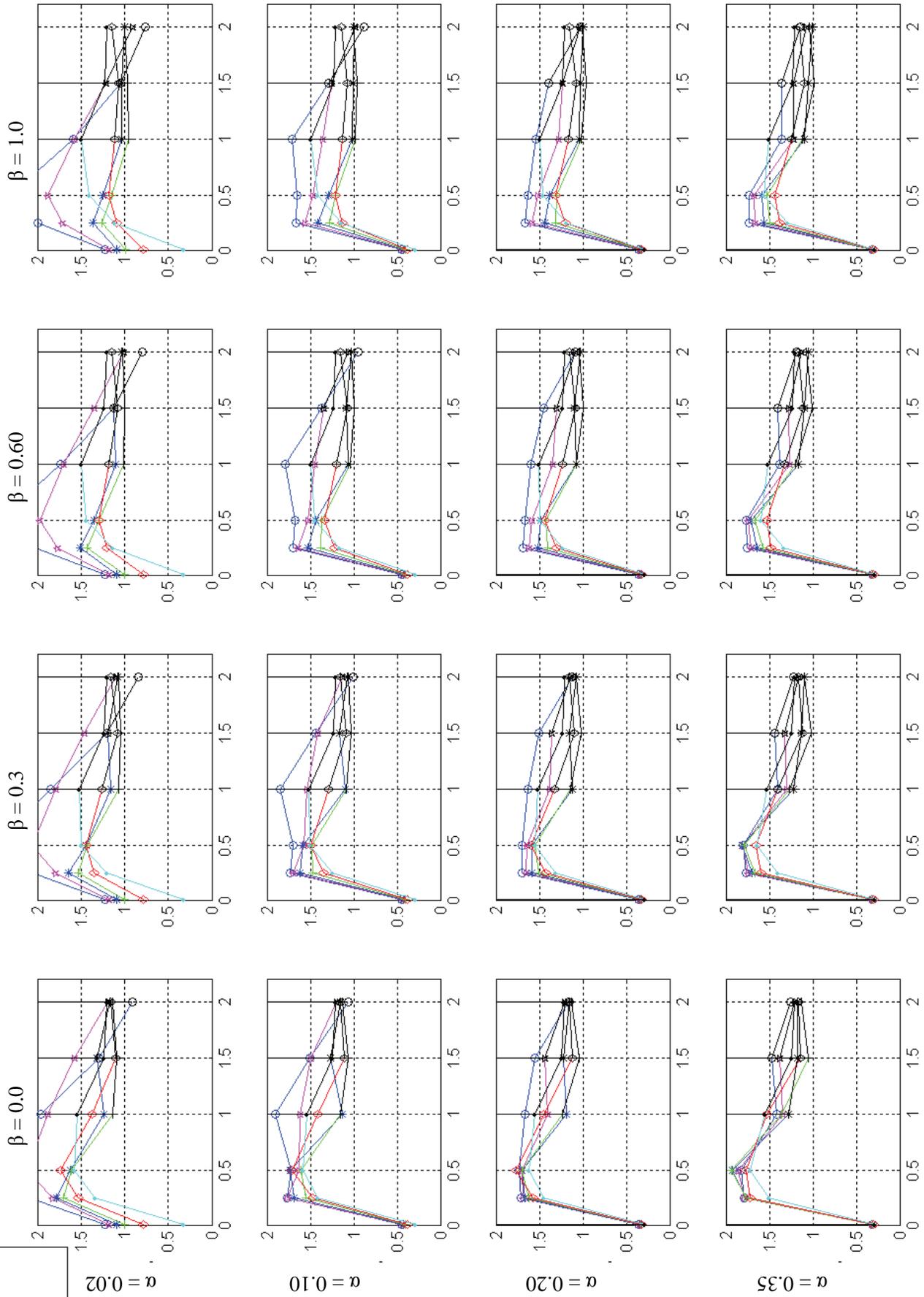
### Relative Performance (0.7\* Disp.Ratio + 0.3\* Accel.Ratio)



$\Psi$  ( $\eta_{EPS} = 0.30; \psi = \eta_{SCS} / \eta_{EPS}$ )



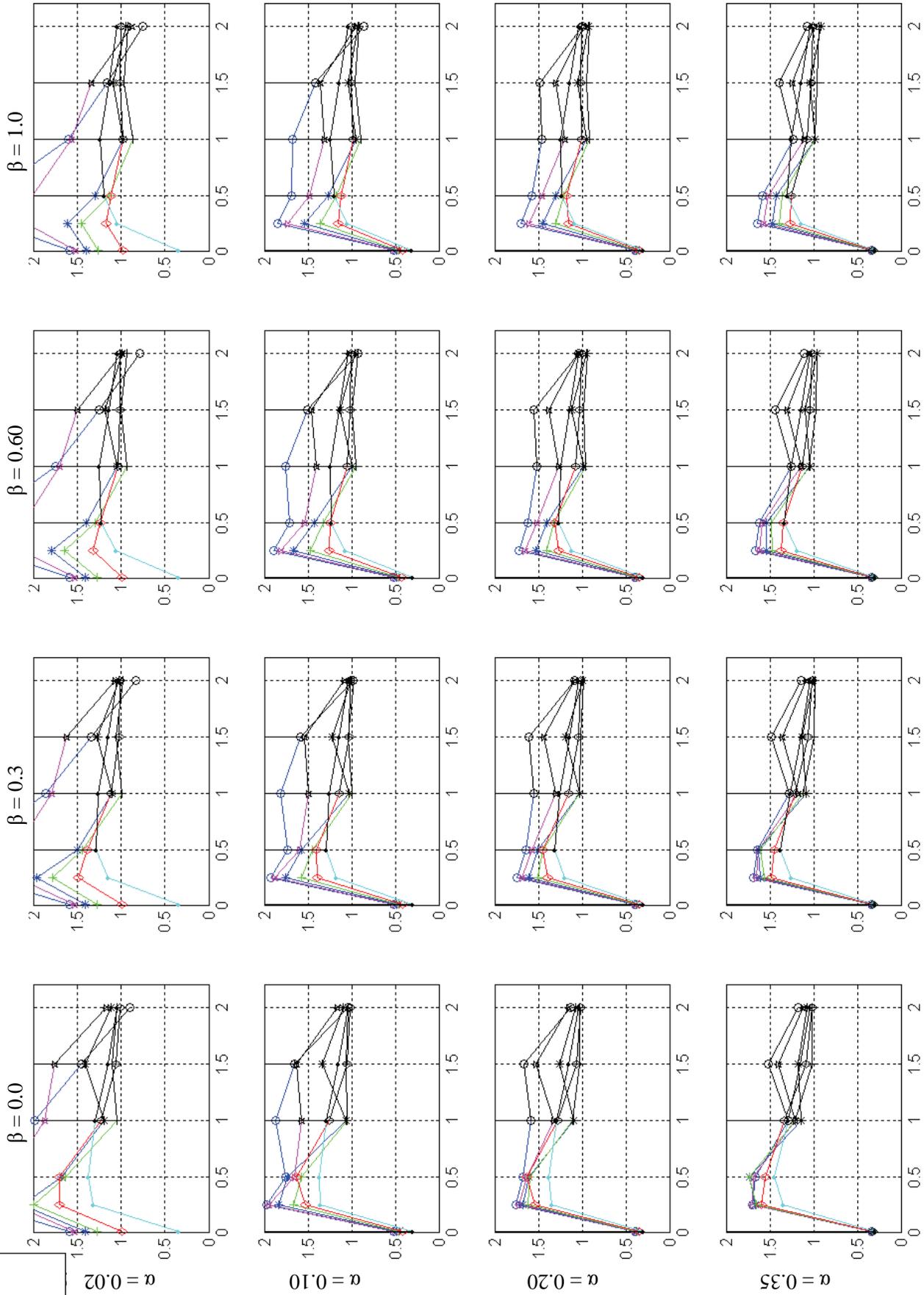
**Relative Performance (0.7\* Disp.Ratio + 0.3\* Accel.Ratio)**



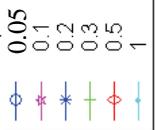
$\Psi$  ( $\eta_{\text{EPS}} = 0.50; \Psi = \eta_{\text{SCS}} / \eta_{\text{EPS}}$ )



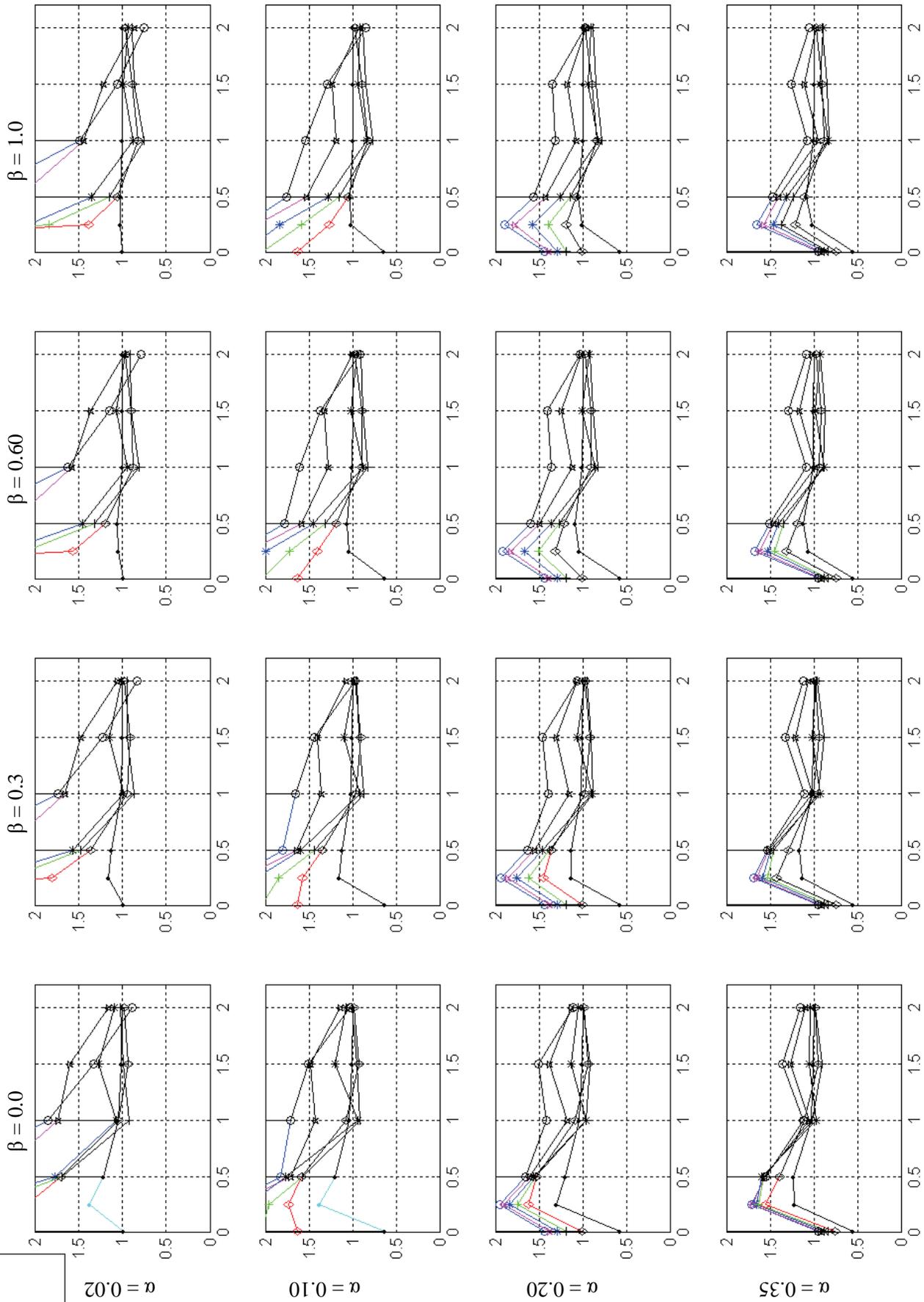
### Relative Performance (0.7\* Disp.Ratio + 0.3\* Accel.Ratio)



$\Psi$  ( $\eta_{EPS} = 1.00$ ;  $\Psi = \eta_{SCS} / \eta_{EPS}$ )

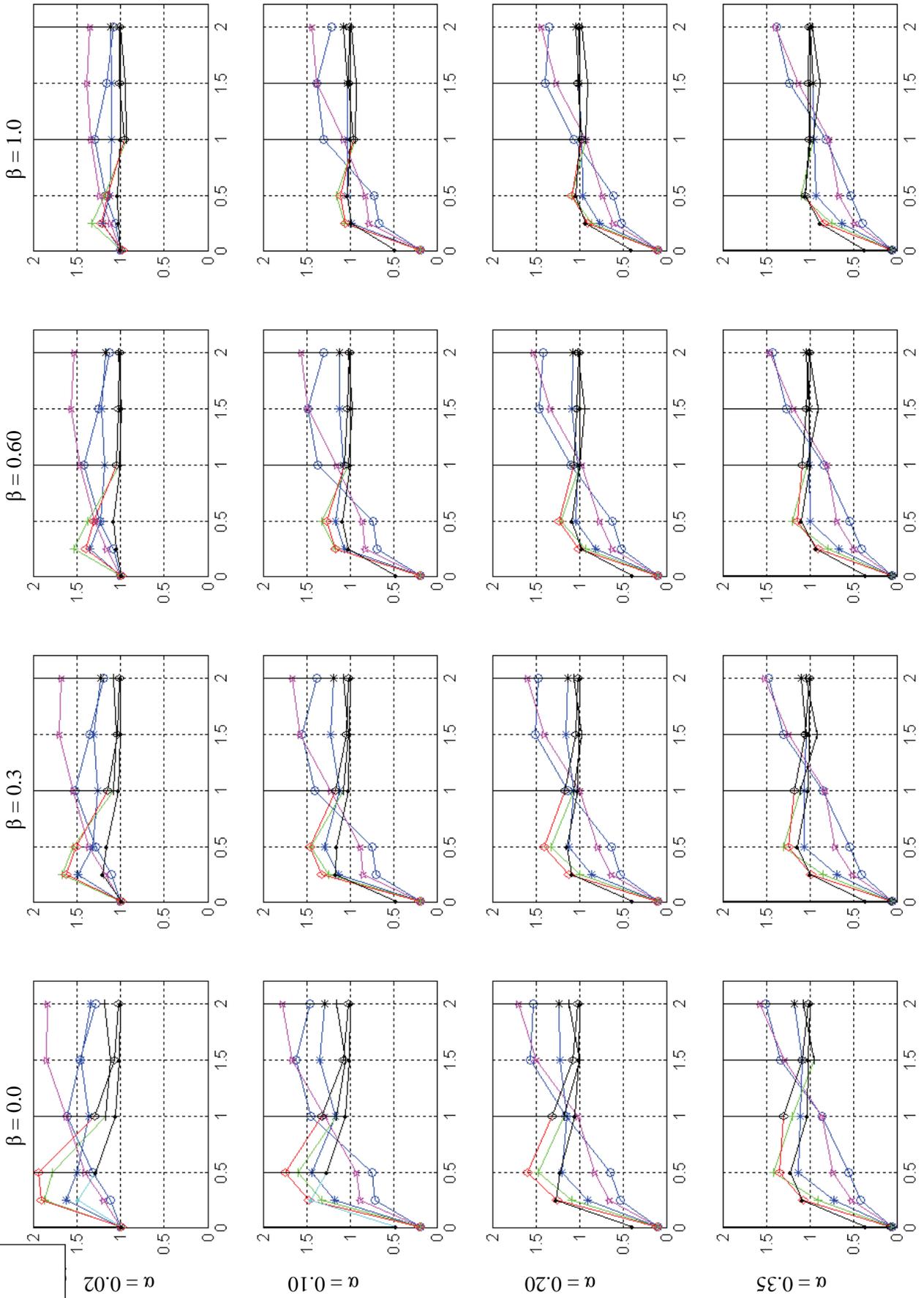
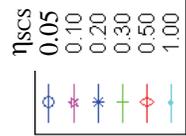


**Relative Performance (0.7\* Disp.Ratio + 0.3\* Accel.Ratio)**



**Relative Performance (1\* Disp.Ratio + 0\* Accel.Ratio)**

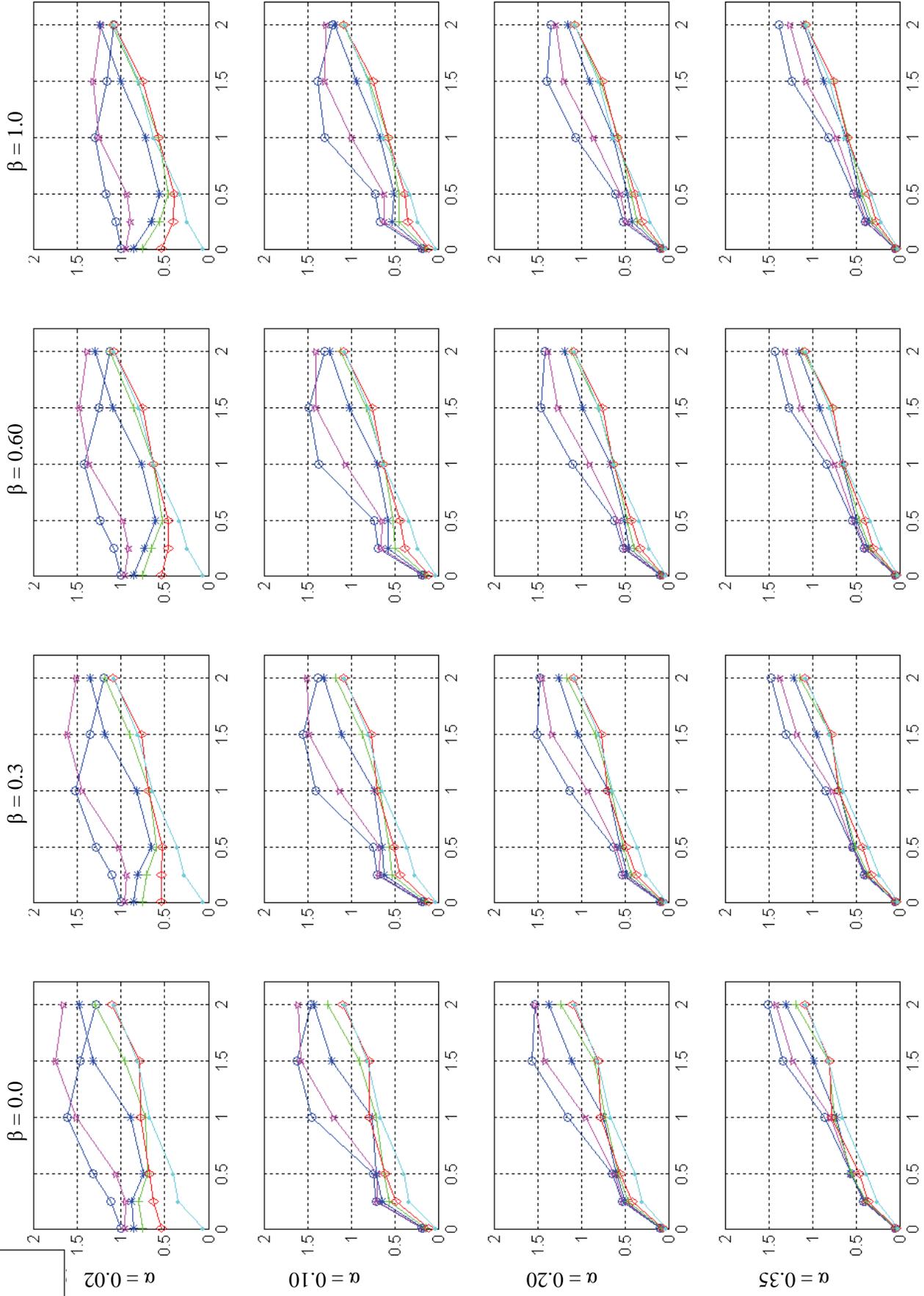
$(\Psi = \eta_{SCS} / \eta_{EPS} = 1)$



$\Psi$  ( $\eta_{\text{EPS}} = 0.05; \Psi = \eta_{\text{SCS}} / \eta_{\text{EPS}}$ )

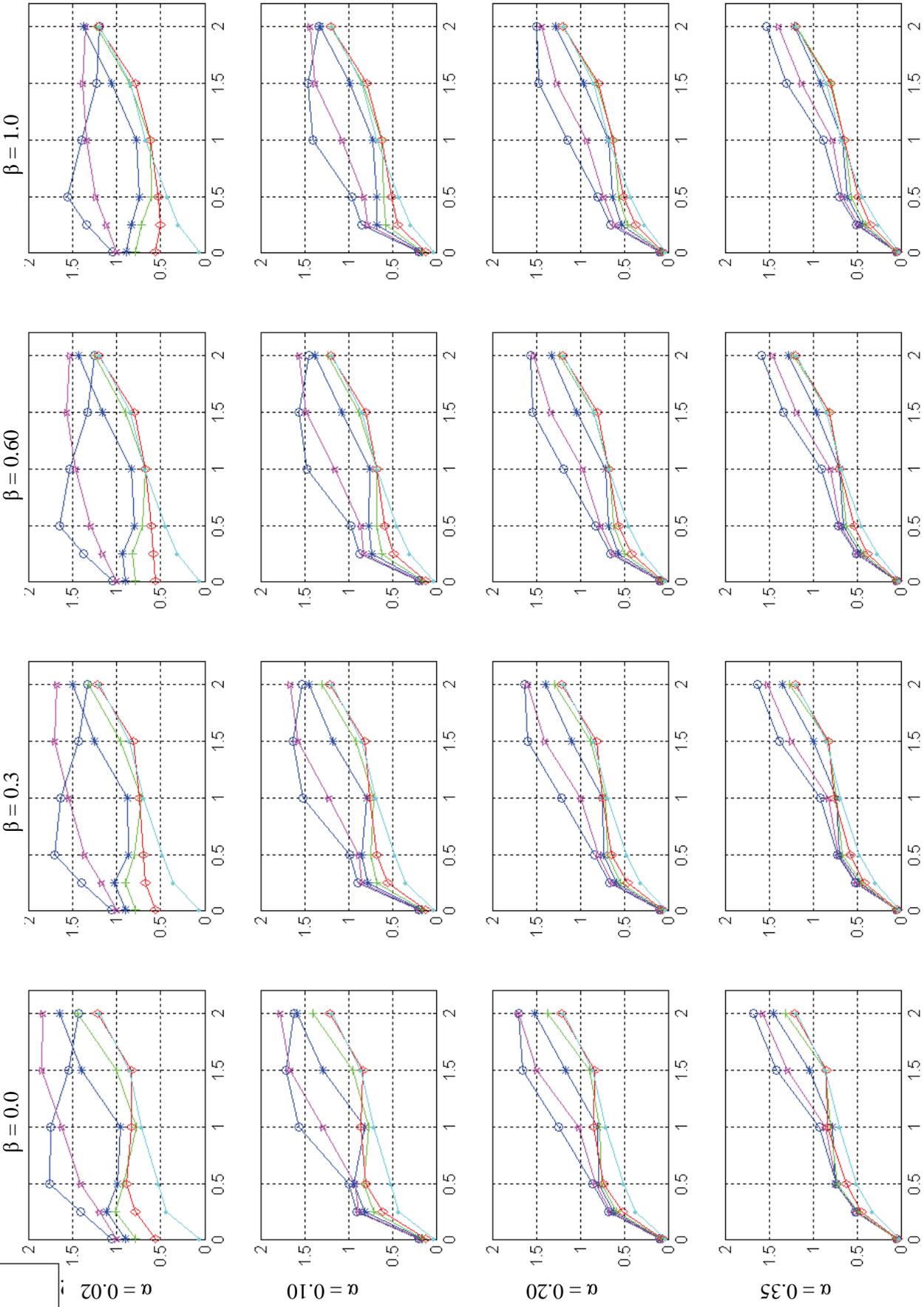
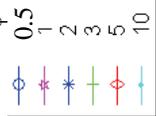


**Relative Performance (1\* Disp.Ratio + 0\* Accel.Ratio)**

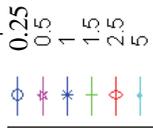


# Relative Performance (1\* Disp.Ratio + 0\* Accel.Ratio)

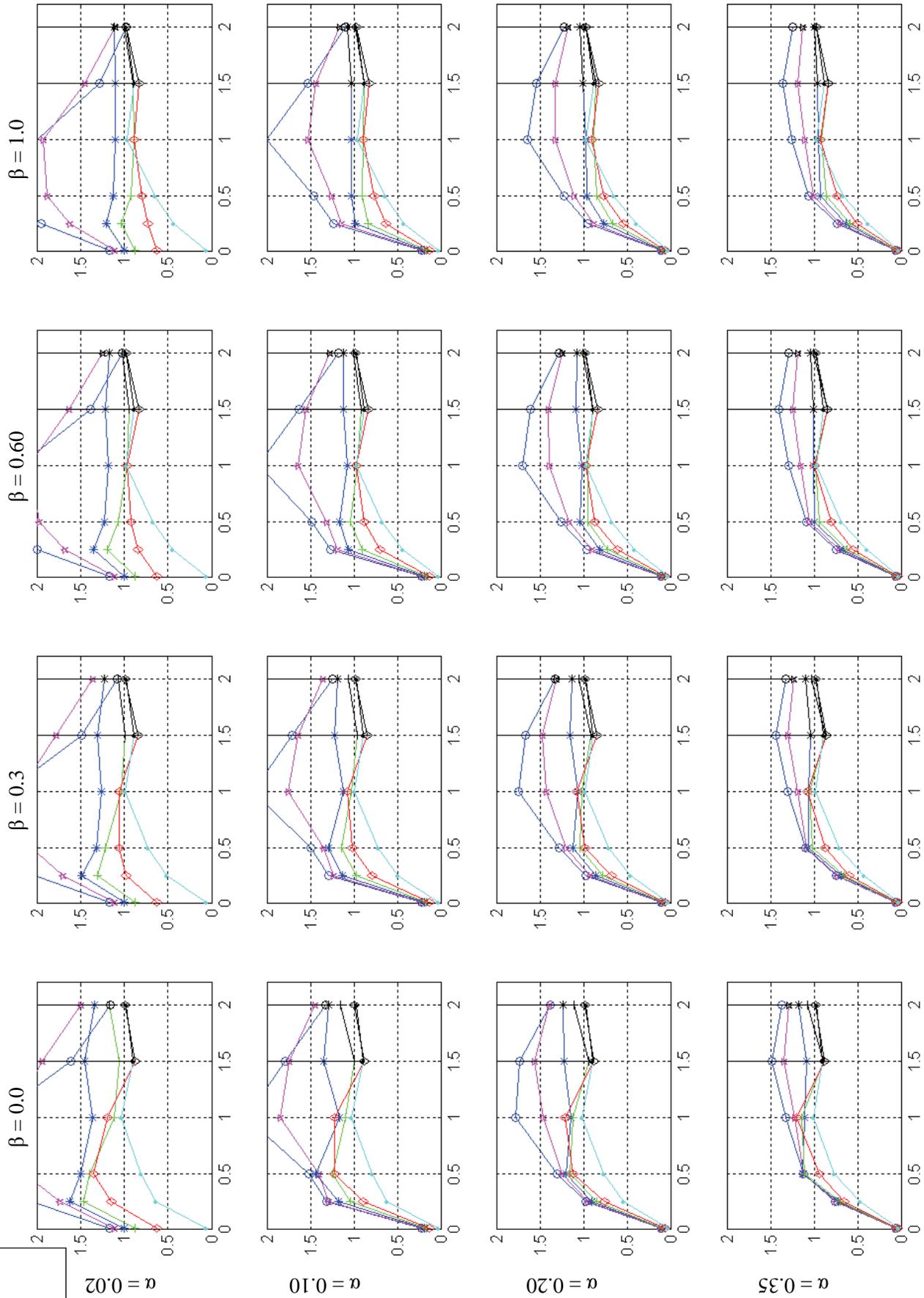
$\Psi$  ( $\eta_{EPS} = 0.10$ ;  $\Psi = \eta_{SCS} / \eta_{EPS}$ )



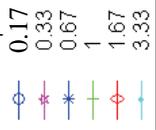
$\Psi$  ( $\eta_{EPS} = 0.20$ ;  $\Psi = \eta_{SCS} / \eta_{EPS}$ )



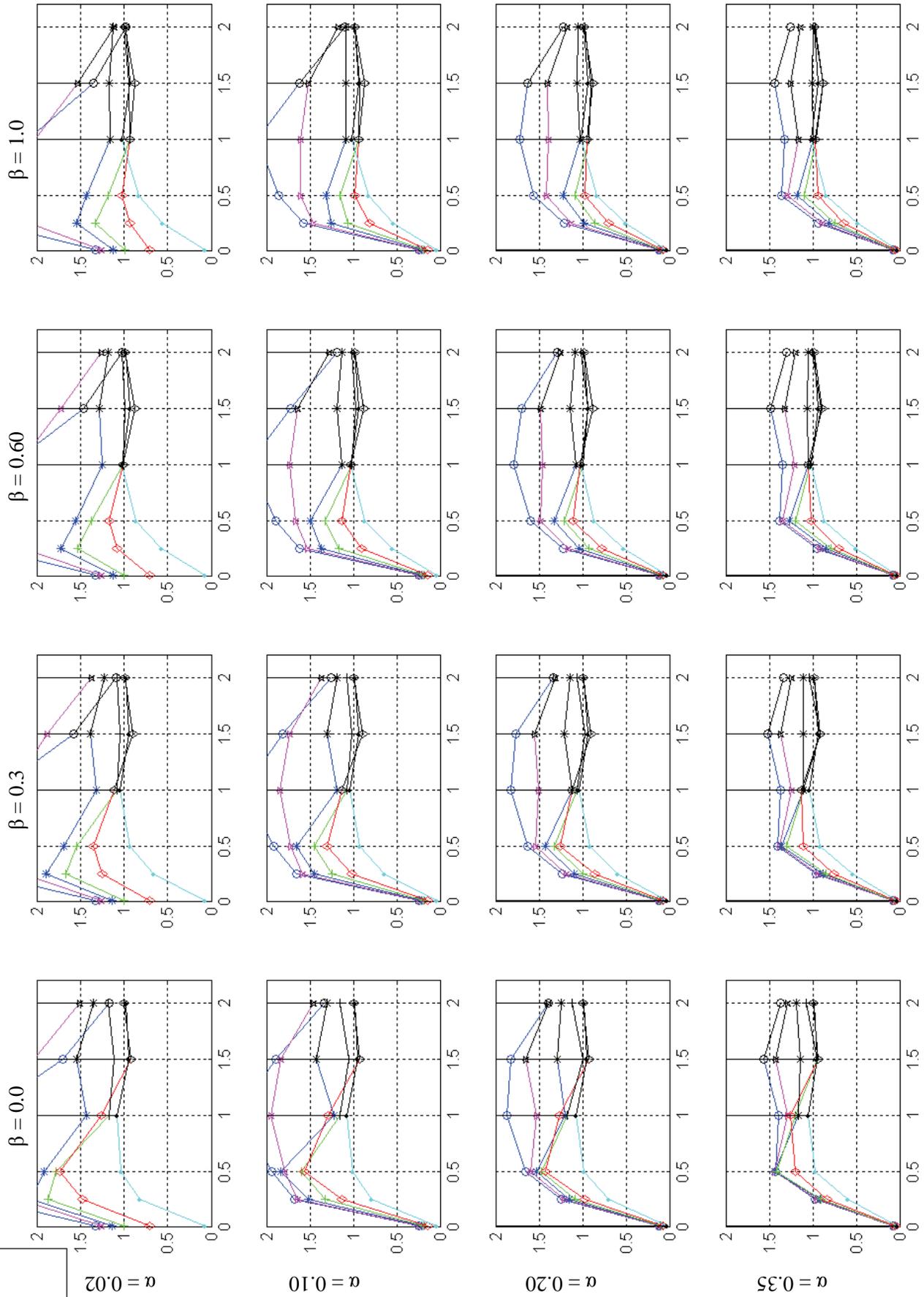
**Relative Performance (1\* Disp.Ratio + 0\* Accel.Ratio)**



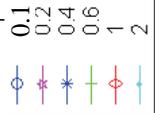
$\Psi$  ( $\eta_{EPS} = 0.30; \psi = \eta_{scs} / \eta_{EPS}$ )



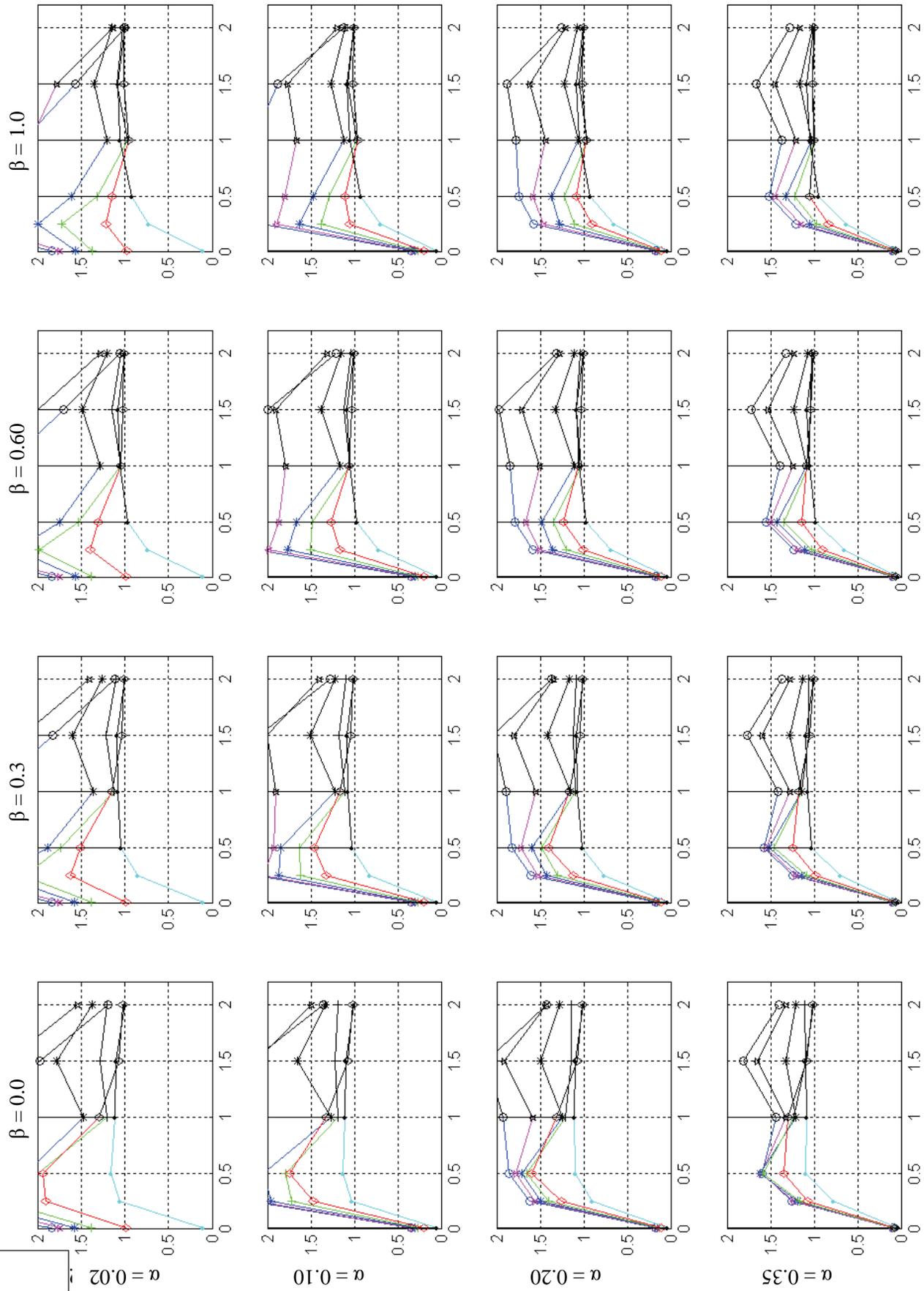
### Relative Performance (1\* Disp.Ratio + 0\* Accel.Ratio)



$\Psi$  ( $\eta_{EPS} = 0.50; \Psi = \eta_{SCS} / \eta_{EPS}$ )

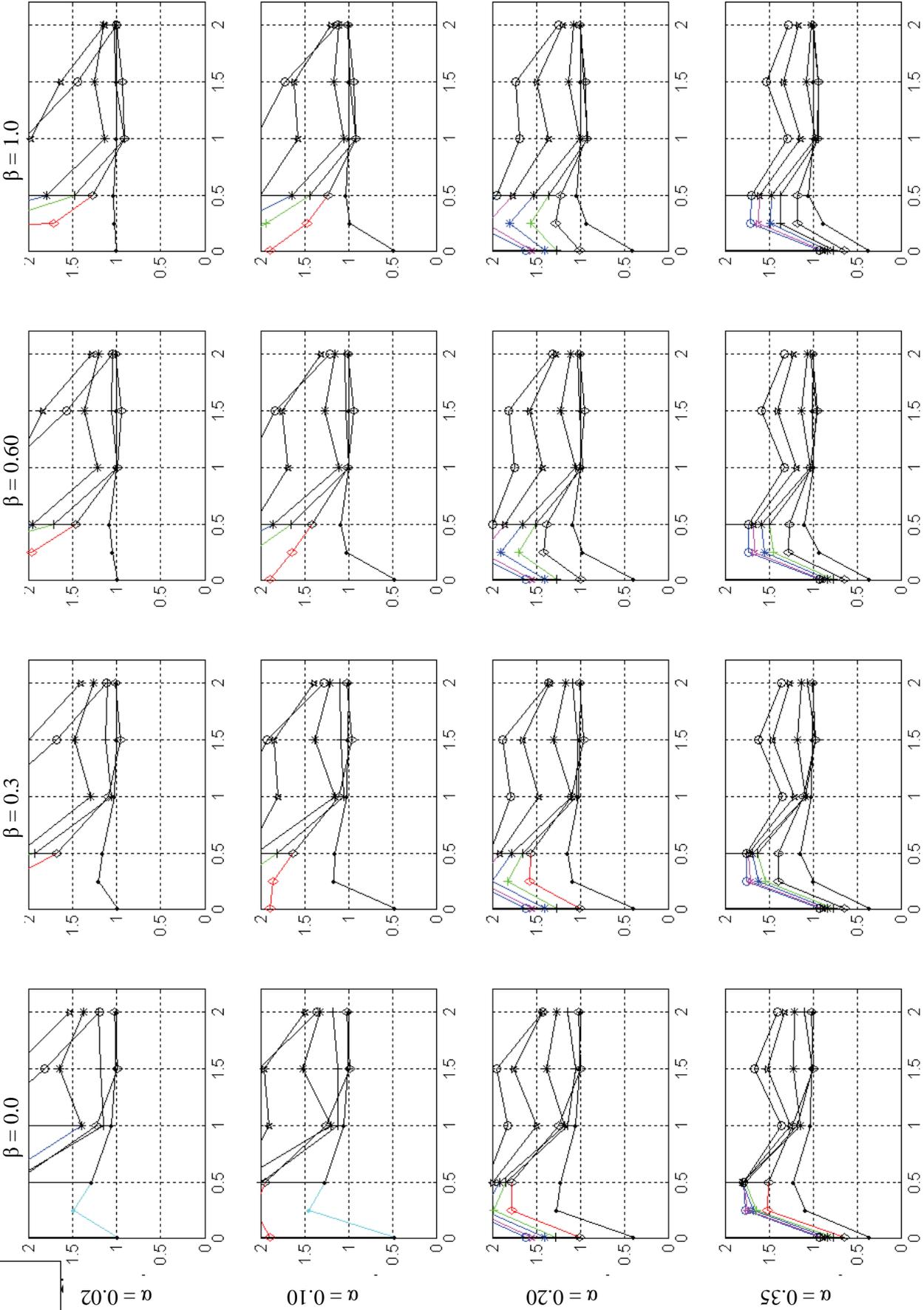
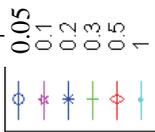


**Relative Performance (1\* Disp.Ratio + 0\* Accel.Ratio)**



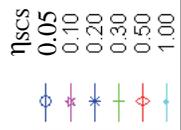
**Relative Performance (1\*Disp.Ratio + 0\*Accel.Ratio)**

$\Psi$  ( $\eta_{EPS} = 1.00; \Psi = \eta_{SCS} / \eta_{EPS}$ )

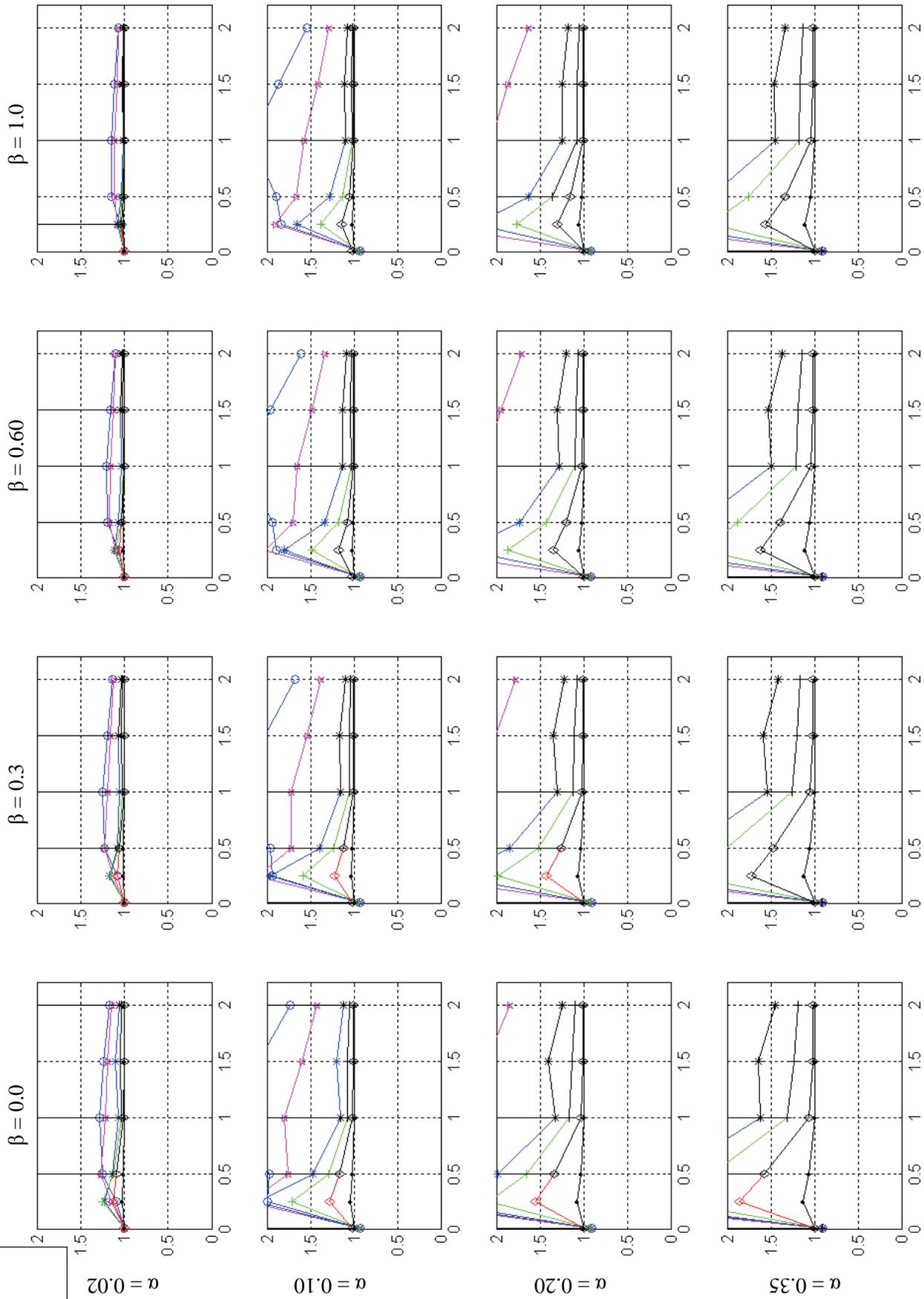


**Appendix B      Relative Performance Index (RPI) under  
MCEER ground motions having a 10%  
probability of exceedance in 50 years in  
California, U.S.**

$\eta_{\text{SCS}} \text{ ( } \Psi = \eta_{\text{SCS}} / \eta_{\text{EPS}} = 1 \text{ )}$



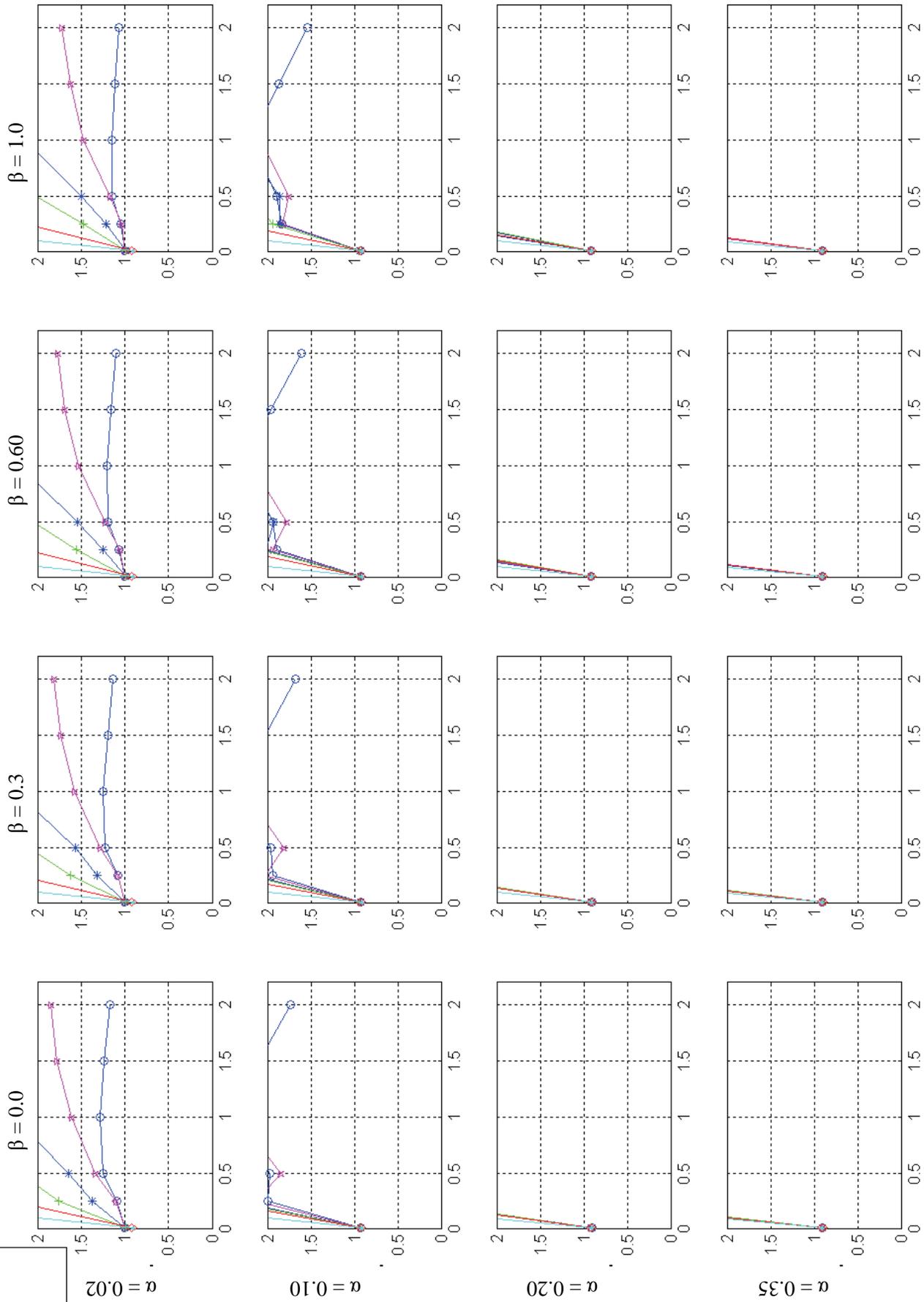
**Relative Performance (0\* Disp.Ratio + 1\* Accel.Ratio)**



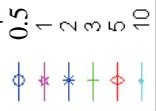
$\Psi$  ( $\eta_{EPS} = 0.05$ ;  $\Psi = \eta_{SCS} / \eta_{EPS}$ )



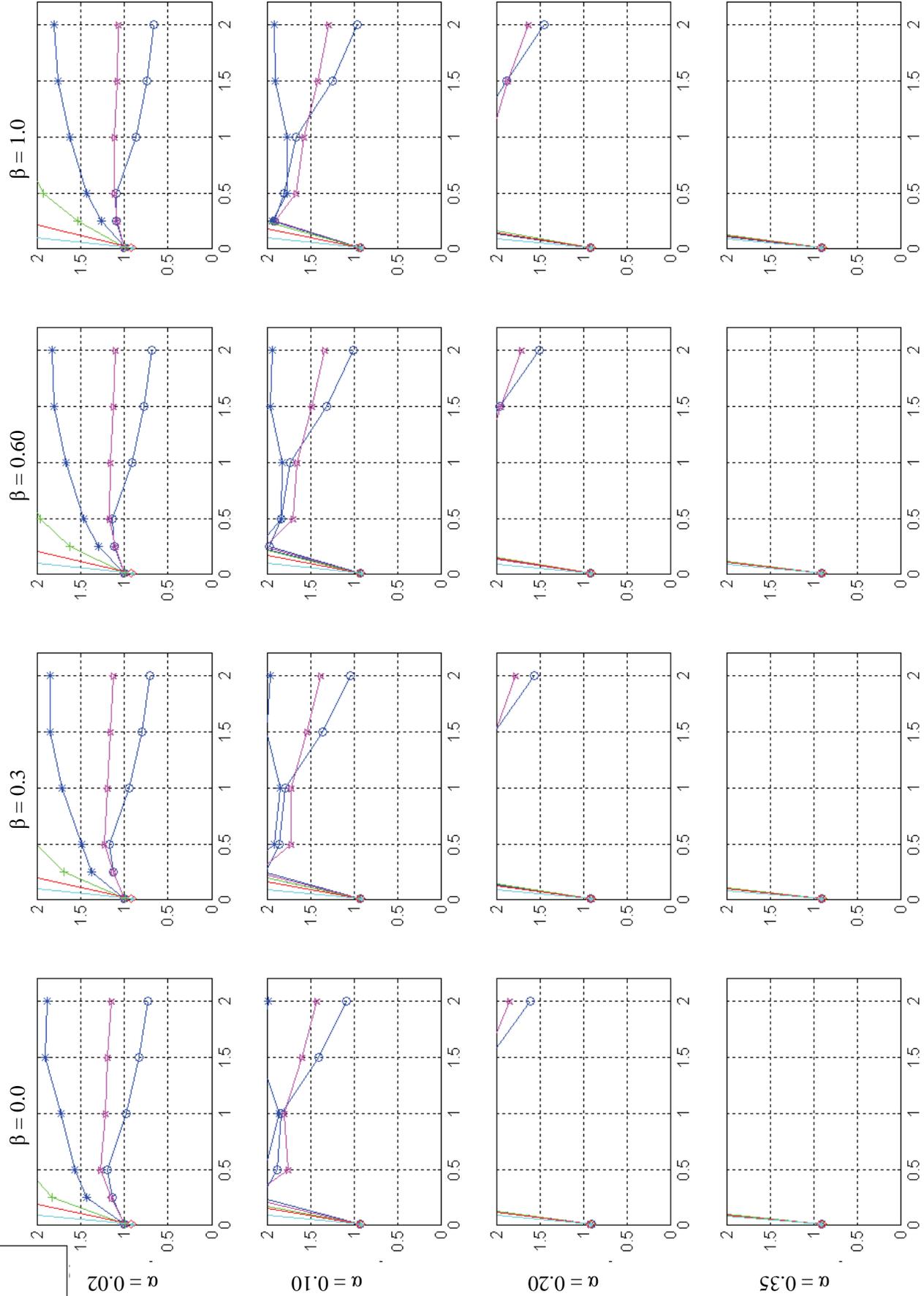
**Relative Performance (0\* Disp.Ratio + 1\* Accel.Ratio)**



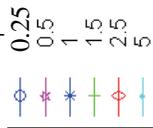
$\Psi$  ( $\eta_{EPS} = 0.10; \Psi = \eta_{SCS} / \eta_{EPS}$ )



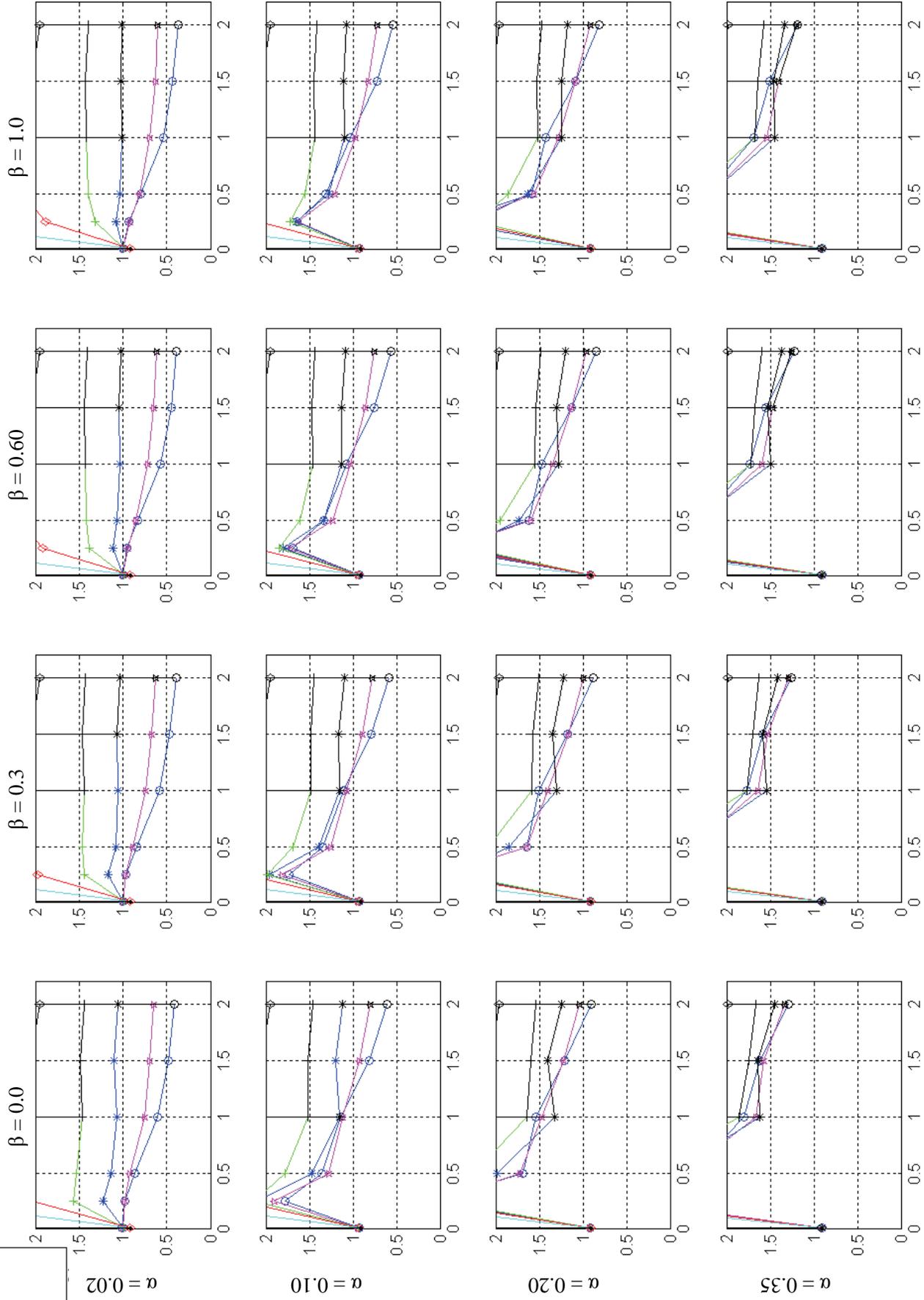
**Relative Performance (0\* Disp.Ratio + 1\* Accel.Ratio)**



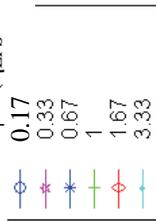
$\Psi$  ( $\eta_{EPS} = 0.20$ ;  $\Psi = \eta_{scs} / \eta_{EPS}$ )



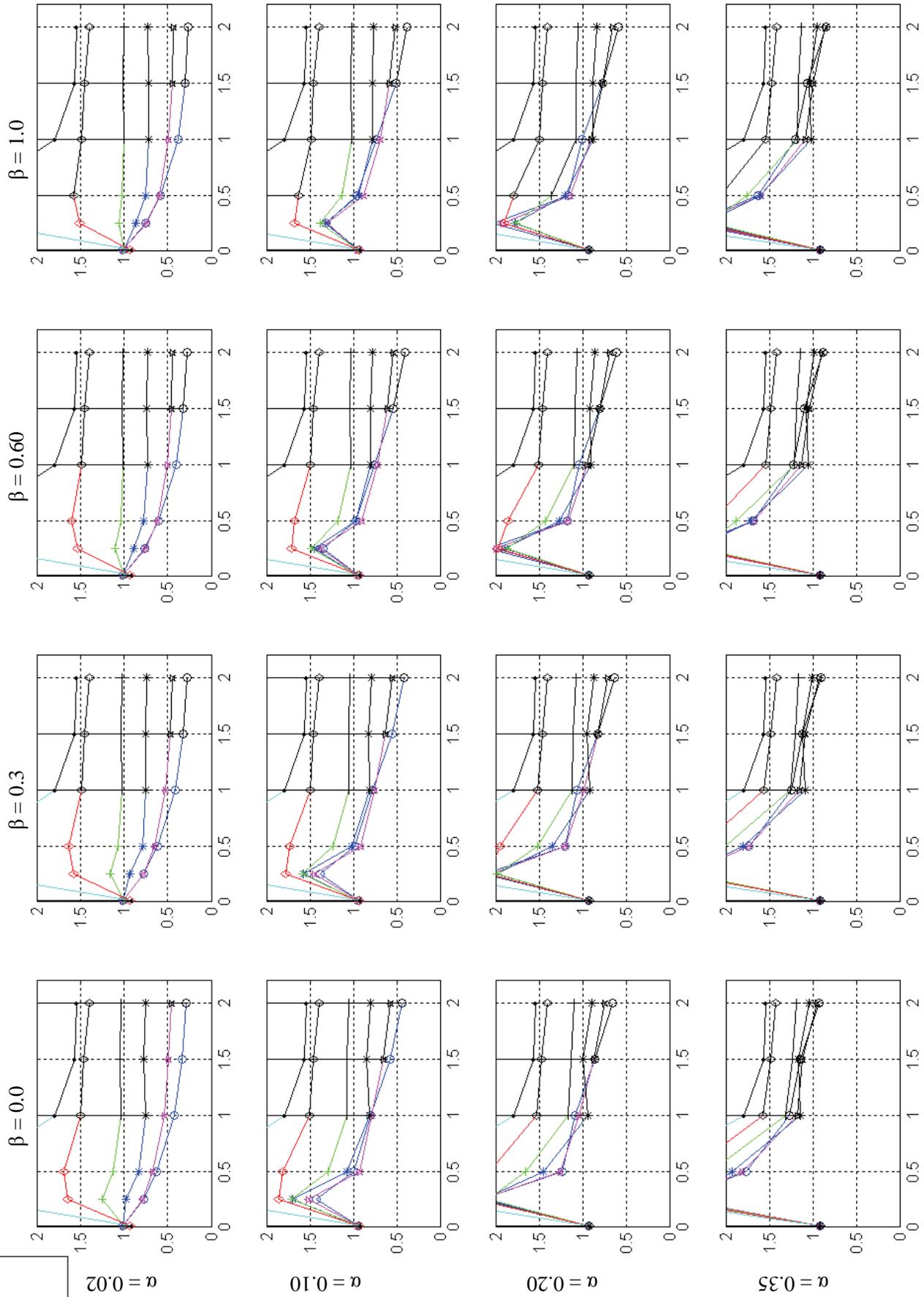
**Relative Performance (0\* Disp.Ratio + 1\* Accel.Ratio)**



$\Psi$  ( $\eta_{\text{EPS}} = 0.30; \Psi = \eta_{\text{SCS}} / \eta_{\text{EPS}}$ )



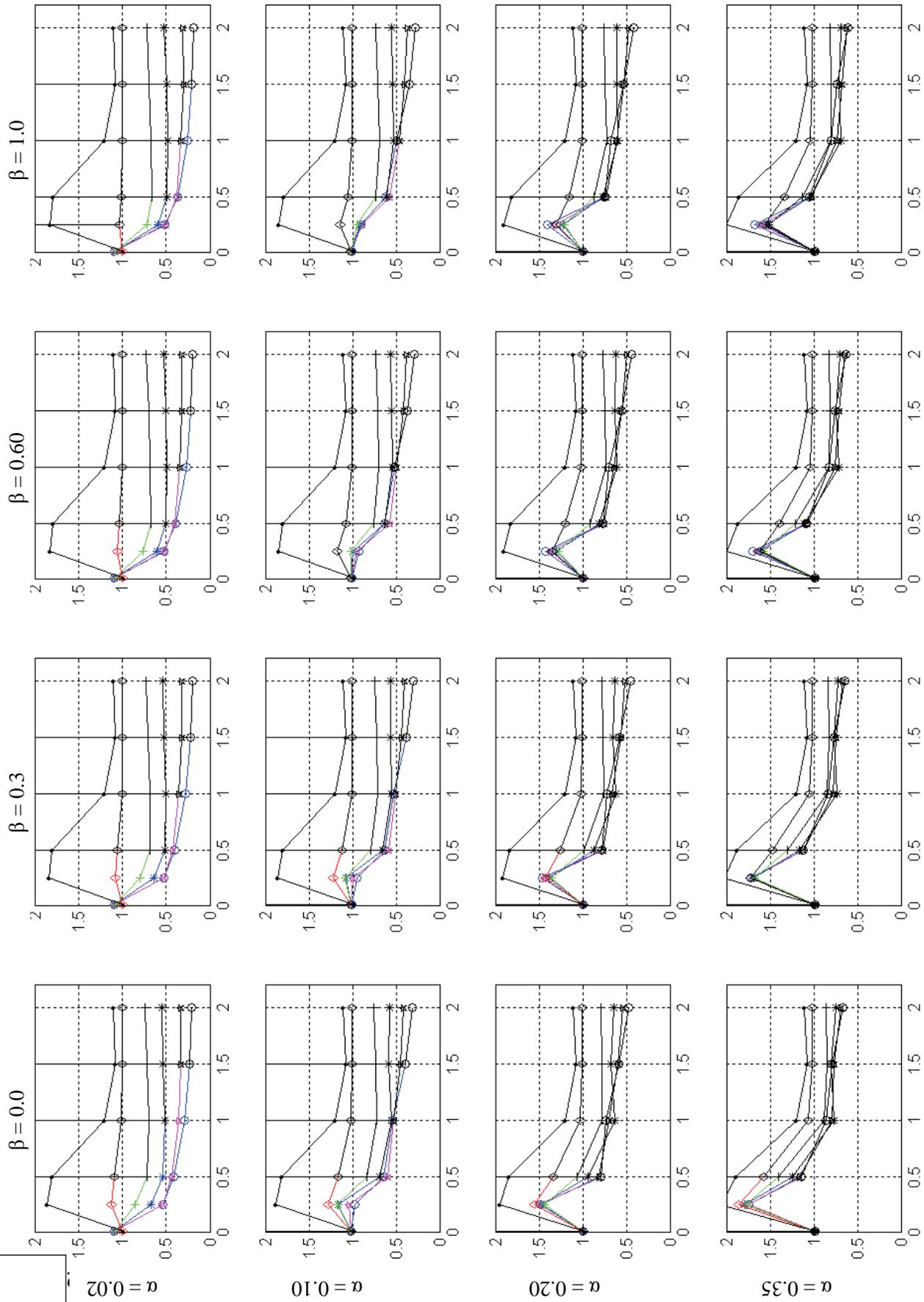
**Relative Performance (0\* Disp.Ratio + 1\* Accel.Ratio)**



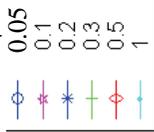
$\Psi$  ( $\eta_{EPS} = 0.50; \Psi = \eta_{SCS} / \eta_{EPS}$ )



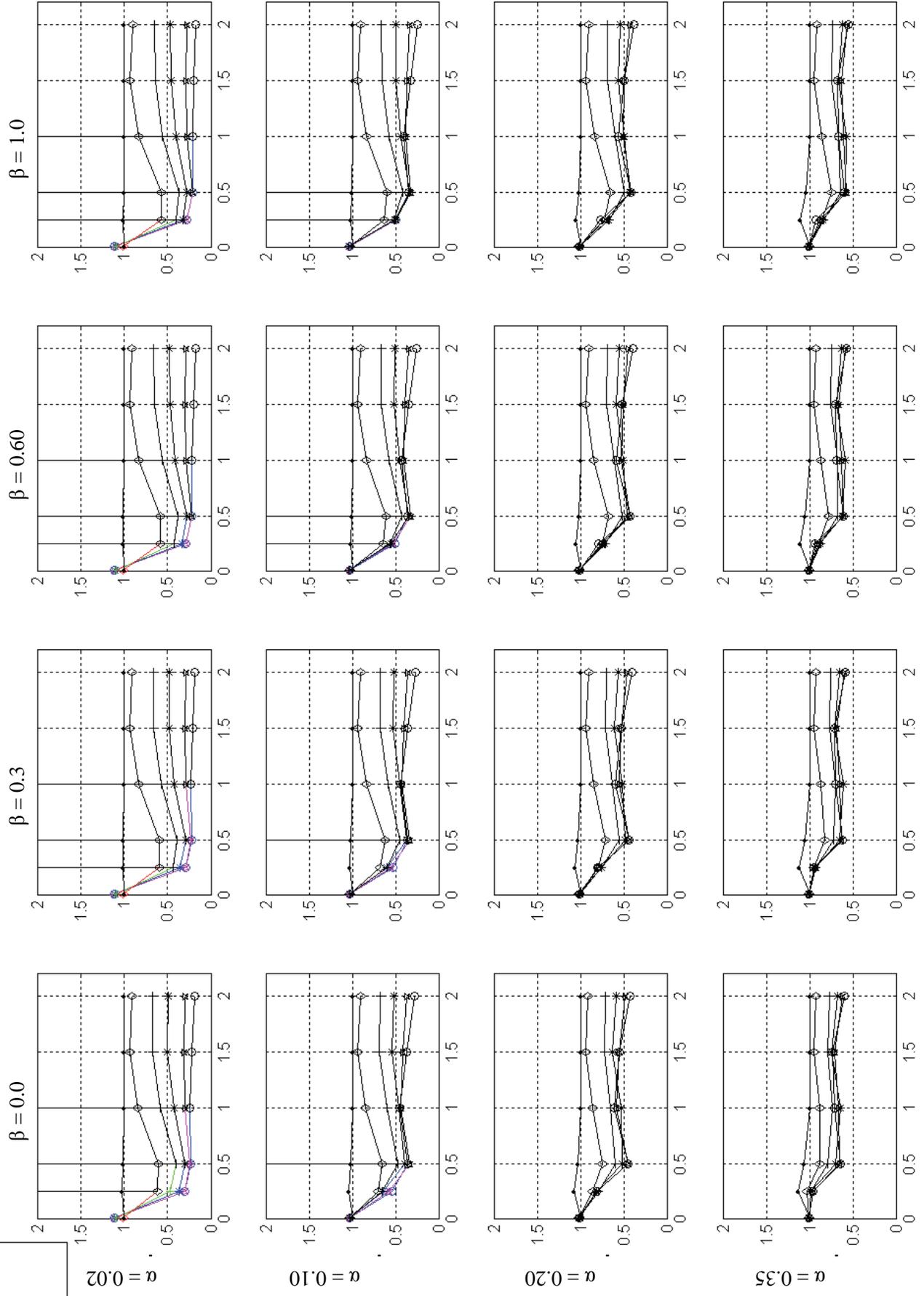
### Relative Performance (0\* Disp.Ratio + 1\* Accel.Ratio)



$\Psi$  ( $\eta_{EPS} = 1.00; \Psi = \eta_{SCS} / \eta_{EPS}$ )



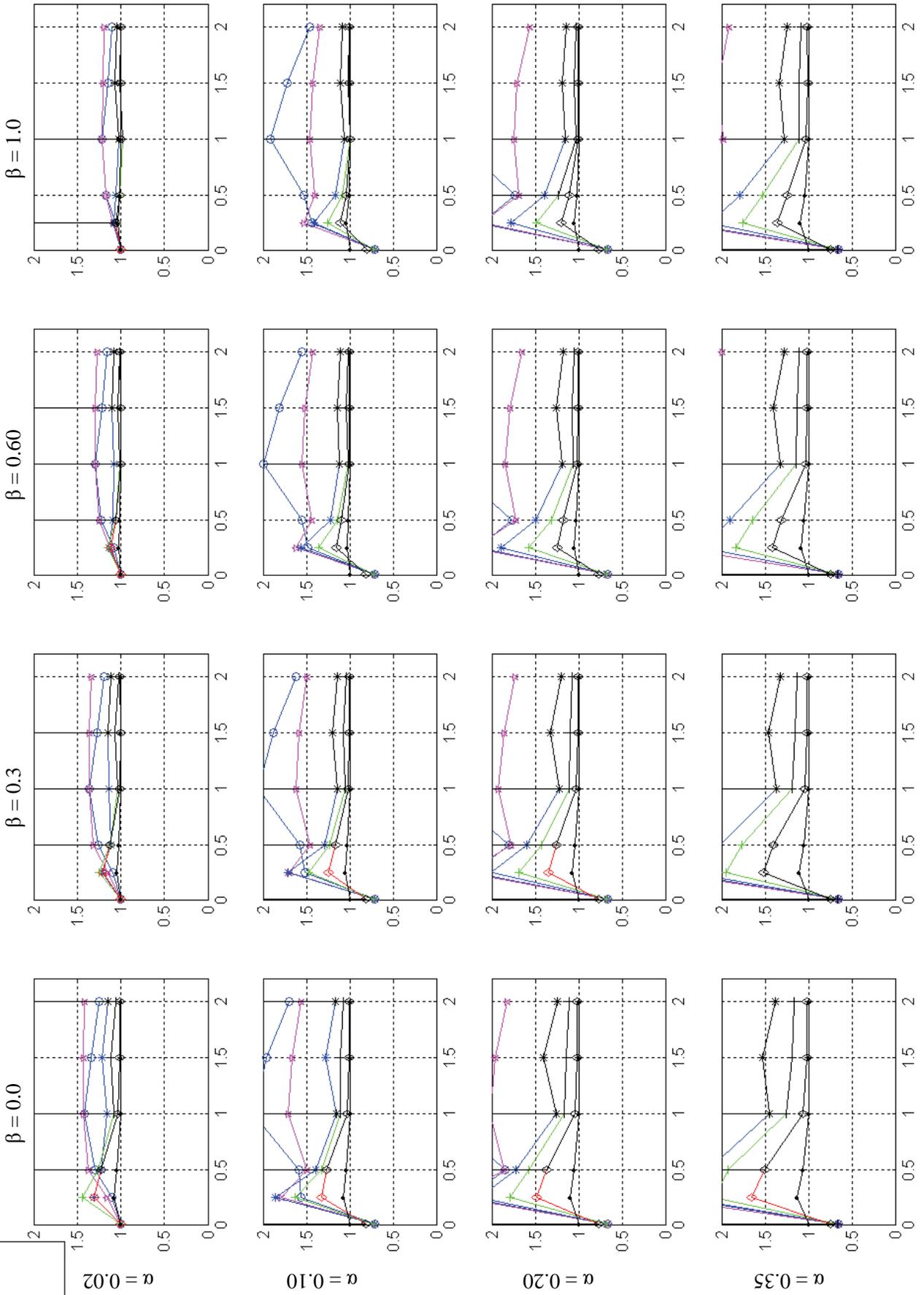
### Relative Performance (0\* Disp.Ratio + 1\* Accel.Ratio)



$\eta_{SCS}$   
( $\psi = \eta_{SCS} / \eta_{EPS} = 1$ )

- $\circ$  0.05
- $*$  0.10
- $*$  0.20
- $*$  0.30
- $*$  0.50
- $*$  1.00

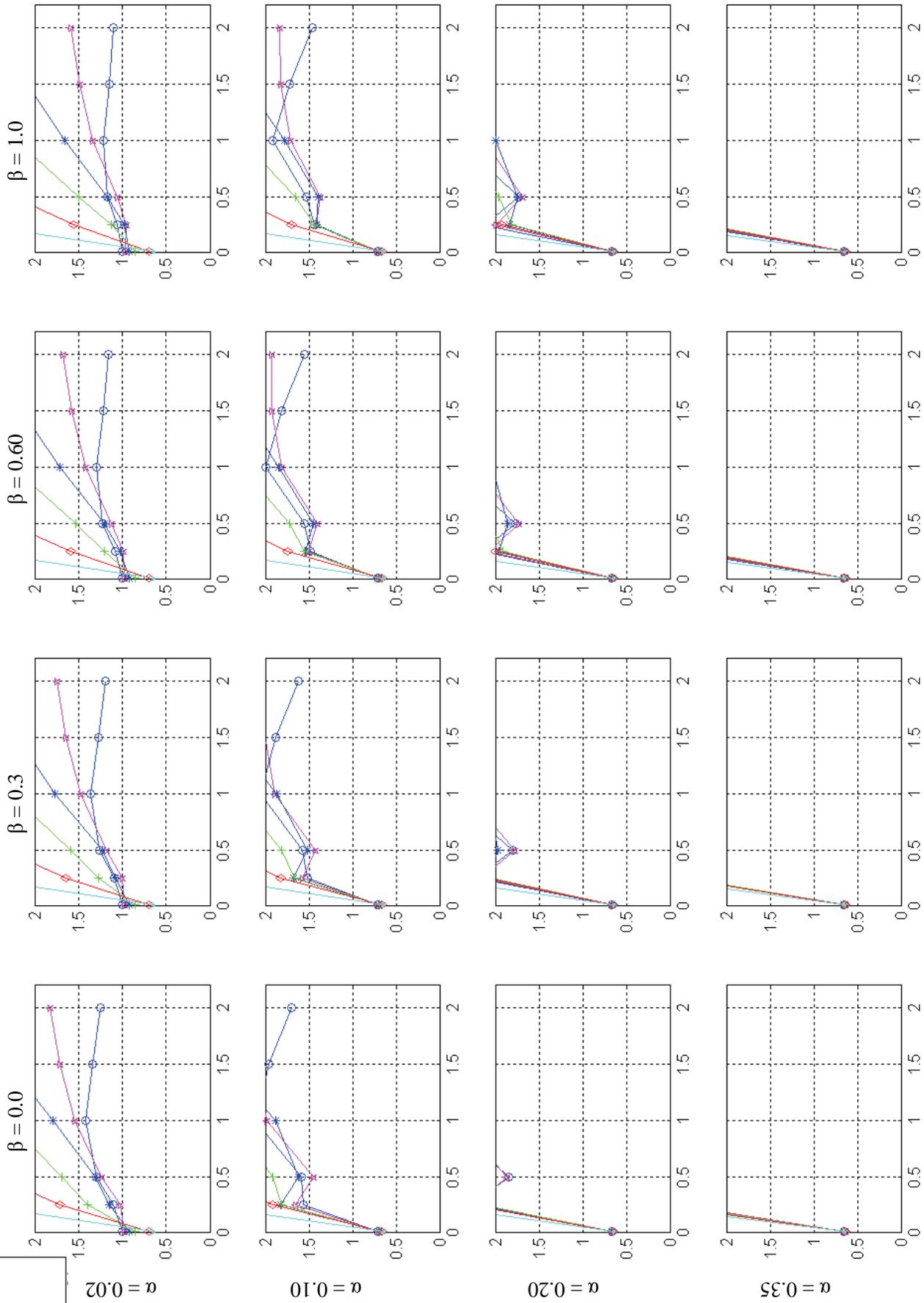
### Relative Performance (0.3\* Disp.Ratio + 0.7\* Accel.Ratio)



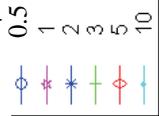
$\Psi$  ( $\eta_{\text{EPS}} = 0.05$ ;  $\psi = \eta_{\text{SCS}} / \eta_{\text{EPS}}$ )



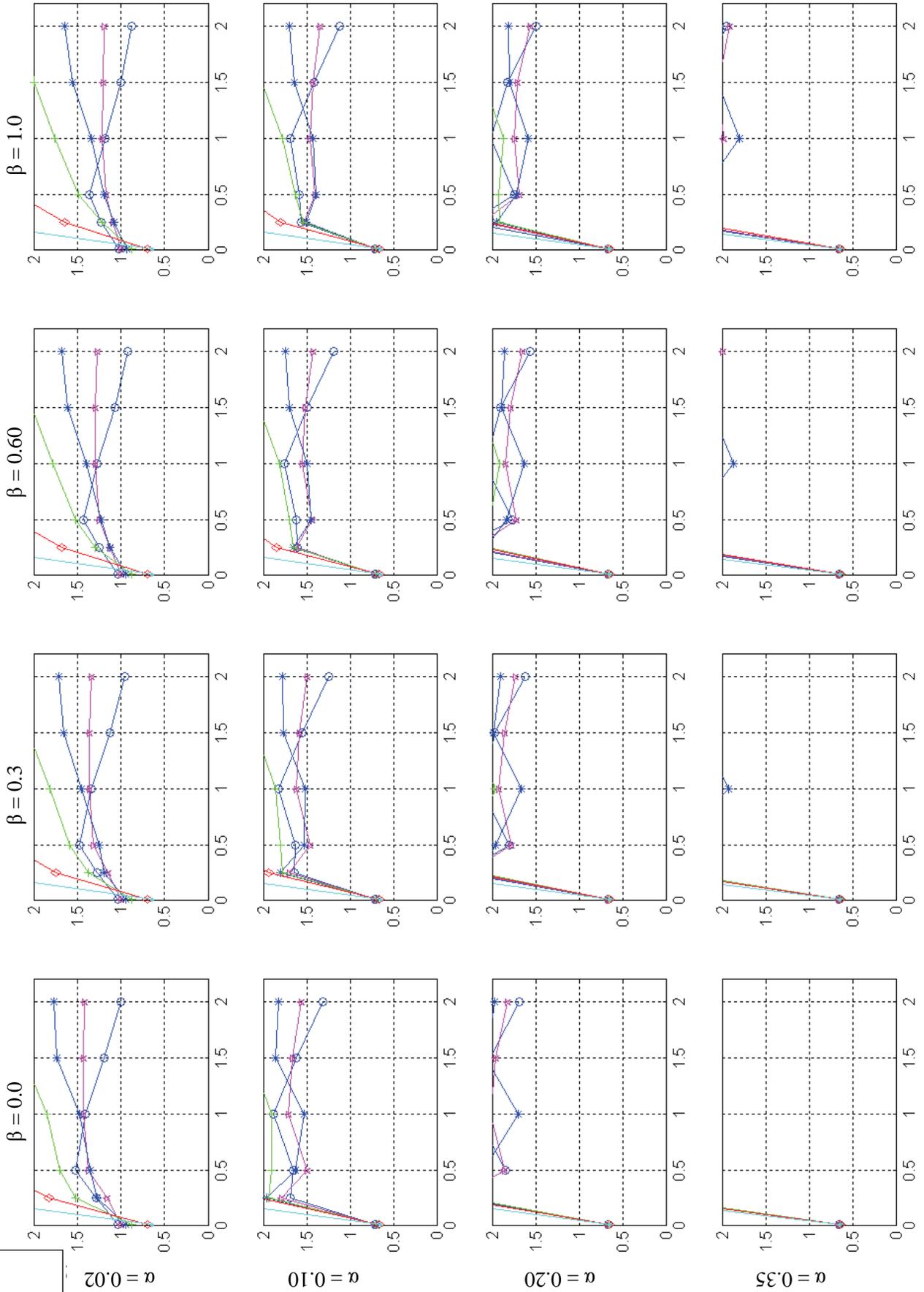
### Relative Performance (0.3\* Disp.Ratio + 0.7\* Accel.Ratio)



$\Psi$  ( $\eta_{EPS} = 0.10$ ;  $\Psi = \eta_{SCS} / \eta_{EPS}$ )



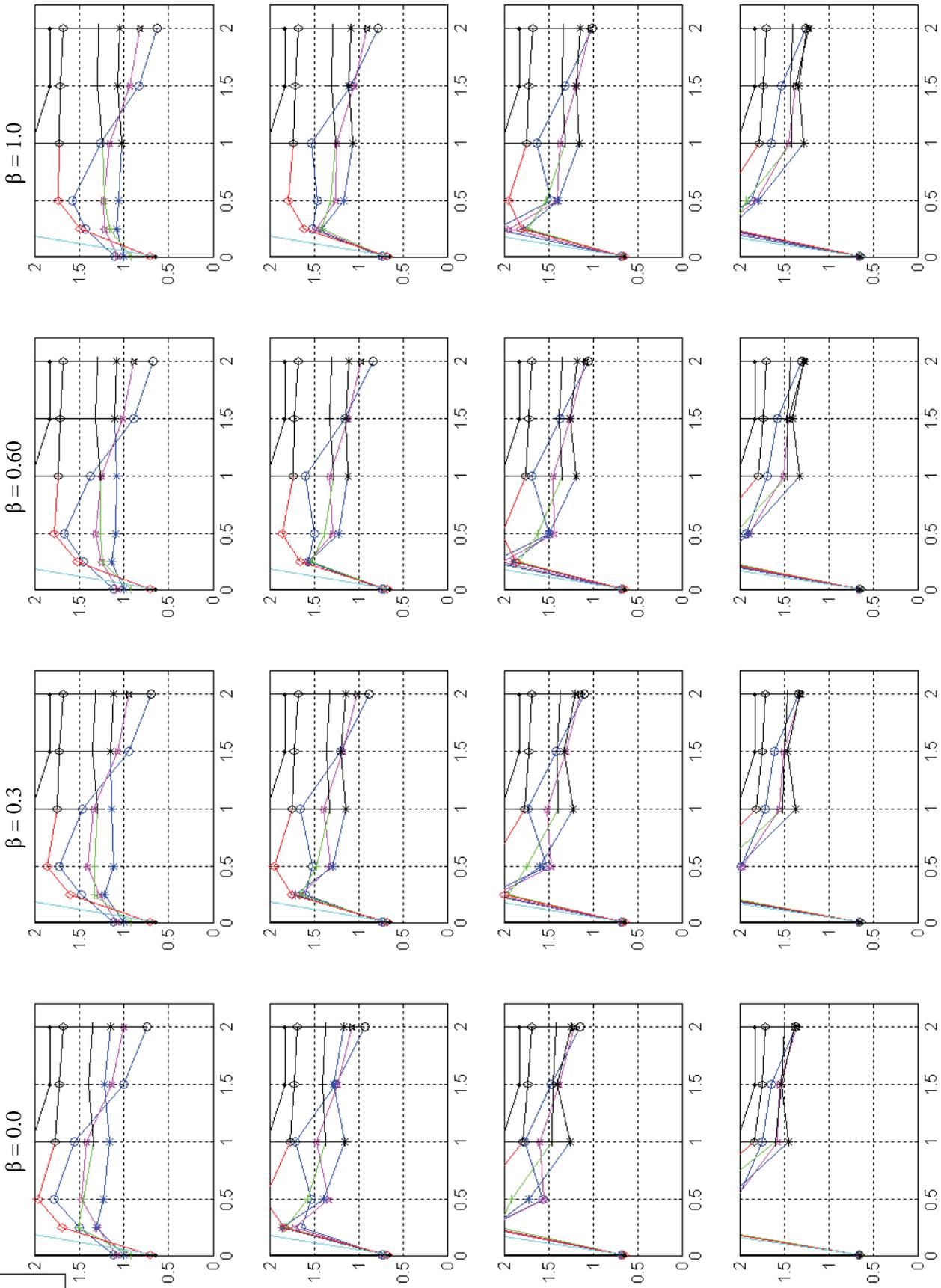
### Relative Performance (0.3\* Disp.Ratio + 0.7\* Accel.Ratio)



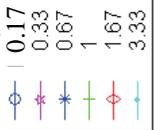
$\Psi$  ( $\eta_{EPS} = 0.20; \psi = \eta_{SCS} / \eta_{EPS}$ )

- 0.25
- 0.5
- 1
- 1.5
- 2.5
- 5

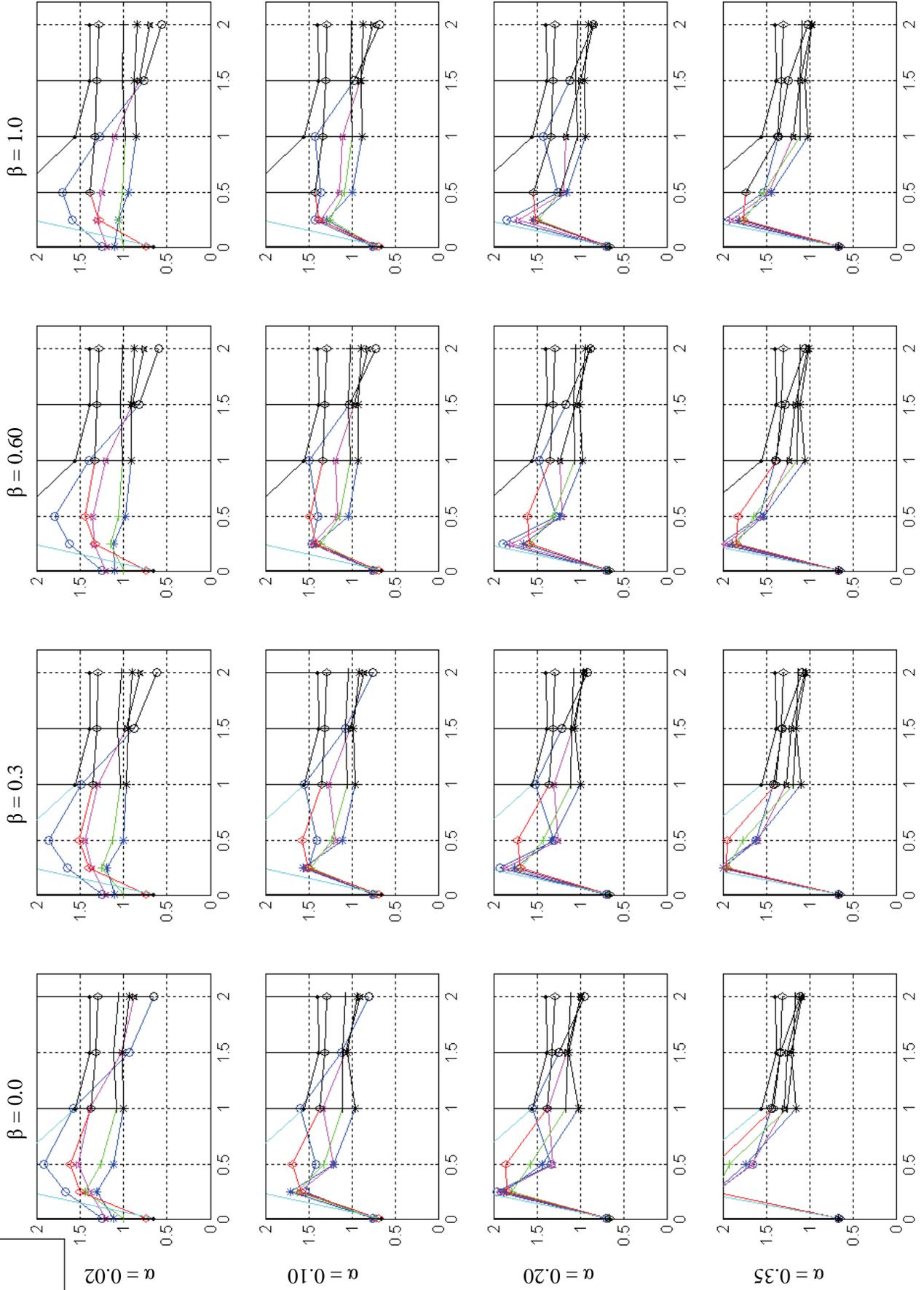
### Relative Performance (0.3\* Disp.Ratio + 0.7\* Accel.Ratio)



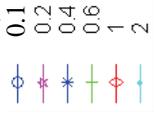
$\psi$  ( $\eta_{EPS} = 0.30$ ;  $\psi = \eta_{SCS} / \eta_{EPS}$ )



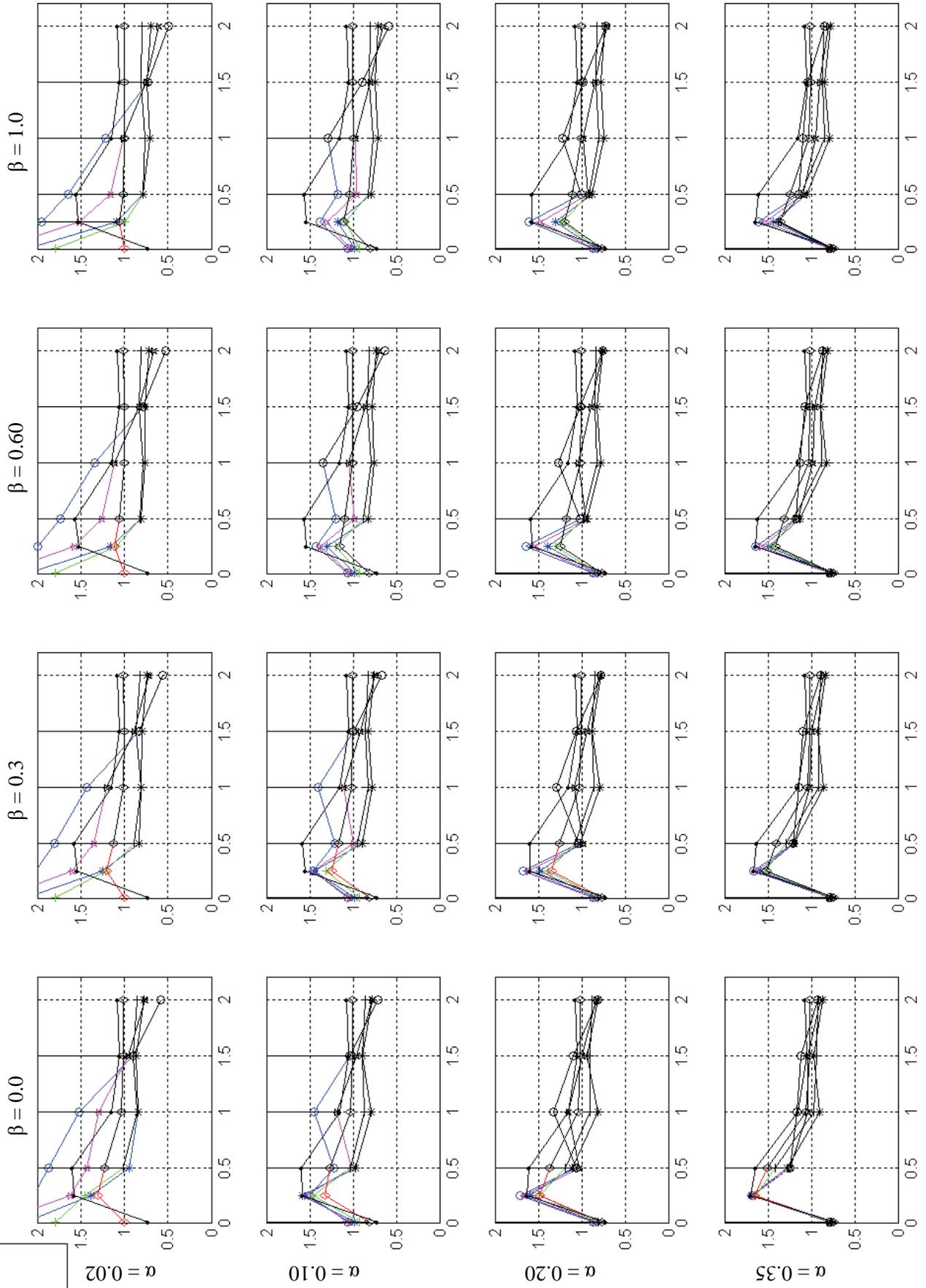
**Relative Performance (0.3\* Disp.Ratio + 0.7\* Accel.Ratio)**



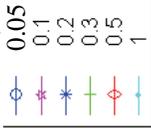
$\Psi$  ( $\eta_{EPS} = 0.50; \Psi = \eta_{SCS} / \eta_{EPS}$ )



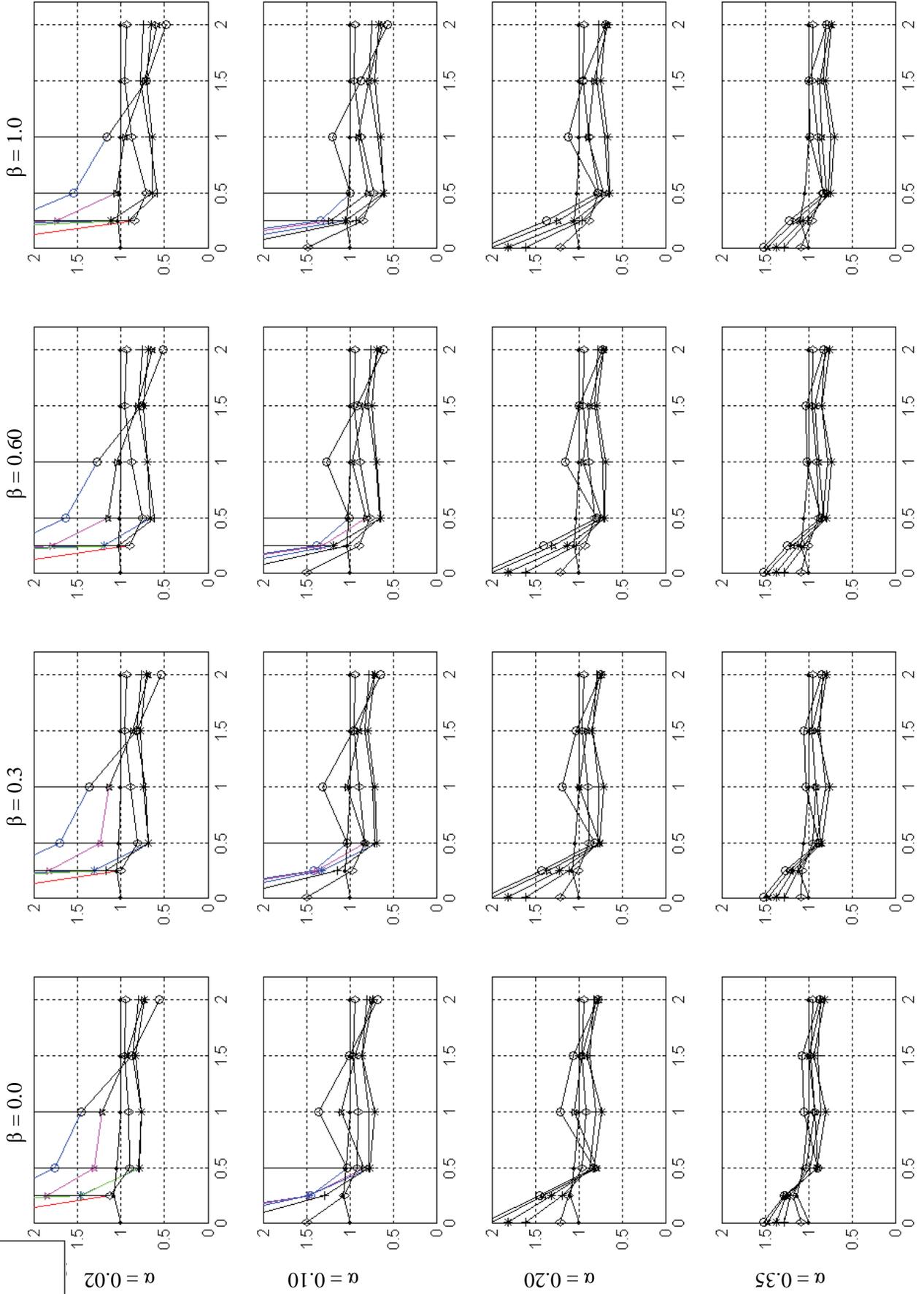
### Relative Performance (0.3\* Disp.Ratio + 0.7\* Accel.Ratio)



$\Psi$  ( $\eta_{EPS} = 1.00$ ;  $\psi = \eta_{SCS} / \eta_{EPS}$ )



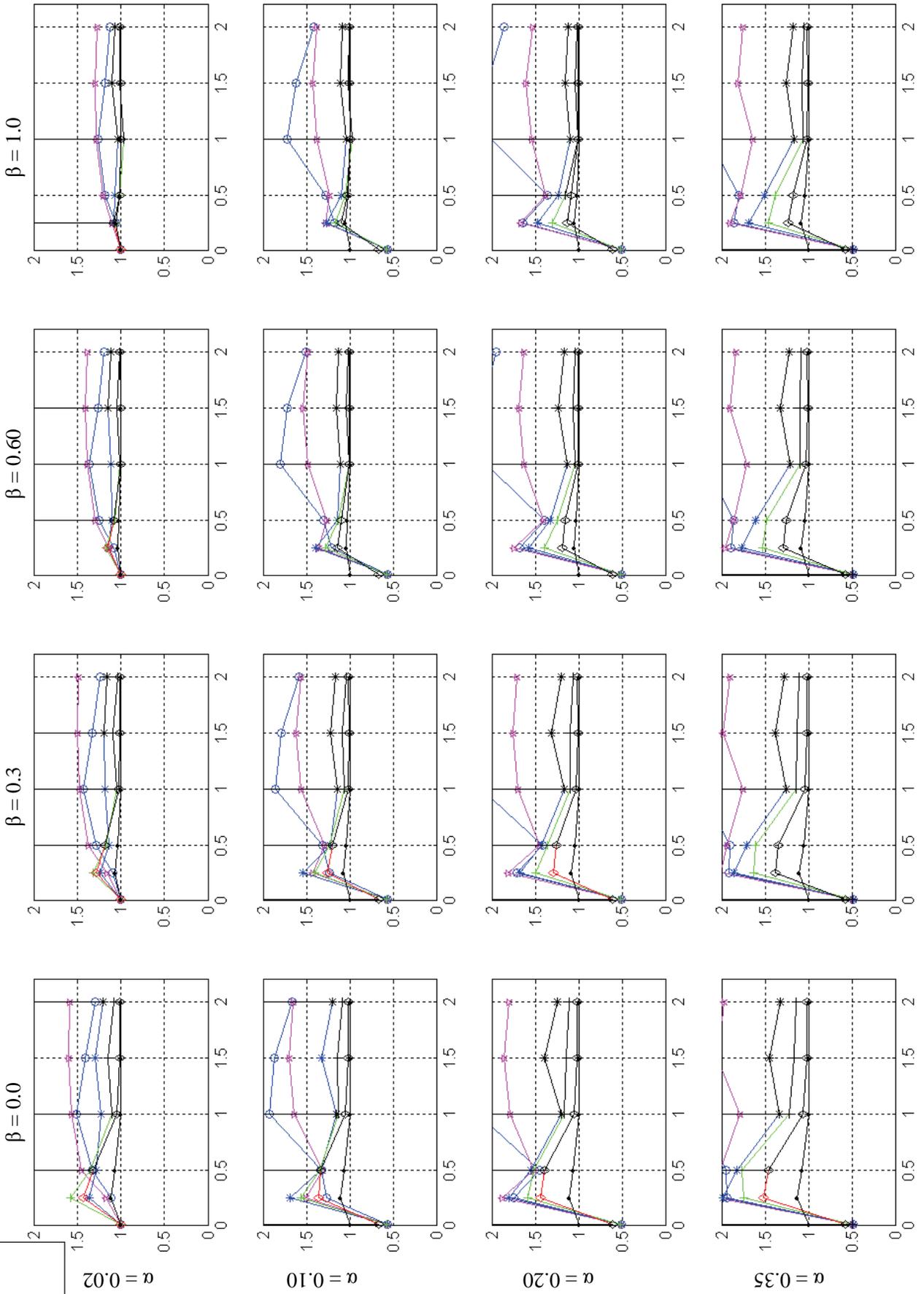
### Relative Performance (0.3\* Disp.Ratio + 0.7\* Accel.Ratio)



$\eta_{SCS}$   
( $\psi = \eta_{SCS} / \eta_{EPS} = 1$ )

- 0.05
- \* 0.10
- ◇ 0.20
- ◇ 0.30
- ◇ 0.50
- ◇ 1.00

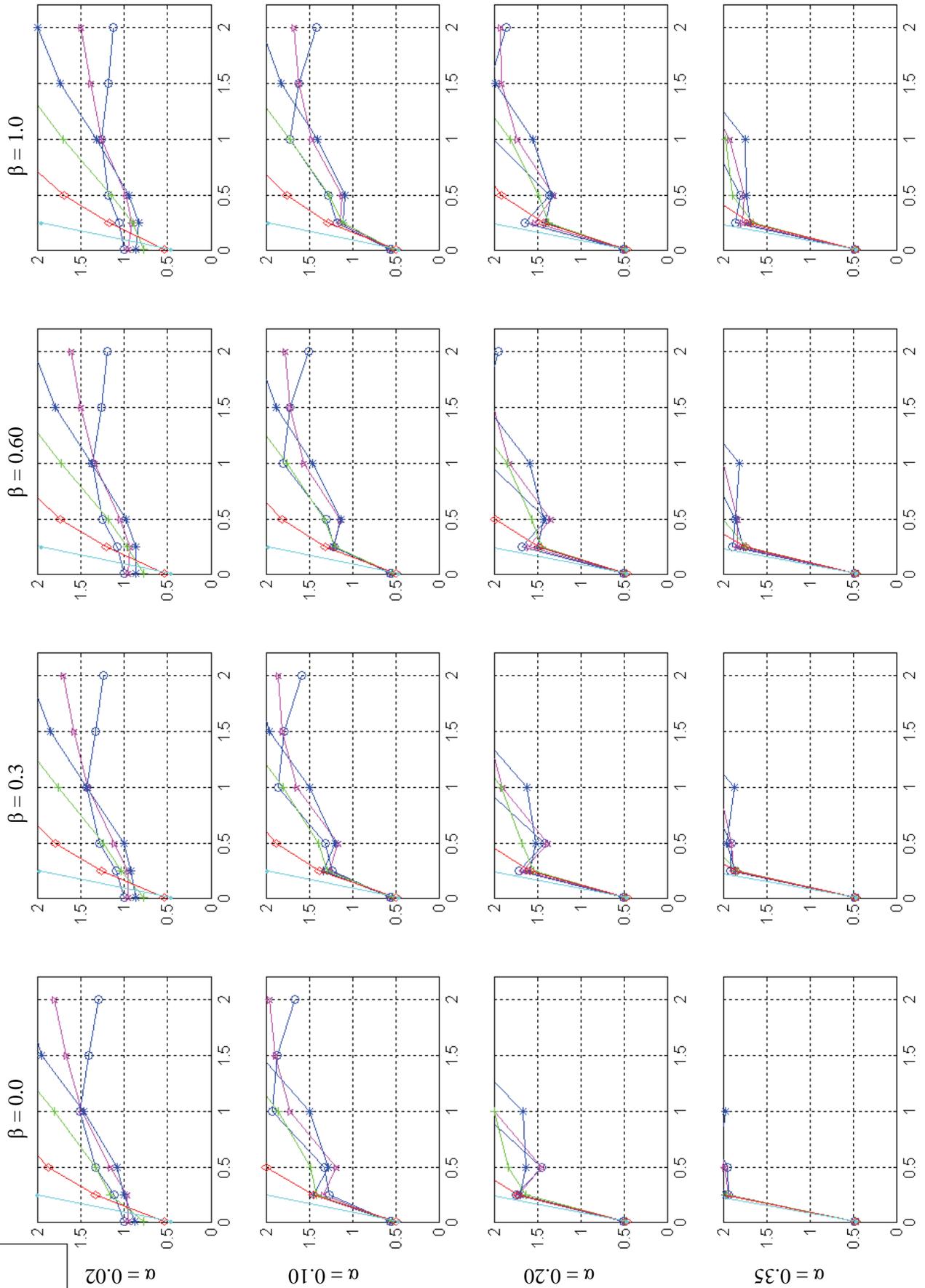
### Relative Performance (0.5\* Disp.Ratio + 0.5\* Accel.Ratio)



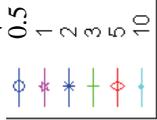
$\Psi$  ( $\eta_{EPS} = 0.05$ ;  $\Psi = \eta_{SCS} / \eta_{EPS}$ )



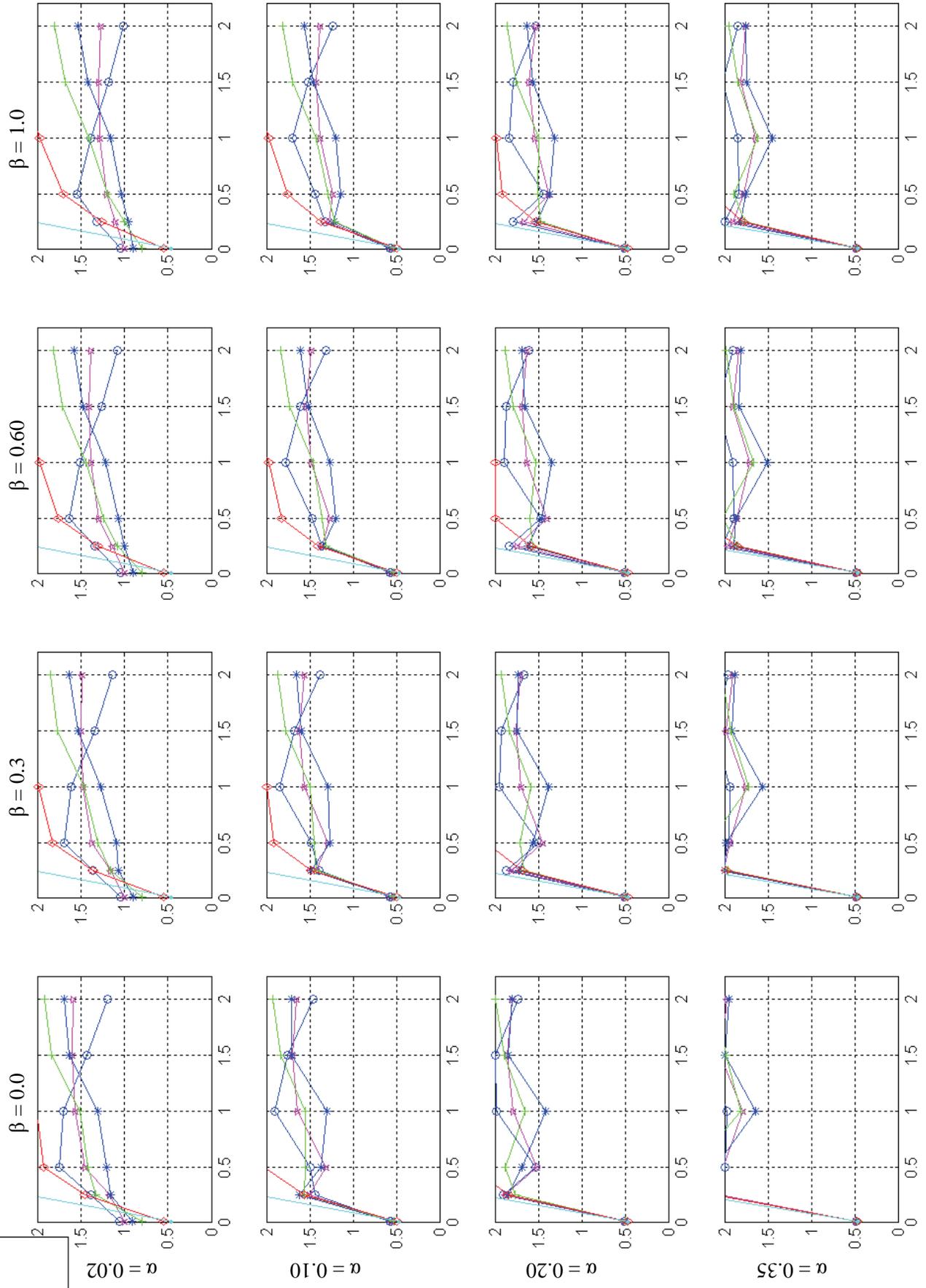
**Relative Performance (0.5\* Disp.Ratio + 0.5\* Accel.Ratio)**



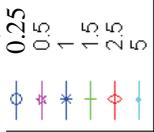
$\Psi$  ( $\eta_{EPS} = 0.10$ ;  $\psi = \eta_{SCS} / \eta_{EPS}$ )



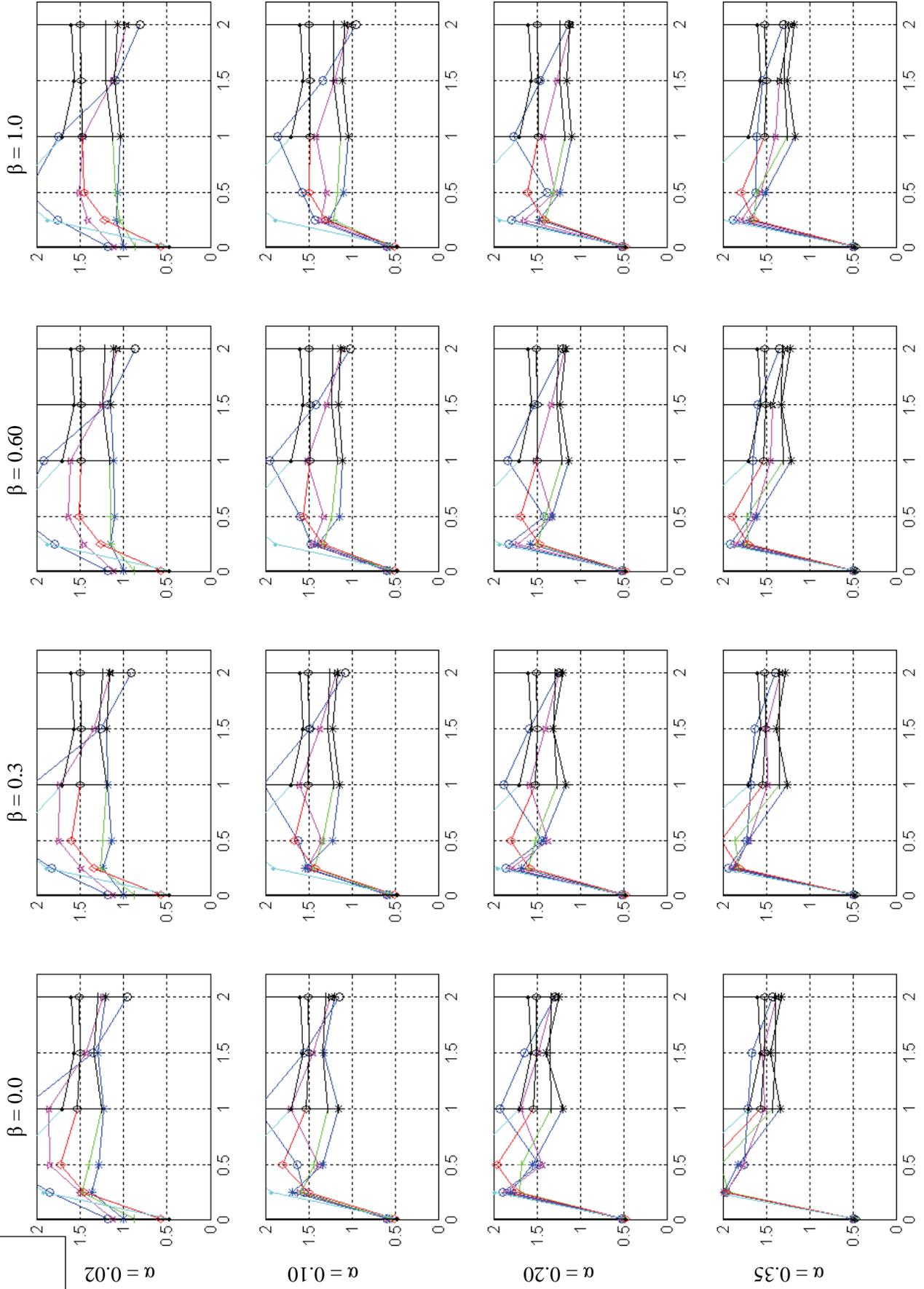
### Relative Performance (0.5\* Disp.Ratio + 0.5\* Accel.Ratio)



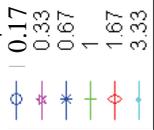
$\Psi$  ( $\eta_{EPS} = 0.20; \psi = \eta_{SCS} / \eta_{EPS}$ )



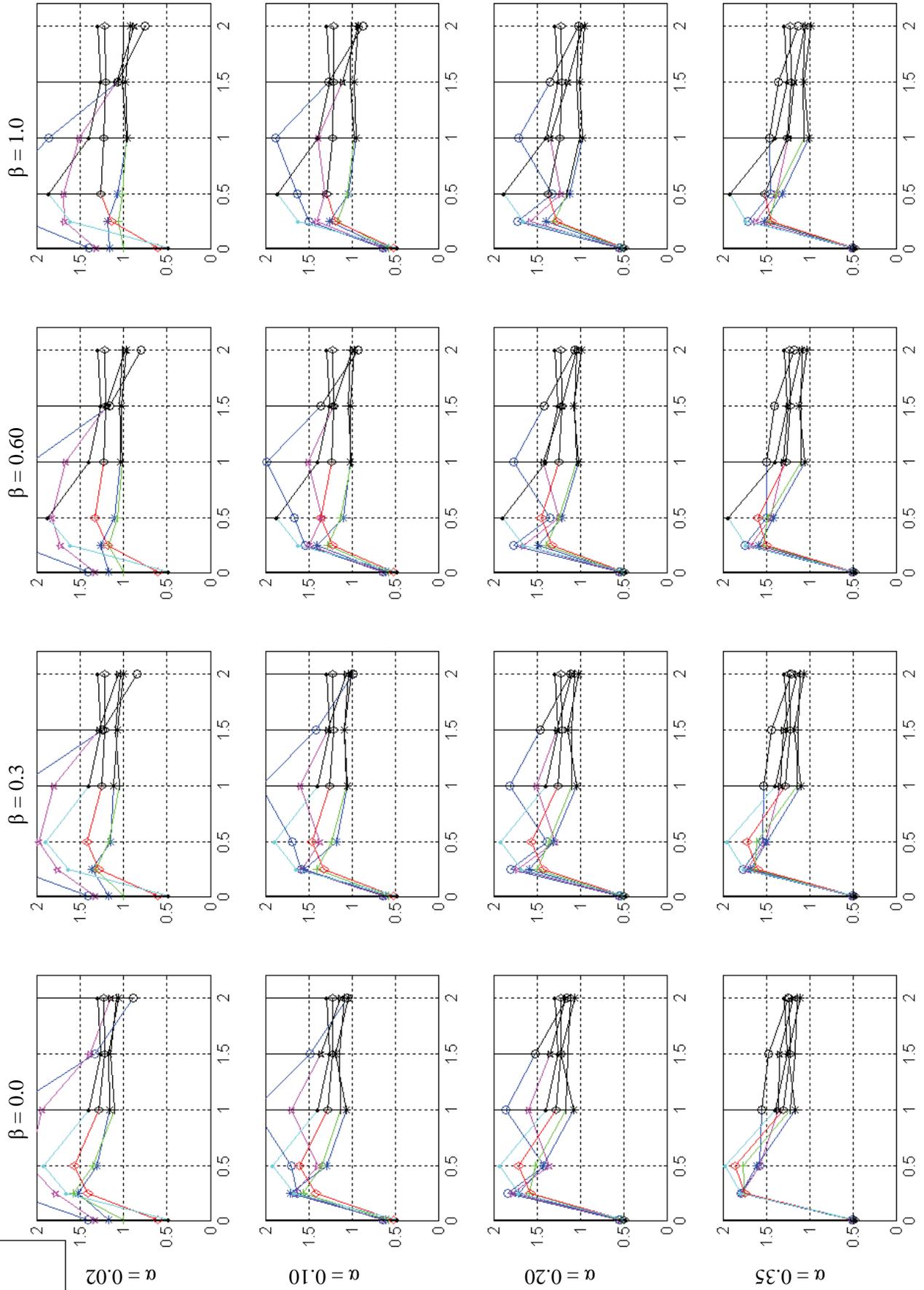
### Relative Performance (0.5\* Disp.Ratio + 0.5\* Accel.Ratio)



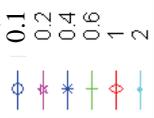
$\Psi$  ( $\eta_{EPS} = 0.30$ ;  $\psi = \eta_{SCS} / \eta_{EPS}$ )



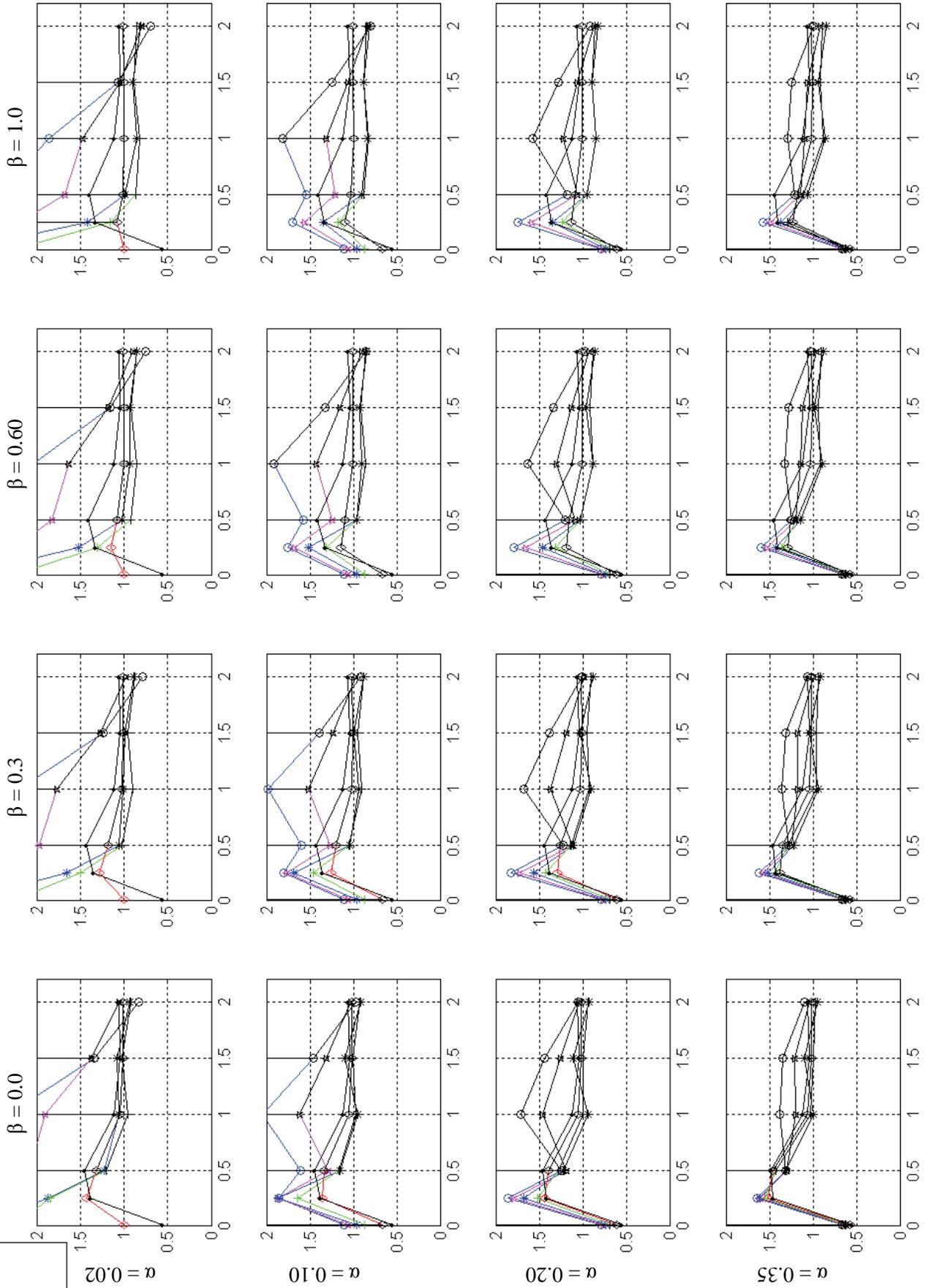
**Relative Performance (0.5\* Disp.Ratio + 0.5\* Accel.Ratio)**



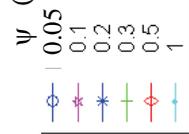
$\psi$  ( $\eta_{EPS} = 0.50$ ;  $\psi = \eta_{SCS} / \eta_{EPS}$ )



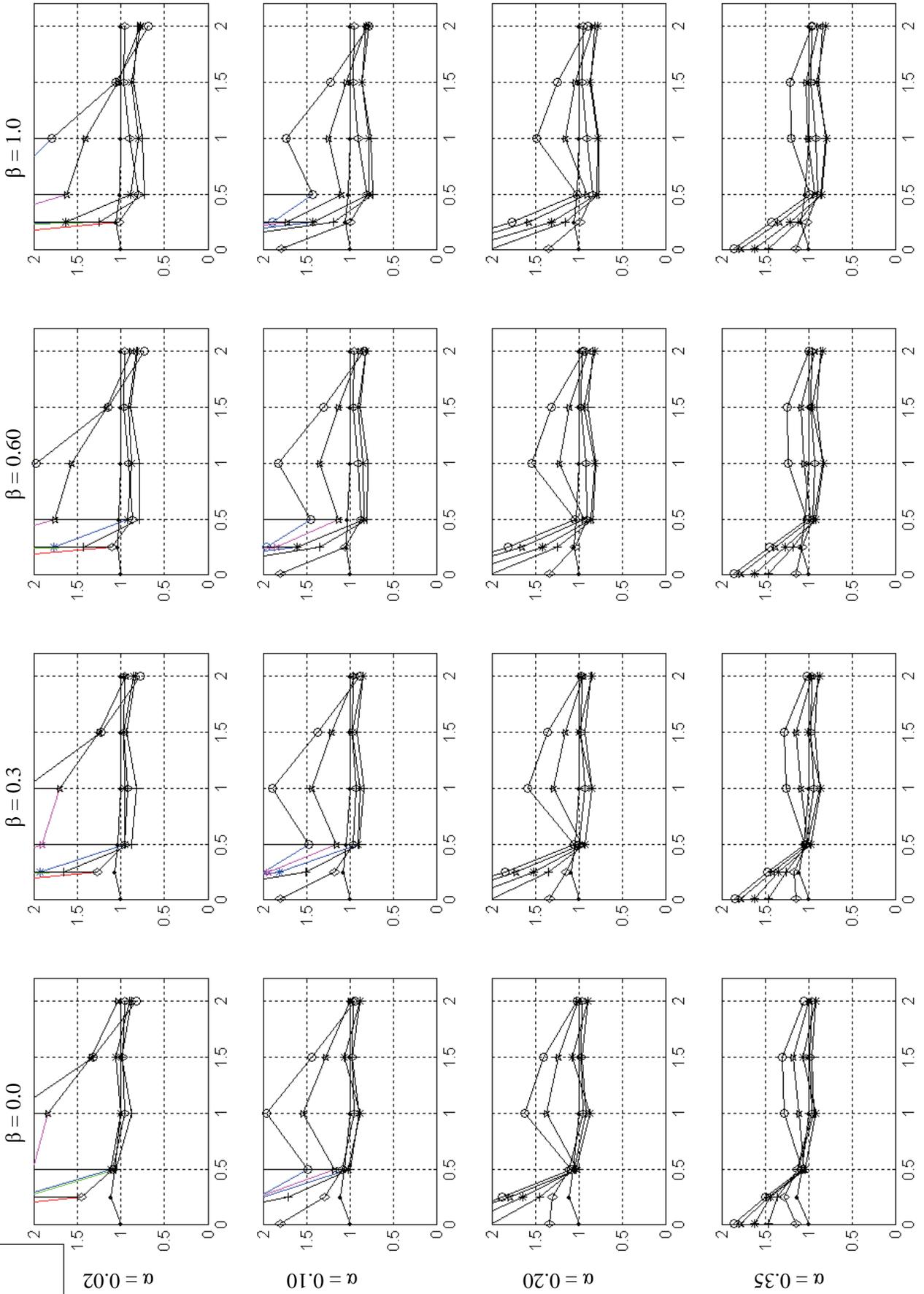
### Relative Performance (0.5\* Disp.Ratio + 0.5\* Accel.Ratio)



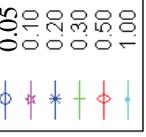
$\Psi$  ( $\eta_{EPS} = 1.00$ ;  $\psi = \eta_{SCS} / \eta_{EPS}$ )



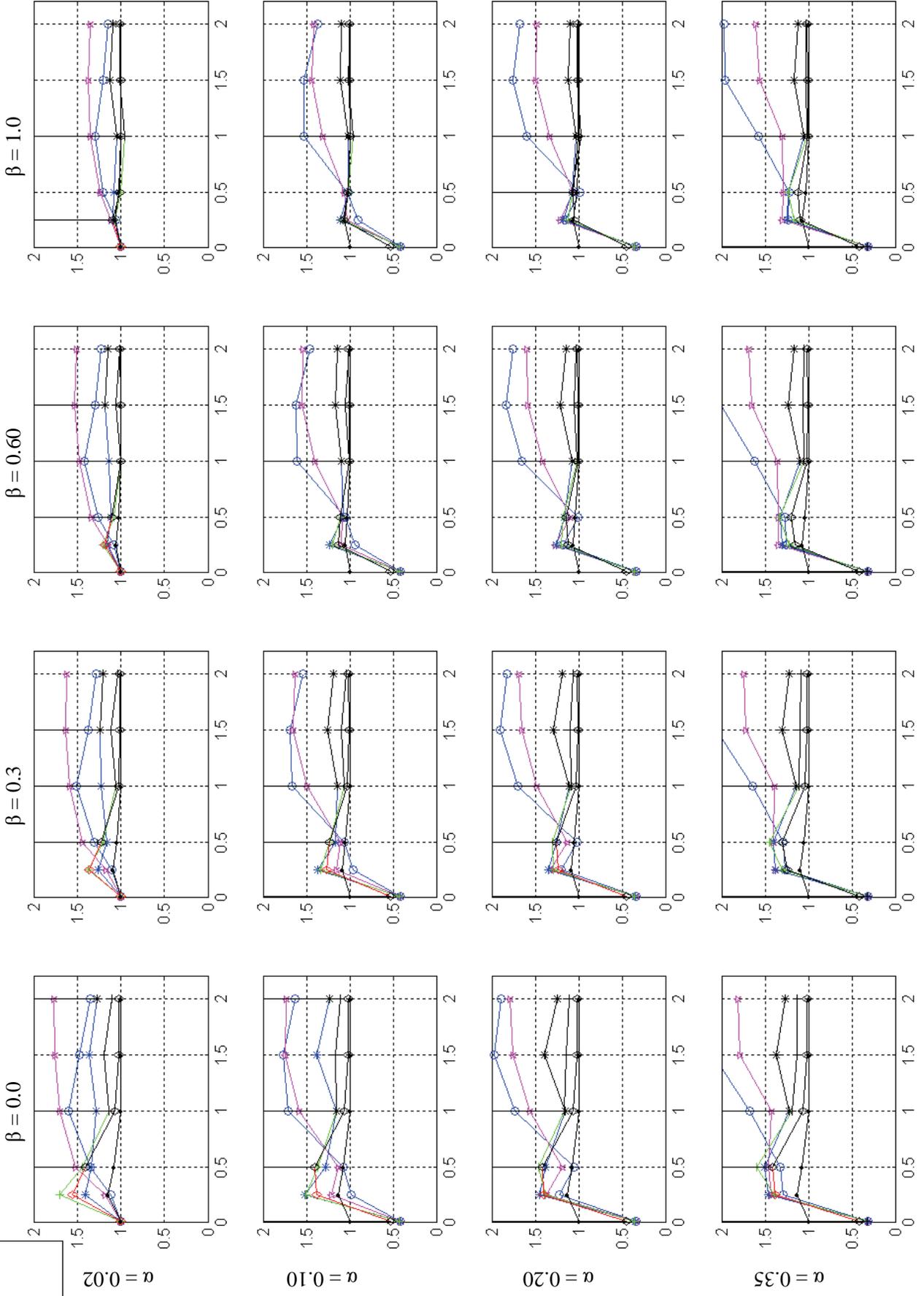
### Relative Performance (0.5\* Disp.Ratio + 0.5\* Accel.Ratio)



$\eta_{SCS}$   
 $(\psi = \eta_{SCS} / \eta_{EPS} = 1)$



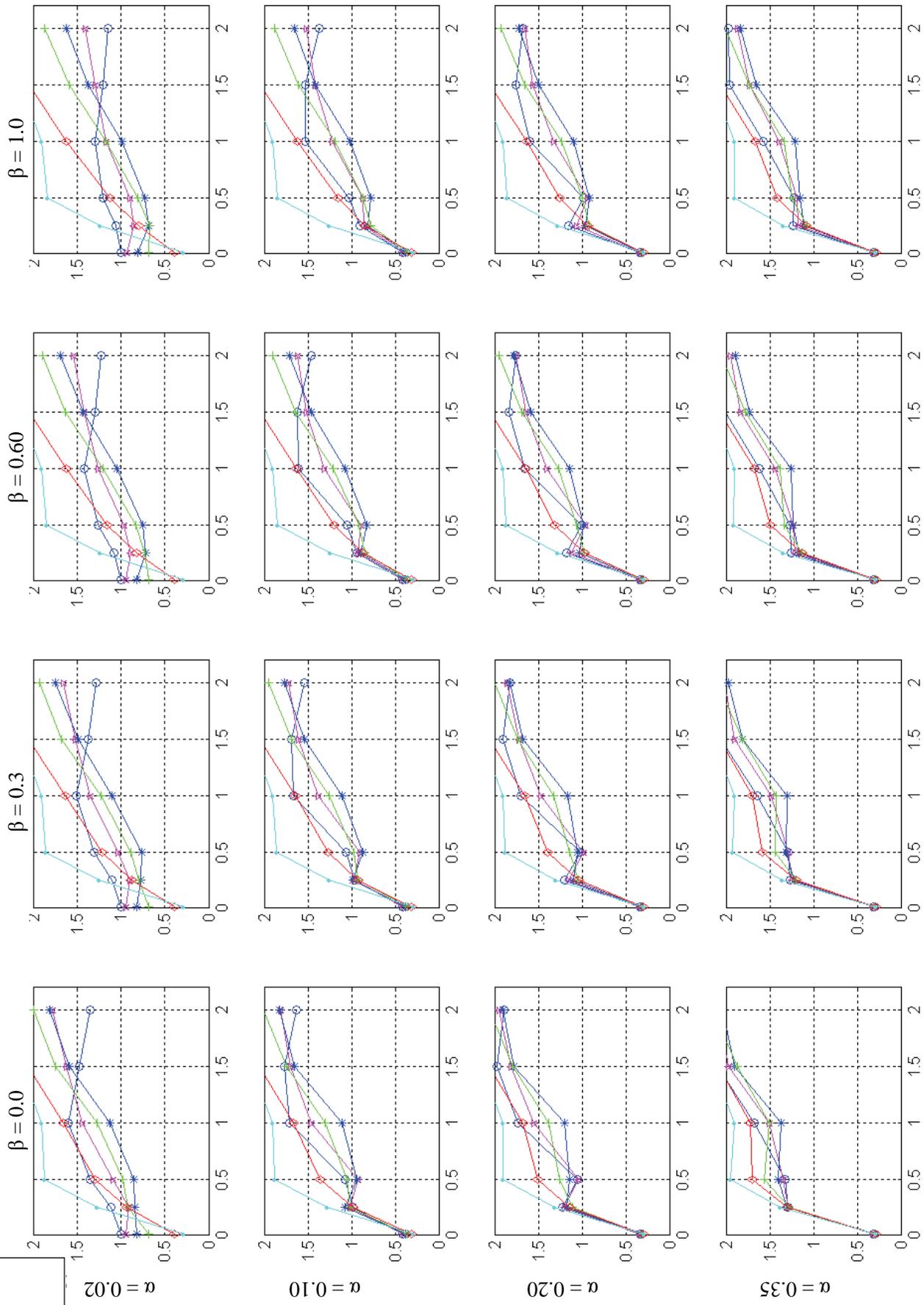
**Relative Performance (0.7\* Disp.Ratio + 0.3\* Accel.Ratio)**



$\Psi$  ( $\eta_{EPS} = 0.05$ ;  $\Psi = \eta_{SCS} / \eta_{EPS}$ )



**Relative Performance (0.7\* Disp.Ratio + 0.3\* Accel.Ratio)**



$\psi$  ( $\eta_{EPS} = 0.10$ ;  $\psi = \eta_{SCS} / \eta_{EPS}$ )

0.5

1

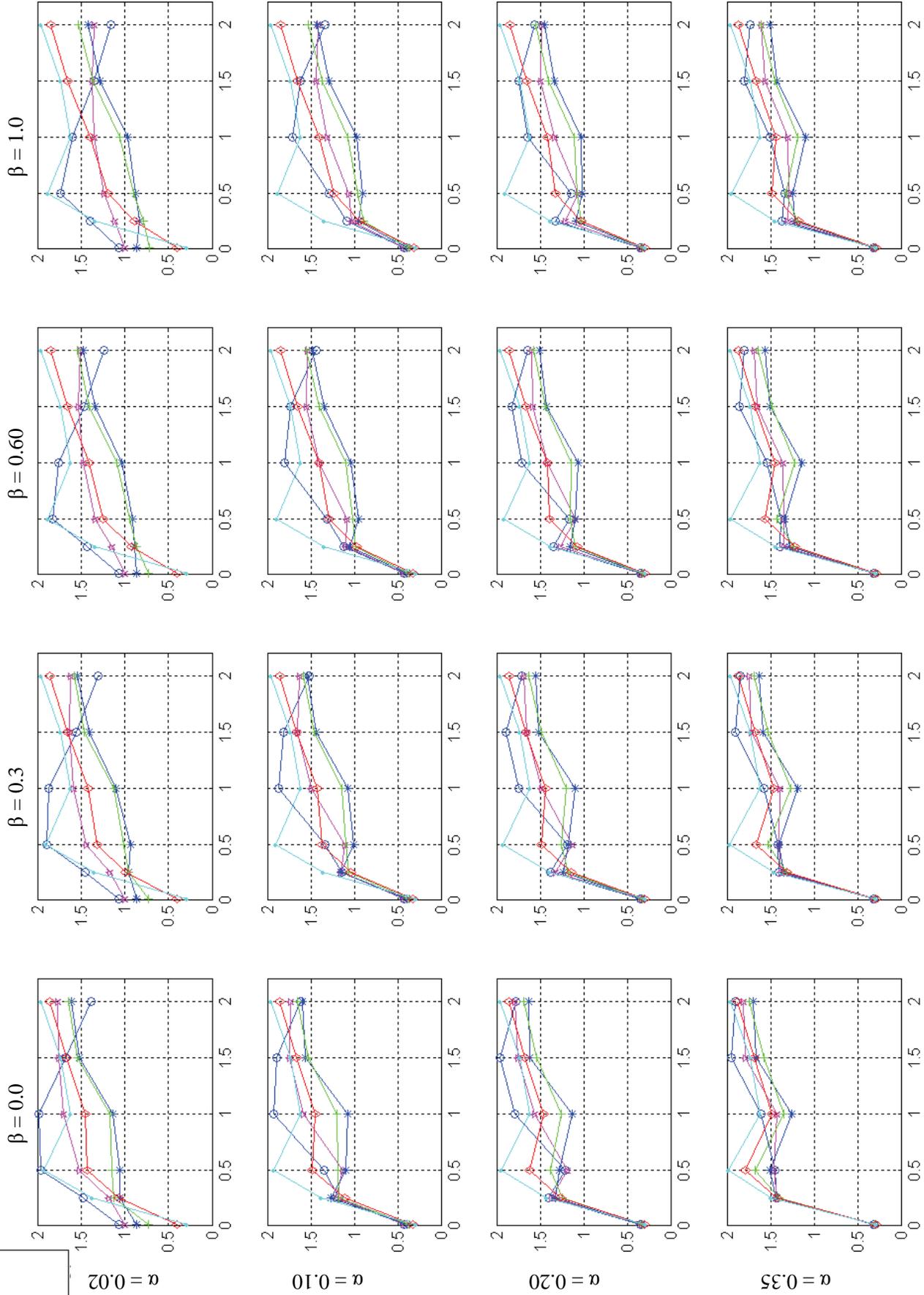
2

3

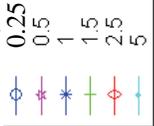
5

10

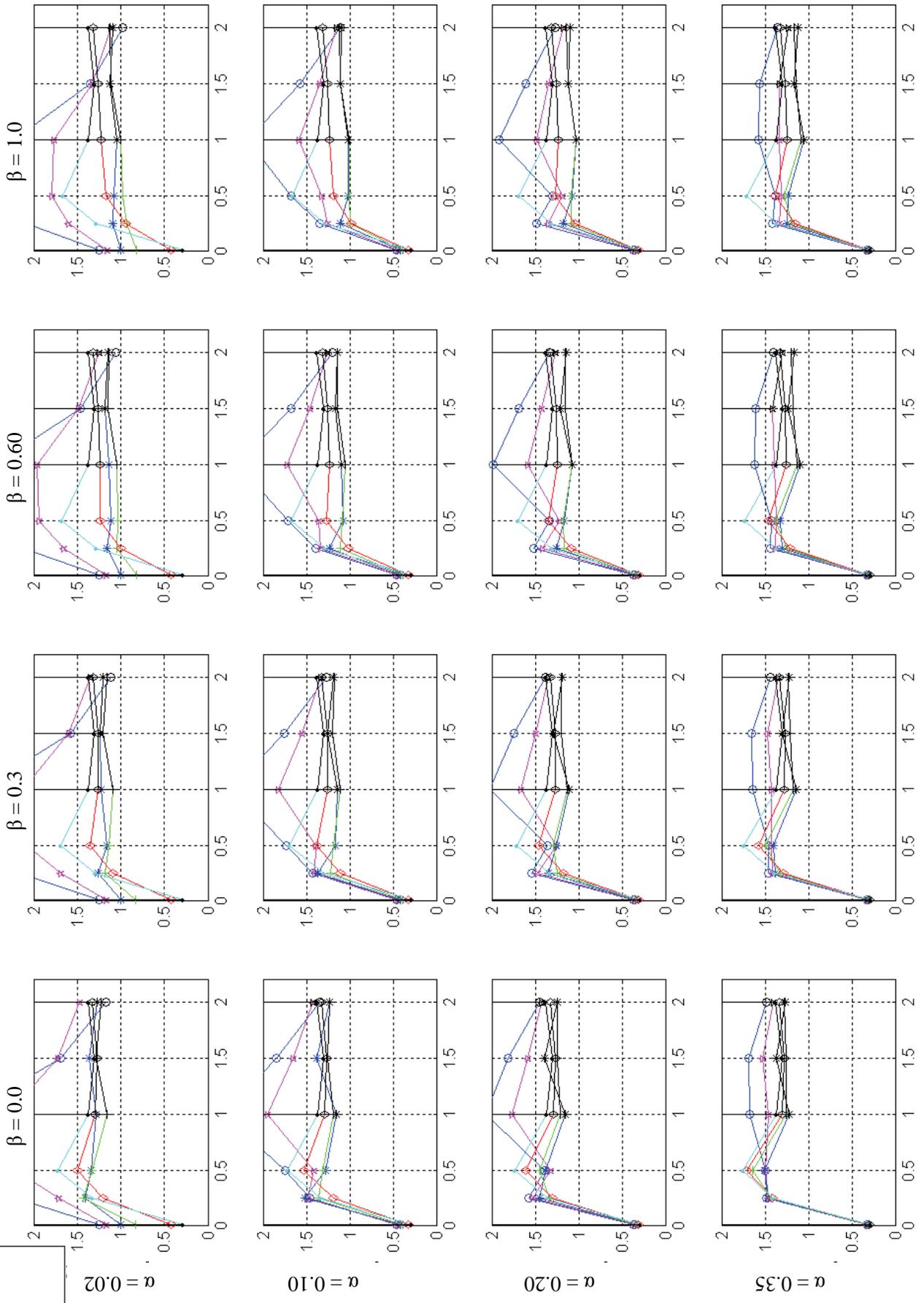
### Relative Performance (0.7\* Disp.Ratio + 0.3\* Accel.Ratio)



$\Psi$  ( $\eta_{\text{EPS}} = 0.20; \psi = \eta_{\text{SCS}} / \eta_{\text{EPS}}$ )



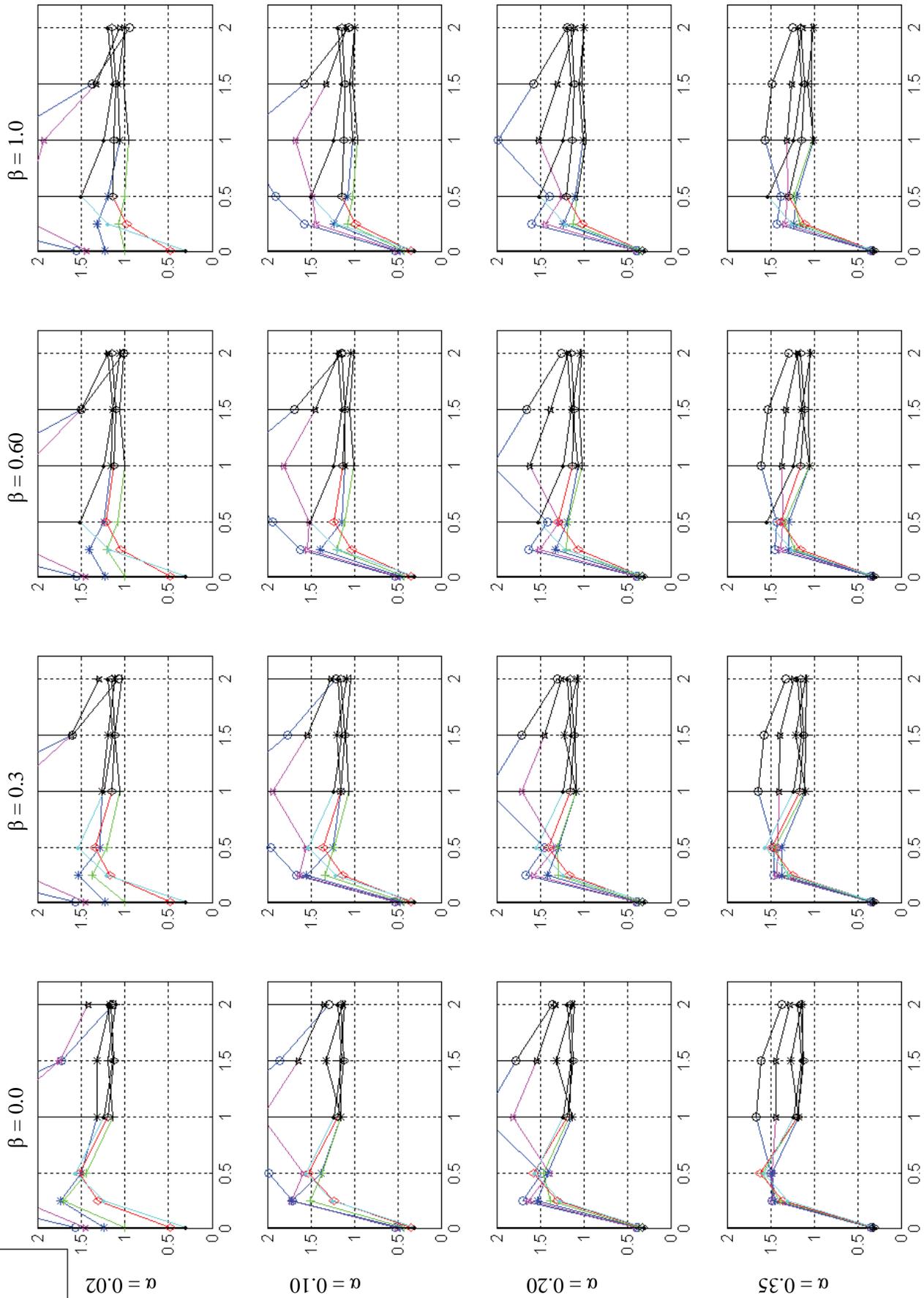
**Relative Performance (0.7\* Disp.Ratio + 0.3\* Accel.Ratio)**



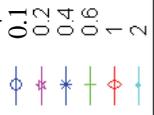
$\Psi$  ( $\eta_{EPS} = 0.30; \psi = \eta_{SCS} / \eta_{EPS}$ )

- 0.17
- \* 0.33
- ✱ 0.67
- ✱ 1
- ✱ 1.67
- ✱ 3.33

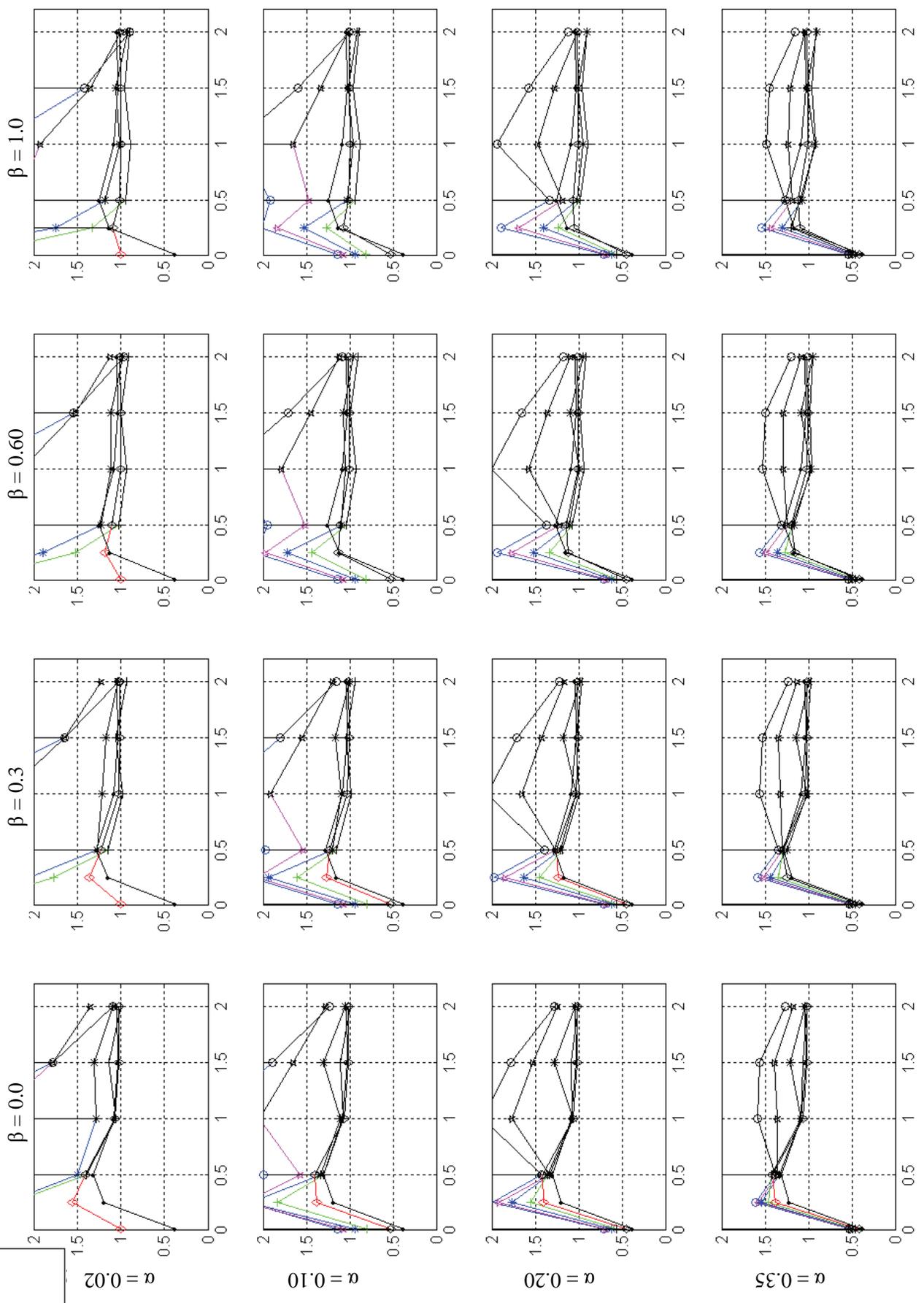
### Relative Performance (0.7\* Disp.Ratio + 0.3\* Accel.Ratio)



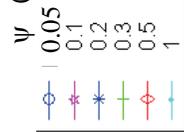
$\Psi$  ( $\eta_{EPS} = 0.50; \psi = \eta_{SCS} / \eta_{EPS}$ )



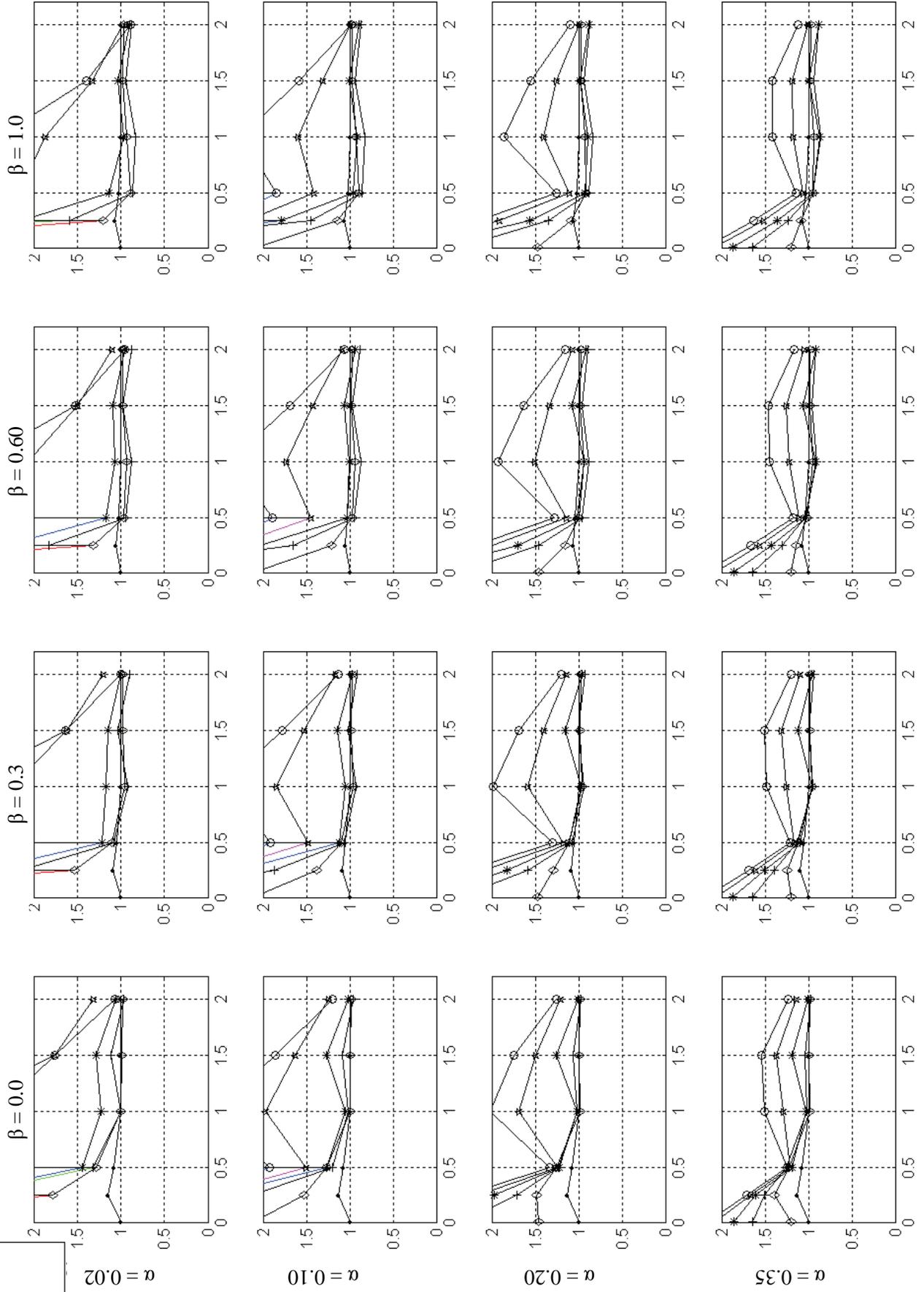
### Relative Performance (0.7\* Disp.Ratio + 0.3\* Accel.Ratio)



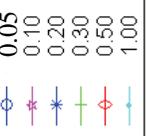
$\Psi$  ( $\eta_{EPS} = 1.00; \psi = \eta_{SCS} / \eta_{EPS}$ )



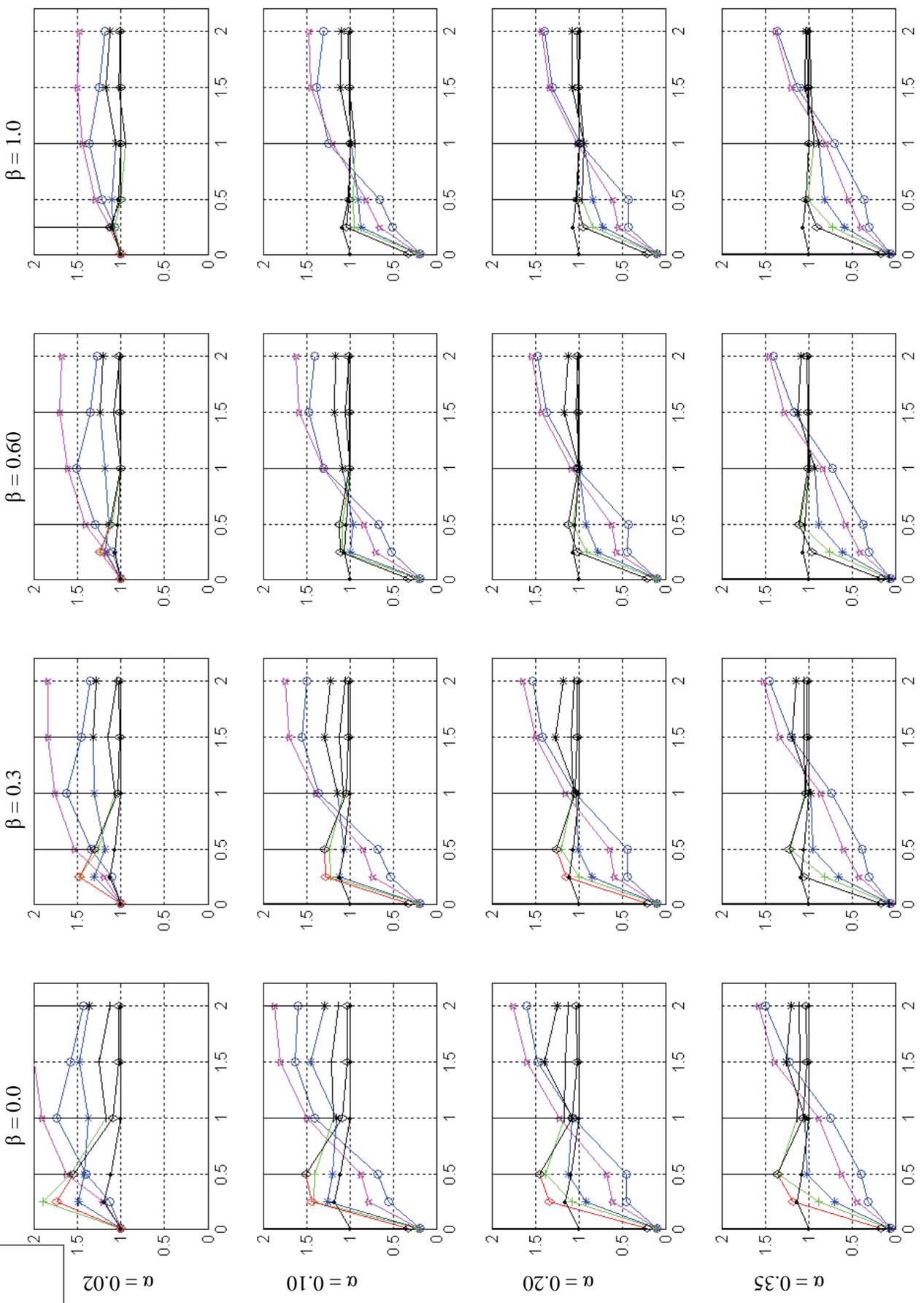
### Relative Performance (0.7\* Disp.Ratio + 0.3\* Accel.Ratio)



$\eta_{SCS}$   
 $(\psi = \eta_{SCS} / \eta_{EPS} = 1)$



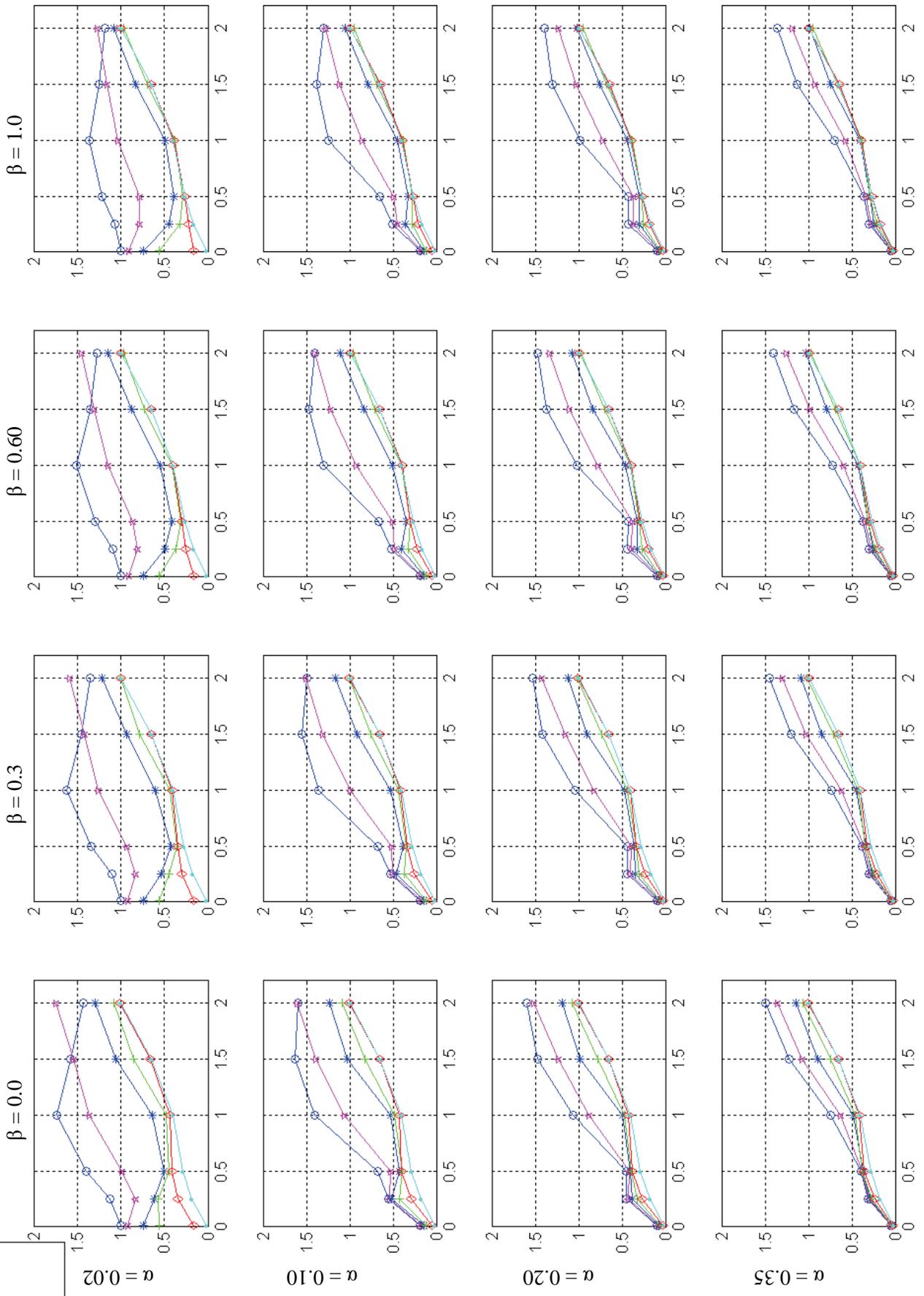
**Relative Performance (1\* Disp.Ratio + 0\* Accel.Ratio)**



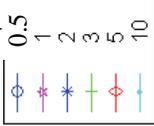
$\Psi$  ( $\eta_{EPS} = 0.05; \Psi = \eta_{SCS} / \eta_{EPS}$ )



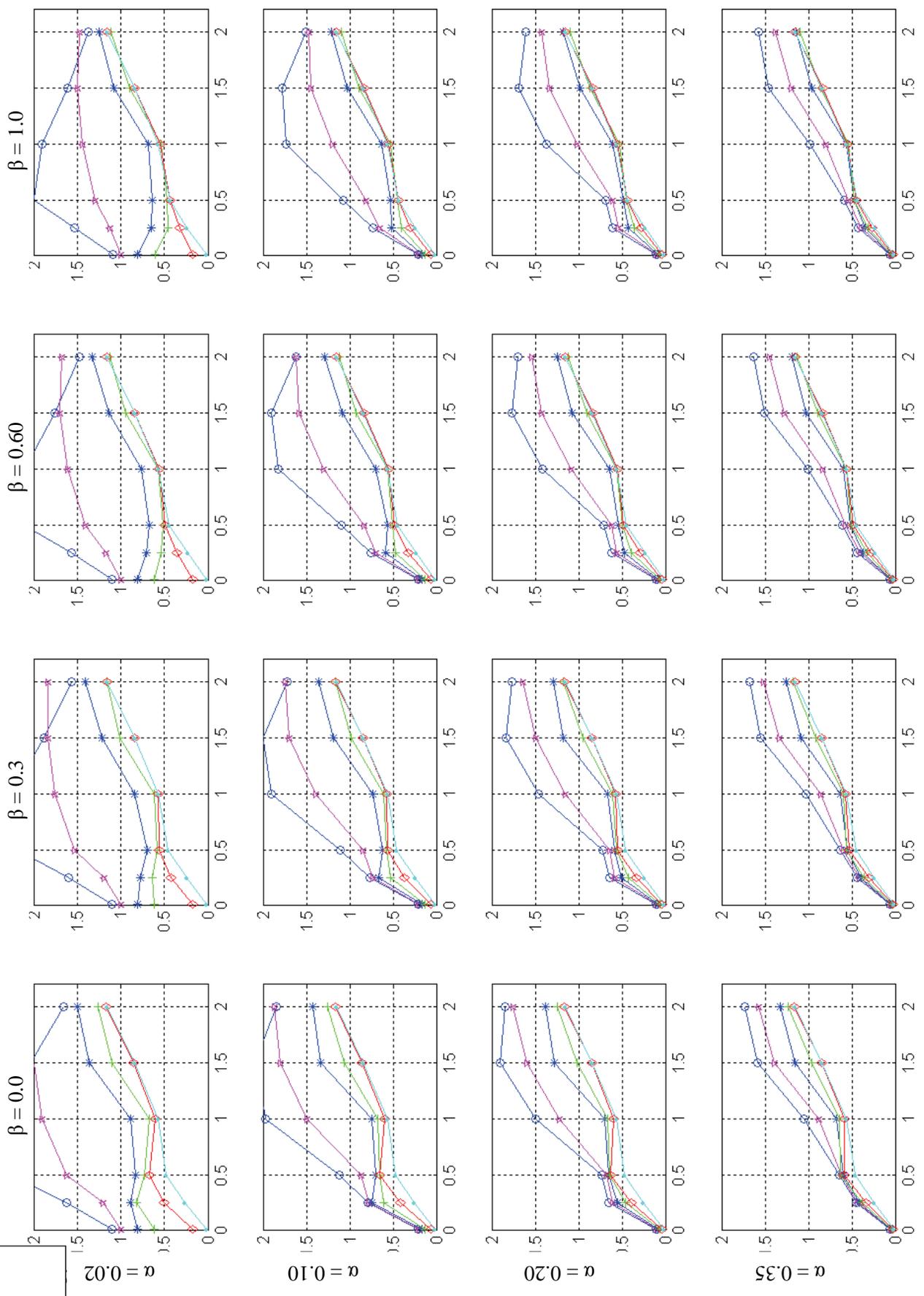
**Relative Performance (1\* Disp.Ratio + 0\* Accel.Ratio)**



$\psi$  ( $\eta_{EPS} = 0.10; \psi = \eta_{SCS} / \eta_{EPS}$ )



**Relative Performance (1\* Disp.Ratio + 0\* Accel.Ratio)**



$\Psi$  ( $\eta_{EPS} = 0.20$ ;  $\Psi = \eta_{SCS} / \eta_{EPS}$ )

0.25

0.5

1

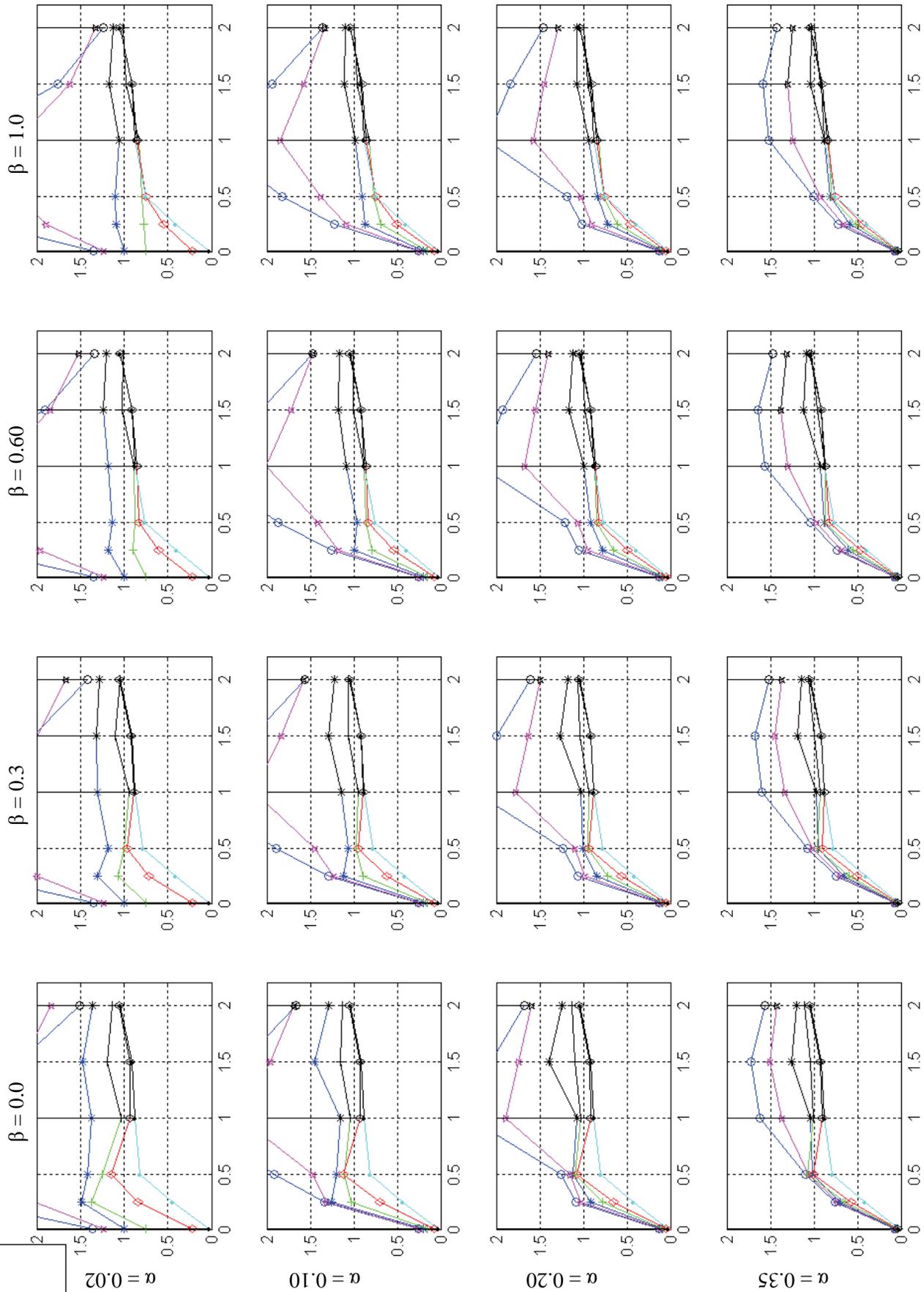
1.5

2.5

5



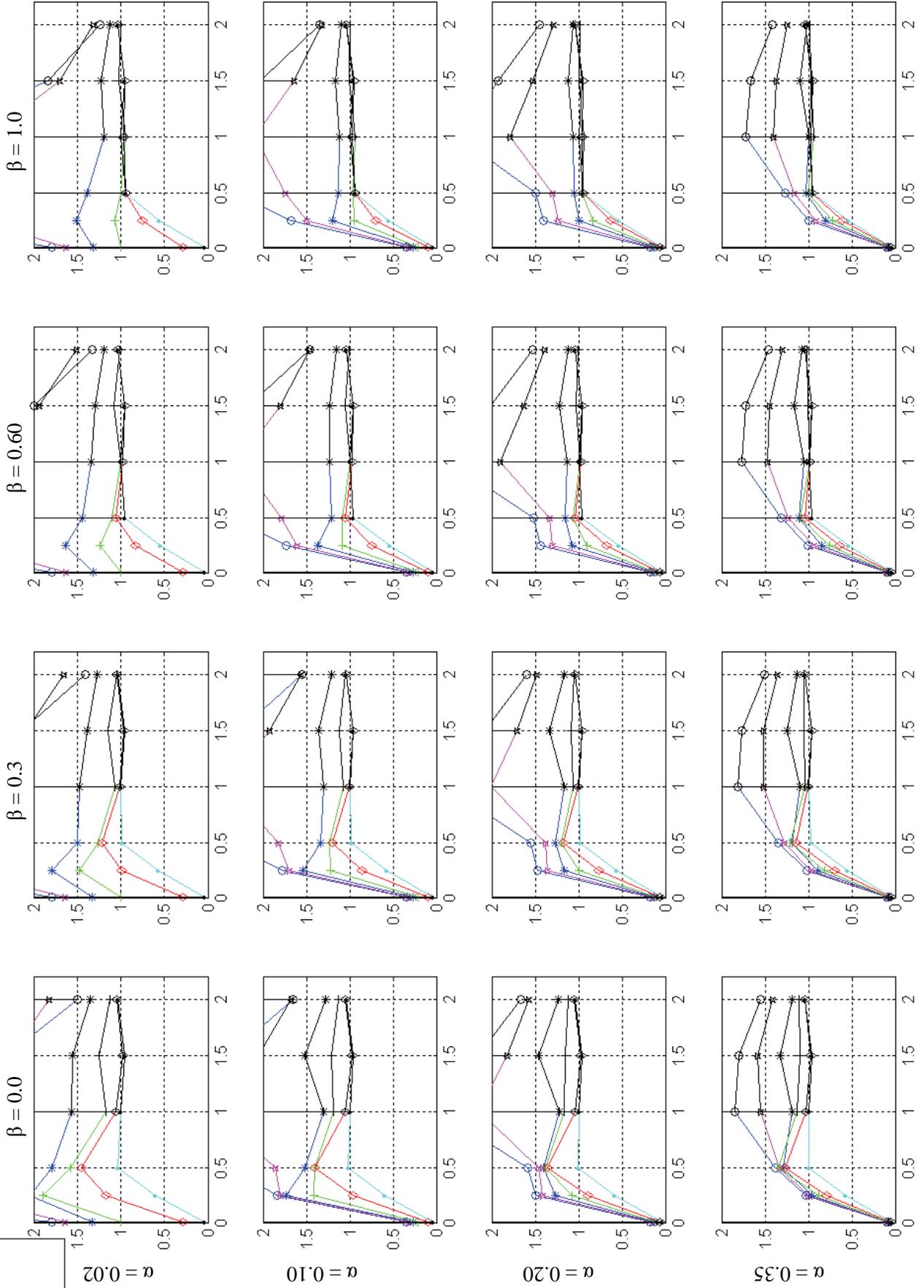
### Relative Performance (1\* Disp.Ratio + 0\* Accel.Ratio)



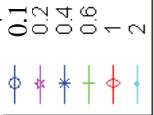
$\Psi$  ( $\eta_{EPS} = 0.30; \Psi = \eta_{SCS} / \eta_{EPS}$ )

- 0.17
- ★ 0.33
- ✱ 0.67
- ✱ 1
- ◇ 1.67
- ⊖ 3.33

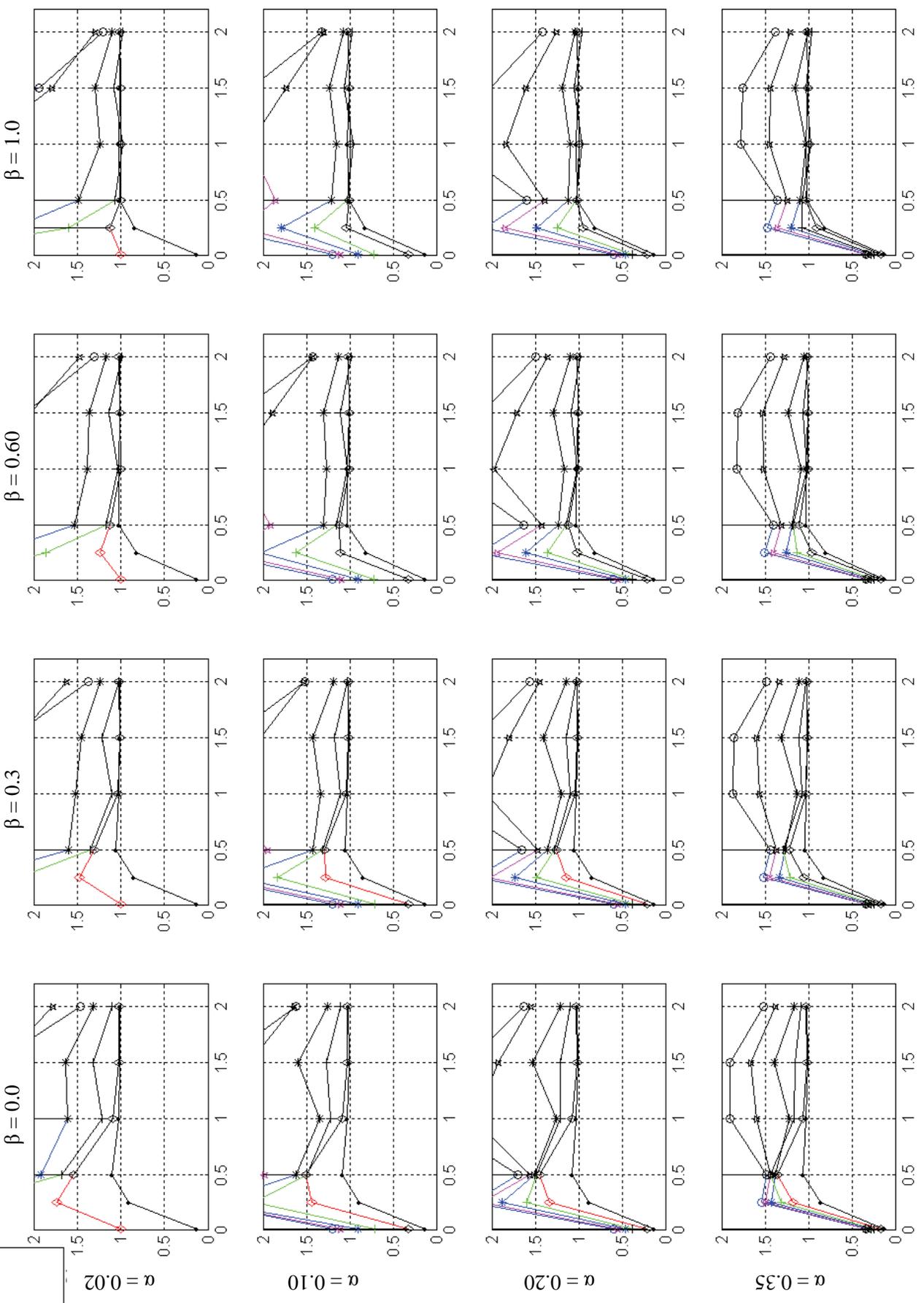
### Relative Performance (1\* Disp.Ratio + 0\* Accel.Ratio)



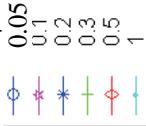
$\Psi$  ( $\eta_{EPS} = 0.50; \psi = \eta_{scs} / \eta_{EPS}$ )



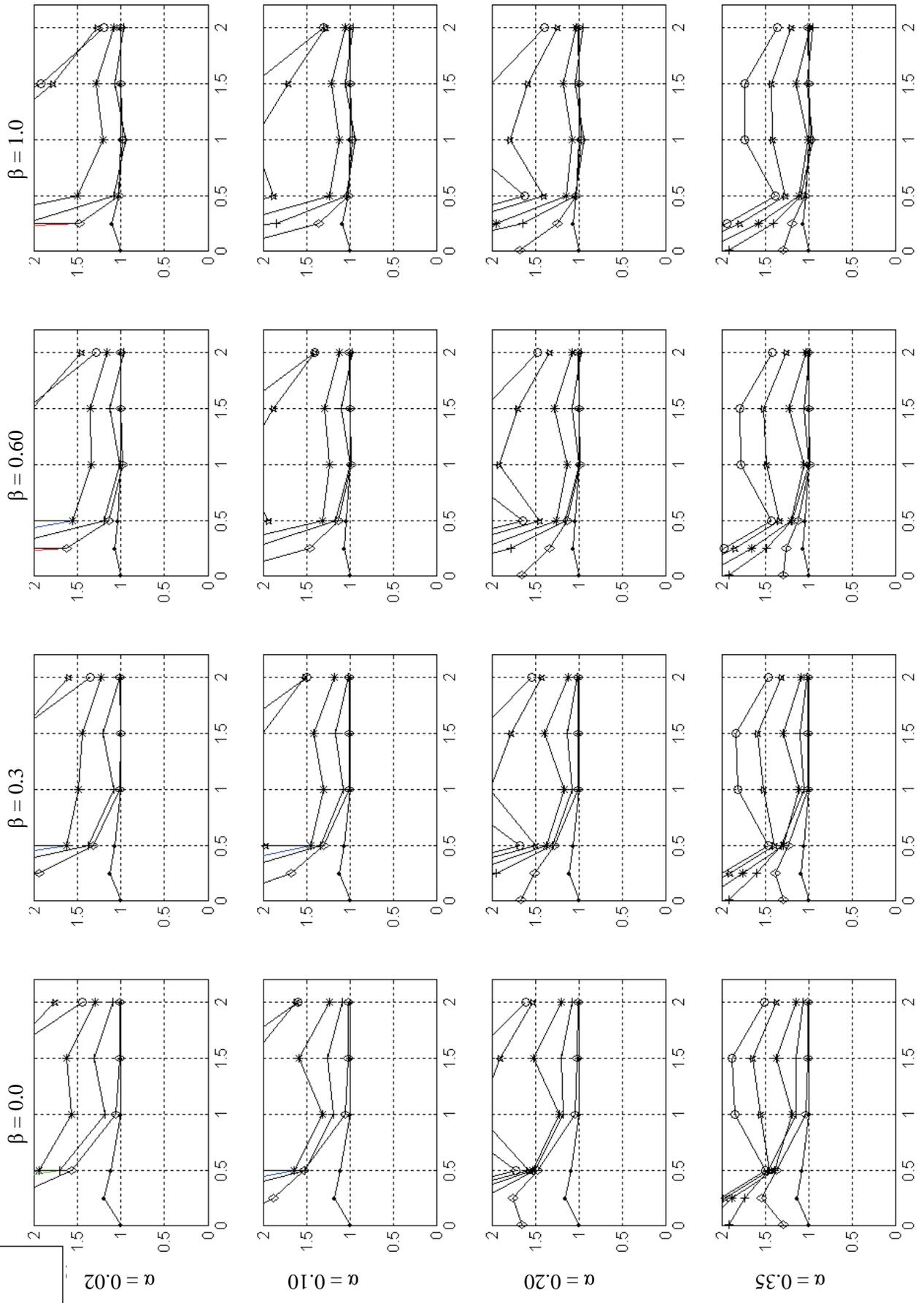
**Relative Performance (1\* Disp.Ratio + 0\* Accel.Ratio)**



$\Psi$  ( $\eta_{\text{EPS}} = 1.00; \psi = \eta_{\text{scs}} / \eta_{\text{EPS}}$ )

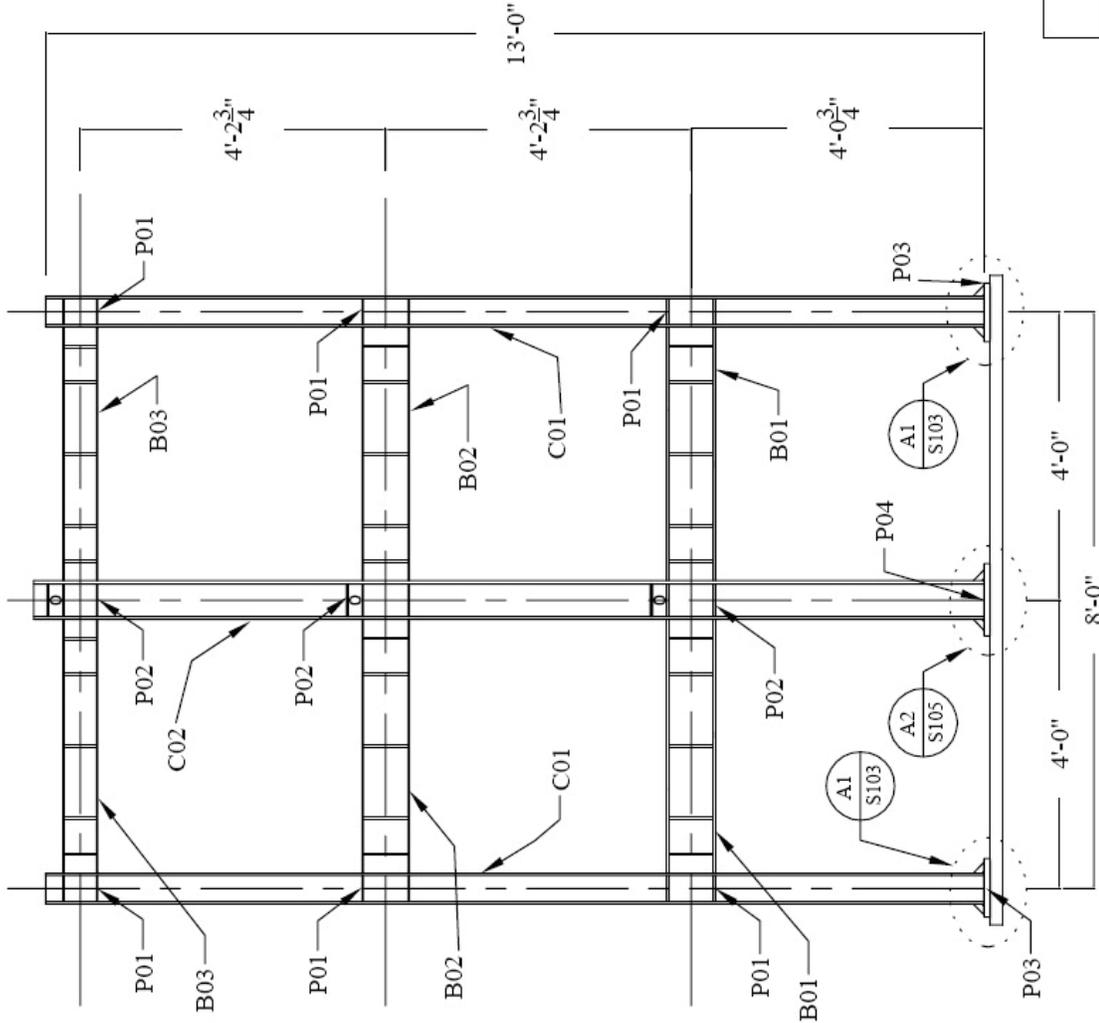


**Relative Performance (1\* Disp.Ratio + 0\* Accel.Ratio)**



## **Appendix C Model of Steel Moment Resisting Frame for Shake Table Testing**

- General Note:**
1. All steels are ASTM A572 Grade 50 unless specified.
  2. All weld material shall be E70 Electrode unless specified.

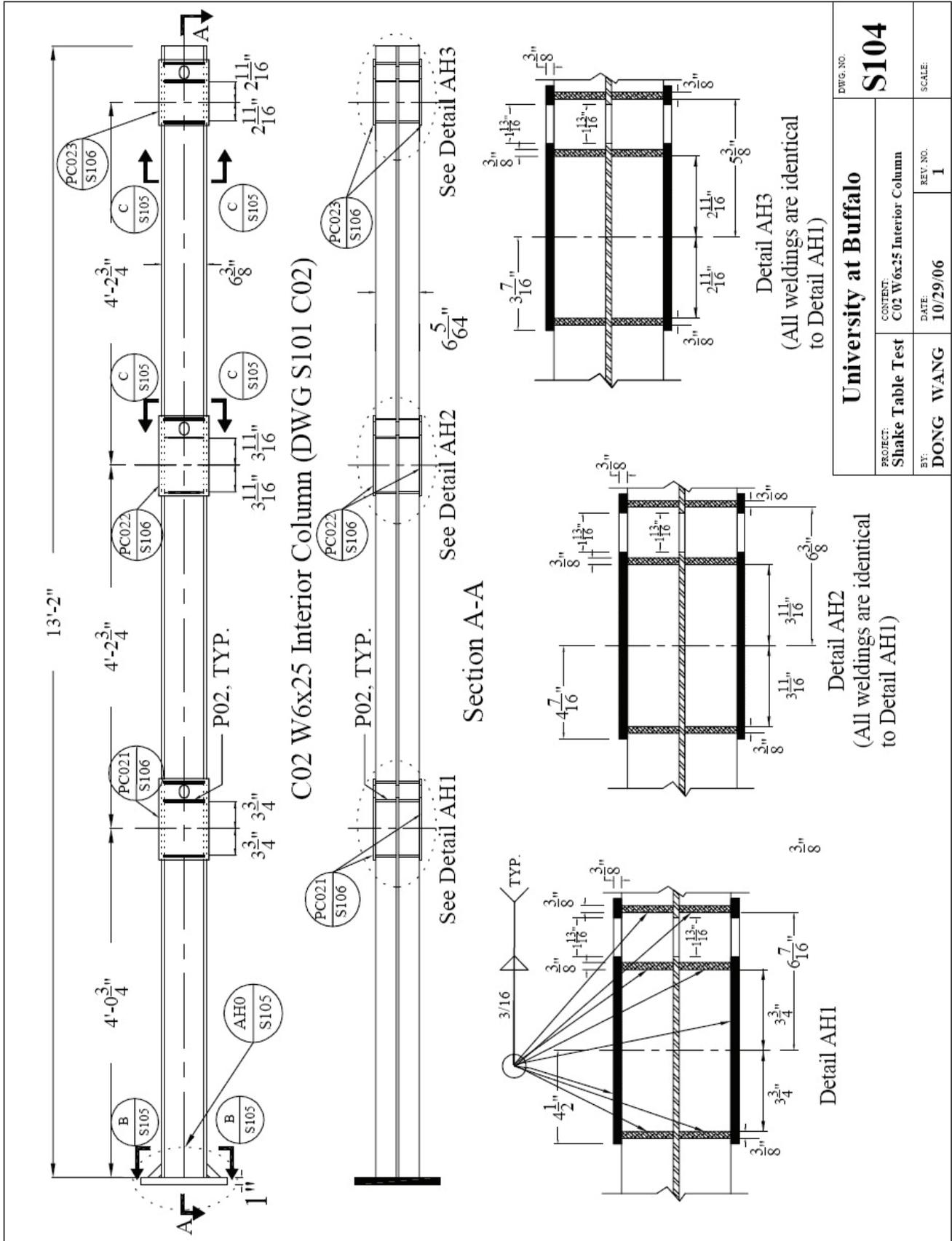


Part Code	Part Description	QTY
C01	Exterior Column, 13-ft length W5x19	2
C02	Interior Column, 13-ft 2-in length W6x25	1
B01	Beam in 1 <sup>st</sup> floor, 3ft 6 1/4 in W8x13	2
B02	Beam in 2 <sup>nd</sup> floor, 3ft 6 1/4 in W8x10	2
B03	Beam in 3 <sup>rd</sup> floor, 3ft 6 1/4 in W6x8.5	2
P01	3/8 in thick Continuity Plate 4 9/32 x 2 3/8 in	24
P02	3/8 in thick Continuity Plate 5 15/32 x 2 7/8 in	12
P03	1 in thick Exterior Base Plate 12 x 9 3/4	2
P04	1 in thick Interior Base Plate 12 x 12	1

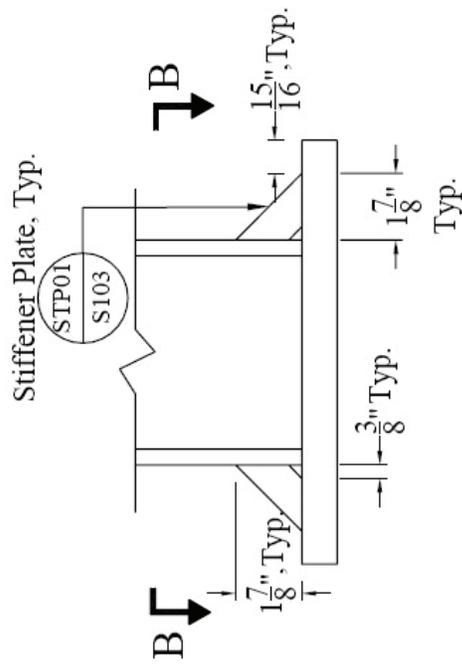
<b>University at Buffalo</b>		DWG. NO.
PROJECT: Shake Table Test		<b>S101</b>
BY: DONG WANG		SCALE
DATE: 9/25/06		REV. NO.
CONTENT: SMRF Model		<b>1</b>







<b>University at Buffalo</b> PROJECT: <b>Shake Table Test</b>		DWG. NO. <b>S104</b>	
CONTENT: <b>C02 W6x25 Interior Column</b>		SCALE:	
BY: <b>DONG WANG</b>	DATE: <b>10/29/06</b>	REV. NO. <b>1</b>	



Detail AH0 in DWG S104  
(Interior Column w6x25 base  
connection with stiffener plates)

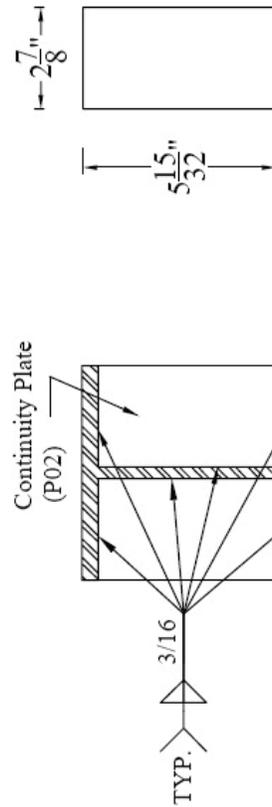
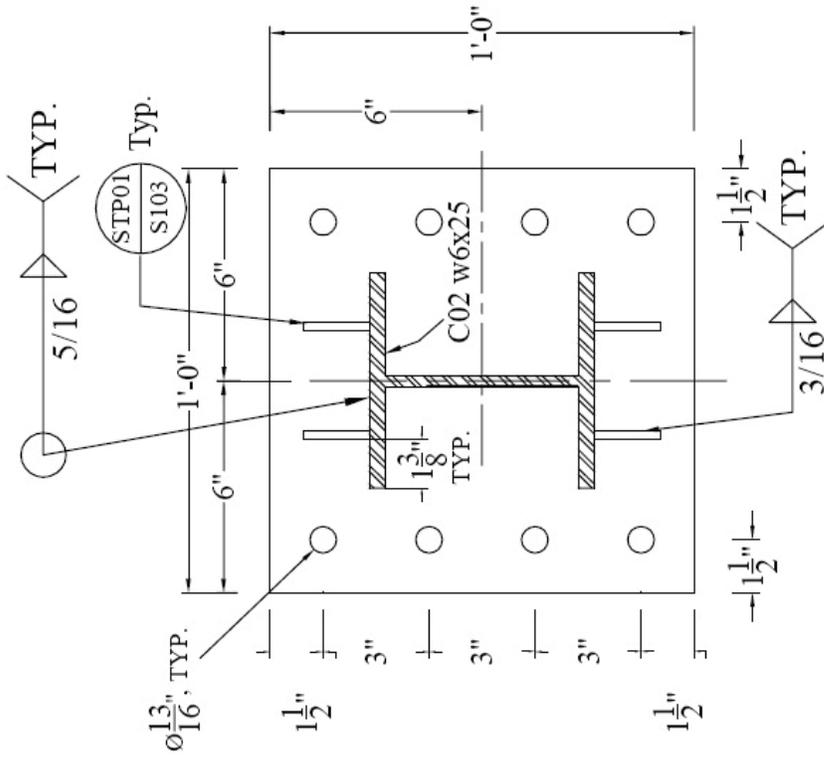


Plate Thk. = 3/8"

P02 Continuity Plate  
(DWG S101 & S104 P02)



Section B-B Base Plate  
(DWG S104 Section B-B,  
DWG S101 Detail A2)

DWG S104 Section C-C

<b>University at Buffalo</b>		DWG. NO.	<b>S105</b>
		PROJECT: Shake Table Test	
BY: <b>DONG WANG</b>	DATE: 10/29/06	REV. NO. <b>1</b>	SCALE

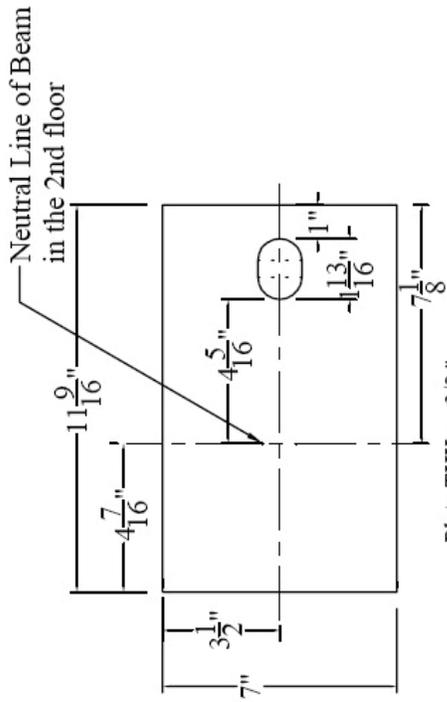


Plate THK. = 3/8 "

PC022 Web Reinforcing Plate (2pc)  
(DWG S104 PC022)

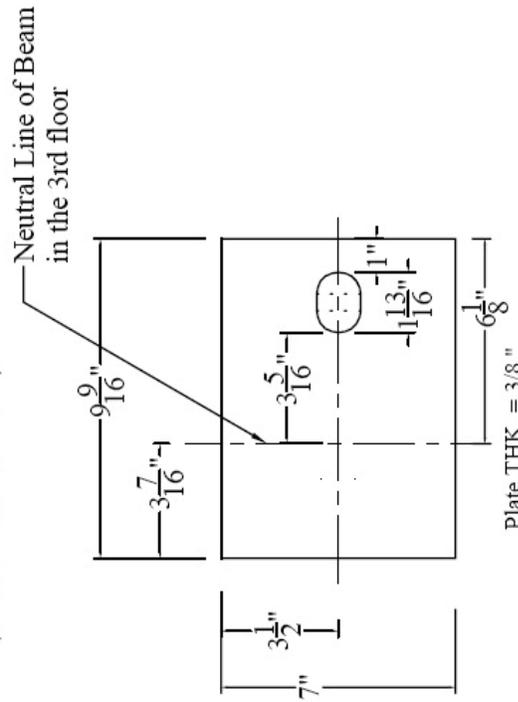
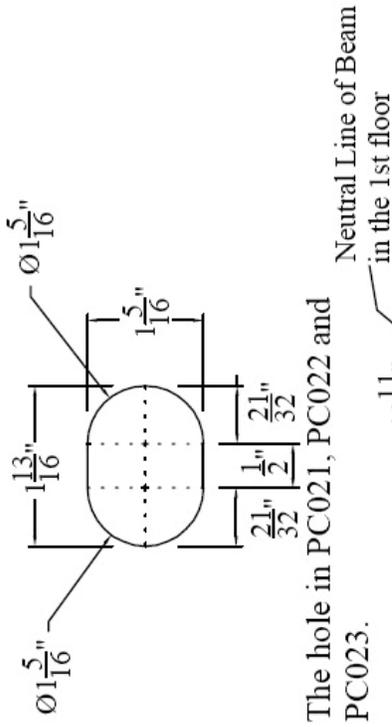


Plate THK. = 3/8 "

PC023 Web Reinforcing Plate (2pc)  
(DWG S104 PC023)



The hole in PC021, PC022 and PC023.

Neutral Line of Beam  
in the 1st floor

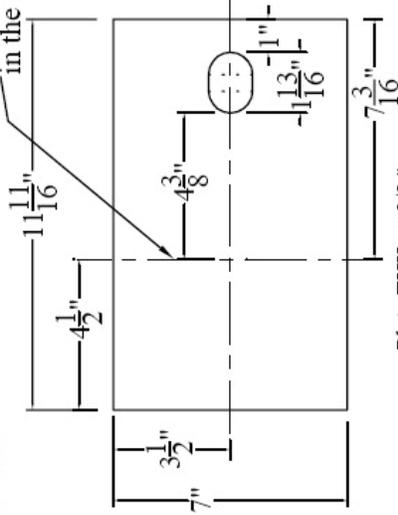
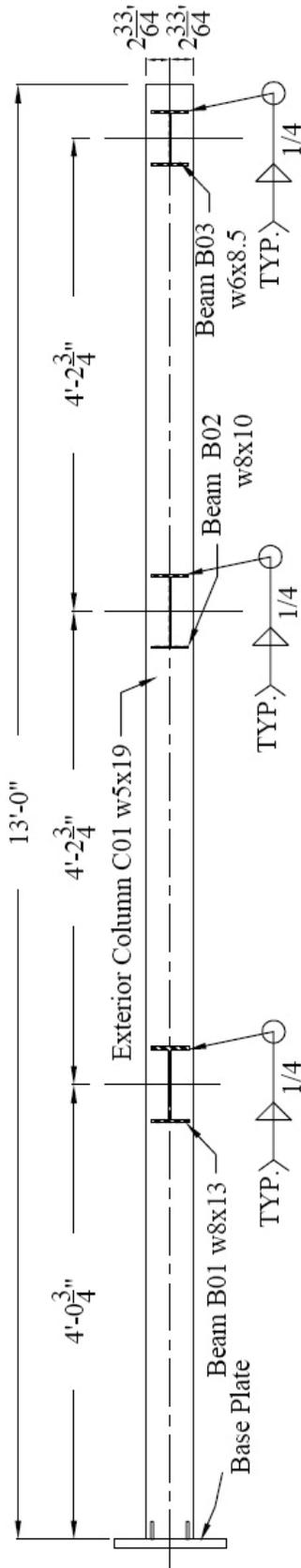


Plate THK. = 3/8 "

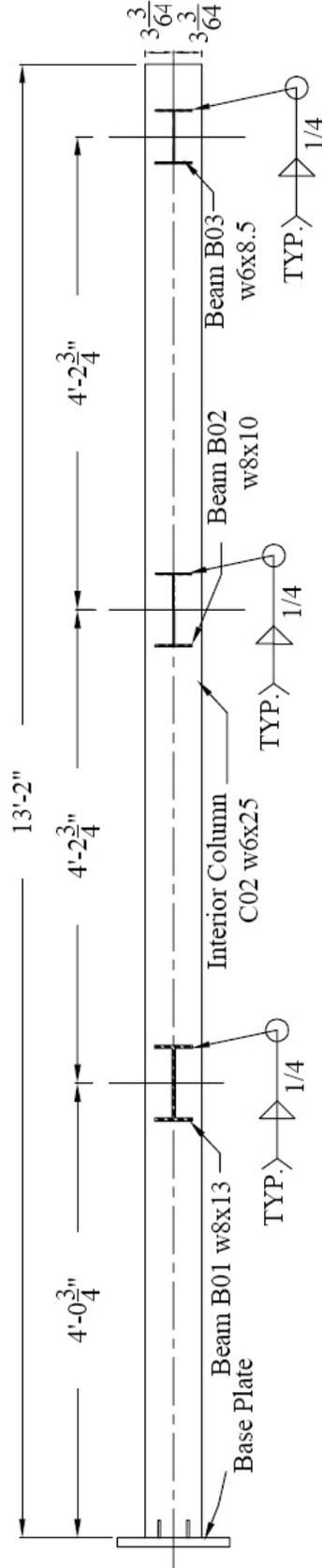
PC021 Web Reinforcing Plate (2pc)  
(DWG S104 PC021.)

**Note: The corresponding part of column (C02) web should be machined with the same holes as shown in these plates.**

<b>University at Buffalo</b>		DWG. NO. <b>S106</b>
PROJECT: Shake Table Test	CONTENT: Plate PC021, PC022 and PC023 in DWG S104.	SCALE:
BY: DONG WANG	DATE: 9/25/06	REV. NO. 1



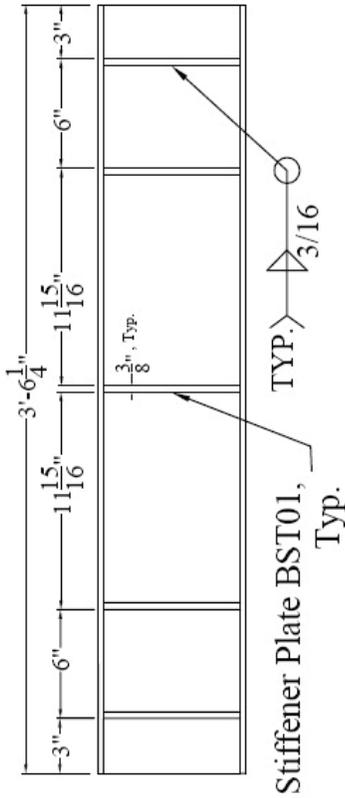
Exterior Beam-Column Connections



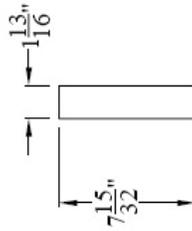
Interior Beam-Column Connections

<b>University at Buffalo</b>		DWG. NO. <b>S107</b>
PROJECT: <b>Shake Table Test</b>	CONTENT: <b>Beam-Column Connections</b>	
BY: <b>DONG WANG</b>	DATE: <b>10/29/06</b>	REV. NO. <b>2</b>
		SCALE:

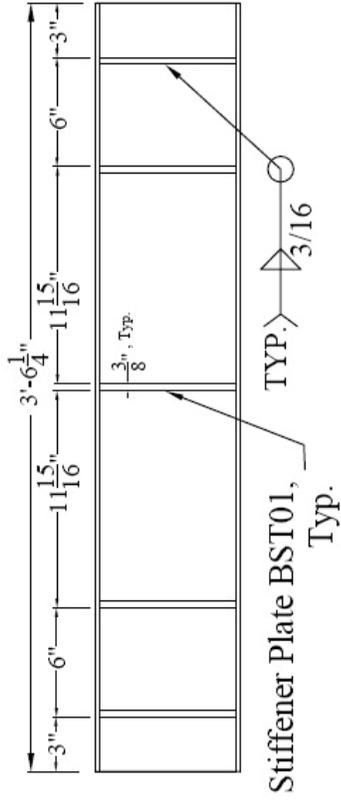
# Beams



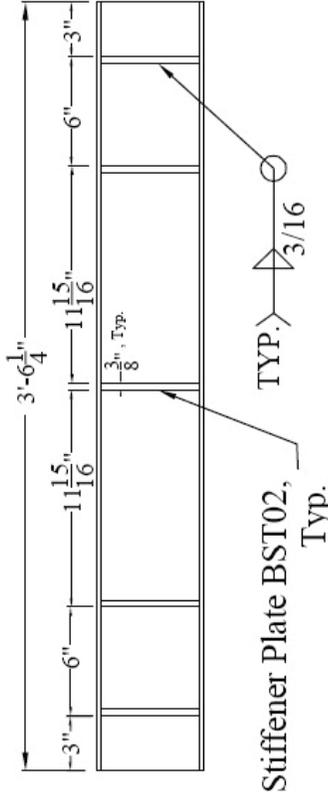
DWG S101 B01 w8x13 (2pc)



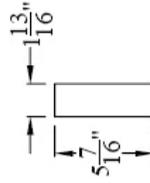
$\frac{3}{8}$ " Thk.  
Stiffener Plate BST01  
(Total 40 pc, all are welded to the beams in the 1st and 2nd floor.)



DWG S101 B02 w8x10 (2pc)



DWG S101 B03 w6x8.5 (2 pc)



$\frac{3}{8}$ " Thk.  
Stiffener Plate BST02  
(Total 20 pc, all are welded to the beams in the 3rd floor.)

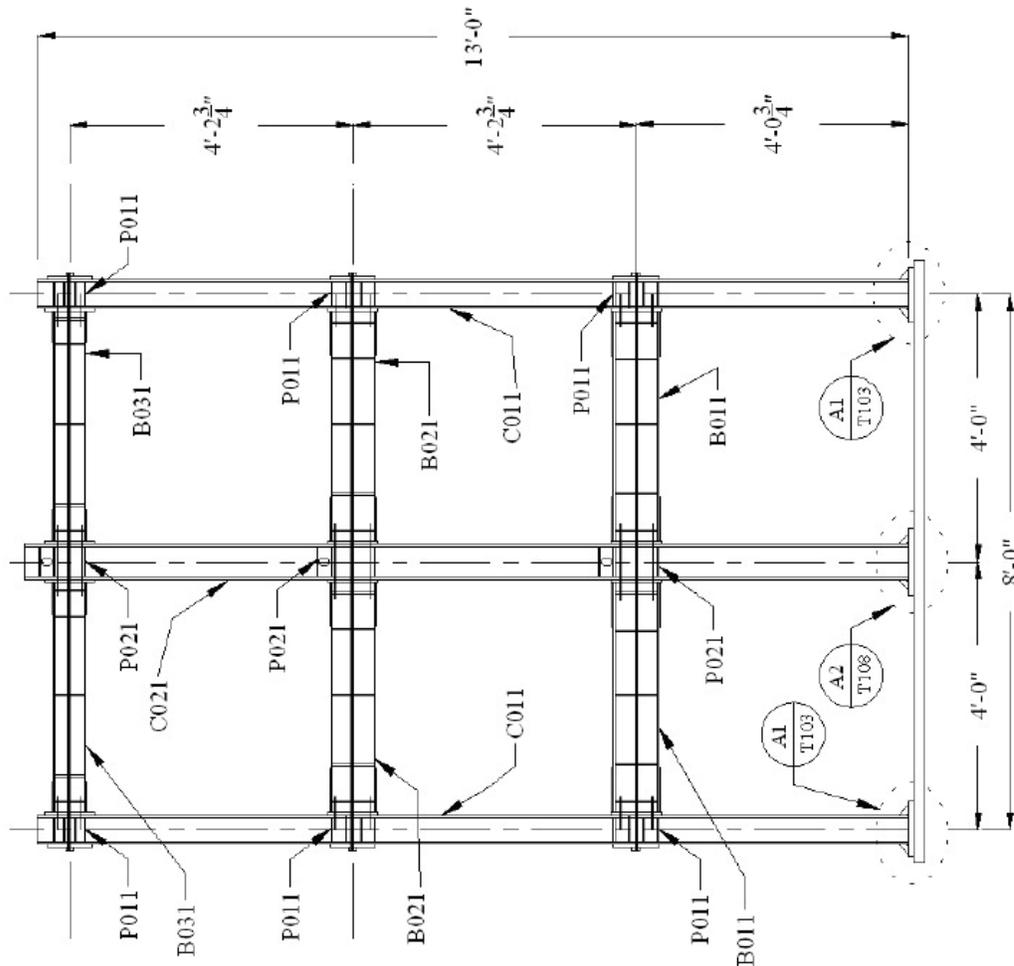
<b>University at Buffalo</b>		DWG. NO. <b>S108</b>
PROJECT: <b>Shake Table Test</b>	CONTENT: <b>Beams</b>	
BY: <b>DONG WANG</b>	DATE: <b>10/29/06</b>	REV. NO. <b>1</b>
		SCALE:



**Appendix D Model of Self-Centering Post-Tensioned Frame  
for Shake Table Testing**

**General Note:**

1. All steels are ASTM A572 Grade 50 unless specified.
2. All weld material shall be E70 Electrode unless specified.
3. Do not weld beams to columns in this SCPT model

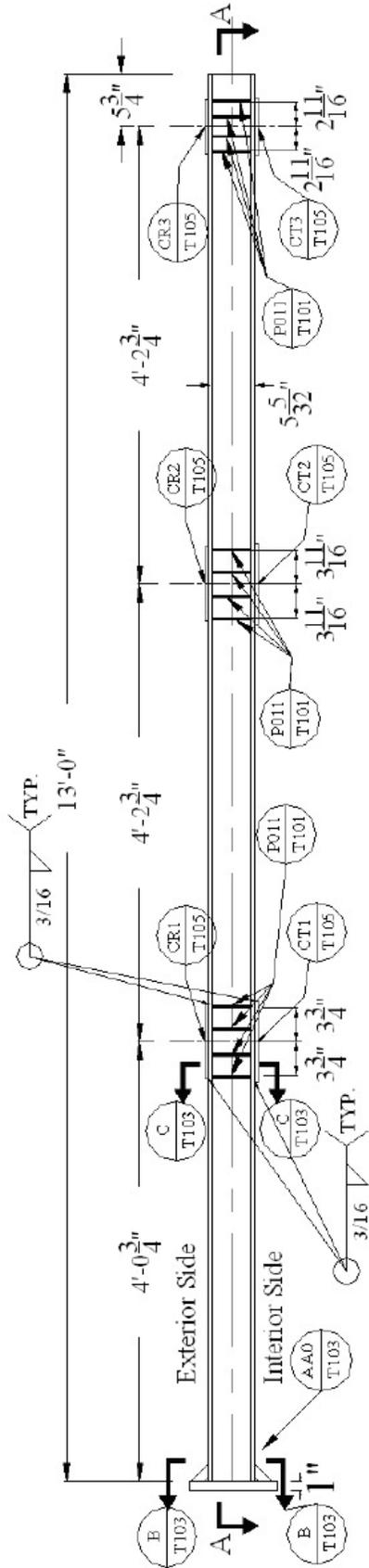


Part Code	Part Description	QTY
C011	Exterior Column, 13-ft length W5x19	2
C021	Interior Column, 13ft 2in length W6x25	1
B011	Beam in 1 <sup>st</sup> floor, 3ft 5 1/2 in W8x13	2
B021	Beam in 2 <sup>nd</sup> floor, 3ft 5 1/2 in W8x10	2
B031	Beam in 3 <sup>rd</sup> floor, 3ft 5 1/2 in W6x8.5	2
P011	3/8 in. thick Continuity Plate 4 9/32 X 2 3/8 in.	48
P021	3/8 in. thick Continuity Plate 5 15/32 X 2 7/8 in.	18

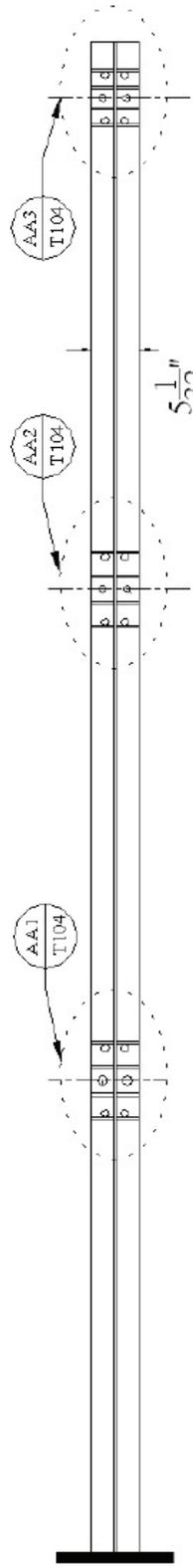
**PTED FRAME MODEL**

<b>University at Buffalo</b>		DWG. NO.
PROJECT Shake Table Test	CONTENT <b>PTED FRAME MODEL</b>	T101
BY: <b>Dong Wang</b>	REV. NO. 1	Date 9/25/2006
		SCALE

Please note that plate CR1, CR2 and CR3 are in the exterior side of the column and plates CT1, CT2 and CT3 are in the interior side of the column.

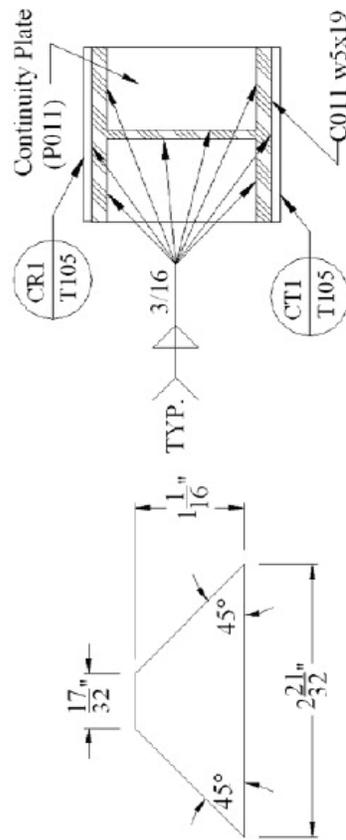
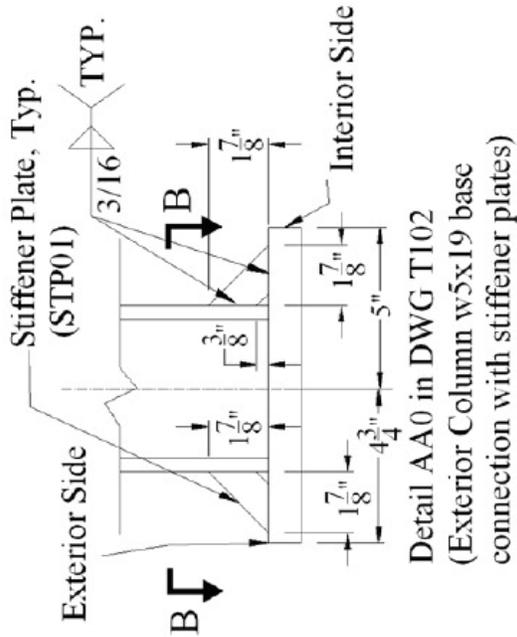


C011 W5x19 Exterior Column (DWG T101 C011)

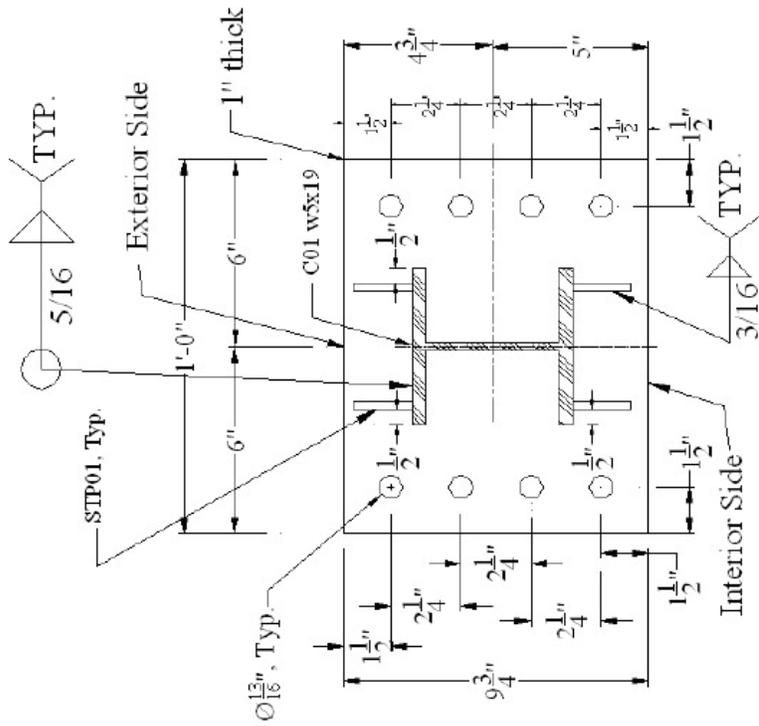


Section A-A

<b>University at Buffalo</b>		DWG. NO.	
PROJECT Shake Table Test	CONTENT C011 W5x19 Exterior Column	T102	
BY: <b>Dong Wang</b>	REV. NO. 1	Date 9/25/2006	SCALE



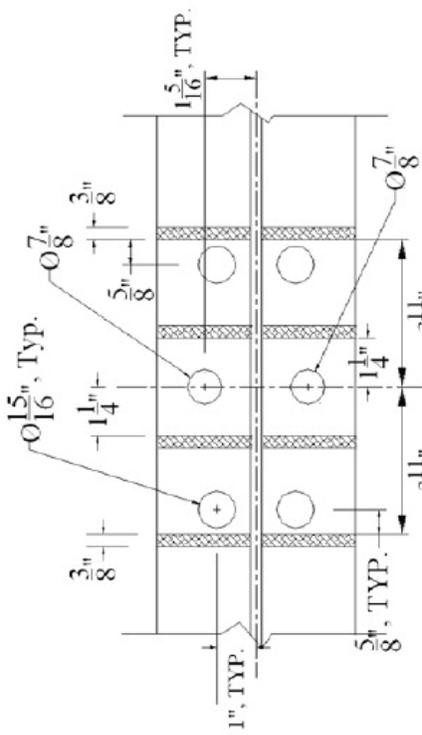
STP01 Stiffener Plate  
(Total 12 pc. Each base-column connection has 4 pc.)



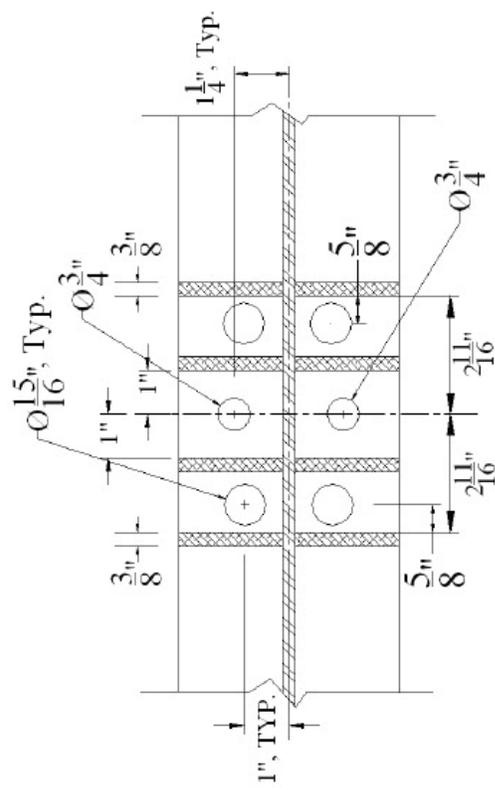
Section B-B Base Connection  
(DWG T102 Section B-B,  
DWG T101 Detail A1)

<b>University at Buffalo</b>		DWG. NO.	
		T103	
PROJECT Shake Table Test	CONTENT Section B-B, C-C in DWG T102	REV. NO.	DATE
BY: <b>Dong Wang</b>	1	10/29/2006	SCALE

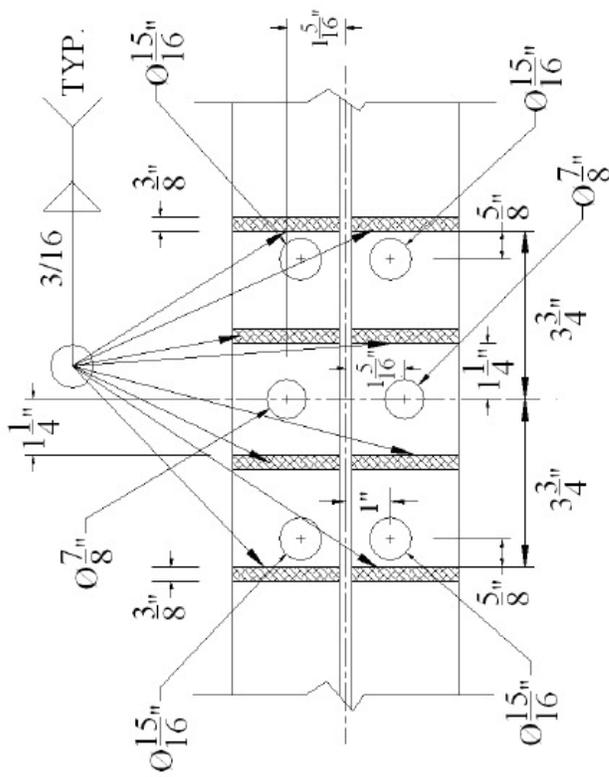
**Note: Do not weld the area near between the continuity plates (shade parts) and holes.**



**DWG T102 Detail AA2**  
(All welds are identical to those of Detail AA1.)



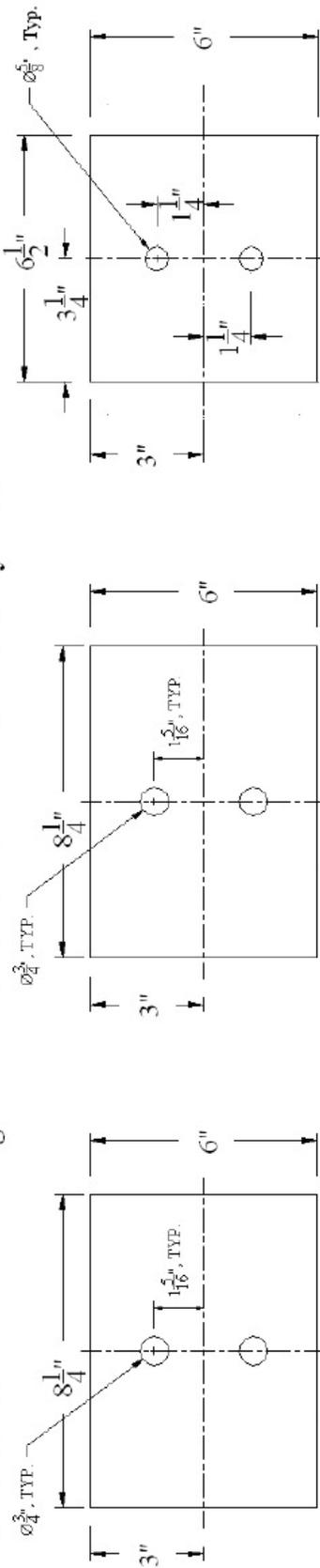
**DWG T102 Detail AA3**  
(All welds are identical to those of Detail AA1.)



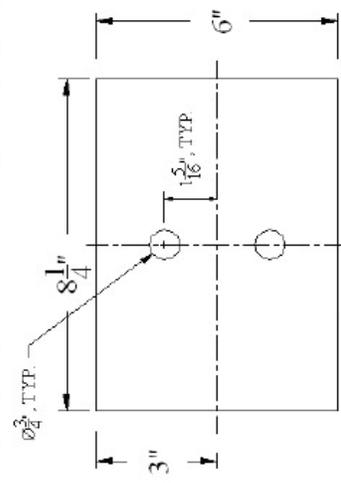
**DWG T102 Detail AA1**

<b>University at Buffalo</b>		DWG. NO.	
PROJECT Shake Table Test	CONTENT Details AA1-3 in DWG T102	T104	
BY: <b>Dong Wang</b>	REV. NO. 1	Date 9/25/2006	SCALE

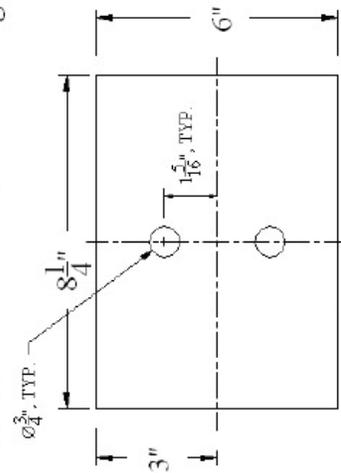
Note: All materials in this figure are made from ASTM A514 with  $F_y=100\text{ksi}$ .



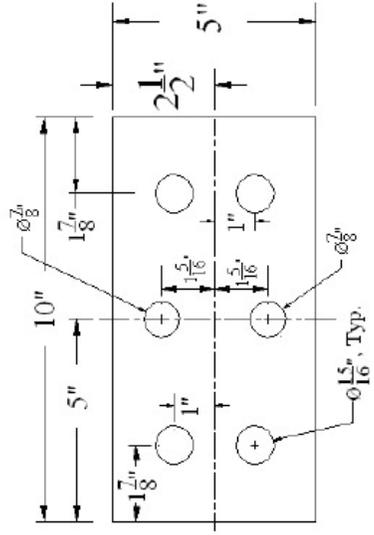
CR1 Reinforcing Plate (2pc)  
Plate Thk.=1/2"



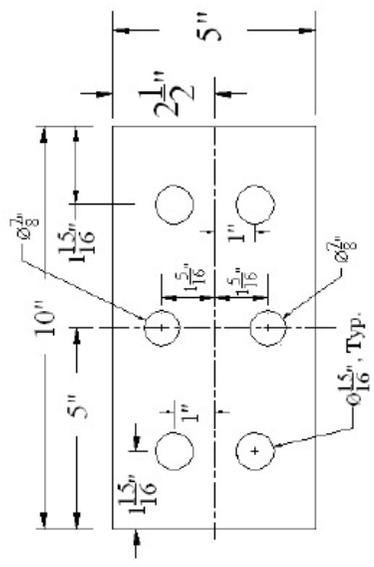
CR2 Reinforcing Plate (2pc)  
Plate Thk.=1/2"



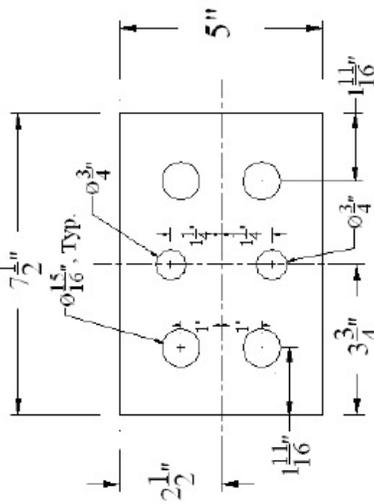
CR3 Reinforcing Plate (2pc)  
Plate Thk.=1/2"



CT1 Contact Plate (4pc)  
Plate Thk.=3/8"



CT2 Contact Plate (4pc)  
Plate Thk.=3/8"

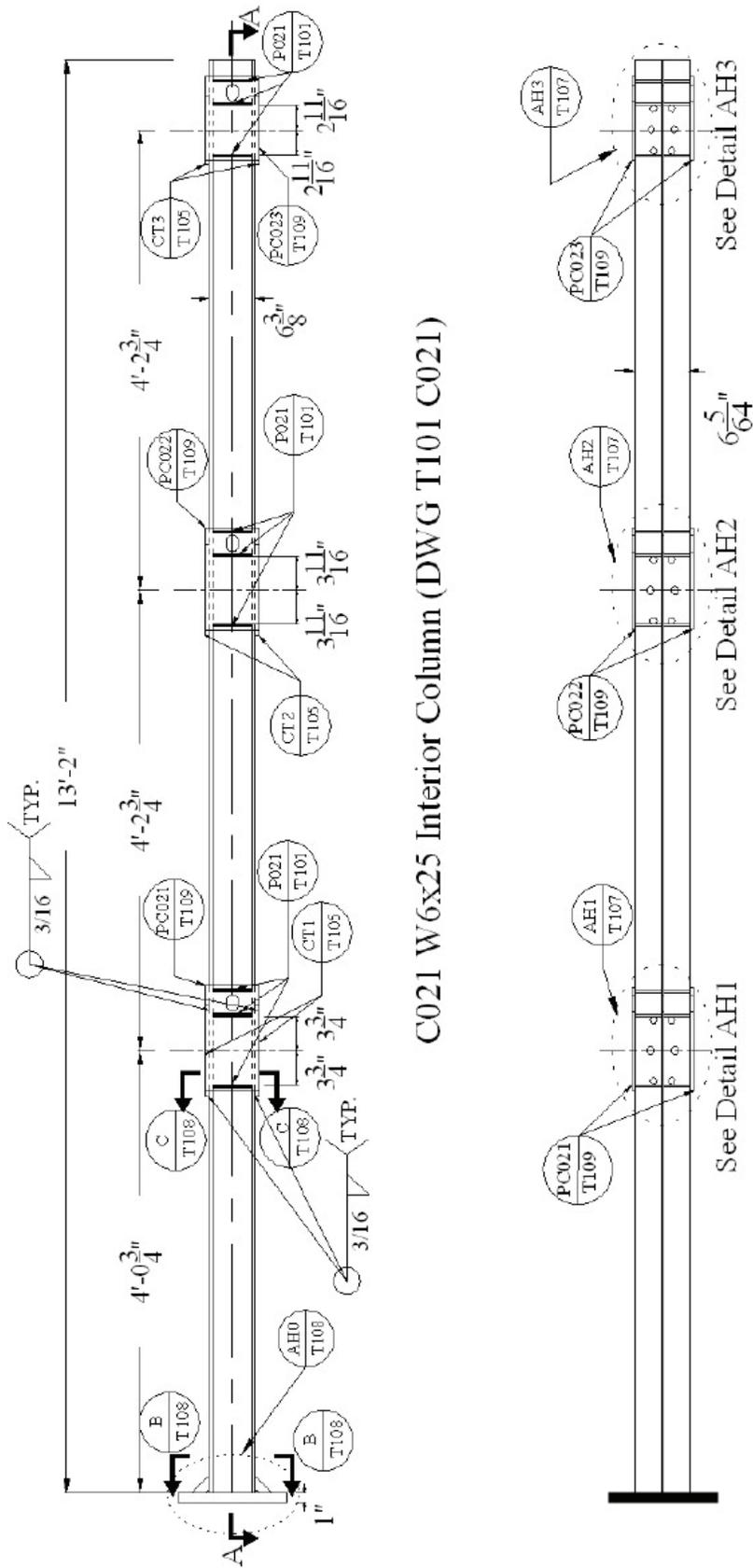


CT3 Contact Plate (4pc)  
Plate Thk.=3/8"

Note: The corresponding part of columns (C011 & C021) flanges should be machined with the same holes as shown in these plates.

<b>University at Buffalo</b>		DWG. NO.	
PROJECT Shake Table Test	CONTENT Plates in DWG T102	T105	
BY: <b>Dong Wang</b>	REV. NO. 1	Date 10/11/2006	SCALE

Note: As shown in the figure of C021 interior column, the contact plate CT1, CT2 and CT3 are between the end of beam and column flange and welded (3/16" welding) by 4 sides to the flange of the column. The contact plates are perpendicular to the web reinforcing plates (PC021, 022 and 023).

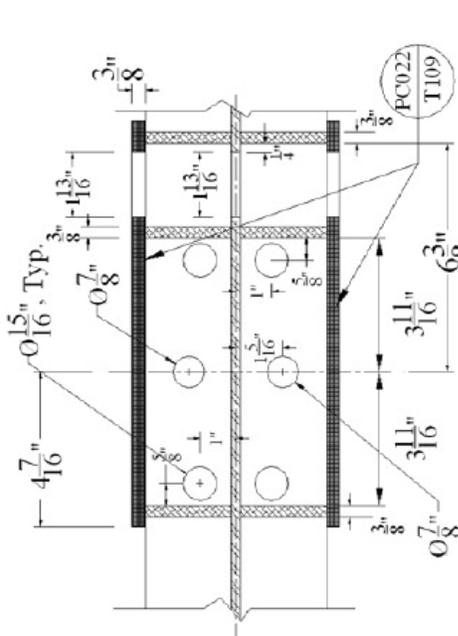


C021 W6x25 Interior Column (DWG T101 C021)

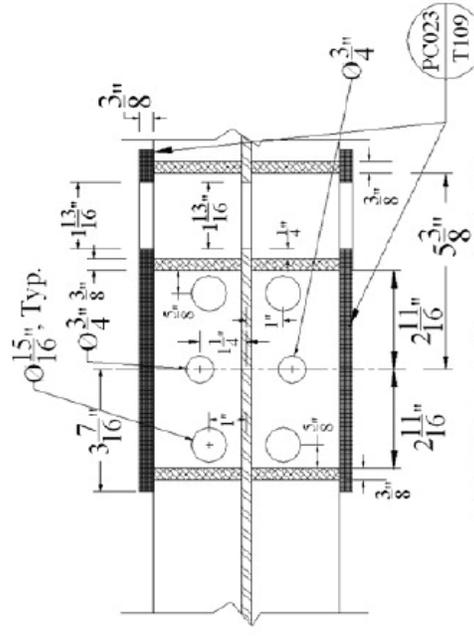
Section A-A

<b>University at Buffalo</b>		DWG. NO.	
PROJECT Shake Table Test	CONTENT C021 Interior Column w6x25	T106	
BY: <b>Dong Wang</b>	REV. NO. 2	Date 9/5/2006	SCALE

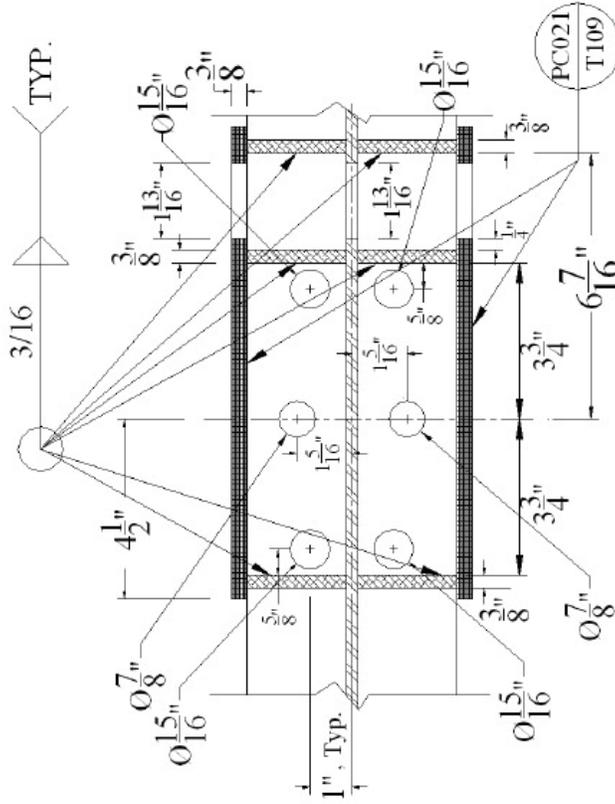
**Note: Do not weld the area near between the continuity plates (shade parts) and holes.  
Do not weld the plates PC021, PC022 and PC023, just manufacture these plates as shown in DWG 109.**



**DWG T106 Detail AH2**  
(All weldings are identical to Detail AH1)

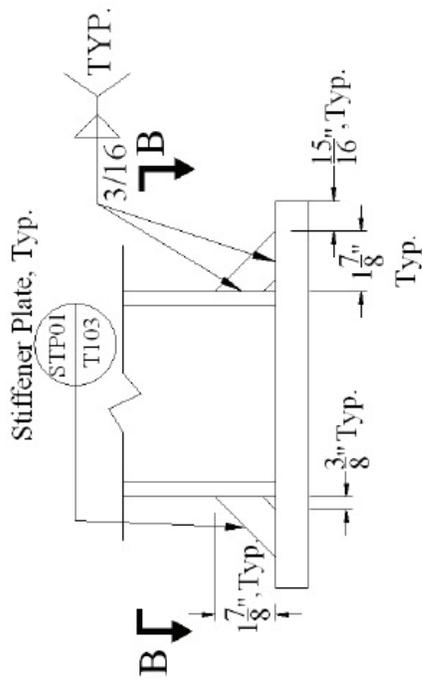


**DWG T106 Detail AH3**  
(All weldings are identical to Detail AH1)

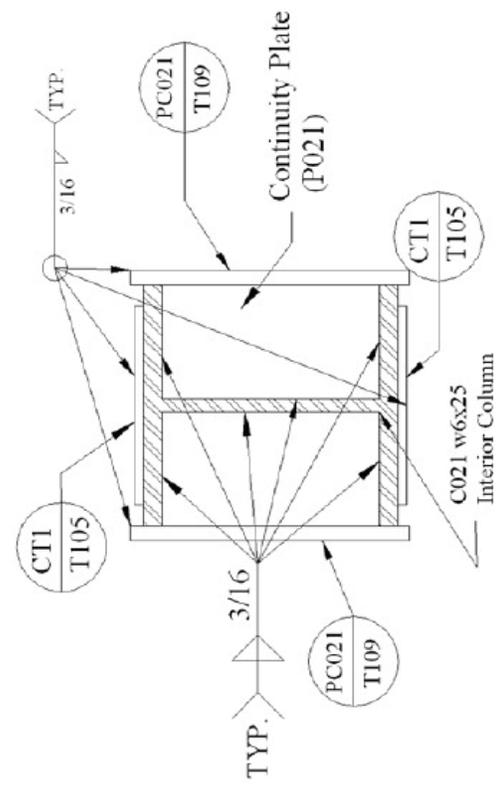


**DWG T106 Detail AH1**

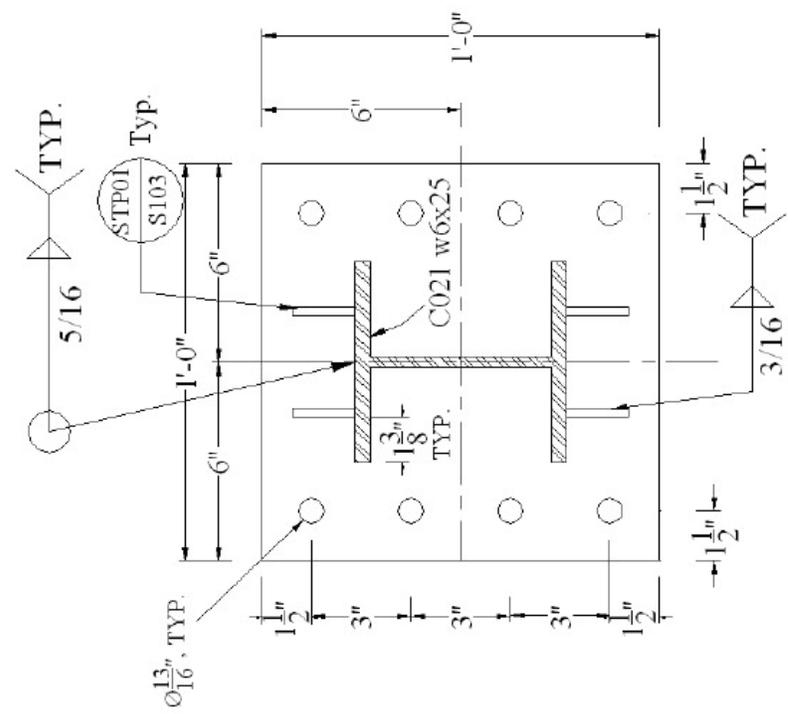
<b>University at Buffalo</b>		DWG. NO.	
PROJECT Shake Table Test	CONTENT Details AH1, 2, 3 in DWG T106	T107	
BY: <b>Dong Wang</b>	REV. NO. 1	Date 10/29/2006	SCALE



Detail AH0 in DWG T106  
(Interior Column w6x25 base  
connection with stiffener plates)

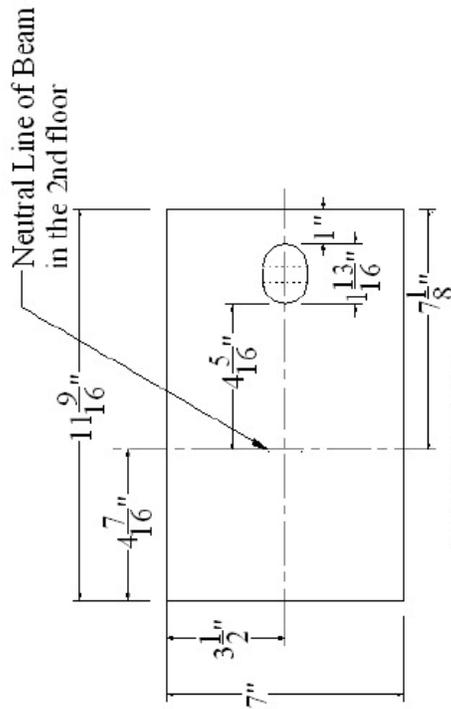


DWG T106 Section C-C

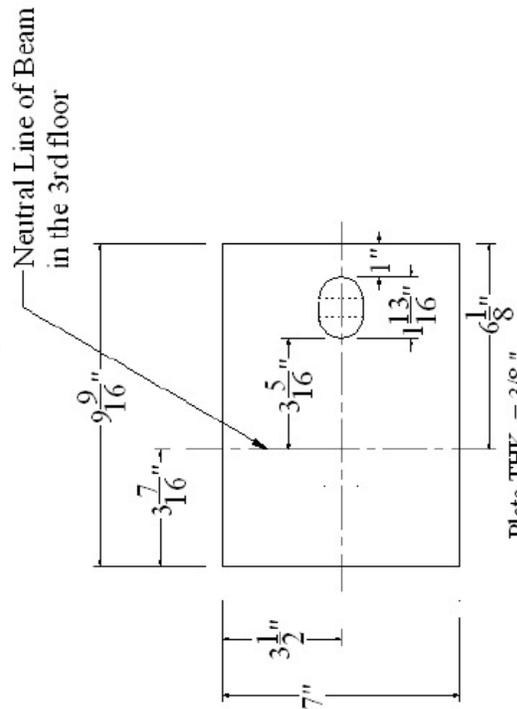


Section B-B Base Connection  
(DWG T106 section B-B,  
DWG T101 Detail A2)

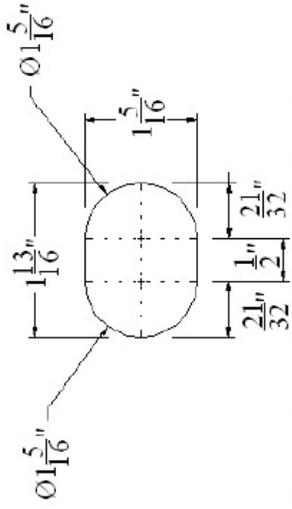
<b>University at Buffalo</b>		DWG. NO.	
PROJECT Shake Table Test	CONTENT Section B-B,C-C in DWG T106	T108	
BY: <b>Dong Wang</b>	REV. NO. 2	Date 10/29/2006	SCALE



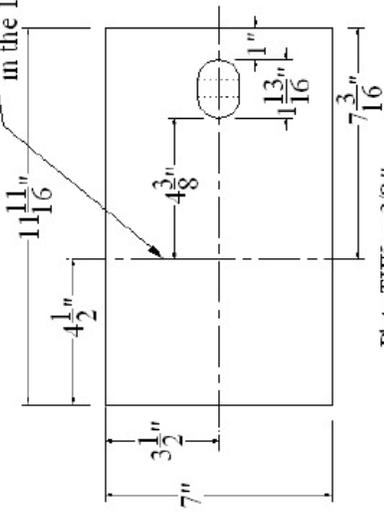
DWG T106 PC022 Web  
Reinforcing Plate (2pc.)



DWG T106 PC023 Web  
Reinforcing Plate (2pc.)



The hole in PC021, PC022 and  
PC023.



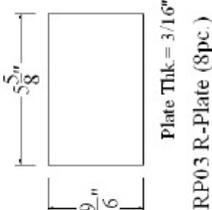
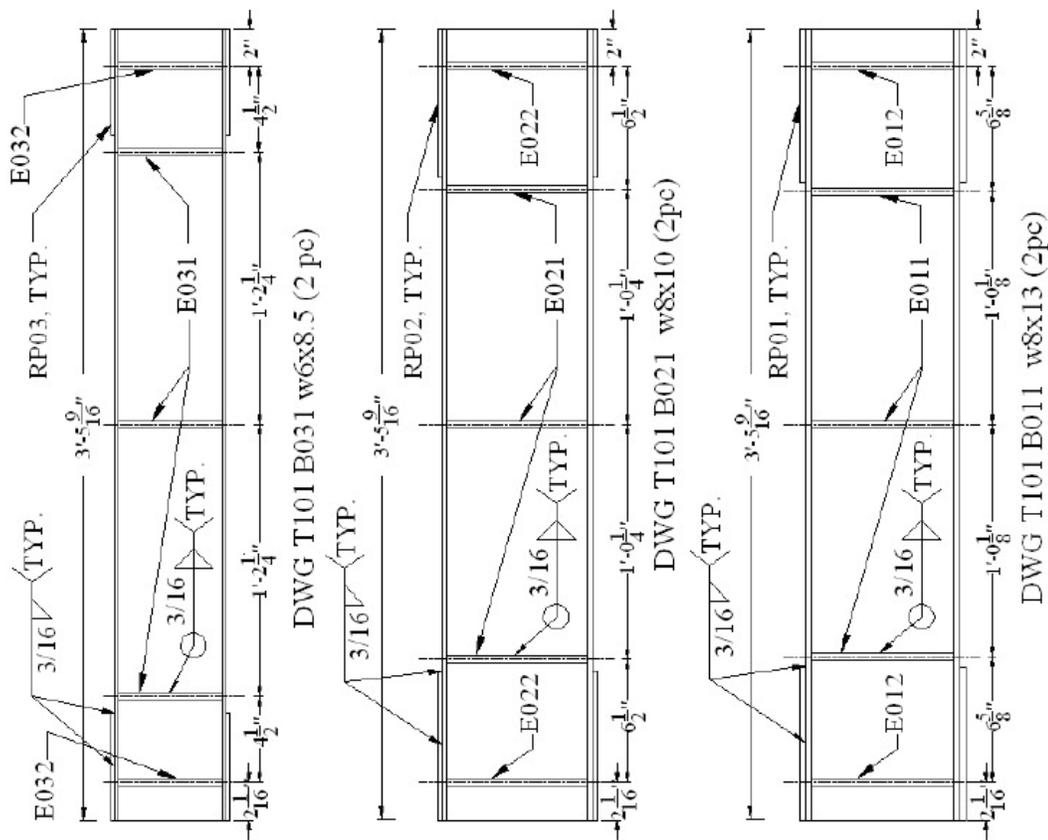
DWG T106 PC021 Web Reinforcing  
Plate (2pc.)

**Note: The corresponding part of  
column (C021) web should be  
machined with the same holes as  
shown in these plates.**

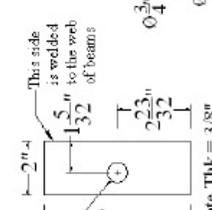
<b>University at Buffalo</b>		DWG. NO.	
PROJECT Shake Table Test	CONTENT Web Reinforcing Plates	T109	
BY: <b>Dong Wang</b>	REV. NO. 1	Date 9/25/2006	SCALE

# Beams

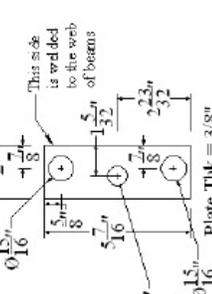
**Note: For the Reinforcing Plates (R-plates, A514, Fy = 100ksi) on the top of beam flanges, only 3-side welding around plates is necessary and do not weld these plates on the 4<sup>th</sup> side at the both ends of**



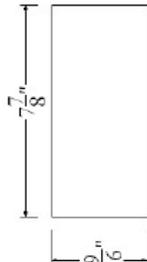
Note: RP03 Plates use ASTM A514 with 100 ksi



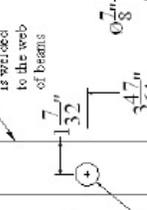
Note: RP03 Plates use ASTM A514 with 100 ksi



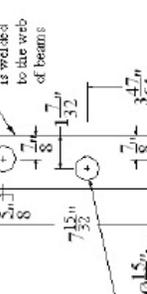
Note: RP03 Plates use ASTM A514 with 100 ksi



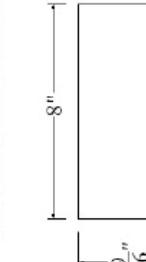
Note: RP02 Plates use ASTM A514 with 100 ksi



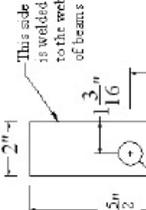
Note: RP02 Plates use ASTM A514 with 100 ksi



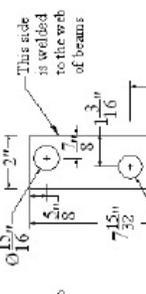
Note: RP02 Plates use ASTM A514 with 100 ksi



Note: RP01 Plates use ASTM A514 with 100 ksi

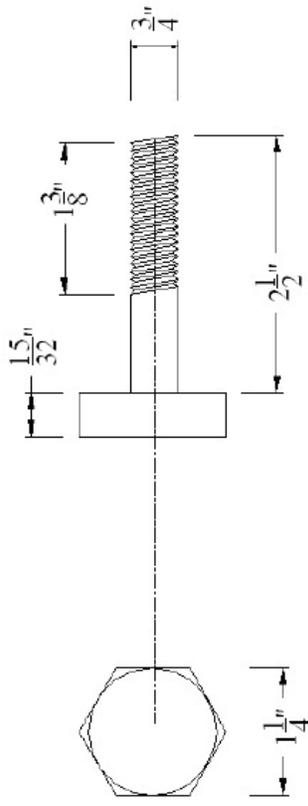


Note: RP01 Plates use ASTM A514 with 100 ksi

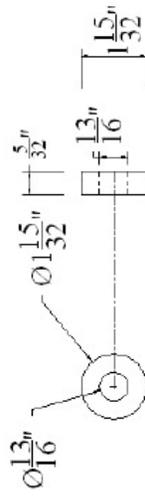


Note: RP01 Plates use ASTM A514 with 100 ksi

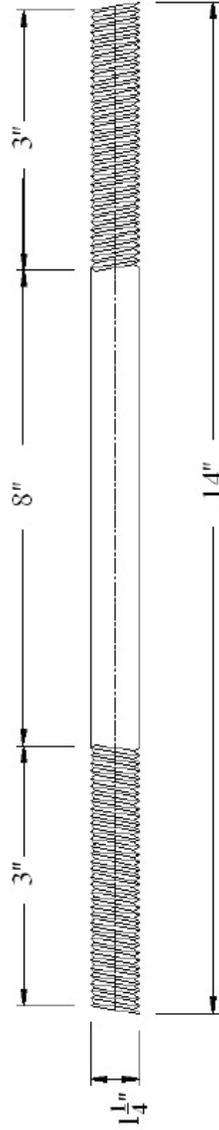
<b>University at Buffalo</b>		DWG. NO.	
PROJECT Shake Table Test	CONTENT Beams	T110	
BY: <b>Dong Wang</b>	REV. NO. 1	Date 11/07/2006	SCALE



Headed Anchor Bolts (ASTM A325  $\text{\O} \frac{3}{4}$ , total 24 pc.)



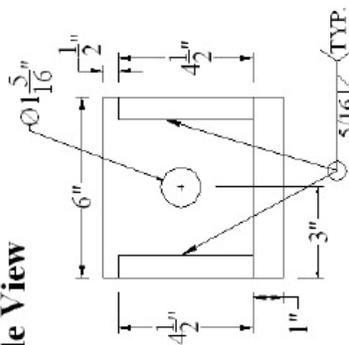
Washer for Headed Anchor Bolts (total 24 pc.)



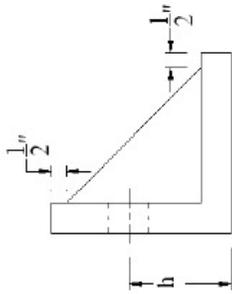
Threaded Rods (ASTM A325  $\text{\O} 1 \frac{1}{4}$ , total 3 pc.  
and corresponding ASTM A563 nuts, 6pc.)

<b>University at Buffalo</b>		DWG. NO.	
PROJECT Shake Table Test	CONTENT Anchor Bolts and Rods	T111	
BY: <b>Dong Wang</b>	REV. NO. 1	Date 9/25/2006	SCALE

**Whole View**



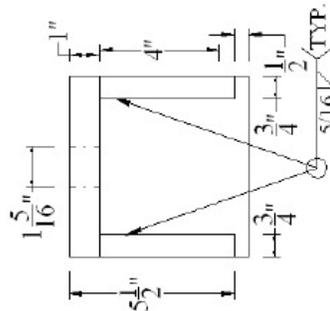
(See Part Dimension for h)



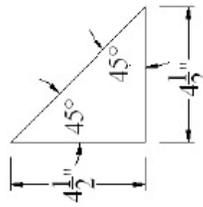
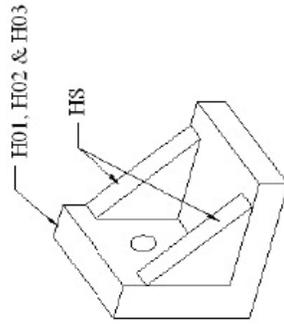
**Part Dimension**

Part	h
H01	$3\frac{17}{32}$
H02	$3\frac{15}{32}$
H03	$2\frac{15}{32}$

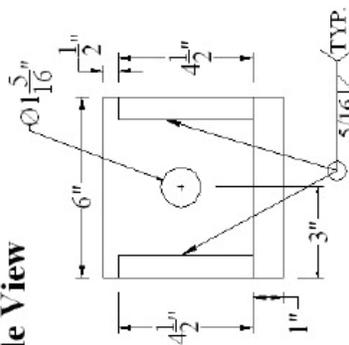
6" long Angles L6x6x1  
Part H01(2pc.), H02(2pc.), H03(2pc.)  
total 6pc.



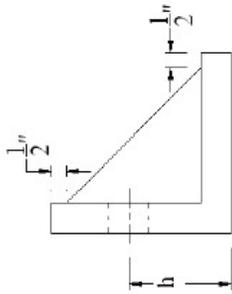
3-D View



**Whole View**



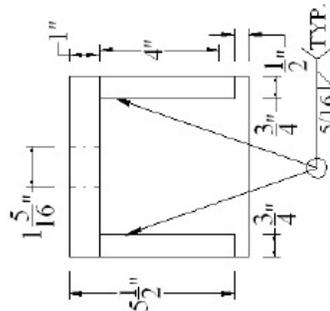
(See Part Dimension for h)



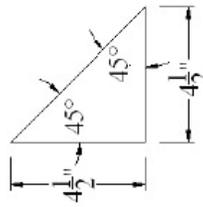
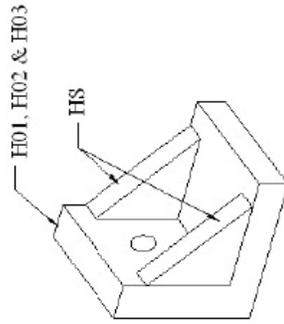
**Part Dimension**

Part	h
H01	$3\frac{17}{32}$
H02	$3\frac{15}{32}$
H03	$2\frac{15}{32}$

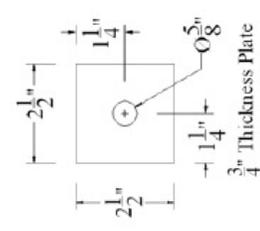
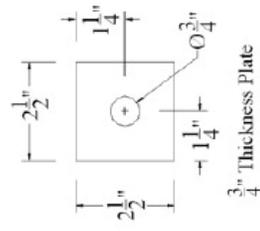
6" long Angles L6x6x1  
Part H01(2pc.), H02(2pc.), H03(2pc.)  
total 6pc.



3-D View



<b>University at Buffalo</b>		DWG. NO.
PROJECT Shake Table Test	CONTENT Connections between Mass Floors and Frame	T112
BY: <b>Dong Wang</b>	REV. NO. 1	Date 10/11/2006
		SCALE SCALE



### Load Cell

6 pieces of **Extra Strong** Steel Pipe (Fy = 50ksi)

Nominal Diameter: **1 1/4"**

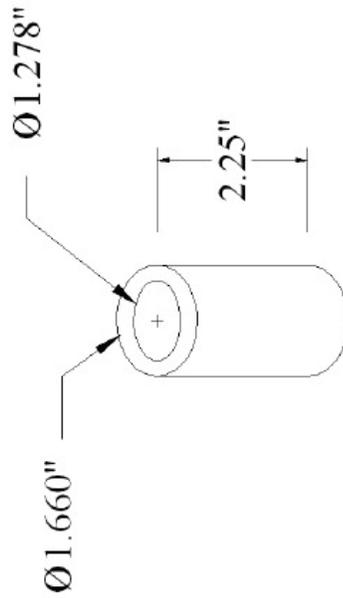
Outside Diameter: **1.66"**

Inside Diameter: **1.278"**

Wall Thickness: **0.191"**

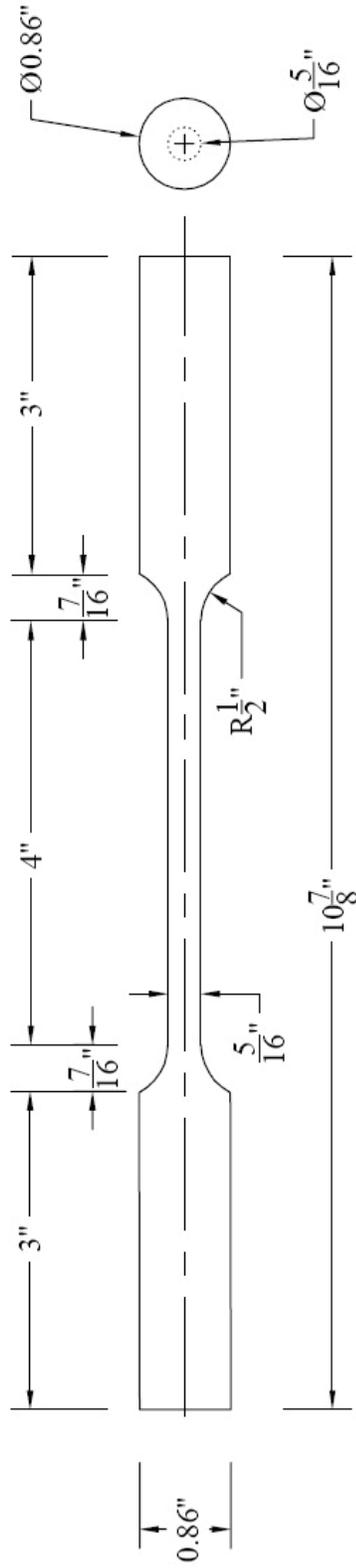
Area: **0.881 in<sup>2</sup>**

Total **13.5"** of steel pipe

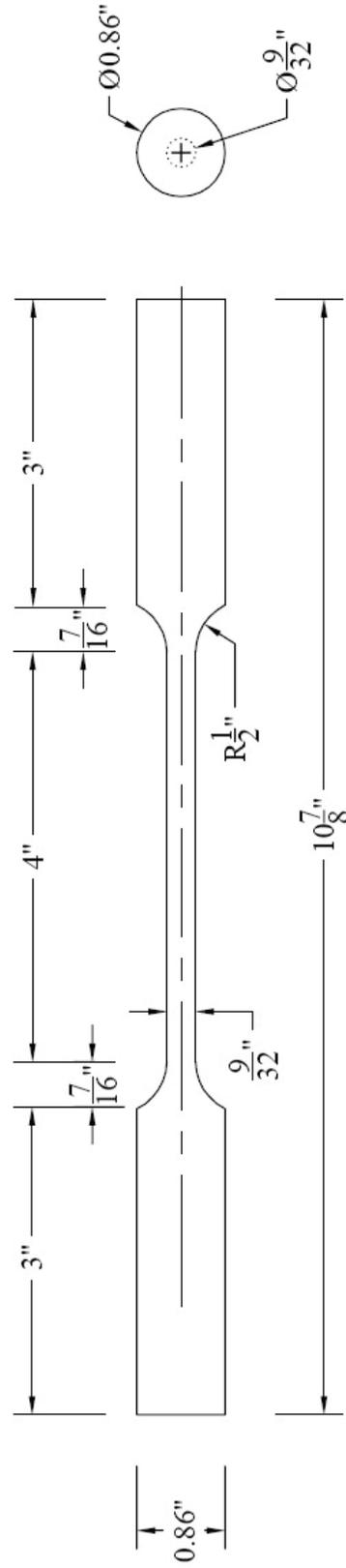


<b>University at Buffalo</b>		DWG. NO.	
PROJECT Shake Table Test	CONTENT Load Cell	T113	
BY: <b>Dong Wang</b>	REV. NO. 1	Date 11/1/2006	SCALE

Materials:  $\frac{3}{4}$ " thread bars with maximum diameter 0.86"



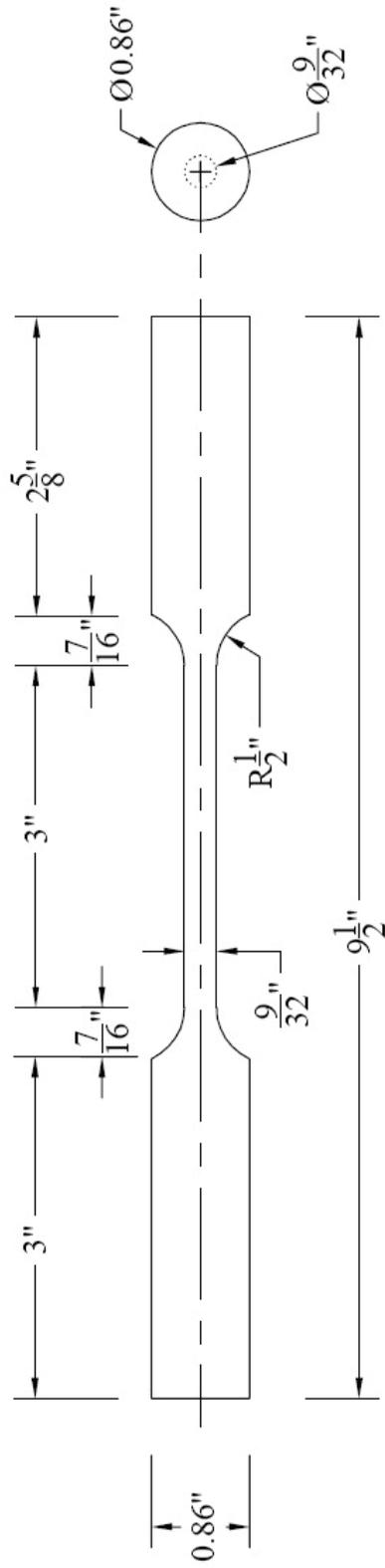
ED Bars in 1st floor (total 16 pc)



ED Bars in 2nd floor (total 16 pc)

<b>University at Buffalo</b>		DWG. NO. <b>ED101</b>	
		PROJECT: <b>Shake Table Test</b>	CONTENT: <b>ED Bars in 1st &amp; 2nd floor</b>
BY: <b>DONG WANG</b>	DATE: <b>11/14/06</b>	REV. NO. <b>1</b>	SCALE:

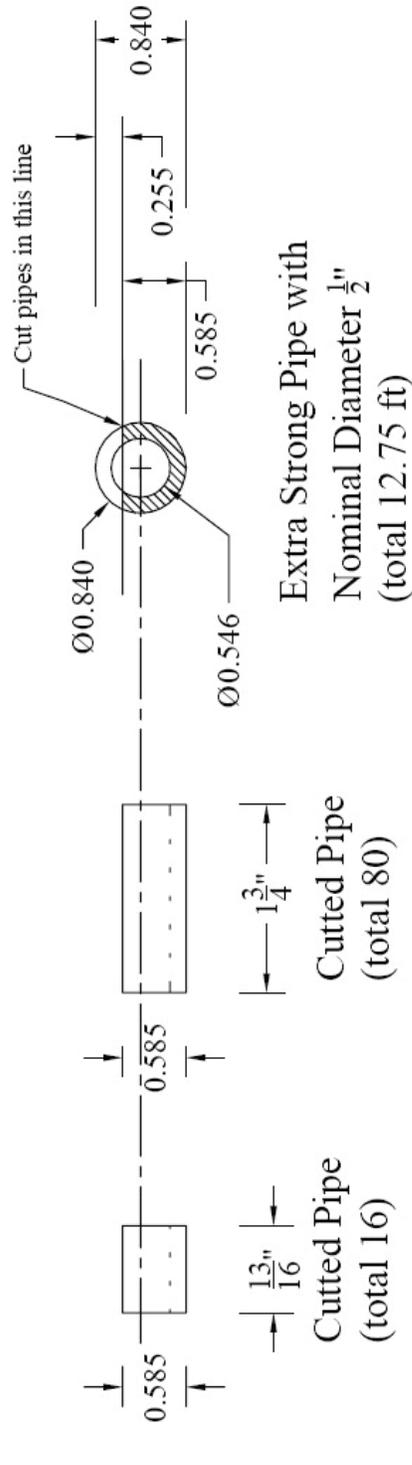
Materials:  $\frac{3}{4}$ " thread bars with maximum diameter 0.86"



ED Bars in the 3rd floor  
(total 16 pc)

<b>University at Buffalo</b>		DWG. NO. <b>ED102</b>
PROJECT: <b>Shake Table Test</b>	CONTENT: <b>ED Bars in 3rd floor</b>	
BY: <b>DONG WANG</b>	DATE: <b>11/14/06</b>	REV. NO. <b>1</b>
		SCALE:

Material: 16-ft Extra Strong Pipe with Nominal Diameter  $\frac{1}{2}$ " (from lab)  
 12" x 16" Plate with  $\frac{3}{8}$ " thickness



Total Area of Plates  
 =172 in<sup>2</sup>

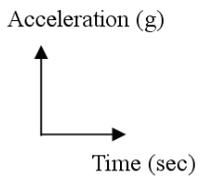
$\frac{3}{8}$ " thickness  
 Reinforcing Plate  
 (total 16)

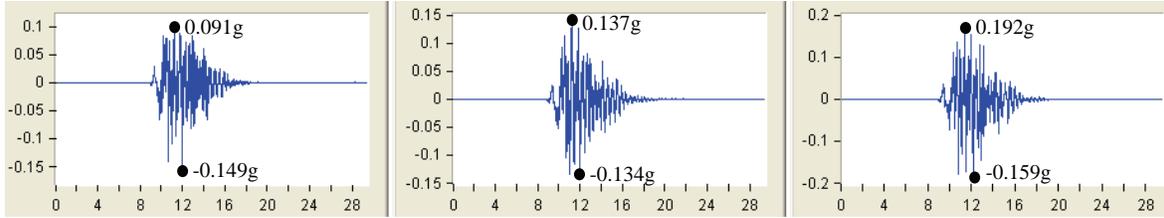
$\frac{3}{8}$ " thickness  
 Reinforcing Plate  
 (total 80)

<b>University at Buffalo</b>		DWG. NO.	ED103
		PROJECT:	Shake Table Test
CONTENT:	Reinforcing Plates	DATE:	11/12/06
BY:	DONG WANG	REV. NO.	1
		SCALE:	

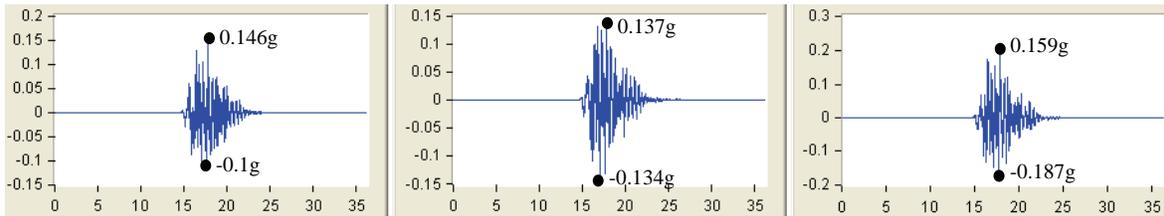


**Appendix E Absolute Acceleration Time History of the  
SCPT and SMRF Models for Seismic Tests**

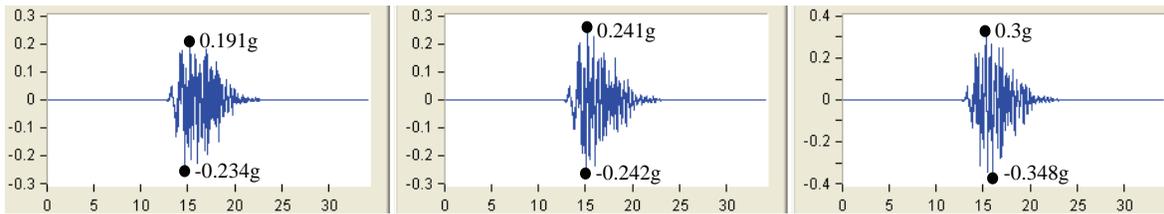
SCPT      1<sup>st</sup> floor                      2<sup>nd</sup> floor                      3<sup>rd</sup> floor                      Acceleration (g)  
 Time (sec)



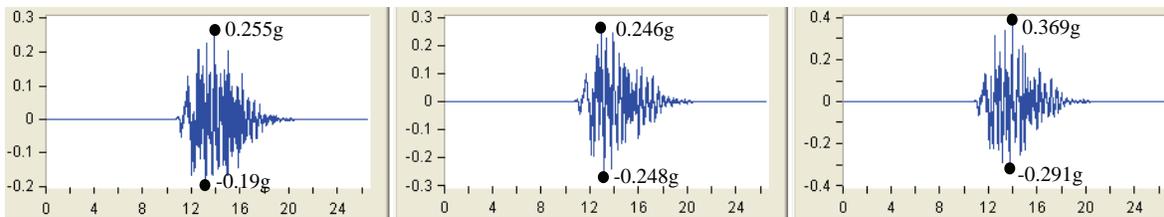
Acceleration Time History of Test PE1B



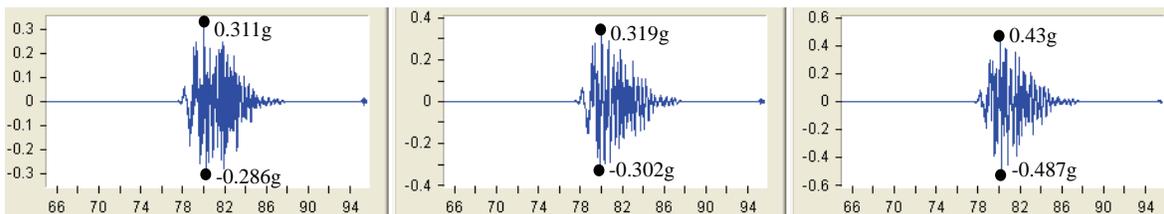
Acceleration Time History of Test PER1



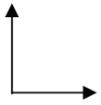
Acceleration Time History of Test PE2

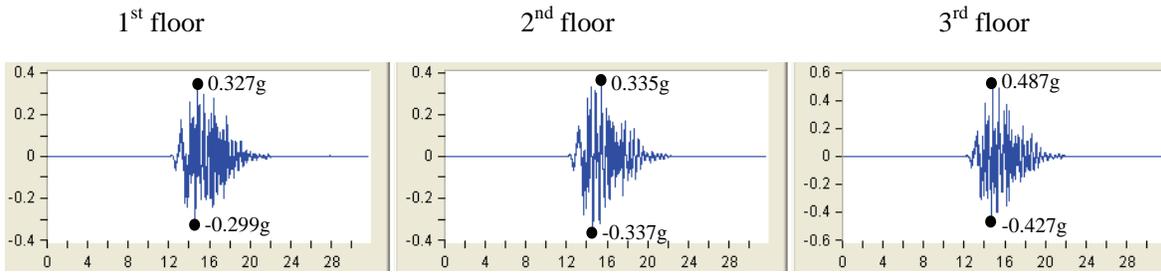


Acceleration Time History of Test PER2

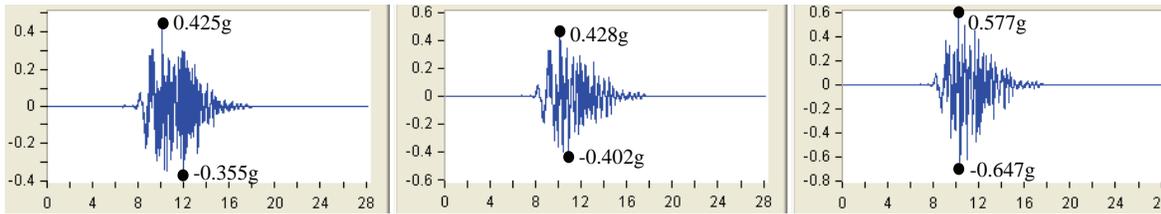


Acceleration Time History of Test PE3

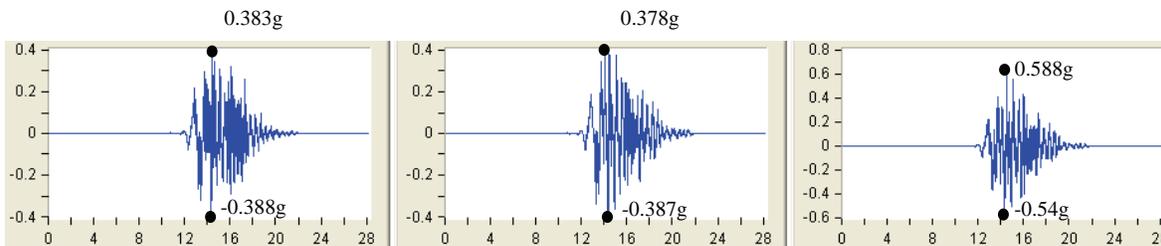
Acceleration (g)  
  
 Time (sec)



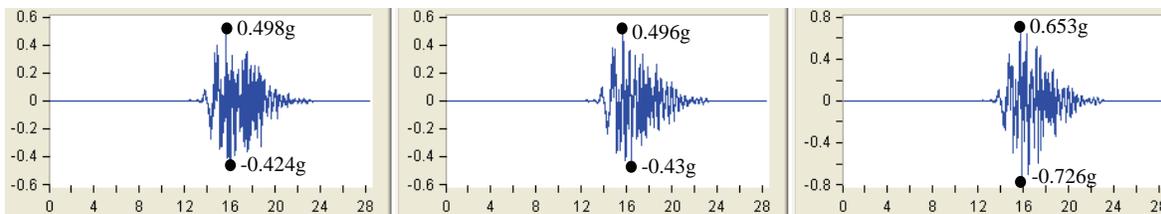
Acceleration Time History of Test PER3



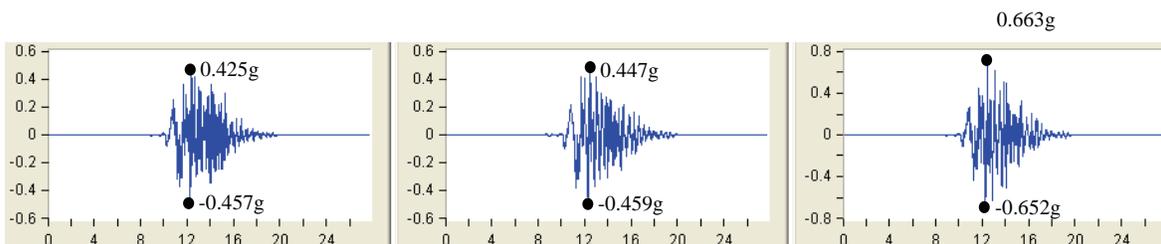
Acceleration Time History of Test PE4



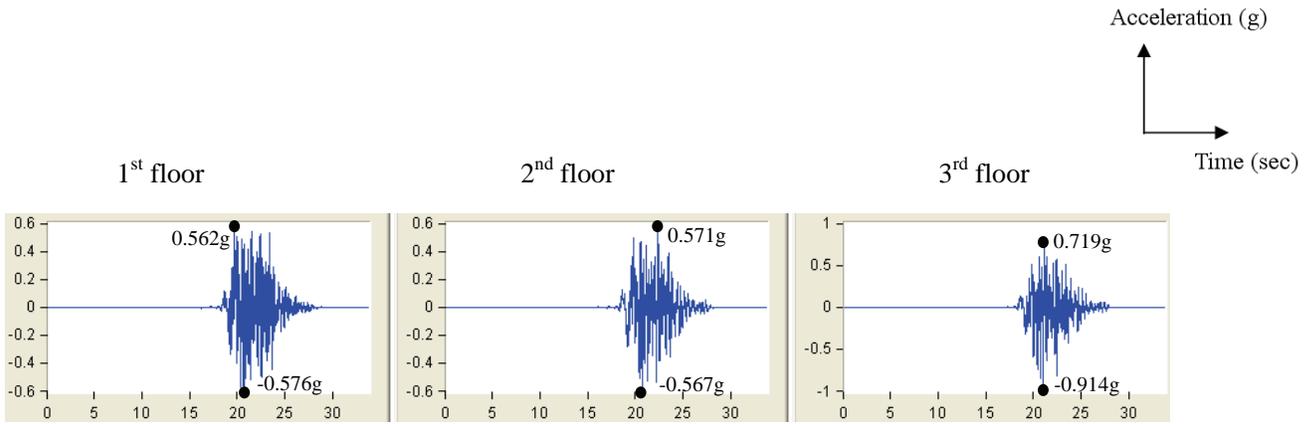
Acceleration Time History of Test PER4



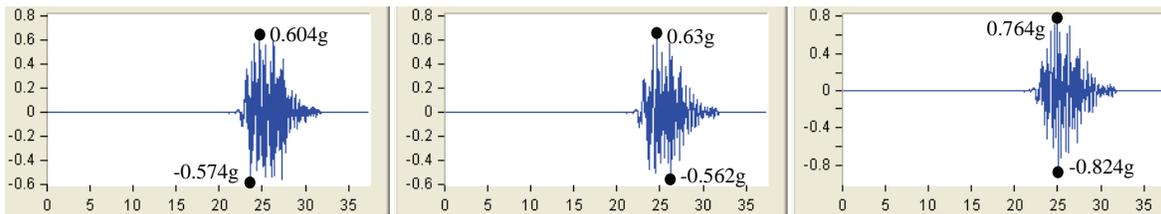
Acceleration Time History of Test PE5A



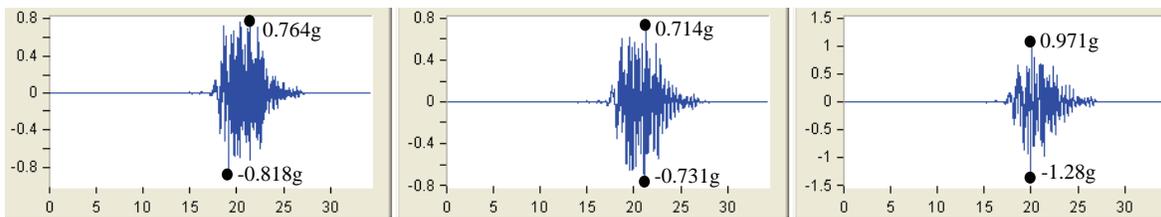
Acceleration Time History of Test PER5



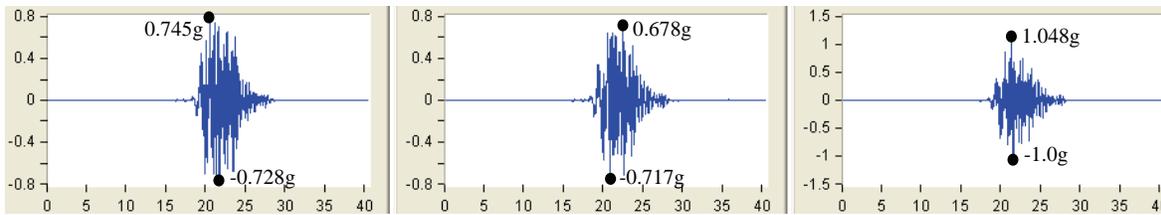
Acceleration Time History of Test PE6A



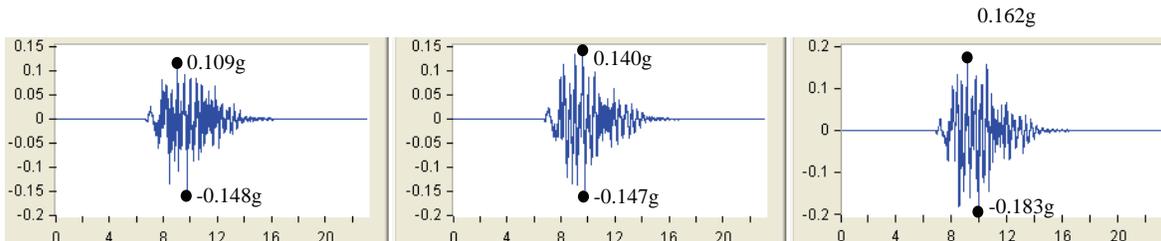
Acceleration Time History of Test PER6



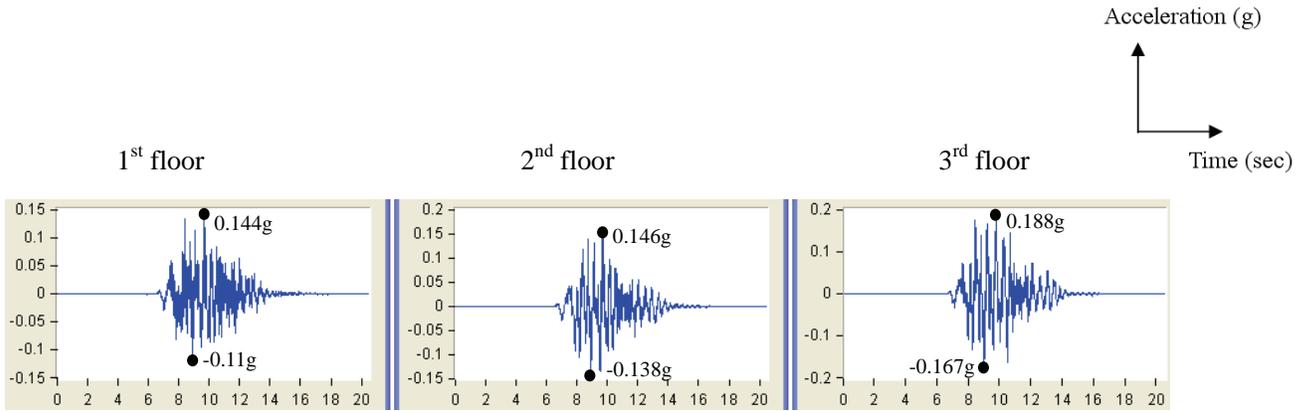
Acceleration Time History of Test PE7A



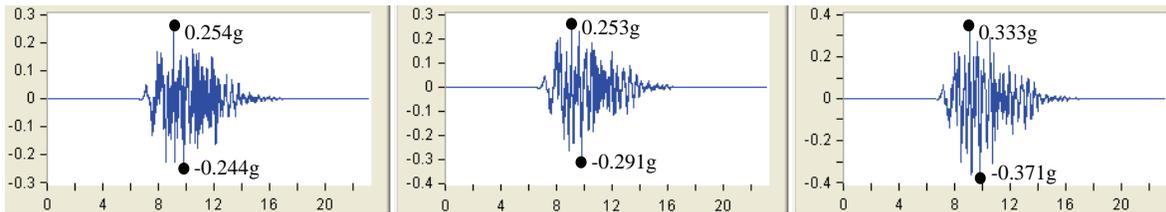
Acceleration Time History of Test PER7



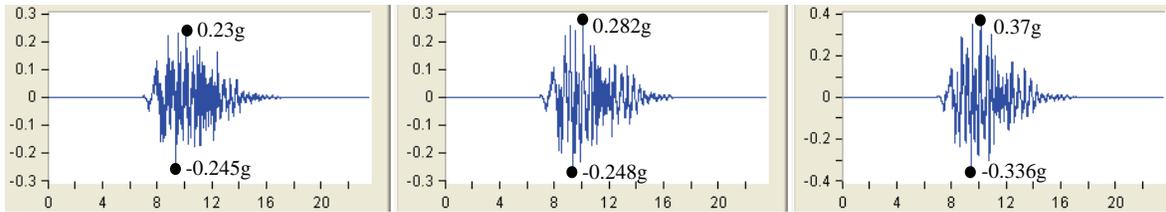
Acceleration Time History of Test PE8



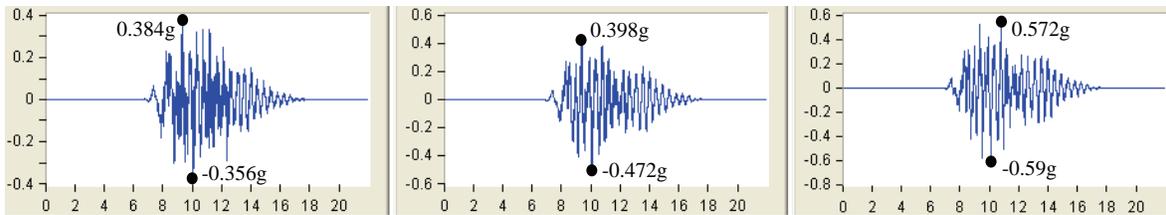
Acceleration Time History of Test PER8



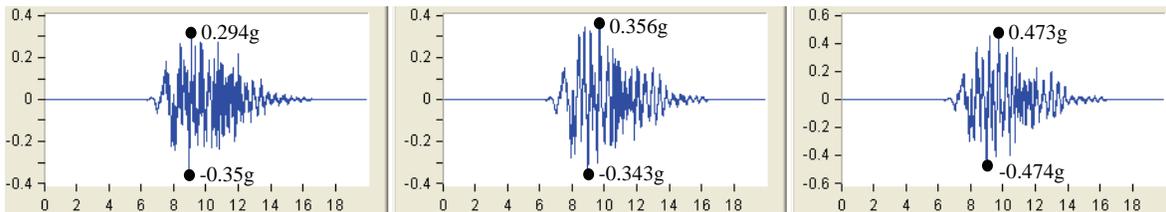
Acceleration Time History of Test PE9



Acceleration Time History of Test PER9

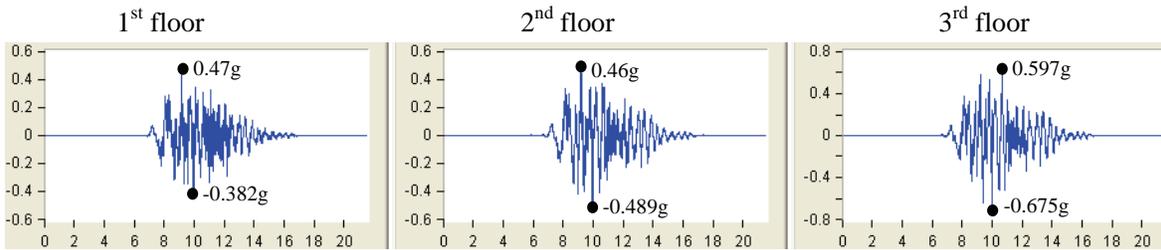


Acceleration Time History of Test PE10

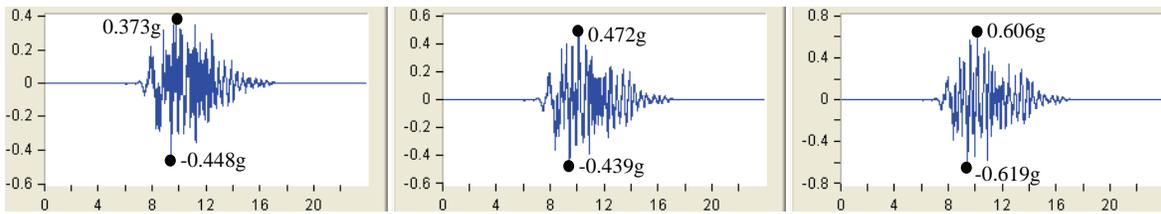


Acceleration Time History of Test PER10

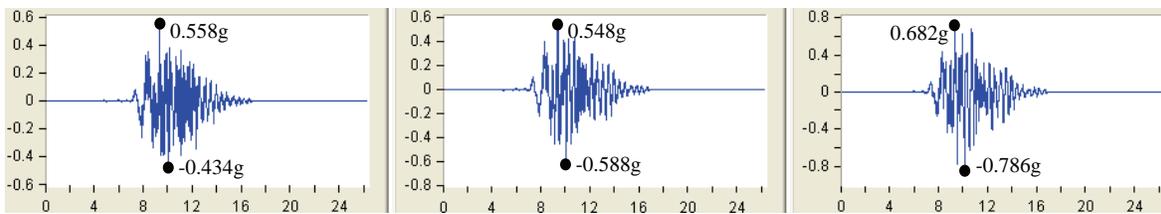
Acceleration (g)  
  
 Time (sec)



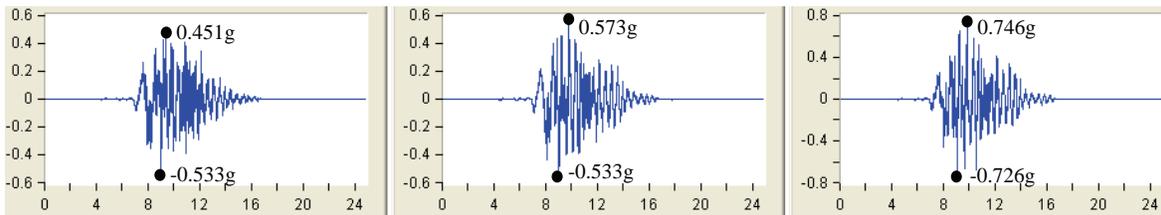
Acceleration Time History of Test PE11



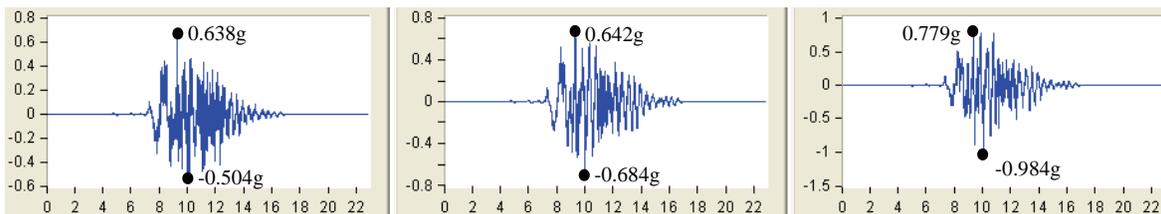
Acceleration Time History of Test PER11



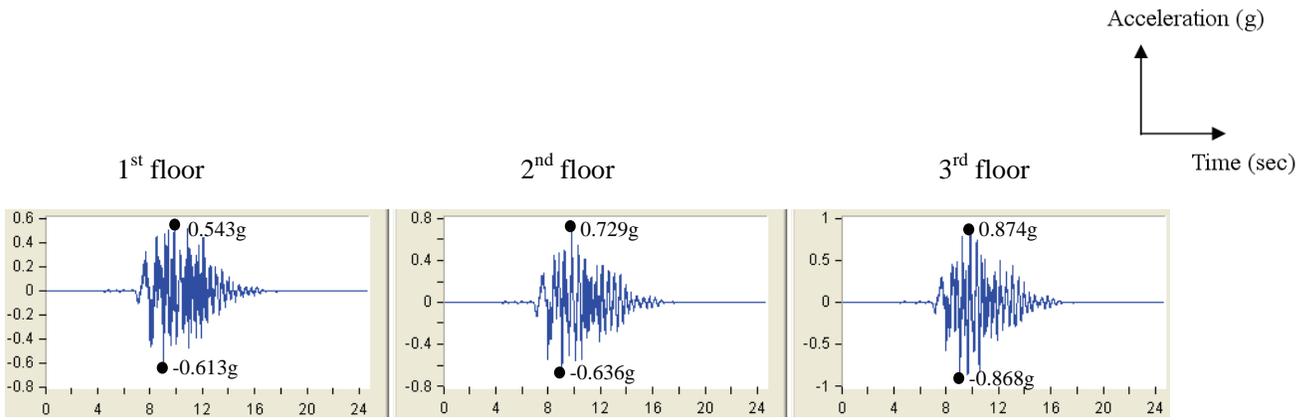
Acceleration Time History of Test PE12



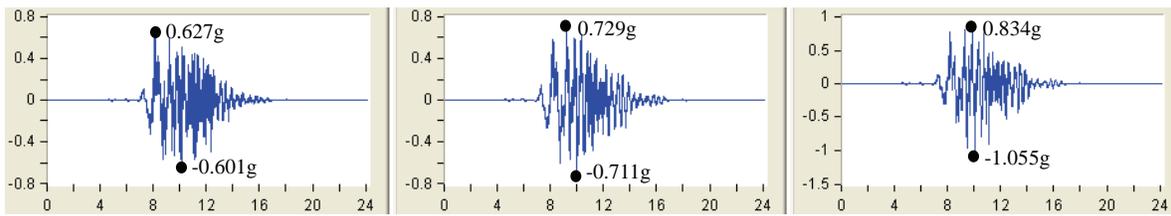
Acceleration Time History of Test PER12



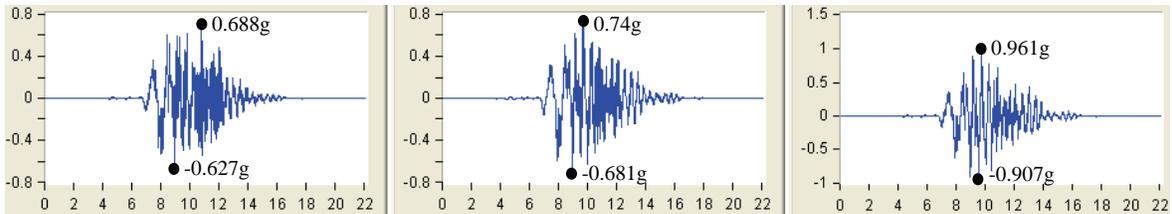
Acceleration Time History of Test PE15



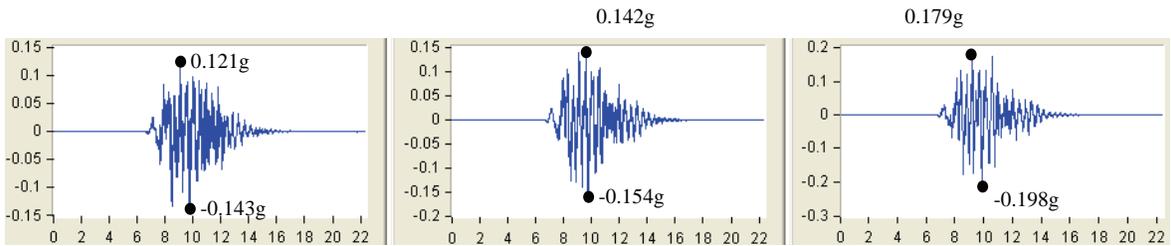
Acceleration Time History of Test PER15



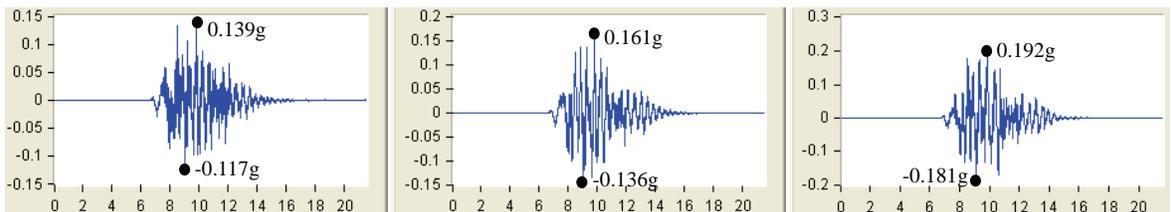
Acceleration Time History of Test PE16



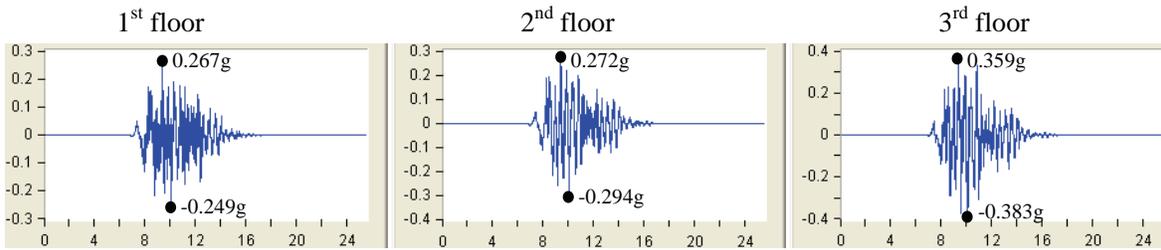
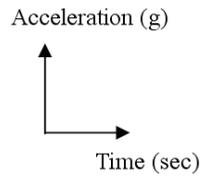
Acceleration Time History of Test PER16



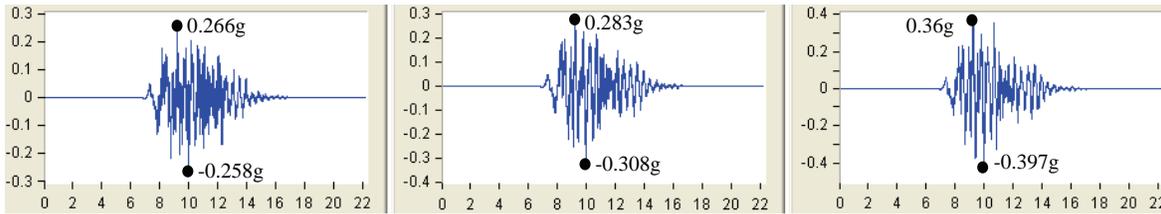
Acceleration Time History of Test PE17



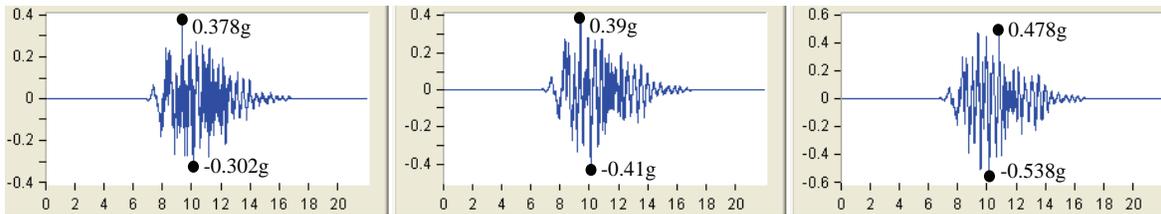
Acceleration Time History of Test PER17



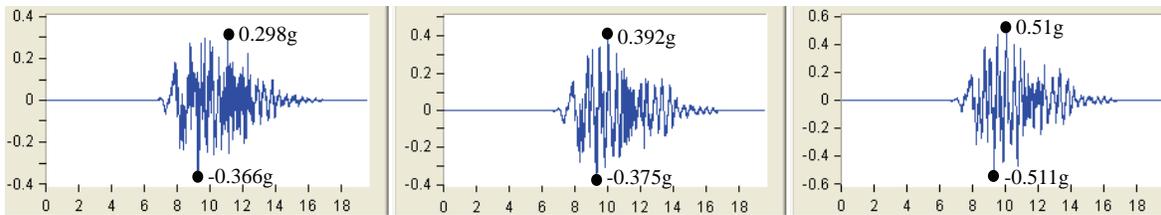
Acceleration Time History of Test PE18



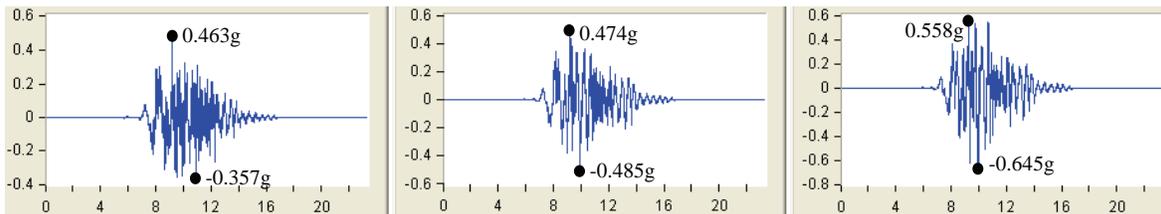
Acceleration Time History of Test PER18



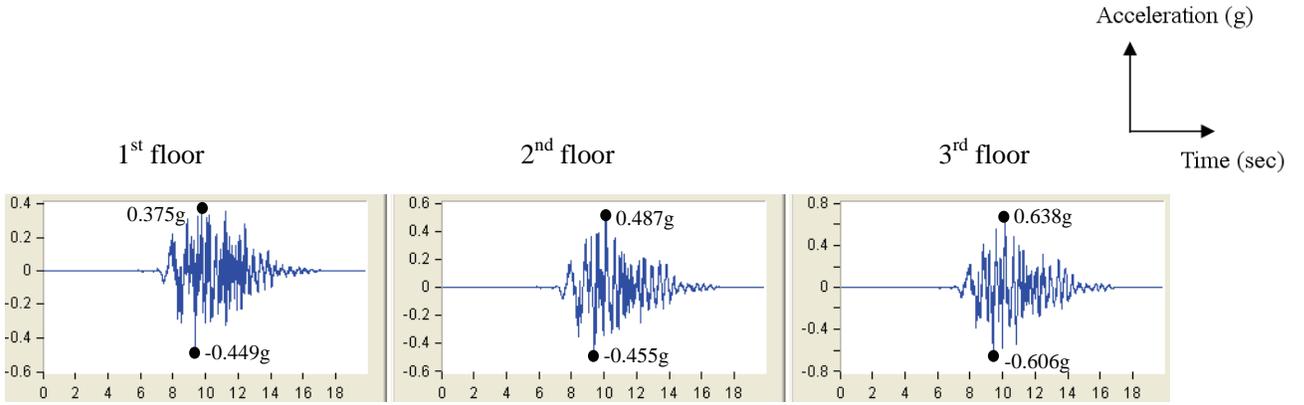
Acceleration Time History of Test PE19



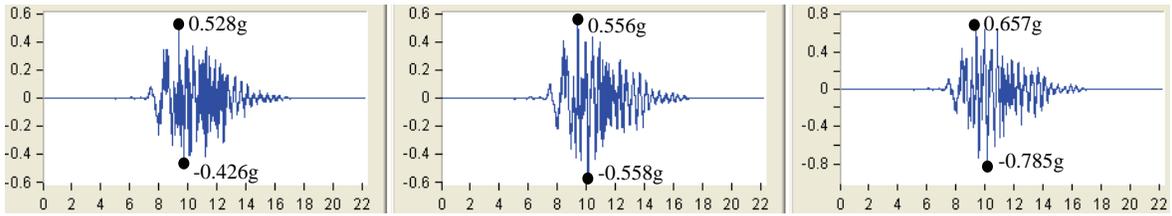
Acceleration Time History of Test PER19



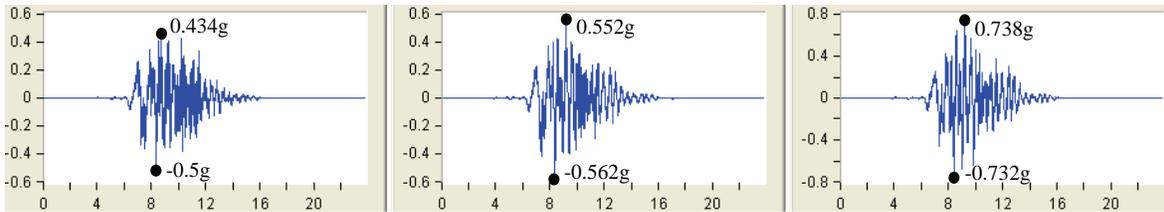
Acceleration Time History of Test PE20



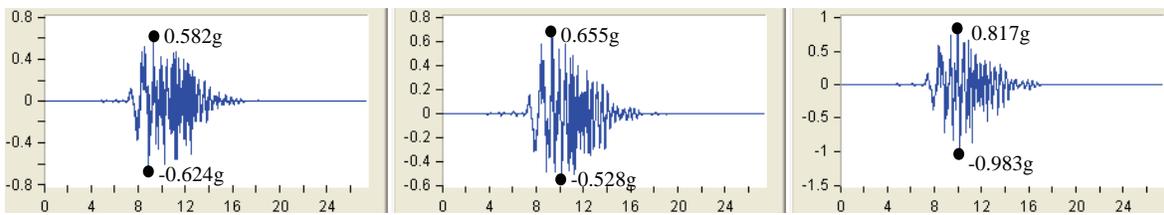
Acceleration Time History of Test PER20



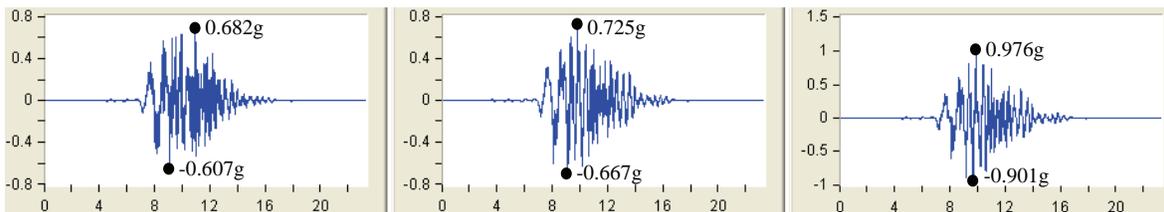
Acceleration Time History of Test PE21



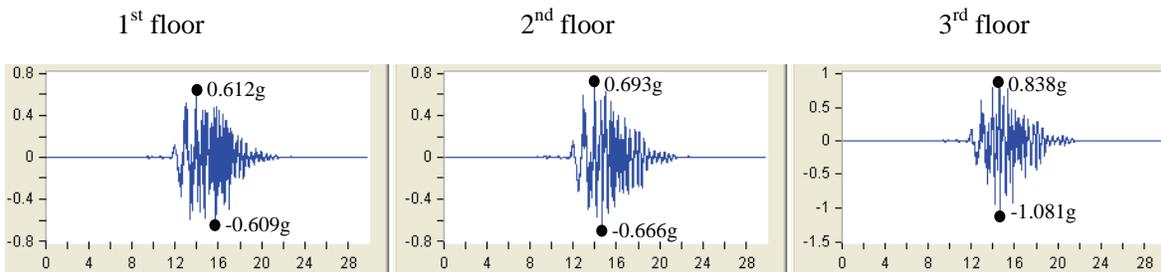
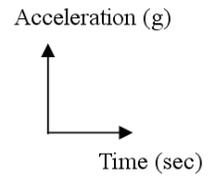
Acceleration Time History of Test PER21



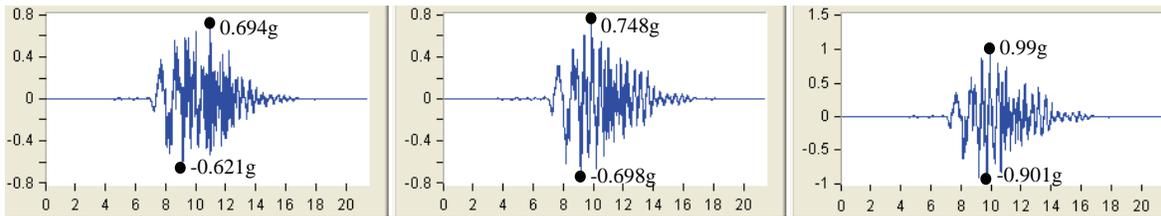
Acceleration Time History of Test PE22



Acceleration Time History of Test PER22

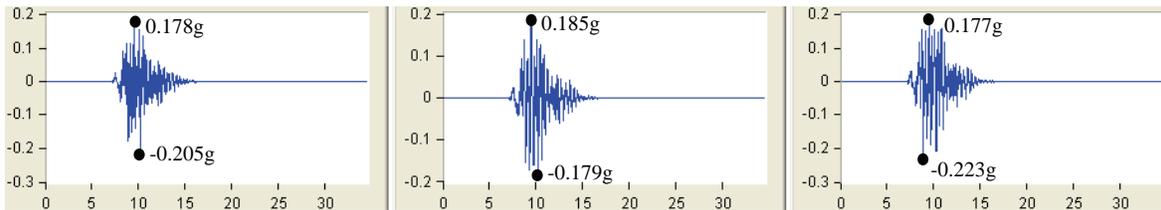


Acceleration Time History of Test PE23

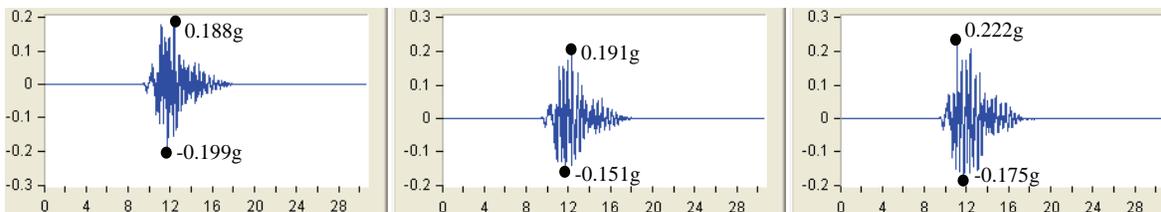


Acceleration Time History of Test PER23

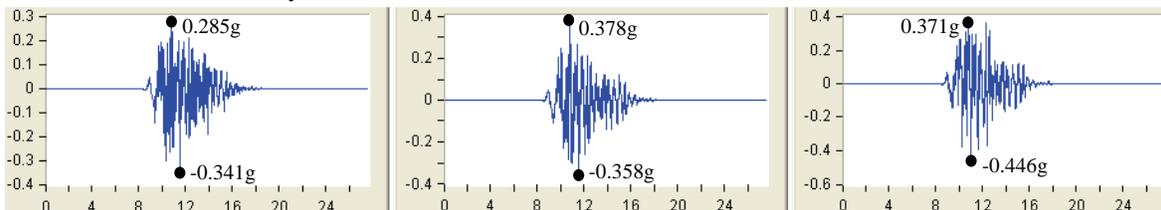
### SMRF



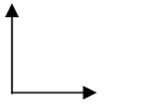
Acceleration Time History of Test ME01

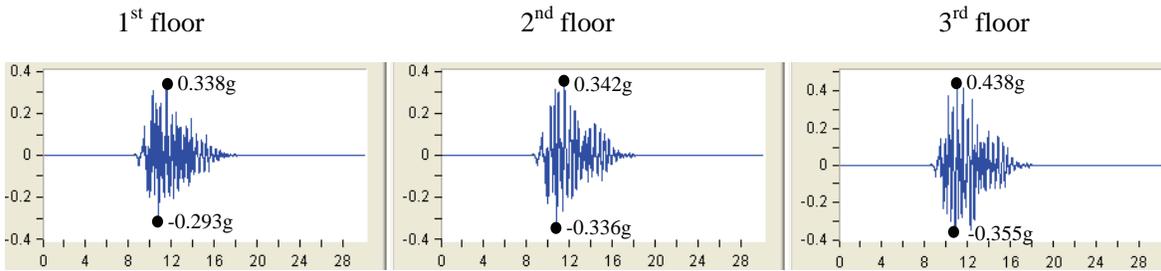


Acceleration Time History of Test MER01

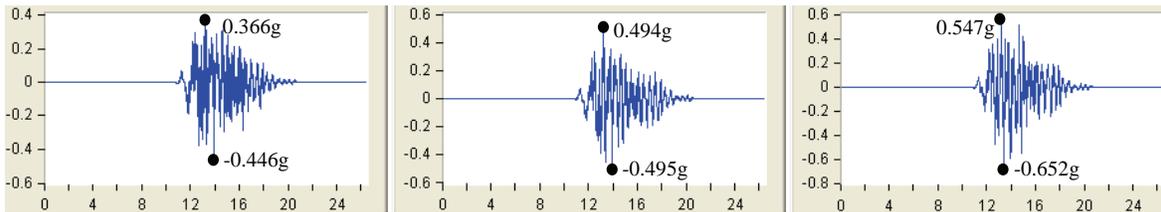


Acceleration Time History of Test ME02

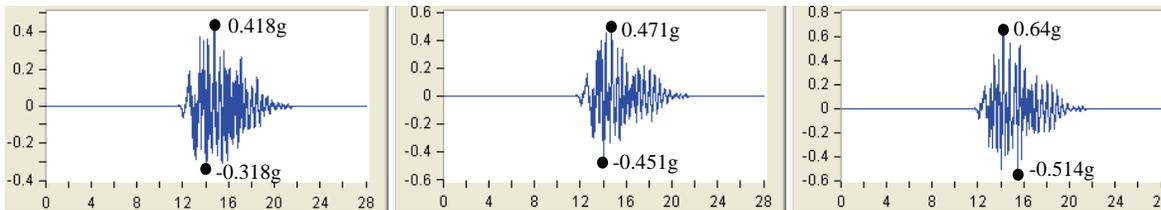
Acceleration (g)  
  
 Time (sec)



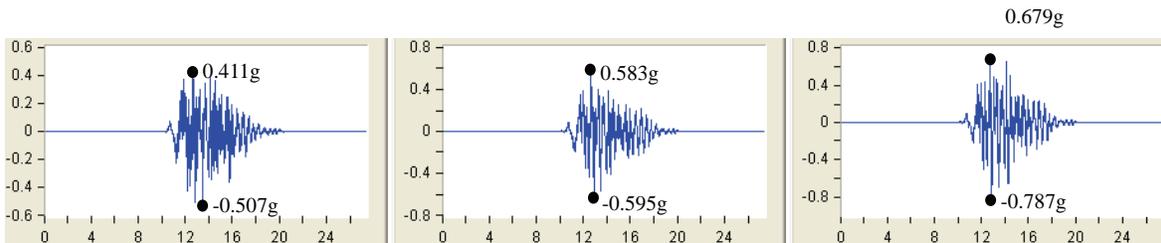
Acceleration Time History of Test MER02A



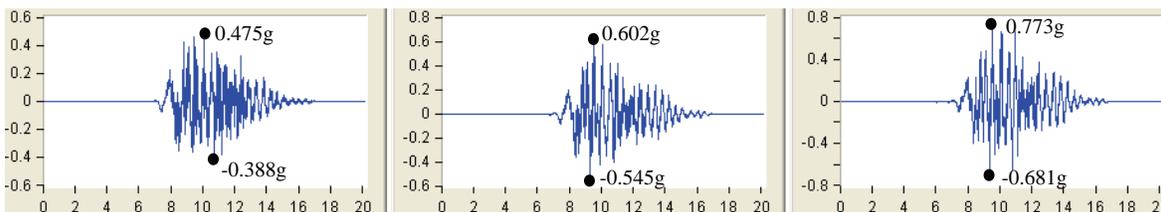
Acceleration Time History of Test ME03



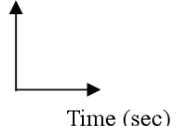
Acceleration Time History of Test MER03

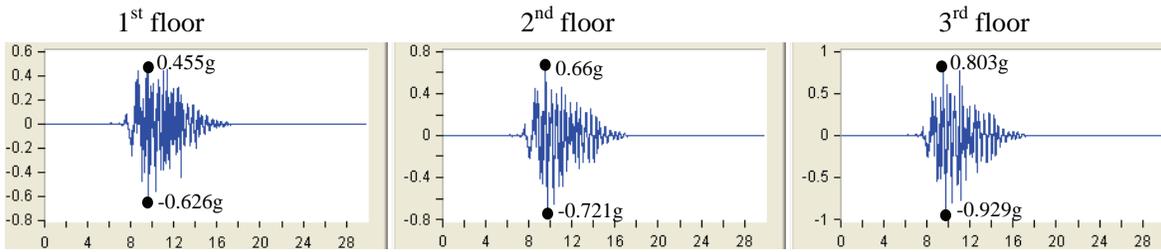


Acceleration Time History of Test ME04

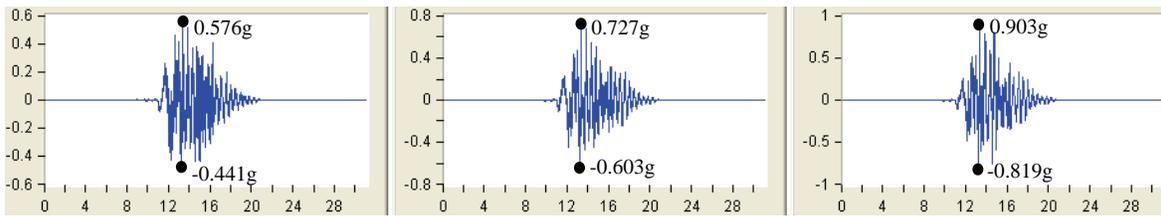


Acceleration Time History of Test MER04

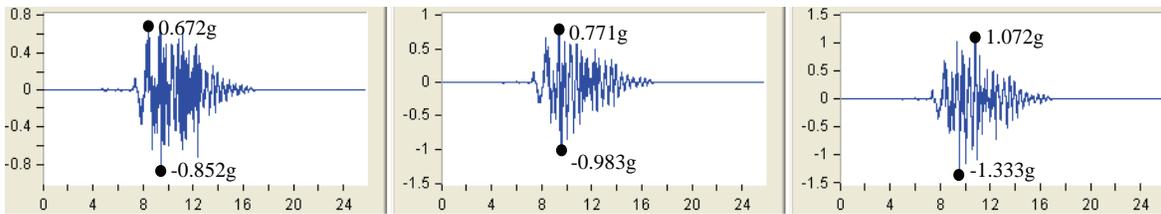
Acceleration (g)  
  
 Time (sec)



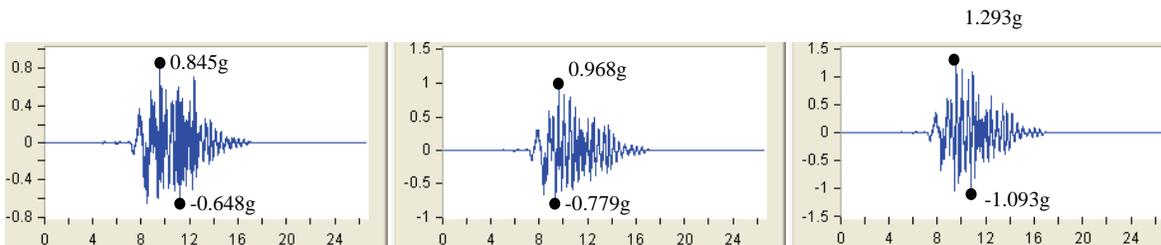
Acceleration Time History of Test ME05



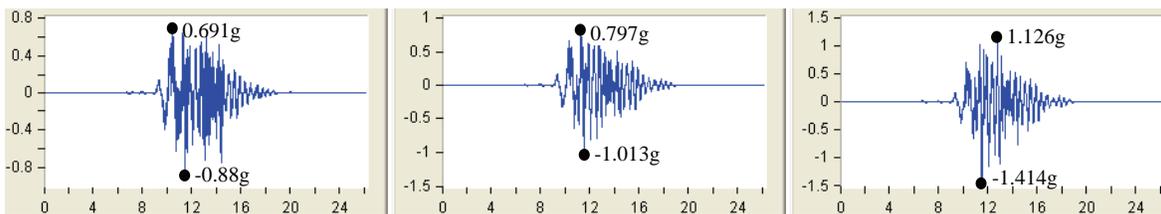
Acceleration Time History of Test MER05



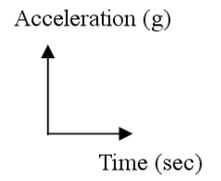
Acceleration Time History of Test ME06A



Acceleration Time History of Test MER06



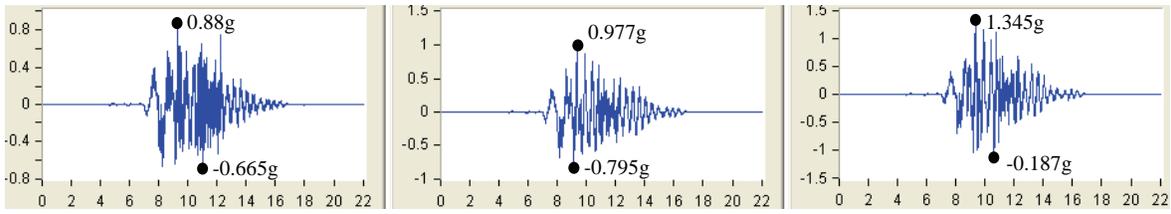
Acceleration Time History of Test ME07



1<sup>st</sup> floor

2<sup>nd</sup> floor

3<sup>rd</sup> floor

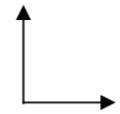


Acceleration Time History of Test MER07



**Appendix F Inter-Story Drift Time History of the SCPT  
and SMRF Models for Seismic Tests**

Inter-Story Drift (%)



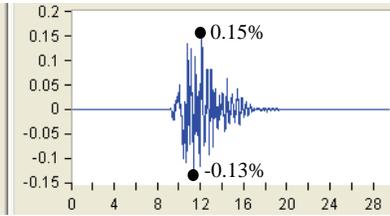
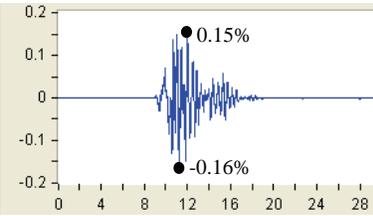
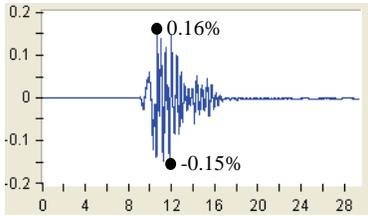
Time (sec)

SCPT

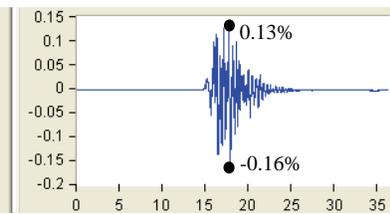
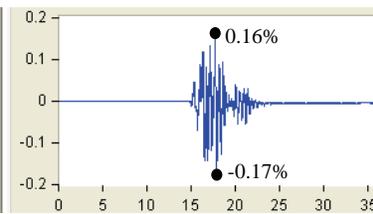
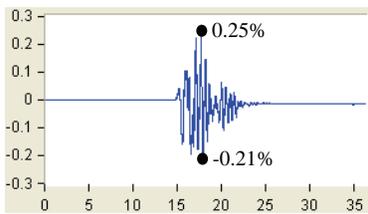
1<sup>st</sup> floor

2<sup>nd</sup> floor

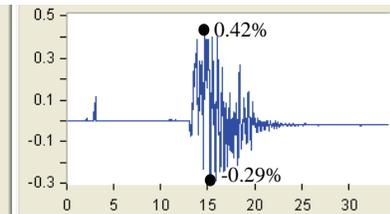
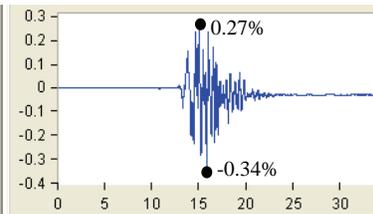
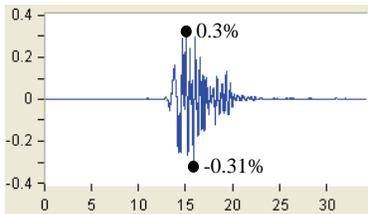
3<sup>rd</sup> floor



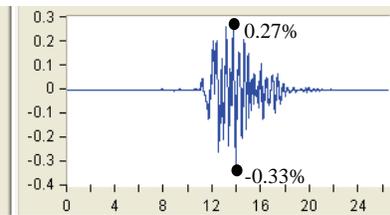
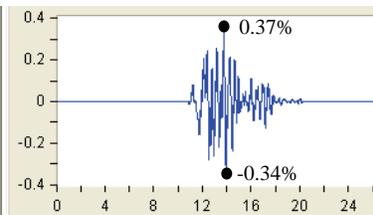
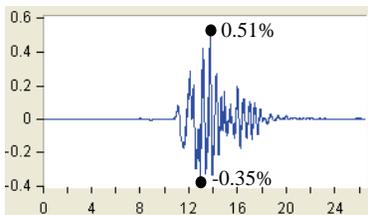
Inter-story Drift Time History of Test PE1B



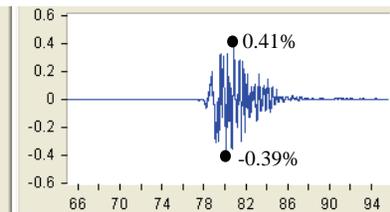
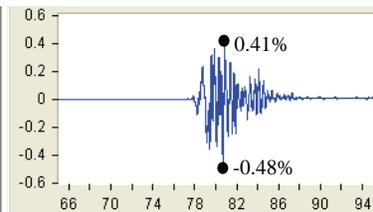
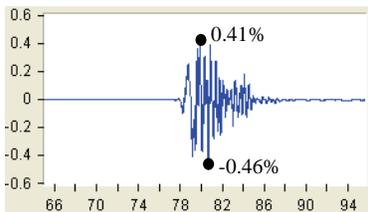
Inter-story Drift Time History of Test PER1



Inter-story Drift Time History of Test PE2

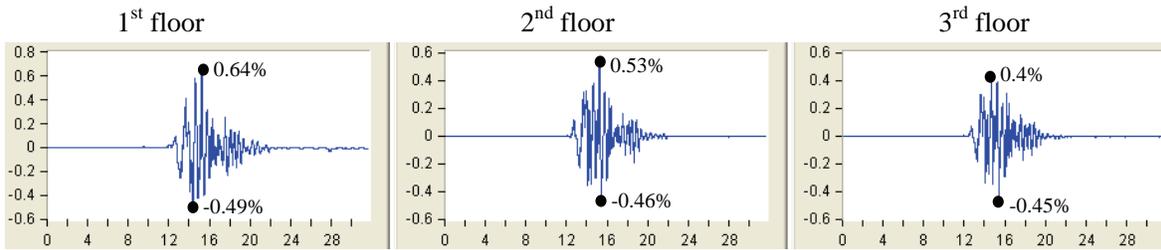
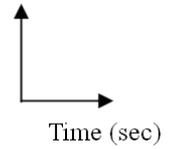


Inter-story Drift Time History of Test PER2

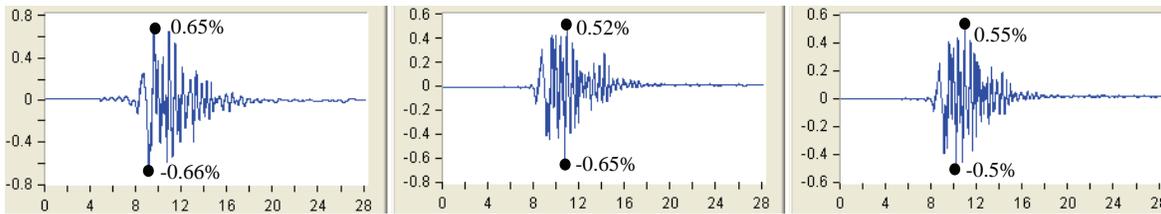


Inter-story Drift Time History of Test PE3

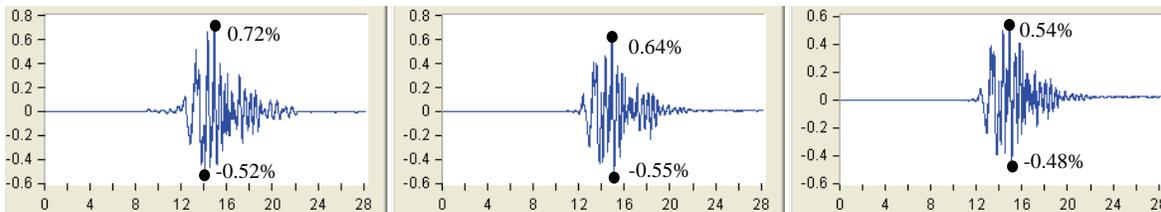
Inter-Story Drift (%)



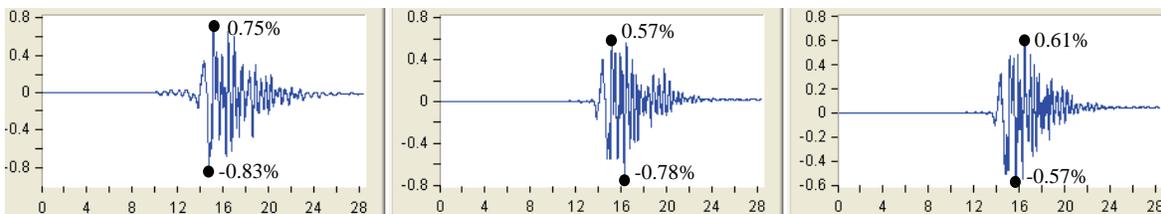
Inter-story Drift Time History of Test PER3



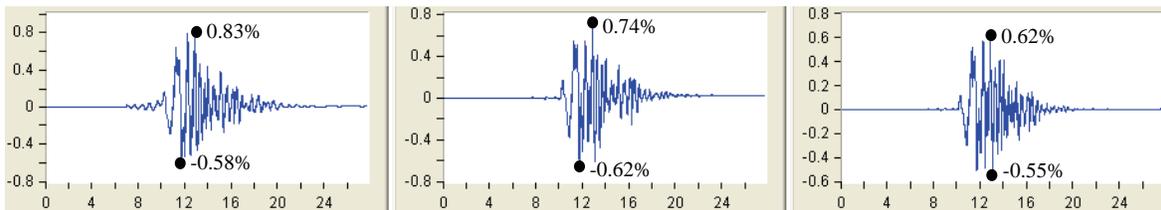
Inter-story Drift Time History of Test PE4



Inter-story Drift Time History of Test PER4

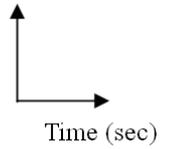


Inter-story Drift Time History of Test PE5A



Inter-story Drift Time History of Test PER5

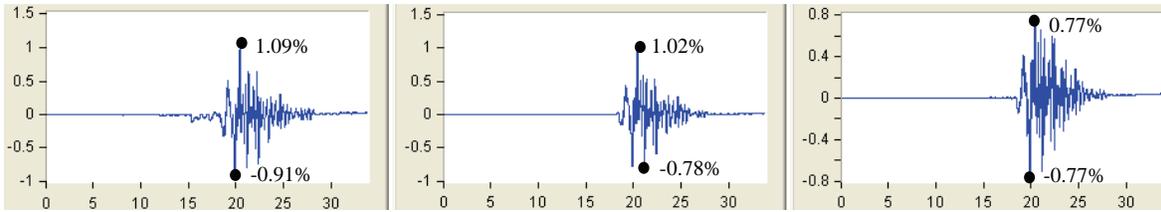
Inter-Story Drift (%)



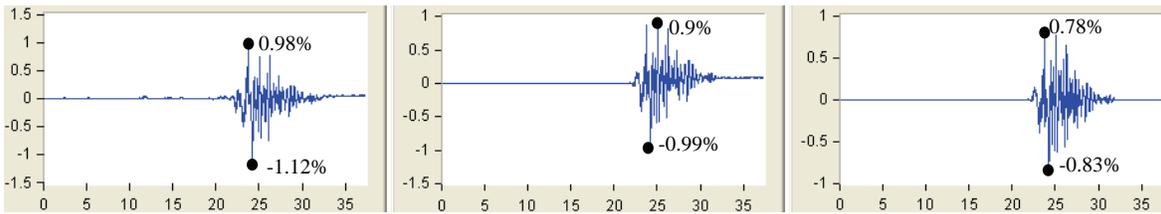
1<sup>st</sup> floor

2<sup>nd</sup> floor

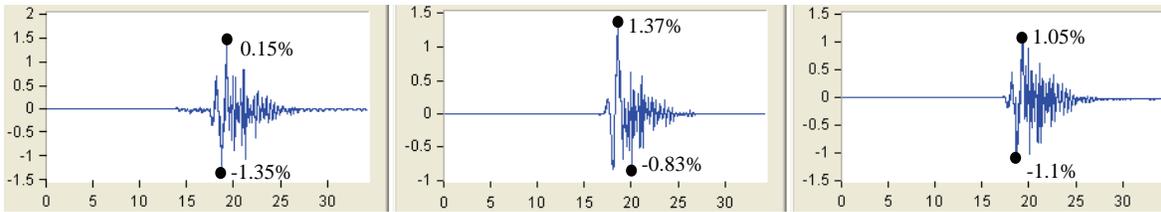
3<sup>rd</sup> floor



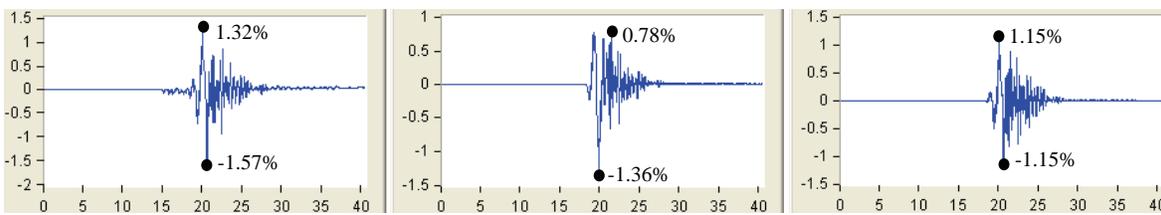
Inter-story Drift Time History of Test PE6A



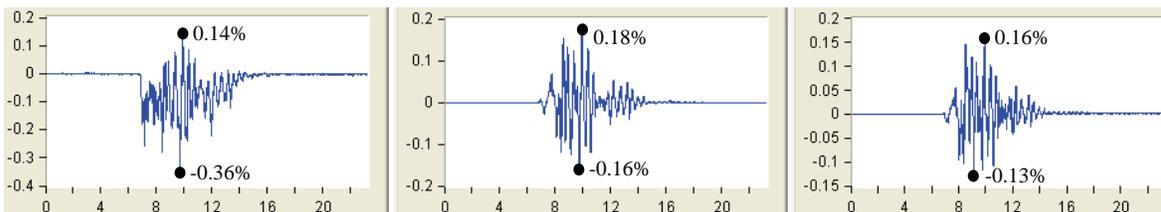
Inter-story Drift Time History of Test PER6



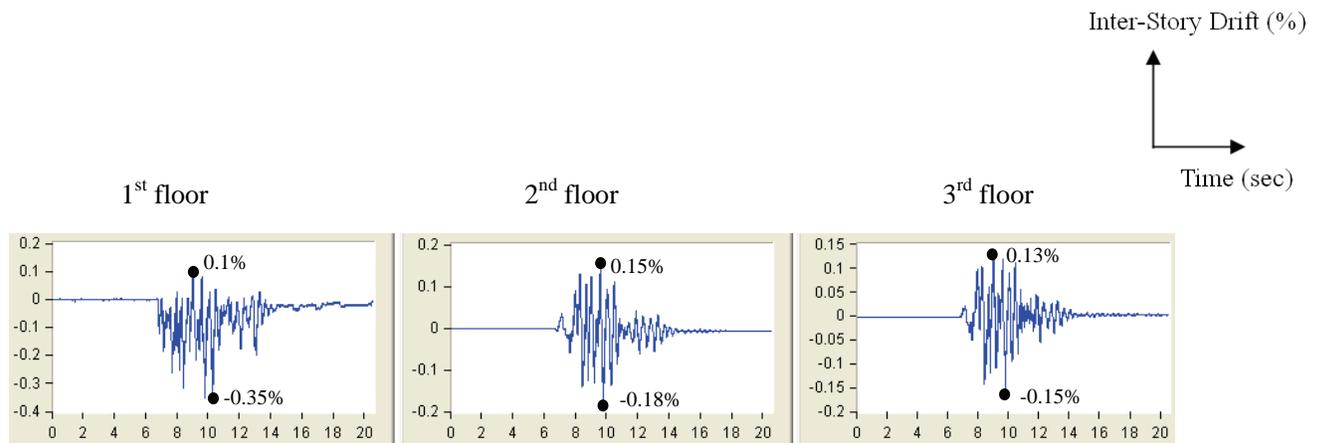
Inter-story Drift Time History of Test PE7A



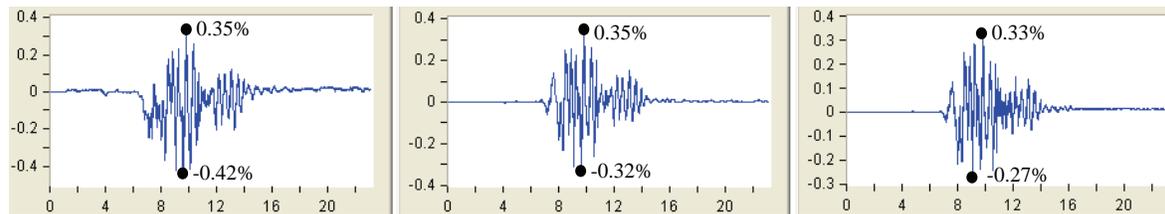
Inter-story Drift Time History of Test PER7



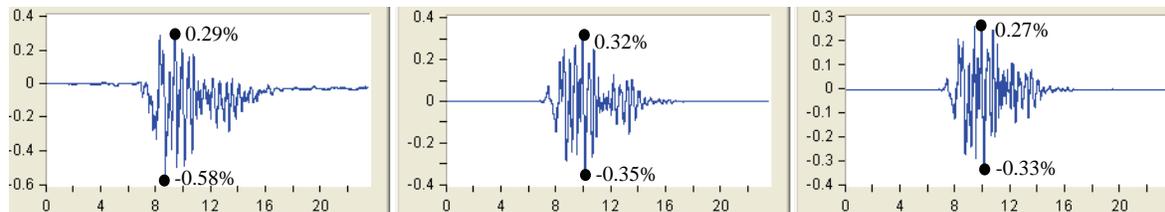
Inter-story Drift Time History of Test PE8



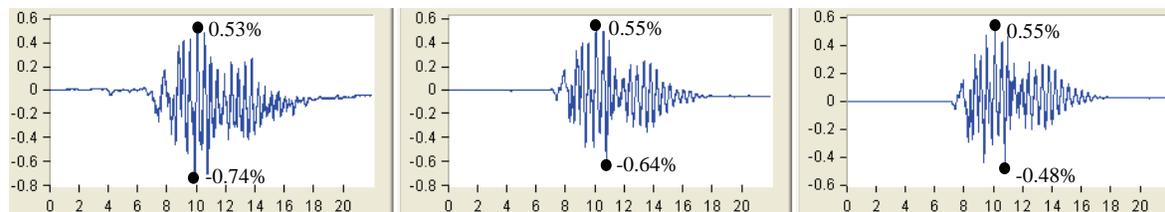
Inter-story Drift Time History of Test PER8



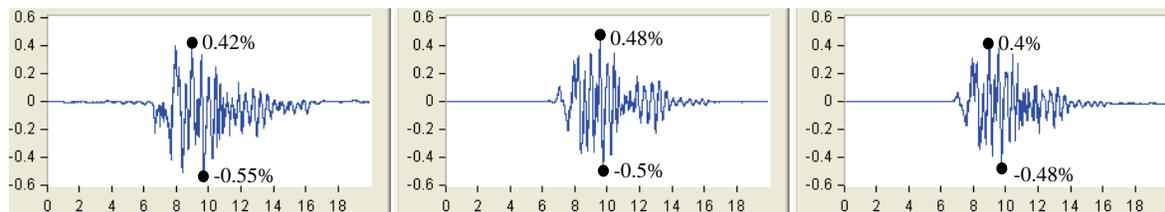
Inter-story Drift Time History of Test PE9



Inter-story Drift Time History of Test PER9

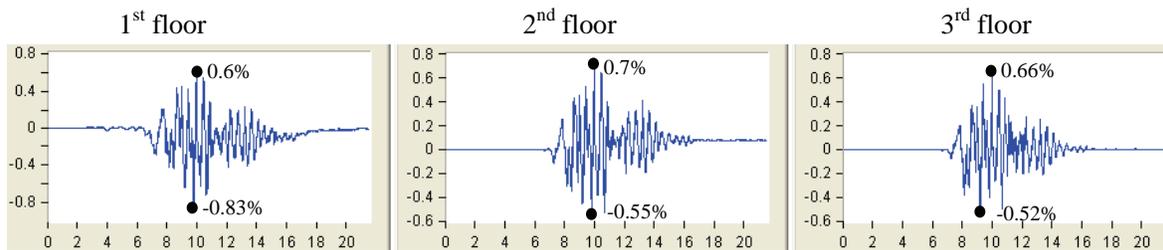
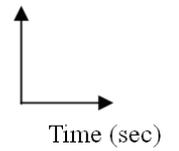


Inter-story Drift Time History of Test PE10

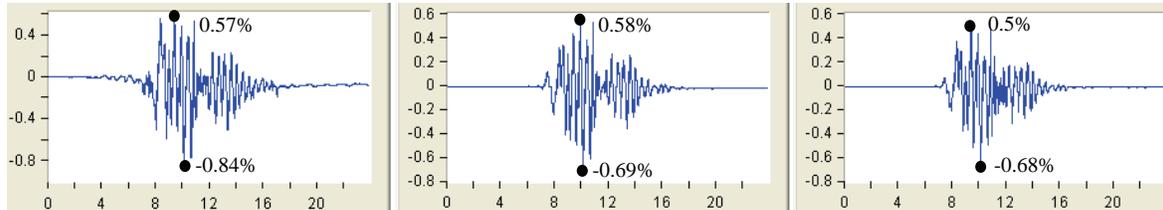


Inter-story Drift Time History of Test PER10

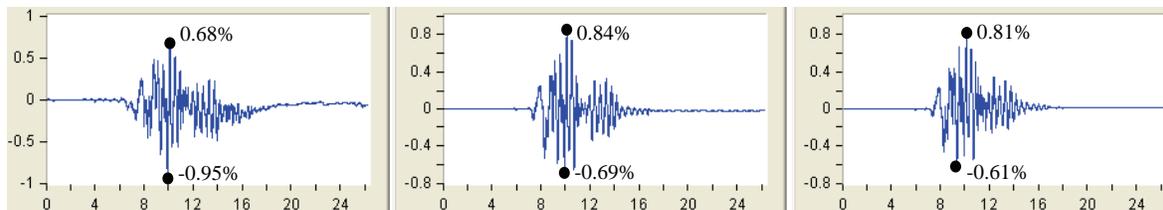
Inter-Story Drift (%)



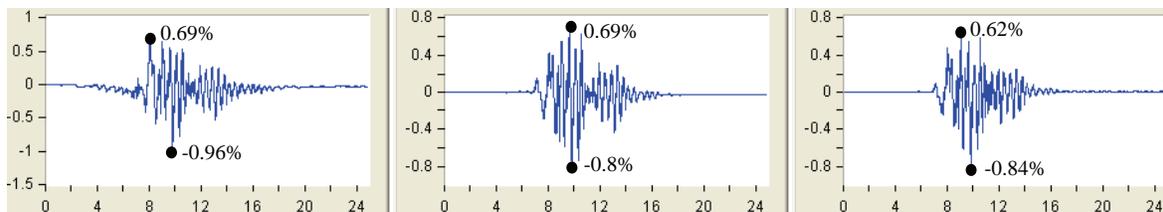
Inter-story Drift Time History of Test PE11



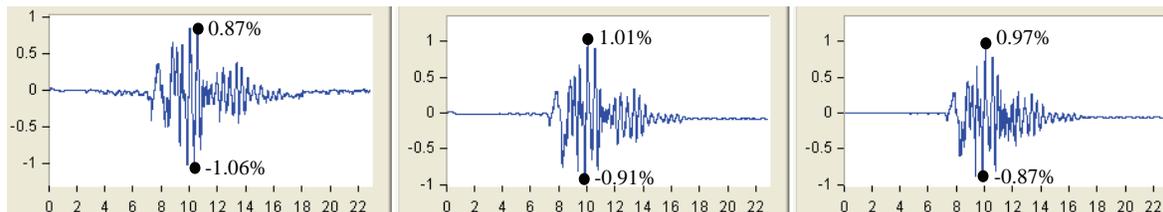
Inter-story Drift Time History of Test PER11



Inter-story Drift Time History of Test PE12

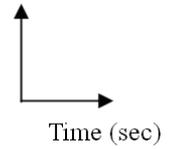


Inter-story Drift Time History of Test PER12



Inter-story Drift Time History of Test PE15

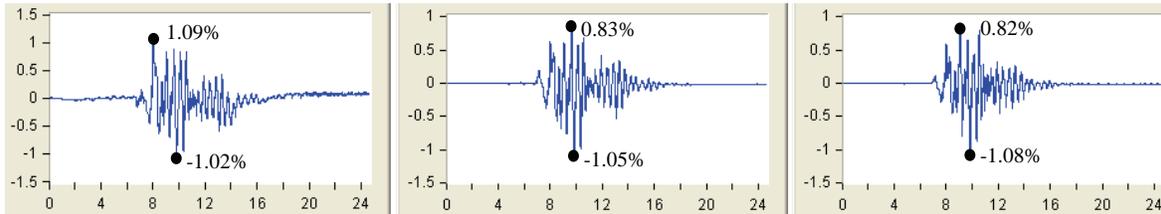
Inter-Story Drift (%)



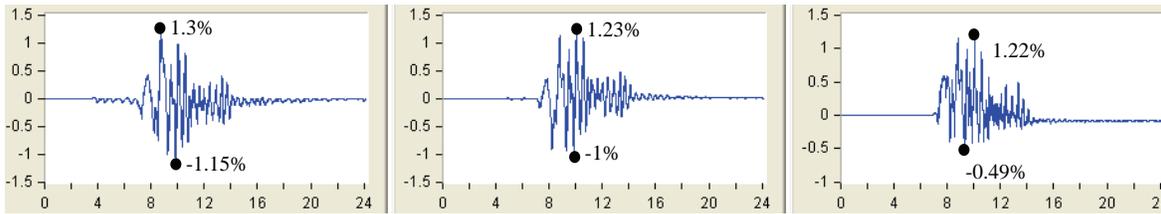
1<sup>st</sup> floor

2<sup>nd</sup> floor

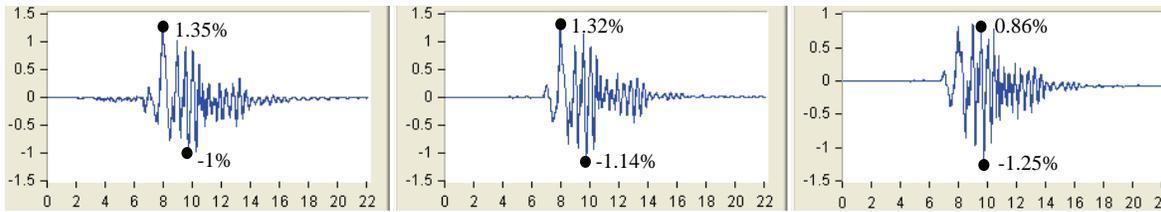
3<sup>rd</sup> floor



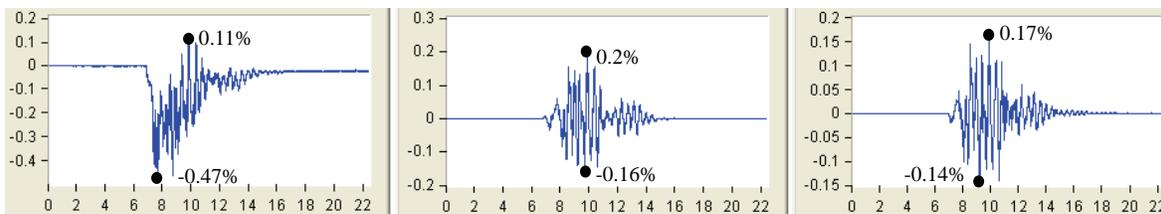
Inter-story Drift Time History of Test PER15



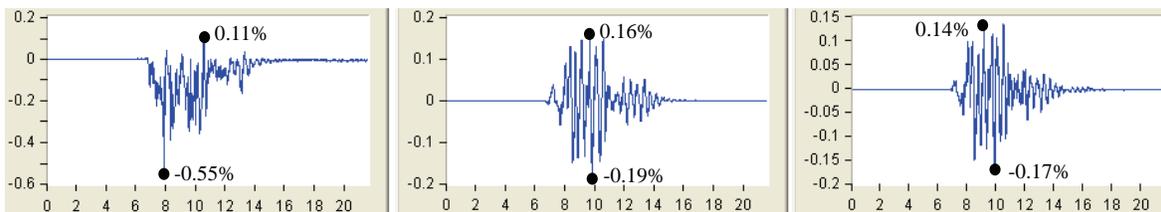
Inter-story Drift Time History of Test PE16



Inter-story Drift Time History of Test PER16

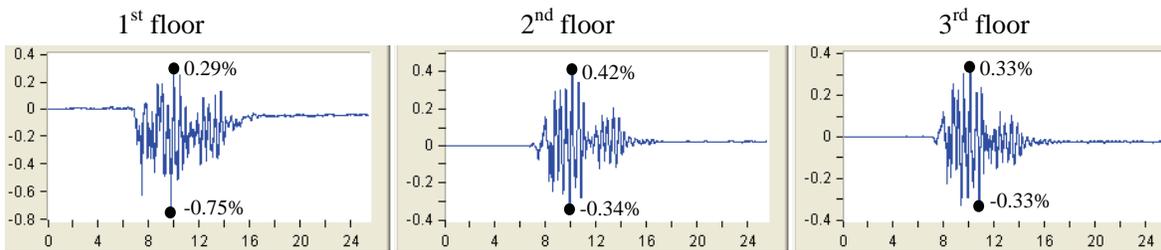
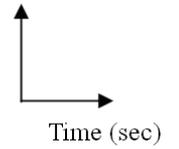


Inter-story Drift Time History of Test PE17

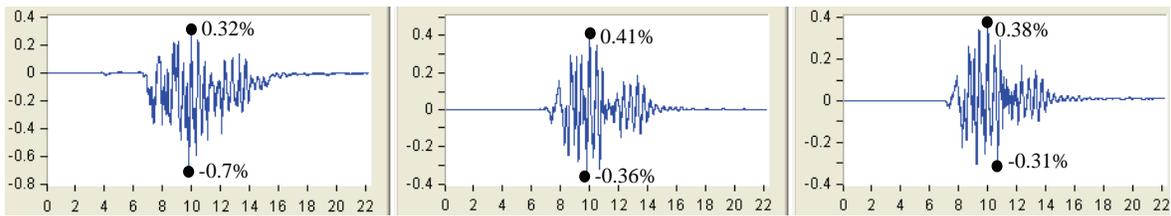


Inter-story Drift Time History of Test PER17

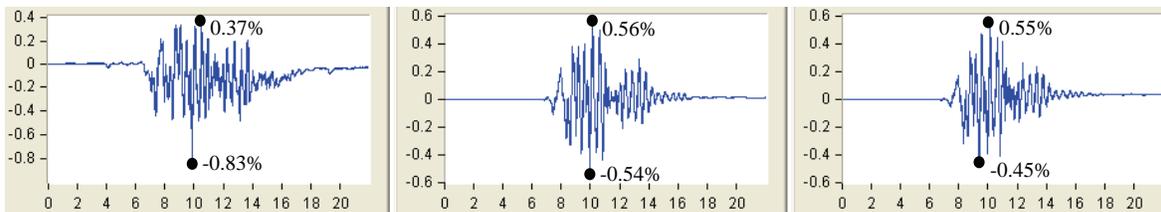
Inter-Story Drift (%)



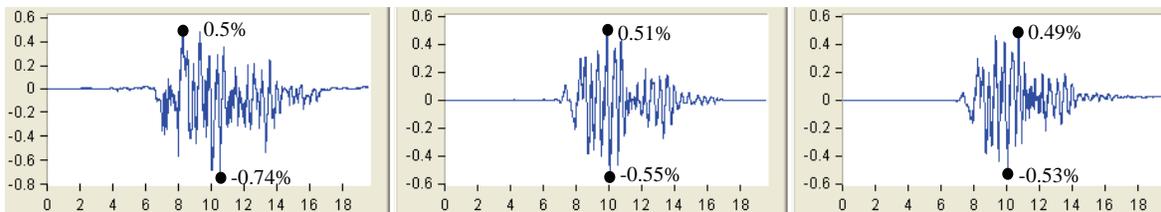
Inter-story Drift Time History of Test PE18



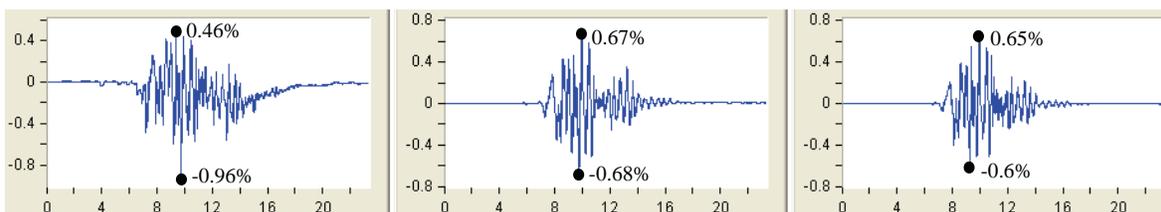
Inter-story Drift Time History of Test PER18



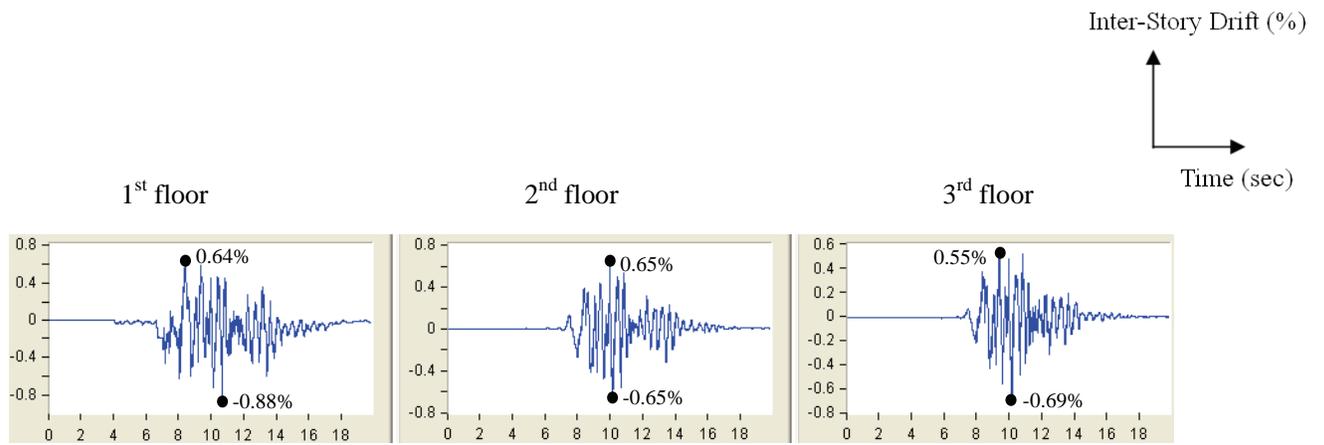
Inter-story Drift Time History of Test PE19



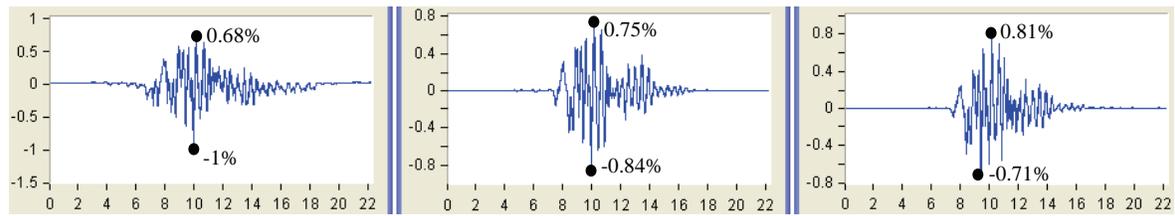
Inter-story Drift Time History of Test PER19



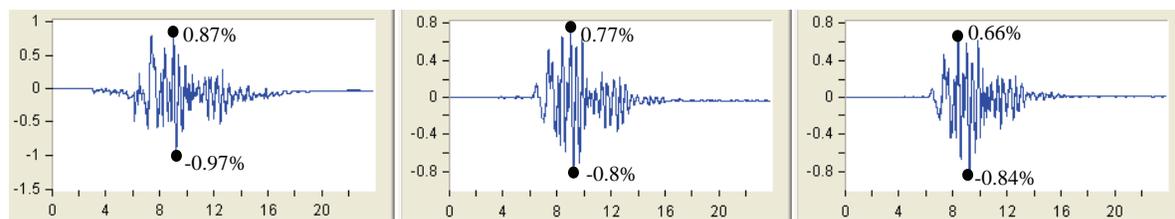
Inter-story Drift Time History of Test PE20



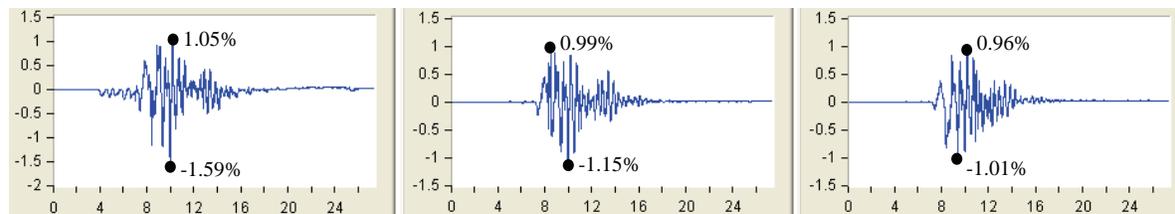
Inter-story Drift Time History of Test PER20



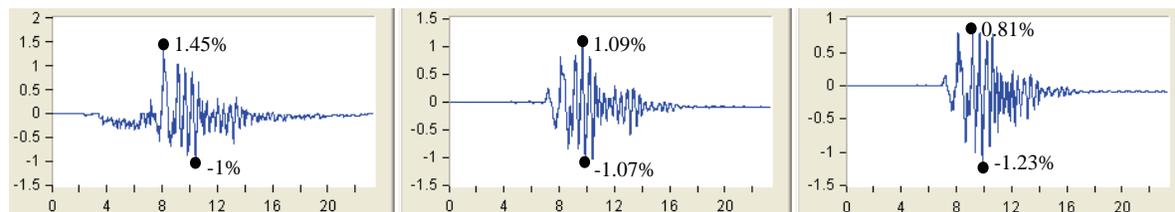
Inter-story Drift Time History of Test PE21



Inter-story Drift Time History of Test PER21

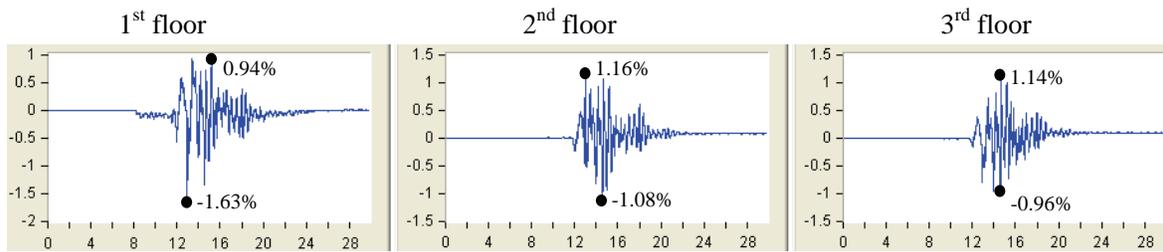
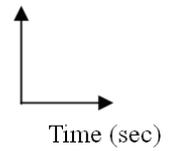


Inter-story Drift Time History of Test PE22

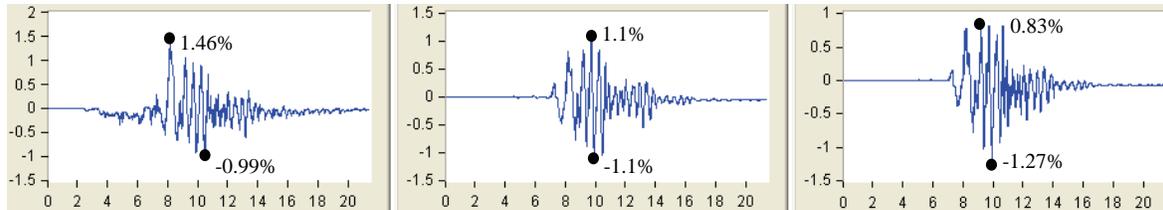


Inter-story Drift Time History of Test PER22

Inter-Story Drift (%)

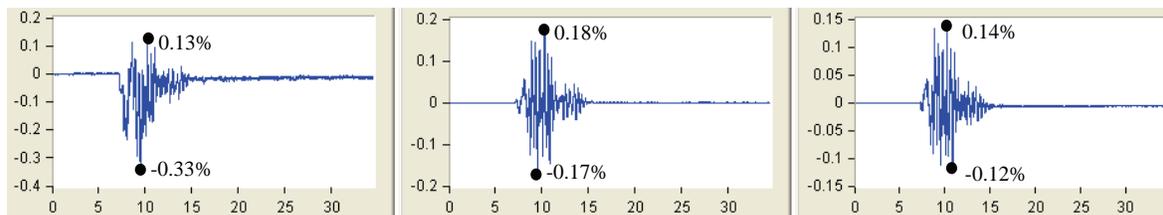


Inter-story Drift Time History of Test PE23

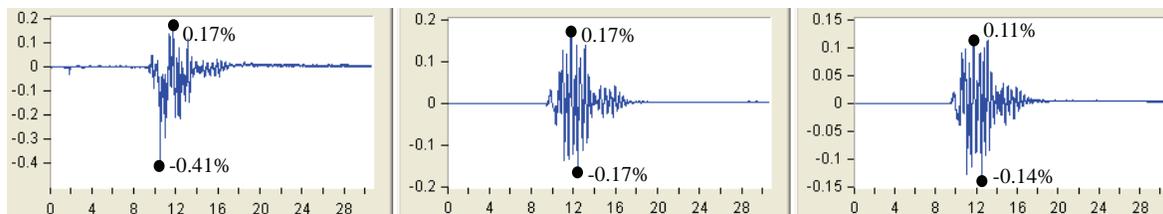


Inter-story Drift Time History of Test PER23

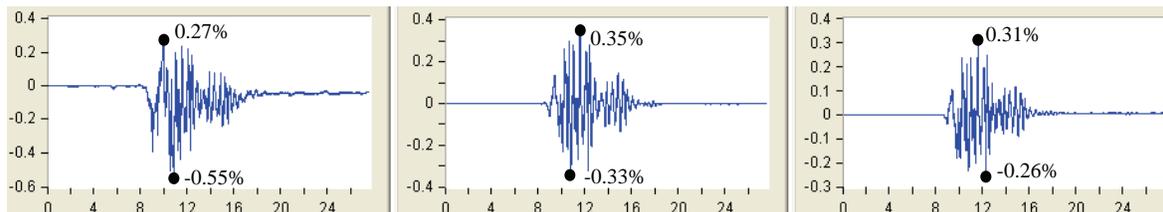
### SMRF



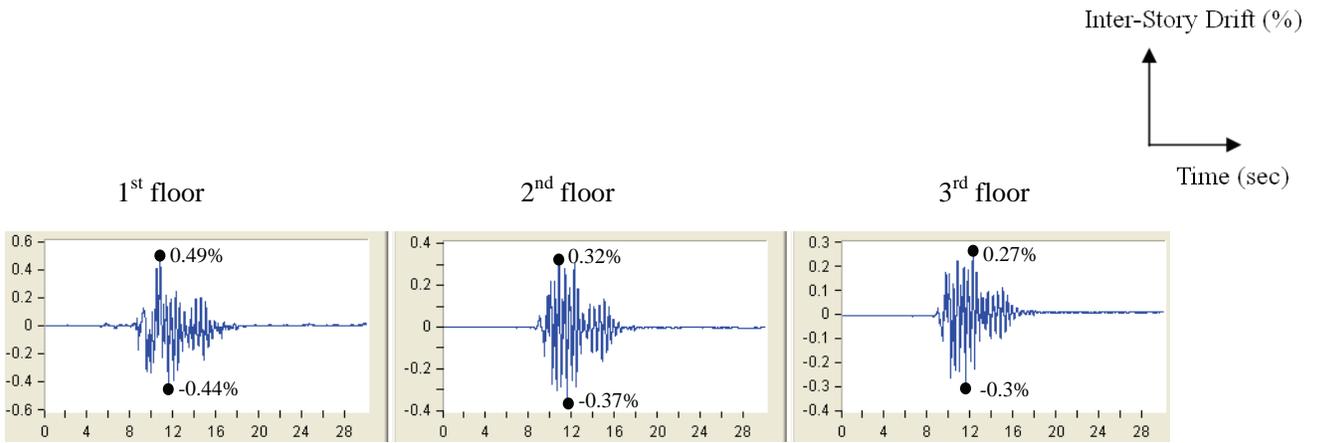
Inter-story Drift Time History of Test ME01



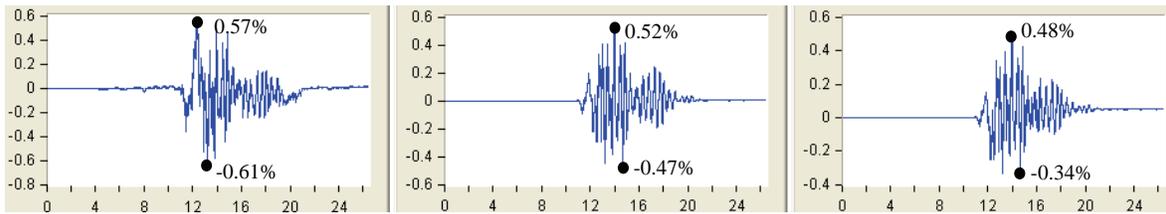
Inter-story Drift Time History of Test MER01



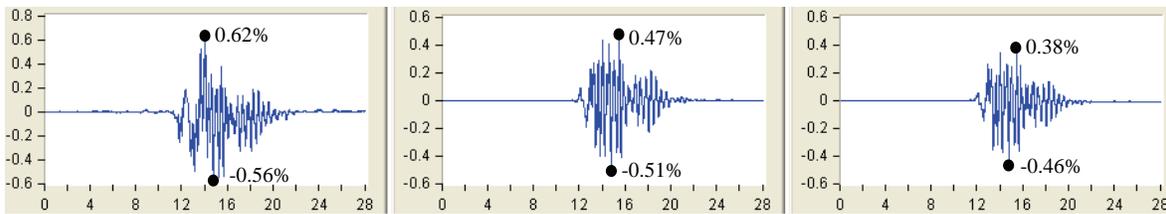
Inter-story Drift Time History of Test ME02



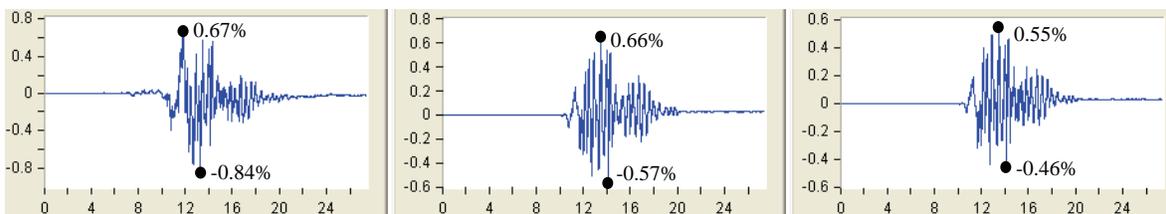
Inter-story Drift Time History of Test MER02A



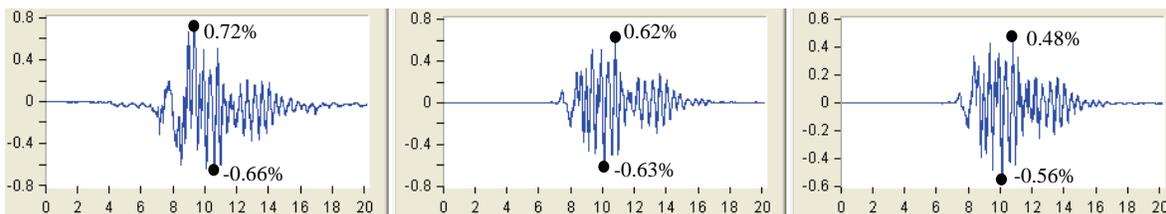
Inter-story Drift Time History of Test ME03



Inter-story Drift Time History of Test MER03

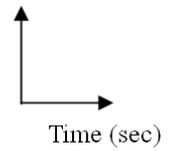


Inter-story Drift Time History of Test ME04



Inter-story Drift Time History of Test MER04

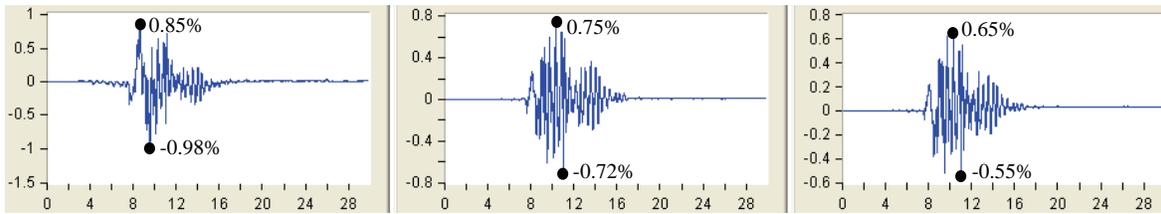
Inter-Story Drift (%)



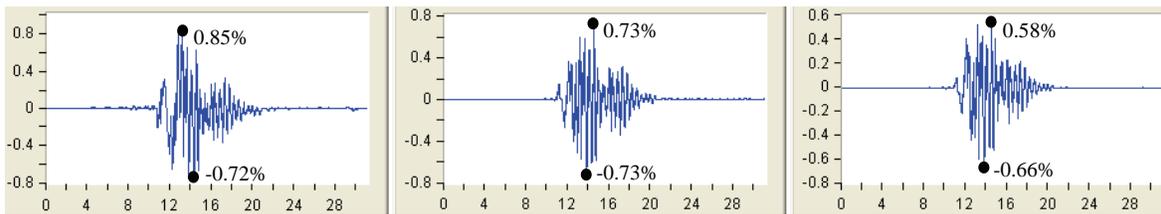
1<sup>st</sup> floor

2<sup>nd</sup> floor

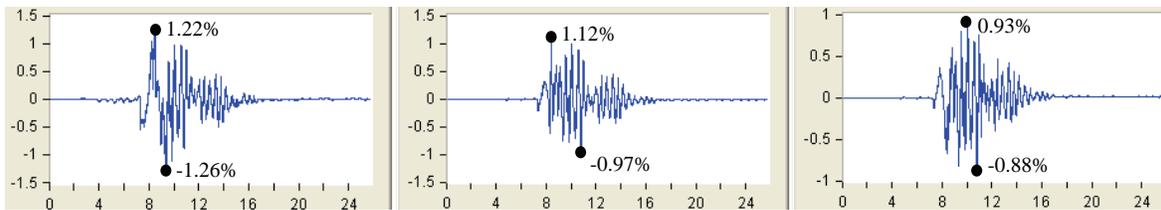
3<sup>rd</sup> floor



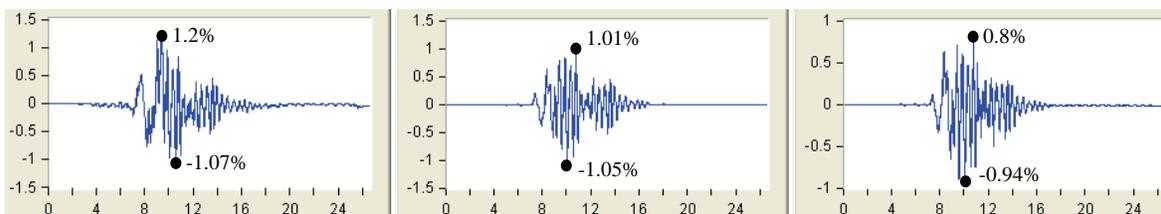
Inter-story Drift Time History of Test ME05



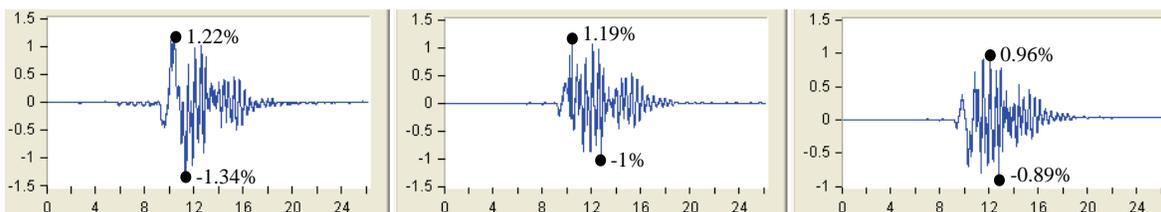
Inter-story Drift Time History of Test MER05



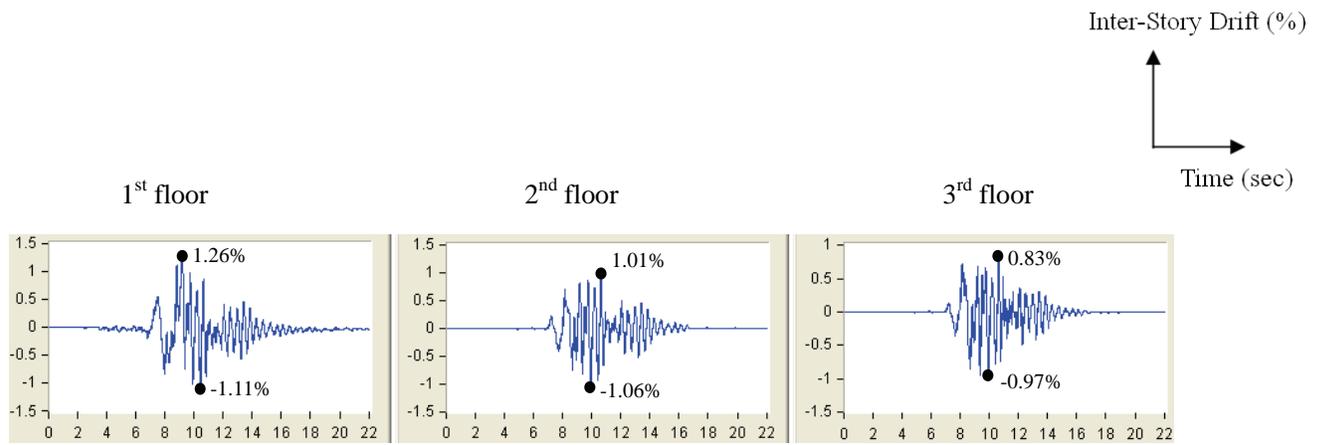
Inter-story Drift Time History of Test ME06A



Inter-story Drift Time History of Test MER06



Inter-story Drift Time History of Test ME07

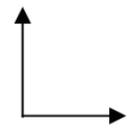


Inter-story Drift Time History of Test MER07



**Appendix G Strain Time History of the SCPT and SMRF  
Models for 125% and 150%-Intensity Seismic Tests**

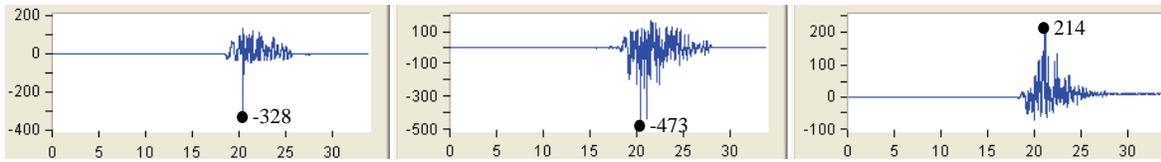
Strain ( $10^{-6}$ )



Time (sec)

Note: In the following figures, the unit of strain is  $10^{-6}$ .

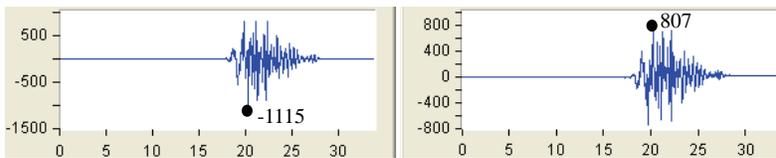
### SCPT



Beam in 1<sup>st</sup> floor

Beam in 2<sup>nd</sup> floor

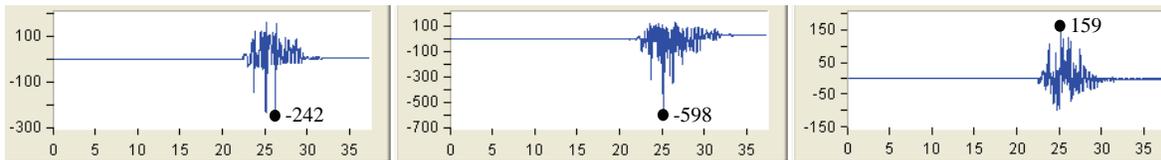
Beam in 3<sup>rd</sup> floor



Left Exterior Base Column

Center Base Column

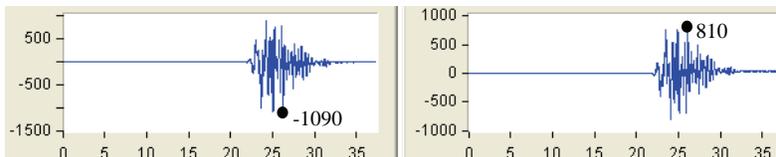
Strains Time History for Test PE6A (125%)



Beam in 1<sup>st</sup> floor

Beam in 2<sup>nd</sup> floor

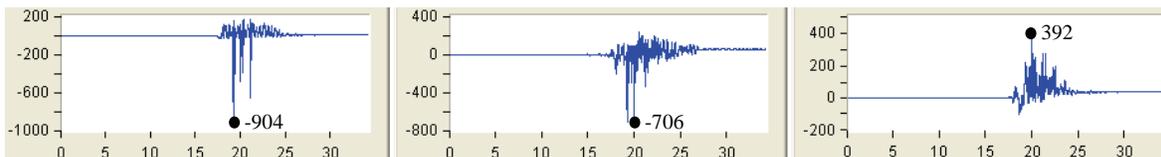
Beam in 3<sup>rd</sup> floor



Left Exterior Base Column

Center Base Column

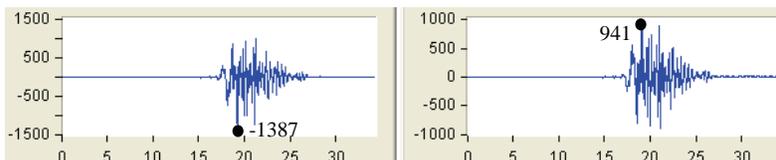
Strains Time History for Test PER6 (-125%)



Beam in 1<sup>st</sup> floor

Beam in 2<sup>nd</sup> floor

Beam in 3<sup>rd</sup> floor



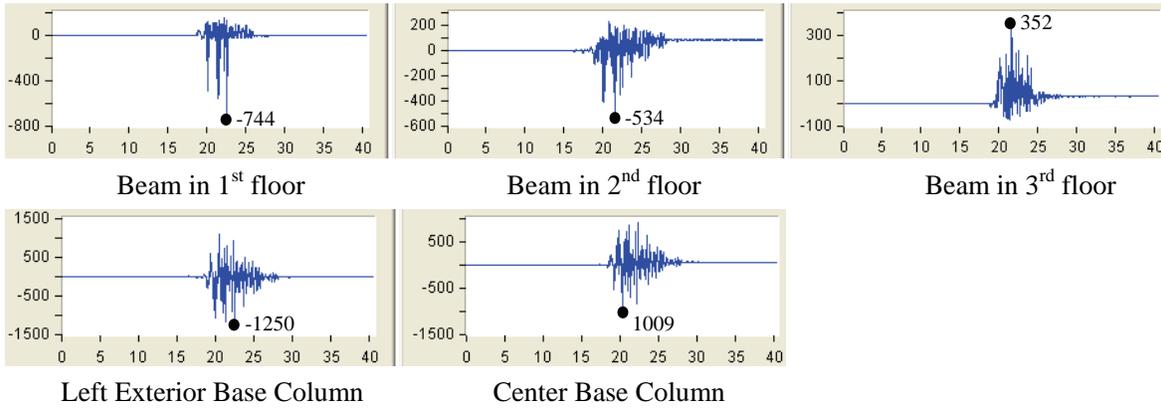
Left Exterior Base Column

Center Base Column

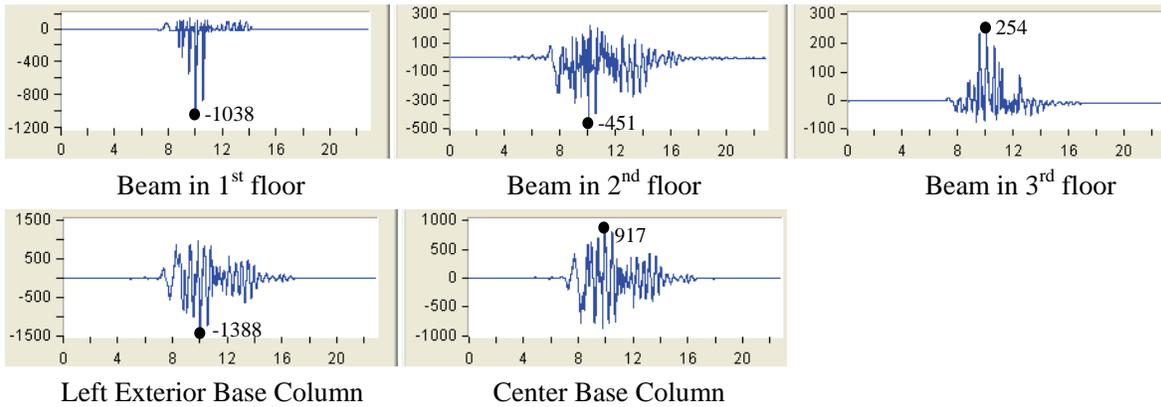
Strains Time History for Test PE7A (150%)

Strain ( $10^{-6}$ )

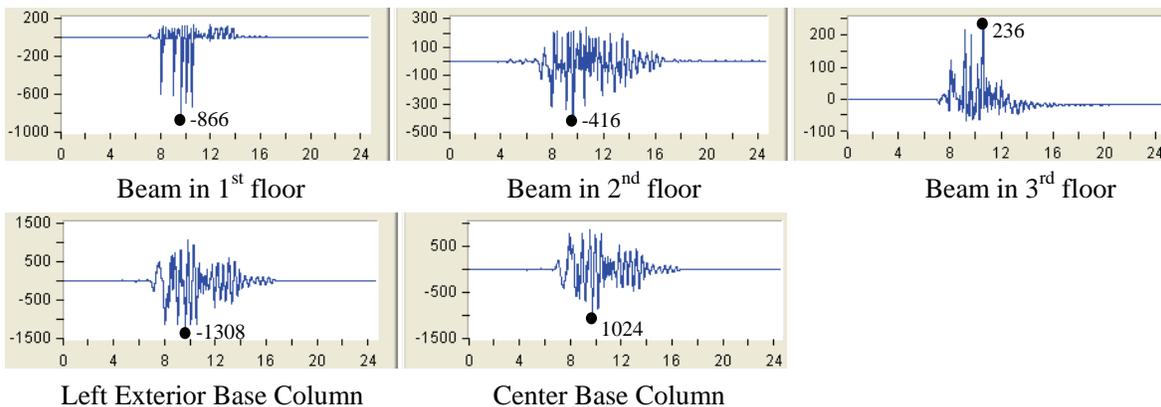
Time (sec)



Strains Time History for Test PER7 (-150%)



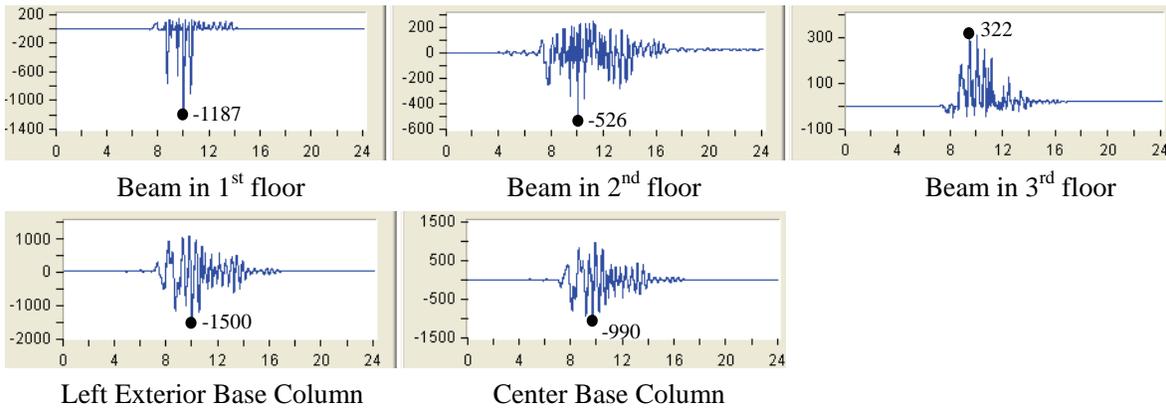
Strains Time History for Test PE15 (125%)



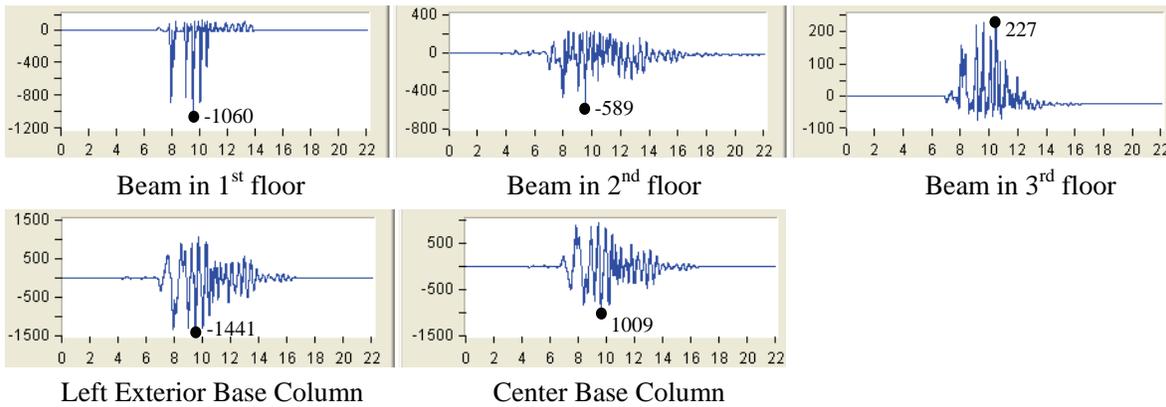
Strains Time History for Test PER15 (-125%)

Strain ( $10^{-6}$ )

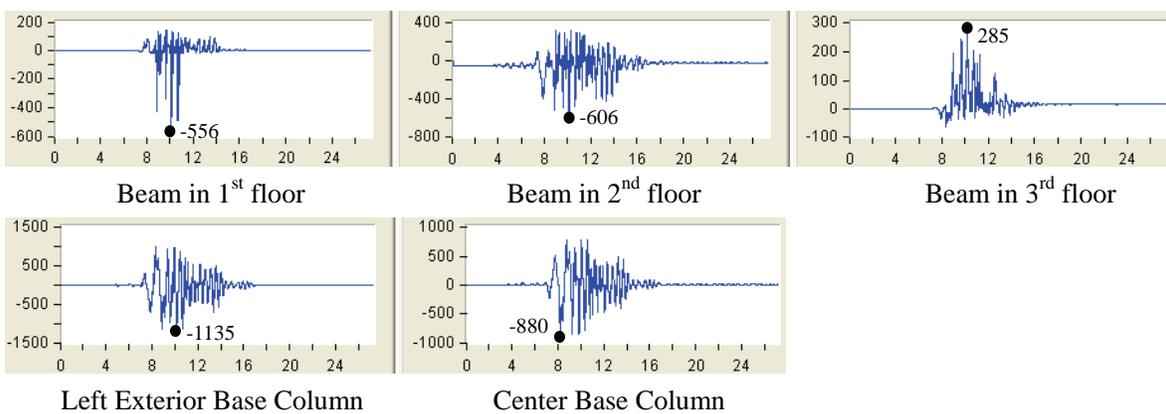
Time (sec)



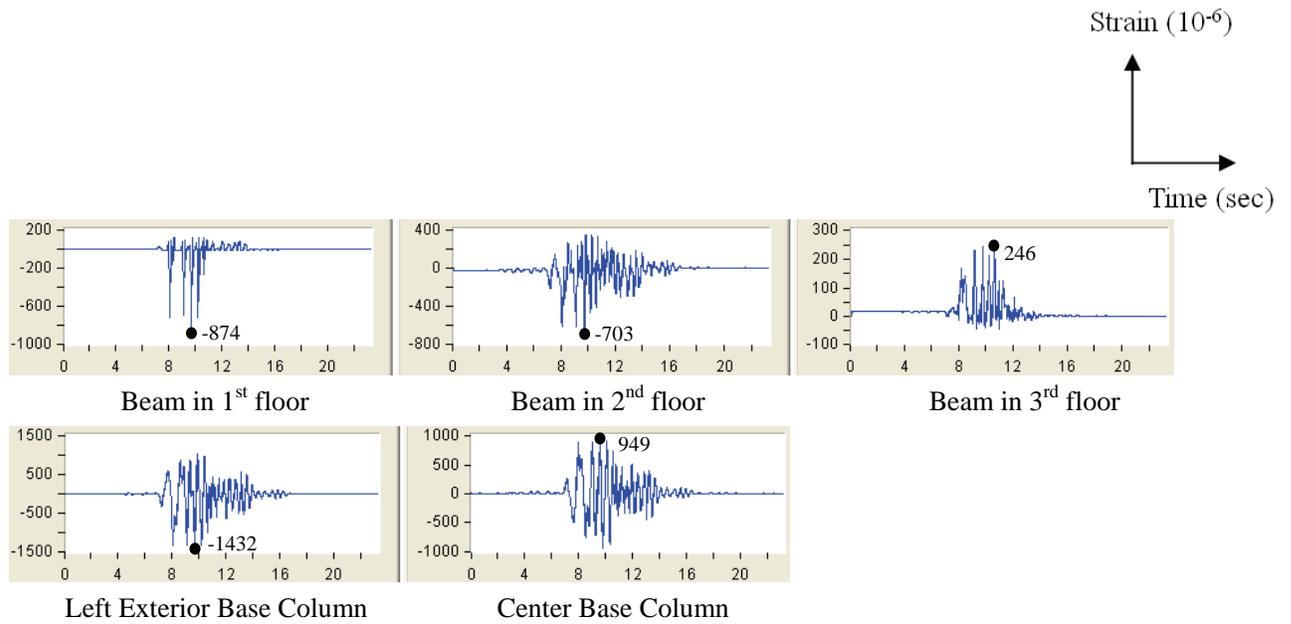
Strains Time History for Test PE16 (150%)



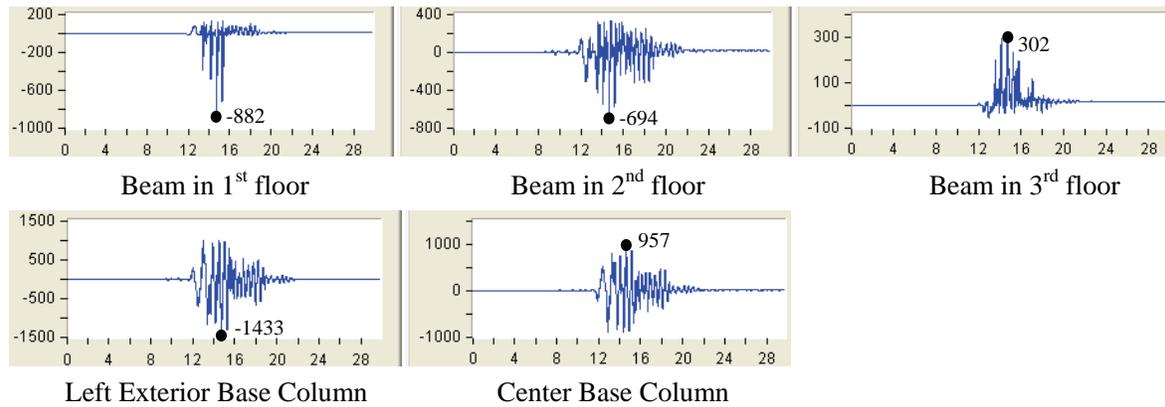
Strains Time History for Test PER16 (-150%)



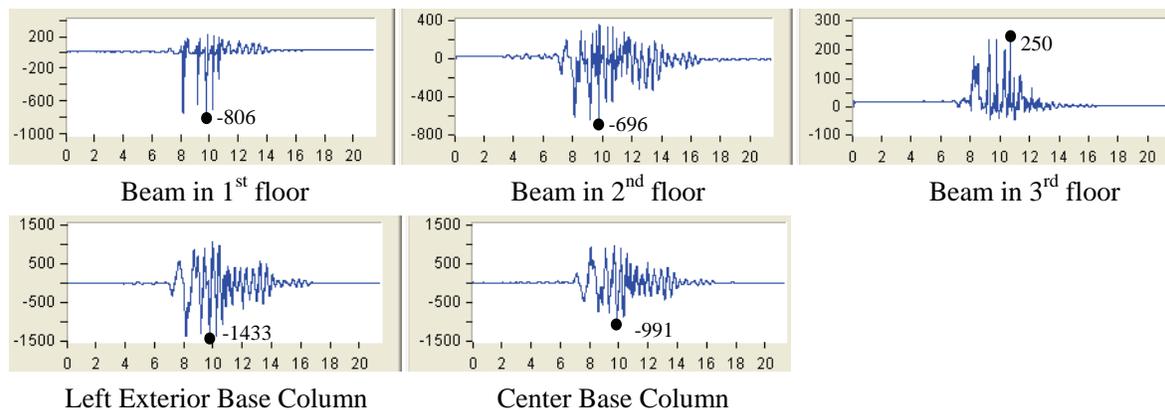
Strains Time History for Test PE22 (125%)



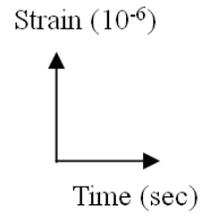
Strains Time History for Test PER22 (-125%)



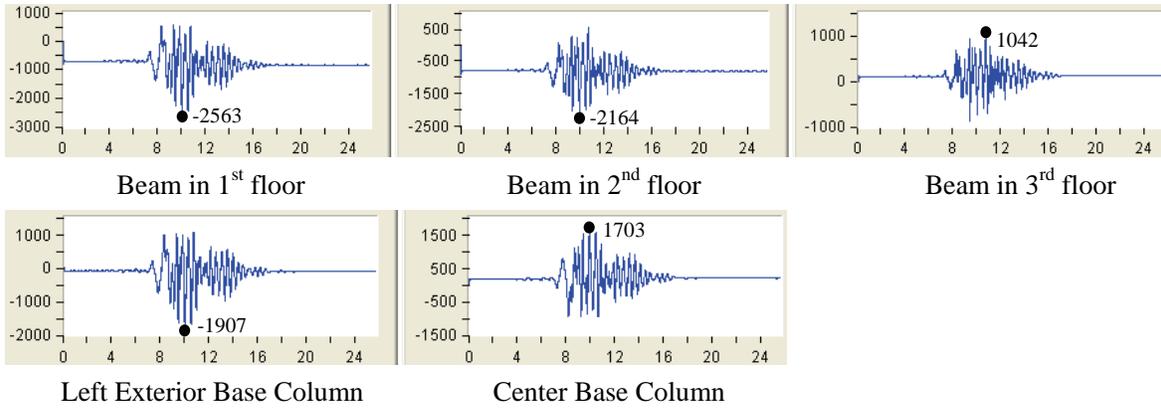
Strains Time History for Test PE23 (150%)



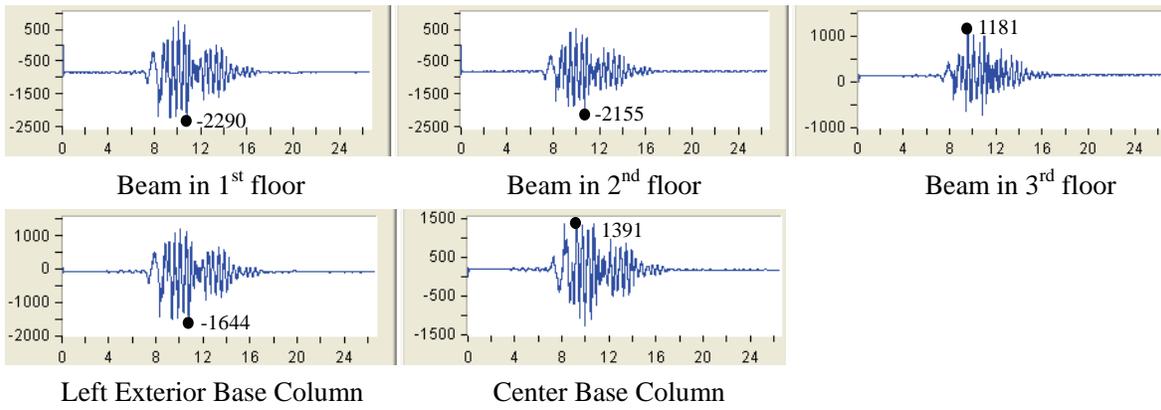
Strains Time History for Test PER23 (-150%)



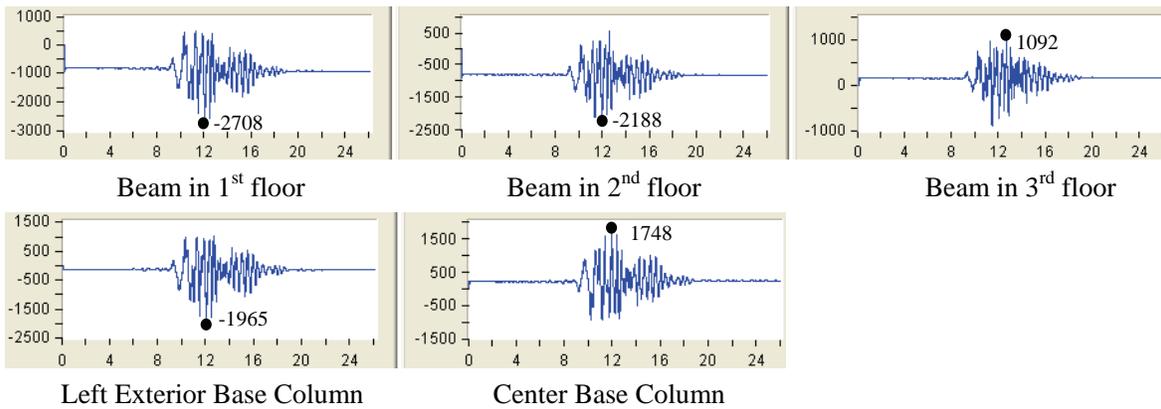
**SMRF**



Strains Time History for Test ME06A (125%)



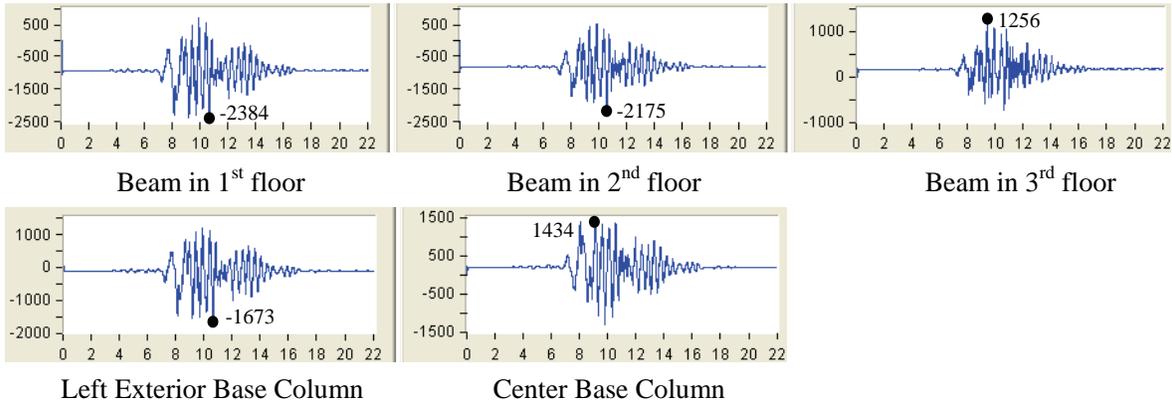
Strains Time History for Test MER06 (-125%)



Strains Time History for Test ME07 (150%)

Strain ( $10^{-6}$ )

Time (sec)



Strains Time History for Test MER07 (-150%)



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